fluid

Bacco, Giacomo

Citing

Please cite this work if you have used it in your project or in your program.

Giacomo Bacco. fluid: Free Fluid Flux-Barriers Rotor for Synchronous Reluctance Motor Drawing. 2018. URL: https://github.com/gbacco5/fluid

Contents

1	File	es needed	2										
2	Hov	w to use	4										
3	Examples												
4	The	eory	6										
	4.1	Flow past a cylinder	6										
	4.2	Conformal mapping	7										
	4.3	Computation of flux-barrier base points	7										
		4.3.1 Magnet insertion	9										
		4.3.2 Central base points	9										
	4.4	Outer base points	10										
	4.5	Flux-barrier sideline points	10										
	4.6	Output	11										
	4.7	Example of Matlab/Octave plot	12										

5	Coc	Code														13					
	5.1	Main file																			13
	5.2	Calc fluid barrier .																			15
	5.3	Draw fluid barrier																			25

Goal

Provide a ready-to-use fully parametric drawing of the Synchronous Reluctance Rotor with fluid flux-barriers.

The scope of this project is the computation of the flux-barriers points. The drawing scripts are for demonstration purposes only.

Requirements

Matlab or Octave or Python (NumPy + SciPy) to compute the points. The points calculation is general, so it could be implemented in any language, but I chose Matlab/Octave because it is my standard interface with FEMM software.

If you do not use FEMM, you can still use the calculation part and make a porting for your CAD engine or FEA software. If you do so, consider contributing to the project adding your interface scripting.

1 Files needed

```
For Matlab/Octave, the files needed for the computation are calc_fluid_barrier.m,
   GetFSolveOptions.m and
   isOctave.m
while for Python it is
   fluid_functions.py.
```

Contacts

You can contact me at giacomo.bacco@phd.unipd.it.

If you find a bug, consider opening an issue at https://github.com/gbacco5/fluid/issues

Last update: April 22, 2018

Notice:

fluid Copyright 2018 Giacomo Bacco

isOctave Copyright (c) 2010, Kurt von Laven All rights reserved.

2 How to use

Open the file fluid and run it.

Change the machine data in the data section. All the variables have a comment next to them.

There are some "hidden" options which should be explained.

- you can provide personal flux barrier angles, or let the program compute them as the average of the final points C and D at the rotor periphery. This means that you have to always provide the fluxbarrier thicknesses and flux-carrier widths. Optionally, you could also provide the electrical flux-barrier angles.
- 2. by default, the flux-barrier-end is round, so the code solves an additional system to determine the correct locations of the fillet points. You can skip such system declaring

```
1 rotor.barrier_end = 'rect';
```

and so selecting "rectangular" flux-barrier-end.

3. the inner radial iron ribs are optional, but you are free to provide different widths for every flux-barrier.

```
1 rotor.wrib = [1,2,4]*mm;
```

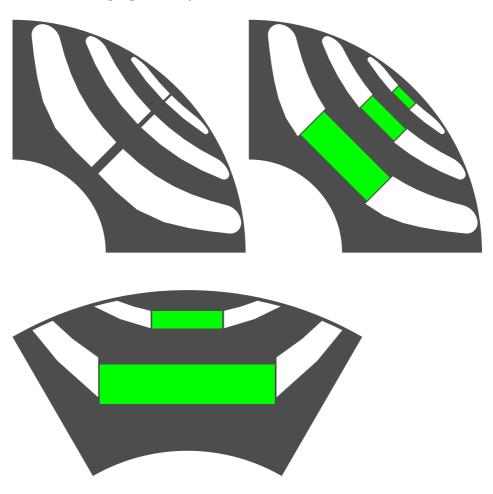
4. if you also input magnet widths, the rib is automatically enlarged to accommodate the magnet, similarly to an IPM (Interior Permanent Magnet) machine.

```
1 rotor.wm = [10,20,40]*mm;
```

In this case, the output structure barrier also contains the location of the magnet base center point.

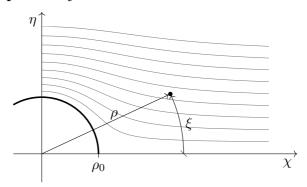
3 Examples

Here are some finished examples based on the output. The drawings are for demonstration purposes only.



