

PX4 Autopilot

Firmware Guide (Spring-22)

Please keep editing if you find mistakes

Multicopter Control Architecture

Multicopter Control Architecture

- <http://docs.px4.io/master/en/concept/architecture.html>
- http://docs.px4.io/master/en/concept/px4_systems_architecture.html
- <https://www.youtube.com/watch?v=nEo4WGl4Lgc&t=118s>

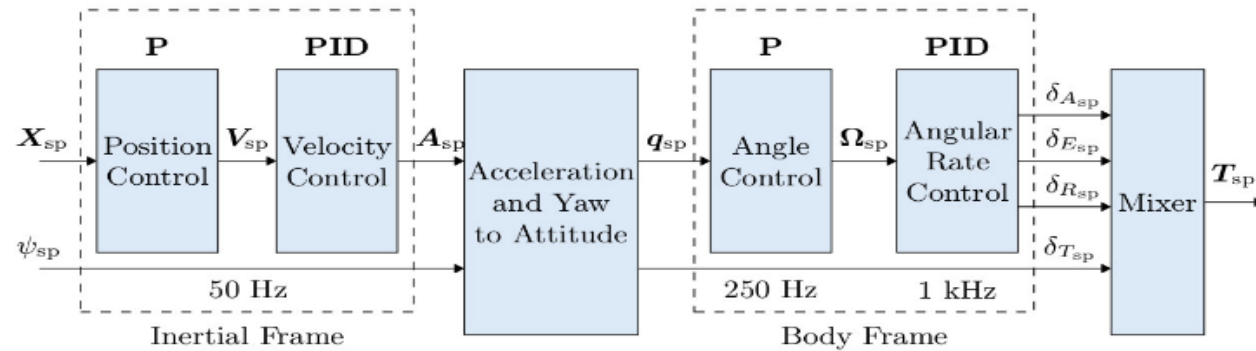
Multicopter Control Architecture

Controller Diagrams

This section contains diagrams for the main PX4 controllers.

The diagrams use the standard **PX4 notation** (and each have an annotated legend).

Multicopter Control Architecture



- This is a standard cascaded control architecture.
- The controllers are a mix of P and PID controllers.
- Estimates come from **EKF2**.
- Depending on the mode, the outer (position) loop is bypassed (shown as a multiplexer after the outer loop). The position loop is only used when holding position or when the requested velocity in an axis is null.

http://docs.px4.io/master/en/flight_stack/controller_diagrams.html

Github Repo

<https://github.com/PX4/PX4-Autopilot/blob/master/src/modules>

■ mc_pos_control

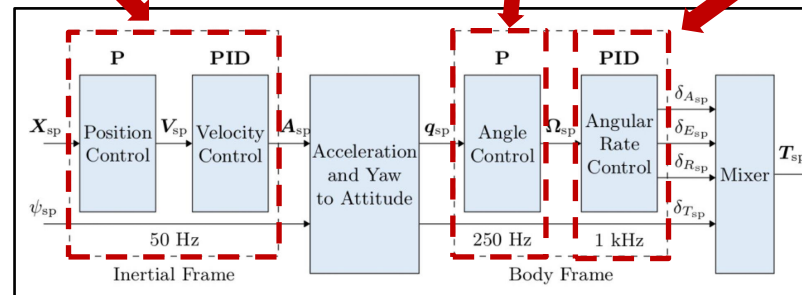
- PositionControl
 - a. PositionControl.cpp
 - b. PositionControl.hpp
- MulticopterPositionControl.cpp
- MulticopterPositionControl.hpp

■ mc_att_control

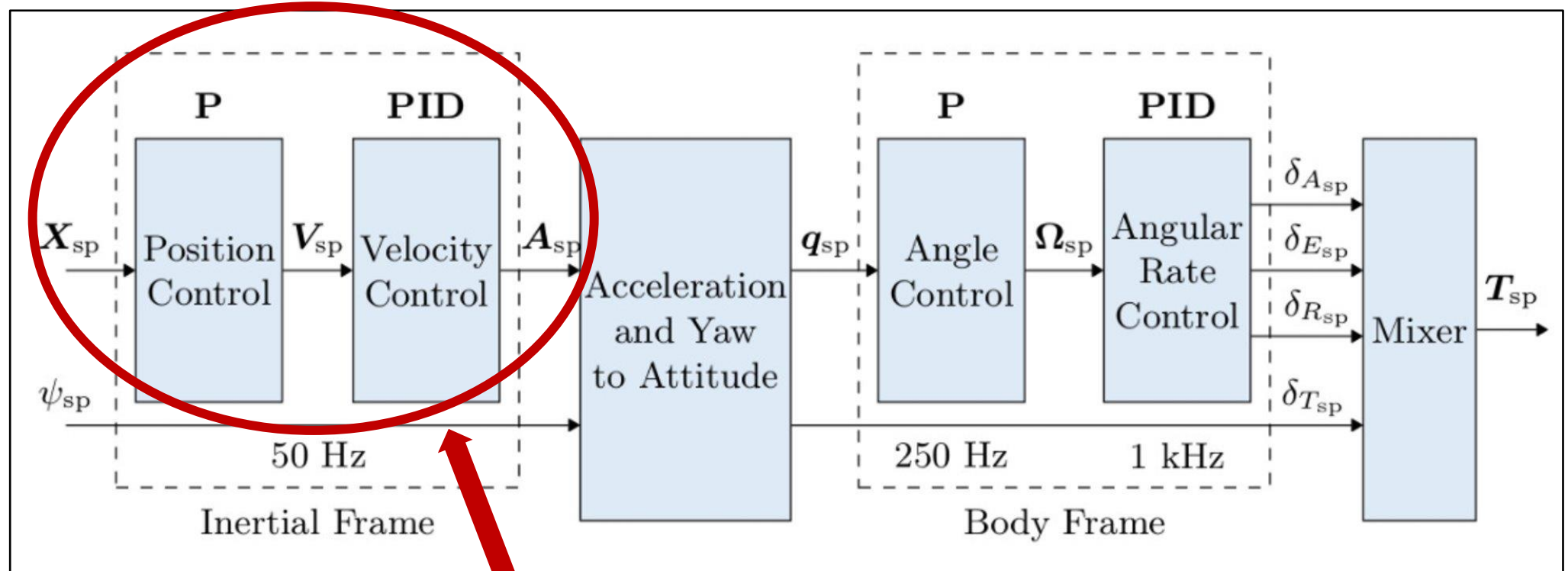
- AttitudeControl
 - a. AttitudeControl.cpp
 - b. AttitudeControl.hpp
- mc_att_control.cpp
- mc_att_control.hpp

