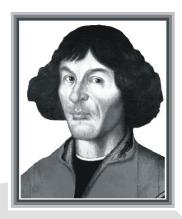
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NEET SYLLABUS

Kepler's laws of planetary motion. The universal law of gravitation. Acceleration due to gravity and its variation with altitude and depth. Gravitational potential energy; gravitational potential. Escape velocity, orbital velocity of a satellite. Geostationary satellites.

Nicolaus Copernicus

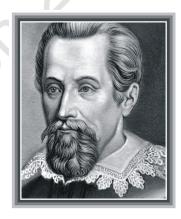
Nicolaus Copernicus was born on February 19, 1473 in Torun, Poland. Circa 1508, Copernicus developed his own celestial model of a heliocentric planetary system. After moving to the Frombork Cathedral Chapter in the early 1500s. Copernicus further developed his heliocentric model, and went on to design and apply a complex mathematical system for proving his theory. In 1513, his dedication prompted him to build his own modest observatory so that he could view the planets in action at any given time. Copernicus's observations did, at times, lead him to form inaccurate conclusions, including his



assumption that planet's orbit occurred in perfect circles. As German astronomer Johannes Kepler would later prove in the 17th century, planetary orbits are actually elliptical in shape.

Johannes Kepler (1571-1630)

Johannes Kepler (1571–1630) was a scientist of German origin. He formulated the three laws of planetary motion based on the painstaking observations of Tycho Brahe and coworkers. Kepler himself was an assistant to Brahe and it took him sixteen long years to arrive at the three planetary laws. He is also known as the founder of geometrical optics, being the first to describe what happens to light after it enters a telescope.



GRAVITATION

1. **GRAVITATION**

The discovery of the law of gravitation

The way the law of universal gravitation was discovered is often considered as the paradigm of modern scientific technique. The major steps involved were:

- The hypothesis about planetary motion given by Nicolaus Copernicus (1473–1543).
- The careful experimental measurements of the positions of the planets and the Sun by Tycho Brahe (1546-1601).
- Analysis of the data and the formulation of empirical laws by Johannes Kepler (1571-1630).
- The development of a general theory by Isaac Newton (1642–1727).

1.1 Newton's law of Gravitation

It states that every particle in the universe attracts all other particle with a force which is directly proportional to the product of their masses and is inversely proportional to the square of the distance between them.

If
$$|\vec{F}_{12}| = |\vec{F}_{21}| = F$$
, then

$$F \propto m_1 m_2$$
 and $F \propto \frac{1}{r^2}$

$$_{SO}~F \propto \frac{m_1 m_2}{r^2}$$





Note: This formula is applicable only for spherically symmetric masses or point masses.

1.2 Vector form of Newton's law of Gravitation:

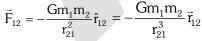
Let
$$\vec{r}_{12}$$
 = Position vector of m_1 w.r.t. m_2 = $\vec{r}_1 - \vec{r}_2$

$$\vec{r}_{21}$$
 = Position vector of m_2 w.r.t. m_1 = $\vec{r}_2 - \vec{r}_1$

$$\vec{F}_{21}$$
 = Gravitational force exerted on m_2 by m_1

 \vec{F}_{12} = Gravitational force exerted on m_1 by m_2

$$\vec{F}_{12} = -\frac{Gm_1m_2}{r_{21}^2}\hat{r}_{12} = -\frac{Gm_1m_2}{r_{21}^3}\vec{r}_{12}$$



Negative sign shows that:

- (i) The direction of \vec{F}_{12} is opposite to that of \hat{r}_{12}
- (ii) The gravitational force is attractive in nature

$$\mbox{Similarly} \quad \vec{F}_{21} = -\frac{Gm_1m_2}{r_{12}^2}\,\hat{r}_{21} \quad \mbox{or} \qquad \vec{F}_{21} = -\frac{Gm_1m_2}{r_{12}^3}\,\vec{r}_{21} \qquad \Rightarrow \ \vec{F}_{12} = -\vec{F}_{21} \;, \quad \vec{r}_{12} = -\vec{r}_{21} \;, \quad \vec{r}_{12} = -\vec{r}_{21} \;, \quad \vec{r}_{21} = -\vec{r}_{22} \;, \quad \vec{r}_{22} = -\vec{r}_{23} \;, \quad \vec{r}_{23} = -\vec{r}_{23} \;, \quad \vec{r}_{24} = -\vec{r}_{24} \;, \quad \vec{r}_{24} = -$$

The gravitational force between two bodies are equal in magnitude and opposite in directions.

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1.3 Universal Gravitational Constant "G"

- Universal Gravitational constant is a scalar quantity.
- Value of G : SI : $G = 6.67 \times 10^{-11} \text{ N-m}^2/\text{kg}^2$;

CGS: $G = 6.67 \times 10^{-8} \text{ dyne-cm}^2/g^2$ **Dimensions**: $[M^{-1}L^3T^{-2}]$

- Its value is same throughout the universe; G does not depend on the nature and size of the bodies; it does not depend even upon the nature of the medium between the bodies.
- Its value was first found out by the scientist "Henry Cavendish" with the help of "Torsion Balance" experiment.

GOLDEN KEY POINTS

- Gravitational force is always attractive.
- Gravitational forces are developed in the form of action and reaction pair. Hence they obey Newton's third law of motion.
- It is independent of the nature of medium between two masses.
- Gravitational forces are central forces as they act along the line joining the centre of gravity of the two bodies.
- Gravitational forces are conservative forces so work done by gravitational force does not depend on path.
- If any particle moves along a closed path under the action of gravitational force then the work done by this force is always zero for round the trip.
- Gravitational force is weaker than the electromagnetic and nuclear forces.
- Force developed between any two masses is called gravitational force and force between Earth and any body is called force of gravity.
- The total gravitational force on a particle due to a number number of particles is the resultant of the forces of attraction exerted on the given particle due to the individual particles i.e. $\vec{F} = \vec{F}_1 + \vec{F}_2 + \vec{F}_3 + \dots$ It means the principle of superposition is valid.
- Gravitational force holds good over a wide range of distances. It is found true from interplanetary distances to interatomic distances.
- It is a two body interaction i.e. gravitational force between the two particles is independent of the presence or absence of other bodies or particles.
- A uniform spherical shell of matter attracts a particle that is outside the shell as if all its mass were concentrated at its centre.

Illustrations

Illustration 1.

Two spherical balls of mass 10 kg each are placed 100 m apart. Find the gravitational force of attraction between them.

Solution

$$F = \frac{Gm_1m_2}{r^2} = \frac{6.67 \times 10^{-11} \times 10 \times 10}{(100)^2} = 6.67 \times 10^{-13} \, N \; . \label{eq:F}$$

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Illustration 2.

Two particles of masses 1 kg and 2 kg are placed at a separation of 50 cm. Assuming that the only forces acting on the particles are their mutual gravitation, find the initial acceleration of the heavier particle.

Solution

Force exerted by one particle on another is $F = \frac{Gm_1m_2}{r^2} = \frac{6.67 \times 10^{-11} \times 1 \times 2}{(0.5)^2} = 5.34 \times 10^{-10} \text{ N}$

Acceleration of heavier particle =
$$\frac{F}{m_2} = \frac{5.3 \times 10^{-10}}{2} = 2.67 \times 10^{-10} \, \text{m/s}^2 \, .$$

Note: This example shows that gravitational force is quite weak but this is the only force keep binds our solar system and also the universe comprising of all galaxies and other interstellar system.

Illustration 3.

Two stationary particles of masses M_1 and M_2 are 'd' distance apart. A third particle lying on the line joining the particles, experiences no resultant gravitational force. What is the distance of this particle from M_1 ?

Solution

Let m be the mass of the third particle

Force on m towards
$$M_1$$
 is $F_1 = \frac{GM_1m}{r^2}$



Force on m towards
$$M_2$$
 is $F_2 = \frac{GM_2m}{(d-r)^2}$

Since net force on m is zero \therefore $F_1 = F_2$

$$\Rightarrow \ \frac{GM_1m}{r^2} = \frac{GM_2m}{\left(d-r\right)^2} \ \ \, \Rightarrow \ \ \, \left(\frac{d-r}{r}\right)^2 = \frac{M_2}{M_1} \ \ \, \Rightarrow \ \ \, \frac{d}{r} - 1 = \frac{\sqrt{M_2}}{\sqrt{M_1}} \ \ \, \Rightarrow \ \ \, r \ \ \, = \ \, d \left[\frac{\sqrt{M_1}}{\sqrt{M_1} + \sqrt{M_2}}\right] = \frac{M_2}{\sqrt{M_1}} \ \ \, \Rightarrow \ \, r = \ \, d \left[\frac{\sqrt{M_1}}{\sqrt{M_1} + \sqrt{M_2}}\right] = \frac{M_2}{\sqrt{M_1}} \ \ \, \Rightarrow \ \, r = \frac{M_2}{\sqrt{M_1}} \ \ \, \Rightarrow \ \, r = \frac{M_2}{\sqrt{M_1}} \ \ \, \Rightarrow \ \, r = \frac{M_2}{\sqrt{M_1}} = \frac{M_2}{\sqrt{M_$$

Illustration 4.

Three masses, each equal to M are placed at the three corners of a square of side a. Calculate the force of attraction on unit mass placed at the fourth corner.

Solution

Force on m = 1 due to masses at corners 1 and 3 are $\vec{F_1}$ and $\vec{F_3}$ with $F_1 = F_3 = \frac{GM}{a^2}$

resultant of $\vec{F_1}$ and $\vec{F_3}$ is $\vec{F_r} = \sqrt{2} \frac{GM}{a^2}$ and its direction is along the diagonal

i.e. toward corner 2

Force on m due to mass M at 2 is $F_2 = \frac{GM}{(\sqrt{2}a)^2} = \frac{GM}{2a^2}$; F_r and F_z act in the same direction.

Resultant of these two is the net force:

$$F_{\text{net}} = \frac{\sqrt{2}GM}{a^2} + \frac{GM}{2a^2} = \frac{GM}{a^2} \bigg[\sqrt{2} + \frac{1}{2} \bigg]; \text{ it is directed along the diagonal as shown in the figure.}$$

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Illustration 5.

Two particles each of equal mass (m) move along a circle of radius (r) under the action of their mutual gravitational attraction. Find the speed of each particle.



Solution

For the circular motion of each particle,
$$\frac{mv^2}{r} = \frac{Gmm}{(2r)^2} \Rightarrow v^2 = \frac{Gm}{4r} \Rightarrow v = \frac{1}{2}\sqrt{\frac{Gm}{r}}$$
.

Illustration 6.

Three particles, each of mass m, are situated at the vertices of an equilateral triangle of side 'a'. The only forces acting on the particles are their mutual gravitational forces. It is intended that each particle moves along a circle while maintaining their original separation 'a'. Determine the initial velocity that should be given to each particle and the time period of the circular motion.

Solution

The resultant force on particle at A due to other two particles is

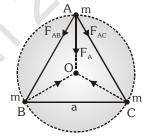
$$F_{_{\!A}} = \sqrt{F_{_{\!AB}}^2 + F_{_{\!AC}}^2 + 2F_{_{\!AB}}F_{_{\!AC}}\cos 60^\circ} = \sqrt{3}\,\frac{Gm^2}{a^2} \qquad ... \text{(i)} \qquad \left[\because F_{_{\!AB}} = F_{_{\!AC}} = \frac{Gm^2}{a^2} \right]$$

Radius of the circle $r = \frac{a}{\sqrt{3}}$

If each particle is given a tangential velocity v, so that the resultant force acts as the centripetal force,

then
$$\frac{mv^2}{r} = \sqrt{3} \frac{mv^2}{a}$$
 ...(ii)

From (i) and (ii) ,
$$\sqrt{3} \frac{mv^2}{a} = \frac{Gm^2\sqrt{3}}{a^2} \implies v = \sqrt{\frac{Gm}{a}}$$

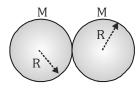


$$T = \frac{2\pi r}{v} = \frac{2\pi a}{\sqrt{3}} \sqrt{\frac{a}{Gm}} = 2\pi \sqrt{\frac{a^3}{3Gm}} \ . \label{eq:T}$$

Illustration 7.

Two solid spheres of same size of a certain metal are placed in contact with each other.

Prove that the gravitational force acting between them is directly proportional to the fourth power of their radius.



Solution

The weights of the spheres may be assumed to be concentrated at their centres.

So
$$F = \frac{G\left[\frac{4}{3}\pi R^{3}\rho\right] \times \left[\frac{4}{3}\pi R^{3}\rho\right]}{(2R)^{2}} = \frac{4}{9}(G\pi^{2}\rho^{2})R^{4}$$

$$F \propto R^4$$

Illustration 8.

A mass (M) is split into two parts (m) and (M-m), which are then separated by a certain distance. What ratio

 $\frac{m}{M}$ will maximise the gravitational force between them ?

Solution

If r is the distance between m and (M - m), the gravitational force will be $F = G \frac{m(M-m)}{r^2} = \frac{G}{r^2} (mM - m^2)$

For F to be maximum $\frac{dF}{dm} = 0$ and $\frac{d^2F}{dm^2} < 0$ as M and r are constants, i.e. $\frac{d}{dm} \left[\frac{G}{r^2} (mM - m^2) \right] = 0$

$$\Rightarrow \qquad \frac{G}{r^2}(M-2m)=0 \qquad \text{i.e. } M\,-\,2m\,=\,0 \qquad \qquad \left[\because \frac{G}{r^2}\neq 0\right]$$

or $\frac{m}{M} = \frac{1}{2}$, i.e., the force will be maximum when the two parts are identical.