4 Theory

4.1 Flow past a cylinder



Let ρ_0 be the radius of the cylinder, ρ, ξ the polar coordinate system in use. One possible solution of this problem have these potential and streamline functions:

$$\phi(\rho,\xi) = \left(\rho + \frac{\rho_0^2}{\rho}\right)\cos\xi\tag{1}$$

$$\psi(\rho,\xi) = \left(\rho - \frac{\rho_0^2}{\rho}\right)\sin\xi\tag{2}$$

Although these equations are deeply coupled, the radius ρ and the phase ξ can be obtained as a function of the other quantities. For our purposes, we use ψ .

$$\rho(\psi,\xi) = \frac{\psi + \sqrt{\psi^2 + 4\rho_0^2 \sin^2 \xi}}{2\sin \xi}$$
 (3)

$$\xi(\psi, \rho) = \arcsin\left(\frac{\rho \,\psi}{\rho^2 - \rho_0^2}\right) \tag{4}$$

The velocity field can also be derived through

$$v_{\rho}(\rho,\xi) = \frac{\partial \phi}{\partial \rho} = \left(1 - \frac{\rho_0^2}{\rho^2}\right) \cos \xi$$

$$v_{\xi}(\rho,\xi) = \frac{1}{\rho} \frac{\partial \phi}{\partial \xi} = -\left(1 + \frac{\rho_0^2}{\rho^2}\right) \sin \xi$$
(5)

4.2 Conformal mapping

From the reference plane, which is equivalent to a two-pole machine, we use a complex map to obtain the quantities in the actual plane. Let p be the number of pole pairs. Then:

$$\zeta \xrightarrow{\mathcal{M}} z = \sqrt[p]{\zeta}$$

$$\rho e^{j\xi} \xrightarrow{\mathcal{M}} r e^{j\vartheta} = \sqrt[p]{\rho} e^{j\xi/p}$$

$$\chi + j\eta \xrightarrow{\mathcal{M}} x + jy$$
(6)

It is easy to find the inverse map:

$$\mathcal{M} \colon \sqrt[p]{\cdot} \qquad \mathcal{M}^{-1} \colon (.)^p \tag{7}$$

In the transformed plane, the velocities have a different expression:

$$v_r(r,\vartheta) = p \left(r^{p-1} - \frac{R_0^{2p}}{r^{p+1}} \right) \cos p\vartheta$$

$$v_{\vartheta}(r,\vartheta) = -p \left(r^{p-1} + \frac{R_0^{2p}}{r^{p+1}} \right) \sin p\vartheta$$
(8)

This vector field is tangent to the streamlines in every point in the transformed plane. In order to work with this field in x, y coordinates, we need a rotational map:

$$v_x(r,\vartheta) = v_r \cos \vartheta - v_\vartheta \sin \vartheta$$

$$v_\vartheta(r,\vartheta) = v_r \sin \vartheta + v_\vartheta \cos \vartheta$$
(9)

4.3 Computation of flux-barrier base points

Refer to Figure 1 for the points naming scheme. Keep in mind that A' is not simply the projection of A onto the q-axis, but it represent the original starting point for the barrier sideline, so it lies on the flux-barrier streamline. The same is true for points B', B, C', C, and D', D.

Let the flux-barrier and flux-carrier thicknesses be given. Then the base points for the flux-barriers can be computed easily. Let $D_{\rm r}$ be the

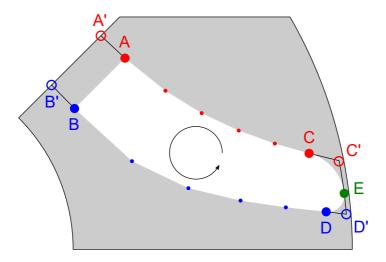


Figure 1: Flux-barrier base points description.

rotor outer diameter, $w_{\text{rib,t}}$ the tangential iron rib width, $w_{\text{c,k}}$ the k-th flux-carrier width, and $t_{\text{b,k}}$ the k-th flux-barrier thickness.¹ Then

$$R_{\rm r} = \frac{D_{\rm r}}{2} - w_{\rm rib,t}$$

$$R_{\rm A'_1} = R_{\rm r} - w_{\rm c,1}$$

$$R_{\rm B'_1} = R_{\rm A'_1} - t_{\rm b,1}$$

$$\vdots$$
(10)

where R represents the radius from the origin. So, in general:

$$R_{\mathsf{A}'_{\mathsf{k}}} = R_{\mathsf{B}'_{\mathsf{k}-1}} - w_{\mathsf{c},k}$$

$$R_{\mathsf{B}'_{\mathsf{k}}} = R_{\mathsf{A}'_{\mathsf{k}-1}} - t_{\mathsf{b},k}$$
(11)

with the exception $R_{\mathsf{B}_0'} = R_{\mathsf{r}}$.

¹You may wonder why the main dimensions of the flux-carrier and flux-barrier differ in the name (width versus thickness). This is due to a choice of mine, because I prefer to refer to width when the flux flows perpendicularly to the dimension, and to thickness when it flows in parallel.

Now we know both the radii and the angle – always $\pi/(2p)$ – of the flux-barrier internal points. So we can compute their respective streamline value.

4.3.1 Magnet insertion

$$w_{\mathrm{rib},k} = w_{\mathrm{rib},k} + w_{\mathrm{m},k}$$

where $w_{m,k}$ is the k-th magnet width.