■ mc_rate_control

- RateControl
 - a. RateControl.cpp
 - b. RateControl.hpp
- MulticopterRateControl.cpp
- MulticopterRateControl.hpp

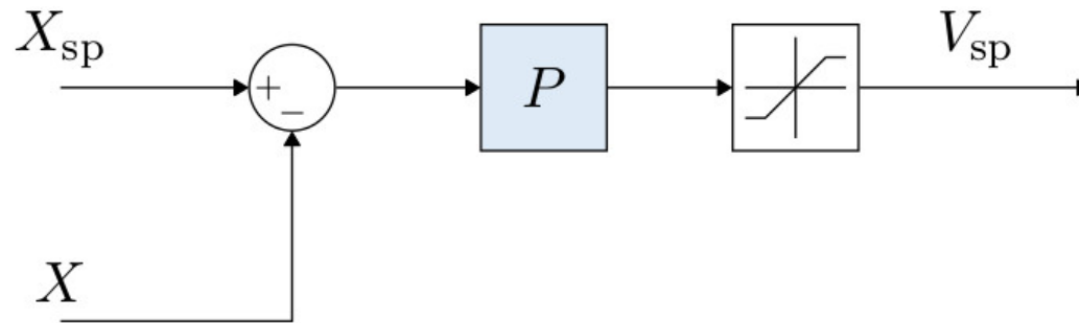


Position & Velocity Control



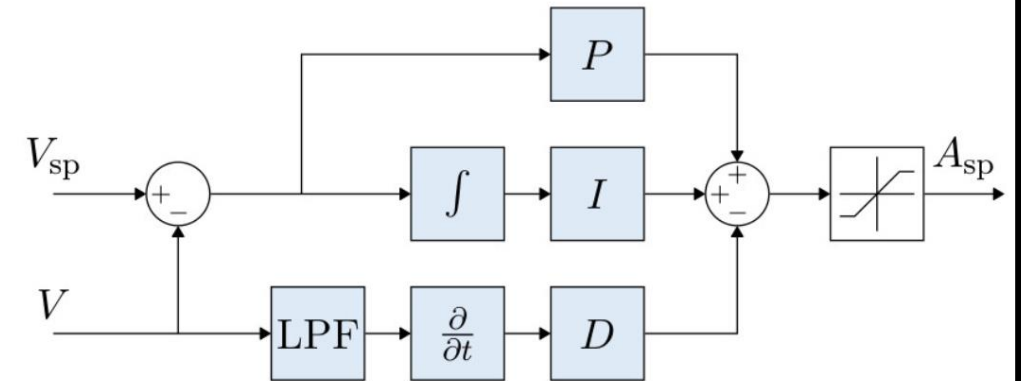
Position & Velocity Control

Multicopter Position Controller



- Simple P controller that commands a velocity.
- The commanded velocity is saturated to keep the velocity in certain limits.

