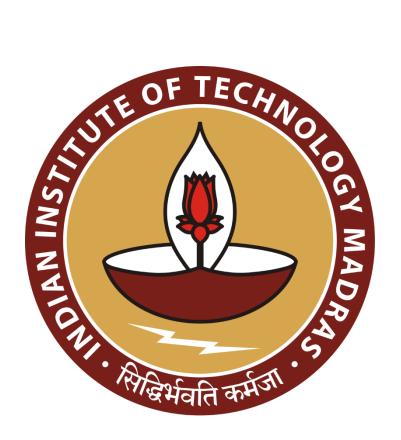
# Report on Finite Element Methods



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ME5204: Finite Element Methods

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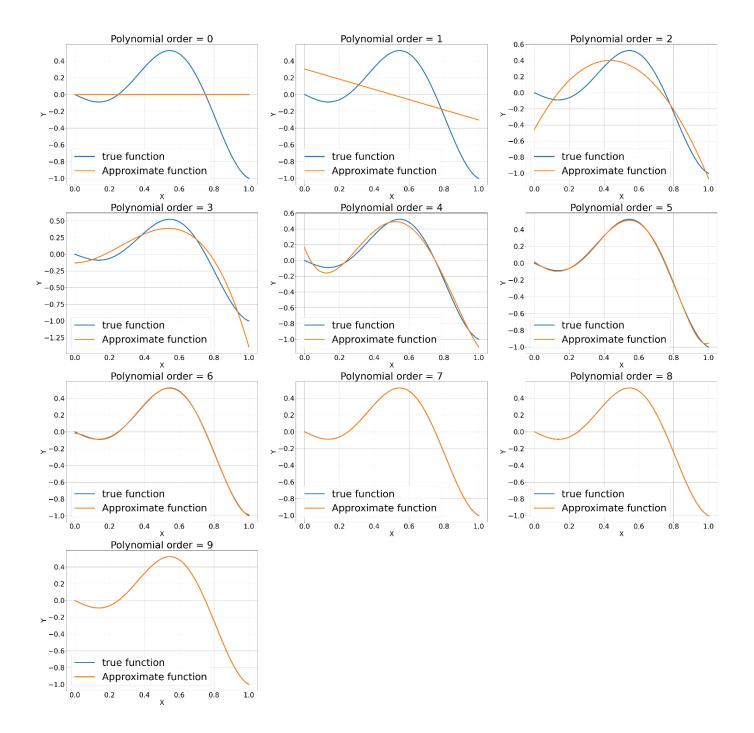
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## Assignment 01

Code

```
#Import libraries
import numpy as np
from scipy import integrate
import matplotlib.pyplot as plt
#define the variables
11 = 0
12 = 1
omega = 2
poly orders = np.arange(1,11)
graph per row = 3 #number of plots per row define by user
row = round((len(poly orders)/graph per row))+1 #total numbers
of rows
col = graph per row #total numbers of columns
fig = plt.figure(figsize=(60,60))
for poly order in poly orders:
    #define the functions
    u = lambda x:-x*np.cos(omega*np.pi*x)
    basis = np.array([lambda x,n=i:x**n for i in
range(poly order)])
    #define the required variables to store values
    k = np.empty((poly order,poly order))
    c = np.empty((poly order))
    f = np.empty((poly order))
    #filling of the required variables
    for i in range (poly order):
        function f = lambda x,i=i:u(x)*basis[i](x)
        f[i] = integrate.quad(function f,11,12)[0]
        for j in range (poly order):
            function k = lambda
x, i=i, j=j:basis[i](x)*basis[j](x)
            k[i,j] = integrate.quad(function k,11,12)[0]
    #solve the governing equation > {c} = inverse([K])@{f}
    c = np.linalg.solve(k,f)
    #define the apporximate function having polynomical basis
    w star = lambda x:sum(c[i]*basis[i](x) for i in
range(poly order))
    #range of X
    x = np.linspace(11, 12, 500)
    plt.subplot(row,col,poly order)
    plt.grid()
    plt.plot(x,u(x),linewidth = 5)
    plt.plot(x,w star(x), linewidth = 5)
```

```
plt.title("Polynomial order = {}".format(poly_order-
1),fontsize=40)
    plt.xlabel("X",fontsize=35)
    plt.ylabel("Y",fontsize=35)
    plt.xticks(fontsize=25)
    plt.yticks(fontsize=25)
    plt.legend(["true function","Approximate
function"],fontsize=30)
plt.savefig('./Assignment01.png')
plt.show()
```

