

Chapter 15

15.1 Eq. (15.15):

$$F_s = \frac{c'}{\gamma H \cos^2 \beta \tan \beta} + \frac{\tan \phi'}{\tan \beta}$$

$$2 = \frac{15}{(15.5)(H)(\cos^2 30)(\tan 30)} + \frac{\tan 20}{\tan 30}$$

$$2 = \frac{2.235}{H} + 0.63$$

$$H = \mathbf{1.63 \text{ m}}$$

15.2 Eq. (15.16):

$$H_{cr} = \frac{c'}{\gamma \cos^2 \beta (\tan \beta - \tan \phi')} = \frac{200}{100} \frac{1}{(\cos^2 22)(\tan 22 - \tan 15)} = \mathbf{17.1 \text{ ft}}$$

15.3 $\gamma_{sat} = \frac{(1900)(9.81)}{1000} = 18.64 \text{ kN/m}^3$

$$\gamma' = 18.64 - 9.81 = 8.83 \text{ kN/m}^3$$

Eq. (15.17):

$$F_s = \frac{c'}{\gamma_{sat} H \cos^2 \beta \tan \beta} + \frac{\gamma' \tan \phi'}{\gamma_{sat} \tan \beta} = \frac{18}{(18.64)(8)(\cos 20)^2 (\tan 20)} + \frac{8.83 \tan 25}{18.64 \tan 20}$$

$$= 0.376 + 0.607 = \mathbf{0.98}$$

15.4 $\gamma_{sat} = \frac{(G_s + e)\gamma_w}{1 + e} = \frac{(2.6 + 0.5)(62.4)}{1 + 0.5} = 128.96 \text{ lb/ft}^3$

$$\gamma' = 128.96 - 62.4 = 66.56 \text{ lb/ft}^3$$

$$\begin{aligned}
 F_s &= \frac{c'}{\gamma_{\text{sat}} H \cos^2 \beta \tan \beta} + \frac{\gamma' \tan \phi'}{\gamma_{\text{sat}} \tan \beta} \\
 &= \frac{600}{(128.96)(25)(\cos^2 20)(\tan 20)} + \frac{66.56 \tan 22}{128.96 \tan 20} \\
 &= 0.579 + 0.573 \approx \mathbf{1.15}
 \end{aligned}$$

15.5 Eq. (15.31):

$$H_{\text{cr}} = \frac{4c'}{\gamma} \left[\frac{\sin \beta \cos \phi'}{1 - \cos(\beta - \phi')} \right] = \left[\frac{(4)(400)}{115} \right] \left[\frac{(\sin 50)(\cos 25)}{1 - \cos(50 - 25)} \right] = \mathbf{103.1 \text{ ft}}$$

15.6 $F_s = 2$; $c'_d = \frac{c'}{F_s} = \frac{400}{2} = 200 \text{ lb/ft}^2$; $\phi'_d = \tan^{-1} \left(\frac{\tan 25}{2} \right) = 13.12^\circ$

$$H = \frac{4c'_d}{\gamma} \left[\frac{\sin \beta \cos \phi'_d}{1 - \cos(\beta - \phi'_d)} \right] = \left[\frac{(4)(200)}{115} \right] \left[\frac{(\sin 50)(\cos 13.12)}{1 - \cos(50 - 13.12)} \right] = \mathbf{25.94 \text{ ft}}$$

15.7 $H = \frac{4c'_d}{\gamma} \left[\frac{\sin \beta \cos \phi'_d}{1 - \cos(\beta - \phi'_d)} \right]$; $\gamma = 115 \text{ lb/ft}^3$

$F_{s(\text{assumed})}$	$\phi'_d = \tan^{-1} \left(\frac{\tan 15}{F_s} \right)$	$c'_d = \frac{c'}{F_s}$	β	H
	(deg)	(lb/ft ²)	(deg)	(m)
1.6	9.5	125	50	13.7
1.8	8.47	111	50	11.63
1.76	8.66	113.6	50	12.0

