

## Dynamics:

Newton's laws:

everything in inertial (or world) frame!

$$\textcircled{1} \quad \sum^w \vec{F}_i = \frac{d^w \vec{p}}{dt}$$

$$\textcircled{2} \quad \sum^w \vec{\tau}_i = \frac{d^w \vec{l}}{dt}$$

$$\begin{aligned} \sum^w \vec{F}_i &= {}^w \vec{F} = \frac{d(m^w \vec{v})}{dt} = m \frac{d^w \vec{v}}{dt} \\ &= m^w \vec{v} \end{aligned}$$

↳ ①

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$${}^w \vec{v} = Q^B \vec{v}$$

$${}^w \vec{v} = \dot{Q}^B \vec{v} + Q^B \vec{\dot{v}}$$

${}^B \vec{\dot{x}} = \vec{\omega}$  (by definition of  
but  ${}^B \vec{v} \neq 0$  rigid body)

also,  $\dot{Q} = Q^B \vec{\omega}$  (see "Rotations.pdf") }

$$\Rightarrow {}^w \vec{F} = m \left( \dot{Q}^B \vec{v} + Q^B \vec{\dot{v}} \right)$$

$$= m \left( Q^B \vec{\omega} \wedge {}^B \vec{v} + Q^B \vec{\dot{v}} \right)$$

$${}^w \vec{F} = m Q \left( {}^B \vec{\omega} \times {}^B \vec{v} + {}^B \vec{\dot{v}} \right)$$

$${}^w \vec{F} = m Q^T Q / {}^B \vec{v} \times {}^B \vec{\dot{v}} + {}^B \vec{\ddot{v}}$$

$$Q^T \dot{F} = m \underbrace{\dot{Q} Q}_{\downarrow} (\omega \times \vec{v} + \vec{v})$$

$= I_{3 \times 3}$  by definition  
of Rotation Matrix

$$\Rightarrow \boxed{^B \dot{F} = m \left( {}^B \vec{\omega} \times {}^B \vec{v} + {}^B \vec{v} \right)}$$

②  $\left\{ \begin{array}{l} {}^W \vec{l} = Q {}^B \vec{l} \\ \Rightarrow {}^W \dot{\vec{l}} = \dot{Q} {}^B \vec{l} + Q {}^B \dot{\vec{l}} \end{array} \right.$

we know,  $\dot{Q} = Q {}^B \hat{\omega}$

$$\Rightarrow {}^W \dot{\vec{l}} = Q {}^B \hat{\omega} {}^B \vec{l} + Q {}^B \dot{\vec{l}} \quad \}$$

$$\Rightarrow \sum {}^W \vec{\tau}_i = {}^W \vec{\tau} = \frac{d {}^W \vec{l}}{dt} = {}^W \dot{\vec{l}}$$

$${}^W \vec{\tau} = Q \left( {}^B \hat{\omega} ({}^B \vec{l}) + {}^B \dot{\vec{l}} \right)$$

$$\underbrace{Q^T {}^W \vec{\tau}}_{^B \vec{\tau}} = \underbrace{Q^T Q}_{I_{3 \times 3}} \left( {}^B \hat{\omega} {}^B \vec{l} + {}^B \dot{\vec{l}} \right)$$

$$^B \vec{\tau} = I_{3 \times 3}$$

$${}^B\vec{\gamma} = {}^B\hat{\omega} {}^B\vec{l} + {}^B\dot{\vec{l}}$$

$$\left\{ {}^B\vec{l} = \underbrace{\begin{bmatrix} J & {}^B\vec{\omega} \\ \downarrow & [ ] \\ 3 \times 3 & 3 \times 1 \end{bmatrix}}_{\text{[ ]}} \Rightarrow {}^B\dot{\vec{l}} = \underbrace{\begin{bmatrix} J & {}^B\vec{\omega} \\ \downarrow & [ ] \\ = 0 & 3 \times 1 \end{bmatrix}}_{\text{[ ]}} + J \cdot {}^B\vec{\dot{\omega}} \right.$$