4.3.2 Central base points

We refer to points A and B. If the rib width is zero $A \equiv A'$ and $B \equiv B'$. The line describing the q-axis is

$$y = mx + q$$

$$m = \tan \frac{\pi}{2p}$$

$$q = \frac{w_{\text{rib}}}{2\cos \frac{\pi}{2p}}$$
(12)

$$\begin{cases} y_{\mathsf{A}} - mx_{\mathsf{A}} - q = 0 \\ x_{\mathsf{A}} - r_{\mathsf{A}}(\psi_{\mathsf{A}'}, \vartheta_{\mathsf{A}})\cos\vartheta_{\mathsf{A}} = 0 \\ y_{\mathsf{A}} - r_{\mathsf{A}}(\psi_{\mathsf{A}'}, \vartheta_{\mathsf{A}})\sin\vartheta_{\mathsf{A}} = 0 \end{cases}$$
(13)

where ϑ_{A} is used as the third degree of freedom and r_{A} is then a function of it. The solution of such system can be determined solving the single equation

$$r_{\mathsf{A}}(\psi_{\mathsf{A}'}, \vartheta_{\mathsf{A}}) \left(\sin \vartheta_{\mathsf{A}} - m \cos \vartheta_{\mathsf{A}} \right) - q = 0 \tag{14}$$

in the unknown ϑ_A . The function $r(\psi, \vartheta)$ is simply

$$r(\psi, \vartheta) = \sqrt[p]{\rho(\psi, \vartheta/p)}$$

The same equation can be written for point B with the proper substitution and repeated for all the flux-barriers.

4.4 Outer base points

We refer to points C, D, and E. If the flux-barrier angle, α_b , is given, then

$$x_{\mathsf{E}} = R_r \cos(\frac{\pi}{2p} - \alpha_{\mathsf{b}})$$

$$y_{\mathsf{E}} = R_r \sin(\frac{\pi}{2p} - \alpha_{\mathsf{b}})$$
(15)

Points C and D results from the connection of the flux-barrier sidelines and point E. This connection should be as smooth as possible in order to avoid dangerous mechanical stress concentrations. We are going to use circular arcs to make this connection. So we impose the tangency between the flux-barrier sideline and the arc, between the arc and the radius through point E. The tangent to the sideline can be obtained through the velocity field described above.

Then we want point C to lay on the flux-barrier sideline. These conditions represent a nonlinear system of 4 equations, in 6 unknowns. So we need two more equations, which are that points C and E belong to the fillet circle with radius R.

$$\begin{cases} x_{\mathsf{C}} - r_{\mathsf{C}}(\psi_{\mathsf{C}}, \vartheta_{\mathsf{C}}) \cos \vartheta_{\mathsf{C}} = 0 \\ y_{\mathsf{C}} - r_{\mathsf{C}}(\psi_{\mathsf{C}}, \vartheta_{\mathsf{C}}) \sin \vartheta_{\mathsf{C}} = 0 \\ (x_{\mathsf{C}} - x_{\mathsf{O}_{\mathsf{C}}})^{2} + (y_{\mathsf{C}} - y_{\mathsf{O}_{\mathsf{C}}})^{2} - R_{\mathsf{EC}}^{2} = 0 \\ (x_{\mathsf{E}} - x_{\mathsf{O}_{\mathsf{C}}})^{2} + (y_{\mathsf{E}} - y_{\mathsf{O}_{\mathsf{C}}})^{2} - R_{\mathsf{EC}}^{2} = 0 \\ (x_{\mathsf{O}_{\mathsf{C}}} - x_{\mathsf{E}})y_{\mathsf{E}} - (y_{\mathsf{O}_{\mathsf{C}}} - y_{\mathsf{E}})x_{\mathsf{E}} = 0 \\ (x_{\mathsf{O}} - x_{\mathsf{C}})v_{x}(r_{\mathsf{C}}, \vartheta_{\mathsf{C}}) + (y_{\mathsf{O}} - y_{\mathsf{C}})v_{y}(r_{\mathsf{C}}, \vartheta_{\mathsf{C}}) \end{cases}$$
(16)

The very same system can be written and solved for point D.

4.5 Flux-barrier sideline points

Consider the top flux-barrier sideline, so the one going from point A to point C. We want to create such sideline using a predetermined number of steps, N_{step} . From now on, let us call this number N, and N_k for the k-th flux-barrier.