$F_s \approx \mathbf{1.76}$

15.8 $\rho = 1700 \text{ kg/m}^3$; $\gamma = \frac{(1700)(9.81)}{1000} = 16.68 \text{ kN/m}^3$; $c' = 18 \text{ kN/m}^2$; $\phi' = 20^\circ$

$$\beta = \tan^{-1} \left(\frac{1}{2} \right) = 26.57^\circ$$

$F_s = 2$; $c'_d = \frac{c'}{F_s} = \frac{18}{2} = 9 \text{ kN/m}^2$

$$\phi'_d = \tan^{-1}\left(\frac{\tan \phi'}{F_s}\right) = \tan^{-1}\left(\frac{\tan 20}{2}\right) = 10.31^\circ$$

$$H = \frac{4c'_d}{\gamma} \left[\frac{\sin \beta \cos \phi'_d}{1 - \cos(\beta - \phi'_d)} \right] = \left[\frac{(4)(9)}{16.68} \right] \left[\frac{(\sin 26.57)(\cos 10.31)}{1 - \cos(26.57 - 10.31)} \right] = \mathbf{23.74 \text{ m}}$$

15.9 $m \approx 0.192$ (From Figure 15.13). Eq. (15.36):

$$H_{cr} = \frac{c_u}{\gamma m} = \frac{30}{(17)(0.192)} = \mathbf{9.19 \text{ m} - \text{Toe failure}}$$

15.10 $m \approx 0.192$ for $\beta = 60^\circ$ (Figure 15.13)

$$c_d = \frac{c_u}{FS_s} = \frac{30}{2} = 15 \text{ kN/m}^2$$

$$H = \frac{c_d}{\gamma m} = \frac{15}{(17)(0.192)} = \mathbf{4.6 \text{ m}}$$

15.11 $\beta = \tan^{-1}\left[\frac{1}{1}\right] = 45^\circ$. For $\beta = 45^\circ$ and $D = 1.2$, $m = 0.18$ (Figure 15.13).

$$c_d = \frac{c_u}{FS_s} = \frac{25}{2} = 12.5 \text{ kN/m}^2$$

$$H = \frac{c_d}{\gamma m} = \frac{12.5}{(17)(0.18)} = \mathbf{3.86 \text{ m}}$$

15.12 $H_{cr} = \frac{c_u}{\gamma m} = \frac{25}{(18)(0.18)} = \mathbf{7.72 \text{ m}}$. It is a **toe circle**.

15.13 a. $D = \frac{12}{8.5} = 1.41$; $\gamma_{sat} = 18.5 \text{ kN/m}^3$. For $\beta = 40^\circ$ and $D = 1.41$, $m = 0.175$.

$$H_{cr} = \frac{c_u}{\gamma m}; \quad c_u = (8.5)(18.5)(0.175) = \mathbf{27.5 \text{ kN/m}^2}$$

b. From Figure 15.13, **midpoint circle**

c. From Figure 15.15, $n \approx 0.7$.

$$\text{Distance} = nH = (0.7)(8.5) = \mathbf{5.95 \text{ m}}$$

$$15.14 \quad F_s = \frac{c_u}{\gamma H} M. \quad \beta = 30^\circ; H = 16 \text{ m}; c_u = 50 \text{ kN/m}^2.$$

From Figure 15.19 for $k_h = 0.4$ and $D = 1$, $M \approx 4$

$$F_s = \frac{50}{(17)(16)}(4) = \mathbf{0.735}$$

$$15.15 \quad F_s = \frac{c_u}{\gamma H} M. \quad \beta = 60^\circ; c_u = 1000 \text{ lb/ft}^2; \gamma = 115 \text{ lb/ft}^3; H = 50 \text{ ft}; k_h = 0.3.$$

From Figure 15.20, $M \approx 3$.

$$F_s = \frac{1000}{(115)(50)}(4) = \mathbf{0.57}$$

$$15.16 \quad \text{a.} \quad \beta = \tan^{-1}\left(\frac{1}{2}\right) = 26.57^\circ$$

$$\frac{F_s}{\tan \phi'} = \frac{1}{\tan 20} = 2.75$$

From Figure 15.27

$$\frac{c'}{\gamma H_{\text{cr}} \tan \phi'} \approx 0.05$$

Or,

$$0.05 = \frac{20}{(16)(H_{\text{cr}})(\tan 20)} = \mathbf{68.7 \text{ m}}$$

$$\text{b.} \quad \frac{F_s}{\tan \phi'} = \frac{1}{\tan 25} = 2.14. \quad \beta = \tan^{-1}\left(\frac{1}{1.5}\right) = 33.7^\circ.$$