BEGINNER'S BOX-1

- 1. Four identical point masses, each equal to M are placed at the four corners of a square of side a. Calculate the force of attraction on another point mass m₁ kept at the centre of the square.
- **2.** Three identical particles each of mass m are placed at the three corners of an equilateral triangle of side "a". Find the gravitational force exerted on one body due to the other two.
- Three identical point masses, each of mass 1 kg lie in the x-y plane at points (0,0)(0,0.2m) and (0.2m,0) respectively. The gravitational force on the mass at the origin is :-

(A) 1.67 x
$$10^{-11}$$
 ($\hat{i} + \hat{j}$) N

(B)
$$3.34 \times 10^{-10} (\hat{i} + \hat{j}) N$$

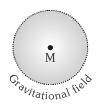
(C)
$$1.67 \times 10^{-9} (\hat{i} + \hat{j}) \text{ N}$$

(D)
$$3.34 \times 10^{-10} (\hat{i} - \hat{j}) N$$

2. GRAVITATIONAL FIELD AND IT'S INTENSITY

2.1 Gravitational Field

The gravitational field is the space around a mass or an assembly of masses within which it can exert gravitational forces on other masses.



Theoretically speaking, the gravitational field extends up to infinity. However, in actual practice, the gravitational field may become too weak to be measured beyond a particular distance.

2.2 Gravitational Field Intensity (\vec{l})

The gravitational field intensity at a point within a gravitational field is defined as the gravitational force exerted on unit mass placed at that point. $\rightarrow m$

$$\vec{I} = \frac{\vec{F}}{m}$$

Gravitational field intensity is a vector quantity whose direction is same as that of the gravitational force. Its **SI** unit is 'N/kg'.

Dimensions of intensity =
$$\frac{[F]}{[m]} = \frac{[M^1L^1T^{-2}]}{[M^1]} = [M^0L^1T^{-2}].$$



2.3 Gravitational Field Intensity Due to a Particle (Point - Mass) :

$$M \stackrel{\overrightarrow{r}}{\longrightarrow} m = 1$$
 unit

Gravitational field intensity = gravitational force exerted on unit mass

$$\Rightarrow \qquad \vec{I} = \frac{GM}{r^2}(-\hat{r}) = \frac{GM}{r^3}(-\vec{r}) \qquad \qquad |\vec{I} = \frac{GM}{r^2}(-\hat{r})|$$

$$\vec{I} = \frac{GM}{r^2}(-\hat{r})$$

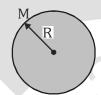
where 'M' is the mass of that particle due to which intensity is to be found.

2.4 Gravitational field intensity due to spherical mass distribution

If the observation point is located on the surface or outside the surface then the spherical mass can be taken as a particle which is situated at the centre of the sphere. i e. point mass.

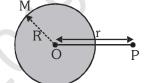
(I) For solid sphere

Let 'M' be the mass of sphere, 'R' the radius of sphere and 'r' the distance of the point under consideration from the centre of sphere.



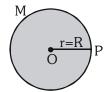
Case I: When r > R, i.e. outside the sphere then $|\vec{l}_{out}| = \frac{GM}{r^2}$

$$\vec{I}_{out} = \frac{GM}{r^2}(-\hat{r})$$



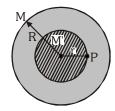
Case II: When r = R, i.e. at the surface then

$$\vec{I}_{\text{surface}} = \frac{GM}{R^2} (-\hat{r})$$



Case III: When r < R, i.e. inside the sphere then

$$\vec{I}_{in} = \frac{GM'}{r^2}(-\hat{r}) \qquad (1)$$



We know,

$$\frac{M^{\,\prime}}{M} = \frac{V^{\,\prime} \! \times \! \rho}{V \! \times \! \rho} = \frac{\frac{4}{3} \, \pi r^3 \times \! \rho}{\frac{4}{3} \, \pi R^3 \times \! \rho} = \frac{r^3}{R^3} \Longrightarrow M^{\,\prime} = \frac{M r^3}{R^3}$$

Putting the expression for M' in eq. (1), we get

$$\vec{I}_{in} = \frac{GMr^3}{r^2R^3}(-\hat{r})$$

$$\vec{I}_{in} = \frac{GMr}{R^3}(-\hat{r})$$

Important conclusions:-

$$I_{out} = \frac{GM}{r^2}$$

$$\therefore I_{\text{out}} \propto \frac{1}{r^2}$$

$$I_{sur} = \frac{GM}{R^2}$$

$$(3) \qquad I_{in} = \frac{GMr}{R^3}$$

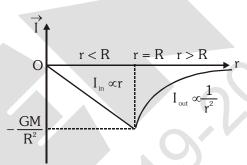
$$I_{in} \propto r$$

(4) So,
$$I_{\text{max}} = I_{\text{sur}} = \frac{GM}{R^2}$$

$$I_{centre} = \frac{GM(0)}{R^3} = 0$$

$$I_{\min} = I_{\text{centre}} = 0$$

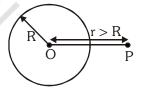
 \Rightarrow Graph between ' \vec{I} ' and 'r' for a solid sphere :



(II) Gravitational field intensity due to a Spherical Shell

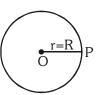
Case I : If r > R, the point is outside the shell then $|\vec{I}_{out}| =$

$$\vec{l}_{out} = \frac{GM}{r^2}(-\hat{r})$$



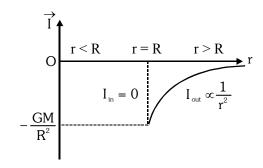
Case II: If r = R, the point is on the surface then

$$\vec{I}_{surface} = \frac{GM}{R^2}(-\hat{r})$$



Case III: If r < R, the point is inside the shell then I = 0

 $\Rightarrow \quad \vec{l} \ \, \text{v/s} \ \, \text{r graph for hollow sphere} :$



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3. ACCELERATION DUE TO GRAVITY

3.1 Gravity

In Newton's law of gravitation, the force of attraction between any two bodies is gravitation. If one of the bodies is Earth then the gravitation is called 'gravity'. Hence, gravity is the force by which Earth attracts a body towards its centre. It is a special case of gravitation.

3.2 Acceleration due to gravity near Earth's surface

Let us assume that Earth is a uniform sphere of mass M and radius R. The magnitude of the gravitational force of Earth on a particle of mass m, located outside the Earth at a distance r from its is centre, is $F = \frac{GMm}{r^2}$

Now according to Newton's second law $F = ma_q$

Therefore
$$a_g = \frac{GM}{r^2}$$
...(i)

At the surface of Earth, acceleration due to gravity $g = \frac{GM}{R^2} = 9.8 \text{ m/s}^2$

However any g value measured at a given location will differ from the g value calculated according to equation due to any three reasons :-

- (i) Earth's mass is not distributed uniformly.
- (ii) Earth is not a perfect sphere and
- (iii) Earth rotates.

3.3 Variation in Acceleration due to gravity

• Due to Altitude (height):

From diagram

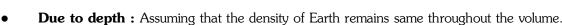
$$\frac{g_{h}}{g} = \frac{R_{e}^{2}}{(R_{e} + h)^{2}} = \frac{R_{e}^{2}}{R_{e}^{2} \left[1 + \frac{h}{R_{e}}\right]^{2}} = \left(1 + \frac{h}{R_{e}}\right)^{-1}$$



$$g_h = g \left[1 - \frac{2h}{R_e} \right]$$

Note:

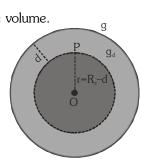
- (i) This formula is valid if h is upto 5% of earth's radius. (320 km from earth's surface)
- (ii) If h is greater than 5% of the earth's radius we use $g_h = \frac{GM_e}{(R_e + h)^2}$

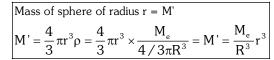


At Earth's surface :
$$g = \frac{4}{3}\pi GR_e \rho$$
 ...(i)

At a depth d inside the Earth:

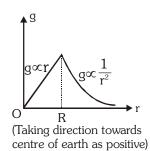
For point P only mass of the inner sphere is effective $g_d = \frac{GM'}{r^2}$





$$g_{\rm d} = \frac{G}{r^2} \times \frac{M_{\rm e} r^3}{R_{\rm e}^3} = \frac{GM_{\rm e}}{R_{\rm e}^2} \times \frac{r}{R_{\rm e}} = \frac{GM_{\rm e}}{R_{\rm e}^2} \times \frac{R_{\rm e} - d}{R_{\rm e}}$$

$$g_d = g \left[1 - \frac{d}{R_e} \right]$$
 valid for any depth



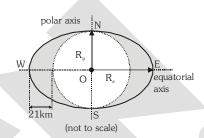
• Due to shape of the Earth:

From the diagram

$$R_{_{p}} < R_{_{e}} (R_{_{e}} = R_{_{p}} + 21 \text{ km}) \quad g_{_{p}} = \frac{GM_{_{e}}}{R_{_{p}}^{2}} \ \& \ g_{_{e}} = \ \frac{GM_{_{e}}}{(R_{_{p}} + 21000)^{2}}$$

$$\therefore g_e < g_p$$

By putting the values $g_{_{\rm p}}$ – $g_{_{\rm e}}$ = 0.02 m/s 2



• Due to Rotation of the Earth:

Net force on particle at P $mg' = mg - mr\omega^2 cos\lambda$

$$g' = g - rω^2 cosλ$$
 from, ΔΟΜΡ $r = R_e cosλ$

where $\lambda = Latitude$

Substituting for r, we have $g' = g - R_e \omega^2 \cos^2 \lambda$

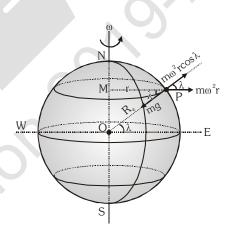
• At the equator $(\lambda = 0^\circ)$:

$$g_{eq} = g - \omega^2 R \cos^2(0^\circ)$$

$$g_{min}$$
 or $g_{eq} = g - \omega^2 R$

• At the poles ($\lambda = 90^\circ$):

$$g_{pole} = g - \omega^2 R\cos^2(90^\circ) = g - \omega^2 R(0) : g_{max} \text{ or } g_{pole} = g$$



It means that acceleration due to gravity at the poles does not depend upon the angular velocity or rotation of earth.

Condition of weightlessness on Earth's surface

If apparent weight of body is zero then angular speed of Earth can be calculated as $mg' = mg - mR_o \omega^2 \cos^2 \lambda$

$$0 = mg - mR_e \omega^2 \cos^2 \lambda \Rightarrow \omega = \frac{1}{\cos \lambda} \sqrt{\frac{g}{R_e}}$$

But at equator $\lambda = 0^{\circ}$ $\therefore \omega = \sqrt{\frac{g}{R_e}} = \frac{1}{800} \, \text{rad/s} = 0.00125 \, \text{rad/s} = 1.25 \times 10^{-3} \, \text{rad/s}.$

Note: If Earth will to rotate with 17 times of its present angular speed then bodies lying on equator would fly off into the space. Time period of Earth's rotation in this case would be 1.4 h.

GOLDEN KEY POINTS

• In terms of density
$$g = \frac{GM}{R^2} = \frac{G}{R^2} \times \frac{4}{3}\pi R^3 \times \rho$$
 \therefore $g = \frac{4}{3}\pi GR\rho$

If ρ is constant then $g \propto R$

• If M is constant then g
$$\propto \frac{1}{R^2}$$
; For % variation in 'g' upto 5%, $\frac{\Delta g}{g} = -2\frac{\Delta R}{R}$

• If mass (M) and radius (R) correspond to a planet and if small changes ΔM and ΔR occur in (M) and (R) respectively then

by
$$g = \frac{GM}{R^2} \Rightarrow \frac{\Delta g}{g} = \frac{\Delta M}{M} - 2 \frac{\Delta R}{R}$$

If R is constant then
$$\frac{\Delta g}{g} = \frac{\Delta M}{M}$$
; If M is constant then $\frac{\Delta g}{g} = -2\frac{\Delta R}{R}$

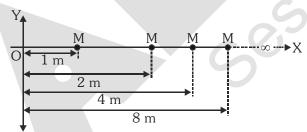
•
$$\Delta g_h = g - g_h = \text{decrement in } g \text{ with small height} = g - g \left[1 - \frac{2h}{R_e} \right] \implies \frac{\Delta g_h}{g} = \frac{2h}{R_e}$$

• If Earth stops rotating about its own axis, then the apparent weight of bodies or effective acceleration due to gravity will increase at all the places except poles.

- Illustrations —

Illustration 9.

Infinite particles each of mass 'M' are placed at positions x = 1 m, x = 2 m, x = 4 m ∞ . Find the gravitational field intensity at the origin.



Solution

$$\vec{l}_{net} = \vec{l}_1 + \vec{l}_2 + \vec{l}_3 + \vec{l}_4 + ... \infty$$
 terms

$$= \frac{GM}{(1)^2} \hat{i} + \frac{GM}{(2)^2} \hat{i} + \frac{GM\hat{i}}{(4)^2} + \dots \infty \ \text{terms} = GM\hat{i} \left(1 + \frac{1}{4} + \frac{1}{16} + \dots \infty \right) \ \left[\text{Here in the GP a=1 and } r = \frac{1}{4} \right]$$

So,
$$\vec{I}_{\text{net}} = GM\hat{i} \left[\frac{1}{1 - \frac{1}{4}} \right] = GM\hat{i} \left[\frac{1}{\left(\frac{3}{4} \right)} \right] \Rightarrow \vec{I}_{\text{net}} = \frac{4}{3}GM\hat{i}$$
.

Illustration 10.

At what depth below the Earth's surface the acceleration due to gravity is decreased by 1%?

Solution

$$\frac{\Delta g_{d}}{g} = \frac{d}{R_{\rm e}} \quad \Rightarrow \quad \frac{1}{100} = \frac{d}{6400} \quad \therefore \quad d = 64 \ \text{km}.$$

Illustration 11.

Which of the following statements are true about acceleration due to gravity?

