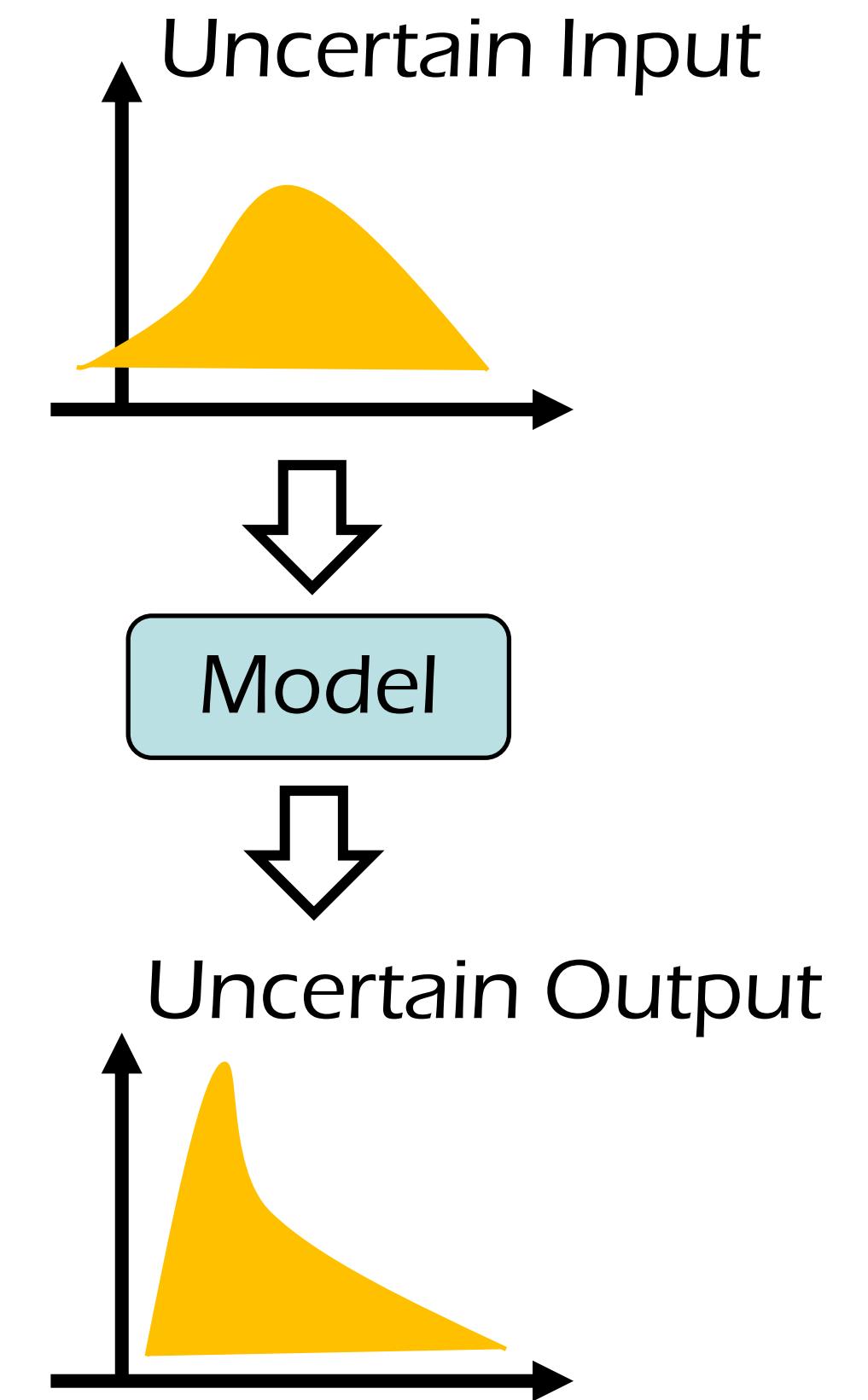


# Quantification and Propagation of Uncertainties in Identification of Flame Impulse Response for Thermoacoustic Stability Analysis

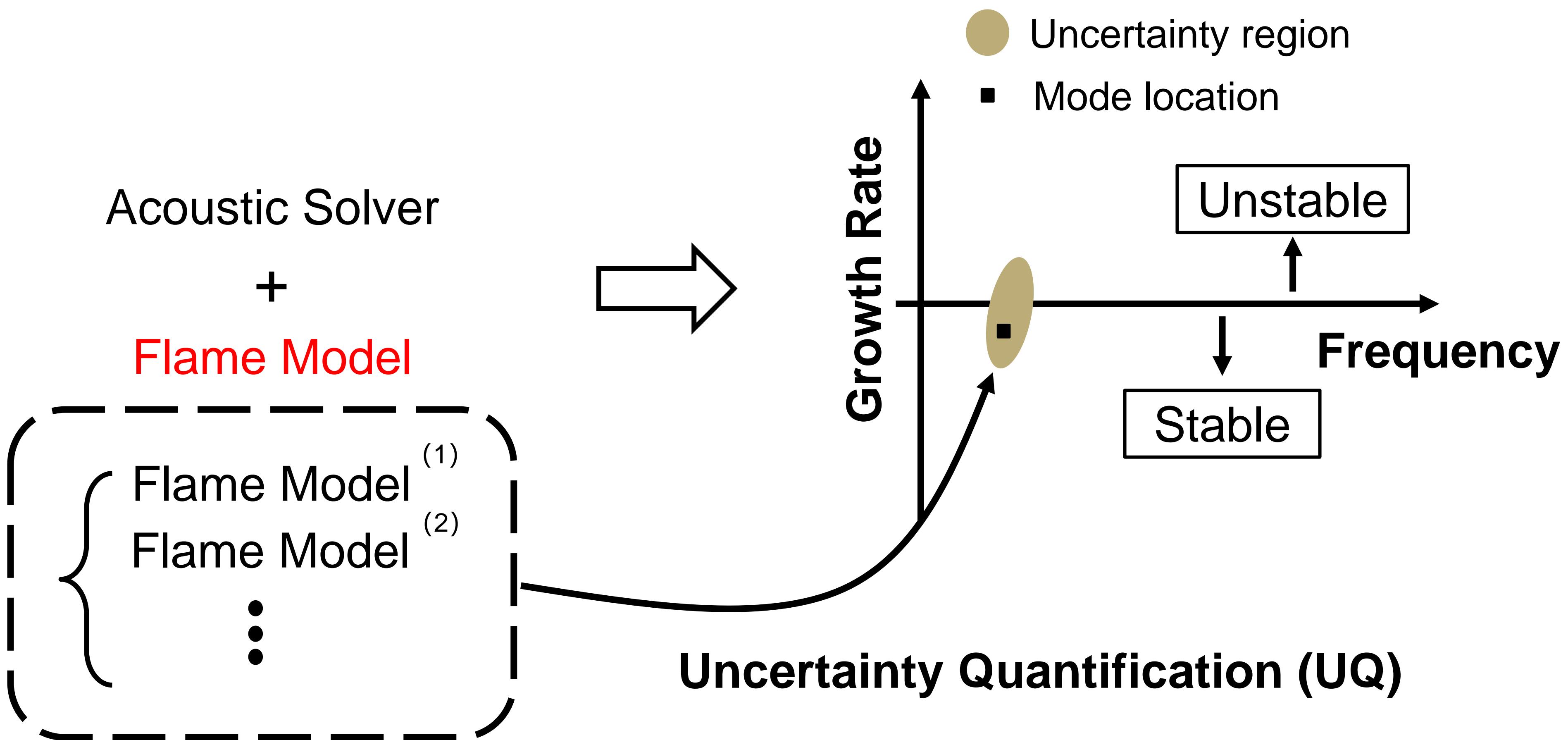
S. Guo, C. Silva, A. Ghani, W. Polifke



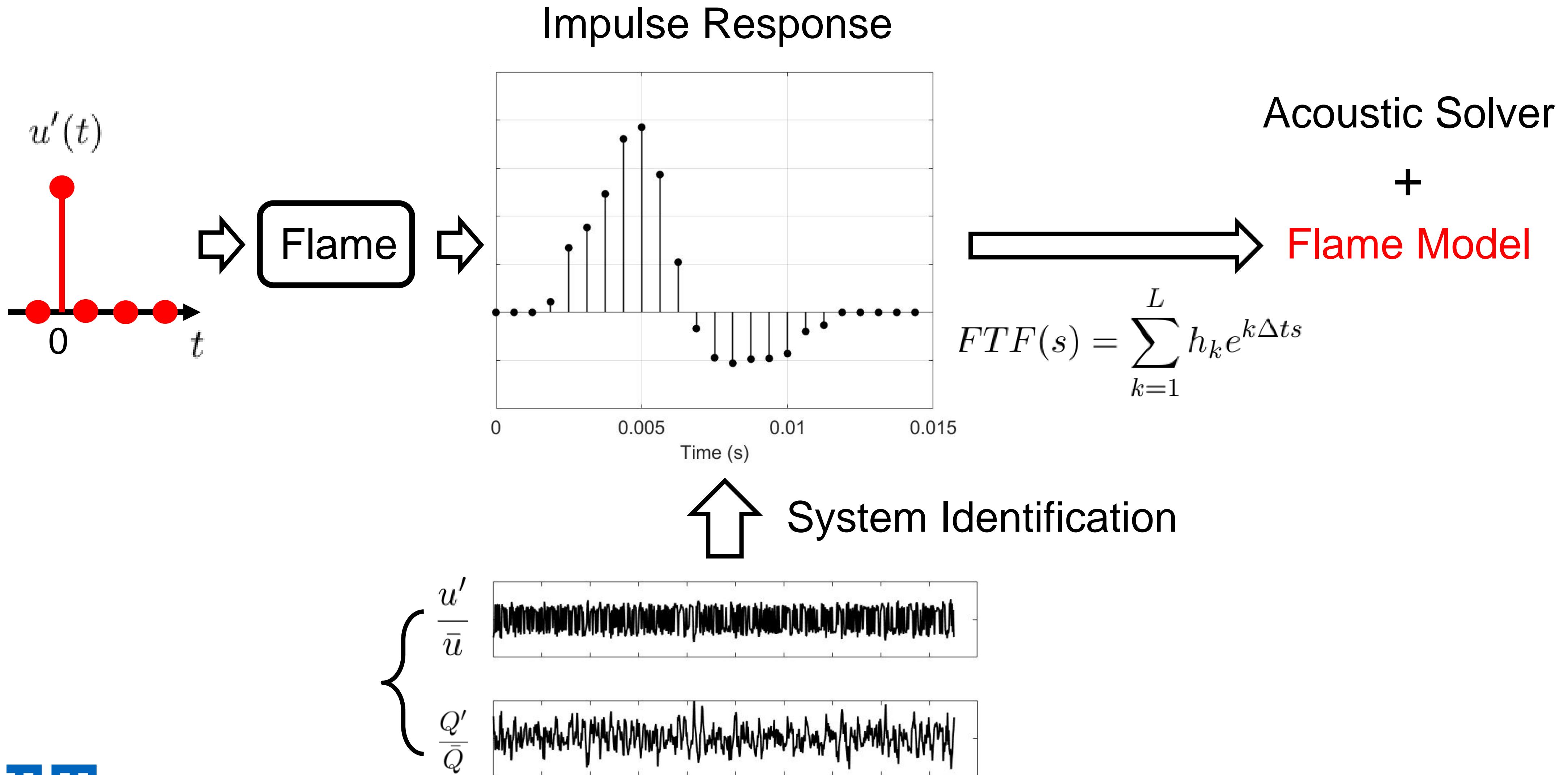
ASME Turbo Expo 2018  
GT2018-75644



# Uncertainty quantification is required to achieve robust thermoacoustic instability prediction

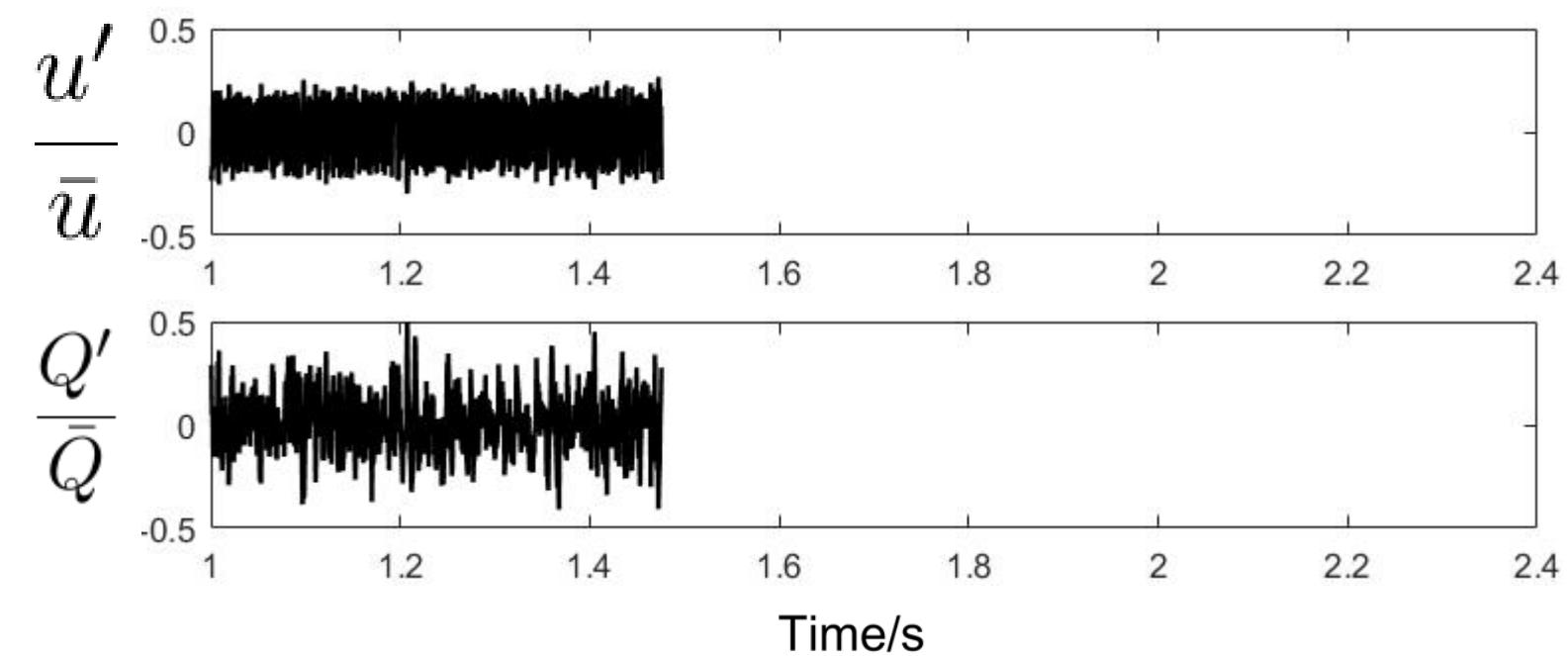


We focus on flame impulse response model in the current study

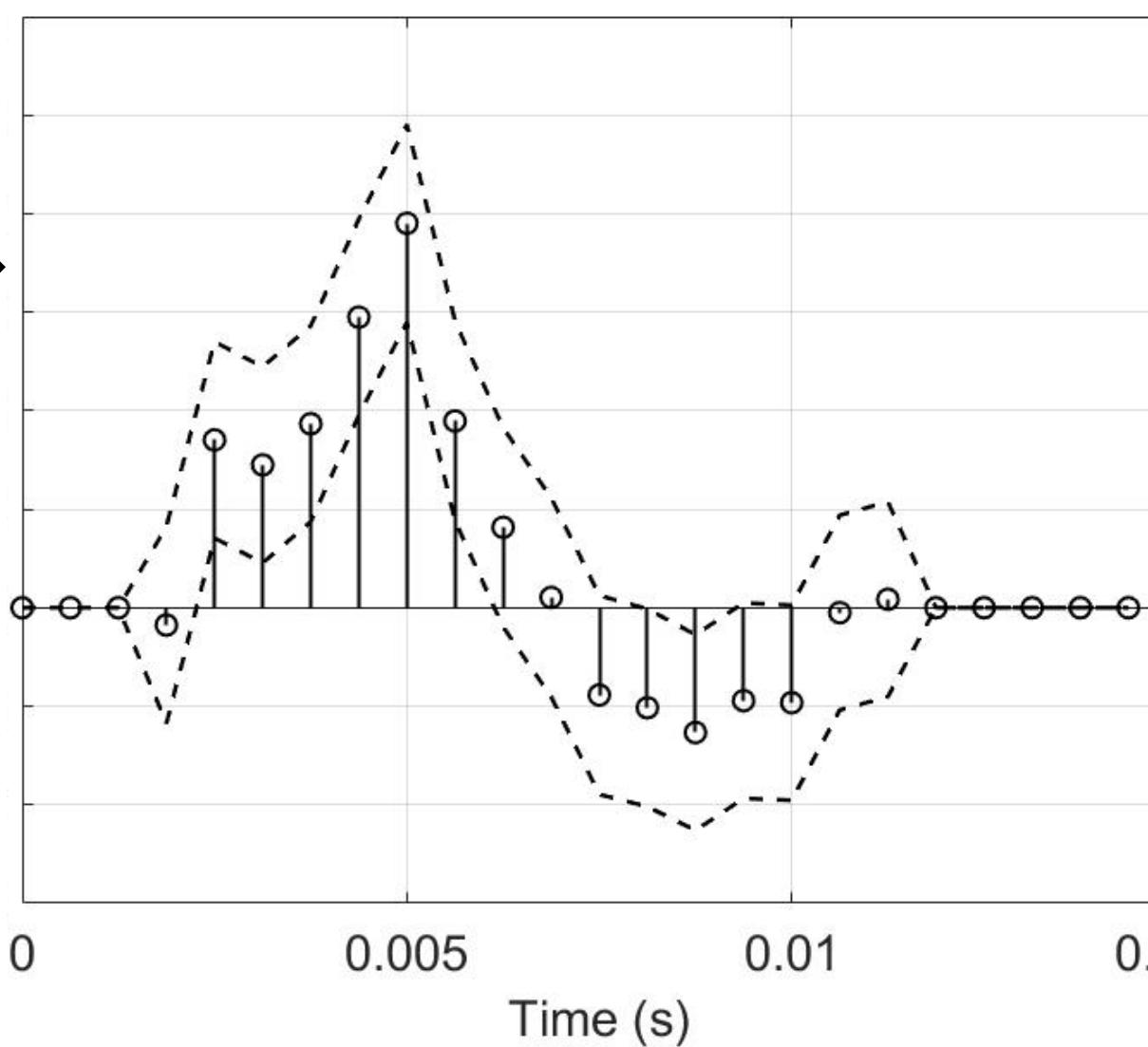


We are interested in the impact of flame impulse response model on thermoacoustic instability prediction

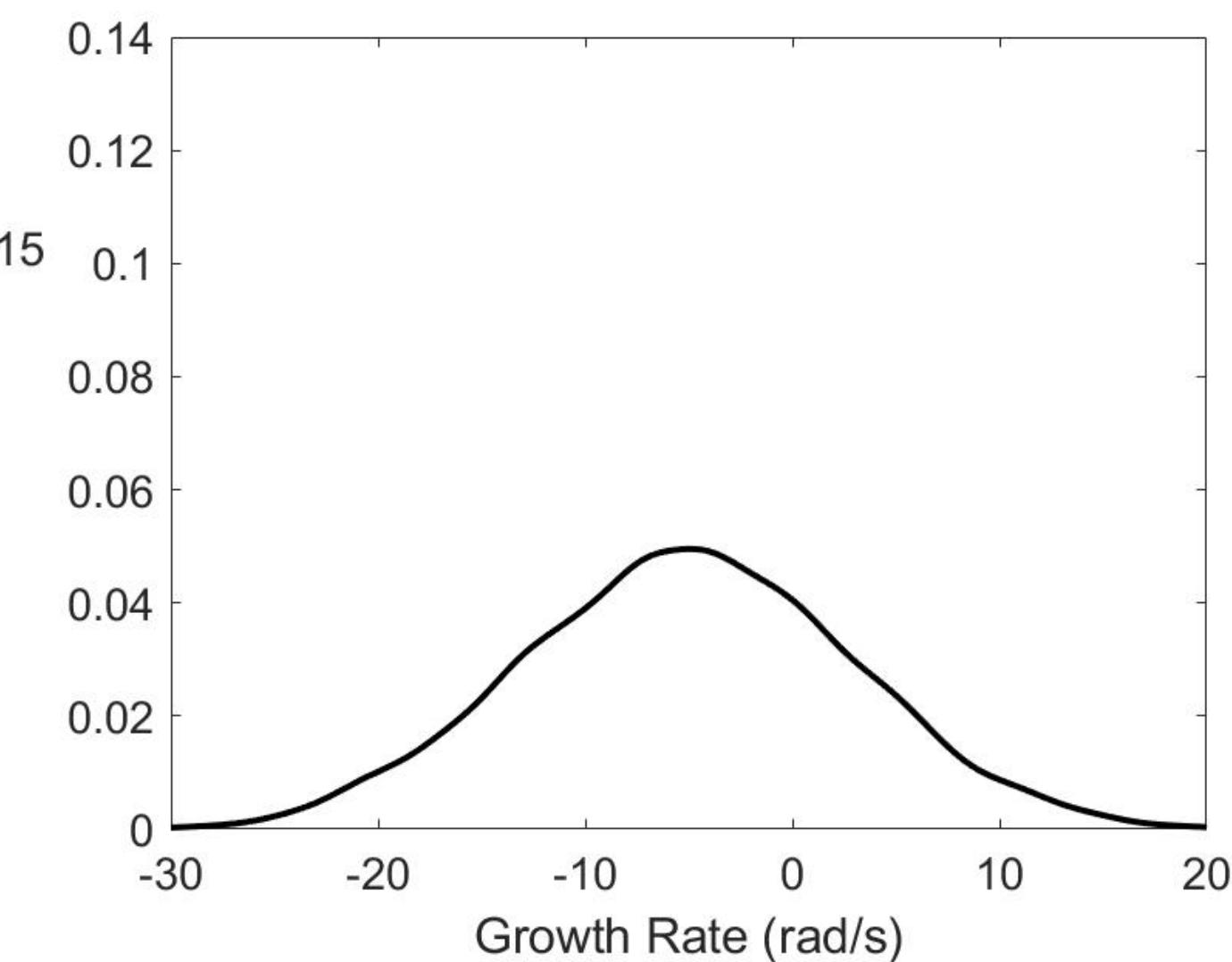
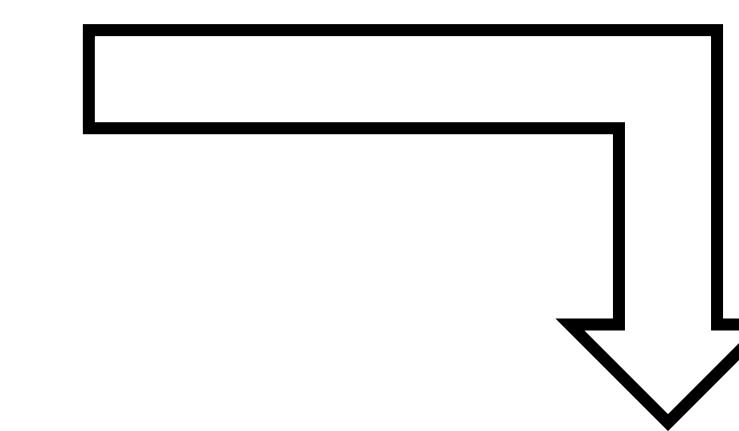
CFD Time Series