Multicopter Velocity Controller

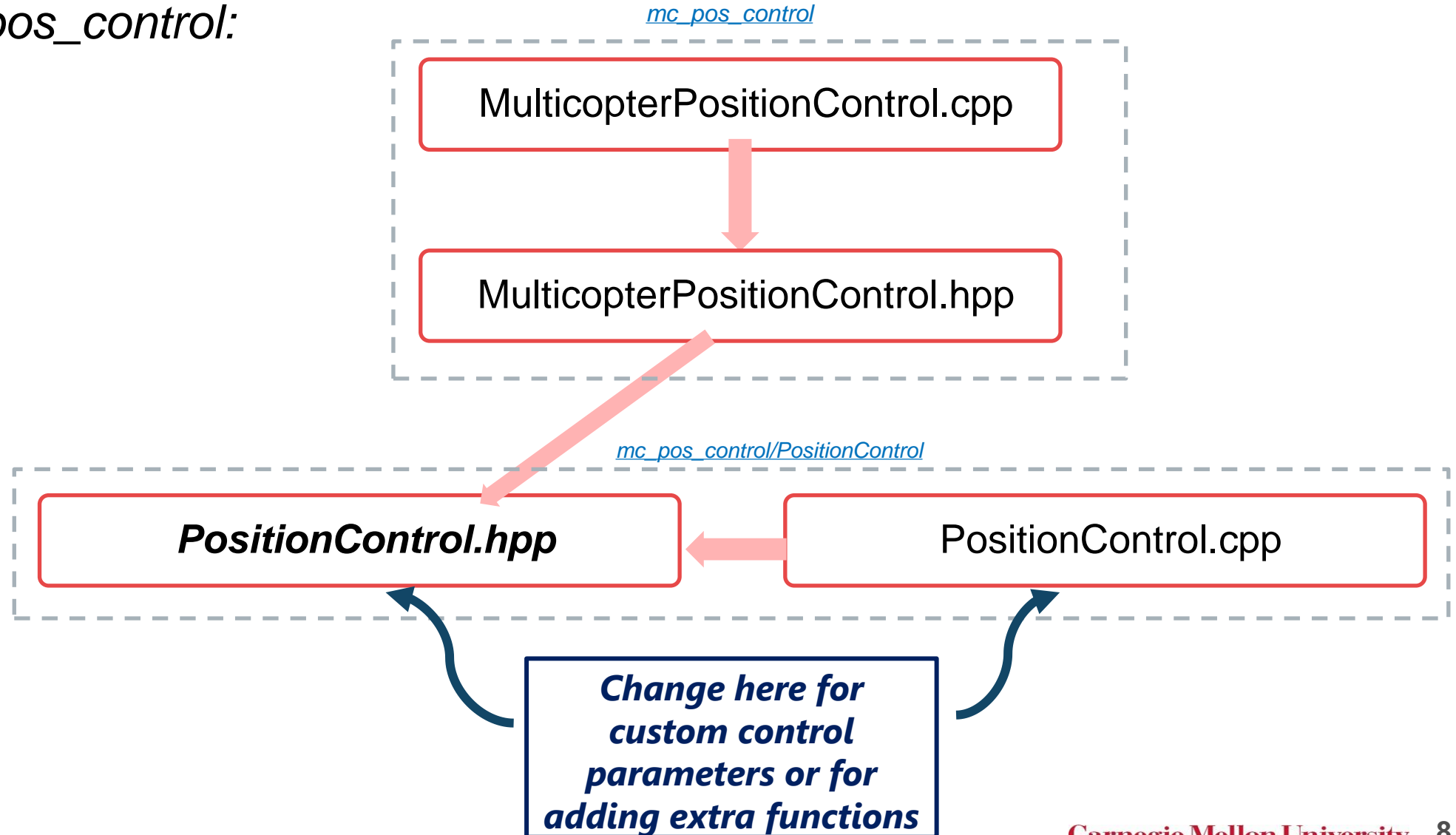


- PID controller to stabilise velocity. Commands an acceleration.
- The integrator includes an anti-reset windup (ARW) using a clamping method.
- The commanded acceleration is saturated.

http://docs.px4.io/master/en/flight_stack/controller_diagrams.html

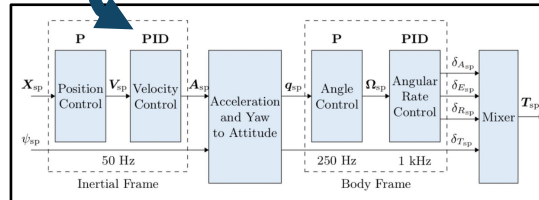
Codeflow

- *mc_pos_control*:



Setpoint Hierarchy

- If position-setpoint && velocity-setpoint true:
 - (Velocity component of P-Controller) >>> (feed-forward component from velocity-setpoint)
- If position/velocity-setpoint && thrust-setpoint true:
 - thrust-setpoint omitted and recomputed from next cascade/step in the architecture (i.e. Velocity PID controller)



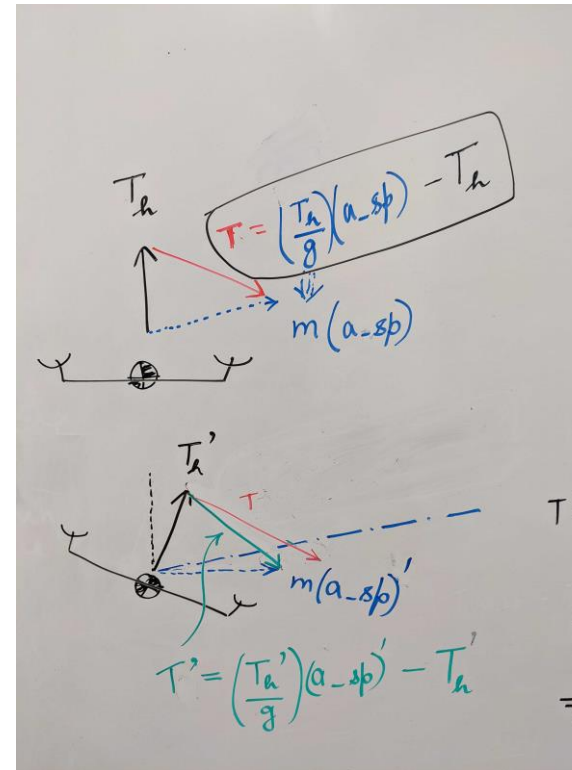
```

52  };
53
54  /**
55   *   Core Position-Control for MC.
56   *   This class contains P-controller for position and
57   *   PID-controller for velocity.
58   *   Inputs:
59   *       vehicle position/velocity/yaw
60   *       desired set-point position/velocity/thrust/yaw/yaw-speed
61   *       constraints that are stricter than global limits
62   *   Output
63   *       thrust vector and a yaw-setpoint
64   *
65   *   If there is a position and a velocity set-point present, then
66   *   the velocity set-point is used as feed-forward. If feed-forward is
67   *   active, then the velocity component of the P-controller output has
68   *   priority over the feed-forward component.
69   *
70   *   A setpoint that is NAN is considered as not set.
71   *   If there is a position/velocity- and thrust-setpoint present, then
72   *   the thrust-setpoint is omitted and recomputed from position-velocity-PID-loop.
73   */
74  class PositionControl
    
```

mc_pos_control/PositionControl/PositionControl.hpp

Hover Thrust Update

- `void PositionControl::updateHoverThrust(const float hover_thrust_new)`
 - Line 73-85



$$T' - T = \left\{ \frac{T'_h}{g}(a_{sp})' - T_h \right\} - \left\{ \frac{T_h}{g}(a_{sp}) - T_h \right\}$$

$$T' - T = T'_h \left(\frac{(a_{sp})'}{g} - 1 \right) - T_h \left(\frac{a_{sp}}{g} - 1 \right)$$

we want $T' - T \approx 0$

$$\Rightarrow T'_h \left(\frac{(a_{sp})'}{g} - 1 \right) = T_h \left(\frac{a_{sp}}{g} - 1 \right)$$