- (A) 'g' decreases in moving away from the centre of earth if r > R
- (B) 'g' decreases in moving away from the centre of earth if r < R
- (C) 'g' is zero at the centre of earth
- (D) 'g' decreases if earth stops rotating on its axis

Solution

Variation of g with distance : If r > R then $g \propto \frac{1}{r^2}$.: (A) is correct

If r < R then $g \propto r$.. (B) is incorrect & (C) is correct

variation of g with ω : $g' = g - \omega^2 R \cos^2 \lambda$

If $\omega=0$ then g will not change at poles where $\cos\lambda=0$. while at other points g increases

: (D) is incorrect.

Illustration 12.

At what height above the Earth's surface the acceleration due to gravity will be 1/9 th of its value at the Earth's surface ? (Radius of Earth is 6400 km)

Solution

Acceleration due to gravity at height h is g' =
$$\frac{g}{\left(1 + \frac{h}{R}\right)^2} = \frac{g}{9} \Rightarrow \left(1 + \frac{h}{R_e}\right) = 3 \Rightarrow h = 2R_e = 12800 \text{ km}.$$

Illustration 13.

Determine the speed with which Earth would have to rotate about its axis so that a person on the equator weighs $\frac{3}{5}$ th of its present value. Write your answer in terms of g and R.

Solution

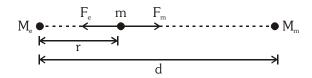
Weight on the equator
$$W' = \frac{3}{5}W \Rightarrow \frac{3}{5}\,mg = mg - m\omega^2R \qquad \Rightarrow \quad \omega = \sqrt{\frac{2g}{5R}}\;.$$

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Illustration 14.

Draw a rough sketch of the variation in weight of a spacecraft which moves from earth to moon.

Solution



Net weight =
$$F_e \sim F_m = \left| \frac{GM_e}{r^2} - \frac{GM_m}{(d-r)^2} \right| m$$

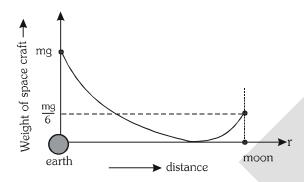


Illustration 15.

A solid sphere of uniform density and radius R exerts a gravitational force of attraction F_1 on a particle P, distant 2R from the centre of the sphere. A spherical cavity of radius R/2 is now formed in the sphere as shown in figure. The sphere with cavity now applies a gravitational force F_2 on the same particle P. Find the ratio F_2/F_1 .

Solution

$$F_1 = \frac{GMm}{4R^2}$$
, $F_2 =$ force due to whole sphere – force due to the sphere forming the cavity

$$= \frac{GMm}{4R^2} - \frac{GMm}{18R^2} \Rightarrow \frac{7GMm}{36R^2} \quad \because \frac{F_2}{F_1} = \frac{7}{9}$$

Illustration 16.

The maximum vertical distance through which an astronaut can jump on the earth is 0.5 m. Estimate the corresponding distance on the moon.

Solution

$$\because \text{ mgh = constant } \therefore \quad h \; \propto \; \frac{1}{g} \; \Rightarrow h_{_m} = \frac{h_{_e}g_{_e}}{g_{_m}} \; = \; \frac{0.5 \times g}{g/6} \; = \; 3 \; \text{m}.$$

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BEGINNER'S BOX-2

1. The magnitudes of the gravitational field at distances r_1 and r_2 from the centre of a uniform sphere of radius R and mass M are F_1 and F_2 respectively. then :-

(A)
$$\frac{F_1}{F_2} = \frac{r_1}{r_2}$$
 if $r_1 < R$ and $r_2 < R$

(B)
$$\frac{F_1}{F_2} = \frac{r_2^2}{r_1^2}$$
 if $r_1 > R$ and $r_2 > R$

(C)
$$\frac{F_1}{F_2} = \frac{r_1^3}{r_2^3}$$
 if $r_1 < R$ and $r_2 < R$

(D)
$$\frac{F_1}{F_2} = \frac{r_1^2}{r_2^2}$$
 if $r_1 < R$ and $r_2 < R$

- 2. A stone dropped from a height 'h' reaches the Earth's surface in 1 s. If the same stone is taken to Moon and dropped freely from the same height then it will reach the surface of the Moon in a time (The 'g' of Moon is 1/6 times that of Earth):—
 - (A) $\sqrt{6}$ seconds
- (B) 9 seconds
- (C) $\sqrt{3}$ seconds
- (D) 6 seconds
- 3. The radius of Earth is about 6400 km and that of Mars is 3200 km. The mass of Earth is 10 times that of Mars. An object weighs 200 N on the surface of Earth. Its weight on the surface of Mars will be :-
 - (A) 80 N
- (B) 40 N
- (C) 20 N
- (D) 8 N
- **4.** Weight of a body decreases by 1% when it is raised to a height h above the Earth's surface. If the body is taken to a depth h in a mine, then its weight will:
 - (A) decrease by 0.5%
- (B) decrease by 2%
- (C) increase by 0.5%
- (D) increase by 1%
- 5. At which height from the earth's surface does the acceleration due to gravity decrease by 1%?
- **6.** Find the percentage decrement in the weight of a body when taken to a height of 16 km above the surface of earth. (radius of earth is 6400 km)
- 7. What is the value of acceleration due to gravity at a height equal to half the radius of earth, from surface of earth? [take $g = 10 \text{ m/s}^2$ on earth's surface]
- **8.** At which height from the earth's surface does the acceleration due to gravity decrease by 75% of its value at earth's surface?
- 9. At which height above earth's surface is the value of 'g' same as in a 100 km deep mine?
- **10.** At what depth below the surface does the acceleration due to gravity becomes 70% of its value on the surface of earth?
- 11. At what depth from earth's surface does the acceleration due to gravity becomes $\frac{1}{4}$ times that of its value at surface ?
- 12. If earth is assumed to be a sphere of uniform density then plot a graph between acceleration due to gravity(g) and distance from the centre of earth.[AIPMT (Mains) 2006]

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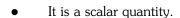
4. GRAVITATIONAL POTENTIAL ENERGY

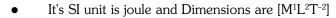
4.1 Gravitational Potential Energy (U)

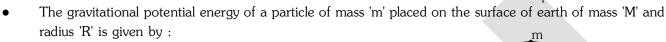
The gravitational potential energy of a particle situated at a point in some gravitational field is defined as the amount of work required to bring it from infinity to that point without changing its kinetic energy.

$$W = U = -\frac{GMm}{r} \text{ or } U = -\frac{Gm_1m_2}{r}$$

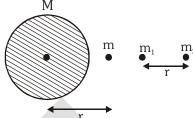
(Here negative sign shows the boundness of the two bodies)





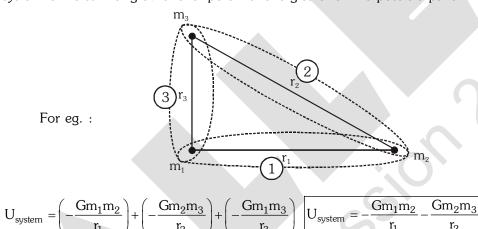


$$U = -\frac{GMm}{R}$$



4.2 Gravitational Potential Energy for three particle system

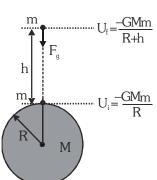
If there are more than two particles in a system, then the net gravitational potential energy of the whole system is the sum of gravitational potential energies of all the possible pairs in that system.



4.3 To find the change in potential energy of body or work done to raise a particle of mass 'm' to 'h' height above the surface of earth.

$$\begin{split} W &= \Delta U = U_f - U_i \\ \Rightarrow W &= -\frac{GMm}{R+h} - \left(-\frac{GMm}{R}\right) \\ \Rightarrow W &= GMm \left(\frac{1}{R} - \frac{1}{R+h}\right) \Rightarrow W = GMm \left(\frac{R+h-R}{R(R+h)}\right) \\ \Rightarrow W &= gR^2m \left(\frac{h}{R^2\left(1 + \frac{h}{R}\right)}\right) \left[\because GM = gR^2\right] \end{split}$$

$$W = \frac{mgh}{\left(1 + \frac{h}{R}\right)}$$



 Gm_1m_3

Special cases:

(i) If
$$h << R$$
, then $\frac{h}{R} \simeq 0$ \therefore $W \simeq \frac{mgh}{1+0} = mgh$

$$\mbox{(ii)} \qquad \mbox{If} \qquad \mbox{ h = R,} \qquad \mbox{ then } \quad \mbox{W} = \frac{mgR}{\left(1 + \frac{R}{R}\right)} = \frac{mgR}{2}.$$

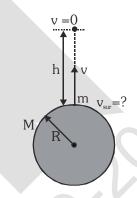
4.4 The velocity required to project a particle to a height 'h' from the surface of earth.

Applying 'COME' on the surface and at a height 'h'.

$$(K.E. + U)_{surface} = (K.E. + U)_{final}$$

$$\Rightarrow \frac{1}{2}mv^2 - \frac{GMm}{R} = 0 - \frac{GMm}{R+h}$$

$$\Rightarrow \frac{1}{2}mv^2 - \frac{GMm}{R} = -\left[\frac{GMm}{R+h}\right] \Rightarrow \frac{1}{2}mv^2 = \frac{mgh}{1+\frac{h}{R}}$$



$$\Rightarrow \qquad v^2 = \frac{2gh}{1 + \frac{h}{R}} \qquad \Rightarrow \qquad \boxed{v = \sqrt{\frac{2gh}{1 + \frac{h}{R}}}}$$

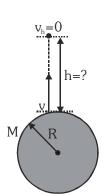
Note: If a body is released from a height 'h' above the surface of earth, then its velocity on reaching the earth's surface is also given by:

$$v = \sqrt{\frac{2gh}{1 + \frac{h}{R}}}$$

4.5 To find the maximum height attained by a body when it is projected with velocity 'v' from the surface of earth.

From
$$v^2 = \frac{2gh}{1 + \frac{h}{R}}$$

$$\Rightarrow v^2 + \frac{v^2h}{R} = 2gh$$



$$\Rightarrow \qquad v^2 = 2gh - \frac{v^2h}{R}$$

$$\Rightarrow \qquad v^2 = h \Bigg(2g - \frac{v^2}{R} \Bigg) \qquad \qquad \Rightarrow \qquad h = \frac{v^2}{2g - \frac{v^2}{R}} = \frac{v^2 R}{2g R - v^2} \,.$$

$$h = \frac{v^2 R}{2gR - v^2}$$

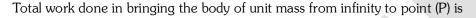
5. **GRAVITATIONAL POTENTIAL**

Gravitational field around a material body can be described not only by gravitational intensity \vec{l}_{σ} , but also by a scalar function, the gravitational potential V. Gravitational potential is the amount of work done by external agent in bringing a body of unit mass from infinity to that point without changing its kinetic energy. $V = \frac{W_{ext}}{m}$

Gravitational force on unit mass at (P) will be =
$$\frac{GM(1)}{x^2} = \frac{GM}{x^2}$$

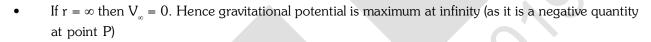
Work done by this force when the unit mass is displaced through the distance dx is

$$dW_{\text{ext}} = Fdx = \frac{GM}{x^2} \ . \ dx$$



$$W_{\text{ext}} = \int\limits_{\infty}^{r} \frac{GM}{x^2} \, dx \, = \, - \, \left(\frac{GM}{x} \right)_{\infty}^{r} \, = \, - \, \frac{GM}{r} \, \, . \label{eq:wext}$$

This work done is the measure of gravitational potential at point (P) $\therefore V_P = -$



• If
$$r = R_e$$
 (on the surface of Earth) $V_S = -\frac{GM_e}{R_e}$

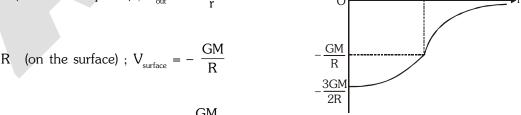
$$V = -\int \vec{I} \cdot \vec{dr} \Rightarrow dV = -\vec{I} \cdot \vec{dr} : I = -\frac{dV}{dr} = -ve$$
 potential gradient.

Gravitational Potential due to solid sphere and spherical shell:

Solid Sphere

r > R (outside the sphere); $V_{out} = -\frac{GM}{r}$ Case I





Case III
$$r < R$$
 (inside the sphere); $V_{in} = -\frac{GM}{2R^3} [3R^2 - r^2]$

It is clear that the potential V will be minimum at the centre (r = 0) but maximum in magnitude.

$$V_{centre} = -\frac{3}{2} \frac{GM}{R}$$
, $V_{centre} = \frac{3}{2} V_{surface}$

r < R

GM R

Spherical shell

Case I
$$r > R$$
 (outside the sphere); $V_{out} = -\frac{GM}{r}$

Case II
$$r = R$$
 (on the surface); $V_{surface} = -\frac{GM}{R}$

Potential is same every where and is equal to its value at the surface

$$V_{_{in}} = - \ \frac{GM}{R}$$

6. **ESCAPE VELOCITY & ESCAPE ENERGY**

6.1 Escape Velocity (v.)

It is the minimum velocity required for an object located at the planet's surface so that it just escapes the planet's gravitational field.

Consider a projectile of mass m, leaving the surface of a planet (or some other astronomical body or system), of radius R and mass M with escape speed v_a.

When the projectile just escapes to infinity, it has neither kinetic energy nor potential energy.

From conservation of mechanical energy
$$\frac{1}{2}mv_e^2 + \left(-\frac{GMm}{R}\right) = 0 + 0 \Rightarrow v_e = \sqrt{\frac{2GM}{R}}$$

The escape velocity of a body from a location which is at height 'h' above the surface of planet, we can use :-

$$v_{es} = \sqrt{\frac{2GM}{r}} = \sqrt{\frac{2GM}{R+h}}$$
 {: $r = R + h$ }

Where,

r = Distance from the centre of the planet.

h = Height above the surface of the planet.