Impulse Response



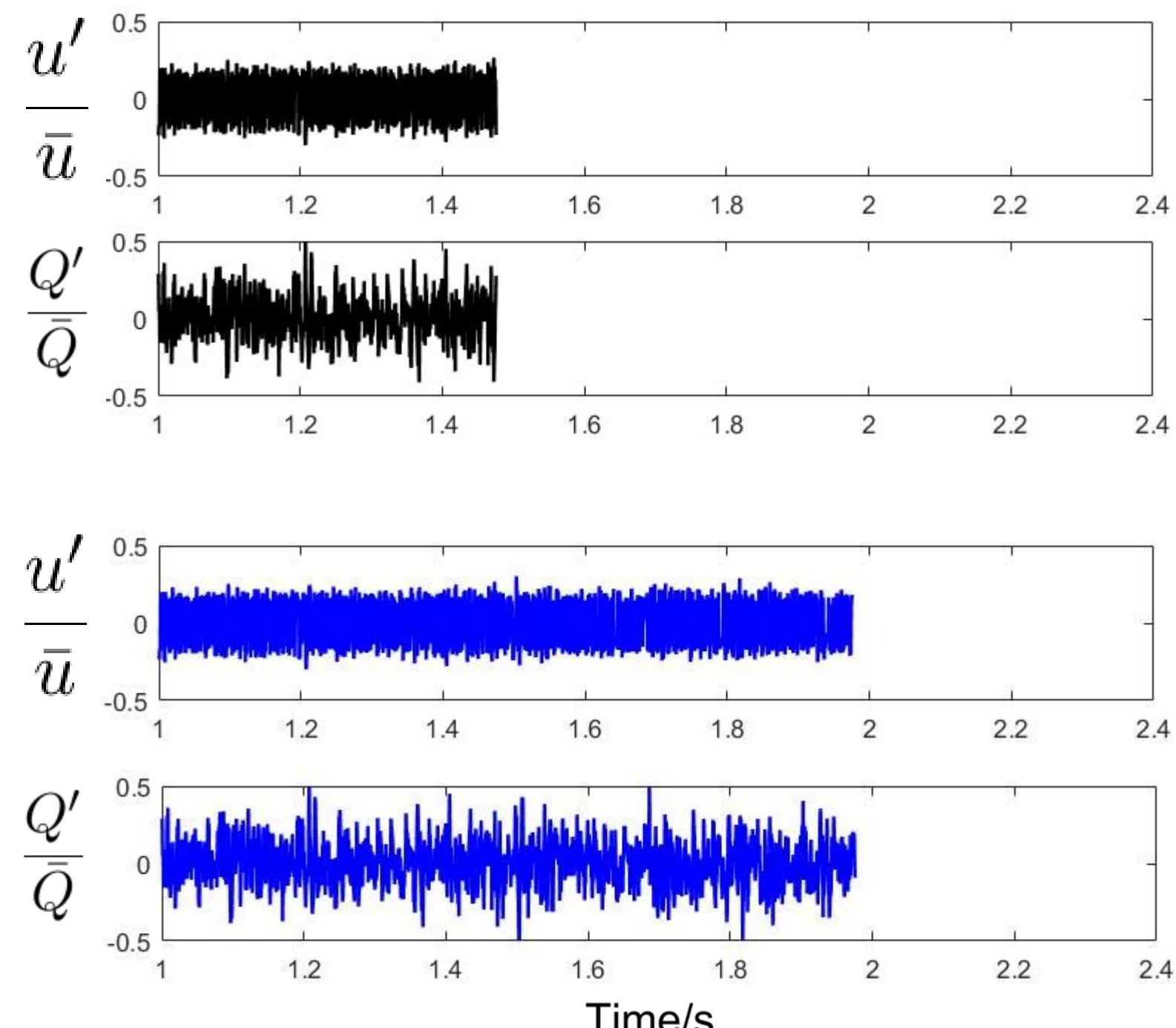
Acoustic  
Solver



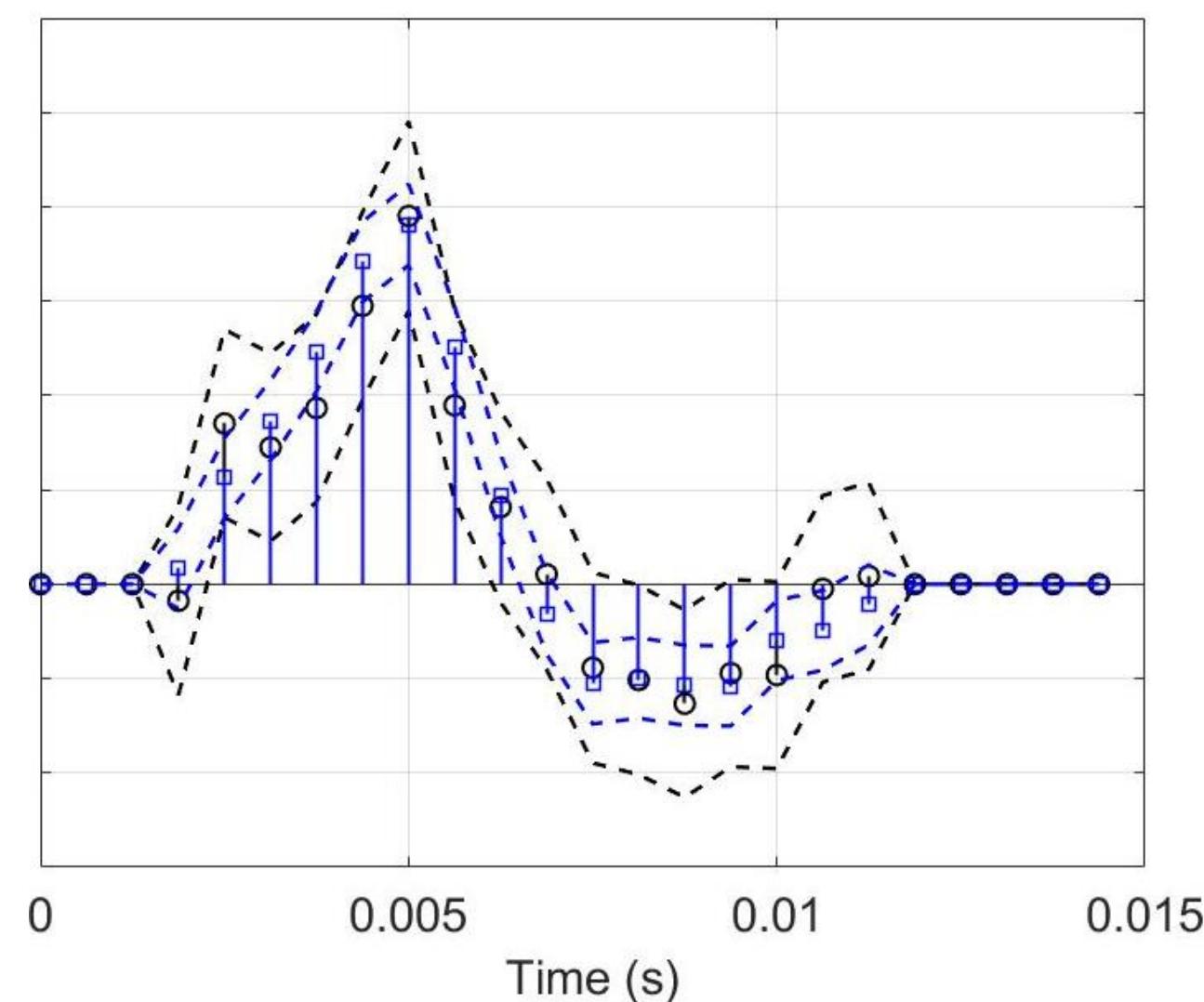
Growth Rate PDF

# Longer CFD time series yields less uncertain identification of impulse response model as well as growth rate prediction

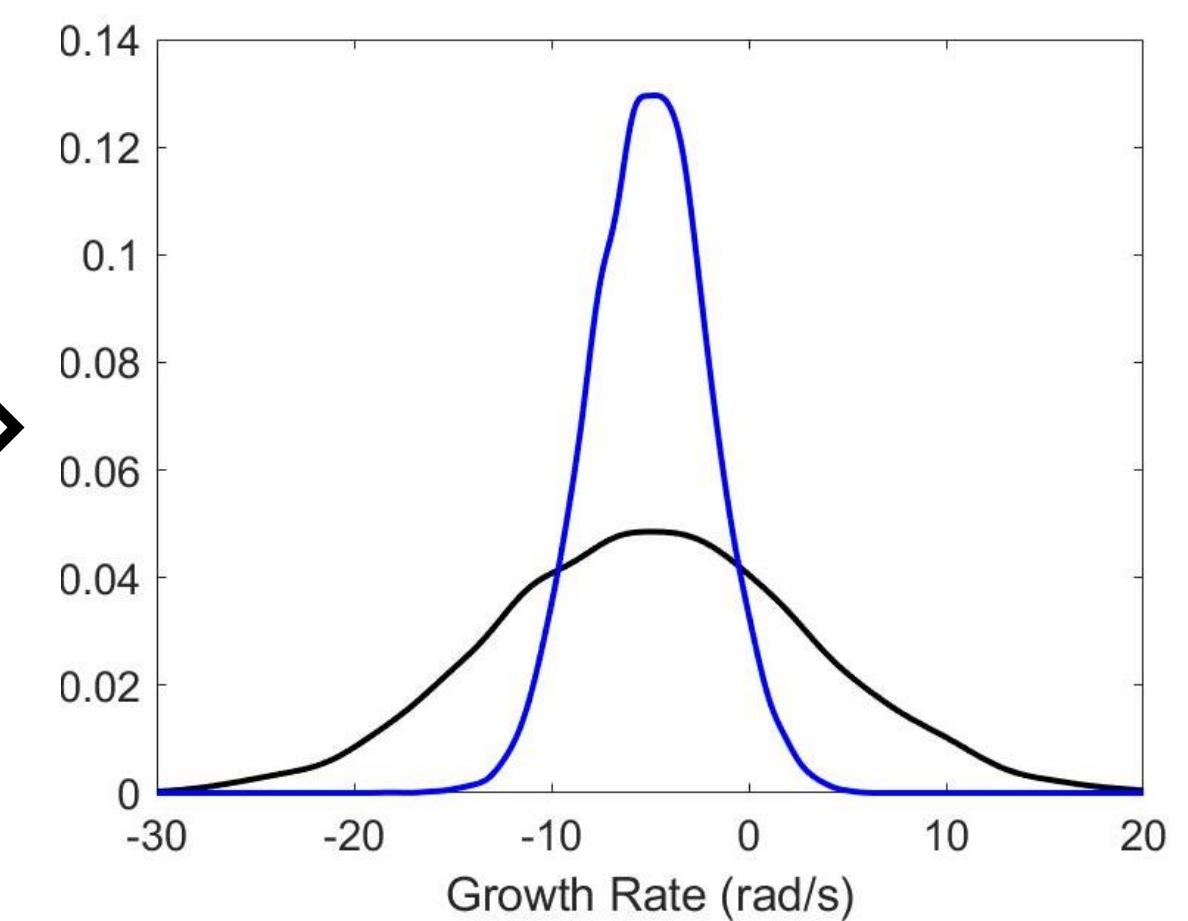
## CFD Time Series



## Impulse Response

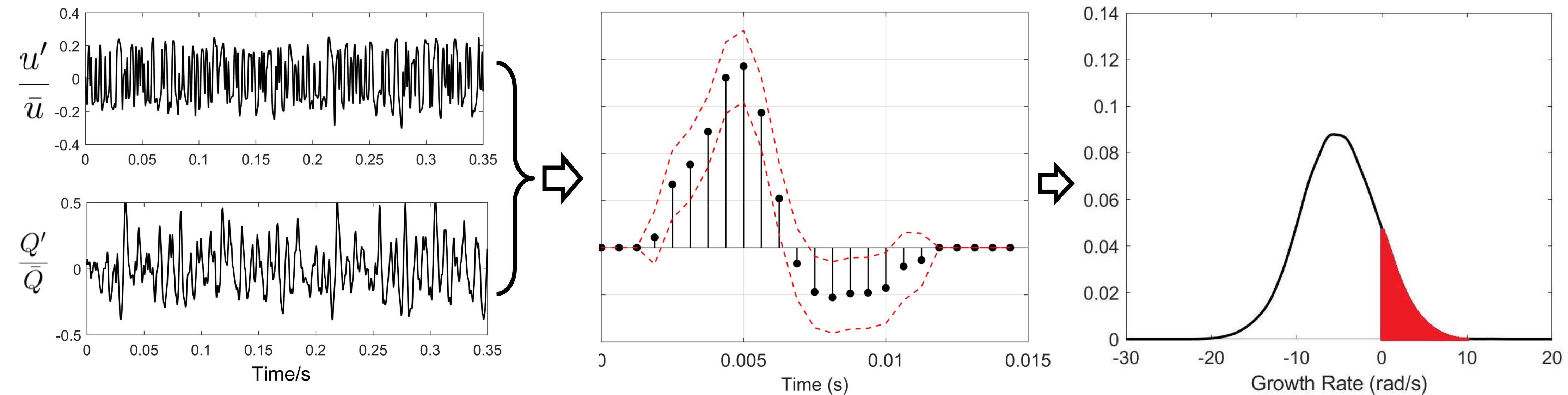


## Growth Rate PDF



# Motivation of our current study

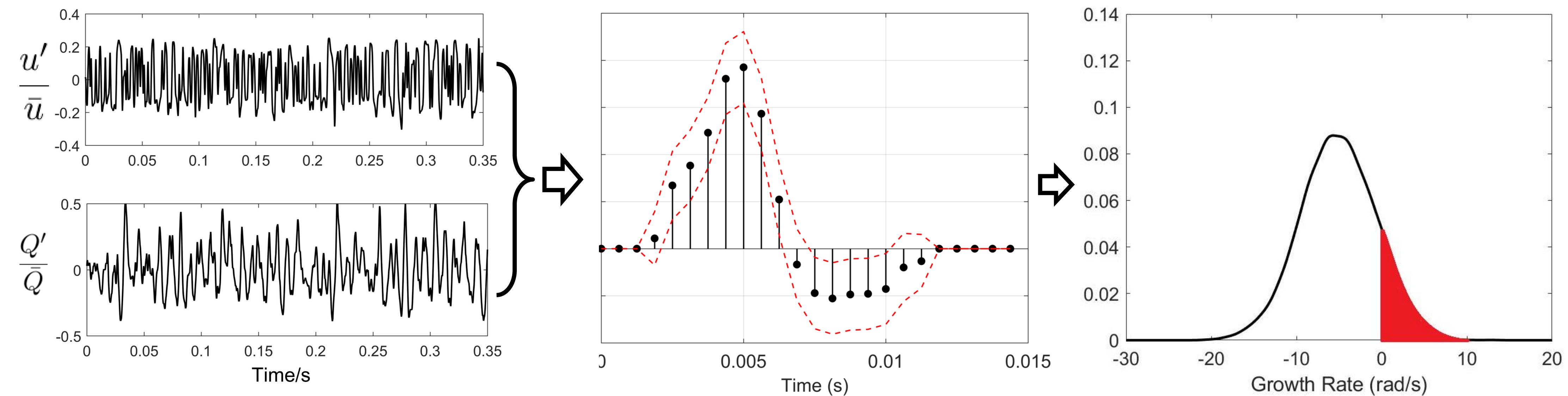
**Problem:** How to determine the adequate length of CFD time series to achieve a satisfactory uncertainty reduction in growth rate prediction?



# Presentation overview

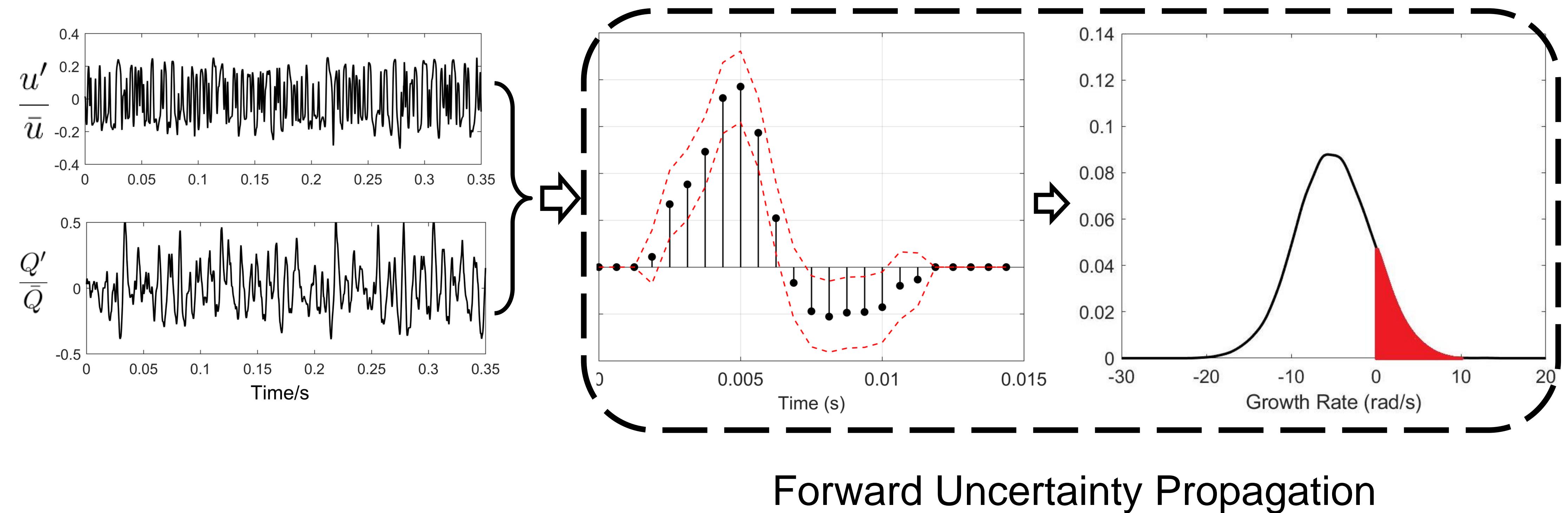
- Motivation
- Methodology
- Thermoacoustic case studies
- Results
- Conclusions

# Methodology – divide and conquer

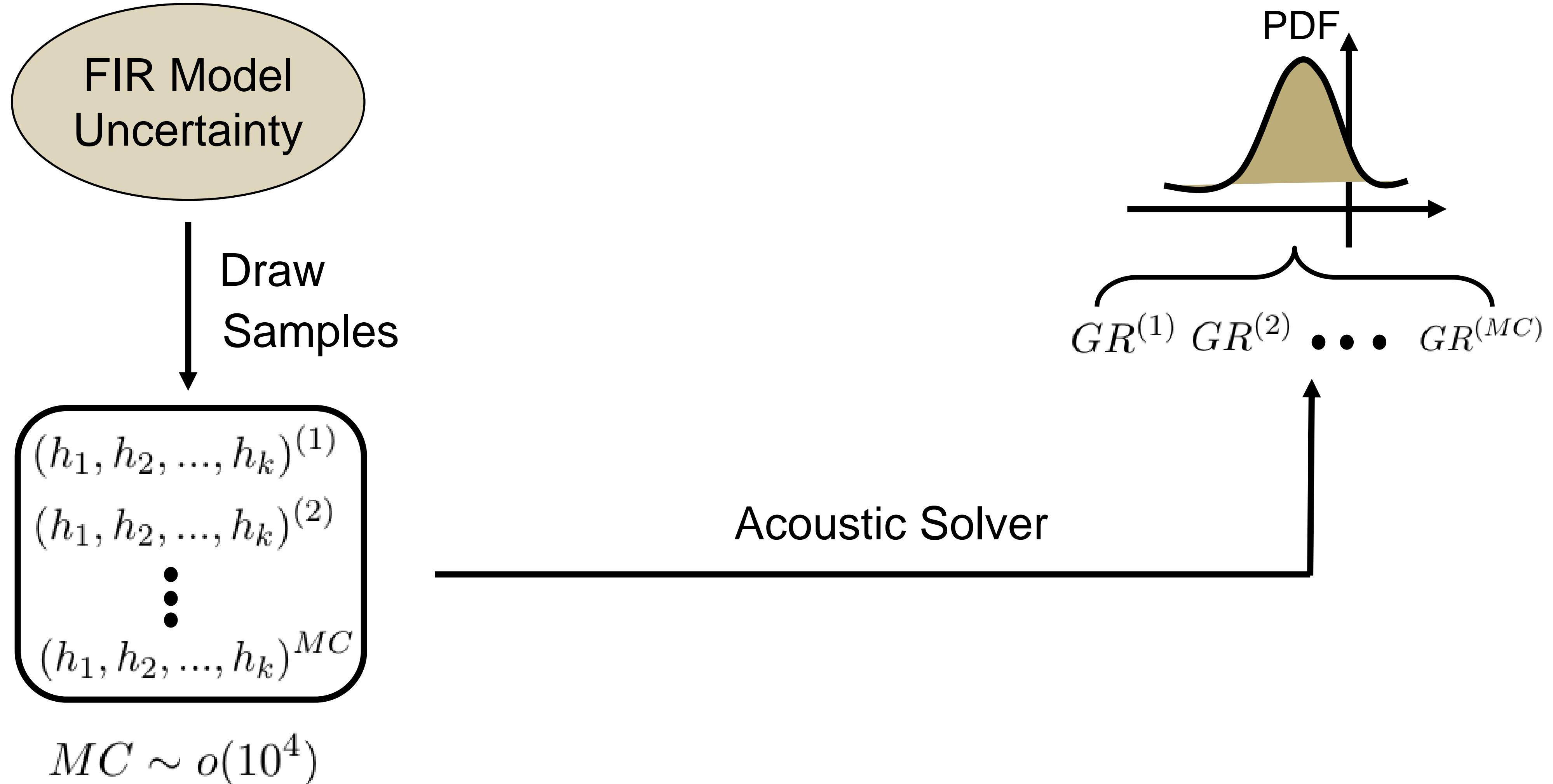


# First we need to solve the problem of efficient forward uncertainty propagation

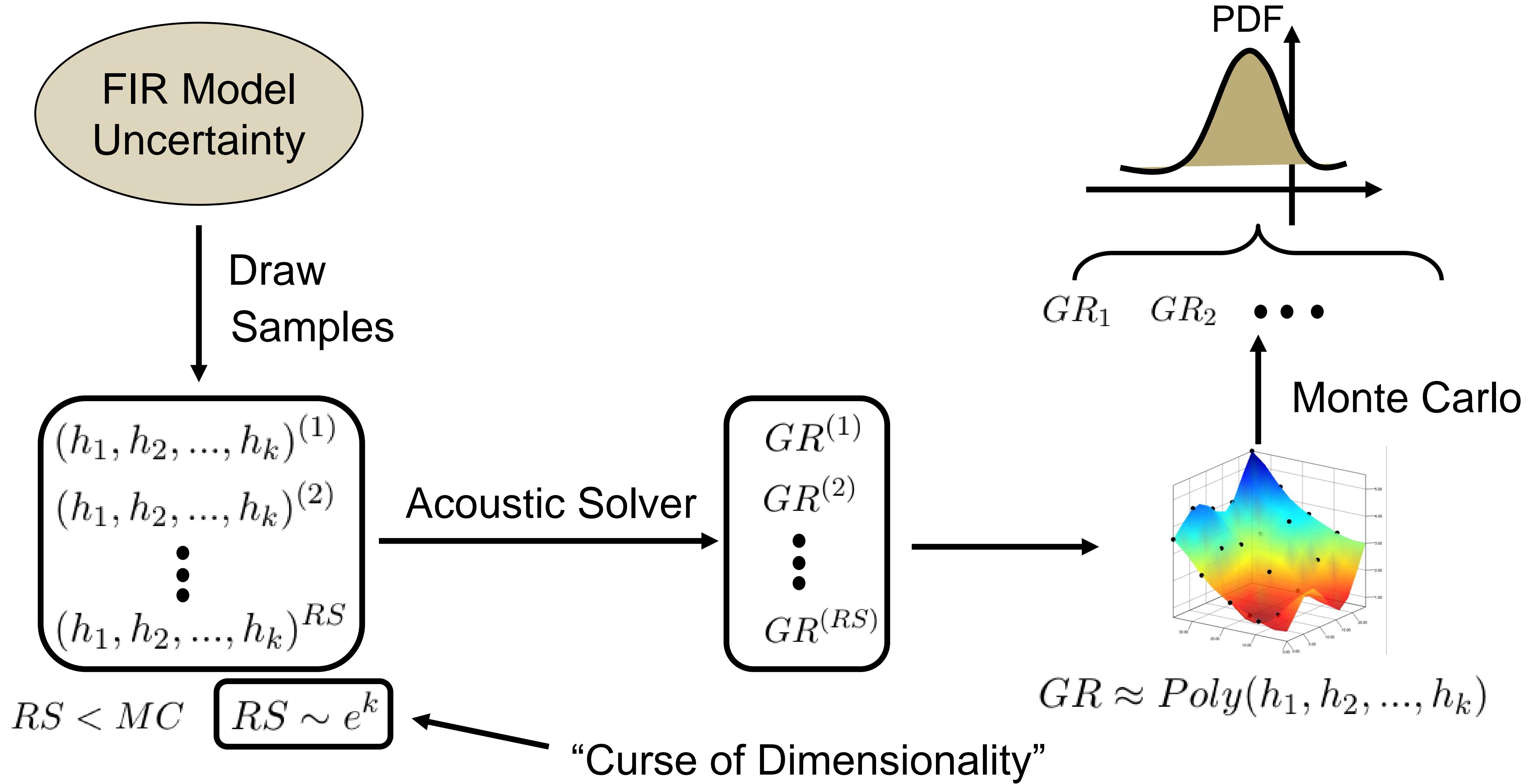
**Problem:** How to efficiently propagate uncertainties in flame impulse response model to the growth rate predictions?