$$\Rightarrow \frac{T_h}{T'_h} = \frac{(a_{sp})' - g}{(a_{sp}) - g}$$

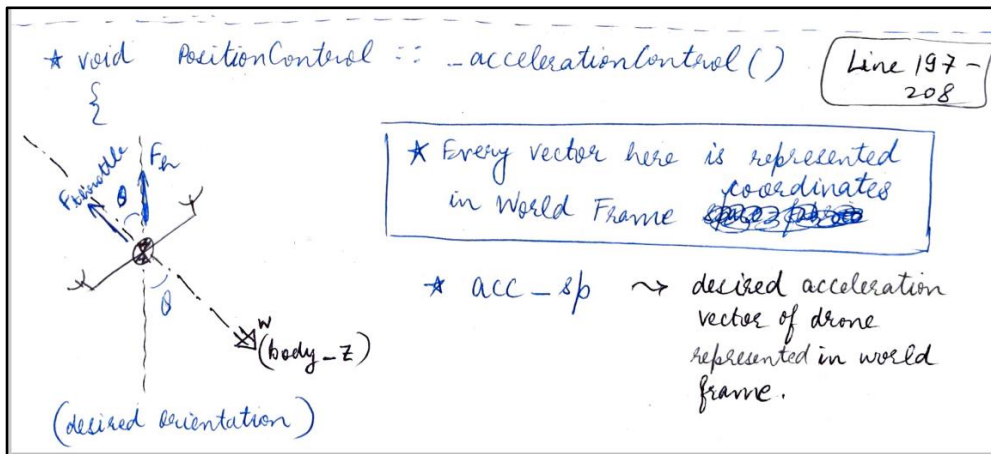
$$\Rightarrow \boxed{(a_{sp})' = \frac{T_h}{T'_h} \left\{ (a_{sp}) - g \right\} + g}$$

New set-point

`mc_pos_control/PositionControl/PositionControl.cpp`

Thrust Setpoint and Acceleration Setpoint (1)

- Output of Velocity-PID block
- void PositionControl::_accelerationControl
 - Line 197-208
 - This is called earlier in void PositionControl::_velocityControl()



★ body-Z → desired z-axis (or orientation) of drone represented in world-frame.

Line 200 ⇒
$$\text{body-Z} = \begin{bmatrix} \text{acc-sp}(0) \\ -\text{acc-sp}(1) \\ \text{CONSTANTS-G} \end{bmatrix}$$

For $\theta \approx \text{small};$

$F_1 = F_{\text{gravity}} = F_g$ (Equilibrium along vertical ^{world} axis)

$F_2 = F_{\text{thrust}} = F_{\text{throttle}}$

~~thrust is along axis~~

$|m a_z| = |F_g| + |F_{\text{extra}}|$ } along vertical world-Z

⇒ $|F_{\text{extra}}| = -|F_g| + |m a_z|$

LINE 203 ⇒ Collective-thrust $\equiv |F_{\text{extra}}|$

⇒ $|F_{\text{extra}}| = m(|\text{acc-sp}(2)|) - |F_g|$

we need to ~~project~~ project $|F_g|$ along drone-Z axis (represented in World Frame)

From figure; $|F_{\text{thrust}}| \cos \theta = |F_{\text{extra}}|$

mc_pos_control/PositionControl/PositionControl.cpp

Thrust Setpoint and Acceleration Setpoint (2)

- Output of Velocity-PID block
- void PositionControl::_accelerationControl
 - Line 197-208
 - This is called earlier in void PositionControl::_velocityControl()

③

$$\Rightarrow |F_{thrust}| = \frac{|F_{extra}|}{\cos \theta}$$

$$= \frac{|F_{extra}|}{\left\{ \frac{(\text{body-}z) \cdot \begin{bmatrix} 0 & 0 & 1 \end{bmatrix}^T}{\left\| \begin{bmatrix} 0 & 0 & 1 \end{bmatrix} \right\|} \right\}}$$

← (Line 203)
Line 205

Line 205 \Rightarrow $\text{collective_thrust} = \frac{\text{collective_thrust}}{|F_{thrust}|} \cdot |F_{extra}|$ (from line 203)

Line 207 \Rightarrow $_{thr_sp} = (\text{body-}z) \cdot \text{collective_thrust}$;
 unit vector from line 200, representing desired drone-orientation represented in the world frame!

Thrust-setpoint = $|F_{thrust}| \begin{bmatrix} -acc_sp(0) \\ -acc_sp(1) \\ \text{CONSTANTS-G} \end{bmatrix}$ (P.T.O)

Overall, ④

$$_{thr_sp} = \frac{\left\{ m(accel_sp(2)) - hover_th \right\} \begin{bmatrix} -acc_sp(0) \\ -acc_sp(1) \\ 9.81 \end{bmatrix}}{\left\{ \begin{bmatrix} -acc_sp(0) \\ -acc_sp(1) \\ 9.81 \end{bmatrix}^T \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \right\}}$$

Line 252
★ void PositionControl::getAttitudeSetpoint()
 {
 ControlMath::thrustToAttitude($_{thr_sp}$, yaw-sp, attitude-setpoint)
 attitude-setpoint.yaw-sp-move-rate = yaw-speed-sp;
 }

mc_pos_control/PositionControl/PositionControl.cpp

Thrust Setpoint To Rotation Matrix

- /PositionControl/ControlMath.cpp
- void thrustToAttitude()
 - Line 49
- void bodyzToAttitude()
 - Line 70 onwards
 - "R" -> Line 104 – 108:
 - **Mapping from body-frame coordinates to world-coordinates....according to Zac's notation -> $X_{world} = R * X_{body}$.*