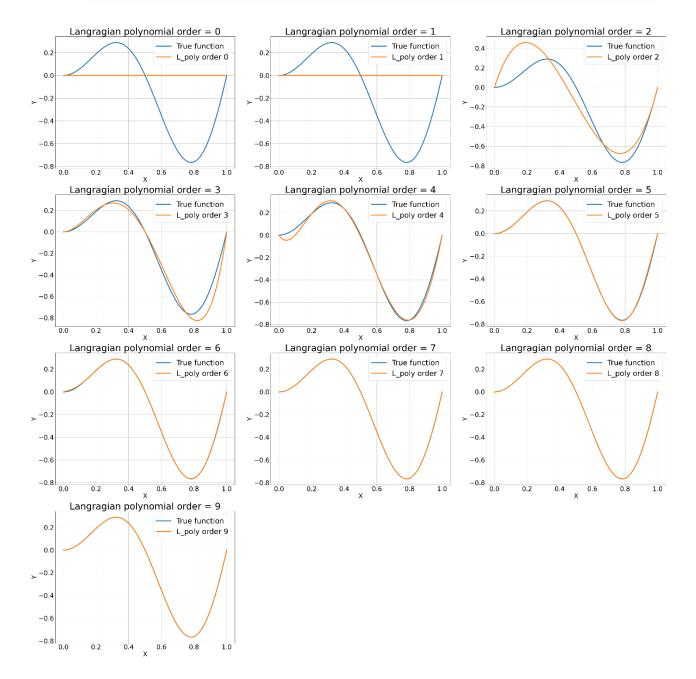


## Assignment 02

Code

```
#Import libraries
import numpy as np
from scipy import integrate
import matplotlib.pyplot as plt
#define the variables
11 = 0
12 = 1
omega = 2
#poly order = 4 #if polynomial order is 1 then total number of
nodes required is 1+1=2
poly orders = np.arange(1,11)
graph per row = 3 #number of plots per row define by user
row = round((len(poly_orders)/graph_per_row))+1 #total numbers
of rows
col = graph per row #total numbers of columns
fig = plt.figure(figsize=(60,60))
for poly order in poly orders:
    #define the original function
    u = lambda x:x*np.sin(np.pi*omega*x)
    nodes = np.linspace(11,12,poly order+1)
    x = np.linspace(11, 12, 100)
    #define the generation of langragian basis
    def langragian basis(x,nodes,i):
        result1 = np.ones like(x)
        for j in range(len(nodes)):
            if j!=i:
                result1*=(x-nodes[j])/(nodes[i]-nodes[j])
        return result1
    #define the Summation of product of langragian basis and
displacement for all the nodes are present
    def langragian poly(x,nodes):
        result2 = np.zeros_like(x)
        for i in range (poly order):
            result2 += u(nodes[i]) *langragian basis(x,nodes,i)
        return result2
    #approximate langragian function
    w star = langragian poly(x,nodes)
    plt.subplot(row,col,poly order)
    plt.grid()
    plt.plot(x,u(x),linewidth = 5)
    plt.plot(x,w star,linewidth = 5)
    plt.title("Langragian polynomial order =
{}".format(poly_order-1),fontsize=40)
    plt.xlabel("X",fontsize=35)
```

```
plt.ylabel("Y", fontsize=35)
  plt.xticks(fontsize=25)
  plt.yticks(fontsize=25)
  plt.legend(["True function", "Langragian polynomical order
{}".format(poly_order-1)], fontsize=30)
plt.savefig('./Assignment02.png')
plt.show()
```