# Monte Carlo is an accurate yet very expensive method to achieve forward uncertainty propagation



# Response Surface Modeling is an alternative for reducing the sample numbers, but it suffers from “Curse of Dimensionality”



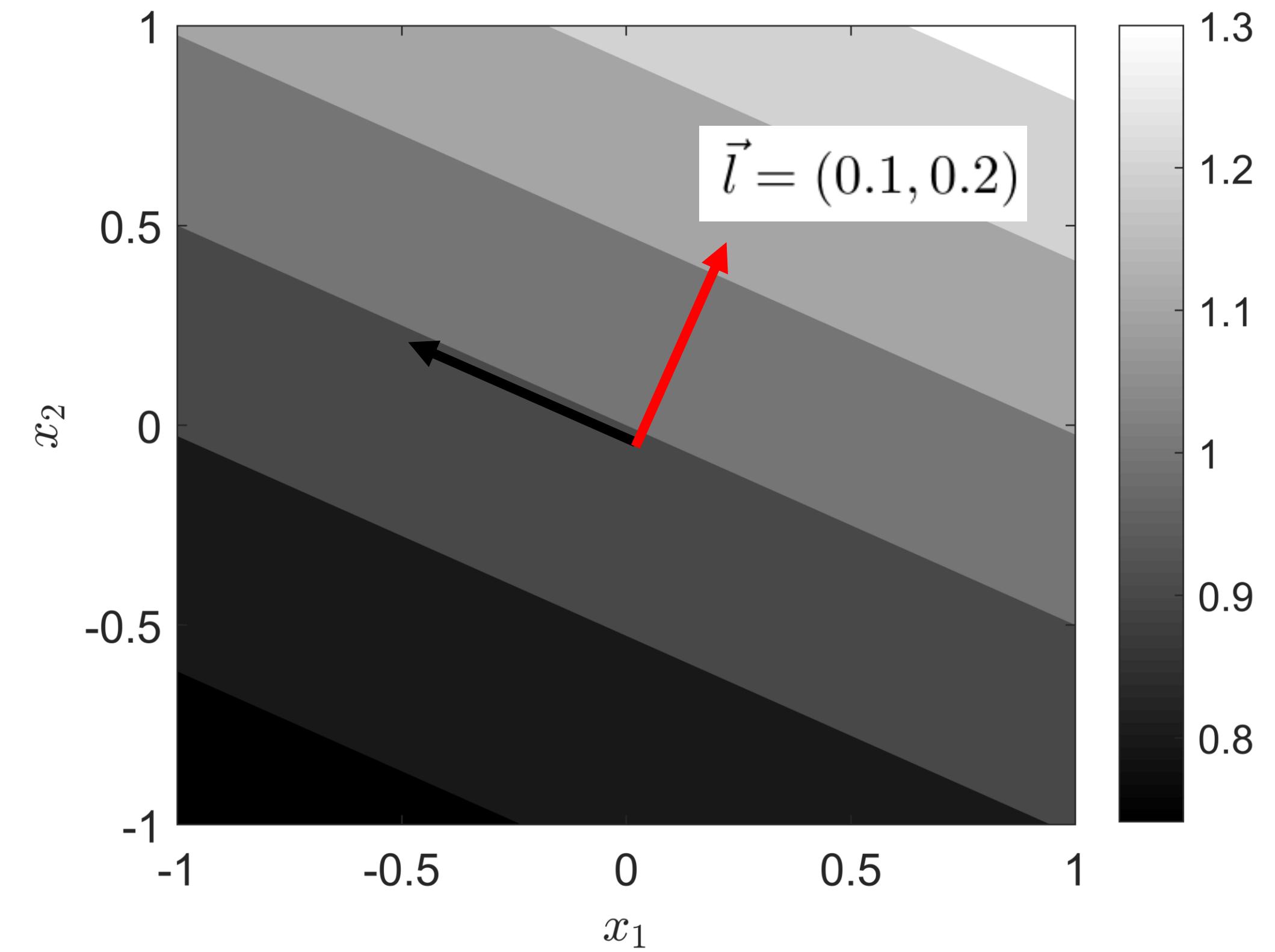
# Active Subspace identifies and exploits important directions in input space: Example

Example:

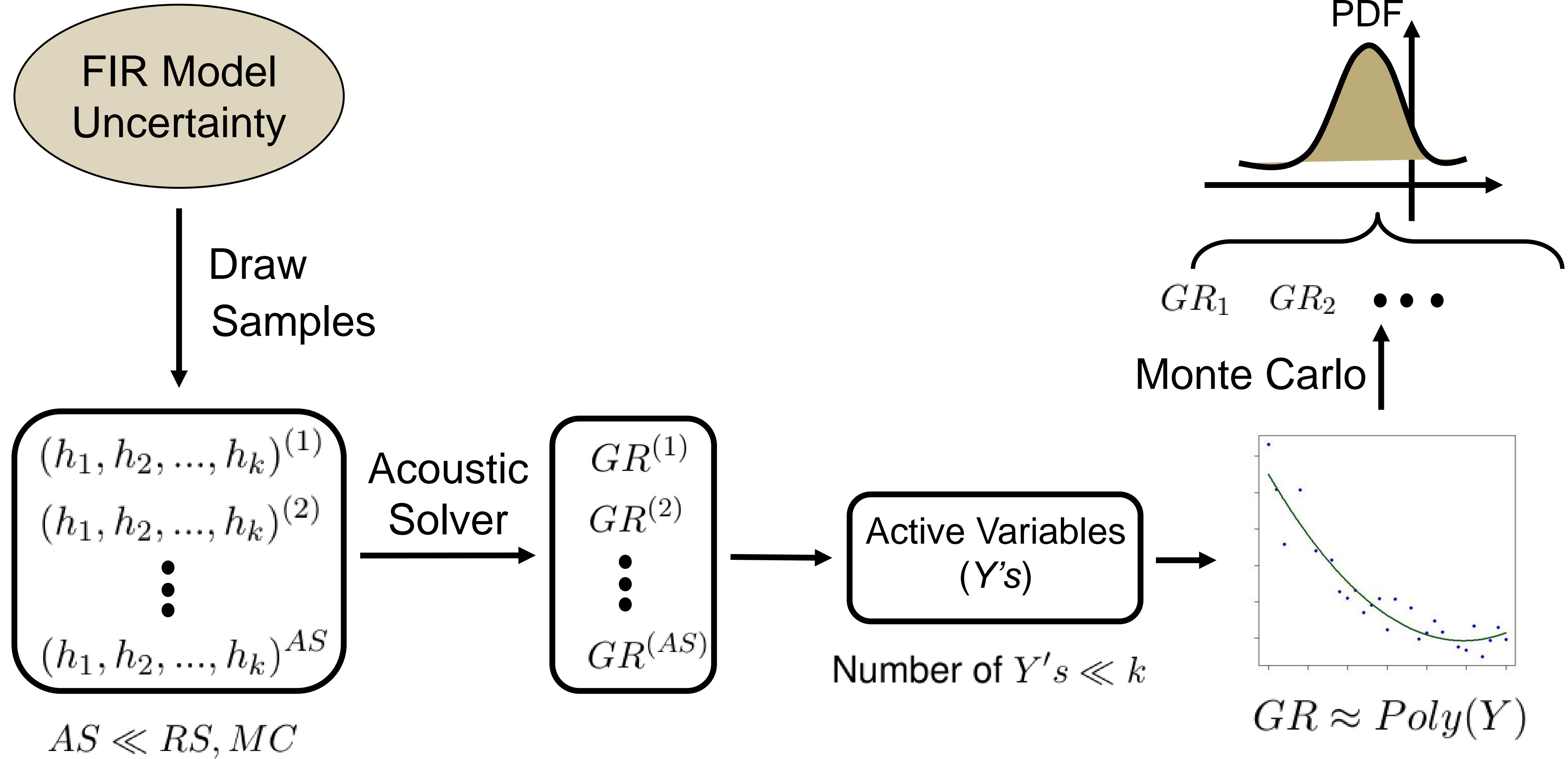
$$y = e^{(0.1x_1 + 0.2x_2)}$$

↑  
Approximate

$$\begin{array}{|c|c|} \hline & AV = 0.1x_1 + 0.2x_2 \\ \hline & y \approx f(AV) \\ \hline \end{array}$$



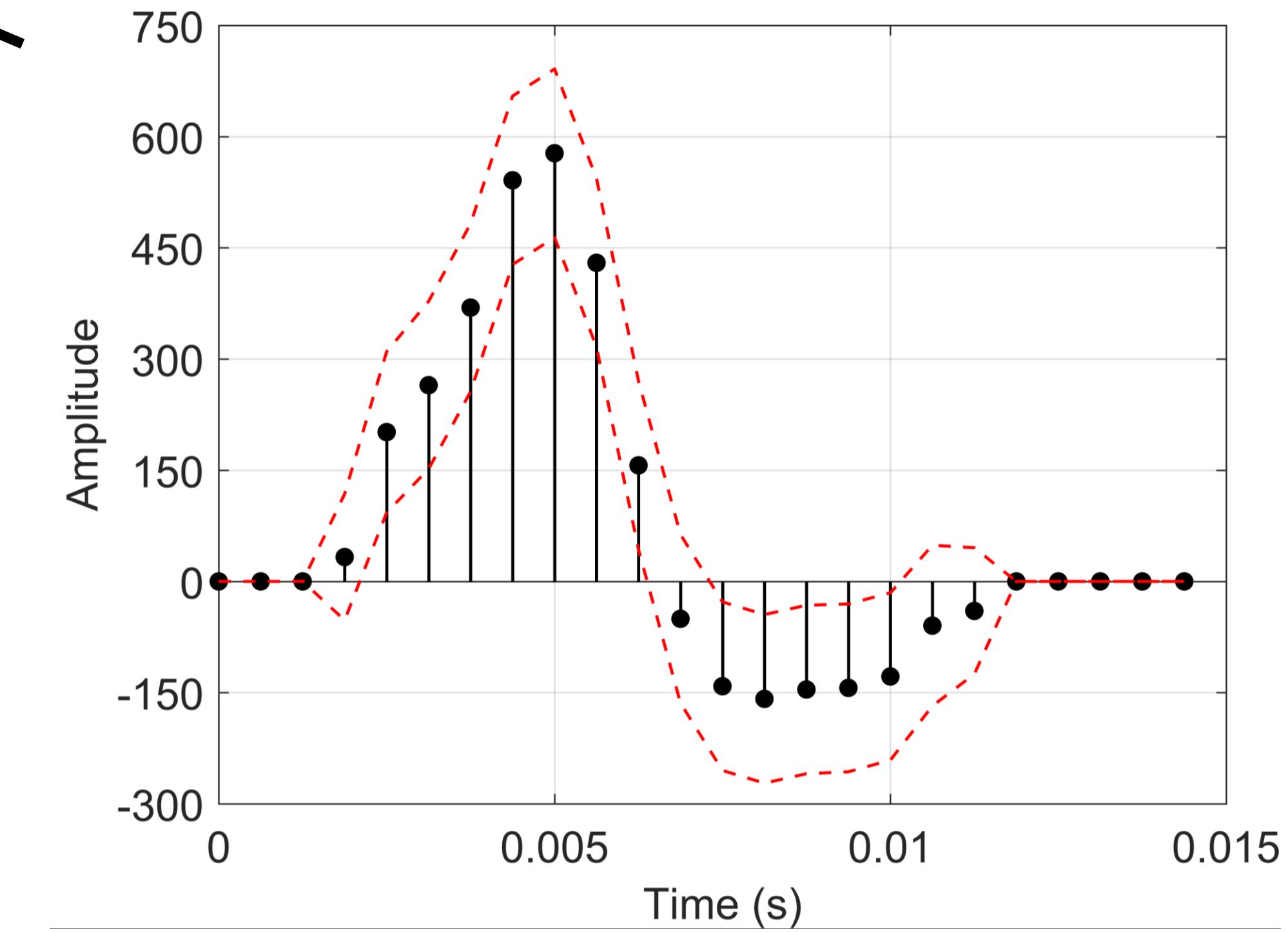
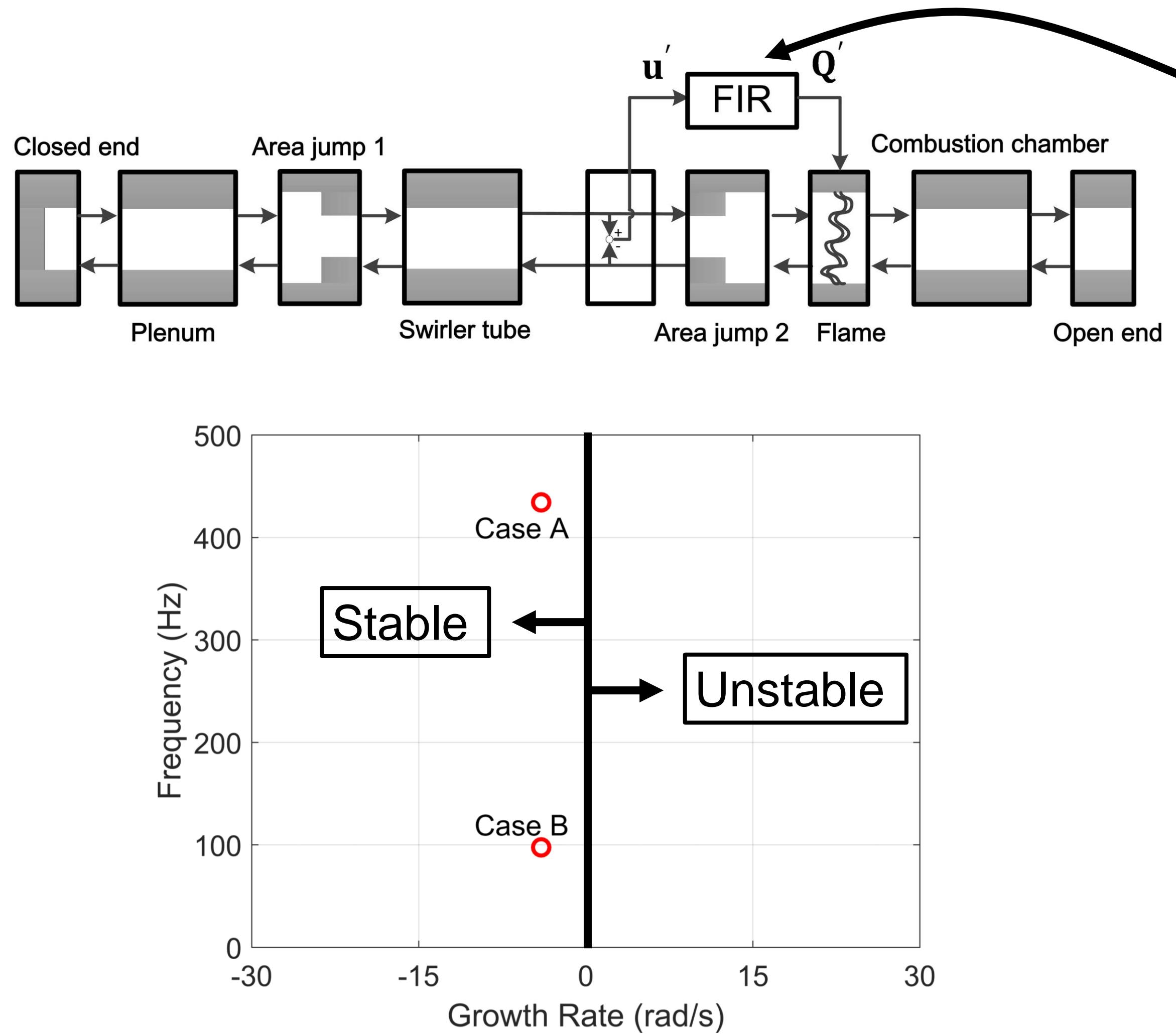
# Active Subspace approach offers much more efficient uncertainty propagation by reducing the problem dimensionality



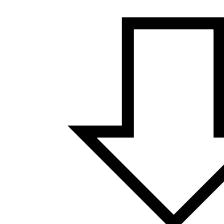
# Presentation overview

- Motivation
- Methodology
- Thermoacoustic case studies
- Results discussion
- Conclusions

# Case studies: acoustic network model, impulse response identification and thermoacoustic mode specification



$$(h_1, h_2, \dots, h_{16})$$



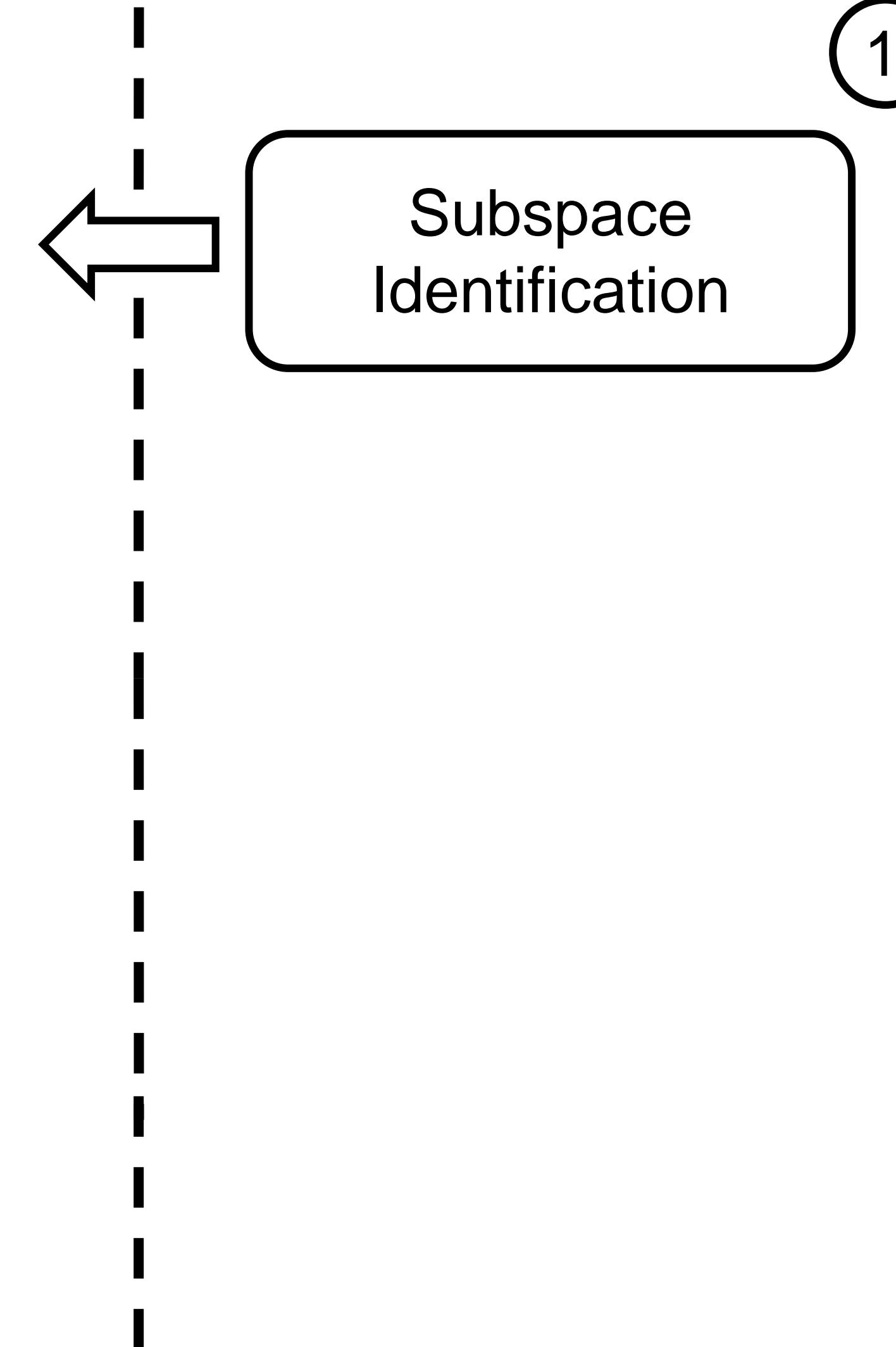
Multivariate Normal Distribution

# Active Subspace approach identified a single active variable in each case

$$Y^A = a_1^A \cdot h_1 + a_2^A \cdot h_2 + \dots + a_{16}^A \cdot h_{16}$$

$$Y^B = a_1^B \cdot h_1 + a_2^B \cdot h_2 + \dots + a_{16}^B \cdot h_{16}$$

Dimensionality Reduction: 16D  $\rightarrow$  1D

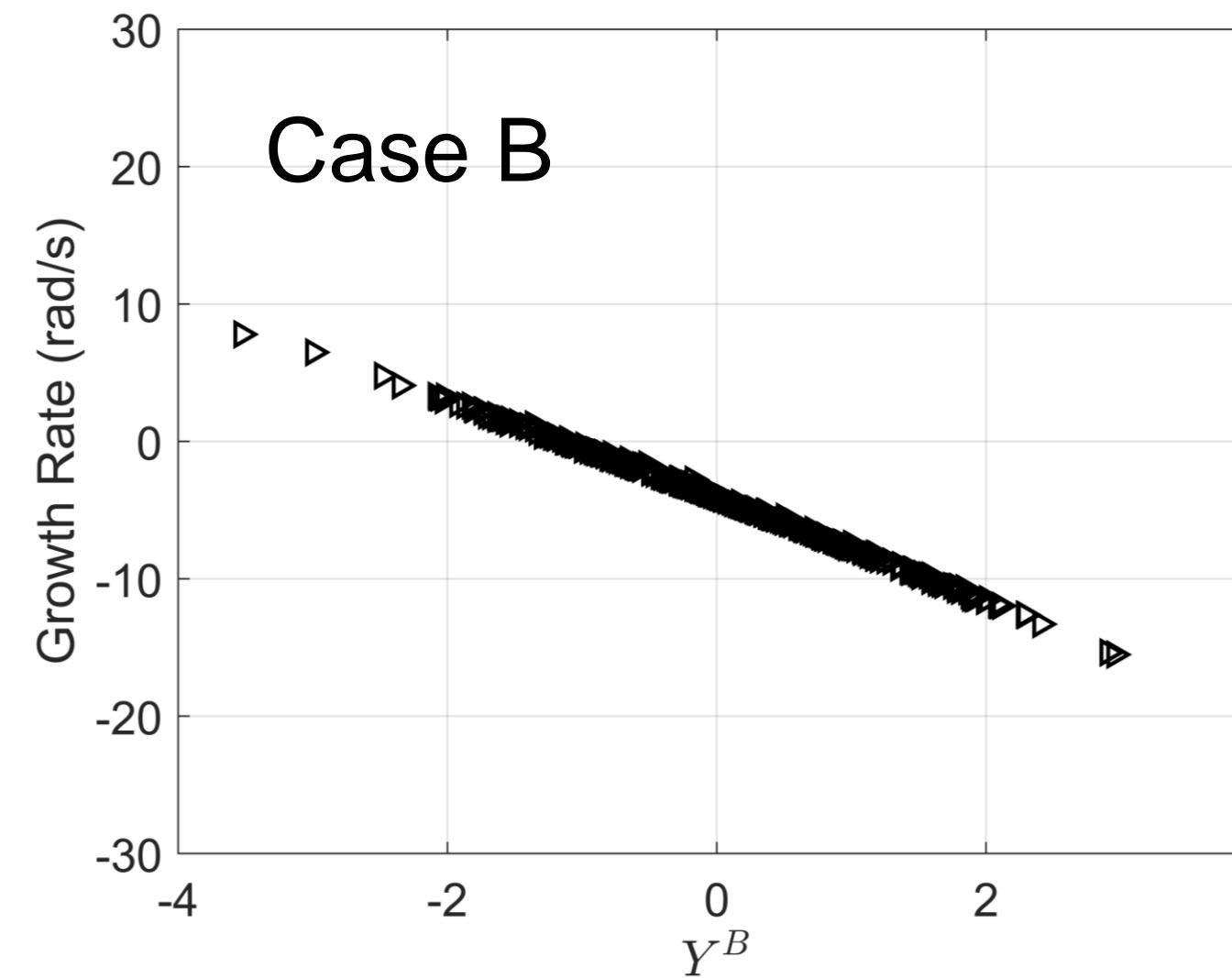
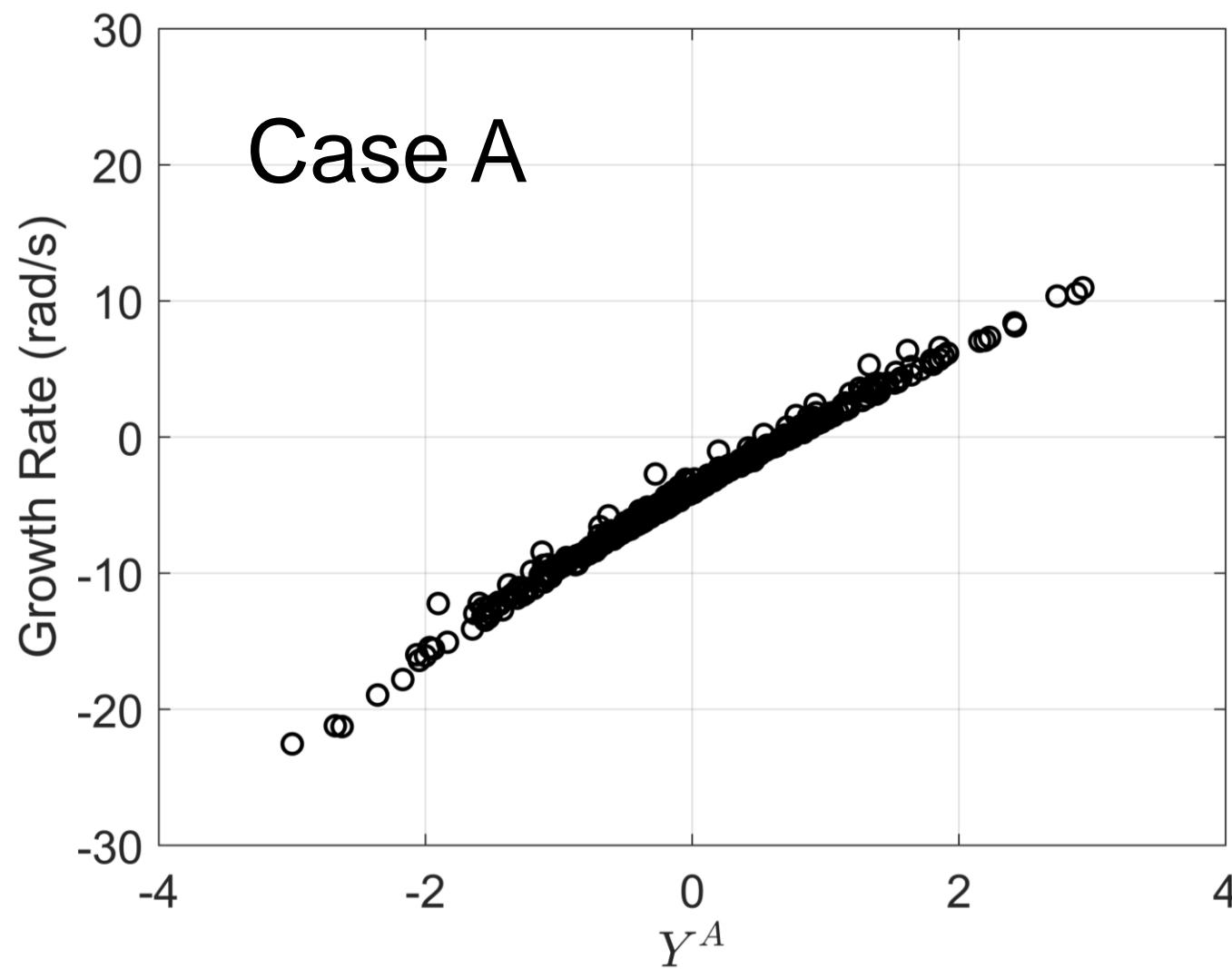