Escape speed depends on:

- (i) Mass (M) and radius (R) of the planet
- (ii) Position from where the particle is projected.

Escape speed does not depend on:

- Mass (m) of the body which is projected
- Angle of projection. (ii)

If a body is thrown from the Earth's surface with escape speed, it goes out of earth's gravitational field and never returns back to the earth's surface.

6.2 Escape energy

Minimum energy given to a particle in the form of kinetic energy so that it can just escape the Earth's gravitational field.

Magnitude of escape energy = $\frac{GMm}{R}$ (-ve of PE on the Earth's surface)

Escape energy = Kinetic Energy corresponding to the escape velocity

$$\Rightarrow \frac{GMm}{R} = \frac{1}{2}mv_e^2$$

Note: In the above discussion it can be any planet for that matter

GOLDEN KEY POINTS

- Gravitational potential energy or potential is a -ve quantity whose maximum value is zero at infinite separation.
- Relation between Force & Gravitational potential energy is

$$F = -\frac{dU}{dr}$$

Above relation is valid only for all conservative forces.

- $\bullet \qquad v_{_{\rm e}} = \, \sqrt{\frac{2GM}{R}} \qquad \quad \mbox{If } M = \mbox{constant then} \qquad v_{_{\rm e}} \propto \, \frac{1}{\sqrt{R}}$
- $v_e = \sqrt{2gR}$ If g = constant then $v_e \propto \sqrt{R}$
- $v_e = R \sqrt{\frac{8\pi G \rho}{3}}$ If $\rho = constant$ then $v_e \propto R$
- Escape velocity does not depend on the mass of the body being projected, angle of projection or direction of projection.

$$v_e^{} \propto m^0$$
 and $v_e^{} \propto \theta^0$

- Escape velocity at : Earth's surface $v_e = 11.2 \text{ km/s}$, Moon surface $v_e = 2.31 \text{ km/s}$.
- Atmosphere on Moon is missing because root mean square velocity of gas particles is greater than escape velocity. i.e., $v_{rms} > v_e$
- Due to absence of atmosphere on moon, atmospheric pressure is zero. Hence, reading of a Barometer is also zero.
- If a hydrogen balloon is released from the surface of earth, then it moves upward because the upward buoyant force due to surrounding air exceeds its downwards weight. But if the balloon is released the surface of moon, then it will fall with g/6 acceleration under the influence of gravitational attraction of moon (upthrust is zero due to absence of atmosphere).
- If a bomb blast occurs on moon then its sound cannot be heard because sound is a mechanical wave which requires medium for propagation, which is absent there on moon.

Illustrations •

Illustration 17.

Three solid spheres of mass M and radius R are placed in contact as shown in figure. Find the potential energy of the system ?

Solution

$$\begin{aligned} \text{PE} &= \text{PE}_{12} + \text{PE}_{23} + \text{PE}_{31} \\ &= -\frac{\text{GM}^2}{2\text{R}} - \frac{\text{GM}^2}{2\text{R}} - \frac{\text{GM}^2}{2\text{R}} \ \ \, \Rightarrow \ \, \text{PE} = -\frac{3\text{GM}^2}{2\text{R}} \end{aligned}$$

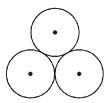


Illustration 18.

Four bodies each of mass m are placed at the different corners of a square of side a. Find the work done on the system to take any one body to infinity.

Solution

Initial potential energy of the system

$$PE_{i} = PE_{12} + PE_{23} + PE_{34} + PE_{41} + PE_{13} + PE_{24}$$

$$PE_{i} = -\frac{4GM^{2}}{a} - \frac{2GM^{2}}{a\sqrt{2}}$$



After taking any one body (say the mass placed at corner 4) to infinity only three bodies remain

:. Final potential energy the system is

$$PE_f = P_{12} + PE_{13} + PE_{23} = -\frac{2GM^2}{a} - \frac{GM^2}{a\sqrt{2}}$$

$$W_{ext.} = PE_f - PE_i = \left(-\frac{2GM^2}{a} - \frac{GM^2}{a\sqrt{2}} \right) - \left(-\frac{4GM^2}{a} - \frac{2GM^2}{a\sqrt{2}} \right) = \frac{2GM^2}{a} + \frac{GM^2}{a\sqrt{2}} \,.$$

Illustration 19.

A body of mass m is placed on the surface of earth. Find the work required to lift this body by a height

(i)
$$h = \frac{R_e}{1000}$$
 (ii) $h = R_e$ ($M_e = \text{mass of earth}$, $R_e = \text{radius of earth}$)

Solution:

(i)
$$h = \frac{R_e}{1000}$$
, as $h << R_e$, so

we can apply
$$W_{\text{ext}} = \text{mgh}$$
; $W_{\text{ext}} = \text{(m)} \left(\frac{GM_e}{R_e^2}\right) \left(\frac{R_e}{1000}\right) = \frac{GM_em}{1000R_e}$

(ii) $h = R_e$, in this case h is not very less than R_e , so we cannot apply $\Delta U = mgh$

$$W_{ext} = U_{f} - U_{i} = m(V_{f} - V_{i}) ; \quad W_{ext} = m \left[\left(-\frac{GM_{e}}{R_{e} + R_{e}} \right) - \left(-\frac{GM_{e}}{R_{e}} \right) \right] ; \quad W_{ext} = \frac{GM_{e}m}{2R_{e}}.$$

Illustration 20.

If velocity given to an object from the surface of the Earth is n times the escape velocity then what will be its residual velocity at infinity?

Solution

Let the residual velocity be v, then from energy conservation $\frac{1}{2} m(nv_e)^2 - \frac{GMm}{R} = \frac{1}{2} mv^2 + 0$

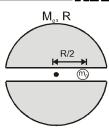
$$\Rightarrow v^{2} = n^{2}v_{e}^{2} - \frac{2GM}{R} = n^{2}v_{e}^{2} - v_{e}^{2} = (n^{2} - 1) v_{e}^{2} \Rightarrow v = (\sqrt{n^{2} - 1})v_{e}.$$

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Illustration 21.

A narrow tunnel is dug along the diameter of the earth, and a particle of mass m_0 is placed at $\frac{R}{2}$ distance from the centre. Find the escape speed of the particle from that place.



Solution

Suppose we project the particle with speed v_e , so that it just reaches infinity ($r \to \infty$).

Applying energy conservation principle

$$\begin{split} &K_{_{\rm I}} + U_{_{\rm I}} = K_{_{\rm f}} + U_{_{\rm f}} \\ &\frac{1}{2} m_{_{\rm 0}} v_{_{\rm e}}^2 + m_{_{\rm 0}} \Bigg[-\frac{G M_{_{\rm e}}}{2 R^3} \bigg\{ 3 R^2 - \bigg(\frac{R}{2} \bigg)^2 \bigg\} \Bigg] = 0 \\ \\ \Rightarrow &v_{_{\rm e}} = \sqrt{\frac{11 G M_{_{\rm e}}}{4 R}} \; . \end{split}$$

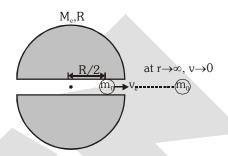


Illustration 22.

The escape velocity for a planet is v_e . A particle starts from rest at a large distance from the planet, reaches the planet only under gravitational attraction, and passes through a smooth tunnel through its centre. Its speed at the centre of the planet will be :-

(A)
$$\sqrt{1.5}v_e$$

(B)
$$\frac{v_e}{\sqrt{2}}$$

Solution (A)

From mechanical energy conservation, $0+0=\frac{1}{2}mv^2-\frac{3GMm}{2R} \Rightarrow v=\sqrt{\frac{3GM}{R}}=\sqrt{1.5}~v_e$.

Illustration 23.

A particle is projected vertically upwards from the surface of the earth (radius R_e) with a speed equal to one fourth of escape velocity. What is the maximum height attained by it?

(A)
$$\frac{16}{15}R_{e}$$

(B)
$$\frac{R_e}{15}$$

(C)
$$\frac{4}{15}$$
R_e

(D) None of these

Solution (B)

From conservation of mechanical energy, $\frac{1}{2} \, \text{mv}^2 = \frac{GMm}{R_e} - \frac{GMm}{R}$

Where R = maximum distance from centre of the earth. Also v = $\frac{1}{4}v_e = \frac{1}{4}\sqrt{\frac{2GM}{R_e}}$

$$\Rightarrow \frac{1}{2} m \times \frac{1}{16} \times \frac{2GM}{R_{_e}} = \frac{GMm}{R_{_e}} - \frac{GMm}{R} \Rightarrow R = \frac{16}{15} R_{_e} \Rightarrow h = R - R_{_e} = \frac{R_{_e}}{15} \; .$$



Illustration 24.

A mass of 6×10^{24} kg (= mass of earth) is to be compressed in a sphere in such a way that the escape velocity from its surface is 3×10^8 m/s (equal to the velocity of light). What should be the radius of the sphere?

- (A) 9 mm
- (B) 8 mm
- (C) 7 mm
- (D) 6 mm

Solution (B)

$$\text{As, } v_{_{e}} = \sqrt{\left(\frac{2GM}{R}\right)} \text{ , } R = \left(\frac{2GM}{v_{_{e}}^{2}}\right), \text{ } \therefore R = \frac{2 \times 6.67 \times 10^{-11} \times 6 \times 10^{24}}{\left(3 \times 10^{8}\right)^{2}} = 9 \times 10^{-3} \, \text{m} = 9 \, \text{mm} \, .$$

Illustration 25.

Gravitational potential difference between a point on the surface of a planet and point 10 m above is 4 J/kg. Considering the gravitational field to be uniform, how much work is done in moving a mass of 2 kg from the surface to a point 5 m above the surface?

- (A) 4 J
- (B) 5 J
- (C) 6 J
- (D) 7 J

Solution (A)

Gravitational field
$$g = -\frac{\Delta V}{\Delta x} = -\left(\frac{-4}{10}\right) = \frac{4}{10} J/kg - m$$

Work done in moving a mass of 2 kg from the surface to a point 5 m above the surface,

W = mgh =
$$(2 \text{ kg}) \left(\frac{4}{10} \frac{J}{\text{kg} - \text{m}} \right) (5 \text{ m}) = 4 \text{ J}$$

Illustration 26.

A body of mass m kg starts falling from a distance 2R above the earth's surface. What is its kinetic energy when it has fallen to a distance 'R' above the earth's surface ? (Where R is the radius of Earth)

Solution

By conservation of mechanical energy,

$$-\frac{GMm}{3R} + 0 = -\frac{GMm}{2R} + K.E. \implies K.E. = \frac{GMm}{R} \left[\frac{1}{2} - \frac{1}{3} \right] = \frac{1}{6} \frac{GMm}{R} = \frac{1}{6} \frac{(gR^2)m}{R} = \frac{1}{6} mgR.$$

Illustration 27.

With what velocity must a body be thrown from the earth's surface so that it may reach a height $4R_e$ above the Earth's surface? (Radius of the Earth $R_e = 6400$ km, g=9.8 m/s²).

Solution

By using conservation of mechanical energy $\frac{1}{2} m_0 v^2 - \frac{GMm_0}{R_e} = 0 - \frac{GMm_0}{(R_e + 4R_e)}$

$$\frac{1}{2} \ m_0 v^2 = - \ \frac{GMm_0}{5R_e} \ + \ \frac{GMm_0}{R_e} \ \Rightarrow \frac{1}{2} \ m_0 v^2 = \frac{4}{5} \frac{GMm_0}{R_e} \ \Rightarrow \ v^2 = \frac{8}{5} \frac{GM}{R_e} \ = \frac{8}{5} \frac{gR_e^2}{R_e}$$

$$v^2 = \frac{8}{5} \times 9.8 \times 6400 \times 10^3 = 10^8 \Rightarrow v = 10 \text{ km/s}.$$

BEGINNER'S BOX-3

- 1. The gravitational acceleration on the surface of earth is g. Find the increase in potential energy in lifting an object of mass m to a height equal to the radius of earth.
- 2. In a certain region of space gravitational field is given by I = -(K/r) (Where r is the distance from a fixed point and K is constant). Taking the reference point to be at $r = r_0$ with $V = V_0$. Find the potential at a distance r.
- 3. Two masses of 10^2 kg and 10^3 kg are separated by 1 m distance. Find the gravitational potential at the mid point of the line joining them.
- **4.** The magnitude of intensity of gravitational field at a point situated at a distance 8000 km from the centre of Earth is 6.0 N-kg. The magnitude of gravitational potential at that point in N-m/kg will be :-

(A) 6

(B) 4.8×10^7

(C) 8×10^5

(D) 4.8×10^2

5. The gravitational field due to a certain mass distribution is $E = \frac{K}{x^3}$ in the x-direction (K is a constant). Taking the gravitational potential to be zero at infinity, its value corresponding to distance x is :-

(A) $\frac{K}{x}$

(B) $\frac{K}{2x}$

(C) $\frac{K}{x^2}$

(D) $\frac{K}{2x^2}$

6. Two bodies of respective masses m and M are placed d distance apart. The gravitational potential (V) at the position where the gravitational field due to them is zero is :-

(A) $V = -\frac{G}{d}(m + M)$

(B) $V = -\frac{G}{d}$

(C) $V = -\frac{GM}{d}$

(D) $V = -\frac{G}{d}(\sqrt{m} + \sqrt{M})^2$

7. A body of mass m is situated at a distance $4R_e$ above the Earth's surface, where R_e is the radius of Earth. What minimum energy should be given to the body so that it may escape?