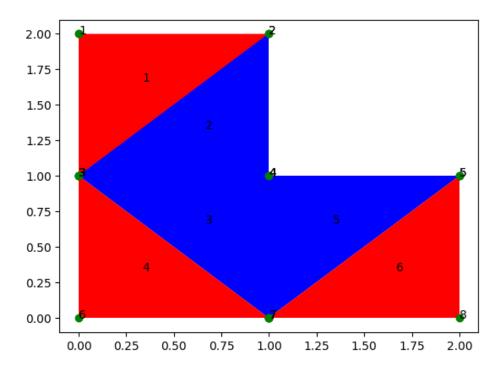
## Assignment 03

#### Code

```
import numpy as np
import matplotlib.pyplot as plt
import copy
#load the required data
coord = np.loadtxt("./Coord.txt", dtype="float32", comments="#",
NCA = np.loadtxt("./NCA.txt", dtype="int16", comments="#",
ndmin=2)
mat data = np.loadtxt("./MAT DATA.txt", dtype="float32",
comments="#", ndmin=2)
print(coord)
print (NCA)
#define problem type
#problem type = input("Press 11 > \n press 12 > \n press 13 \n
press 21 > plane stress \n press 22 > plane strain n press 3 >
for 3-D")
problem type = 21
thickness = 1
#verify the given data
Num nodes = coord.shape[0]-1
Num elements = NCA.shape[0]-1
Num mats = mat data.shape[0]-1
print("Number of node = {}, Number of elements = {} and Number
of different material =
{}".format(Num nodes, Num elements, Num mats))
for elem idx in range(1,Num elements+1):
    N1 = NCA[elem idx][1]
    N2 = NCA[elem_idx][2]
    N3 = NCA[elem idx][3]
    X1N1 = coord[N1][1]
    X1N2 = coord[N2][1]
    X1N3 = coord[N3][1]
    X2N1 = coord[N1][2]
    X2N2 = coord[N2][2]
    X2N3 = coord[N3][2]
    X1 = [X1N1, X1N2, X1N3, X1N1]
    X2 = [X2N1, X2N2, X2N3, X2N1]
    if NCA[elem idx][4] == 1:
        plt.fill(X1,X2,facecolor = "red")
    elif NCA[elem idx][4] == 2:
        plt.fill(X1,X2,facecolor = "blue")
    X1CG = (X1N1+X1N2+X1N3)/3.0
    X2CG = (X2N1+X2N2+X2N3)/3.0
```

```
plt.scatter(X1,X2,color = "green")
           plt.text(X1CG,X2CG,elem idx)
           plt.text(X1N1,X2N1,N1)
           plt.text(X1N2,X2N2,N2)
           plt.text(X1N3,X2N3,N3)
     mat data
[[0. 0. 0.]
[1. 0. 2.]
[2. 1. 2.]
[3. 0. 1.]
[4. 1. 1.]
[5. 2. 1.]
[6. 0. 0.]
[7. 1. 0.]
[8. 2. 0.]]
[[00000]]
[1 1 3 2 1]
[2 3 4 2 2]
[3 7 4 3 2]
[4 3 6 7 1]
[5 5 4 7 2]
[67851]]
Number of node = 8, Number of elements = 6 and Number of different material = 2
array([[0.e+00, 0.e+00, 0.e+00],
   [1.e+00, 1.e+11, 3.e-01],
```

[2.e+00, 2.e+11, 2.e-01]], dtype=float32)



```
#D-martix
D store = []
for elem idx in range(1,Num elements+1):
    mat num = NCA[elem idx][4]
    E = mat data[mat num][1]
    PR = mat data[mat num][2]
    if problem type == 21:
        #print("For plane stress condition, D matrix = ")
        const1 = E/(1-PR**2)
        D = np.multiply([[1,PR,0],[PR,1,0],[0,0,(1-
PR) /2]], const1)
    elif problem type == 22:
        #print("For plane strain condition, D martix = ")
        const2 = E/((1+PR)*(1-2*PR))
        D = np.multiply([[1-PR,PR,0],[PR,1-PR,0],[0,0,1-PR])
PR]],const2)
    else:
        print("Problem type is not defined. Please, check once
again.")
    print("D matrix for element \{\} = n\{\} \setminus n".format(elem idx,D))
    D store.append(D)
print(np.shape(D store))
#D matrix > There are only 2 types.
```

D matrix for element 1 =

[[1.09890109e+11 3.29670339e+10 0.00000000e+00]

[3.29670339e+10 1.09890109e+11 0.00000000e+00]