Figure 15.27:

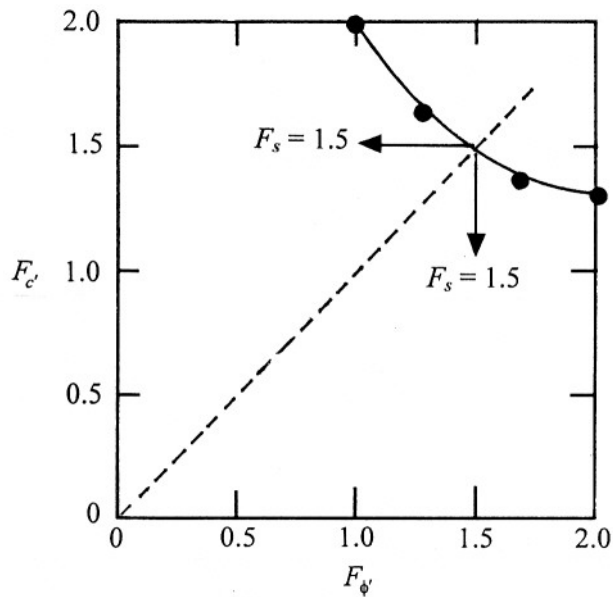
$$\frac{c'}{\gamma H_{cr} \tan \phi'} \approx 0.035; \quad \frac{750}{(110)(H_{cr})(\tan 25)} = 0.035$$

$$H_{cr} = \mathbf{417.8 \text{ ft}}$$

15.17 $\beta = \tan^{-1}\left(\frac{1}{2}\right) = 26.57^\circ$; $\phi' = 10^\circ$; $c' = 700 \text{ lb/ft}^2$; $\gamma = 110 \text{ lb/ft}^3$.

ϕ'_d (deg)	$F_{\phi'} = \frac{\tan \phi'}{\tan \phi'_d}$	m	$c'_d = m\gamma H$ (lb/ft ²)	$F_{c'} = \frac{c'}{c'_d}$
5	2.01	0.098	539	1.30
6	1.68	0.090	495	1.41
8	1.25	0.078	429	1.63
10	1.00	0.064	352	1.99

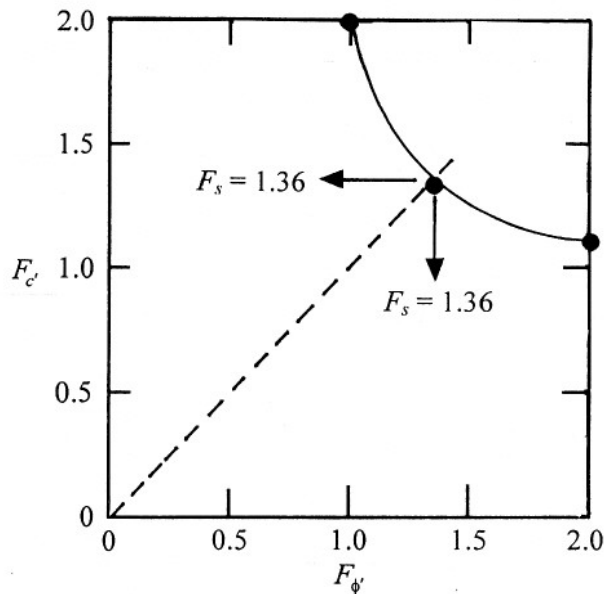
The plot of $F_{c'}$ vs. $F_{\phi'}$ is shown. From the figure, $F_{c'} = F_{\phi'} = F_s = \mathbf{1.5}$.