One of the best way to distribute the points along the streamline is to use the potential function, ϕ , defined in Equation 1. We start computing

the potential for points A and C:

$$\phi_{\mathsf{A}} = \phi(\rho_{\mathsf{A}}, \xi_{\mathsf{A}})$$
$$\phi_{\mathsf{C}} = \phi(\rho_{\mathsf{C}}, \xi_{\mathsf{C}})$$

Then, we want to find N-1 points along the streamline between points A and C with a uniform distribution of the potential function. We define

$$\Delta\phi_{\mathsf{AC}} = \frac{\phi_{\mathsf{C}} - \phi_{\mathsf{A}}}{N}$$

So we can compute the potentials we are looking for

$$\phi_i = \phi_{\mathsf{A}} + i\Delta\phi_{\mathsf{AC}}, \quad i = 1, \dots, N-1$$

and finally the location of the point with this potential value and the streamline function value required to lie on the flux-barrier sideline. This translates to the following system of equations:

$$\begin{cases} \psi_{\mathsf{AC}} - \psi(\rho, \xi) = 0\\ \phi_i - \phi(\rho, \xi) = 0 \end{cases}$$

$$\tag{17}$$

The system is well-defined because there are two unknowns and two independent equations. This system must be solved for every flux-barrier sideline point, for the two sides, and for every flux-barrier.²

4.6 Output

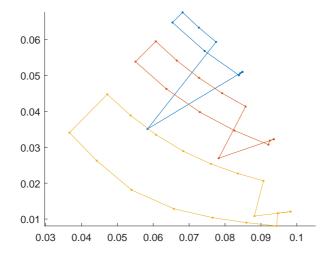
The output of the computation function in Matlab/Octave is one vector of structures (barrier(:)) which contains at least two fields (X and Y). The X vector is made in this way:

$$X = \begin{bmatrix} x_{\mathsf{E}} & x_{\mathsf{O}_{\mathsf{C}}} & x_{\mathsf{C}} & x_{\mathsf{AC}_{N_{\mathsf{step}}-1}} & \cdots & x_{\mathsf{AC}_{1}} & x_{\mathsf{A}} \\ x_{\mathsf{B}} & x_{\mathsf{BD}_{1}} & \cdots & x_{\mathsf{BD}_{N_{\mathsf{step}}-1}} & x_{\mathsf{D}} & x_{\mathsf{O}_{\mathsf{D}}} & x_{\mathsf{E}} \end{bmatrix}^{\mathsf{T}}$$

and similarly the Y vector. So the points are ordered starting from the point E and then moving counter-clockwise until E is reached again.

 $^{^2}$ In Matlab/Octave, the "for every flux-barrier sideline point" loop has been vectorized, while the two sides has been manually split.

4.7 Example of Matlab/Octave plot



Here is an example of a Matlab/Octave output plot. The V-shaped lines represent the radii of the fillet arcs, which were not worth to be shown in Matlab/Octave.

5 Code

5.1 Main file

```
% FLUID
   % Free Fluid Flux-Barriers Rotor for Synchronous
       Reluctance Motor Drawing
3
   % Bacco, Giacomo 2018
  clear all; close all; clc;
   addpath('draw', 'tools');
   응응 DATA
  rotor.p = 2; % number of pole pairs
10
   mm = 1e-3; % millimeters
11
   rotor.De = 200*mm; % [m], rotor outer diameter
12
   rotor.Nb = 3; % number of flux-barriers
14
   rotor.tb = [4 8 15]*mm; % flux-barrier thicknesses
   rotor.wc = [3 7 12 10]*mm; % flux-carrier widths
16
   rotor.Nstep = 3*[2, 4, 6]; % number of steps to draw the
17
       flux-barrier side
   rotor.wrib_t = 1*mm; % [m], tangential iron rib width
18
   % you can input flux-barrier angles or let the program
20
       compute them
   % rotor.barrier_angles_el = [14,26,38]*2; % [deg],
       electrical flux-barrier angles
   % rotor.barrier_end = 'rect'; % choose 'rect' or comment
   % you can define the rib width or comment
   rotor.wrib = [0,1,1]*mm; % [m], radial iron rib widths
   % You can define the magnet width or comment
   % rotor.wm = [10, 20, 40]*mm;
27
   %% barrier points computation
   barrier = calc_fluid_barrier(rotor);
32
   %% simple matlab plot
33
   figure
```

```
34 hold all
35 axis equal
36 for bkk = 1:rotor.Nb
plot(barrier(bkk).X, barrier(bkk).Y, '.-')
  end
38
  if isfield(rotor,'wm')
   RM = [barrier(:).Rm];
    thM = pi/2/rotor.p;
    Xm = RM.*cos(thM);
42
    Ym = RM.*sin(thM);
43
   plot(Xm, Ym, 'ko')
44
45 end
   axis auto
46
   %% FEMM drawing
48
49
   try
50
    openfemm(1)
    newdocument(0);
51
    draw_fluid_barrier(barrier);
   catch
     disp('FEMM not available.');
    end
```