(A) mgR_e

(B) $2mgR_e$

(C) $\frac{\text{mgR}}{5}$

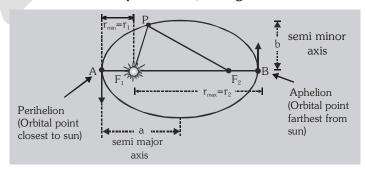
(D) $\frac{\text{mgR}}{16}$

7. KEPLER'S LAWS OF PLANETARY MOTION

Kepler found important regularities in the motion of the planets. These regularities are known as 'Kepler's three laws of planetary motion'.

(a) First Law (Law of Orbits):

All planets move around the Sun in elliptical orbits, having the Sun at one focus of the orbit.



When a particle moves with respect to two fixed points in such a way that the sum of the distances from these two points is always constant then the path of the particle is an ellipse and the two fixed points are called focal points.

According to Figure :-

$$PF_1 + PF_2 = AF_1 + AF_2 = BF_1 + BF_2 = constant$$

But in ellipse $AF_1 = BF_2$ (minimum distance from both focal is same)

$$PF_1 + PF_2 = BF_2 + AF_2 = BF_1 + AF_1 = 2a = length of major axis$$

$$r_1 + r_2 = r_{min} + r_{max} = 2a$$

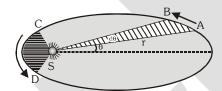
$$\therefore \quad a = \frac{r_1 + r_2}{2} = \frac{r_{min} + r_{max}}{2} \quad \text{(Mean distance)}$$

(b) Second Law (Law of Areas):

A line joining any planet to the Sun sweeps out equal areas in equal intervals of time, i.e., the areal speed of the planet remains constant.

According to the second law, if a planet moves from A to B in a given time interval, and from C to D in the same time interval, then the areas ASB and CSD will be equal.

dA = area of the curved triangle $SAB = \frac{1}{2}(AB \times SA) = \frac{1}{2}(r d\theta \times r) = \frac{1}{2}r^2 d\theta$



Thus, the instantaneous areal speed of the planet is $\frac{dA}{dt} = \frac{1}{2}r^2 \frac{d\theta}{dt} = \frac{1}{2}r^2\omega = \frac{1}{2}rv$...(i)

where ω is the angular speed of the planet.

Let L be the angular momentum of the planet about the Sun S and m the mass of the planet.

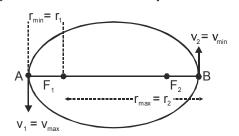
Then
$$L = I\omega = mr^2\omega = mvr$$
 ...(ii)

where I (=mr²) is the instantaneous moment of inertia of the planet about the Sun S.

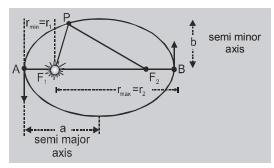
From eq. (i) and (ii),
$$\frac{dA}{dt} = \frac{L}{2m}$$
 ...(iii)

Now, the areal speed dA/dt of the planet is constant, according to Kepler's second law. Therefore, according to eq. (iii), the angular momentum L of the planet is also constant, that is, the angular momentum of the planet is conserved. Thus, Kepler's second law is equivalent to conservation of angular momentum.

Applying conservation of angular momentum between points A and B



A planet moves around the sun in an elliptical orbit of semi major axis a and eccentricity e



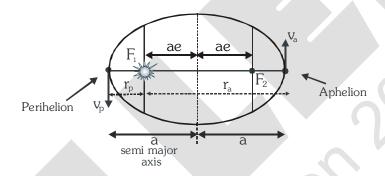
For an ellipse: its general equation is given as

$$\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$$

If a > b then a is semi major axis, b is semi minor axis and e is eccentricity where

$$b^2 = a^2 (1 - e^2)$$
 \Rightarrow $e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$

Applying the conservation of angular momentum (COAM) at the perihelion and aphelion $mv_{pp} = mv_{pp}$



$$r_{\text{max.}} = a(1 + e) \; ; \; r_{\text{min.}} = a(1 - e) \implies \left[\frac{v_p}{v_a} = \frac{v_{\text{max}}}{v_{\text{min}}} = \frac{1 + e}{1 - e} \right] \qquad(1)$$

By conservation of mechanical energy

$$\frac{1}{2} m v_p^2 - \frac{GMm}{r_p} = \frac{1}{2} m v_a^2 - \frac{GMm}{r_a} = -\frac{GMm}{2a} \qquad(2)$$

By solving eqⁿ (1) and (2),

$$\Rightarrow \qquad v_{_{a}} = \sqrt{\frac{GM}{a} \bigg(\frac{1\!-\!e}{1\!+\!e}\bigg)} \ ; \ v_{_{P}} = \sqrt{\frac{GM}{a} \bigg(\frac{1\!+\!e}{1\!-\!e}\bigg)}$$

(c) Third Law (Law of Periods): The square of the period of revolution of any planet around the Sun is directly proportional to the cube of the semi-major axis of its elliptical orbit.

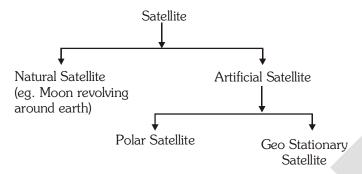
$$T^2 \propto a^3$$

Note: For a circular orbit semi major axis = Radius of the orbit

$$T^2 \propto R^3$$

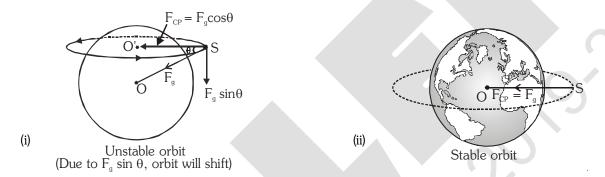
8. SATELLITE MOTION

A light body revolving round a heavier planet due to gravitational attraction, is called a satellite. Moon is a natural satellite of Earth.



8.1 Essential Conditions for Satellite Motion

- The centre of satellite's orbit should coincide with the centre of Earth.
- Plane of the orbit of satellite should pass through the centre of Earth.

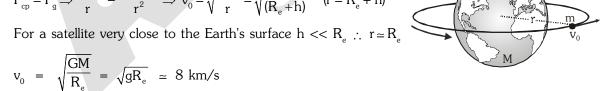


• It follows that a satellite can revolve round the earth only in those circular orbits whose centres coincide with the centre of earth. Circles drawn on globe with centres coincident with earth are known as 'great circles'. Therefore, a satellite revolves around the earth along circles concentric with great circles.

8.2 Orbital velocity (v_o)

A satellite of mass m moving in an orbit of radius r with speed v_0 . The required centripetal force is provided by gravitation.

$$F_{\rm cp} = F_{\rm g} \Rightarrow \frac{m v_0^2}{r} = \frac{GMm}{r^2} \Rightarrow v_0 = \sqrt{\frac{GM}{r}} = \sqrt{\frac{GM}{(R_{\rm e} + h)}} \quad \ (r = R_{\rm e} + h)$$



- If a body is taken to some height (small) from Earth and given a horizontal velocity of magnitude 8 km/s then it becomes a satellite of Earth.
- ullet $v_{_{\! o}}$ depends upon : Mass of planet, Radius of the circular orbit of satellite.
- If orbital velocity of a satellite becomes $\sqrt{2} \, v_o$ (or increased by 41.4%) or K.E. is doubled then it escapes from the gravitational field of Earth.



8.3 Time Period of a Satellite

$$T = \frac{2\pi r}{v_0} = \frac{2\pi r^{\frac{3}{2}}}{\sqrt{GM}} = \frac{2\pi r^{\frac{3}{2}}}{R\sqrt{g}} \Rightarrow T^2 = \frac{4\pi^2}{GM}r^3 \Rightarrow T^2 \propto r^3 \quad (r = R+h)$$

For a satellite close to Earth's surface $v_0 = \sqrt{\frac{GM_e}{R_e}} \simeq 8 \text{ km/s}$

$$T_0 = 2\pi \sqrt{\frac{R_e}{g}} = 84 \text{ minutes} = 1 \text{ hour } 24 \text{ minute} = 1.4 \text{ h} = 5063 \text{ s}$$

In terms of density
$$T_0 = \frac{2\pi (R_e)^{1/2}}{(G \times 4/3\pi R_e \times \rho)^{1/2}} = \sqrt{\frac{3\pi}{G\rho}}$$

Time period of a near by satellite only depends on the density of the planet.

For Moon

$$h_m = 380,000 \text{ km} \text{ and } T_m = 27 \text{ days}$$

$$v_{o'} \ = \ \frac{2\pi (R_{_e} + h)}{T_{_m}} = \frac{2\pi (386400 \times 10^3)}{27 \times 24 \times 60 \times 60} \simeq 1.04 \ km/s.$$

8.4 Energy of a satellite

Kinetic energy K.E. = $\frac{1}{2}$ mv₀² = $\frac{GMm}{2r}$ = $\frac{L^2}{2mr^2}$ (L = mrv₀ = m \sqrt{GMr})

Potential energy P.E. =
$$-\frac{GMm}{r} = -mv_0^2 = -\frac{L^2}{mr^2}$$

$$\label{eq:Total mechanical energy T.E.} \text{Total mechanical energy T.E.} = P.E. + K.E. = -\frac{mv_0^2}{2} = -\frac{GMm}{2r} = -\frac{L^2}{2mr^2} \; .$$

• Binding energy:

Total mechanical energy (potential + kinetic) of a closed system is negative. The modulus of this total mechanical energy is known as the binding energy of the system. This is the energy due to which system is bound or the different parts of the system are bonded to each other.

Binding energy of a satellite (system)

$$B.E. \ = \ - \ T.E. \\ B.E. \ = \ \frac{1}{2} m v_0^2 = \frac{GMm}{2r} = \frac{L^2}{2mr^2} \, . \\ Hence \ B.E. \ = \ K.E. \ = \ - \ T.E. \ = \frac{- \ P.E.}{2} \, .$$

Escape energy and ionisation energy are the practical examples of binding energy.

Work done in Changing the Orbit of a Satellite

W = Change in mechanical energy of the system but E = $\frac{-GMm}{2r}$

so
$$W = E_2 - E_1 = \frac{GMm}{2} \left(\frac{1}{r_1} - \frac{1}{r_2} \right)$$

In the given graph of energy v/s position

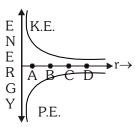
At point A:

$$|(P.E.)| > K.E.$$
 $|(P.E.)| > KE + PE = -ve$

At A, B & C System is bounded.

At point D:
$$\therefore$$
 | PE | = K.E. \therefore TE = KE + PE = 0

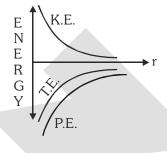
So, the system is unbounded.



Graphs:

$$KE = \frac{GMm}{2r}$$

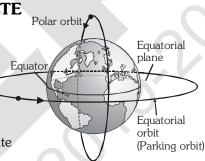
$$TE = -\frac{GMm}{2r}$$



9. GEO-STATIONARY SATELLITE & POLAR SATELLITE

Geo-Stationary Satellite

- It rotates in an equatorial plane.
- Its height from the Earth's surface is 36000 km. ($\approx 6R_o$)
- Its angular velocity and time period should be same as that of Earth.
- Its rotating sense should be same as that of Earth (West to East).
- Geo Stationary/Telecommunication/Parking/Synchronus/Satellite are always projected from equator (for example Singapore).
- Its orbit is called parking orbit and its orbital velocity is 3.1 km/s.



Polar Satellite

It is that satellite which revolves in the polar orbit around Earth. A polar orbit is one whose angle of inclination with the equatorial plane of Earth is 90° and a satellite in polar orbit will pass over both the north and south geographical poles once per revolution. Its time period is 100 min. and height is between 500 Km to 800 Km.

Polar satellites are employed to obtain the cloud images, atmospheric data, information regarding ozone layer in the atmosphere and it detected the ozone hole over Antarctica etc.

10. WEIGHTLESSNESS

When the apparent weight of a body becomes zero, the body is said to be in a state of weightlessness. In a satellite around the earth, every part and parcel of the satellite has an acceleration towards the centre of the earth which is exactly the value of earth's acceleration due to gravity at that position. Thus in the satellite everything inside it is in a state of free fall. If a body is in a satellite (which does not produce its own gravity) orbiting the Earth at a height h above its surface then

True weight =
$$mg_h = \frac{mGM}{(R+h)^2} = \frac{mg}{\left(1 + \frac{h}{R}\right)^2}$$
 and Apparent weight = $m(g_h - a)$

But
$$a = \frac{v_0^2}{r} = \frac{GM}{r^2} = \frac{GM}{(R+h)^2} = g_h \implies Apparent weight = m(g_h - g_h) = 0.$$

Note: Condition of weightlessness can be overcome by creating artificial gravity by rotating the satellite in addition to its revolution.

GOLDEN KEY POINTS

• The time period of the longest pendulum on the surface of earth is given by $T=2\pi\sqrt{\frac{R}{g}}=84.6$ minutes.

Note: The time period of a satellite orbiting close to the earth's surface is also 84.6 minutes.

- The angular velocity and time period of revolution of a G.S.S. is same as that of earth. It means that a G.S.S. completes its revolution around the earth once in 24 hours.
- Height of a G.S.S. (Geo-stationary satellite) from the surface of earth is about 36,000 km. Therefore, its distance from the centre of earth is about $R + H = 36,000 + 6400 = 42,400 \text{ km} \approx 7R_a$
- It is used as a communication satellite. It is also known as parking satellite, telecommunication satellite or synchronous satellite.
- One G.S.S. can cover nearly one-third surface area of earth. Therefore a minimum of three G.S.S. are required to cover the whole earth.
- Orbital velocity depends upon the mass of the central body and orbital radius (distance of satellite from the centre of the central body). If the distance of satellite increases, then the orbital velocity (v_0) decreases.
- Orbital velocity does not depend on the mass of satellite.
- If a body is taken to a small height and given a horizontal velocity of 8 km/s, it will start revolving around the earth in a circular orbit which means that it will become a satellite close to the earth's surface.
- If a body is released from a revolving satellite, then it will continue to move in the same orbit with the same orbital velocity which means that it will also become a satellite close to the earth.
- When the total energy of a satellite is negative, it will be moving in either a circular or an elliptical orbit.
- When the total energy of a satellite is zero, it will escape away from its orbit and its path becomes parabolic.
- If the gravitational force is inversely proportional to the n^{th} power of distance r, then the orbital velocity of a satellite $v_0 \propto r^{\frac{l-n}{2}}$ and time period $T \propto r^{\frac{n+1}{2}}$
- The total energy of any planet revolving around the sun is negative (: it is bounded).
- First Satellite of Earth is Sputnik I. First Geo-satellite of India is Aryabhatt I.