15.18 $n' = 1$; $\phi' = 20^\circ$; $c' = 400 \text{ lb/ft}^2$; $\gamma = 115 \text{ lb/ft}^3$; $H = 30 \text{ ft}$

ϕ'_d (deg)	$F_{\phi'} = \frac{\tan \phi'}{\tan \phi'_d}$	m	$c'_d = m\gamma H$ (lb/ft ²)	$F_{c'} = \frac{c'}{c'_d}$
5	4.16	0.133	458.9	0.87
8	2.06	0.105	362.3	1.1
10	1.36	0.080	276.0	1.45
20	1.00	0.058	200.1	2.0

The plot of $F_{c'}$ vs. $F_{\phi'}$ is shown. From the figure, $F_{\phi'} = F_{c'} = F_s \approx \mathbf{1.36}$.

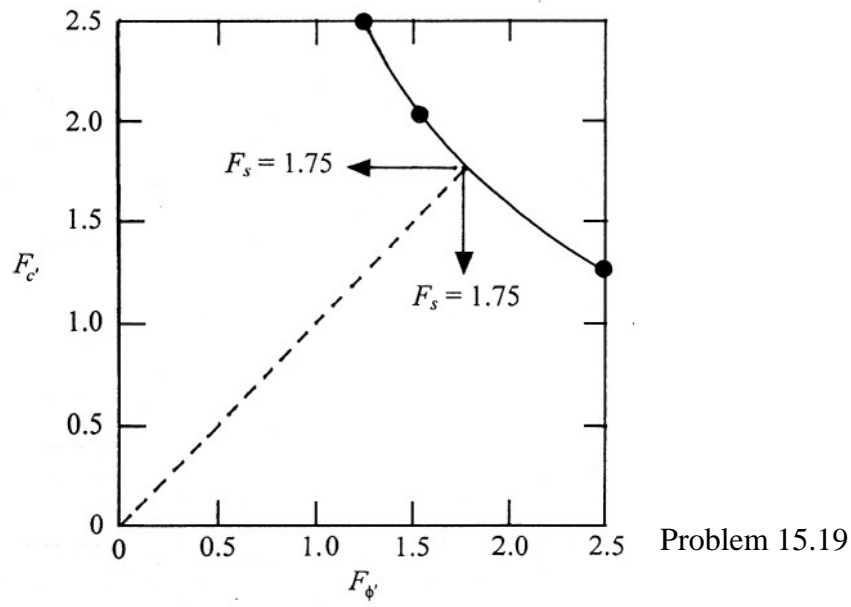


15.19 $n' = 2.5$; $\beta = \tan^{-1}\left(\frac{1}{2.5}\right) = 21.8^\circ$; $\phi' = 12^\circ$; $c' = 24 \text{ kN/m}^2$; $\gamma = 16.5 \text{ kN/m}^3$;
 $H = 12 \text{ m}$

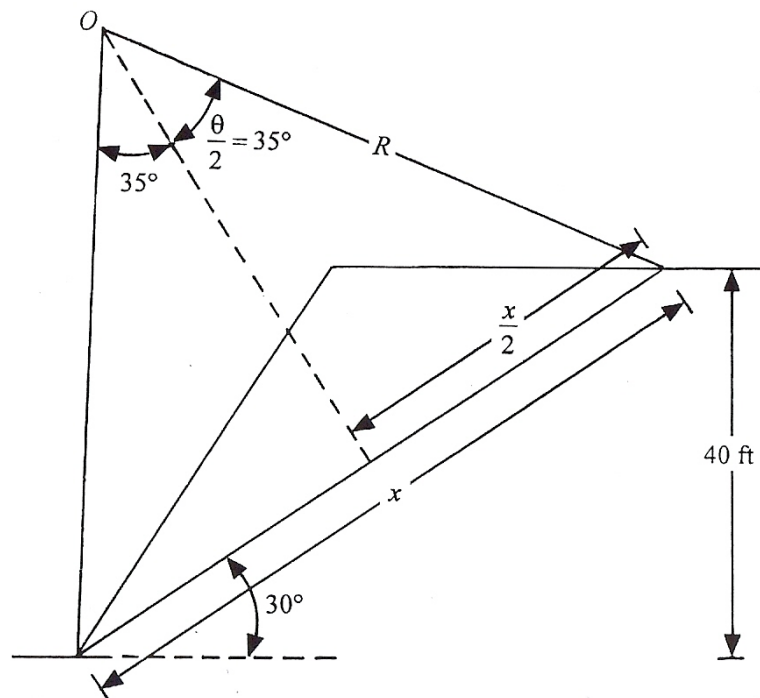
ϕ'_d (deg)	$F_{\phi'} = \frac{\tan \phi'}{\tan \phi'_d}$	m	$c'_d = m\gamma H$ (kN/m ²)	$F_{c'} = \frac{c'}{c'_d}$
5	2.43	0.088	17.42	1.38
8	1.51	0.060	11.88	2.02
10	1.21	0.048	9.50	2.53
12	1.00	0.038	7.52	3.19