5.2 Calc fluid barrier

```
function barrier = calc_fluid_barrier(r)
   % CALC_FLUID_BARRIER computes the flux-barrier points
       along the streamline
   % function.
   용용 DATA
   global deb
   Dr = r.De; % [m], rotor outer diameter
   ScalingFactor = 1/( 10^(round(log10(Dr))) );
   % ScalingFactor = 1;
   Dr = Dr*ScalingFactor;
11
   p = r.p; % number of pole pairs
   Nb = r.Nb; % number of flux-barriers
   tb = r.tb*ScalingFactor; % flux-barrier widths
   wc = r.wc*ScalingFactor; % flux-carrier widths
17
   Nstep = r.Nstep; % number of steps to draw the flux-
       barrier side
   wrib_t = r.wrib_t*ScalingFactor; % [m], tangential iron rib
       width
   if isfield(r,'barrier_angles_el')
21
    barrier_angles_el = r.barrier_angles_el; % [deg], electrical
22
        flux-barrier angles
     AutoBarrierEndCalc = 0;
23
24
   else
    barrier_angles_el = zeros(1,Nb);
    AutoBarrierEndCalc = 1;
   end
27
   if isfield(r,'wm')
30
   wm = r.wm*ScalingFactor;
   else
31
    wm = 0;
32
33
   end
   if isfield(r,'wrib')
34
    wrib = r.wrib*ScalingFactor + wm; % [m], radial iron rib
```

```
widths
    else
36
      wrib = zeros(1,Nb) + wm;
37
    end
38
    Dend = Dr - 2*wrib_t; % [m], flux-barrier end diameter
40
    Dsh = Dend - 2*( sum(tb) + sum(wc) ); % [m], shaft diameter
    RO = Dsh/2; % [m], shaft radius
    barrier_angles = barrier_angles_el/p; % [deg], flux-barrier
    if isfield(r,'barrier_end')
44
    barrier_end = r.barrier_end;
45
46
    else
    barrier_end = '';
47
48
    end
   %% IMPLICIT FUNCTIONS
50
   % definition of fluid past a cylinder functions
51
   psi_fluid = @(rho,xi,rho0) (rho.^2 - rho0^2)./rho.*sin(xi);
   phi_fluid = @(rho,xi,rho0) (rho.^2 + rho0^2)./rho.*cos(xi);
   xi_fluid = @(psi,rho,rho0) asin(psi.*rho./(rho.^2 - rho0^2));
    rho_fluid = @(psi,xi,rho0) ( psi + sqrt(psi.^2 + 4*sin(xi).^2*
       rho0^2) )./(2*sin(xi));
   r_map = @(rho) rho.^(1/p);
   th_map = @(xi) xi./p;
   rho_map = @(r) r.^p;
    xi_map = 0(th) th.*p;
60
   vr = Q(r,th,R0) p*(r.^(p-1) - R0^(2*p)./r.^(p+1)).*cos(p*th);
62
   vt = Q(r,th,R0) -p*(r.^(p-1) + R0^(2*p)./r.^(p+1)).*sin(p*th);
63
    vx = @(vr_v, vth_v, th) vr_v.*cos(th) - vth_v.*sin(th);
    vy = @(vr_v, vth_v, th) vr_v.*sin(th) + vth_v.*cos(th);
65
    %% Precomputations
    rho0 = rho_map(R0);
    %% Central base points
    RAprime = Dend(1)/2 - [0, cumsum(tb(1:end-1))] - cumsum(wc(1:end-1))]
        end-1)); % top
    RBprime = RAprime - tb; % bottom
```

```
te_qAxis = pi/(2*p); % q-axis angle in rotor reference
        frame
    % get A' and B' considering rib and magnet widths
75
    mCentral = tan(te_qAxis); % slope
    qCentral = repmat( -wrib/2/cos(te_qAxis), 1, 2); % intercept
    psiCentralPtA = psi_fluid(rho_map(RAprime), xi_map(te_qAxis),
    psiCentralPtB = psi_fluid(rho_map(RBprime), xi_map(te_qAxis),
80
        rho0);
    psiCentralPt = [psiCentralPtA, psiCentralPtB];
82
    psiA = psiCentralPtA;
    psiB = psiCentralPtB;
83
    CentralPt_Eq = @(th) ...
85
      r_map( rho_fluid(psiCentralPt, xi_map(th), rho0) ).*...
86
       ( sin(th) - mCentral*cos(th) ) - qCentral;
87
    if deb == 1
89
      options.Display = 'iter'; % turn off folve display
90
91
    else
92
      options.Display = 'off'; % turn off folve display
    options.Algorithm = 'levenberg-marquardt'; % non-square
        systems
    options.FunctionTolerance = 10*eps;
    options.StepTolerance = 1e4*eps;
96