# A surrogate model is trained to map from active variable value to growth rate value for each case

$$Y^A = a_1^A \cdot h_1 + a_2^A \cdot h_2 + \dots + a_{16}^A \cdot h_{16}$$

$$Y^B = a_1^B \cdot h_1 + a_2^B \cdot h_2 + \dots + a_{16}^B \cdot h_{16}$$

Dimensionality Reduction: 16D  $\rightarrow$  1D



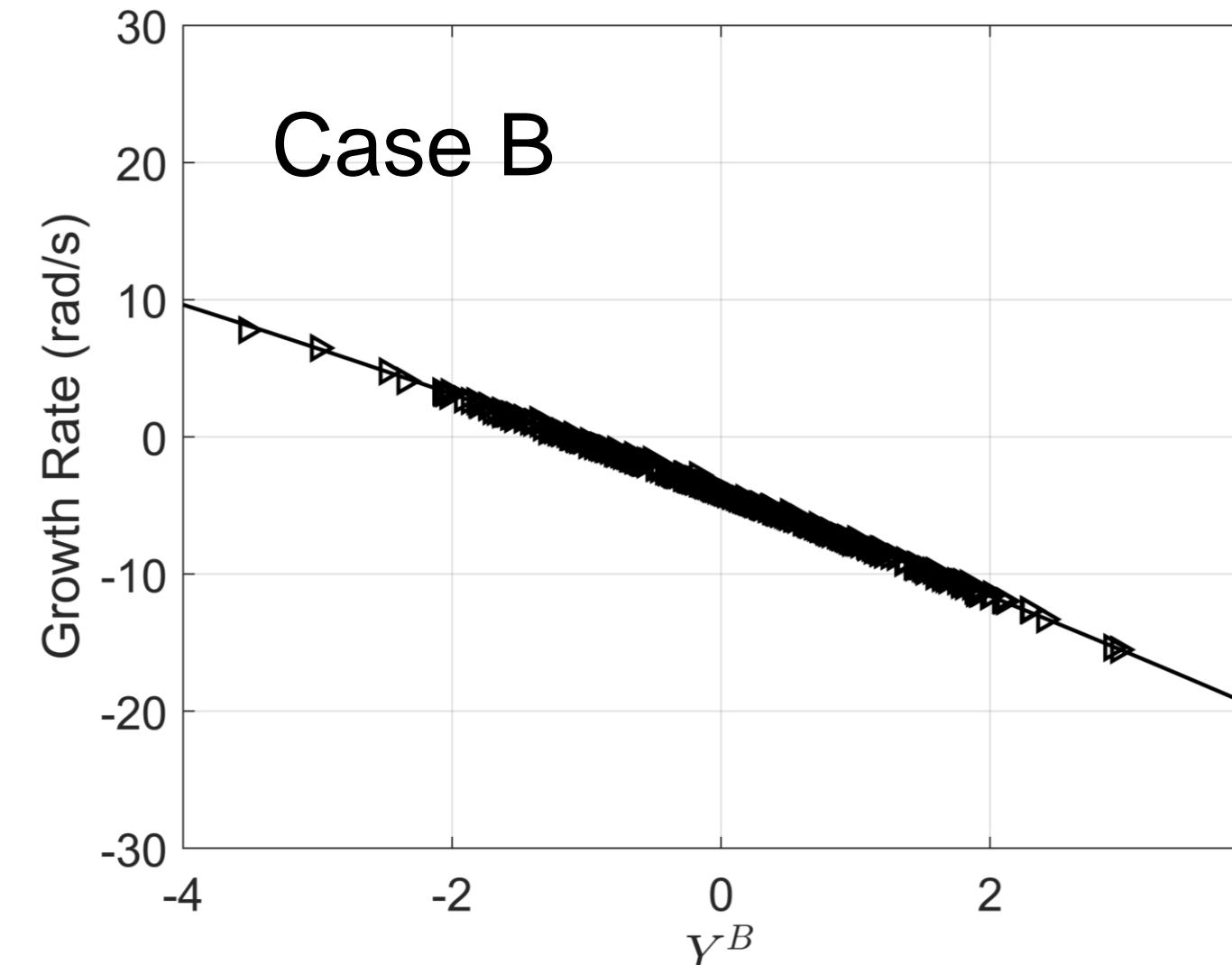
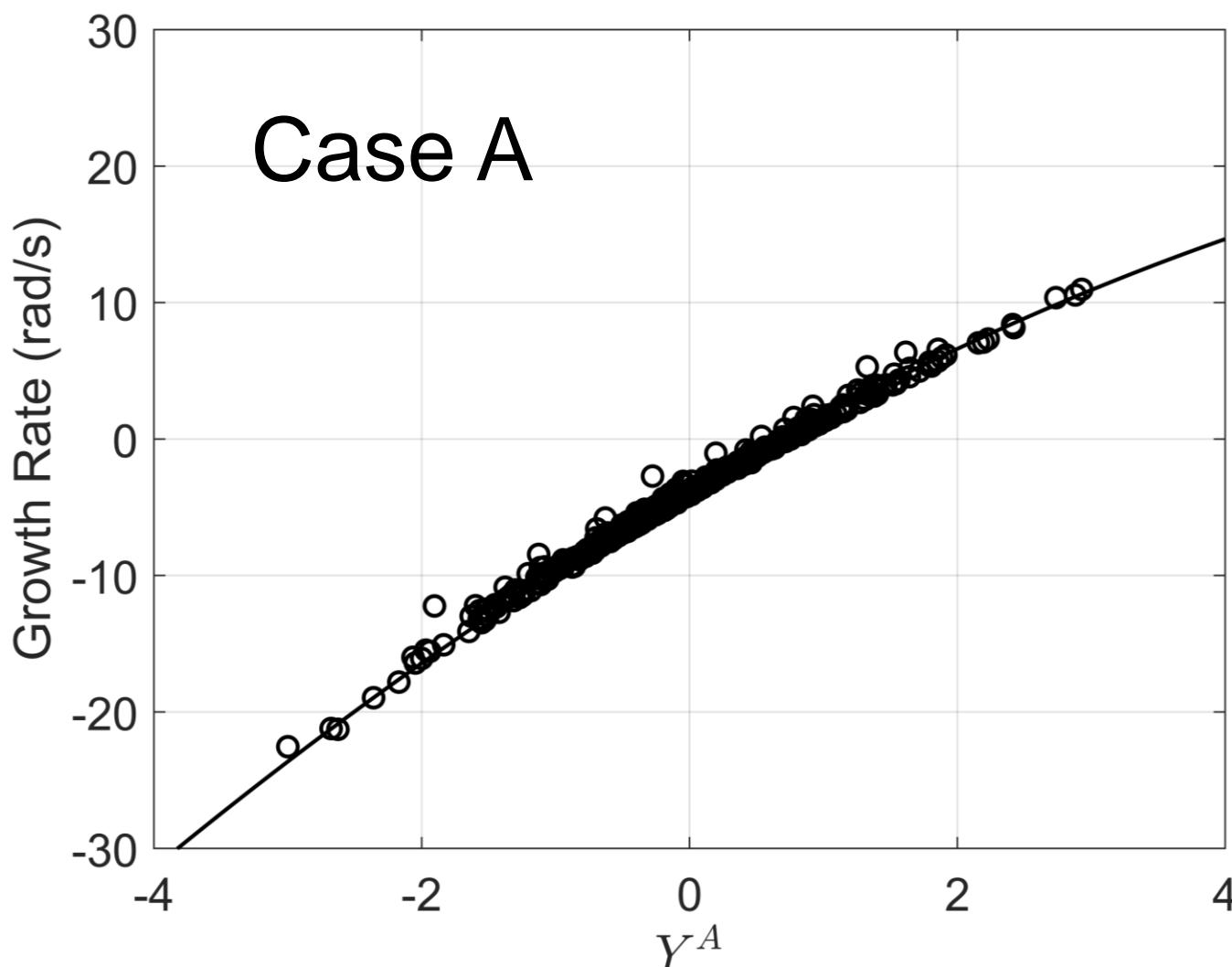
Subspace  
Identification

Surrogate  
Model Training

# A quadratic function is trained to map from active variable value to growth rate value for each case

$$Y^A = a_1^A \cdot h_1 + a_2^A \cdot h_2 + \dots + a_{16}^A \cdot h_{16}$$

$$Y^B = a_1^B \cdot h_1 + a_2^B \cdot h_2 + \dots + a_{16}^B \cdot h_{16}$$



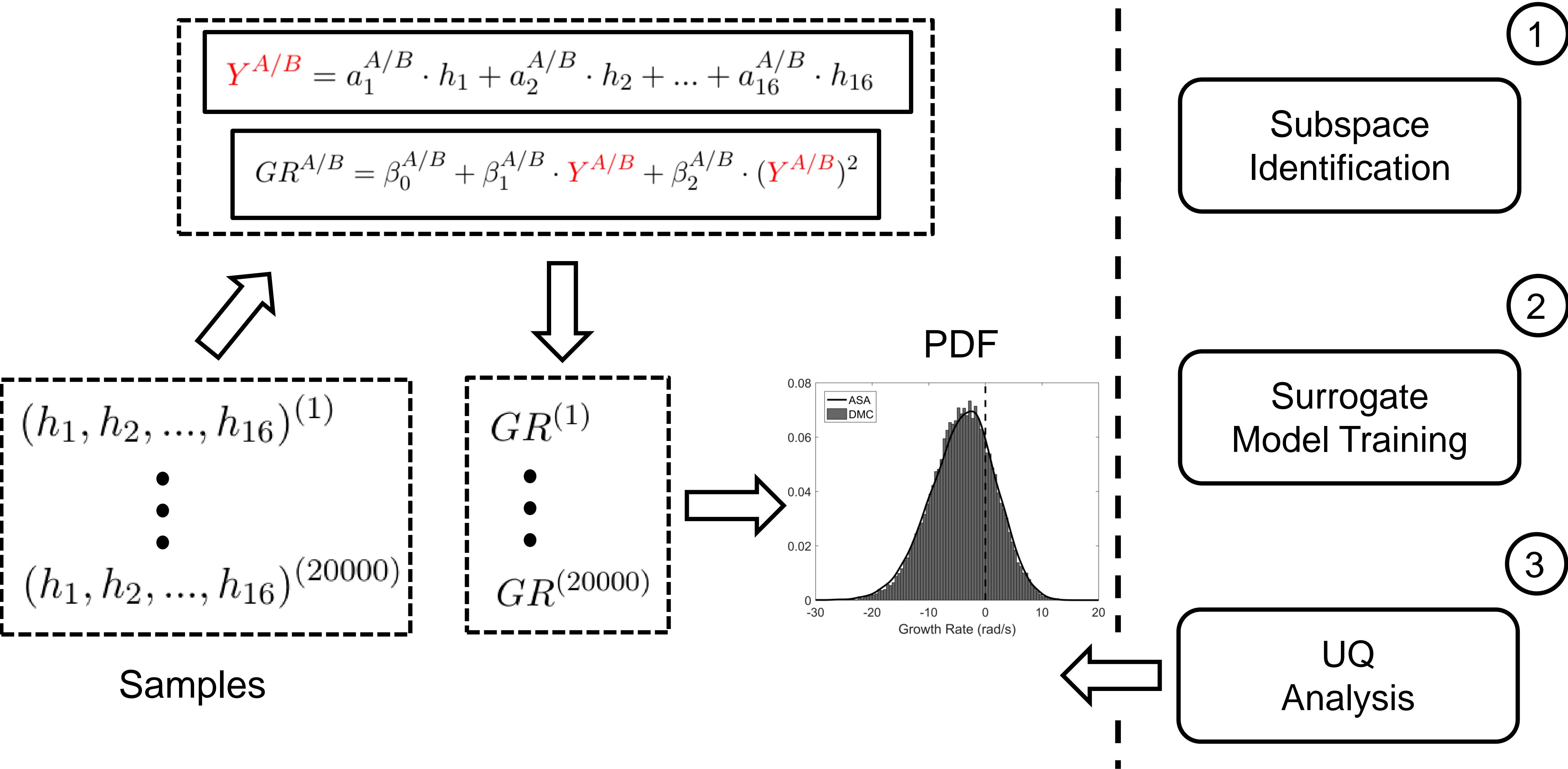
$$GR^A = \beta_0^B + \beta_1^B(Y^A) + \beta_2^B(Y^A)^2$$

$$GR^B = \beta_0^B + \beta_1^B(Y^B) + \beta_2^B(Y^B)^2$$

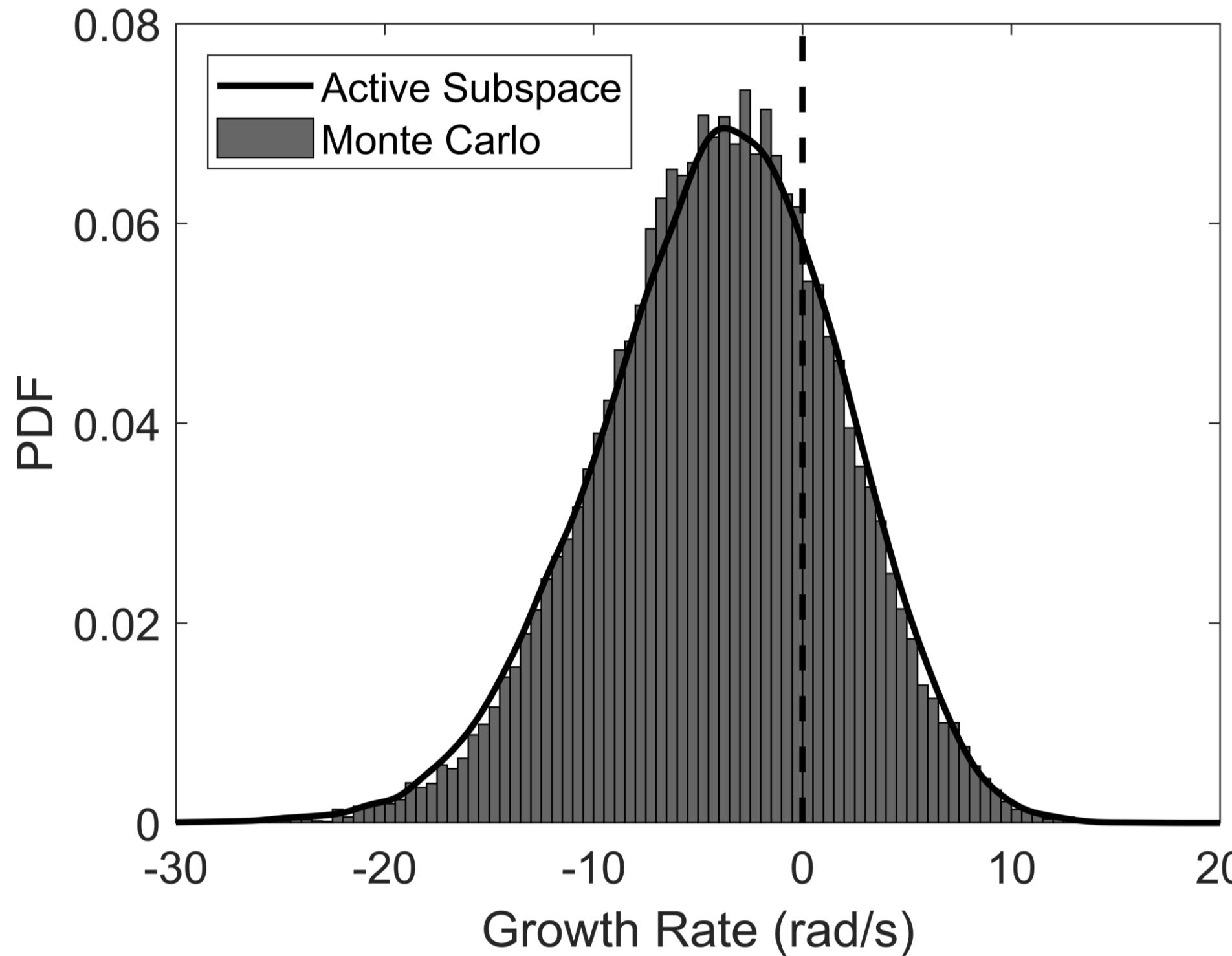
1  
Subspace Identification

2  
Surrogate Model Training

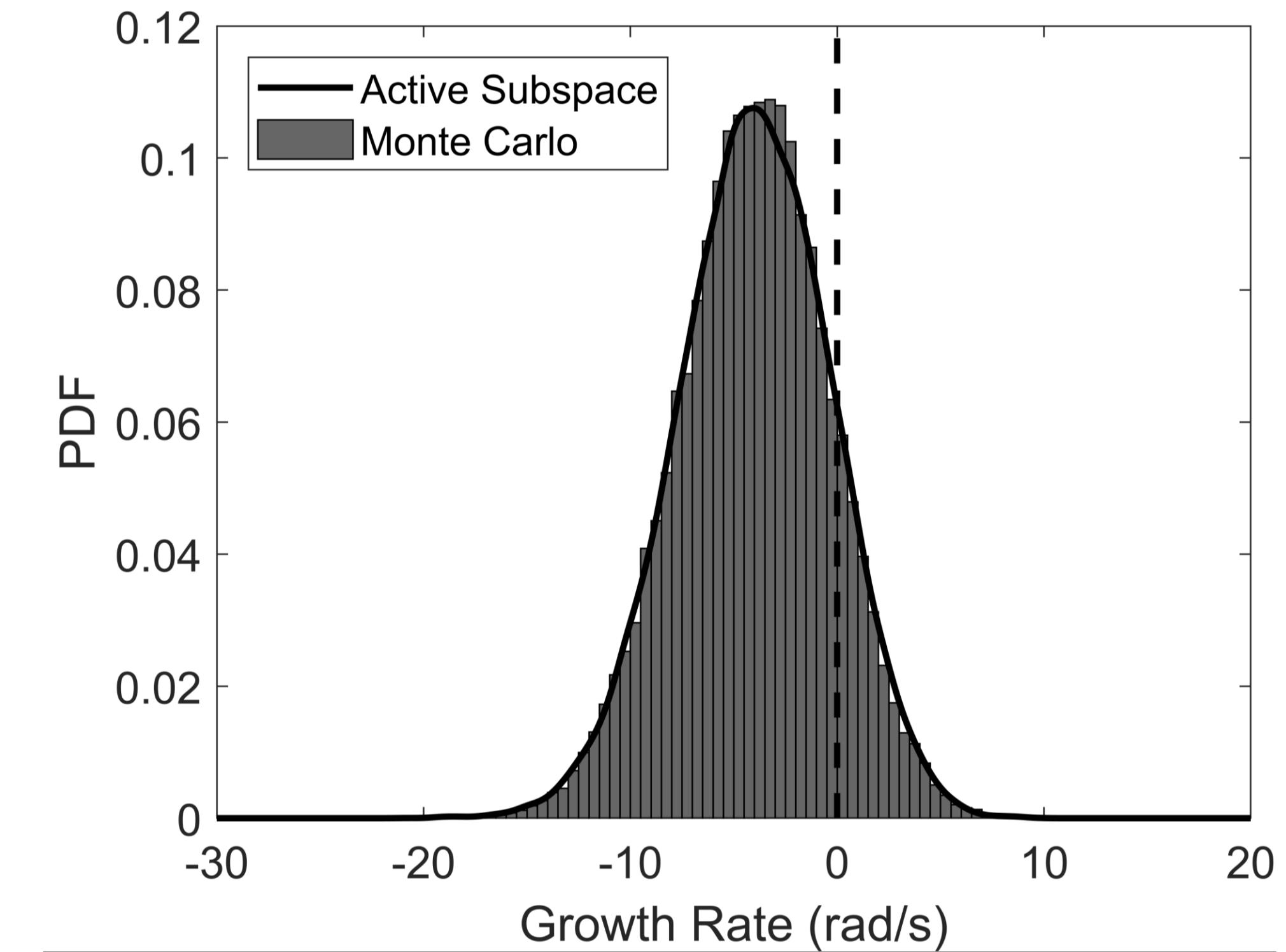
# Monte Carlo simulation can be directly applied on the surrogate model to accelerate the UQ analysis



# Active Subspace replicated the reference Monte Carlo results with 50 times less computational cost



Case A



Case B

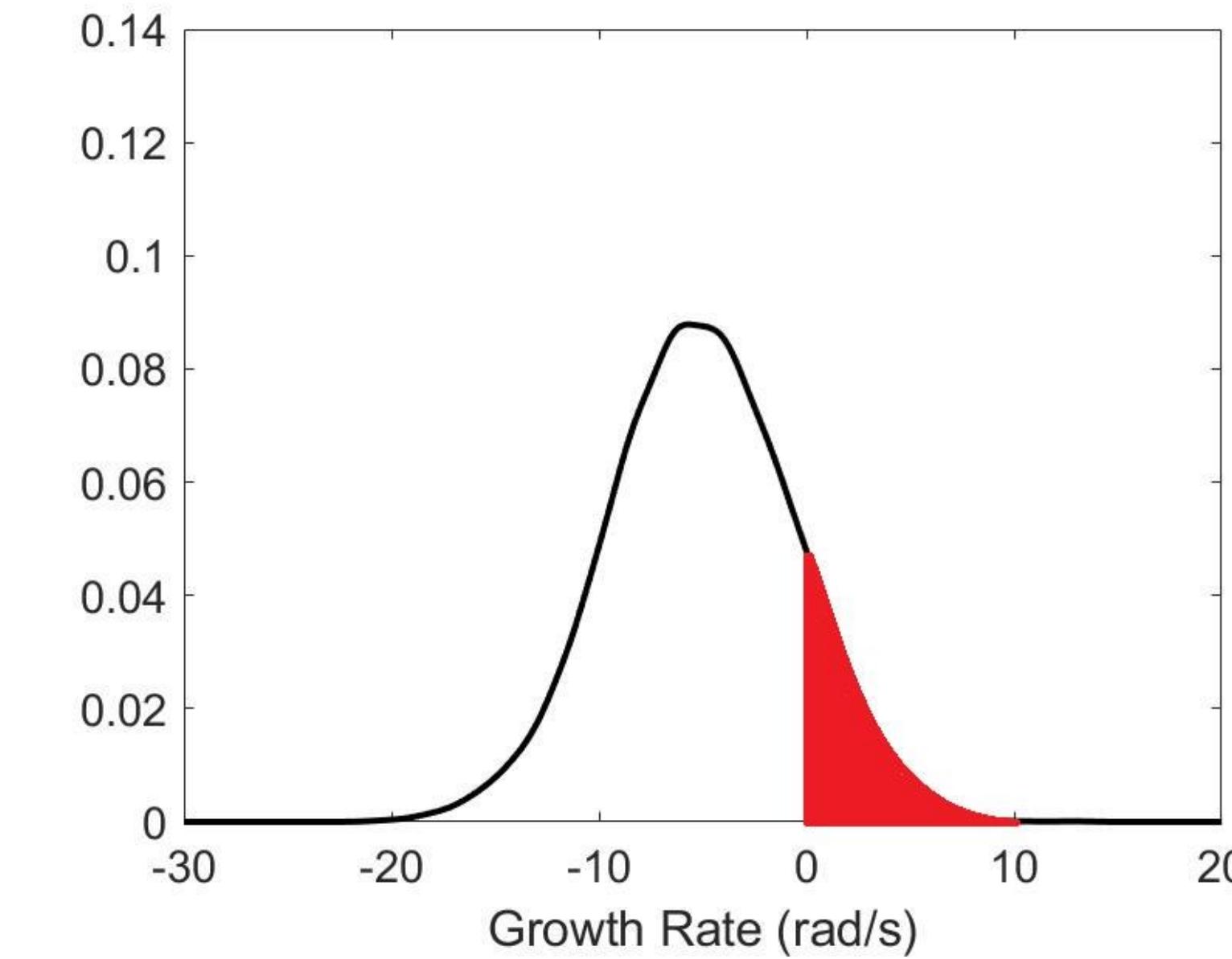
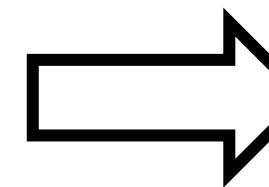
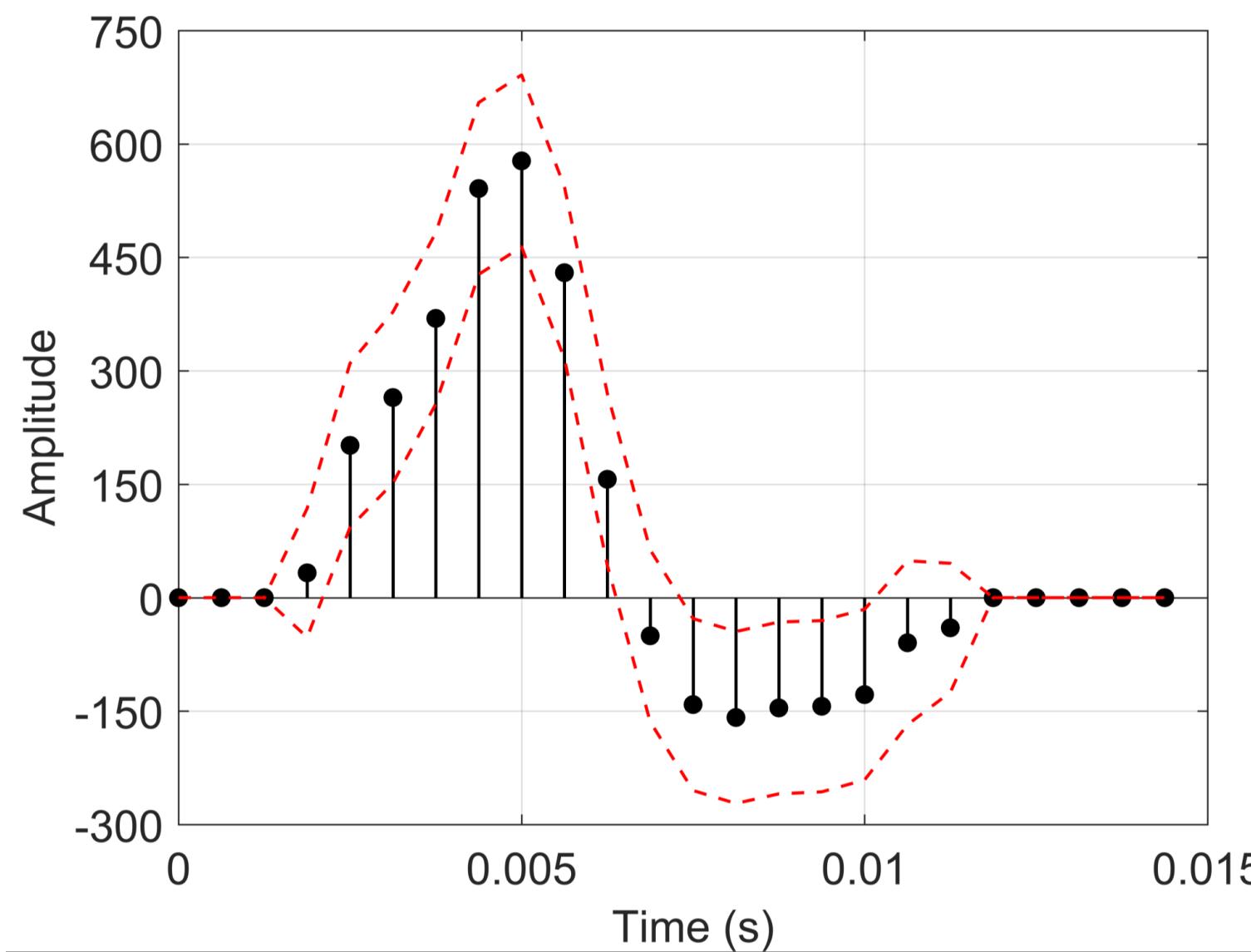
Computational Cost:

Monte Carlo: **20000** acoustic network calculations  
Active Subspace: **400** acoustic network calculations

# Active Subspace is very efficient in quantifying the impact of FIR model uncertainty on thermoacoustic instability prediction

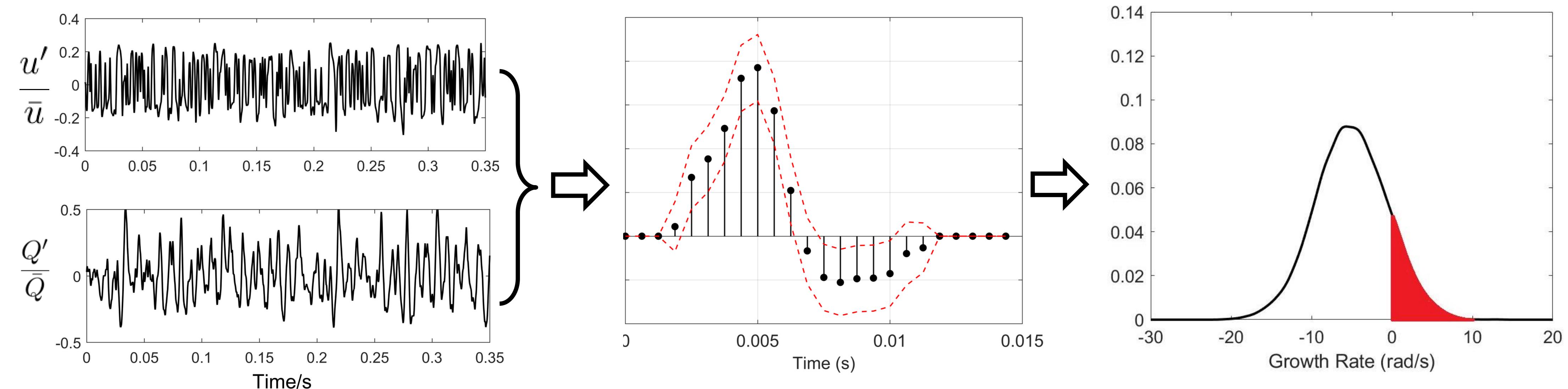
Problem Solved:

How to efficiently propagate uncertainties in flame impulse response model to the growth rate predictions?

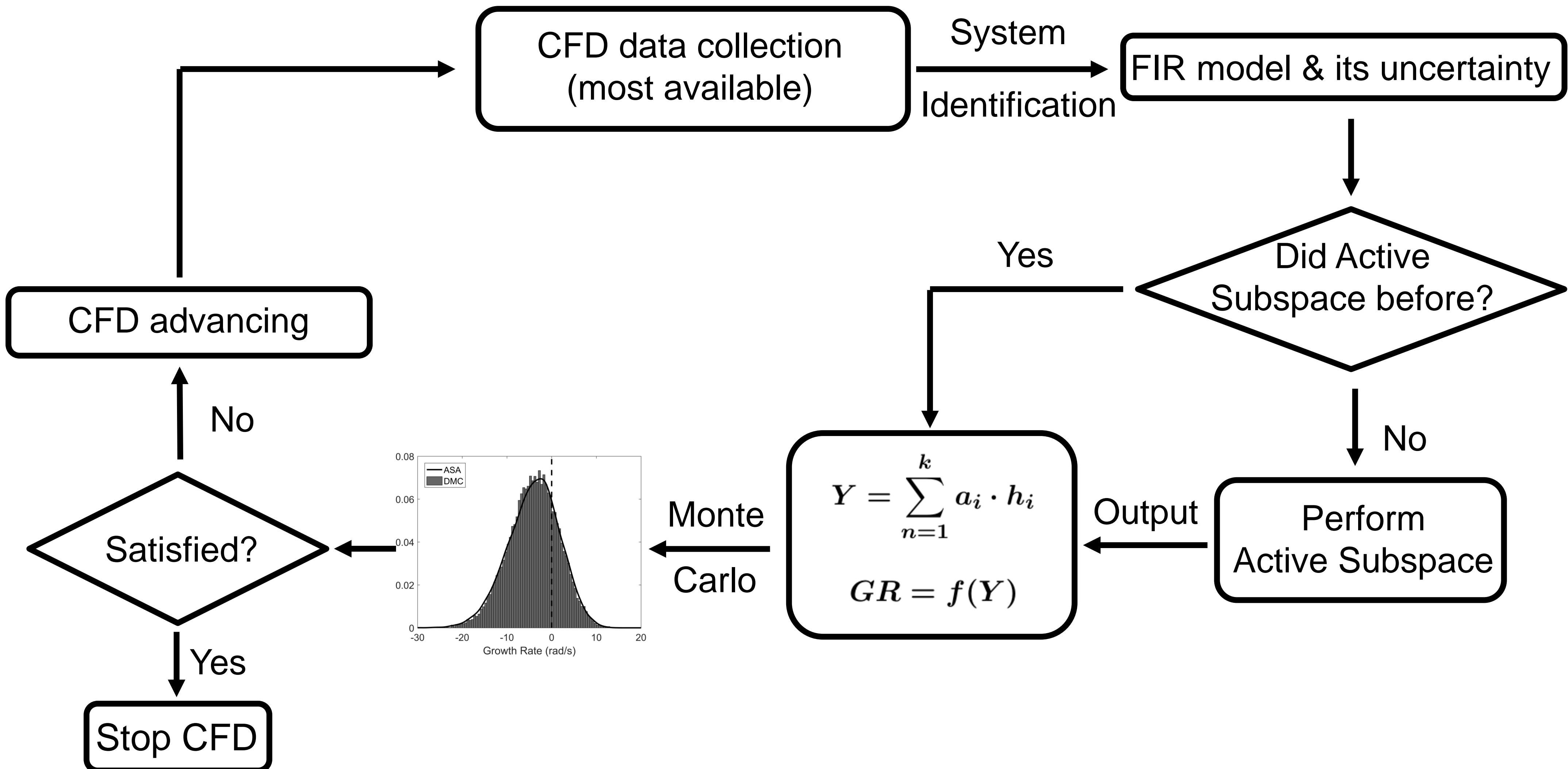


# The main question we want to address

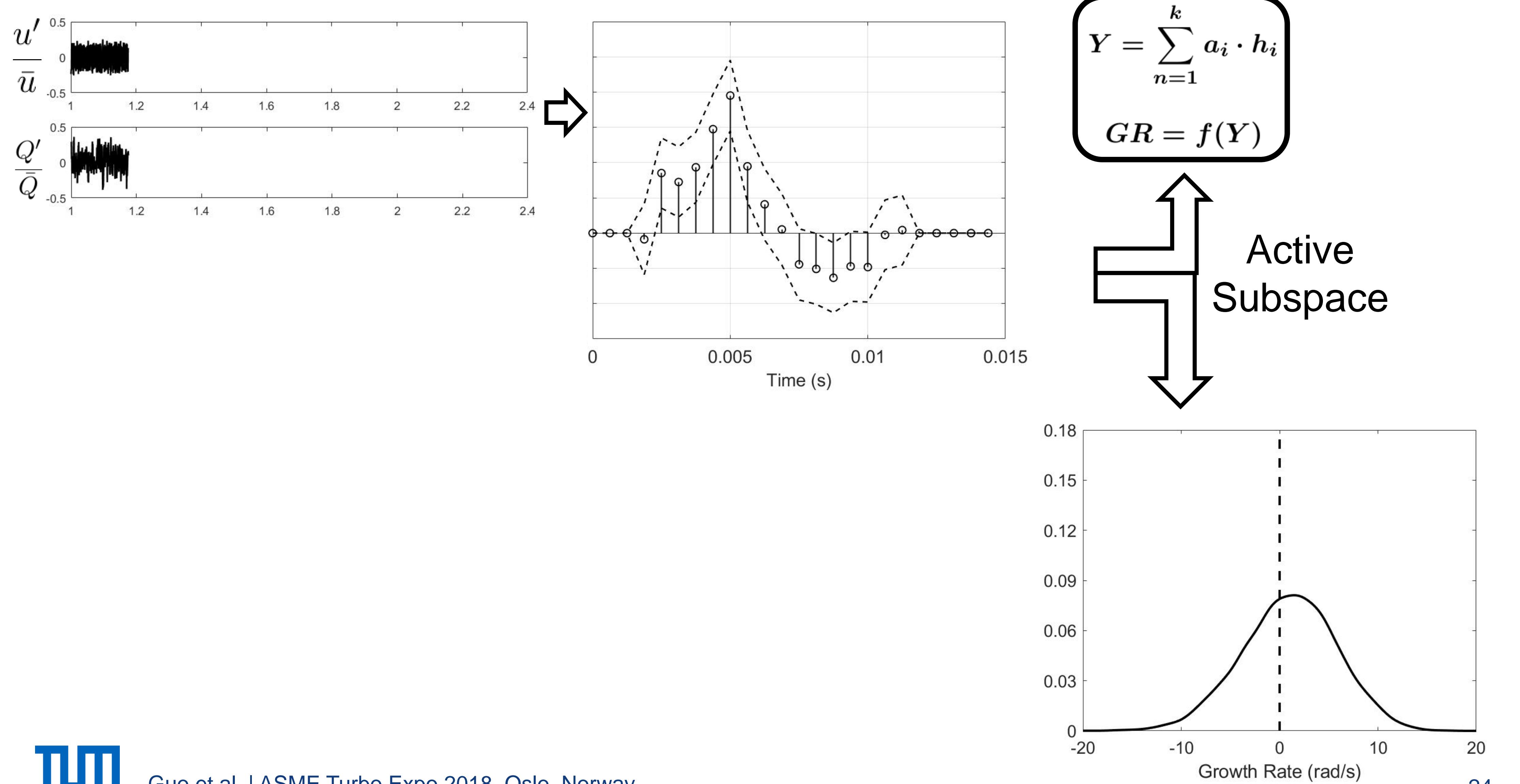
**Problem:** How to determine the adequate length of CFD time series to achieve a satisfactory uncertainty reduction in growth rate prediction?



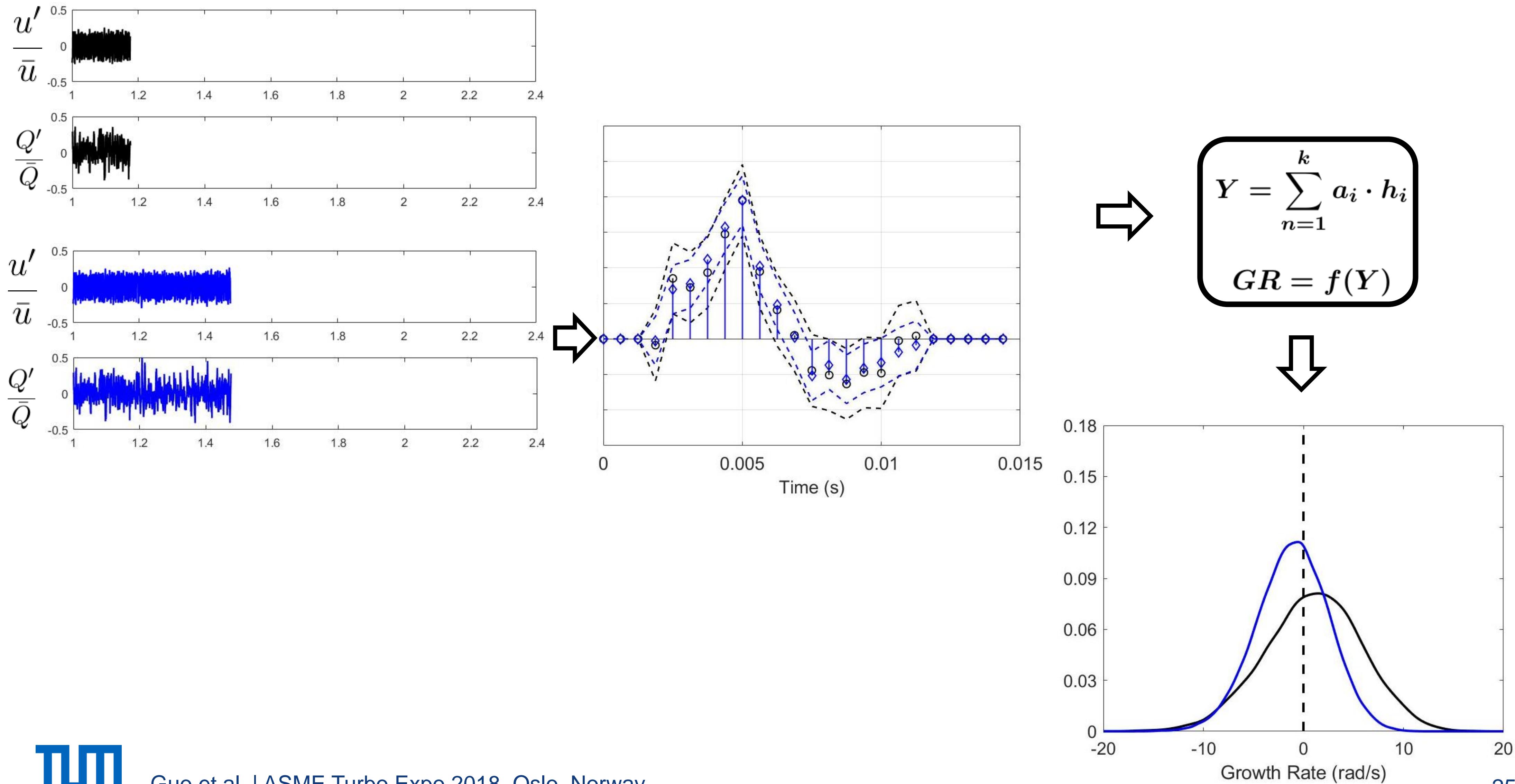
# Procedure for estimating the CFD time series length based on Active Subspace approach



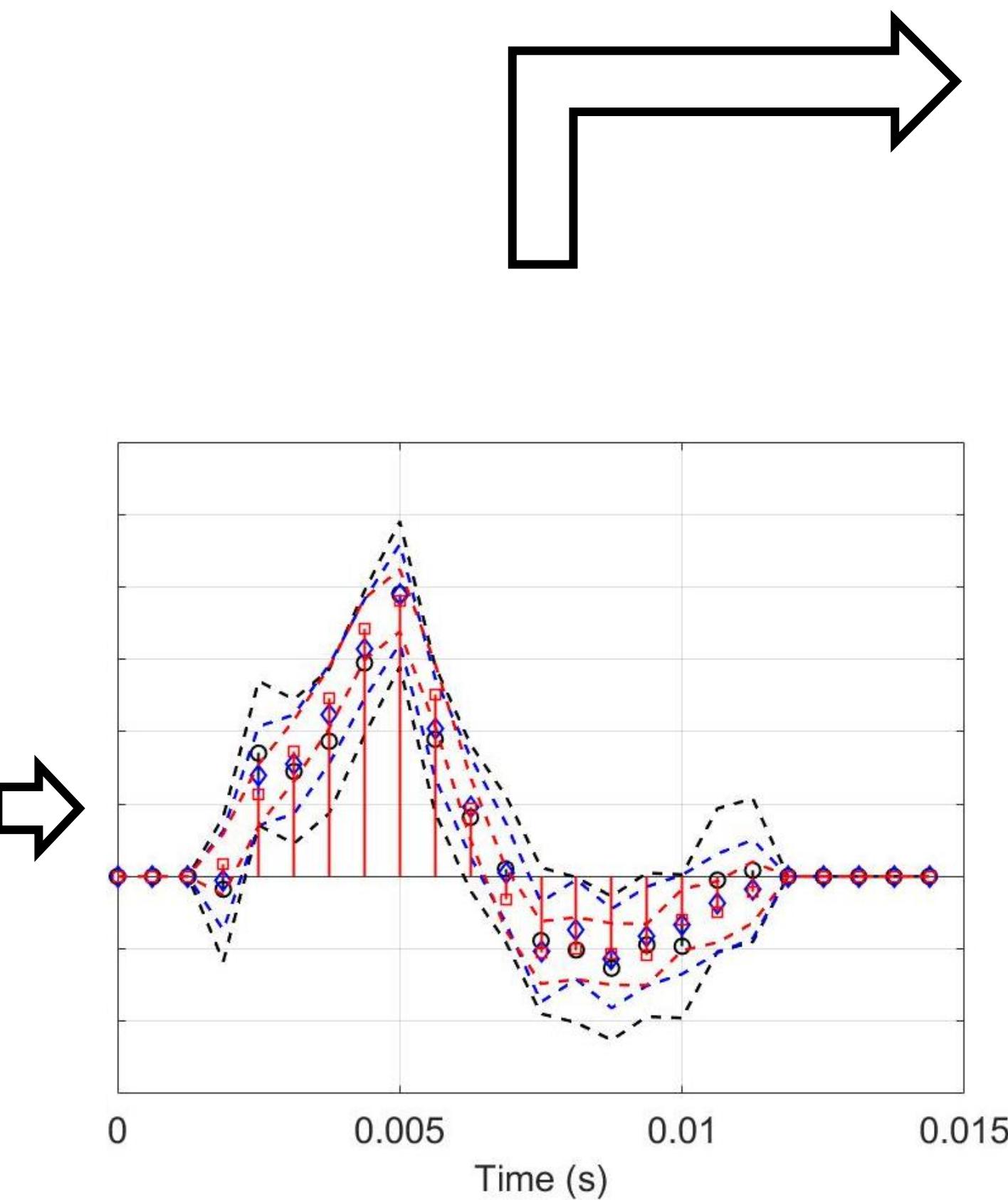
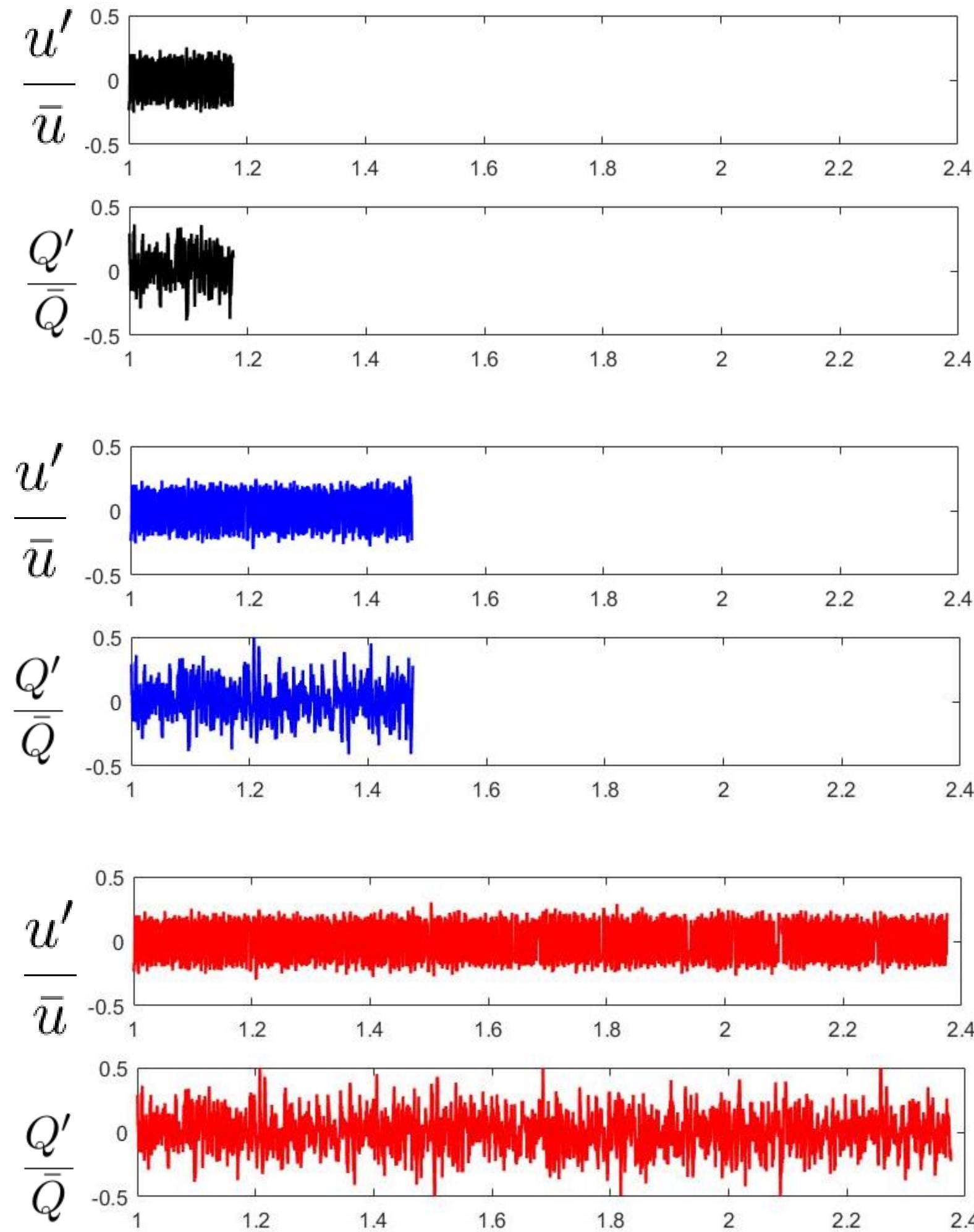
# Procedure demonstration