The plot of $F_{c'}$ vs. $F_{\phi'}$ is shown on the next page.

From the figure, $F_{\phi'} = F_{c'} = F_s = \mathbf{1.75}$.



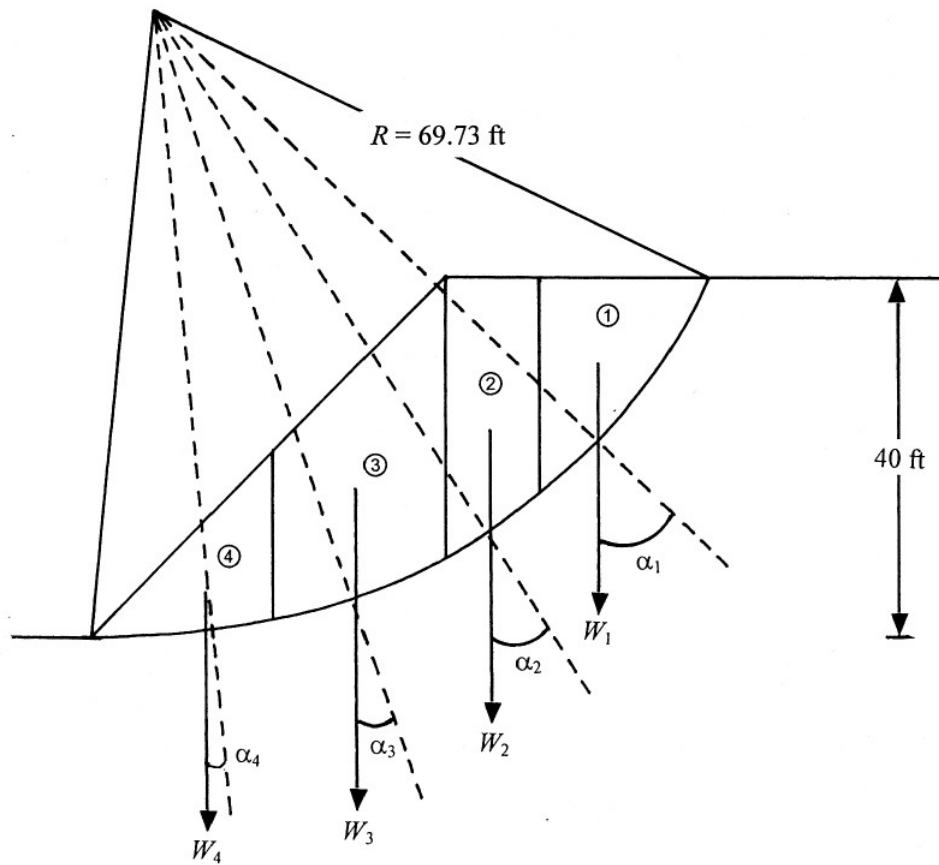
15.20 a. Refer to the figure.



$$\frac{40}{x} = \sin 30^\circ; x = 80 \text{ ft}$$

$$\frac{40}{\text{Radius}, R} = \sin 35^\circ; R = \frac{40}{\sin 35} = 69.73 \text{ ft}$$

With radius $R = 69.73$ ft, the trial surface circle has been drawn.



Now the following table can now be prepared.