[0.00000000e+00 0.00000000e+00 3.84615373e+10]]

```
D matrix for element 2 =
[[2.08333329e+11 4.1666665e+10 0.00000000e+00]
[4.16666665e+10 2.08333329e+11 0.00000000e+00]
[0.00000000e+00 0.00000000e+00 8.33333314e+10]]
D matrix for element 3 =
[[2.08333329e+11 4.1666665e+10 0.00000000e+00]
[4.1666665e+10 2.08333329e+11 0.00000000e+00]
[0.00000000e+00 0.00000000e+00 8.33333314e+10]]
D matrix for element 4 =
[[1.09890109e+11 3.29670339e+10 0.00000000e+00]
[3.29670339e+10 1.09890109e+11 0.00000000e+00]
[0.00000000e+00 0.00000000e+00 3.84615373e+10]]
D matrix for element 5 =
[[2.08333329e+11 4.16666665e+10 0.00000000e+00]
[4.16666665e+10 2.08333329e+11 0.00000000e+00]
[0.00000000e+00 0.00000000e+00 8.33333314e+10]]
D matrix for element 6 =
[[1.09890109e+11 3.29670339e+10 0.00000000e+00]
[3.29670339e+10 1.09890109e+11 0.00000000e+00]
[0.00000000e+00 0.00000000e+00 3.84615373e+10]]
(6, 3, 3)
     DOF_PN = 2 #DOF per node = 2; X1 and X2
     Num nodes=8
     Total DOF = DOF PN*Num nodes
     #print("Total degrees of freedom = {}".format(Total DOF))
     GSTIFF = np.zeros([Total_DOF,Total_DOF])
```

```
#print("Global stifness matrix = \n{}\n".format(GSTIFF))
    #print("Force vector = {}".format(F))
    #B-matrix
    B store = []
    for elem idx in range(1,Num elements+1):
        N1 = NCA[elem idx, 1]
        N2 = NCA[elem idx, 2]
        N3 = NCA[elem idx, 3]
        X1N1 = coord[N1,1]
        X1N2 = coord[N2,1]
        X1N3 = coord[N3,1]
        X2N1 = coord[N1,2]
        X2N2 = coord[N2,2]
        X2N3 = coord[N3,2]
        two delta matrix =
    [[1,X1N1,X2N1],[1,X1N2,X2N2],[1,X1N3,X2N3]]
        two delta = np.linalg.det(two delta matrix)
        #print(two delta)
        b1 = x2n2-x2n3
        b2 = X2N3-X2N1
        b3 = X2N1-X2N2
        v1 = X1N3-X1N2
        v2 = X1N1-X1N3
        v3 = x1n2-x1n1
        B = [[b1,0,b2,0,b3,0],[0,v1,0,v2,0,v3],[v1,b1,v2,b2,v3,b3]]
        B store.append(B)
        #print(B)
        ESTIFF =
    np.matmul(np.matmul(np.transpose(B),D store[elem idx-
    1]),B)*(two delta/2)*thickness
        col idx = [[2*N1-2,2*N1-1,2*N2-2,2*N2-1,2*N3-2,2*N3-1]]
        row idx = np.transpose(col idx)
        GSTIFF[row idx,col idx] = GSTIFF[row idx,col idx]+ESTIFF
        #print(GSTIFF)
        #print(ESTIFF)
    #print(D)
    print(B)
    #print(np.shape(GSTIFF), np.linalg.det(GSTIFF))
    #print(np.shape(B store))
[[-1.0, 0, 1.0, 0, 0.0, 0], [0, 0.0, 0, -1.0, 0, 1.0], [0.0, -1.0, -1.0, 1.0, 1.0, 0.0]]
    load BC = np.loadtxt("./Load BC.txt", dtype="float32",
    comments="#", ndmin=2)
    #print(load BC)
    num_force_bc = load_BC.shape[0]
    F = np.zeros([Total DOF])
```

```
for idx in range(1, num force bc):
        node num = int(load BC[idx,0])
         load type = load BC[idx,1]
         #print(node num)
        match load type:
             case 1:
                 load X1 = load BC[idx, 2]
                 F[2*node num-2] = load X1
                 #print("369")
             case 2:
                 load X2 = load BC[idx,3]
                 F[2*node num-1] = load X2
                 #print("Hariprabodham")
             case 12:
                 load X1 = load BC[idx, 2]
                 F[2*node num-2] = load X1
                 load X2 = load BC[idx,3]
                 F[2*node num-1] = load X2
                 #print("I am Akshar")
    print("Nodal force vector = {}".format(F))
Nodal force vector = \begin{bmatrix} 0. & 0. & 0. & 0. & 0. & 0. & 0. & -1000. & 0. & 0. & 0. & 0. & -1000. \end{bmatrix}
    GSTIFF mod = copy.deepcopy(GSTIFF)
    Disp BC data = np.loadtxt("./Disp BC data.txt", dtype="float32",
    comments="#", ndmin=2)
    Num disp BC = Disp BC data.shape[0]
    P = 10e16
    for idx in range (Num disp BC):
         node num = int(Disp BC data[idx,0])
        data type = int(Disp BC data[idx,1])
        match data_type:
             case 1:
                 X1 disp = Disp BC data[idx,2]
                 GSTIFF mod[2*node num-2,2*node num-2] = P
                 F[2*node num-2] = X1 disp*P
                 #print("369")
             case 1:
                 X2 disp = Disp BC data[idx,3]
                 GSTIFF mod[2*node num-1,2*node num-1] = P
                 F[2*node num-1] = X2 disp*P
                 #print("Hariiii")
             case 12:
                 X1 disp = Disp BC data[idx,2]
                 X2 disp = Disp BC data[idx,3]
                 GSTIFF mod[2*node num-2,2*node num-2] = P
                 GSTIFF mod[2*node num-1,2*node num-1] = P
                 F[2*node num-2] = X1 disp*P
                 F[2*node num-1] = X2 disp*P
                 #print("Prabodham")
    #Solution
    u = np.linalg.solve(GSTIFF mod,F)
```

```
print(np.shape(B),np.shape(D),np.shape(u))
    #strain = np.matmul(B,u)
    #stress = np.matmul(D,strain)
    F new = np.matmul(GSTIFF,u)
    print("\nCorrected nodal force vector F = \n{}\n".format(F new))
    print("Sum of all the above elements =
    {}\n".format(np.sum(F new)))
    #plt.plot(F new)
    print("Deformation matrix = \n{} \n".format(u))
    #print("Nodal force vector = \n{} \n".format(F))
    \#u 2d = u.reshape(-1,2)
(3, 6) (3, 3) (16,)
Corrected nodal force vector F =
[ 1.07843128e+03 -2.00000000e+03 -1.07843128e+03 4.00000000e+03
4.54747351e-13 0.00000000e+00 1.81898940e-12 1.81898940e-12
4.54747351e-13 -1.00000000e+03 0.00000000e+00 -4.54747351e-13
1.81898940e-12 -1.42108547e-12 2.27373675e-12 -1.00000000e+03]
Sum of all the above elements = 3.410605131648481e-12
Deformation matrix =
[-1.07843208e-14 2.00000148e-14 1.07843232e-14 -4.00000494e-14
-3.55407601e-08 2.39607859e-08 -4.03598887e-08 -3.28025085e-08
-3.63948342e-08 -1.62182017e-07 -1.27563909e-07 3.04299926e-08
-1.34033115e-07 -3.75646735e-08 -1.47204452e-07 -1.75010682e-07
    #for getting strain and stress values. It looks wrong :(
     strain store, stress store, von mises store = [],[],[]
     for elem idx in range(1,Num elements+1):
         u store = []
         #print(u)
         N1 = NCA[elem idx, 1]
         N2 = NCA[elem idx, 2]
         N3 = NCA[elem idx, 3]
         #print(np.round(u[2*N1-2],7),np.round(u[2*N1-
    1],7),np.round(u[2*N2-2],7),np.round(u[2*N2-
    1],7), np.round(u[2*N3-2],7), np.round(u[2*N3-1],7))
         #print(N1,N2,N3)
         u store.append(u[2*N1-2])
         u store.append(u[2*N1-1])
```

```
u store.append(u[2*N2-2])
         u store.append(u[2*N2-1])
         u store.append(u[2*N3-2])
         u store.append(u[2*N3-1])
         #print(np.shape(B store[elem idx]),np.shape(u store))
         strain = np.matmul(B_store[elem_idx-1],u_store)
         stress = np.matmul(D store[elem idx-1],strain)
         strain store.append(strain)
         stress store.append(stress)
         print("Strain for element {} = {}".format(elem idx,strain))
        print("Stress for element {} = {}
    Pa".format(elem idx,stress))
         #print("Strain matrices:")
         #print(strain)
         #print("Stress matrices:")
         #print(stress)
         sigma xx = stress[0]
         sigma yy = stress[1]
         sigma xy = stress[2]
         von mises = np.sqrt((sigma xx**2+sigma yy**2-
    sigma xx*sigma yy+3*sigma xy**2))
         von mises store.append(von mises)
         print("Von-mises stress for element {} = {} Pa
    \n".format(elem idx,von mises))
    #print(np.round(u,10))
    #print(np.shape(strain store),np.shape(stress store))
    plt.grid()
    plt.scatter(x=range(1,Num elements+1), y=von mises store,
    c=von mises store)
    plt.colorbar(label="Von-mises stress", orientation="vertical")
    plt.xlabel("Elements")
    plt.ylabel("Von-mises stress")
    plt.title("Von-mises Stress (Pa) Vs. Elements")
    plt.show()
Strain for element 1 = [2.15686440e-14-2.39607659e-08\ 3.55406893e-08]
Stress for element 1 = [ -789.91301035 -2633.05045274 1366.94954726] Pa
Von-mises stress for element 1 = 3329.0673822155422 Pa
Strain for element 2 = [-4.81912862e-09 \ 3.28024685e-08 -1.64033949e-08]
Stress for element 2 = [ 362.78440556 6633.05045274 -1366.94954726] Pa
Von-mises stress for element 2 = 6879.553529282195 Pa
```