# Procedure demonstration

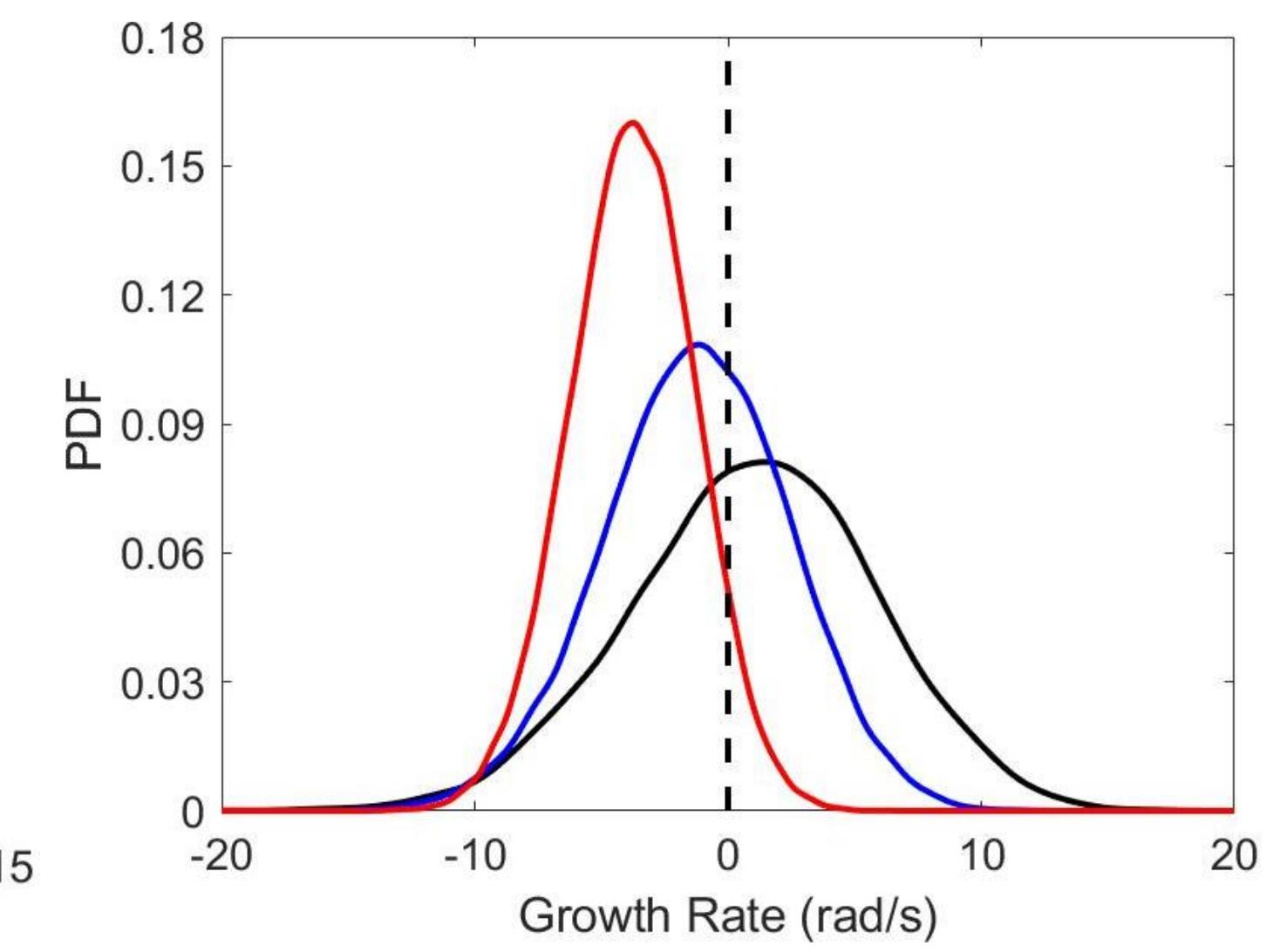


# Procedure demonstration



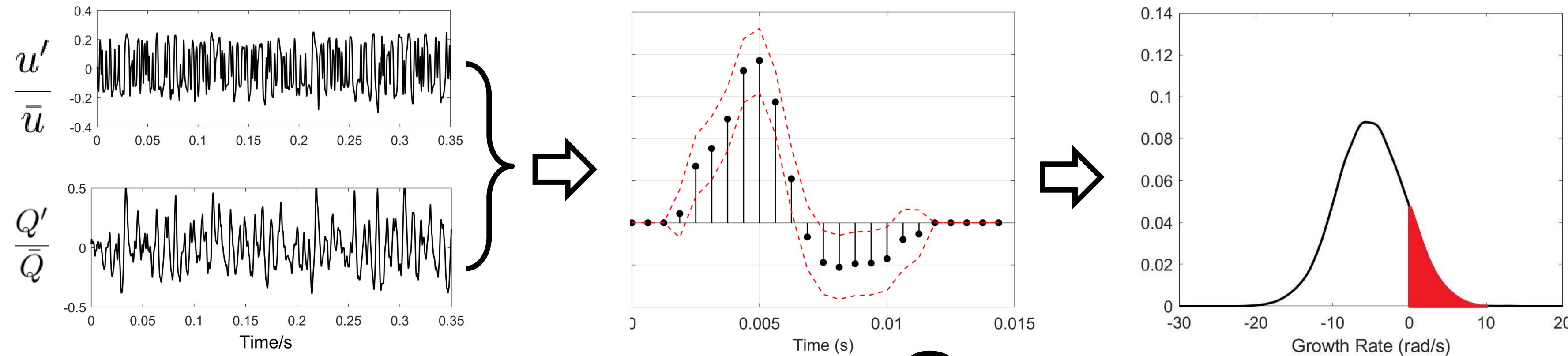
$$Y = \sum_{n=1}^k a_i \cdot h_i$$

$$GR = f(Y)$$

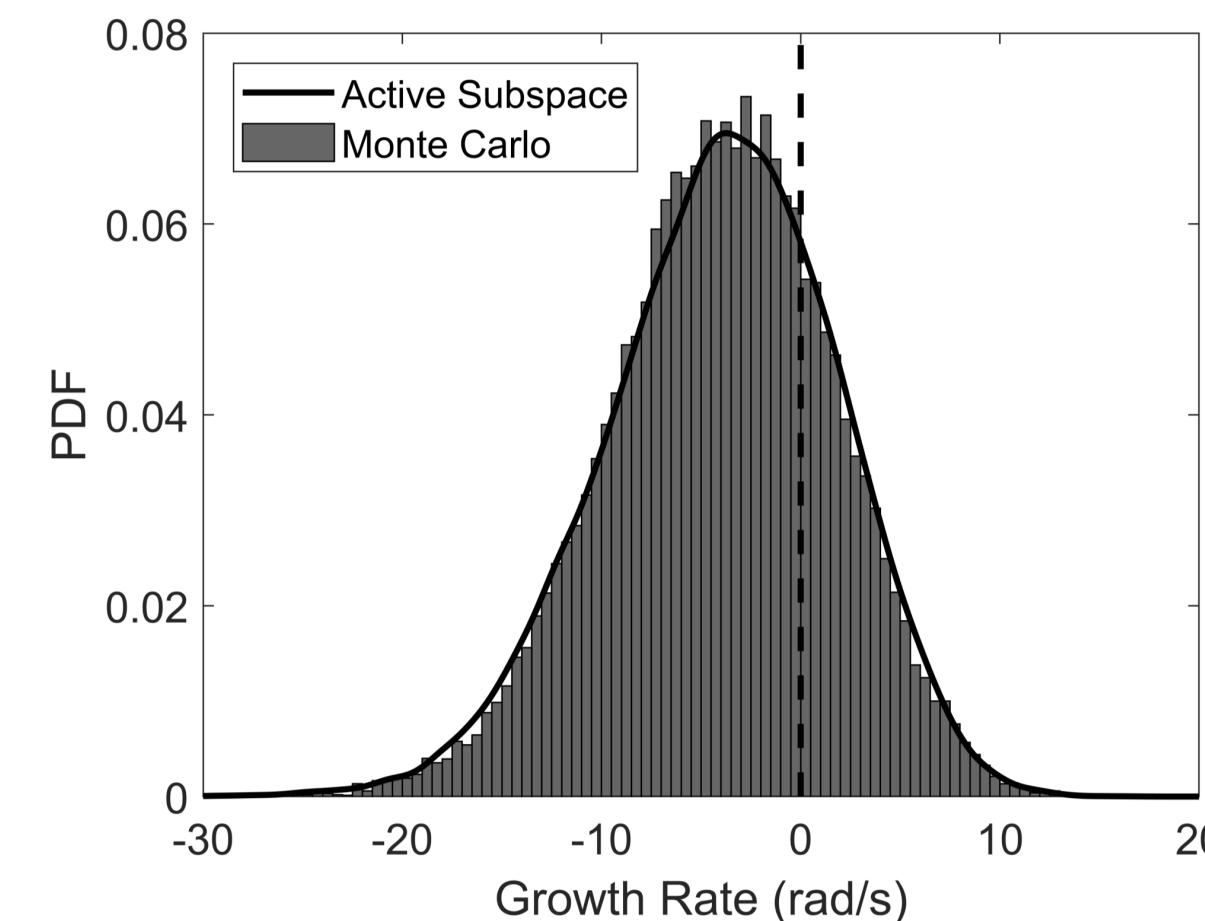


# Conclusions

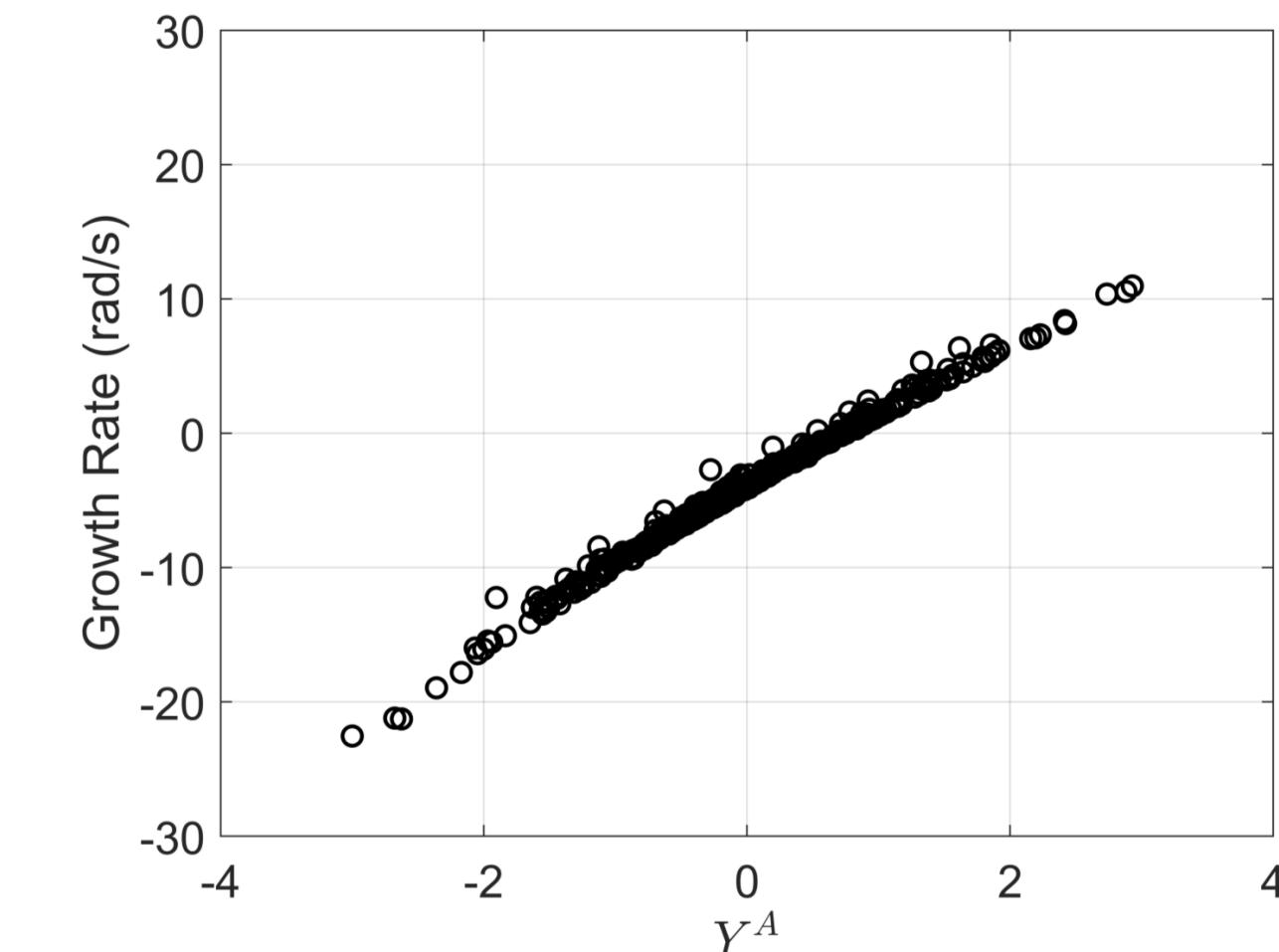
- 1 A procedure to estimate adequate length of CFD time series



- 2 Active Subspace offers more efficient UQ analysis



- 3 One Active Variable is sufficient

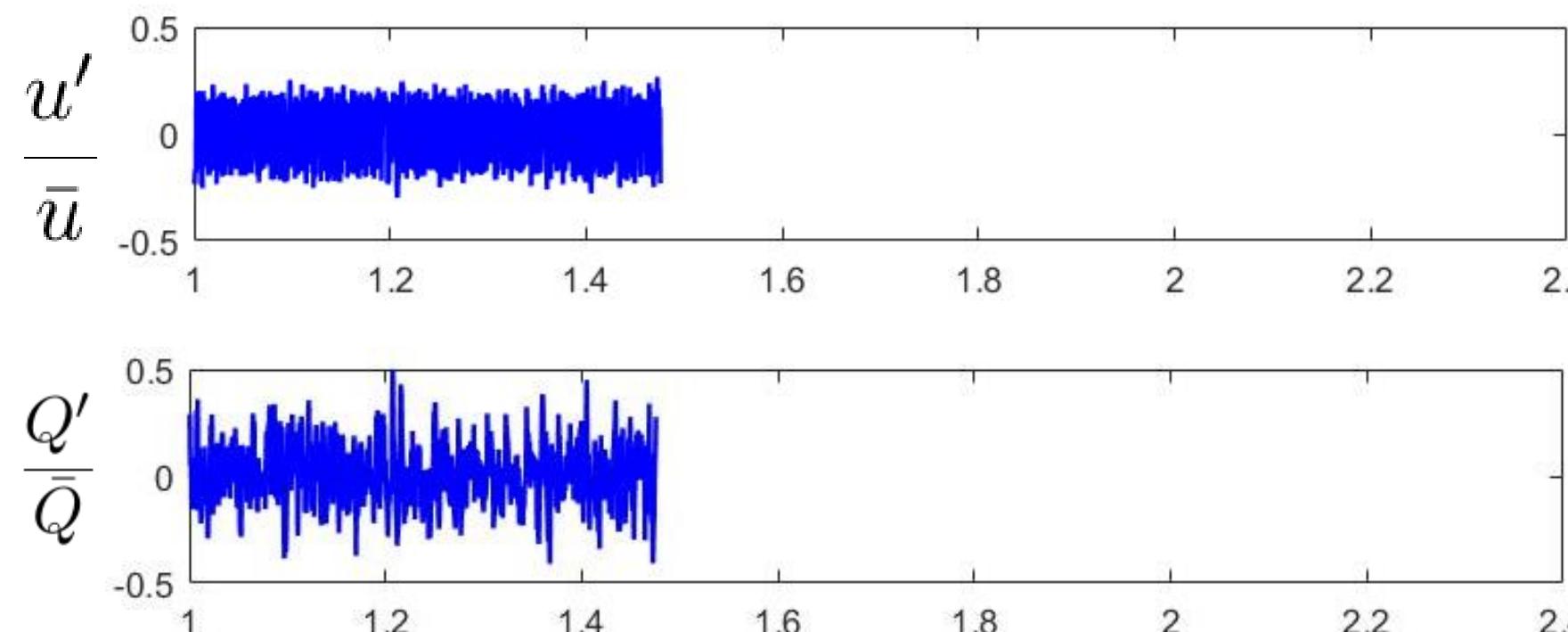


# Back-up Slides

# Impulse Response Identification

$$\frac{\dot{Q}'_n}{\bar{Q}} = \frac{1}{\bar{u}_u} \sum_{k=0}^{L-1} h_k u'_{u,n-k}$$

Convolution



$u'_u$  Velocity fluctuation upstream of the flame  
 $\dot{Q}'$  Global heat release rate fluctuation

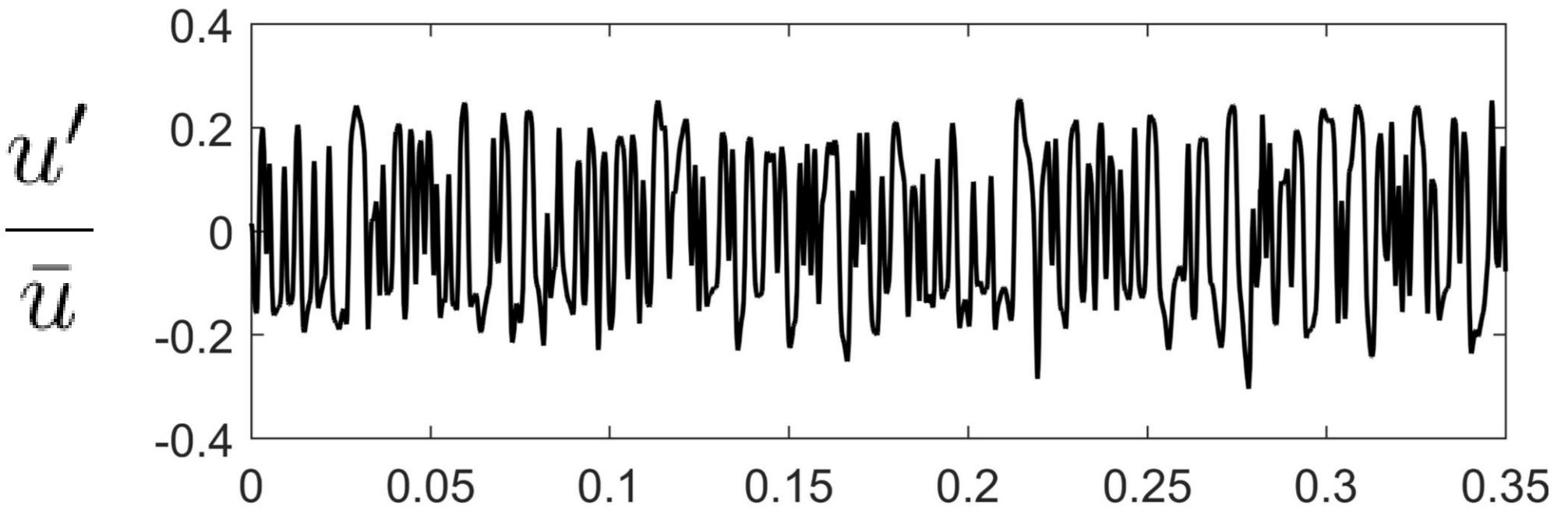
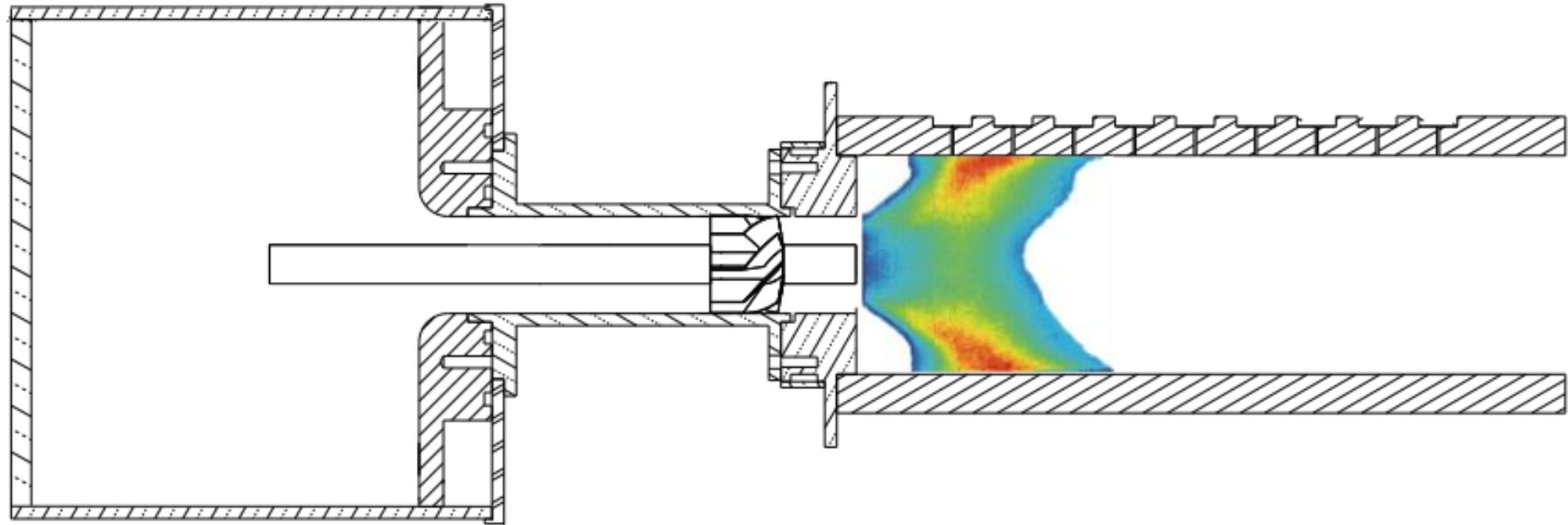
Least Square



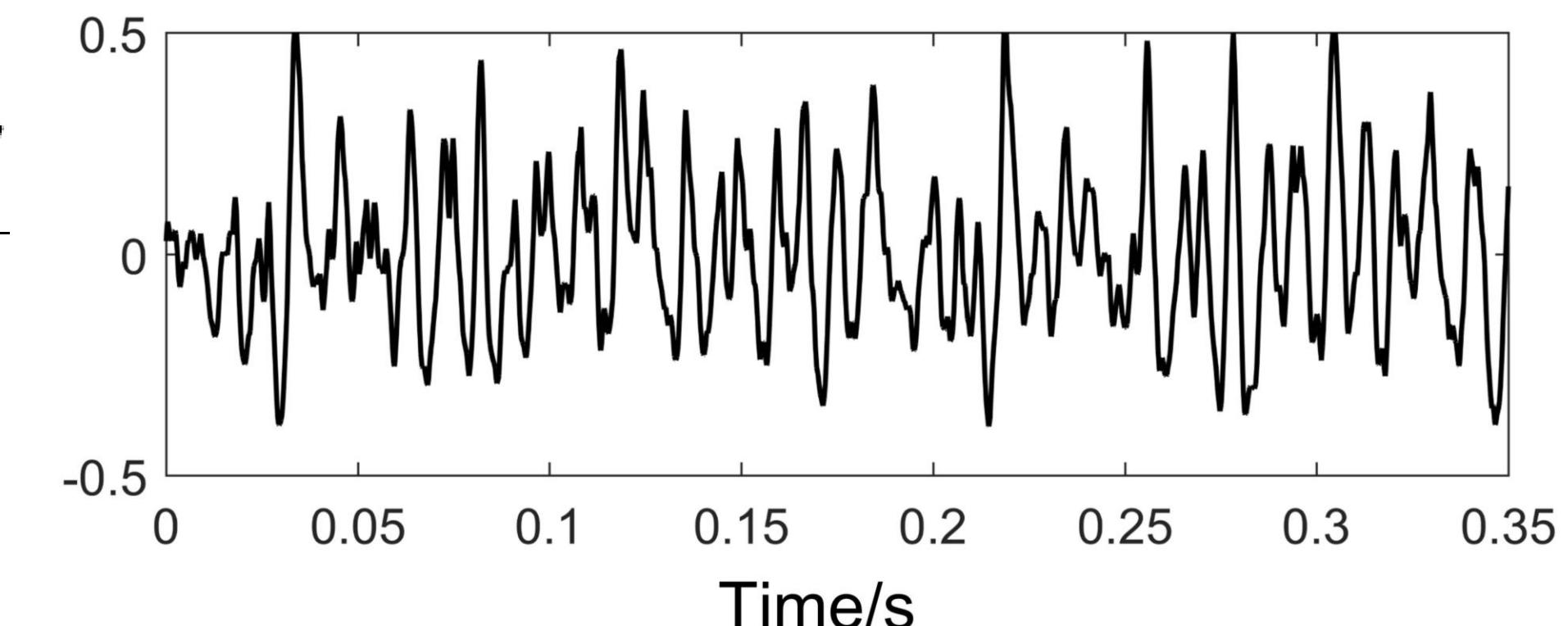
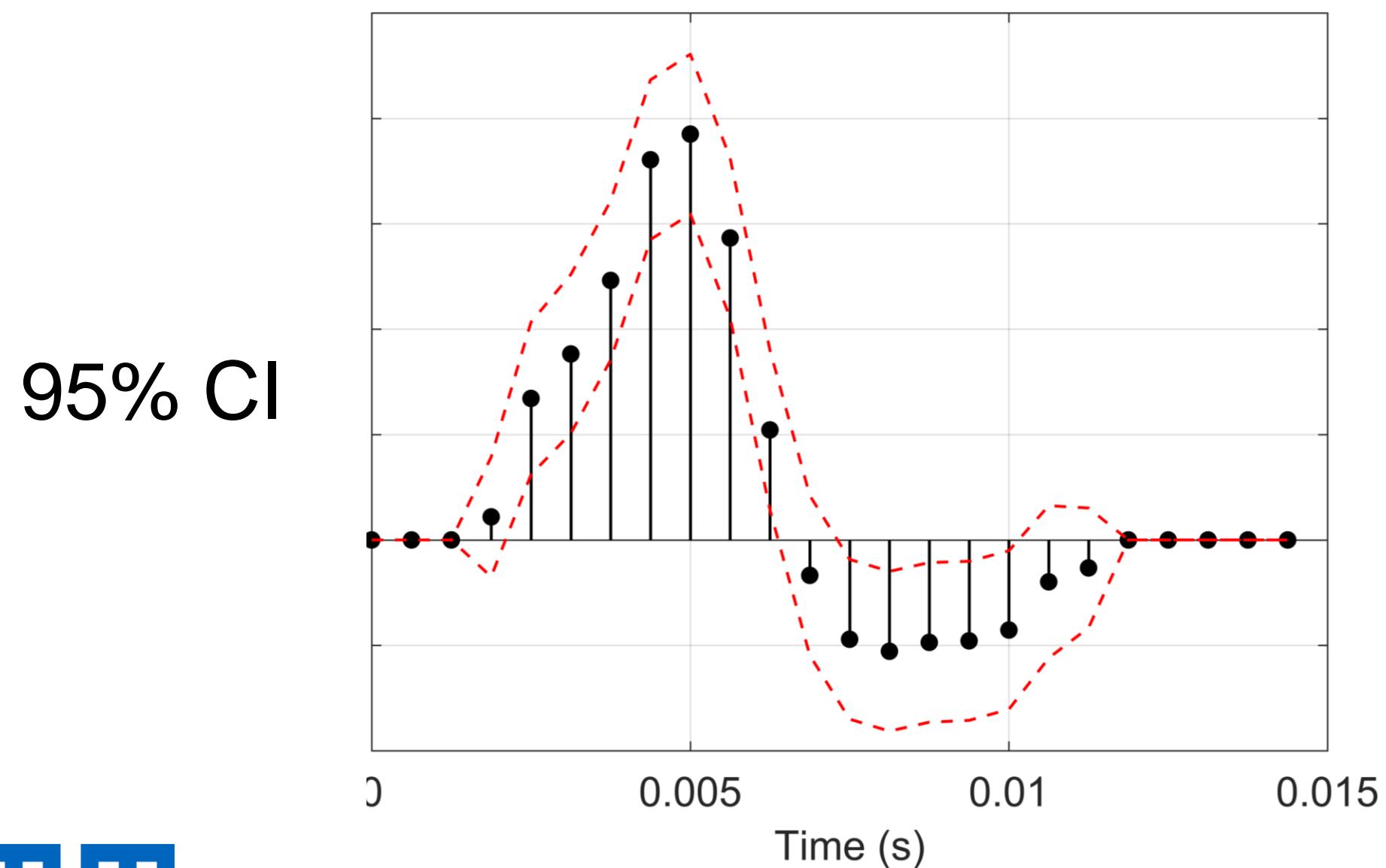
Estimations