    X0 = repmat(te_qAxis,1,2*Nb);
98
    options = GetFSolveOptions(options);
99
100
    teAB = fsolve(CentralPt_Eq, X0, options);
    teA = teAB(1:Nb);
101
    teB = teAB(Nb+1:end);
102
    RA = r_map( rho_fluid(psiA, xi_map(teA), rho0) );
103
    RB = r_map( rho_fluid(psiB, xi_map(teB), rho0) );
104
    % magnet central base point radius computation
106
    RAsecond = RA.*cos(te_qAxis - teA);
    RBsecond = RB.*cos(te_qAxis - teB);
```

```
Rmag = (RAprime + RAsecond + RBprime + RBsecond)/4;
    %% Outer base points C,D preparation
112
    RCprime = Dend/2;
113
    teCprime = th_map( xi_fluid(psiA, rho_map(RCprime), rho0) );
114
    xCprime = Dend/2.*cos(teCprime);
115
    yCprime = Dend/2.*sin(teCprime);
116
    RDprime = Dend/2;
118
    teDprime = th_map( xi_fluid(psiB, rho_map(RDprime), rho0) );
119
120
    xDprime = Dend/2.*cos(teDprime);
    yDprime = Dend/2.*sin(teDprime);
121
    if AutoBarrierEndCalc
123
     teE = (teCprime + teDprime)/2;
124
125
     aphE = pi/2/p - teE;
126
     barrier_angles = 180/pi*aphE;
     barrier_angles_el = p*barrier_angles;
127
    else
128
      aphE = barrier_angles*pi/180;
129
     teE = pi/2/p - aphE;
130
    end
131
    xE = Dend/2.*cos(teE);
132
    yE = Dend/2.*sin(teE);
    %% Outer base points C (top)
136
    if strcmp(barrier_end, 'rect')
     RC = RCprime;
137
      teC = teCprime;
138
      xC = xCprime;
139
      yC = yCprime;
140
      xOC = xC;
141
      yOC = yC;
142
    else
144
      options.Algorithm = 'trust-region-dogleg'; % non-square
145
        systems
      BarrierEndSystem = @(th,xd,yd,xo,yo,R) ...
147
         [xd - r_map(rho_fluid(psiA', p*th, rho0)).*cos( th )
        yd - r_map(rho_fluid(psiA', p*th, rho0)).*sin( th )
149
```

```
150
         (xd - xo).^2 + (yd - yo).^2 - R.^2
         (xE' - xo).^2 + (yE' - yo).^2 - R.^2
151
         (xo - xd).*vx( vr( r_map(rho_fluid(psiA', p*th, rho0)),th,R0
152
         ), vt( r_map(rho_fluid(psiA', p*th, rho0)) ,th,R0 ), th) +
         (yo - yd).*vy( vr(r_map(rho_fluid(psiA', p*th, rho0)),th,
        RO ), vt( r_map(rho_fluid(psiA', p*th, rhoO)) ,th,RO ), th)
         (xo - xE').*yE' - (yo - yE').*xE'
153
         % th - xi_fluid((rho_fluid(p*th, psiA', rho0)),
        psiA', rho0)/p % serve?
155
        ];
      XO = [1.5*teE', 0.9*xE', 0.9*yE', 0.8*xE', 0.8*yE', 0.25*xE]
157
        '];
       % X0 = [aph_b, 0, 0, 0, 0, 0];
158
      X = fsolve(@(x) BarrierEndSystem(x(:,1),x(:,2),x(:,3),x(:,4))
159
         x(:,5),x(:,6) ), X0, options);
      xOC = X(:,4);
161
      yOC = X(:,5)';
162
      xC = X(:,2)';
163
      yC = X(:,3)';
164
      RC = hypot(xC, yC);
166
      teC = atan2(yC, xC);
167
     %% Outer base points D (bottom)
170
    if strcmp(barrier_end, 'rect')
      RD = RDprime;
171
      teD = teDprime;
172
      xD = xDprime;
173
      yD = yDprime;
174
175
      xOD = xD;
      yOD = yD;
176
    else
178
       options.Algorithm = 'levenberg-marquardt'; % non-square
179
        systems
      BarrierEndSystem = @(th,xd,yd,xo,yo,R) ...
181
182
         [xd - r_map(rho_fluid(psiB', p*th, rho0)).*cos( th )
         yd - r_map(rho_fluid(psiB', p*th, rho0)).*sin( th )
183
```

```
184
         (xd - xo).^2 + (yd - yo).^2 - R.^2
         (xE' - xo).^2 + (yE' - yo).^2 - R.^2
185
         (xo - xd).*vx( vr( r_map(rho_fluid(psiB', p*th, rho0)),th,R0
186
         ), vt( r_map(rho_fluid(psiB', p*th, rho0)) ,th,R0 ), th) +
        (yo - yd).*vy( vr( r_map(rho_fluid(psiB', p*th, rho0)),th,