 First Geo-stationary satellite of India is Apple I. Example of other Satellites of India are: Bhaskar-I,
 Rohini-I, Bhaskar-II (Geo Satellite); Insat-I (A), Insat-I (B) (Geo Stationary Satellite)

Illustrations

Illustration 28.

Two satellites S_1 and S_2 are revolving round a planet in coplanar and concentric circular orbits of radii R_1 and R_2 in the same sense respectively. Their respective periods of revolution are 1 h and 8 h. The radius of the orbit of satellite S_1 is equal to 10^4 km. Find the relative speed in km/h when they are closest.

Solution

By Kepler's
$$3^{rd}$$
 law, $\frac{T^2}{R^3}$ = constant $\therefore \frac{T_1^2}{R_1^3} = \frac{T_2^2}{R_2^3}$ or $\frac{1}{(10^4)^3} = \frac{64}{R_2^3}$ or $R_2 = 4 \times 10^4$ km

Distance travelled in one revolution, $S_1 = 2\pi R_1 = 2\pi \times 10^4$ and $S_2 = 2\pi R_2 = 2\pi \times 4 \times 10^4$

$$v_1 = \frac{S_1}{t_1} = \frac{2\pi \times 10^4}{1} = 2\pi \times 10^4 \, \text{km/h} \ \text{and} \ v_2 = \frac{S_2}{t_2} = \frac{2\pi \times 4 \times 10^4}{8} = \pi \times 10^4 \, \text{km/h}$$

:. Relative velocity = $v_1 - v_2 = 2\pi \times 10^4 - \pi \times 10^4 = \pi \times 10^4 \text{ km/h}$

Illustration 29.

A space—ship is launched into a circular orbit close to the Earth's surface. What additional speed should now be imparted to the spaceship so that it overcomes the gravitational pull of the Earth.

Solution

Let ΔK be the additional kinetic energy imparted to the spaceship to overcome the gravitation pull then

$$\Delta K = - \text{ (total energy of spaceship)} = \frac{GMm}{2R}$$

$$Total \ kinetic \ energy = \frac{GMm}{2R} + \Delta K = \frac{GMm}{2R} + \frac{GMm}{2R} = \frac{GMm}{R} \,, \quad then \ \frac{1}{2} \, mv_2^2 = \frac{GMm}{R} \Rightarrow v_2 = \sqrt{\frac{2GMm}{R}} \,.$$

But
$$v_1 = \sqrt{\frac{GM}{R}}$$
. So additional velocity required = $v_2 - v_1 = \sqrt{\frac{2GM}{R}} - \sqrt{\frac{GM}{R}} = \left(\sqrt{2} - 1\right)\sqrt{\frac{GM}{R}}$

Alternate solution:

Additional velocity = Escape velocity - Orbital velocity

$$= v_{es} - v_{0}$$

$$= \sqrt{\frac{2GM}{R}} - \sqrt{\frac{GM}{R}}$$

$$= (\sqrt{2} - 1) \sqrt{\frac{GM}{R}}$$

Illustration 30.

An astronaut, inside an earth's satellite experiences weightlessness because:

(A) he is falling freely

- (B) no external force is acting on him
- (C) no reaction is exerted by the floor of the satellite
- (D) he is far away from the earth's surface

Solution (Ans. A, C)

As astronaut's acceleration = g; so he is falling freely. Also no reaction is exerted by the floor of the satellite.

Illustration 31.

If a satellite orbits as close to the earth's surface as possible

- (A) its speed is maximum
- (B) time period of its revolution is minimum
- (C) the total energy of the 'earth plus satellite' system is minimum
- (D) the total energy of the 'earth plus satellite' system is maximum

Solution (Ans. A, B, C)

For (A): orbital speed
$$v_0 = \sqrt{\frac{GM}{r}}$$
, $r_{min} = R$ so $v_0 = maximum$

For (B) : Time period of revolution $T^2 \propto r^3$

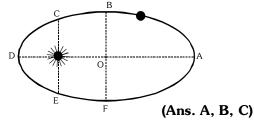
For (C/D) : Total energy =
$$-\frac{GMm}{2r}$$

ALLEN

Illustration 32.

A planet is revolving round the sun in an elliptical orbit as shown in figure. Select correct alternative(s)

- (A) Its total energy is negative at D.
- (B) Its angular momentum is constant
- (C) Net torque on the planet about sun is zero
- (D) Linear momentum of the planet is conserved



Solution

- **For (A)**: For a bound system, the total energy is always negative.
- For (B): For central force field, angular momentum is always conserved.
- **For (C)**: For central force field, torque = 0.
- **For (D)**: In presence of external force, linear momentum is not conserved.

Illustration 33.

An artificial satellite is moving in a circular orbit around the earth with a speed equal to half the magnitude of escape velocity from the earth.

- (a) Determine the height of the satellite above the earth's surface.
- (b) If the satellite is stopped suddenly in its orbit and allowed to fall freely on the earth, find the speed with which it hits the surface of earth. Given M = mass of earth and R = Radius of earth

Solution

(a) Let height above the earth's surface = h then

$$v_{\text{orbital}} = \sqrt{\frac{GM}{R+h}} = \frac{1}{2}v_e = \frac{1}{2}\sqrt{\frac{2GM}{R}} = \sqrt{\frac{GM}{2R}} \Rightarrow R+h = 2R \Rightarrow h = R$$

(b) If the satellite is stopped suddenly then it total energy $E_1 = -\frac{GMm}{2R}$

Let its speed be v when it hits the earth's surface then its total energy on earth surface

$$E_2 = -\frac{GMm}{R} + \frac{1}{2}mv^2$$

 $\text{conservation law for mechanical energy yields } E_1 = E_2 \Rightarrow \frac{-GMm}{2R} = -\frac{GMm}{R} + \frac{1}{2}mv^2 \Rightarrow v = \sqrt{\frac{GM}{R}}$

Illustration 34.

Is it possible to place an artificial satellite in an orbit such that it is always visible over Kota? Write down the reason.

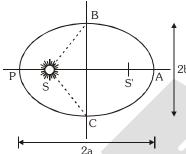
Solution

No, Kota is not in the equatorial plane.

BEGINNER'S BOX-4

- 1. The mean radius of the earth's orbit around the sun is 1.5×10^{11} m. The mean radius of the orbit of mercury around the sun is 6×10^{10} m. Calculate the year of the mercury.
- 2. If earth describes an orbit round the sun of double its present radius, what will be the year on earth?
- **3.** If the gravitational force were to vary inversely as mth power of the distance, then the time period of a planet in circular orbit of radius r around the Sun will be proportional to :-
 - (A) r^{-3m/2}
- (B) $r^{3m/2}$
- (C) $r^{m+1/2}$
- (D) $r^{(m+1)/2}$

- A planet is revolving around the Sun in an elliptical orbit. Its closest distance from the Sun is r_{min} . The farthest distance from the Sun is r_{max} . If the orbital angular velocity of the planet when it is nearest to the Sun is ω, then the orbital angular velocity at the point when it is at the farthest distance from the Sun is :-
- $\text{(A)} \ \left(\sqrt{\frac{r_{\text{min}}}{r_{\text{max}}}} \right) \omega \qquad \qquad \text{(B)} \ \left(\sqrt{\frac{r_{\text{max}}}{r_{\text{min}}}} \right) \omega \qquad \qquad \text{(C)} \ \left(\frac{r_{\text{max}}}{r_{\text{min}}} \right)^2 \omega \qquad \qquad \text{(D)} \ \left(\frac{r_{\text{min}}}{r_{\text{max}}} \right)^2 \omega$
- The time period of revolution of moon around the earth is 28 days and radius of its orbit is 4×10^5 km. **5**. If $G = 6.67 \times 10^{-11} \text{ Nm}^2/\text{kg}^2$ then find the mass of the earth.
- 6. Let the speed of the planet at the perihelion P in Fig. be v_p and the Sun-planet distance SP be r_p . Relate (r_p, v_p) to the corresponding quantities at the aphelion (r_A, v_A) . Will the planet take equal times to traverse BAC and CPB?



- 7. A comet orbits the sun in a highly elliptical orbit. Does the comet have a constant (a) linear speed, (b) angular speed, (c) angular momentum, (d) kinetic energy, (e) potential energy (f) total energy throughout its orbit? Neglect any mass loss of the comet when it comes very close to the Sun.
- 8. A satellite moves in a circular orbit around the earth. The radius of this orbit is one half that of the moon's orbit. Find the time in which the satellite completes one revolution.
- 9. A small satellite revolves round a planet in an orbit just above planet's surface. Taking the mean density of planet as p, calculate the time period of the satellite.
- Two satellites A and B, having ratio of masses 3: 1 are in circular orbits of radius r and 4r. Calculate the **10**. ratio of total mechanical energies of A to B.
- A satellite orbits the Earth at a height of 400 km above the surface. How much energy must be expended to rocket the satellite out of the Earth's gravitational influence? Mass of the satellite = 200 kg; mass of Earth = 6.0×10^{24} kg; radius of the Earth = 6.4×10^6 m; G = 6.67×10^{-11} N-m²/kg².
- **12**. An artificial satellite orbiting the earth in very thin atmosphere loses its energy gradually due to small but continuous dissipation against atmospheric resistance. Then explain why its speed increases progressively as it comes closer and closer to the earth.
- Write the answer of the following questions in one word -
 - What is the orbital speed of Geo-stationary satellite? (a)
 - (b) For a satellite moving in an orbit around the earth what is the ratio of kinetic energy to potential energy?
- An object weighs 10 N at the north pole of the Earth. In a geostationary satellite distant 7R from the centre of the Earth (of radius R), the true weight and the apparent weight are respectively:
 - (A) 0, 0
- (B) 0.2 N, 0
- (C) 0.2 N, 9.8 N
- (D) 0.2 N, 0.2 N

ANSWERS

BEGINNER'S BOX-1

- 1. Zero
- $2. \qquad \sqrt{3} \frac{\text{Gm}^2}{\text{a}^2}$
- **3.** (C)
- **BEGINNER'S BOX-2**
- **1.** A, B
- **2.** (A)
- **3.** (A)

- **4.** (A)
- **5.** 32 km

8.

6. 0.5%

- **7.** 4.44 m/s²
- h = R = 6400 km
- **9.** 50 km
- **10.** 1920 km
- **11.** $d = \frac{3}{4}R$
- g R

BEGINNER'S BOX-3

- 1. $\frac{1}{2}$ mgR
- $2. v = v_0 + K \log \left(\frac{r}{r_0}\right)$
- 3. $-2200 \times 6.67 \times 10^{-11} \text{ J/kg}$
- **4.** (B)
- **5.** (D)
- **6.** (D)

7. (C)

BEGINNER'S BOX-4

- 1. $\left(\frac{2}{5}\right)^{3/2}$ years
- **2.** $2\sqrt{2}$ years
- **3.** (D)
- **4.** (D)
- **5.** $6.47 \times 10^{24} \text{ kg}$
- **6.** \therefore angular momentum $L = m_p r_p v_p = m_p r_A v_A$

$$\label{eq:vp} \therefore \qquad \frac{v_{\rm p}}{v_{\rm A}} = \frac{r_{\rm A}}{r_{\rm p}} \qquad \text{Since } r_{\rm A} > r_{\rm p}, v_{\rm p} > v_{\rm A}.$$

The area SBAC bounded by the ellipse and the radius vectors SB and SC is larger than SBPC in Fig. From Kepler's second law, equal areas are swept in equal time intervals. Hence the planet will take a longer time to traverse BAC than CPB.