Overall, ④

$$-thr_sp = \frac{\{m[acc_sp(2)] - hover_th\} \begin{bmatrix} -acc_sp(0) \\ -acc_sp(1) \\ 9.81 \end{bmatrix}}{\left\{ \begin{bmatrix} -acc_sp(0) \\ -acc_sp(1) \\ 9.81 \end{bmatrix}^T \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \right\}}$$

Line 252

```

★ void PositionControl::getAttitudeSetpoint(
{
    ControlMath::thrustToAttitude(-thr_sp, yaw_sp, attitude_setpoint);
    attitude_setpoint.yaw_sp_move_rate = yaw_speed_sp;
}
    
```

This is probably just a message topic

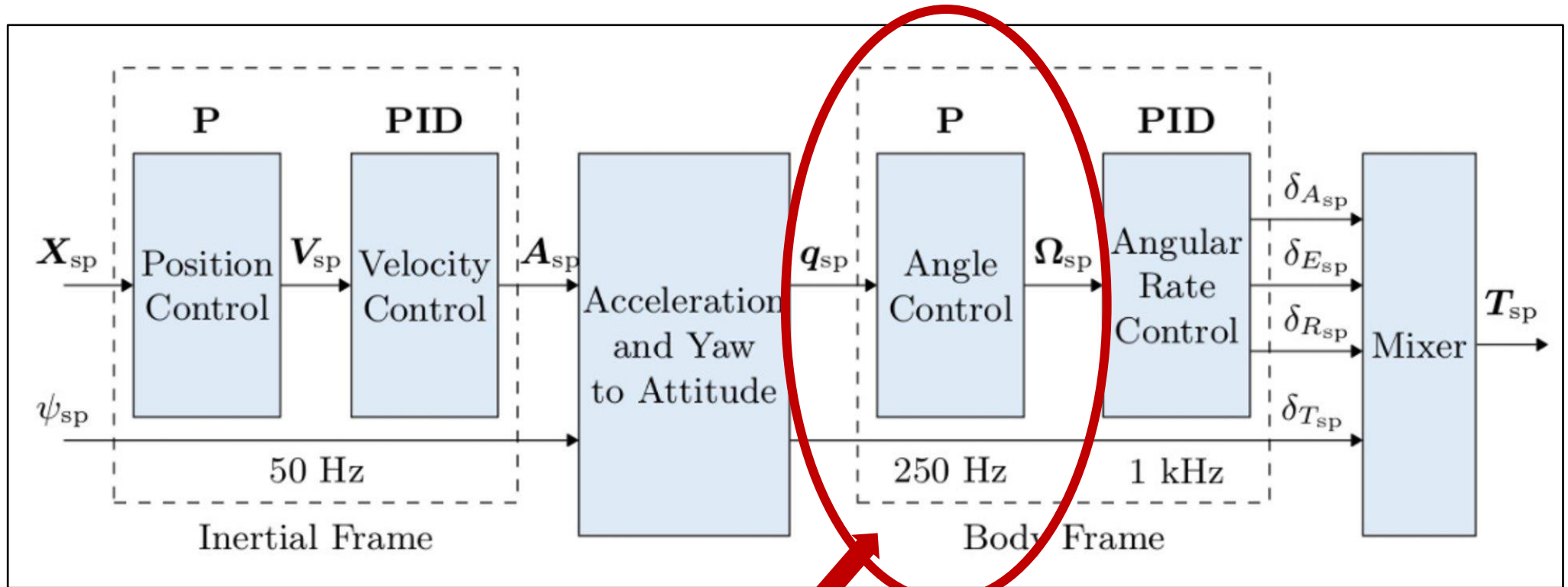
mc_pos_control/PositionControl/PositionControl.cpp

How is all of this called?

- `/mc_pos_control/MulticopterPositionControl.cpp`
- `void MulticopterPositionControl::Run()`
 - `_control` object, throughout this file, refers to `"/PositionControl/PositionControl.cpp"`
 - Line 474 and 495 calls `_control.update` that triggers the Position-Control algorithm!

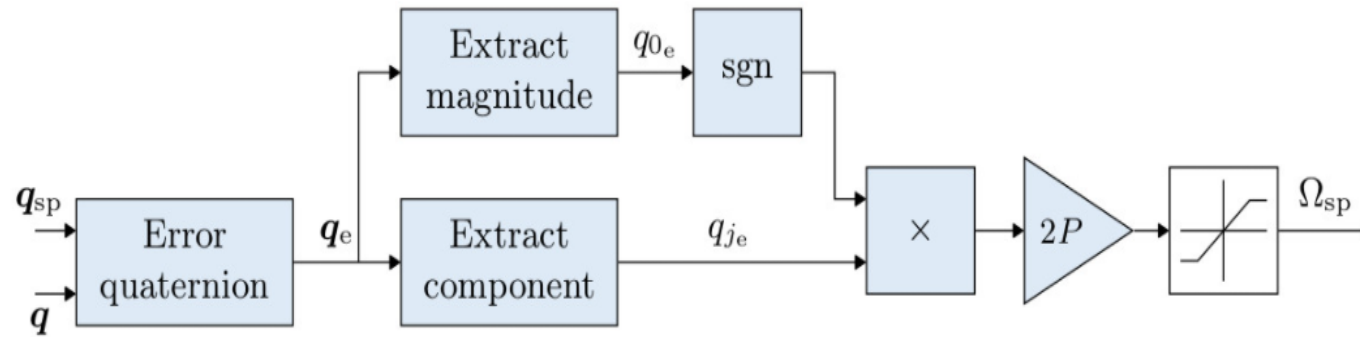
`mc_pos_control/MulticopterPositionControl.cpp`