Strain for element 3 = [-4.81912862e-09 4.76216493e-09 3.69099323e-08]

Stress for element 3 = [-805.56157241 791.32064998 3075.82761959] Pa

Von-mises stress for element 3 = 5504.064131297851 Pa

Strain for element 4 = [-6.46920669e-09 -6.46920669e-09 2.40284826e-08]

Stress for element 4 = [-924.17238041 -924.17238041 924.17238041] Pa

Von-mises stress for element 4 = 1848.3447608145686 Pa

Strain for element 5 = [ 3.96505453e-09 4.76216493e-09 -3.57062822e-08]

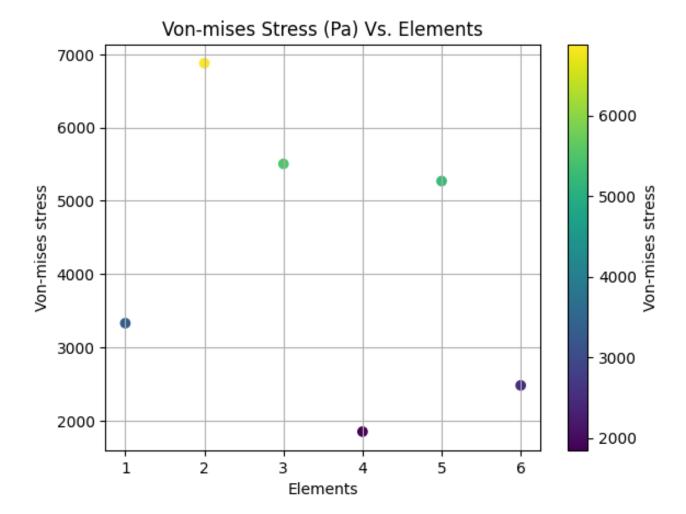
Stress for element 5 = [ 1024.47654935 1157.32827978 -2975.52345065] Pa

Von-mises stress for element 5 = 5269.205317778571 Pa

Strain for element 6 = [-1.31713362e-08 1.28286645e-08 -2.66363911e-08]