Slice	Area of slices (ft ²)	Weight of slice			
		$W_n = A \times \gamma$ (kip/ft)	α_n (deg)	$W_n \cos \alpha_n$ (kip/ft)	$W_n \sin \alpha_n$ (kip/ft)
1	$\frac{(26)(20)}{2} = 260$	29.9	47	20.39	21.86
2	$\frac{(10)(26 + 32)}{2} = 290$	33.35	32	28.28	17.67
3	$\frac{(20)(32 + 20)}{2} = 520$	59.8	20	59.19	20.45
4	$\frac{(20)(20)}{2} = 200$	23	5	22.91	2.00
				$\Sigma 127.77$	$\Sigma 61.98$

$$F_s = \frac{R\theta c' + (\sum W_n \cos \alpha_n) \tan \phi'}{\sum W_n \sin \alpha_n}$$

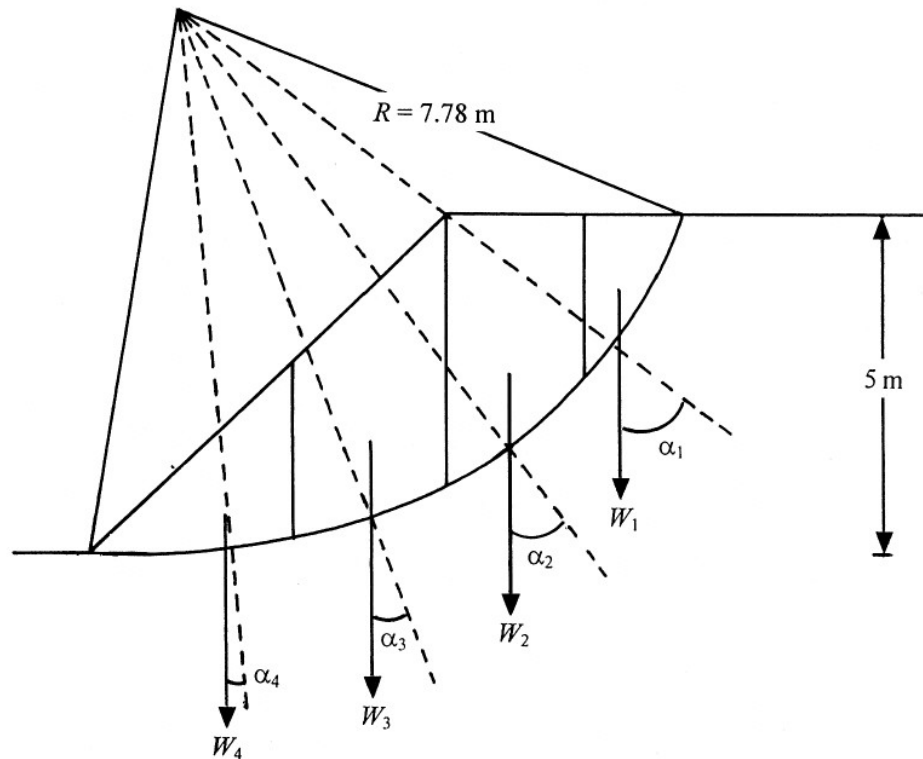
$$= \frac{(69.73) \left[\left(\frac{\pi}{180} \right) (70) \right] (0.4) + (127.77) (\tan 20)}{61.98} = \mathbf{1.3}$$

Note: The accuracy can be increased by increasing the number of slices.

b. As in Part a, $\frac{H}{x} = \sin \alpha$; $x = \frac{H}{\sin \alpha} = \frac{5}{\sin 30} = 10 \text{ m}$

$$\frac{\left(\frac{x}{2} \right)}{R} = \sin \frac{\theta}{2}, \text{ or } \frac{5}{\sin 40} = R = 7.78 \text{ m}$$

With a radius $R = 7.78 \text{ m}$, the trial surface has been drawn.



The following table can now be prepared.