Estimation Error  
(Covariance matrix)

# Current Impulse Response Model is identified through LES simulations



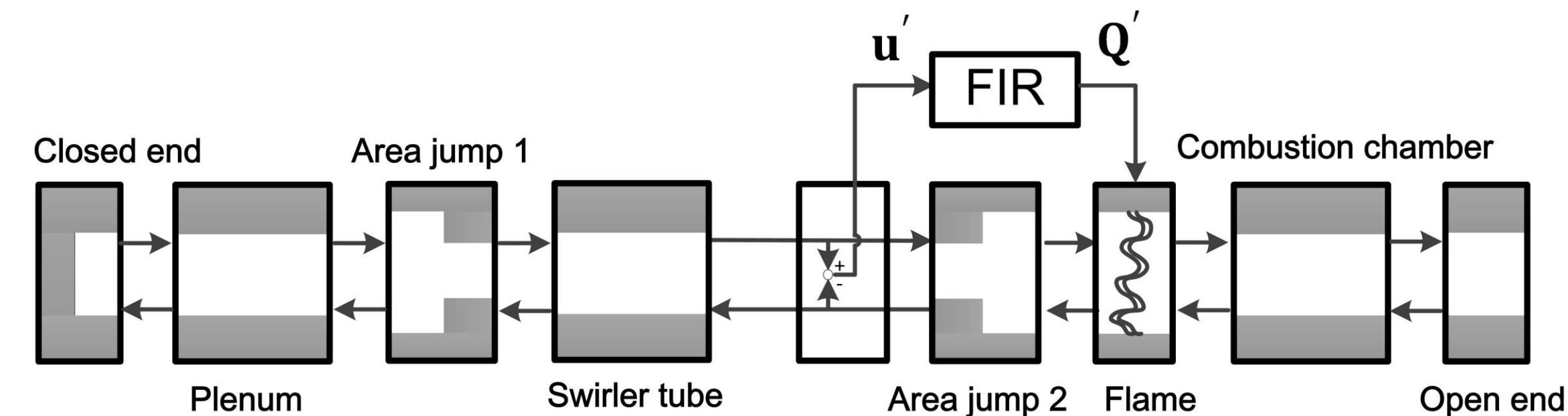
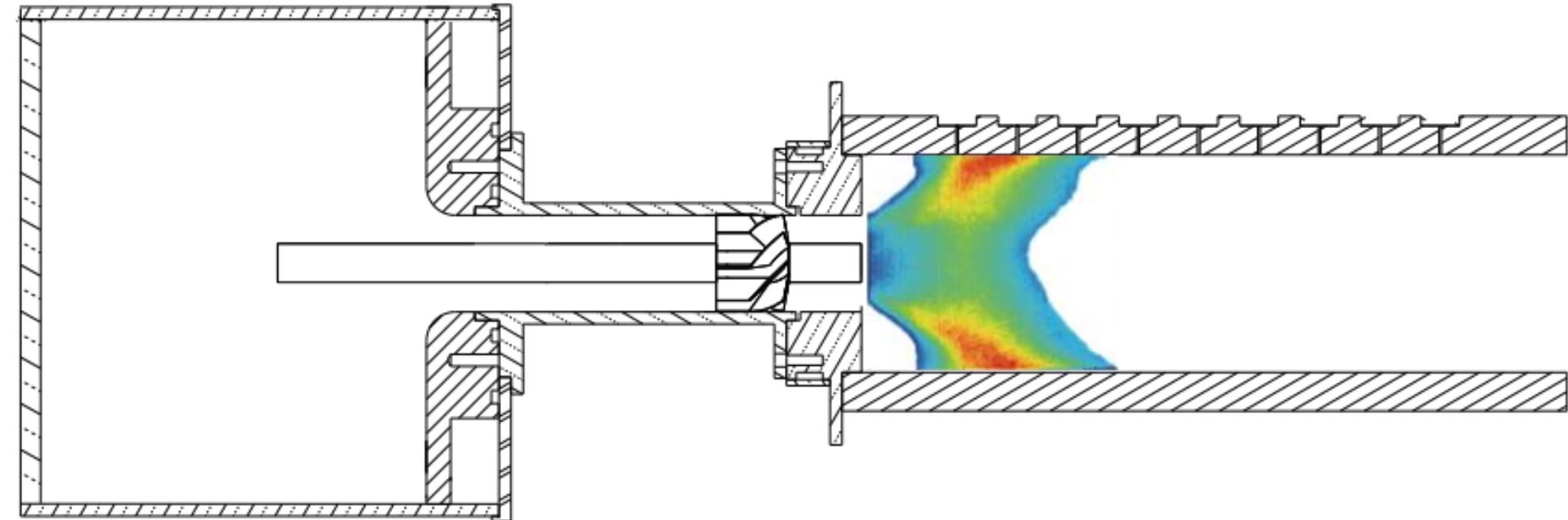
Test Rig: A turbulent, premixed Swirl Burner



LES time series<sup>1</sup>

[1] Luis Tay-Wo-Chong et al., 2010 in ASME

# Current Acoustic Network Model and its model parameters



Acoustic element	Parameters
Closed end	Reflection coefficient = 1
Plenum	Length = 0.17m Sound speed = 343 m/s
Area jump 1	Area ratio = 29.76
Swirler tube	Length = 0.18m Sound speed = 343 m/s
Area jump 2	Area ratio = 0.13
Flame	Relative temperature jump = 5.59 Ratio of specific impedances = 2.57
FIR	Impulse response model
Combustion chamber	Length <sub>A</sub> = 0.51m; Length <sub>B</sub> = 0.6m Sound speed = 880 m/s
Combustor exit	Reflection coefficient <sub>A</sub> = -0.9883 Reflection coefficient <sub>B</sub> = -0.6351

# Active Subspace Identification

$$f = f(\vec{x}) \quad \vec{x} \in R^m$$

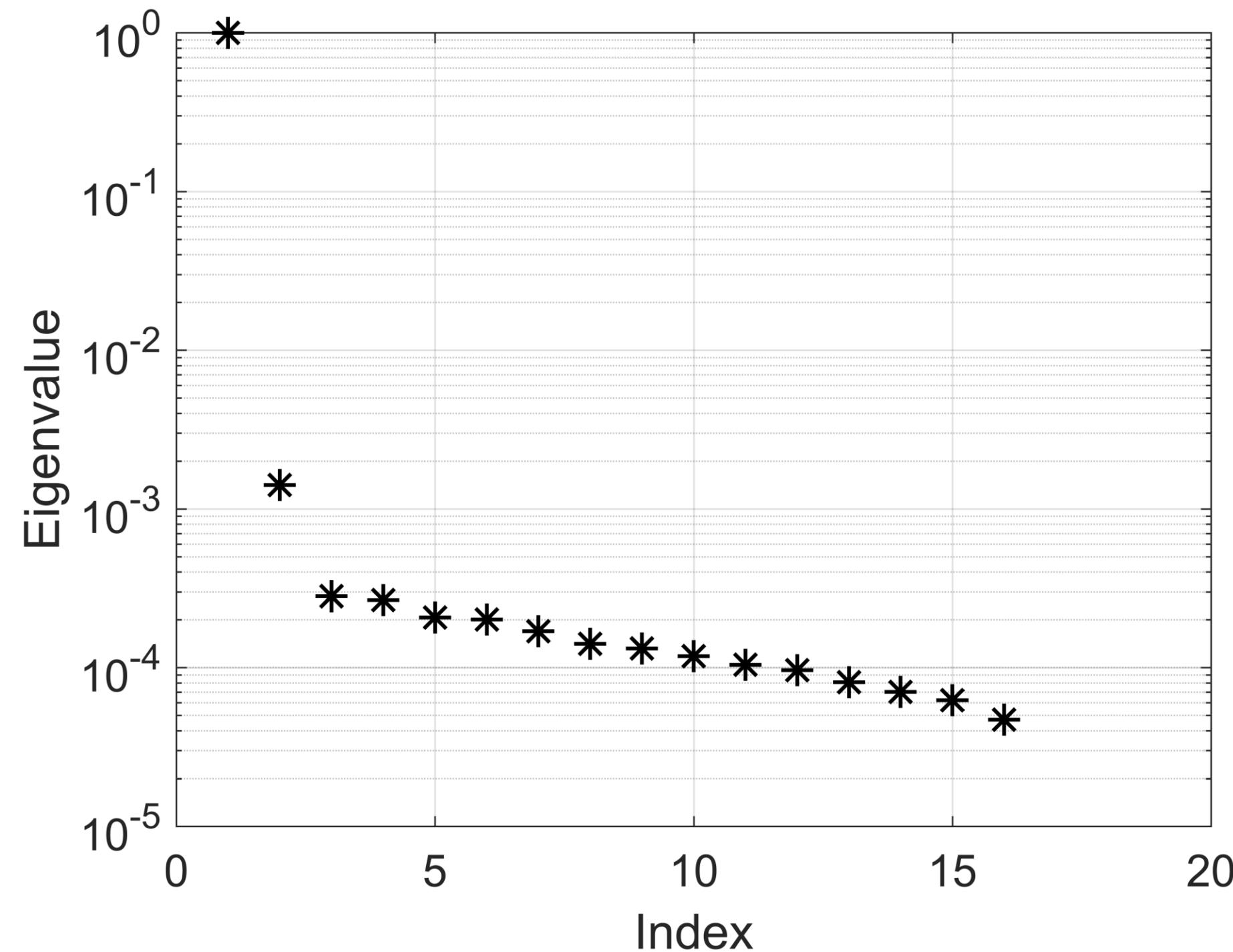
  $C = E[(\nabla_x f)(\nabla_x f)^T] =$   
 $(m \times m) \quad (m \times 1)(1 \times m)$

Covariance matrix of  
gradient vector

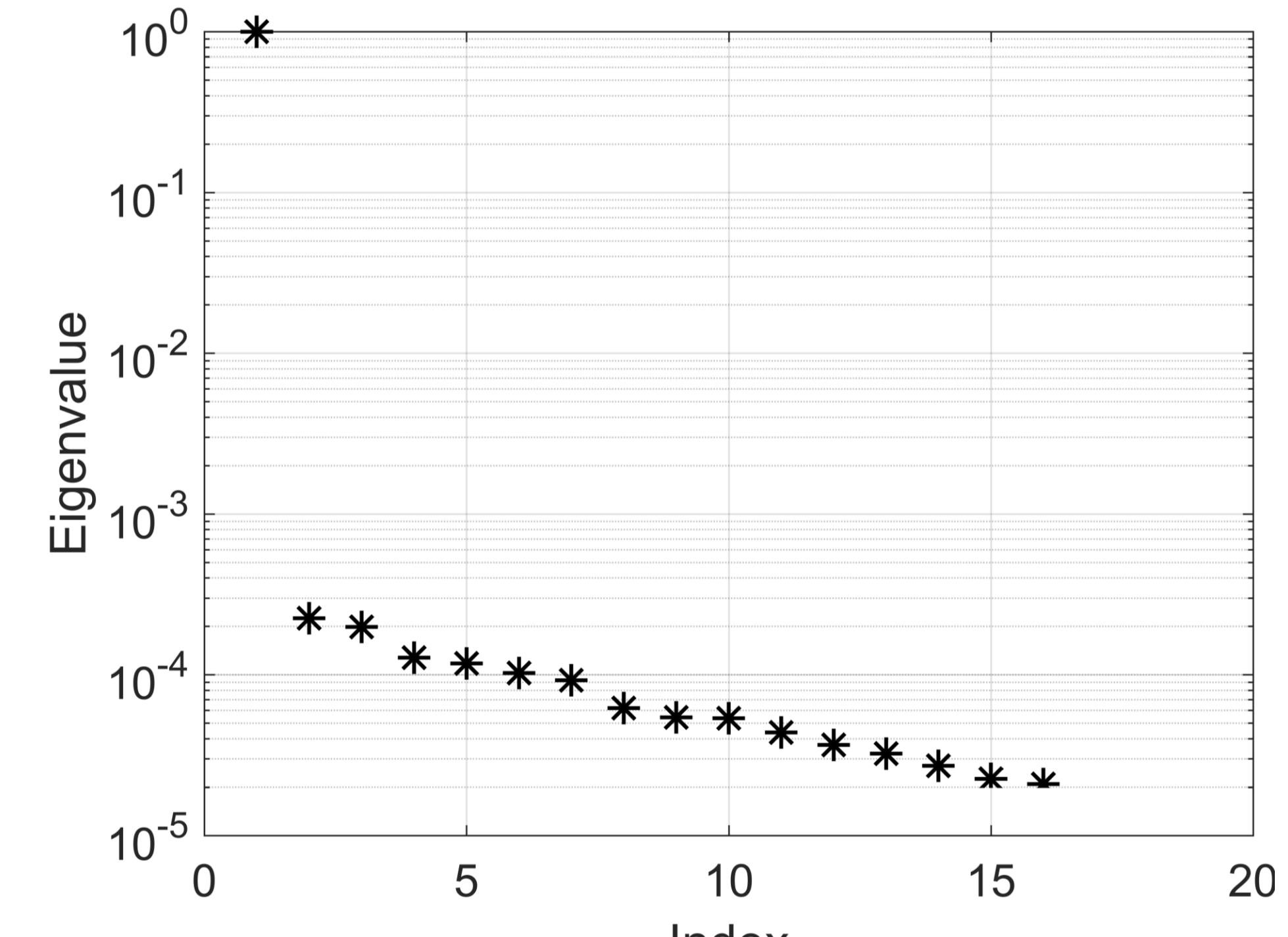
$$W \begin{bmatrix} \text{---} \\ \text{---} \\ \text{---} \end{bmatrix} \begin{bmatrix} \lambda_1 \\ \lambda_2 \\ \vdots \\ \lambda_m \end{bmatrix} \begin{bmatrix} \text{---} \\ \text{---} \\ \text{---} \end{bmatrix} W^T$$

The diagram illustrates the decomposition of the covariance matrix  $C$  into a product of matrices  $W$  and  $W^T$ . The matrix  $W$  is represented by three vertical ellipses, each representing a principal component. The matrix  $W^T$  is also represented by three vertical ellipses, corresponding to the columns of  $W$ . The eigenvalues  $\lambda_1, \lambda_2, \dots, \lambda_m$  are shown as labels next to the ellipses.