Stress for element 6 = [-1024.47654935 975.52345065 -1024.47654935] Pa

Von-mises stress for element 6 = 2479.769284021626 Pa



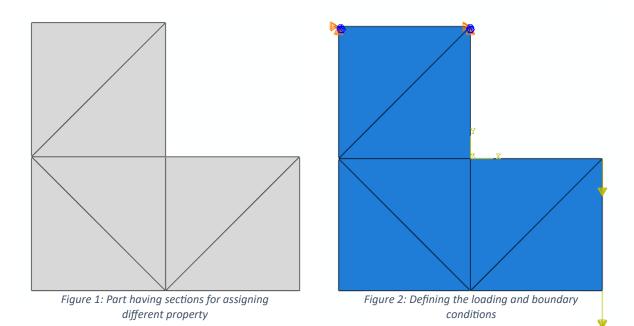
#### Abaqus

Here, I have performed 2 FEA Simulations to replicate our original problem. Here I am going to share the procedure to do run FEA simulation in Abaqus.

- 1. Create a part of given dimensions
- 2. Assign property to part after creating sections in part (fig. 1)
- 3. Create a mesh in part according to problem statement. In this case, it is static FEA analysis and plane stress problem (fig. 3 and 4)

Note: In Abaqus, we couldn't provide too much coarse mesh size which we have used for our coding problem. So, I have use mesh size of 0.5 m and 0.25 m respectively for two FEA simulations.

- 4. Apply loading condition at nodes (fig. 2)
- 5. Define boundary conditions at the node for fixed joint (fig. 2)
- 6. Start the job and export the desired result.



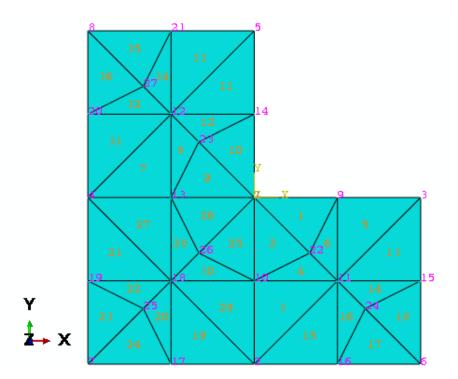


Figure 3: Node and element order for mesh size = 0.5 m

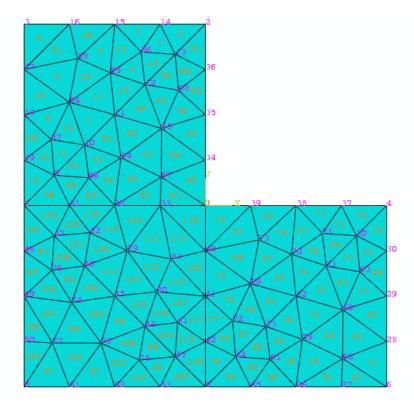


Figure 4: Node and element order for mesh size = 0.25 m

# Deformation study

Comparison of deformation between code Vs. Abaqus (Mesh size = 0.5 m)

| Node Label | Abaqus-U1 | Code-U1     | Error-U1 | Abaqus-U2 | Code-U2     | Error-U2 |
|------------|-----------|-------------|----------|-----------|-------------|----------|
| 1          | 0.00E+00  | 0.00E+00    | 0%       | 0.00E+00  | 0.00E+00    | 0%       |
| 2          | 0.00E+00  | 0.00E+00    | 0%       | 0.00E+00  | 0.00E+00    | 0%       |
| 3          | -9.43E-08 | -1.34E-07   | 42%      | 8.45E-08  | -3.76E-08   | 144%     |
| 4          | -8.16E-08 | -4.04E-08   | 51%      | -8.66E-08 | -3.28E-08   | 62%      |
| 5          | -5.65E-08 | -3.64E-08   | 36%      | -4.24E-07 | -1.62E-07   | 62%      |
| 6          | -3.12E-07 | -1.28E-07   | 59%      | 1.11E-07  | 3.04E-08    | 73%      |
| 7          | -3.37E-07 | -3.55E-08   | 89%      | -1.07E-07 | 2.40E-08    | 122%     |
| 8          | -3.71E-07 | -1.47E-07   | 60%      | -4.32E-07 | -1.75E-07   | 60%      |
|            |           | Total error | 337%     |           | Total error | 523%     |