Attitude Control



Attitude Control

Multicopter Attitude Controller



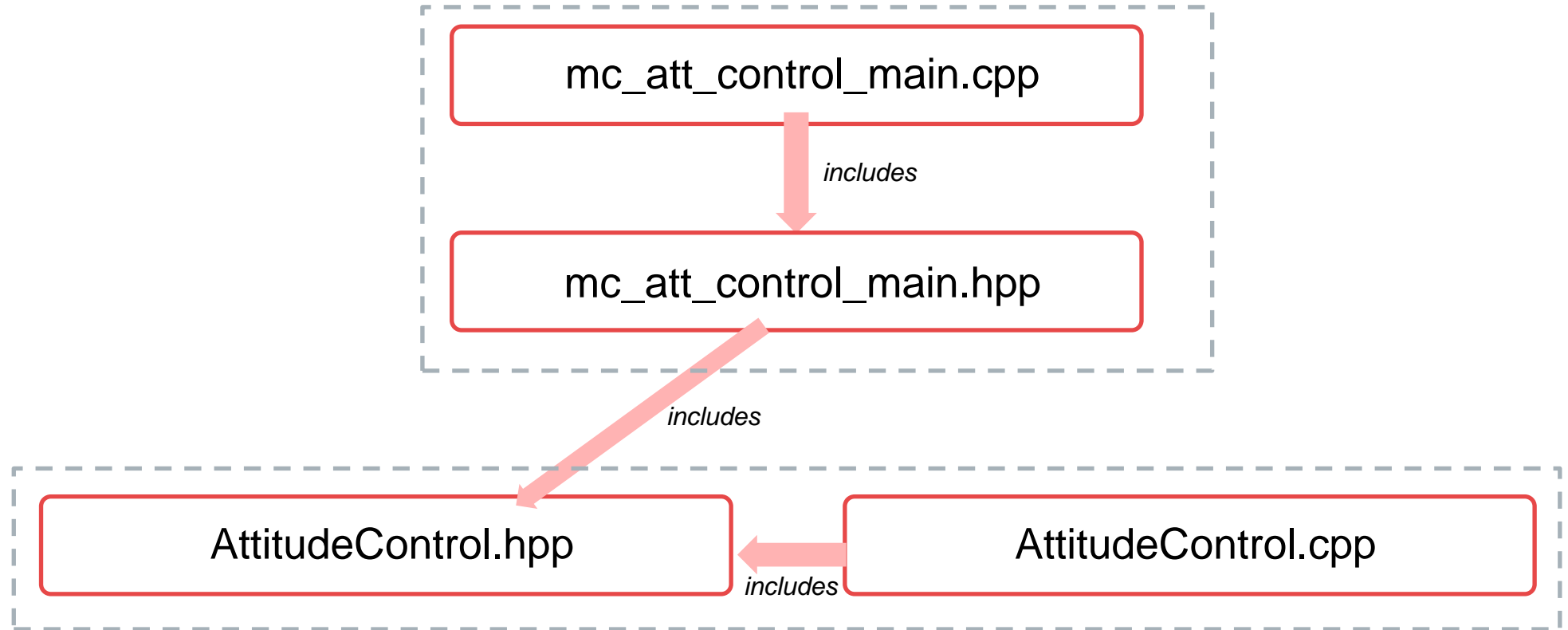
- The attitude controller makes use of [quaternions](#) .
- The controller is implemented from this [article](#) .
- When tuning this controller, the only parameter of concern is the P gain.
- The rate command is saturated.

http://docs.px4.io/master/en/flight_stack/controller_diagrams.html

***Refer this link for an additional note on IMU pipeline*

Codeflow

- *mc_att_control*: (quaternion based control : <https://www.research-collection.ethz.ch/bitstream/handle/20.500.11850/154099/eth-7387-01.pdf>)



Mc_att_control_main.cpp

generate_attitude_setpoint

- `/mc_att_control/mc_att_control_main.cpp`
- `void MulticopterAttitudeControl::generate_attitude_setpoint()`
 - Line 155 – 157
 - ✓ For axis angle representation, refer this: 1) https://en.wikipedia.org/wiki/Axis%E2%80%93angle_representation
 - ✓ 2) <https://github.com/Optimal-Control-16-745/lecture-notebooks-2021/blob/main/Lecture%2013/Lecture%2013.pdf>
 - Line 165, 166, 177 spits out the Roll, Pitch and Yaw setpoints -->

<ul style="list-style-type: none">• <code>attitude_setpoint.roll_body</code>• <code>attitude_setpoint.pitch_body</code>• <code>attitude_setpoint.yaw_body</code>
--
- `void MulticopterAttitudeControl::Run()`
 - **Line 304 – 308**
 - **Generates attitude setpoints if we are in Manual/Stabilized mode!**
 - Line 310 calls "`generate_attitude_setpoint()`" mentioned above
 - Line 318 calls "`_attitude_control.update(q)`". This belongs to "AttitudeControl.cpp" and is discussed in next slides