- **7.** All quantities vary over an orbit except angular momentum and total energy.
- **8.** 9.7 days
- 9. $\sqrt{\frac{3\pi}{G\rho}}$
- **10.** $\frac{12}{1}$
- **11.** 5.89 × 10⁹ J
- 12. Kinetic energy increases, but potential energy decreases, and the sum decreases due to dissipation against friction.
- **13.** (a) 3.1 km/s; (b) $-\frac{1}{2}$
- **14.** (B)

Ε

Build Up Your Understanding

NEWTON'S LAW OF GRAVITATION & GRAVITATIONAL FIELD

- 1. Newton's law of gravitation:
 - (1) is not applicable out side the solar system
 - (2) is used to govern the motion of satellites only
 - (3) control the rotational motion of satellites and
 - (4) control the rotational motion of electrons in atoms
- 2. Mass particles of 1 kg each are placed along x-axis at $x = 1, 2, 4, 8, \dots, \infty$. Then gravitational force on a mass of 3kg placed at origin is (G = universal)gravitational constant) :-
 - (1) 4G
- (2) $\frac{4G}{3}$ (3) 2G
- $(4) \infty$
- **3**. Gravitational force between two masses at a distance 'd' apart is 6N. If these masses are taken to moon and kept at same separation, then the force between them will become:
 - (1) 1 N
- (2) $\frac{1}{6}$ N
- (3) 36 N
- (4) 6 N
- 4. The value of universal gravitational constant G depends upon:
 - (1) Nature of material of two bodies
 - (2) Heat constant of two bodies
 - (3) Acceleration of two bodies
 - (4) None of these
- **5**. Three identical bodies (each mass M) are placed at vertices of an equilateral triangle of arm L, keeping the triangle as such by which angular speed the bodies should be rotated in their gravitational fields so that the triangle moves along circumference of circular orbit:
 - (1) $\sqrt{\frac{3GM}{I^3}}$
- (2) $\sqrt{\frac{GM}{I^3}}$
- (3) $\sqrt{\frac{GM}{3I^3}}$
- (4) $3\sqrt{\frac{GM}{L^3}}$

6. Four particles of masses m, 2m, 3m and 4m are kept in sequence at the corners of a square of side a. The magnitude of gravitational force acting on a particle of mass m placed at the centre of the square will be:

- (1) $\frac{24\text{m}^2\text{G}}{\text{a}^2}$
- (2) $\frac{6m^2G}{2^2}$
- (3) $\frac{4\sqrt{2}Gm^2}{3^2}$
- (4) Zero

7. The tidal waves in the seas are primarily due to:

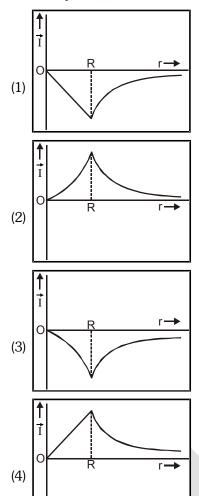
- (1) The gravitational effect of the sun on the earth
- (2) The gravitational effect of the moon on the earth
- (3) The rotation of the earth
- (4) The atmospheric effect of the earth it self

8. During the journey of space ship from earth to moon and back, the maximum fuel is consumed :-

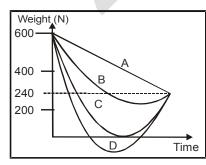
- (1) Against the gravitation of earth in return journey
- (2) Against the gravitation of earth in onward journey
- (3) Against the gravitation of moon while reaching the moon
- (4) None of the above
- 9. If the distance between the centres of earth and moon is D and mass of earth is 81 times that of moon. At what distance from the centre of earth gravitational field will be zero:
- (1) $\frac{D}{2}$ (2) $\frac{2D}{3}$ (3) $\frac{4D}{5}$ (4) $\frac{9D}{10}$
- **10**. An earth's satellite is moving in a circular orbit with a uniform speed v. If the gravitational force of the earth suddenly disappears, the satellite will:-
 - (1) vanish into outer space
 - (2) continue to move with velocity v in original orbit
 - (3) fall down with increasing velocity
 - (4) fly off tangentially from the orbit with velocity v

Pre-Medical: Physics

Following curve shows the variation of intensity of gravitational field (\vec{I}) with distance from the centre of solid sphere(r):



12. Suppose the acceleration due to gravity at the earth's surface is 10m/s² and at the surface of mars it is 4.0 m/s². A 60kg passenger goes from the earth to the mars in a spaceship moving with a constant velocity. Neglect all other objects in the sky. Which part of figure best represent the weight (Net gravitational force) of the passenger as a function of time:



200 (1) A

(2) B

(3) C

(4) D

13. Assume that a tunnel is dug through earth from North pole to south pole and that the earth is a non-rotating, uniform sphere of density ρ. The gravitational force on a particle of mass m dropped into the tunnel when it reaches a distance r from the centre of earth is

(1)
$$\left(\frac{3}{4\pi}\text{mG}\rho\right)$$

(1)
$$\left(\frac{3}{4\pi} \text{mG}\rho\right) \text{r}$$
 (2) $\left(\frac{4\pi}{3} \text{mG}\rho\right) \text{r}$

(3)
$$\left(\frac{4\pi}{3}\text{mG}\rho\right)r^2$$
 (4) $\left(\frac{4\pi}{3}\text{m}^2\text{G}\rho\right)r$

(4)
$$\left(\frac{4\pi}{3}\text{m}^2\text{G}\rho\right)$$

14. Mars has a diameter of approximately 0.5 of that of earth, and mass of 0.1 of that of earth. The surface gravitational field strength on mars as compared to that on earth is a factor of -

(1) 0.1

(2) 0.2

(3) 2.0

(4) 0.4

Three equal masses of 1 kg each are placed at the vertices of an equilateral triangle PQR and a mass of 2 kg is placed at the centroid O of the triangle which is at a distance of $\sqrt{2}$ m from each of the vertices of the triangle. The force, in newton, acting on the mass of 2 kg is :-

(1) 2

(3) 1

(4) zero

One can easily "weigh the earth" by calculating the **16**. mass of earth using the formula (in usual notation)

(1)
$$\frac{G}{g}R_E^2$$

(2) $\frac{g}{G}R_E^2$

(3) $\frac{g}{G}R_E$

(4) $\frac{G}{\sigma}R_E^3$

ACCELERATION DUE TO GRAVITY

Acceleration due to gravity at the centre of the **17**. earth is :-

(1) g

(2) $\frac{g}{2}$

(3) zero

(4) infinite

- The value of 'g' on earth surface depends :-
 - (1) only an earth's structure
 - (2) only an earth's rotational motion
 - (3) on above both
 - (4) on none these and is same
- **19**. The value of 'g' reduces to half of its value at surface of earth at a height 'h', then :-
 - (1) h = R
- (2) h = 2R
- (3) $h = (\sqrt{2} + 1)R$ (4) $h = (\sqrt{2} 1)R$
- At some planet 'g' is 1.96 m/sec². If it is safe to jump **20**. from a height of 2m on earth, then what should be corresponding safe height for jumping on that planet
 - (1) 5m
- (2) 2m
- (3) 10m
- (4) 20m
- **21**. If the earth stops rotating suddenly, the value of g at a place other than poles would :-
 - (1) Decrease
 - (2) Remain constant
 - (3) Increase
 - (4) Increase or decrease depending on the position of earth in the orbit round the sun
- **22**. Diameter and mass of a planet is double that earth. Then time period of a pendulum at surface of planet is how much times of time period at earth surface :-
 - (1) $\frac{1}{\sqrt{2}}$ times
- (2) $\sqrt{2}$ times
- (3) Equal
- (4) None of these
- Gravitation on moon is 1/6th of that on earth. When **23**. a balloon filled with hydrogen is released on moon then, this :-
 - (1) Will rise with an acceleration less then $\left(\frac{g}{6}\right)$
 - (2) Will rise with acceleration $\left(\frac{g}{6}\right)$
 - (3) Will fall down with an acceleration less than $\left(\frac{5g}{6}\right)$
 - (4) Will fall down with acceleration $\left(\frac{g}{6}\right)$

- 24. The acceleration due to gravity g and mean density of earth p are related by which of the following relations? [G = gravitational constant and R = radius]of earth1:
 - $(1) \ \rho = \frac{4\pi g R^2}{3G}$
- (2) $\rho = \frac{4\pi g R^3}{3G}$
- (3) $\rho = \frac{3g}{4\pi GR}$
- (4) $\rho = \frac{3g}{4\pi GR^3}$
- **25**. More amount of sugar is obtained in 1kg weight:
 - (1) At North pole
 - (2) At equator
 - (3) Between pole and equator
 - (4) At South pole
- **26**. When you move from equator to pole, the value of acceleration due to gravity (g) :-
 - (1) increases
 - (2) decreases
 - (3) remains the same
 - (4) first increases then decreases
- **27**. When the radius of earth is reduced by 1% without changing the mass, then the acceleration due to gravity will
 - (1) increase by 2%
- (2) decrease by 1.5%
- (3) increase by 1%
- (4) decrease by 1%
- 28. Weight of a body of mass m decreases by 1% when it is raised to height h above the earth's surface. If the body is taken to a depth h in a mine, then in its weight will
 - (1) decrease by 0.5%
- (2) decrease by 2%
- (3) increase by 0.5%
- (4) increase by 1%
- **29**. Acceleration due to gravity at earth's surface is 'g' m/s². Find the effective value of acceleration due to gravity at a height of 32 km from sea level : $(R_e = 6400 \text{ Km})$
 - (1) 0.5 g m/s^2
- (2) 0.99 g m/s^2
- (3) 1.01 g m/s²
- (4) 0.90 g m/s²
- **30**. The mass of the moon is 1% of mass of the earth. The ratio of gravitational pull of earth on moon to that of moon on earth will be:
 - (1) 1 : 1
- (2) 1 : 10 (3) 1 : 100 (4) 2 : 1

- Imagine a new planet having the same density as that of earth but its radius is 3 times bigger than the earth in size. If the acceleration due to gravity on the surface of earth is g and that on the surface of the new planet is g', then:
 - (1) g' = 3g
- (2) g' = g/9
- (3) g' = 9g
- (4) g'=27 g
- **32**. The change in the value of 'g' at a height 'h' above the surface of the earth is same as at a depth 'd'. If 'd' and 'h' are much smaller than the radius of earth, then which one of the following is correct?
 - (1) d = h
- (2) d = 2h
- (3) $d = \frac{3h}{2}$
- (4) d = h/2
- **33**. If the rotational speed of earth is increased then weight of a body at the equator
 - (1) increases
- (2) decreases
- (3) becomes double
- (4) does not changes
- A body weighs W newton at the surface of the earth. **34**. Its weight at a height equal to half the radius of the earth will be:
- (1) $\frac{W}{2}$ (2) $\frac{2W}{3}$ (3) $\frac{4W}{9}$ (4) $\frac{W}{4}$
- **35**. The imaginary angular velocity of the earth for which the effective acceleration due to gravity at the equator shall be zero is equal to

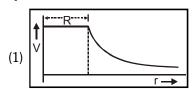
 - (1) 1.25×10^{-3} rad/s (2) 2.50×10^{-3} rad/s
 - (3) 3.75×10^{-3} rad/s (4) 5.0×10^{-3} rad/s

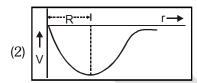
Take $g = 10 \text{m/s}^2$ for the acceleration due to gravity if the earth were at rest and radius of earth equal to 6400 km.]

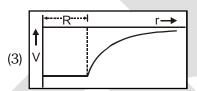
GRAVITATIONAL POTENTIAL ENERGY & POTENTIAL

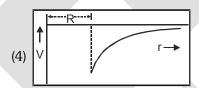
- **36**. Two different masses are droped from same heights. When these just strike the ground, the following is same :
 - (1) kinetic energy
- (2) potential energy
- (3) linear momentum
- (4) acceleration

Which of the following curve expresses the variation **37**. of gravitational potential with distance for a hollow sphere of radius R:









- 38. Gravitational potential difference between surface of a planet and a point situated at a height of 20m above its surface is 2 J/kg. If gravitational field is uniform, then the work done in taking a 5kg body upto height 4 meter above surface will be :
 - (1) 2 J
- (2) 20 J
- (3) 40 J
- (4) 10 J
- If $\mathbf{M}_{_{\!\mathit{e}}}$ is the mass of earth and $\mathbf{M}_{_{\!\mathit{m}}}$ is the mass of **39**. moon $(M_e = 81 M_m)$. The potential energy of an object of mass m situated at a distance R from the centre of earth and r from the centre of moon, will be :-

(1)
$$-GmM_m \left(\frac{R}{81} + r\right) \frac{1}{R^2}$$
 (2) $-GmM_e \left(\frac{81}{r} + \frac{1}{R}\right)$

- (3) $-\text{GmM}_{m} \left(\frac{81}{R} + \frac{1}{r} \right)$ (4) $\text{GmM}_{m} \left(\frac{81}{R} \frac{1}{r} \right)$
- **40**. The gravitational potential energy is maximum at:
 - (1) infinity
 - (2) the earth's surface
 - (3) The centre of the earth
 - (4) Twice the radius of the earth

- A missile is launched with a velocity less than the escape velocity. Sum of its kinetic energy and potential energy is :-
 - (1) Positive
 - (2) Negative
 - (3) May be negative or positive depending upon its initial velocity
 - (4) Zero
- 42 A body attains a height equal to the radius of the earth when projected from earth' surface. The velocity of the body with which it was projected is:
 - (1) $\sqrt{\frac{GM}{P}}$
- (2) $\sqrt{\frac{2GM}{R}}$
- (3) $\sqrt{\frac{5}{4} \frac{GM}{R}}$
- (4) $\sqrt{\frac{3GM}{P}}$
- **43**. The gravitational potential energy of a body at a distance r from the center of the earth is U. The force at that point is:

 - $(1) \frac{U}{r^2} \qquad (2) \frac{U}{r}$
- (3) Ur
- (4) Ur²
- A particle falls from infinity to the earth. Its velocity **44**. on reaching the earth surface is:
 - (1) 2Rg
- $(3) \sqrt{Rq}$
- $(4) \sqrt{2Rq}$
- **45.** A projectile of mass m is thrown vertically up with an initial velocity v from the surface of earth (mass of earth = M). If it comes to rest at a height h, the change in its potential energy is
 - (1) GMmh/R(R + h)
- (2) $GMmh^2/R(R + h)^2$
- (3) GMmhR/R(R + h)
- (4) GMm/hR(R+h)
- **46**. Two small and heavy spheres, each of mass M, are placed a distance r apart on a horizontal surface. The gravitational potential at the mid-point on the line joining the centre of the spheres is :-
 - (1) Zero

- An artificial satellite moving in a circular orbit around 47. the earth has a total (kinetic + potential) energy E_0 . Its potential energy is :-
 - $(1) E_0$
- (2) E_0
- $(3) 2E_0$
- $(4) 2E_0$
- 48. A particle of mass m is moving in a horizontal ciricle of radius R under a centripetal force equal to - $\frac{A}{L^2}$ (A = constant). The total energy of the particle

(Potential energy at very large distance is zero)