## Comparison of deformation between code Vs. Abaqus (Mesh size = 0.25 m)

| •          |           |           |       | ,         | ,         |        |
|------------|-----------|-----------|-------|-----------|-----------|--------|
| Node Label | Abaqus-U1 | Code-U1   | Error | Abaqus-U2 | Code-U2   | Error2 |
| 1          | 0.00E+00  | 0.00E+00  | 0%    | 0.00E+00  | 0.00E+00  | 0%     |
| 2          | 0.00E+00  | 0.00E+00  | 0%    | 0.00E+00  | 0.00E+00  | 0%     |
| 3          | -1.34E-07 | -1.34E-07 | 0%    | 1.18E-07  | -3.76E-08 | 132%   |
| 4          | -1.18E-07 | -4.04E-08 | 66%   | -1.42E-07 | -3.28E-08 | 77%    |
| 5          | -7.84E-08 | -3.64E-08 | 54%   | -6.08E-07 | -1.62E-07 | 73%    |
| 6          | -4.48E-07 | -1.28E-07 | 72%   | 1.49E-07  | 3.04E-08  | 80%    |
| 7          | -4.74E-07 | -3.55E-08 | 93%   | -1.64E-07 | 2.40E-08  | 115%   |
| 8          | -5.19E-07 | -1.47E-07 | 72%   | -6.12E-07 | -1.75E-07 | 71%    |
|            |           |           | 355%  |           |           | -547%  |

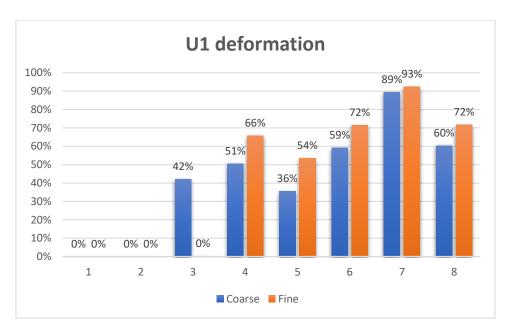


Figure 5: Deformation U1 for coarse and fine mesh

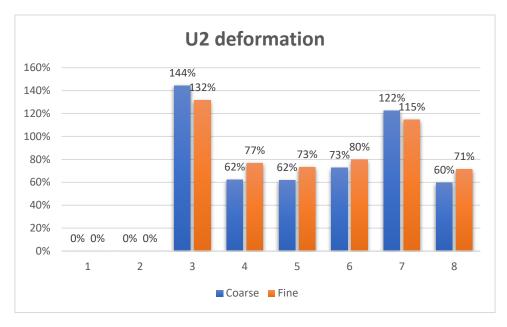


Figure 6: Deformation U2 for coarse and fine mesh

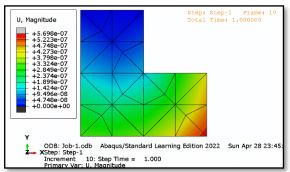


Figure 7: Resultant deformation on mesh size = 0.5 m

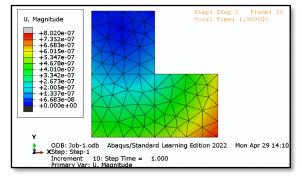


Figure 8: Resultant deformation on mesh size = 0.25 m

From fig 3 and 4, we can conclude that answer which we derived from the coding is way far from the real answer. So, as we increase mesh size from 0.5 m to 0.25 m. Our error will get increased for majority of case. Where error is defined as, | ((deformation from Abaqus – deformation from code)/deformation from code) | \*100 %.

From fig 5 and 6, those are the results of Abaqus simulations with different mesh size. We couldn't compare the result directly due to mismatching of element numbers. But we could comment that our code is working very fine by observing the nature of deformation of the or direction of deformation. From fig 5 and 6, it looks like red area is the one where highest deformation is happening due to initial and boundary conditions. This area has node no. 6 as per Abaqus notation and node no. 8 as per coding problem.

## Reaction forces study

Comparison of reaction forces between code Vs. Abaqus (Mesh size = 0.5 m)

| Node Label | RF-RF1      | Code-RF1 | RF-RF2    | Code-RF2 |
|------------|-------------|----------|-----------|----------|
| 1          | 151.370361  | 1078     | -2.00E+03 | -2000    |
| 2          | -151.370361 | -1078    | 4.00E+03  | 4000     |
| 3          | 0           | 0        | 0         | 0        |
| 4          | 0           | 0        | 0         | 0        |
| 5          | 0           | 0        | 0         | 0        |
| 6          | 0           | 0        | 0         | 0        |
| 7          | 0           | 0        | 0         | 0        |
| 8          | 0           | 0        | 0         | 0        |

## Comparison of reaction forces between code Vs. Abaqus (Mesh size = 0.25 m)

|            |         |          | 1 1       | ,        |
|------------|---------|----------|-----------|----------|
| Node Label | RF-RF1  | Code-RF1 | RF-RF2    | Code-RF2 |
| 1          | -92.889 | 1078     | -2.00E+03 | -2000    |
| 2          | 92.8891 | -1078    | 4.00E+03  | 4000     |
| 3          | 0       | 0        | 0         | 0        |
| 4          | 0       | 0        | 0         | 0        |
| 5          | 0       | 0        | 0         | 0        |
| 6          | 0       | 0        | 0         | 0        |
| 7          | 0       | 0        | 0         | 0        |
| 8          | 0       | 0        | 0         | 0        |