AttitudeControl.cpp

AttitudeControl.cpp

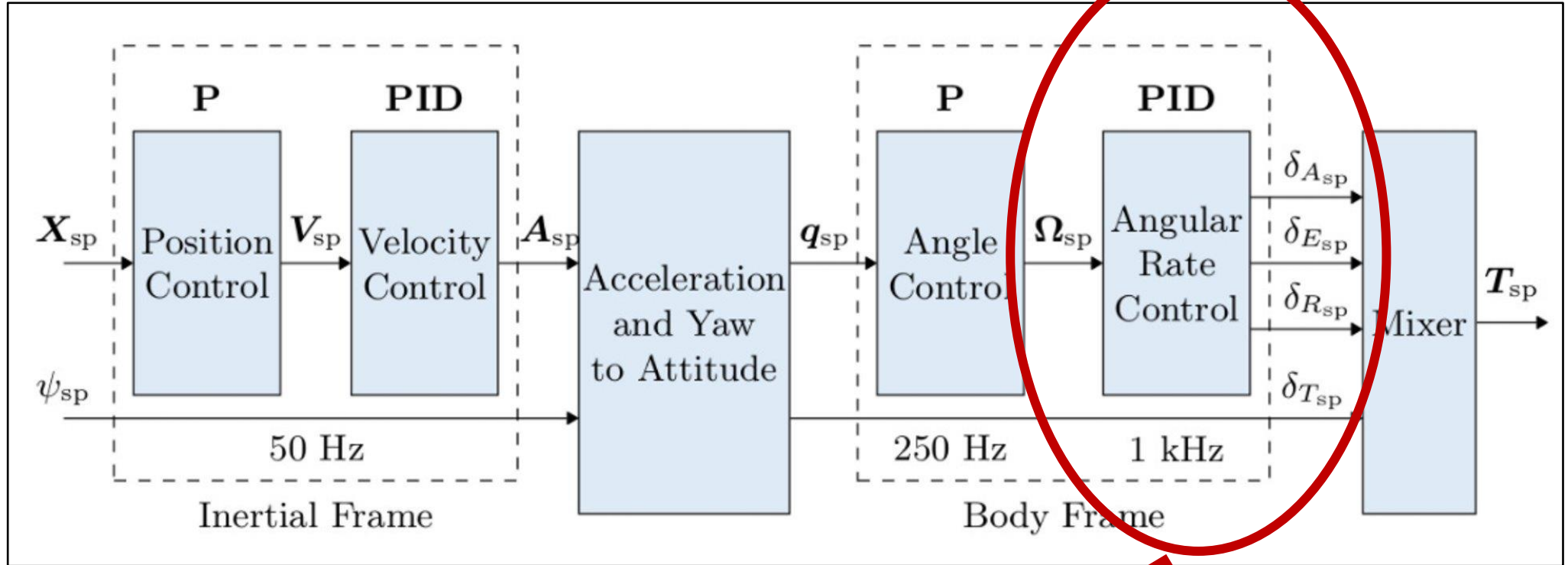
- */mc_att_control/AttitudeControl/AttitudeControl.cpp*
- `void AttitudeControl::setProportionalGain()`
- `void AttitudeControl::update()`
 - Refer <https://github.com/PX4/PX4-Autopilot/blob/master/src/lib/matrix/matrix/Quaternion.hpp>

mc_att_control/M.cpp

Quaternion Starter-Pack

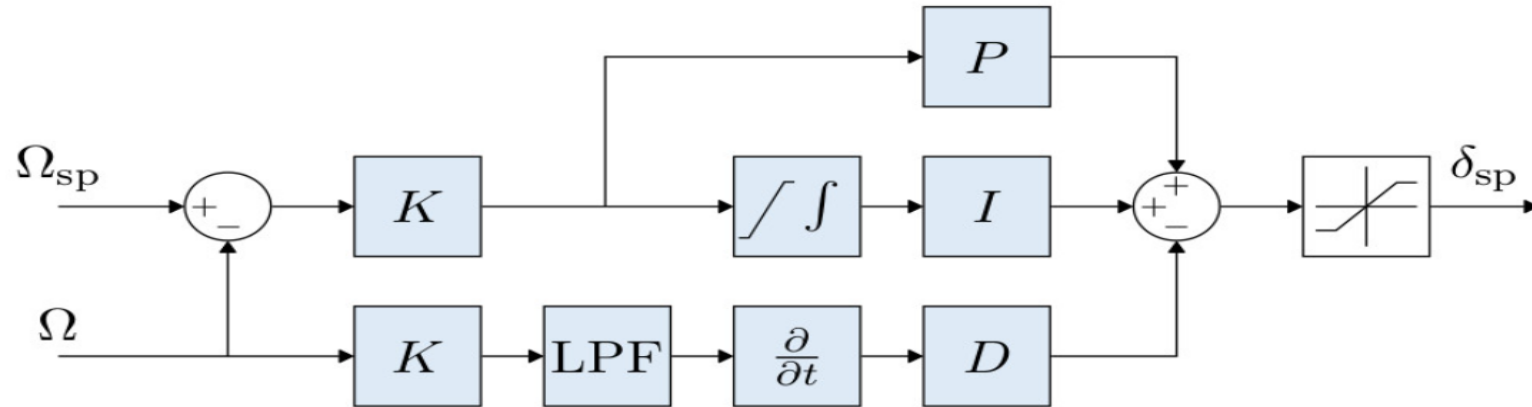
- <https://www.youtube.com/watch?v=zjMulxRvygQ>
- <https://github.com/Optimal-Control-16-745/lecture-notebooks-2021/blob/main/Lecture%2013/Lecture%2013.pdf>
- <https://github.com/Optimal-Control-16-745/lecture-notebooks-2021/blob/main/Lecture%2014/Lecture%2014.pdf>
- <https://www.research-collection.ethz.ch/bitstream/handle/20.500.11850/154099/eth-7387-01.pdf>
- <https://ieeexplore.ieee.org/document/9326337>