Slice	Area of slices (m ²)	Weight of slice			
		$W_n = A \times \gamma$ (kN/m)	α_n (deg)	$W_n \cos \alpha_n$ (kN/m)	$W_n \sin \alpha_n$ (kN/m)
1	$\frac{(2.6)(1.5)}{2} = 1.95$	33.35	54	19.60	29.98
2	$\frac{(2.6 + 4.2)(2)}{2} = 6.80$	116.28	38	91.63	71.59
3	$\frac{(4.2 + 2.8)(2)}{2} = 7.00$	119.70	20	112.48	40.94
4	$\frac{(3)(2.8)}{2} = 4.20$	71.82	6	71.43	7.52
				$\Sigma 295.14$	$\Sigma 150.02$

$$F_s = \frac{R\theta c' + (\Sigma W_n \cos \alpha_n) \tan \phi'}{\Sigma W_n \sin \alpha_n}$$

$$= \frac{(7.78) \left[\left(\frac{\pi}{180} \right) (80) \right] (18) + (295.14) (\tan 15)}{150.02} = \mathbf{1.83}$$

Note: The accuracy will improve with smaller slices.

$$15.21 \quad \phi' = 20^\circ; \beta = 26.57^\circ; r_u = 0.5; \frac{c'}{\gamma H} = \frac{300}{(120)(25)} = 0.1.$$

Using Table 15.3, the following table can be prepared.

D	m'	n'	$F_s = m' - n' r_u$
Toe circle	1.804	2.101	0.75
1.00	1.841	1.143	1.27
1.25	1.874	1.301	1.22
1.50	2.079	1.528	1.32

$$F_s \approx \mathbf{0.75}.$$

$$15.22 \quad \phi' = 20^\circ; \beta = 18.43^\circ; r_u = 0.5; \frac{c'}{\gamma H} = \frac{6}{(20)(6)} = 0.05.$$

Using Table 15.3, the following table can be prepared.

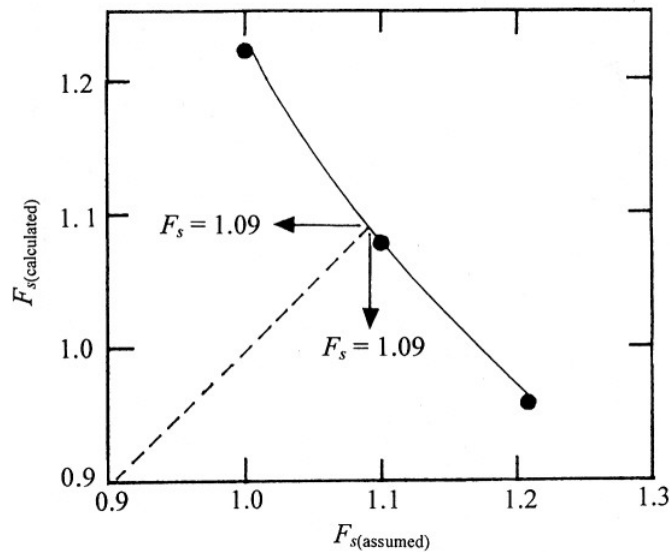
D	m'	n'	$F_s = m' - n' r_u$
1.00	1.840	1.387	1.15
1.25	1.834	1.493	1.09
1.50	2.011	1.705	1.16

$F_s \approx 1.09$.

15.23 $\beta = 20^\circ$; $\phi' = 15^\circ$; $r_u = 0.5$; $\gamma = 17.5 \text{ kN/m}^3$; $c' = 20 \text{ kN/m}^2$; $H = 15 \text{ m}$

$F_{s(\text{assumed})}$	$\frac{c'}{\gamma H F_s}$	$\phi'_d \text{ (deg)}$	$F_{s(\text{calculated})} = \frac{\tan \phi'}{\tan \phi'_d}$
1.2	0.0635	15.5	0.966
1.1	0.0693	14.0	1.075
1.0	0.0762	12.5	1.209
0.9	0.0847	11.5	1.317

From the plot, $F_s = 1.09$.



15.24 $\beta = \tan^{-1}\left(\frac{1}{3}\right) = 18.43^\circ$; $\phi' = 25^\circ$; $c' = 12 \text{ kN/m}^2$; $\gamma = 19 \text{ kN/m}^3$; $r_u = 0.25$;

$H = 12.63 \text{ m}$

$F_{s(\text{assumed})}$	$\frac{c'}{\gamma H F_s}$	ϕ'_d (deg)	$F_{s(\text{calculated})} = \frac{\tan \phi'}{\tan \phi'_d}$
1.7	0.0294	15	1.74
1.8	0.0278	16	1.63
1.9	0.0263	17	1.53

From the graph, $F_s \approx \mathbf{1.72}$.