# Active Subspace Identification

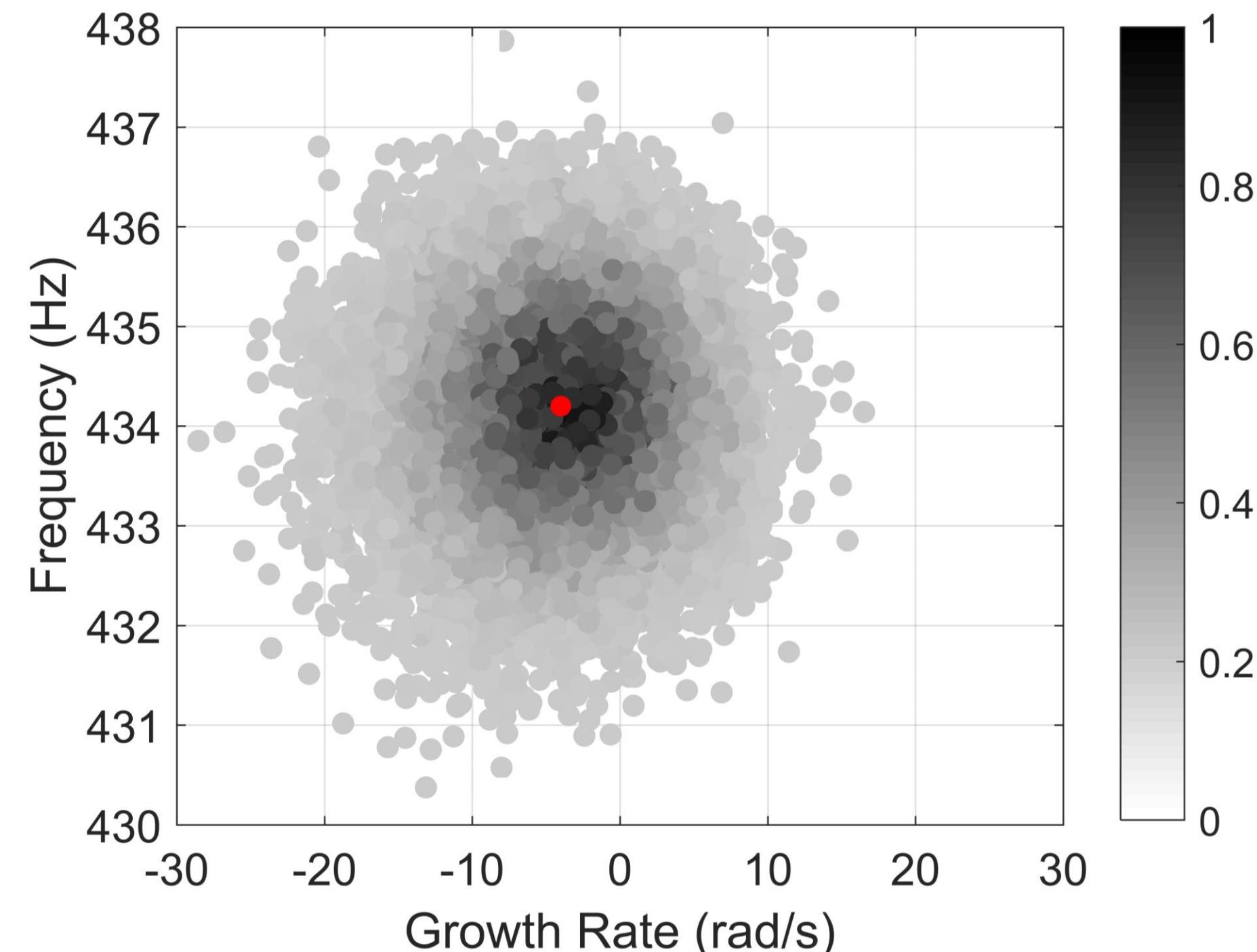


Case A

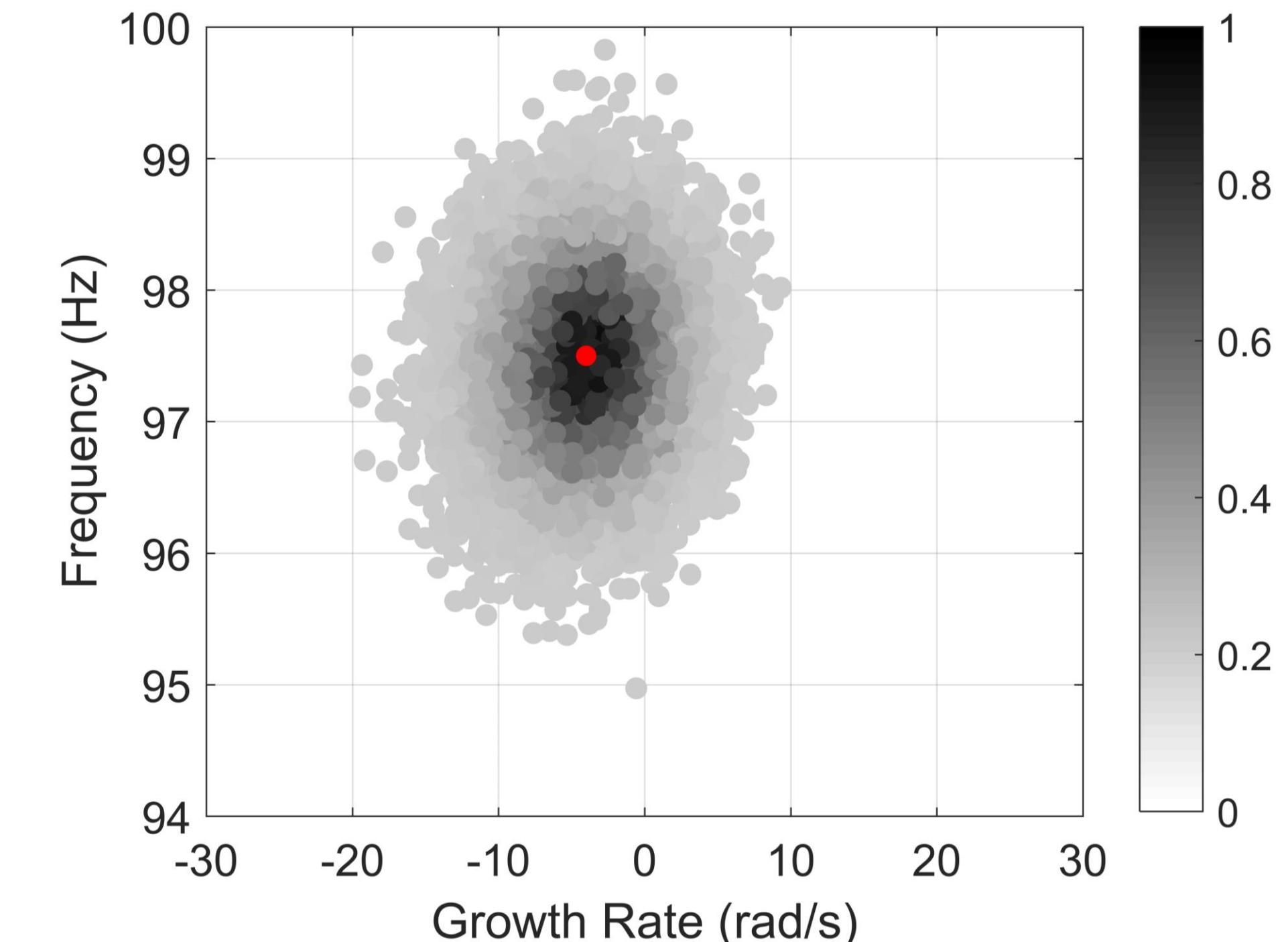


Case B

# Monte Carlo results for case A & B

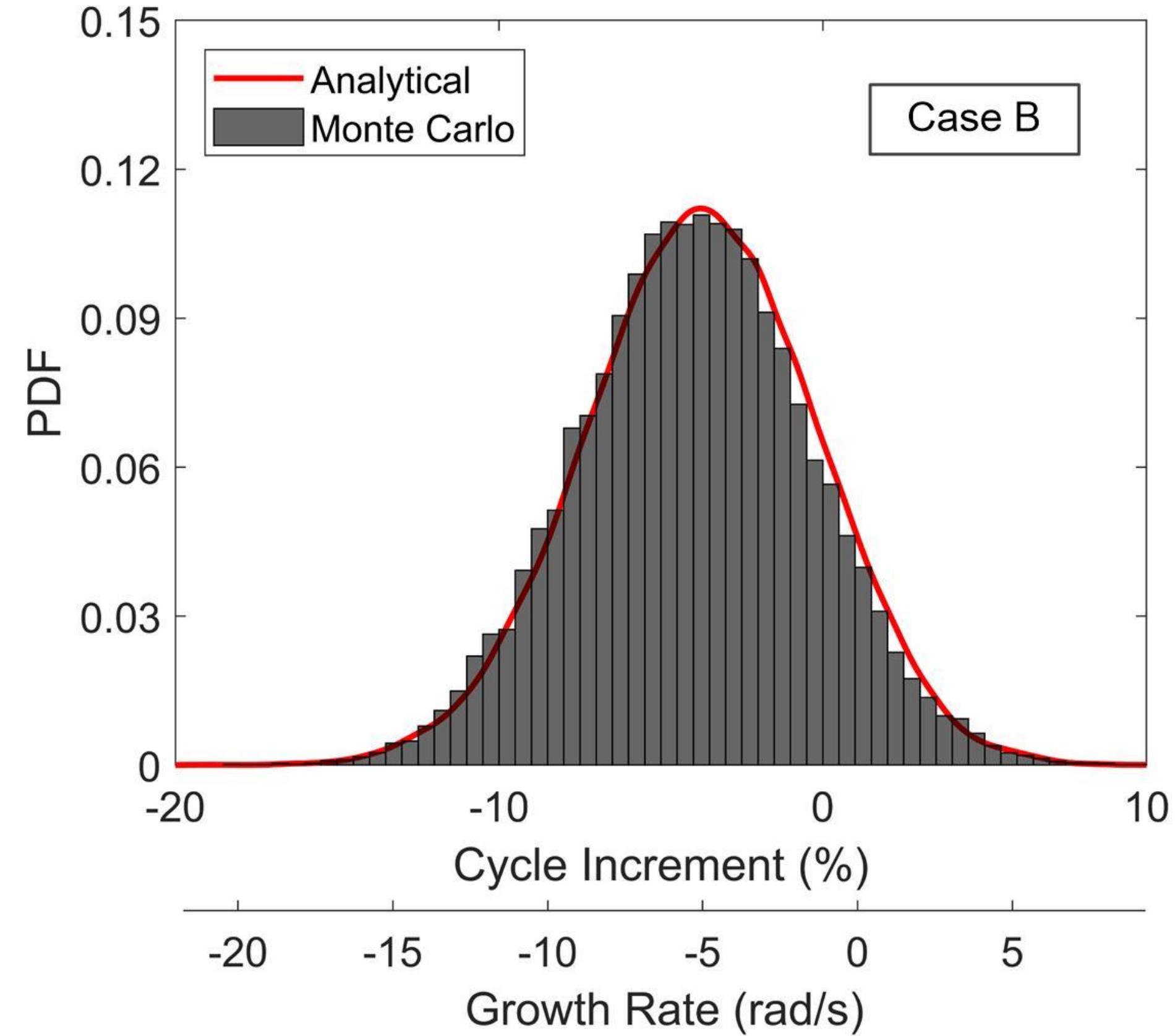
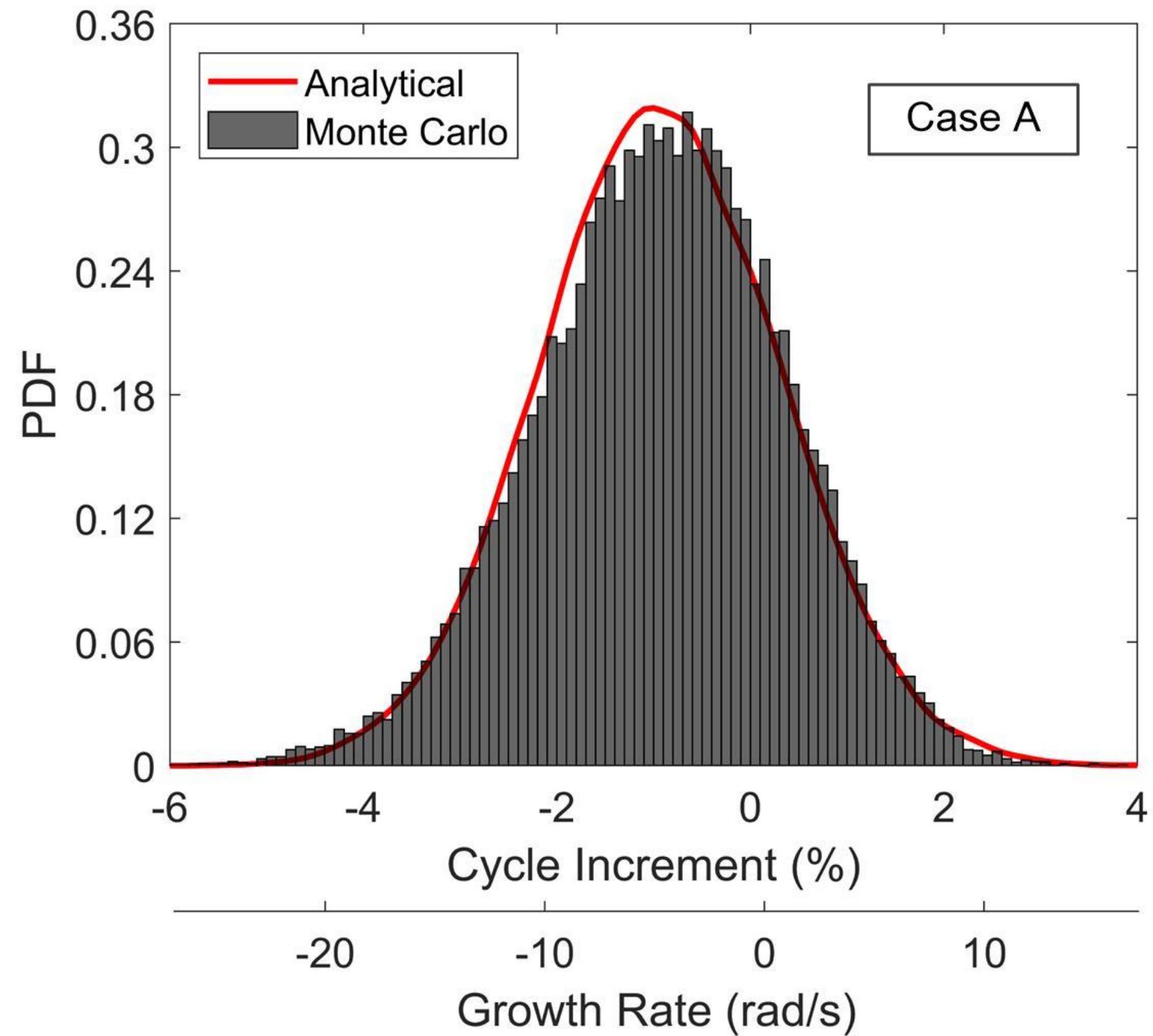


Case A

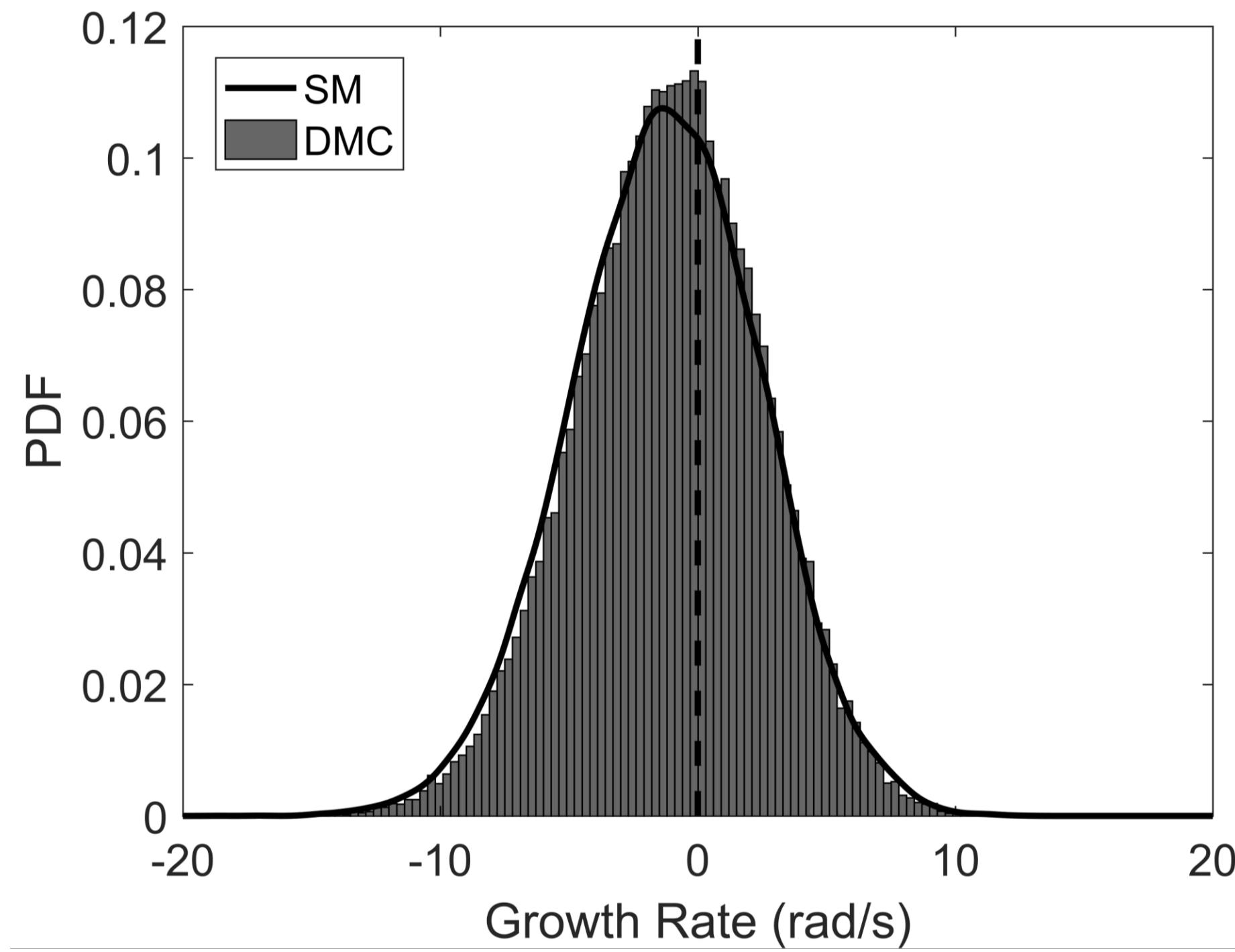


Case B

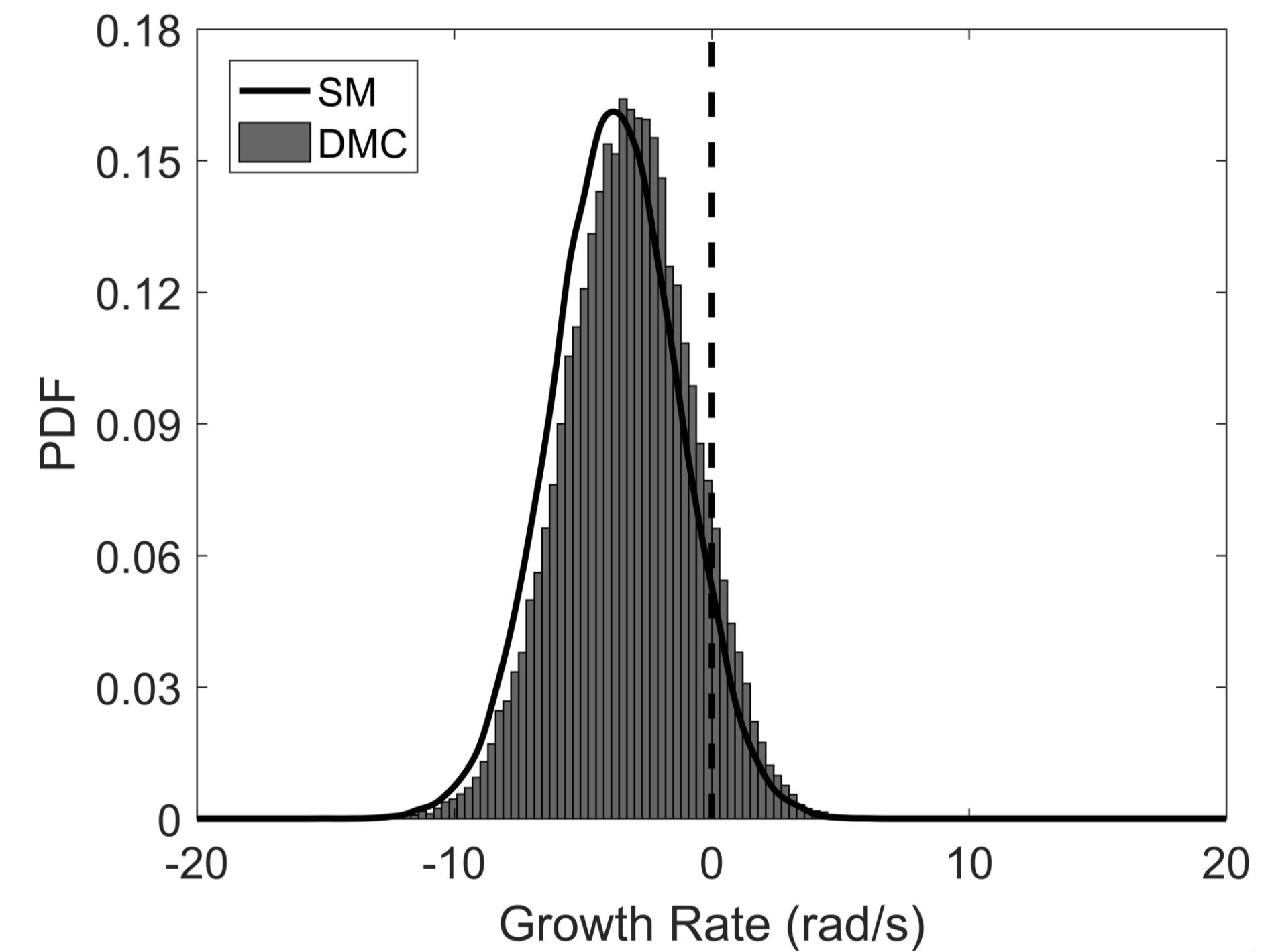
# PDF of modal growth rate in terms of cycle increment



# Procedure validation



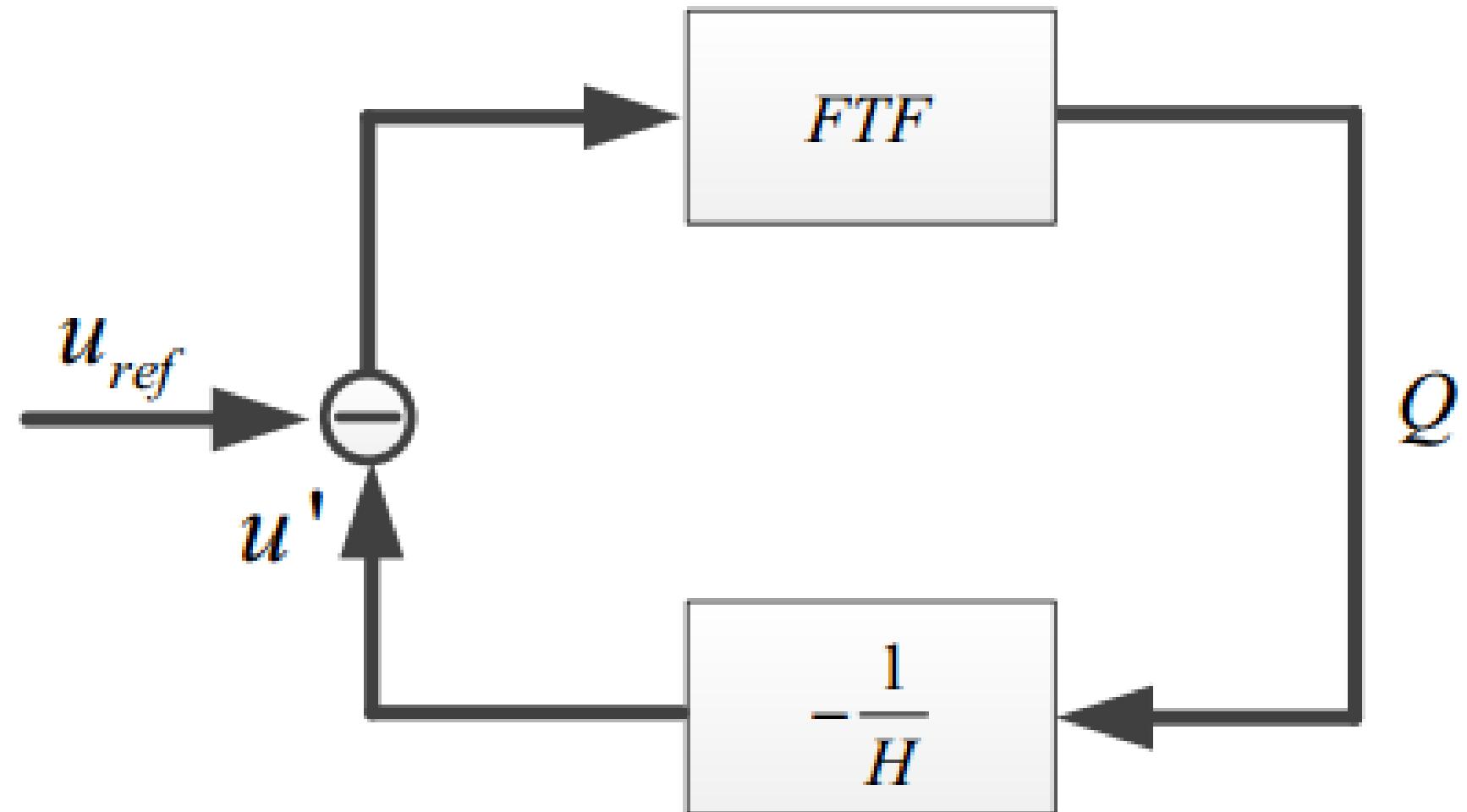
500 ms



1400 ms

## Discussion on the single active variable

1. This active variable approximates to first order the causal relationship between variations of the FIR model coefficients and variations of the model growth rate



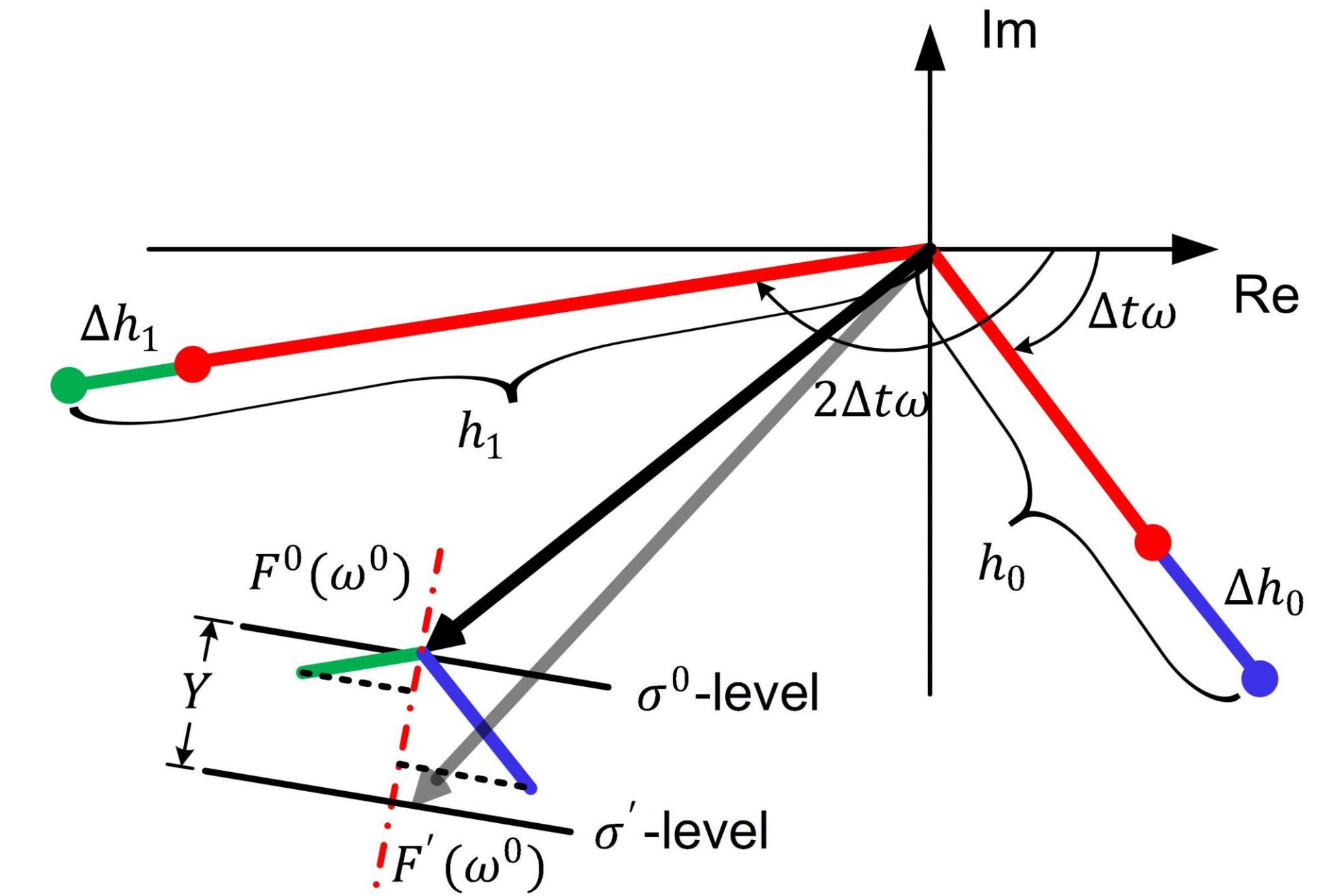
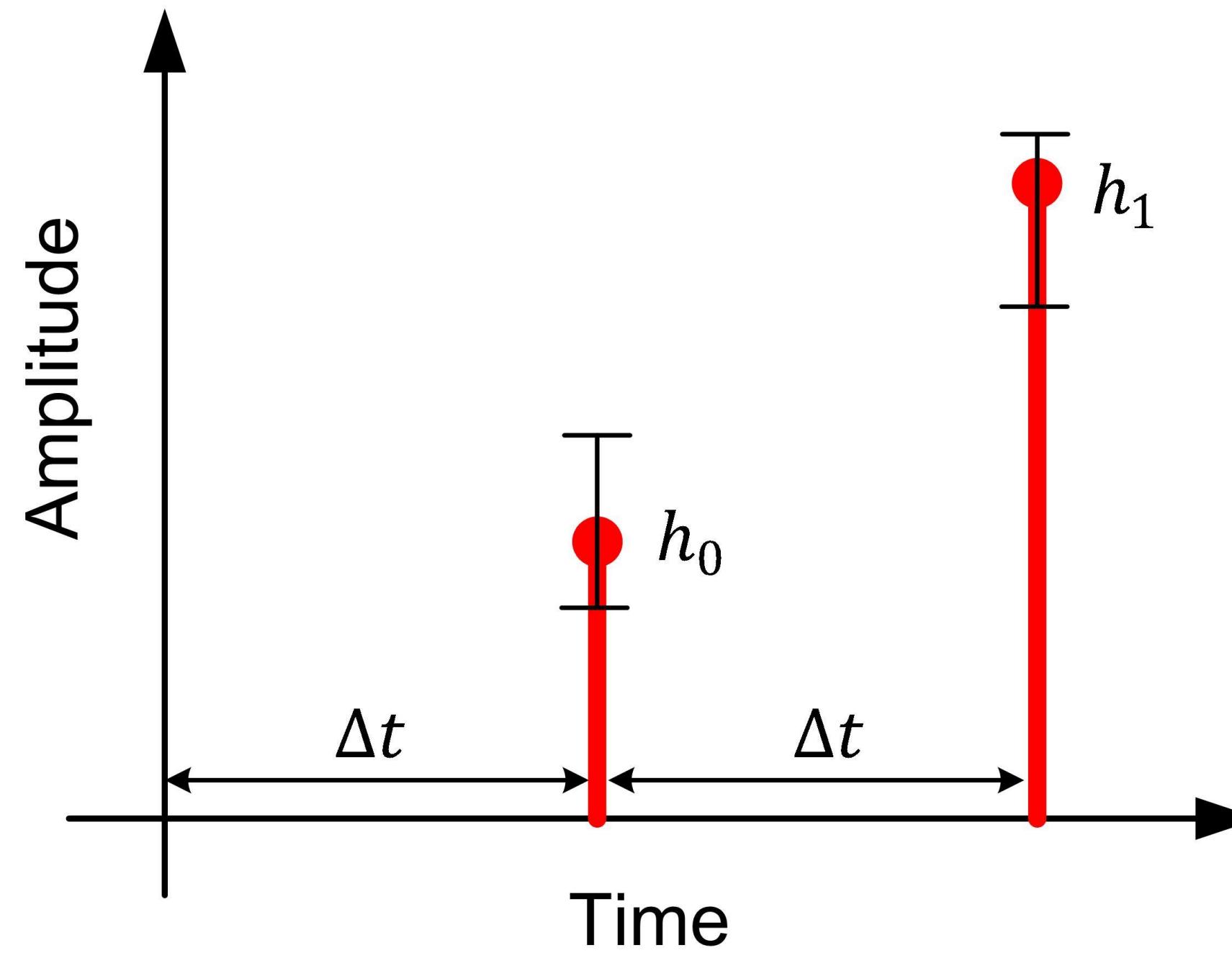
$$FTF(\omega - i\sigma) = H(\omega - i\sigma) \quad \omega, \sigma \in \mathbb{R}$$

2. It is possible to analytically derive the linear combination coefficients in active variable expression

## Discussion on the single active variable

3. This first-order approximation fails when uncertainties in impulse response model is too large, which is not allowed in real application of flame impulse response model

# Discussion on the single active variable (phasor plot demo)



$$F(\omega) = \sum_{k=0}^{L-1} h_k e^{-i(k+1)\Delta t \omega}, \quad \omega \in \mathbb{R} \quad \Rightarrow \quad F(\omega^0) = h_0 e^{-i\Delta t \omega^0} + h_1 e^{-i(2\Delta t)\omega^0}$$