        RO ), vt( r_map(rho_fluid(psiB', p*th, rhoO)) ,th,RO ), th)
        (xo - xE').*yE' - (yo - yE').*xE'
           th - xi_fluid((rho_fluid(p*th, psi_d, rho0)),
        psi_d, rho0)/p % serve?
189
        ];
      XO = [0.8*teE', 0.8*xE', 0.8*yE', xE'*.9, yE'*.9, xE'*.2];
191
      X = fsolve(@(x) BarrierEndSystem(x(:,1),x(:,2),x(:,3),x(:,4))
192
        ,x(:,5),x(:,6) ), XO, options);
      xOD = X(:,4);
194
      yOD = X(:,5)';
195
      xD = X(:,2)';
196
      yD = X(:,3)';
197
      RD = hypot(xD, yD);
198
199
      teD = atan2(yD, xD);
200
    end
     %% Flux-barrier points
    % We already have the potentials of the two flux-barrier
203
         sidelines
204
    phiA = phi_fluid( rho_map(RA), xi_map(teA), rho0);
    phiB = phi_fluid( rho_map(RB), xi_map(teB), rho0);
205
    phiC = phi_fluid( rho_map(RC), xi_map(teC), rho0);
207
    phiD = phi_fluid( rho_map(RD), xi_map(teD), rho0);
208
    %% Code for single Nstep
210
    % dphiAC = (phiC - phiAprime)./Nstep;
211
    % dphiBD = (phiD - phiBprime)./Nstep;
212
213
214
    % % we create the matrix of potentials phi needed for
        points intersections
    % PhiAC = phiAprime + cumsum( repmat(dphiAC, Nstep - 1,
215
    % PhiBD = phiBprime + cumsum( repmat(dphiBD, Nstep - 1,
```

```
1));
217
218
    % PhiAC vec = reshape(PhiAC, numel(PhiAC), 1);
    % PhiBD vec = reshape(PhiBD, numel(PhiBD), 1);
219
    % PsiAC vec = reshape( repmat( psiA, Nstep-1, 1), numel(
        PhiAC), 1);
    % PsiBD_vec = reshape( repmat( psiB, Nstep-1, 1), numel(
        PhiBD), 1);
222
223
    % % we find all the barrier points along the streamline
224
    % PsiPhi = @(rho,xi, psi,phi, rho0) ...
        [psi - psi_fluid(rho, xi, rho0)
225
226
         phi - phi_fluid(rho, xi, rho0)];
227
    % X0 = [repmat(rho0*1.1, numel(PhiAC vec), 1), repmat(pi
228
        /4, numel(PhiAC_vec), 1)];
    % RhoXi_AC = fsolve(@(x) PsiPhi(x(:,1),x(:,2),
229
        PsiAC vec, PhiAC vec, rho0), X0, options);
    % RhoXi BD = fsolve(@(x) PsiPhi(x(:,1),x(:,2),
230
        PsiBD_vec, PhiBD_vec, rho0), X0, options);
231
    % R AC = reshape( r map(RhoXi AC(:,1)), Nstep-1, Nb );
232
233
    % te_AC = reshape( th_map(RhoXi_AC(:,2)), Nstep-1, Nb );
    % R_BD = reshape( r_map(RhoXi_BD(:,1)), Nstep-1, Nb );
    % te_BD = reshape( th_map(RhoXi_BD(:,2)), Nstep-1, Nb );
237
    %% Code for different Nsteps
    % we find all the barrier points along the streamline
238
    PsiPhi = @(rho,xi, psi,phi, rho0) ...
239
      [psi - psi_fluid(rho, xi, rho0)
240
      phi - phi_fluid(rho, xi, rho0)];
241
    % barrier(Nb).R AC = 0;
243
    % barrier(Nb).R BD = 0;
244
    % barrier(Nb).te AC = 0;
245
    % barrier(Nb).te BD = 0;
246
    barrier(Nb) = struct;
247
    for bkk = 1:Nb
249
      dphiAC = (phiC(bkk) - phiA(bkk))./Nstep(bkk);
      dphiBD = (phiD(bkk) - phiB(bkk))./Nstep(bkk);
251
```

```
% we create the matrix of potentials phi needed for
        points intersections
      PhiAC = phiA(bkk) + cumsum( repmat(dphiAC', Nstep(bkk) - 1, 1)
253
      PhiBD = phiB(bkk) + cumsum( repmat(dphiBD', Nstep(bkk) - 1, 1)
254
         );
      PsiAC = repmat( psiA(bkk), Nstep(bkk)-1, 1);
255
      PsiBD = repmat( psiB(bkk), Nstep(bkk)-1, 1);
258
     % 1st trv
259
         X0 = [repmat(rho0*1.1, numel(PhiAC), 1), repmat(pi
        /4, numel(PhiAC), 1)];
     % 2nd try
260
         X0 = [repmat(rho0*1.1, numel(PhiAC), 1), repmat(
261