becuz, Inertia tensor  
doesn't change in  
Body frame

$$\therefore {}^B\dot{\vec{l}} = J \cdot {}^B\vec{\dot{\omega}} \quad \left. \right\}$$

$$\Rightarrow {}^B\vec{\gamma} = {}^B\hat{\omega} {}^B\vec{l} + {}^B\vec{\dot{l}}$$

$$= {}^B\hat{\omega} J {}^B\vec{\omega} + J \cdot {}^B\vec{\dot{\omega}}$$

$$\left\{ \vec{a} \times \vec{b} = \begin{bmatrix} \hat{a} & b \\ \downarrow & [ ] \\ 3 \times 3 & 3 \times 1 \end{bmatrix} \right\}$$

$$\Rightarrow \boxed{{}^B\vec{\gamma} = {}^B\vec{\omega} \times J {}^B\vec{\omega} + J \cdot {}^B\vec{\dot{\omega}}}$$

$${}^B\vec{F} = m \left( {}^B\vec{\omega} \times {}^B\vec{v} + {}^B\vec{\dot{v}} \right)$$

$$\left. \begin{array}{l} {}^B\vec{\gamma} = {}^B\vec{\omega} \times J {}^B\vec{\omega} + J \cdot {}^B\vec{\dot{\omega}} \\ \\ \begin{bmatrix} \gamma_x \\ \gamma_y \\ \gamma_z \end{bmatrix} = \begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} \times \begin{bmatrix} J_{xx} & & & \omega_x \\ & J_{yy} & & \omega_y \\ & & J_{zz} & \omega_z \end{bmatrix} \end{array} \right\}$$

$$+ J \cdot \begin{bmatrix} \omega_n \\ \omega_n \\ \omega_n \end{bmatrix}$$

$$\begin{bmatrix} \dot{\omega}_x \\ \dot{\omega}_y \\ \dot{\omega}_z \end{bmatrix}$$

$$= \begin{bmatrix} \hat{i} & \hat{j} & \hat{k} \\ \omega_x & \omega_y & \omega_z \\ J_{xx}\omega_x & J_{yy}\omega_y & J_{zz}\omega_z \end{bmatrix}$$

$$+ \begin{bmatrix} J_{xx}^B \dot{\omega}_x \\ J_{yy}^B \dot{\omega}_y \\ J_{zz}^B \dot{\omega}_z \end{bmatrix}$$

$$= \begin{bmatrix} (J_{zz} - J_{yy})\omega_y \omega_z \\ (J_{xx} - J_{zz})\omega_x \omega_z \\ (J_{yy} - J_{xx})\omega_x \omega_y \end{bmatrix}$$

$$+ \begin{bmatrix} J_{xx} & \dot{\omega}_x \\ J_{yy} & \dot{\omega}_y \\ J_{zz} & \dot{\omega}_z \end{bmatrix}$$

$$\begin{bmatrix} {}^B\tau_x \\ {}^B\tau_y \\ {}^B\tau_z \end{bmatrix}$$

$$= \begin{bmatrix} (J_{zz} - J_{yy})\omega_y \omega_z + \dot{\omega}_x J_{xx} \\ (J_{xx} - J_{zz})\omega_x \omega_z + \dot{\omega}_y J_{yy} \\ (J_{yy} - J_{xx})\omega_x \omega_y + \dot{\omega}_z J_{zz} \end{bmatrix}$$

choice = 1

choice = ?