Angular Rate Control



Angular Rate Control

Multicopter Angular Rate Controller



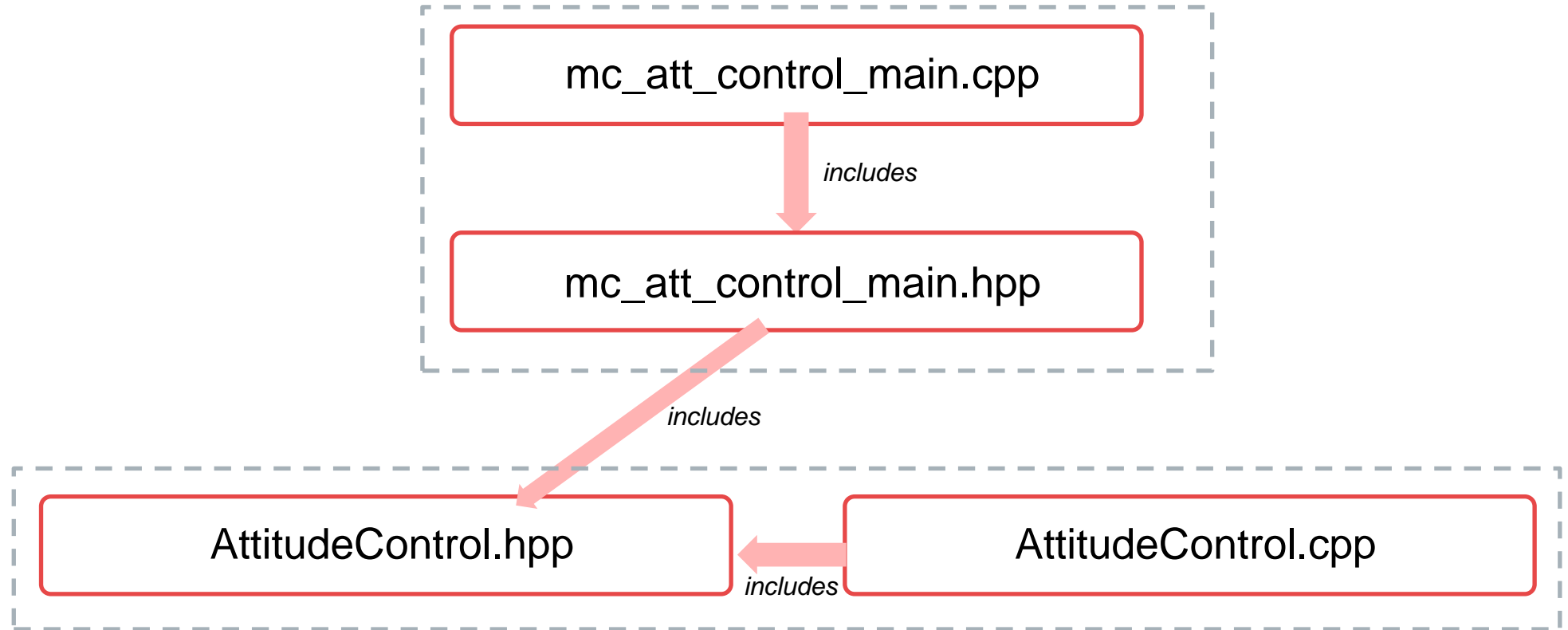
- K-PID controller. See [Rate Controller](#) for more information.
- The integral authority is limited to prevent wind up.
- The outputs are limited (in the mixer), usually at -1 and 1.
- A Low Pass Filter (LPF) is used on the derivative path to reduce noise (the gyro driver provides a filtered derivative to the controller).

http://docs.px4.io/master/en/flight_stack/controller_diagrams.html

***Refer this link for an additional note on IMU pipeline*

Codeflow

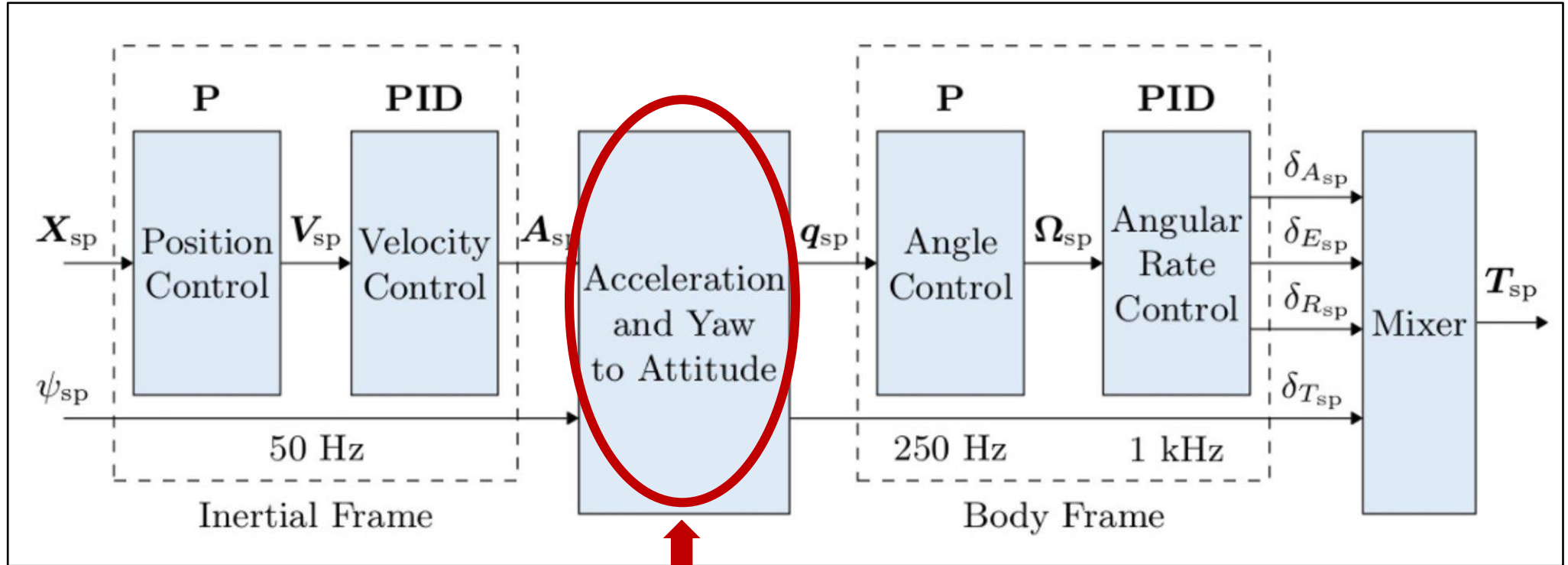
- *mc_rate_control*: (quaternion based control : <https://www.research-collection.ethz.ch/bitstream/handle/20.500.11850/154099/eth-7387-01.pdf>)



MulticopterRateControl.cpp

RateControl.cpp

Attitude Setpoint



UUV Control Architecture *(coming soon...)*