        xi_map(teE(bkk)), numel(PhiAC), 1)];
262
     % 3rd try
      X0 = [linspace(rho0, Dend/2, numel(PhiAC))', linspace(pi/4,
263
        xi_map(teE(bkk)), numel(PhiAC))'];
      RhoXi_AC = fsolve(@(x) PsiPhi(x(:,1),x(:,2), PsiAC, PhiAC,
264
        rho0), XO, options);
      RhoXi_BD = fsolve( @(x) PsiPhi( x(:,1),x(:,2), PsiBD, PhiBD,
265
        rho0), XO, options);
      R_AC = r_map(RhoXi_AC(:,1));
267
      te_AC = th_map(RhoXi_AC(:,2));
      R_BD = r_map(RhoXi_BD(:,1));
270
      te_BD = th_map(RhoXi_BD(:,2));
      if deb
272
         barrier(bkk).R_AC = R_AC/ScalingFactor;
273
        barrier(bkk).R_BD = R_BD/ScalingFactor;
274
275
        barrier(bkk).te_AC = te_AC;
        barrier(bkk).te_BD = te_BD;
276
      end
277
       % output of points
279
280
        barrier(bkk).Zeta = [...
      Zeta = [...]
281
         % top side
        xE(bkk) + 1j*yE(bkk)
        xOC(bkk) + 1j*yOC(bkk)
284
```

```
285
         xC(bkk) + 1j*yC(bkk)
         flipud( R_AC.*exp(1j*te_AC) )
286
         RA(bkk).*exp(1j*teA(bkk))
287
         % bottom side
288
         RB(bkk).*exp(1j*teB(bkk))
289
         R_BD.*exp(1j*te_BD)
290
         xD(bkk) + 1j*yD(bkk)
291
         xOD(bkk) + 1j*yOD(bkk)
         xE(bkk) + 1j*yE(bkk)
293
294
         ]/ScalingFactor;
       barrier(bkk).X = real(Zeta);
296
       barrier(bkk).Y = imag(Zeta);
297
       % magnet central base point
299
       barrier(bkk).Rm = Rmag(bkk)/ScalingFactor;
300
     end
302
     %% plot
304
     if deb
305
       % draw the rotor
307
       figure
       hold on
309
       tt = linspace(0,pi/p,50);
310
311
       plot(RO/ScalingFactor*cos(tt), RO/ScalingFactor*sin(tt), 'k');
       plot(Dr/2/ScalingFactor*cos(tt), Dr/2/ScalingFactor*sin(tt), ')
312
        k');
       axis equal
313
       % plot the flux-barrier central point
314
315
       plot(RA/ScalingFactor.*exp(1j*teA), 'rd')
       plot(RB/ScalingFactor.*exp(1j*teB), 'bo')
316
       plot(xE/ScalingFactor, yE/ScalingFactor,'ko')
318
       plot(x0C/ScalingFactor, y0C/ScalingFactor, 'go')
320
       plot(xC/ScalingFactor, yC/ScalingFactor, 'ro')
321
       plot(x0D/ScalingFactor, y0D/ScalingFactor, 'co')
       plot(xD/ScalingFactor, yD/ScalingFactor,'bo')
```

```
325
326
       % plot(R_AC.*exp(j*te_AC),'r.-')
       % plot(R_BD.*exp(j*te_BD),'b.-')
327
      for bkk = 1:Nb
329
         % plot flux-barrier sideline points
330
        plot(barrier(bkk).R_AC.*exp(1j*barrier(bkk).te_AC),'r.-')
        plot(barrier(bkk).R_BD.*exp(1j*barrier(bkk).te_BD),'b.-')
332
         % plot all the complete flux-barrier
334
        plot(barrier(bkk).X, barrier(bkk).Y, '.-')
335
336
      end
      pause(1e-3)
337
339
     end
341
     end
```

5.3 Draw fluid barrier

```
function draw_fluid_barrier(b)
    for bkk = 1:length(b)
      xE = b(bkk).X(1);
      yE = b(bkk).Y(1);
      xEOC = b(bkk).X(2);
      yEOC = b(bkk).Y(2);
      xC = b(bkk).X(3);
      yC = b(bkk).Y(3);
      xD = b(bkk).X(end-2);
11
      yD = b(bkk).Y(end-2);
12
      xDOE = b(bkk).X(end-1);
      yDOE = b(bkk).Y(end-1);
      X = b(bkk).X(3:end-2);
      Y = b(bkk).Y(3:end-2);
17
      mi_drawpolyline([X, Y])
19
      if xEOC == xC && yEOC == yC
        mi_drawline(xE,yE, xC,yC)
22
23
      else
        mi_draw_arc(xE,yE, xEOC,yEOC, xC,yC, 1)
      end
25
27
      if xDOE == xD && yDOE == yD
        mi_drawline(xD,yD, xE,yE)
      else
        mi_draw_arc(xD,yD, xDOE,yDOE, xE,yE, 1)
      end
33
    end
    end
```