## Drone State Space:

$$X = \begin{bmatrix} {}^W\vec{r}_c \\ q \\ {}^B\vec{v} \\ {}^B\vec{\omega} \end{bmatrix}$$

(3x1)      (4x1)      (3x1)      (3x1)

13x1

$$X = \begin{bmatrix} {}^W\vec{r}_c \\ q \\ {}^W\vec{v} \\ {}^B\vec{\omega} \end{bmatrix}$$

13x1

## choice - 1: Dynamics:

$$\dot{X} = f(X, u)$$

$$\dot{X} = \begin{bmatrix} \dot{{}^W\vec{r}_c} \\ \dot{q} \\ \dot{{}^B\vec{v}} \\ \dot{{}^B\vec{\omega}} \end{bmatrix} \rightsquigarrow \begin{aligned} &= \begin{bmatrix} {}^Q{}^B\vec{v} \\ \frac{1}{2}(q \otimes [0 \quad {}^B\vec{\omega}]) \\ \frac{{}^B\vec{F}}{m} - {}^B\vec{\omega} \times {}^B\vec{v} \end{bmatrix} \\ &\rightsquigarrow \text{extract from } {}^B\vec{\tau} = {}^B\vec{\omega} \times J {}^B\vec{\omega} + J {}^B\vec{c}\vec{\omega} \end{aligned}$$

## choice - 2:

$$\dot{X} = \begin{bmatrix} \dot{{}^W\vec{r}_c} \\ \dot{q} \\ \dot{{}^W\vec{v}} \\ \dot{{}^B\vec{\omega}} \end{bmatrix} \rightsquigarrow \begin{aligned} &= \begin{bmatrix} {}^W\vec{v} \\ \frac{1}{2}(q \otimes [0 \quad {}^B\vec{\omega}]) \\ Q {}^B\vec{v} = Q \left( \frac{{}^B\vec{F}}{m} - {}^B\vec{\omega} \times {}^B\vec{v} \right) \end{bmatrix} \\ &\rightsquigarrow \text{extract from } {}^B\vec{\tau} = {}^B\vec{\omega} \times J {}^B\vec{\omega} + J {}^B\vec{c}\vec{\omega} \end{aligned}$$

${}^W\vec{r}_c \Rightarrow$  Motion capture

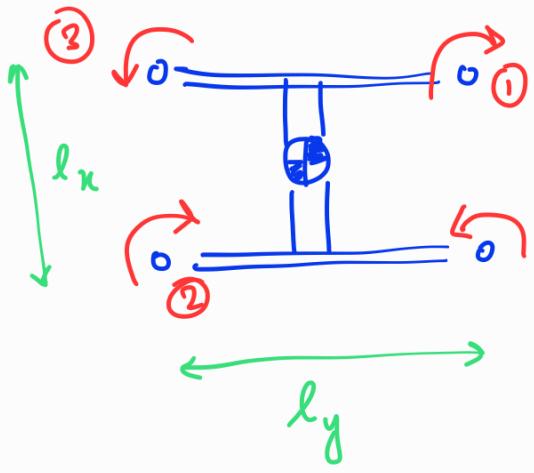
$q \rightarrow$  integrating Rotation Matrix "Q" from known initial orientation "Q<sub>0</sub>"

will give us orientation ( $Q$  or quaternion "q") at any future time given that we have IMU (gyroscope) readings ( ${}^B\vec{\omega}_i$ ) at every time step.

${}^B\vec{v}$  → motion capture?  
 Integration?

${}^B\vec{\omega}$  → gyroscope readings of IMU sensor.

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 Finding  ${}^B\vec{F}$  &  ${}^B\vec{\tau}$  for a drone...



$${}^B\vec{F} = \vec{f}_1 + \vec{f}_2 + \vec{f}_3 + \vec{f}_4 + \vec{F}_{wind}$$

$${}^B\vec{\tau}_x = {}_w\vec{M}_x + \{ F_1 + F_3 - (F_2 + F_4) \}$$

$${}^B\vec{\tau}_y = {}_w\vec{M}_y + \{ F_1 + F_3 - (F_2 + F_4) \}$$

$${}^B\vec{\tau}_z = {}_w\vec{M}_z + \{ M_1 + M_2 - (M_3 + M_4) \}$$

Yaw Torque of propellers working against air

If wind induces moments too,

*then*  $\overset{\rightarrow}{wM_x}$   $\overset{\rightarrow}{wM_y}$   $\overset{\rightarrow}{wM_z}$  are moments  
generated due  
to wind)