### Von-mises stress analysis

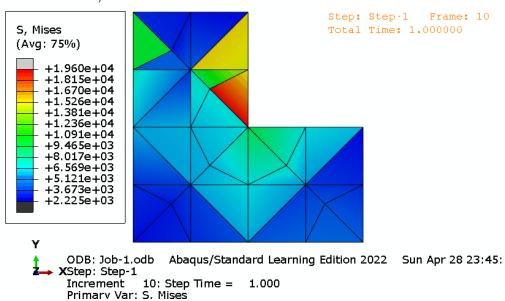
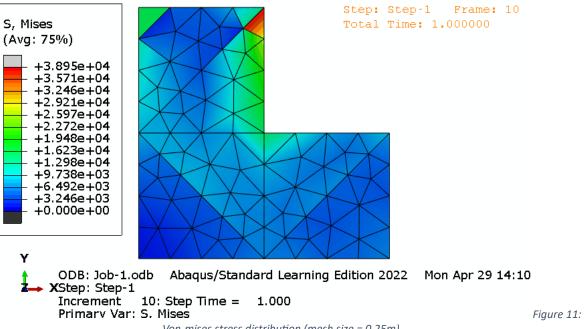


Figure 9: Von-mises stress distribution (mesh size = 0.5m)

| Element Label | Node Label | X         | Υ        | S-Mises         |
|---------------|------------|-----------|----------|-----------------|
| 15            | 24         | -3.33E-01 | 3.33E-01 | 6.81E+03        |
| 16            | 7          | 0         | 0        | 1.96E+04        |
| 16            | 16         | 0         | 5.00E-01 | 1.49E+04        |
| 16            | 24         | -3.33E-01 | 3.33E-01 | 1.96E+04        |
| 17            | 2          | 0         | 1        | <b>1.57E+04</b> |
| 17            | 10         | -5.00E-01 | 5.00E-01 | <b>1.57E+04</b> |

Figure 10: Von-mises result from mesh size = 0.5 m



*Von-mises stress distribution (mesh size = 0.25m)* 

| Element Label                      | Node Label | X         | Υ        | S-Mises  |
|------------------------------------|------------|-----------|----------|----------|
| 50                                 | 12         | -3.33E-01 | 6.67E-01 | 8.46E+03 |
| 51                                 | 2          | 0         | 1        | 3.90E+04 |
| 51                                 | 13         | -1.67E-01 | 8.33E-01 | 3.90E+04 |
| 51                                 | 36         | 0         | 7.50E-01 | 2.73E+04 |
| 52                                 | 10         | -6.67E-01 | 3.33E-01 | 5.36E+03 |
| 52                                 | 64         | -4.66E-01 | 2.63E-01 | 5.10E+03 |
| 50_cm_mesh 25_cm_mesh Code_results | +          | I (       |          |          |

Figure 12: Von-mises result from mesh size = 0.25 m

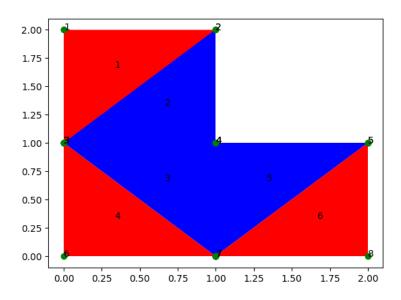


Figure 13: Node and element notation as per code

| Element            | Von-mises stress    |  |
|--------------------|---------------------|--|
| 1                  | <b>332</b> 9.067382 |  |
| 2                  | 6879.553529         |  |
| 3                  | <b>5504.064131</b>  |  |
| 4                  | <b>18</b> 48.344761 |  |
| 5                  | <b>526</b> 9.205318 |  |
| 6                  | <b>247</b> 9.769284 |  |
|                    |                     |  |
| 50_cm_mesh 25_cm_m | nesh Code_results + |  |

Figure 13: Von-mises result from code results

Fig 9 and 10 are the result generated from abaqus with mesh size 0.5m and 0.25m respectively. After simulating the FEA analysis, I've exported the necessary data to excel for post processing analysis. In fig 11, it's shows that element 16 have highest von-misses stress. Similarly in fig 12, it's shows that element 51 have highest von-misses stress compared to other elements. In code result, element 2 have highest von-misses stress compared to other elements. These result which generates from the abaqus is having same nature of vonmises stress which generates from code results. Element 2 in code result, element 16 in mesh size of 0.5 m simulation and element 51 in mesh size 0.25 m simulation are excepted to high stress as per the concept of strength of material. As we increase the mesh size in analysis we approch towards the correct area where high stress is being generated.

#### Conclusion

So, we can conclude that finite element analysis using python is being use as backend in all the FEM solver like Abaqus and Ansys. As we can see in fig.5 and fig.6, the result derived from codes are away from the actual result. As increasing mesh size from 0.5m to 0.25m we are getting the results which are more realistically correct. So, finite element methods are one of the way to find the deformation. From deformation we can derived all the required quantities like stress, strain, von-mises stress for each element. This process is called post-processiong which we have done in our code also.

Click here <a href="https://github.com/SetuThacker/FEM">https://github.com/SetuThacker/FEM</a> Assignments to get a pdf of the report.