ESCAPE VELOCITY

- Potential energy of a 3kg body at the surface of a **49**. planet is - 54J, then escape velocity will be:
 - (1) 18 m/s
- (2) 162 m/s
- (3) 36 m/s
- (4) 6 m/s
- **50**. Escape velocity of a 1kg body on a planet is 100 m/s. Potential energy of body at that planet is:
 - (1) 5000J
- (2) -1000J
- (3) -2400J
- (4) -10000J
- The ratio of radii of two satellites is p and the ratio of their acceleration due to gravity is q. The ratio of their escape velocities will be:
 - (1) $\left(\frac{q}{p}\right)^{\frac{1}{2}}$
- (2) $\left(\frac{p}{q}\right)^{\frac{1}{2}}$
- (3) pq

- $(4) \sqrt{pq}$
- **52**. Escape velocity of a body from earth is 11.2 km/s. Escape velocity, when thrown at an angle of 45° from horizontal will be :-
 - (1) 11.2 km/s
 - (2) 22.4 km/s
 - (3) $11.2/\sqrt{2}$ km/s
 - (4) $11.2\sqrt{2}$ km/s

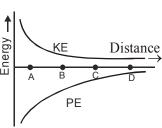
Pre-Medical: Physics

- The escape velocity from the earth is 11.2 km/s the **53**. mass of another planet is 100 times of mass of earth and its radius is 4 times the radius of earth. The escape velocity for the planet is :-
 - (1) 56.0 km/s
- (2) 280 km/s
- (3) 112 km/s
- (4) 11.2 km/s
- **54**. Body is projected vertically upward from the surface of the earth with a velocity equal to half the escape velocity. If R is radius of the earth, the maximum height attained by the body is :-
- (1) $\frac{R}{6}$ (2) $\frac{R}{3}$ (3) $\frac{2}{3}R$
- (4) R

PLANETARY MOTION & WEIGHTLESSNESS

- Binding energy of moon and earth is :-
 - $(1) \frac{GM_eM_m}{r_{om}}$
- $(2) \frac{GM_e M_m}{2r_{m}}$
- $(3) -\frac{GM_eM_m}{r}$
- **56**. Two artificial satellites A and B are at a distance r. and $r_{_{\rm B}}$ above the earth's surface. If the radius of earth is R, then the ratio of their speed will be :-
 - $(1) \left(\frac{r_B + R}{r_L + R} \right)^{\frac{1}{2}}$
- $(2) \left(\frac{r_B + R}{r_A + R} \right)^2$
- (3) $\left(\frac{r_B}{r}\right)^2$
- The average radii of orbits of mercury and earth **57**. around the sun are 6×10^7 km and 1.5×10^8 km respectively. The ratio of their orbital speeds will be :-
 - (1) $\sqrt{5}:\sqrt{2}$
- (2) $\sqrt{2} \cdot \sqrt{5}$
- (3) 2.5 : 1
- (4) 1 : 25
- **58**. A body is dropped by a satellite in its geo-stationary
 - (1) it will burn on entering in to the atmosphere
 - (2) it will remain in the same place with respect to the earth
 - (3) it will reach the earth is 24 hours
 - (4) it will perform uncertain motion

- **59**. Two ordinary satellites are revolving round the earth in same elliptical orbit, then which of the following quantities is conserved :-
 - (1) Velocity
 - (2) Angular velocity
 - (3) Angular momentum
 - (4) None of above
- **60**. Kepler's second law is a consequence of :-
 - (1) conservation of kinetic energy
 - (2) conservation of linear momentum
 - (3) conservation of angular momentum
 - (4) conservation of speed
- One projectile after deviating from its path starts **61**. moving round the earth in a cirular path of radius equal to nine times the radius of earth R. Its time period will be :-
 - (1) $2\pi\sqrt{\frac{R}{\sigma}}$
- (2) $27 \times 2\pi \sqrt{\frac{R}{g}}$
- (3) $\pi \sqrt{\frac{R}{\epsilon}}$
- (4) $0.8 \times 3\pi \sqrt{\frac{R}{R}}$
- **62**. In adjoining figure earth goes around the sun in elliptical orbit on which point the orbital speed is maximum:
 - (1) On A
 - (2) On B
 - (3) On C
 - (4) On D
- 63. Potential energy and kinetic energy of a two particle system under imaginary force field are shown by curves KE and PE. respectively in figure. This system is bound at:
 - (1) only point A
 - (2) only point D
 - (3) only point
 - A, B, and C
 - (4) All points A,
 - B, C and D



- A satellite of earth of mass 'm' is taken from orbital radius 2R to 3R, then minimum work done is :-

 - (1) $\frac{\text{GMm}}{\text{GR}}$ (2) $\frac{\text{GMm}}{12\text{R}}$ (3) $\frac{\text{GMm}}{24\text{R}}$ (4) $\frac{\text{GMm}}{3\text{R}}$
- If a graph is plotted between T² and r³ for a planet **65**. then its slope will be :-
 - (1) $\frac{4\pi^2}{GM}$ (2) $\frac{GM}{4\pi^2}$ (3) $4\pi GM$ (4) Zero

- A planet is revolving round the sun. Its distance from **66**. the sun at Apogee is r_A and that at Perigee is r_D . The mass of planet and sun is m and M respectively, $v_{_{A}}$ and $v_{_{P}}$ is the velocity of planet at Apogee and Perigee respectively and T is the time period of revolution of planet round the sun.
 - (a) $T^2 = \frac{\pi^2}{2GM} (r_A + r_P)^2$ (b) $T^2 = \frac{\pi^2}{2GM} (r_A + r_P)^3$
 - (c) $v_{A}r_{A} = v_{P}r_{P}$
- (d) $v_A < v_P$, $r_A > r_P$

- (1) a, b, c (2) a, b, d (3) b, c, d (4) all
- **67**. A satellite launching station should be:
 - (1) near the equatorial region
 - (2) near the polar region
 - (3) on the polar axis
 - (4) all locations are equally good
- A space shuttle is launched in a circular orbit near **68**. the earth's surface. The additional velocity be given to the space - shuttle to get free from the influence of gravitational force, will be:
 - (1) 1.52 km/s
- (2) 2.75 km/s
- (3) 3.28 km/s
- (4) 5.18 km/s
- A satellite is moving in a circular orbit around earth with a speed v. If its mass is m, then its total energy
- (1) $\frac{3}{4}$ mv² (2) mv² (3) $\frac{1}{2}$ mv² (4) $-\frac{1}{2}$ mv²
- If the length of the day is T, the height of that TV satellite above the earth's surface which always appears stationary from earth, will be:

 - (1) $h = \left[\frac{4\pi^2 GM}{T^2}\right]^{\frac{1}{3}}$ (2) $h = \left[\frac{4\pi^2 GM}{T^2}\right]^{\frac{1}{2}} R$

 - (3) $h = \left[\frac{GMT^2}{4\pi^2}\right]^{\frac{1}{3}} R$ (4) $h = \left[\frac{GMT^2}{4\pi^2}\right]^{\frac{1}{3}} + R$

- If two bodies of mass M and m are revolving around 71. the centre of mass of the system in circular orbit of radii R and r respectively due to mutual interaction. Which of the following formula is applicable:
 - (1) $\frac{GMm}{(R+r)^2} = m\omega^2 r$ (2) $\frac{GMm}{R^2} = m\omega^2 r$
 - (3) $\frac{GMm}{r^2} = m\omega^2 R$ (4) $\frac{GMm}{R^2 + r^2} = m\omega^2 r$
- **72**. Two satellites of same mass m are revolving round of earth (mass M) in the same orbit of radius r. Rotational directions of the two are opposite therefore, they can collide. Total mechanical energy of the system (both satallites and earths) is $(m \ll M)$:-
 - $(1) -\frac{GMm}{r}$
- $(3) \frac{GMm}{2r}$
- (4) Zero
- **73**. A planet of mass m is moving in an elliptical orbit about the sun (mass of sun = M). The maximum and minimum distances of the planet from the sun are r₁ and r₂ respectively. The period of revolution of the planet will be proportional to:

- (4) $(r_1 + r_2)^{\frac{3}{2}}$
- **74**. The relay satellite transmits the television programme continuously from one part of the world to another because its:
 - (1) Period is greater than the period of rotation of the earth about its axis
 - (2) Period is less than the period of rotation of the earth about its axis
 - (3) Period is equal to the period of rotation of the earth about its axis
 - (4) Mass is less than the mass of earth
- If the satellite is stopped suddenly in its orbit which **75**. is at a distance radius of earth from earth's surface and allowed to fall freely into the earth. The speed with which it hits the surface of earth will be:
 - (1) 7.919 m/s
- (2) 7.919 km/s
- (3) 11.2 m/s
- (4) 11.2 km/s

Pre-Medical: Physics

- A planet is moving in an elliptical orbit. If T, U, E and Lare its kinetic energy, potential energy, total energy and magnitude of angular momentum respectively, then which of the following statement is true :-
 - (1) T is conserved
 - (2) U is always positive
 - (3) E is always negative
 - (4) L is conserved but the direction of vector \vec{L} will continuously change
- The gravitational force between two bodies is directly proportional to $\frac{1}{R}$ (not $\frac{1}{R^2}$), where 'R' is the distance between the bodies. Then the orbital speed for this force in circular orbit is proportional to :-
 - (1) $1/R^2$
- (2) R°
- (3) R
- (4) 1/R
- **78**. What will be velocity of a satellite revolving around the earth at a height h above surface of earth if radius of earth is R:-

 - (1) $R^2 \sqrt{\frac{g}{R+h}}$ (2) $R \frac{g}{(R+h)^2}$

 - (3) $R\sqrt{\frac{g}{R+h}}$ (4) $R\sqrt{\frac{R+h}{g}}$
- Two artificial satellites of masses m₁ and m₂ are moving with speeds v_1 and v_2 in orbits of radii r_1 and r_2 respectively. If $r_1 > r_2$ then which of the following statements in true :-
 - $(1) v_1 = v_2$
- (3) $v_1 < v_2$
- (2) $v_1 > v_2$ (4) $v_1/r_1 = v_2/r_2$
- Orbital radius of a satellite S of earth is four times that of a communication satellite C. Period of revolution of S is :-
 - (1) 4 days
- (2) 8 days
- (3) 16 days
- (4) 32 days
- If a satellite is revolving very close to the surface of **81**. earth, then its orbital velocity does not depend upon:-
 - (1) Mass of satellite
- (2) Mass of earth
- (3) Radius of earth
- (4) Orbital radius

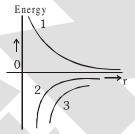
- Two identical satellites are at the heights R and 7R **82**. from the earth's surface. Then which of the following statement is incorrect :-
 - (R = Radius of the earth)
 - (1) Ratio of total energy of both is 5
 - (2) Ratio of kinetic energy of both is 4
 - (3) Ratio of potential energy of both 4
 - (4) Ratio of total energy of both is 4
- **83**. The minimum projection velocity of a body from the earth's surface so that it becomes the satellite of the earth ($R_a = 6.4 \times 10^6 \,\text{m}$).
 - (1) 11×10^3 m/s
- (2) 8×10^3 m/s
- (3) 6.4×10^3 m/s
- (4) 4×10^3 m/s
- Geostationary satellite :-**84**.
 - (1) is situated at a great height above the surface of earth
 - (2) moves in equatorial plane
 - (3) have time period of 24 hours
 - (4) have time period of 24 hours and moves in equatorial plane
- **85**. The maximum and minimum distances of a comet from the sun are 8×10^{12} m and 1.6×10^{12} m respecting. If its velocity when it is nearest to the sun is 60 m/s then what will be its velocity in m/s when it is farthest?
 - (1) 12
- (2)60
- (3) 112
- (4) 6
- 86. A satellite of mass m goes round the earth along a circular path of radius r. Let $m_{\scriptscriptstyle F}$ be the mass of the earth and $R_{\scriptscriptstyle E}$ its radius then the linear speed of the satellite depends on.
 - (1) m, m_E and r
- (2) m, R_E and r
- (3) m_F only
- (4) m_F and r
- Near the earth's surface time period of a satellite **87**. is 1.4 hrs. Find its time period if it is at the distance '4R' from the centre of earth:
 - (1) 32 hrs.
- (2) $\left(\frac{1}{8\sqrt{2}}\right)$ hrs.
- (3) $8\sqrt{2}$ hrs.
- (4) 16 hrs.

- A communication satellite of earth which takes 24 hrs. to complete one circular orbit eventually has to be replaced by another satellite of double mass. If the new satellites also has an orbital time period of 24 hrs, then what is the ratio of the radius of the new orbit to the original orbit?
 - (1) 1 : 1
- (2) 2 : 1
- (3) $\sqrt{2}:1$
- $(4)\ 1:2$
- **89**. Escape velocity for a projectile at earth's surface is V_a. A body is projected form earth's surface with velocity 2 V_e. The velocity of the body when it is at infinite distance from the centre of the earth is :-
 - $(1) V_{o}$
- (2) $2V_e$ (3) $\sqrt{2} V_e$ (4) $\sqrt{3} V_e$
- For a satellite moving in an orbit around the earth, **90**. the ratio of kinetic energy to potential energy is :-
 - (1) 2

- (2) 1/2 (3) $\frac{1}{\sqrt{2}}$ (4) $\sqrt{2}$
- The orbital velocity of an artificial satellite in a 91. circular orbit just above the earth's surface is v_0 . The orbital velocity of satellite orbiting at an altitude of half of the radius is :-

- (1) $\frac{3}{2}v_0$ (2) $\frac{2}{3}v_0$ (3) $\sqrt{\frac{2}{3}}v_0$ (4) $\sqrt{\frac{3}{2}}v_0$
- **92**. The earth revolves around the sun in one year. If distance between them becomes double, the new time period of revolution will be :-
 - (1) $4\sqrt{2}$ years
- (2) $2\sqrt{2}$ years
- (3) 4 years
- (4) 8 years

- A satellite of mass m revolves in a circular orbit of 93. radius R a round a planet of mass M. Its total energy
 - (1) $-\frac{GMm}{2R}$
- $(2) + \frac{GMm}{3R}$
- (3) $-\frac{GMm}{R}$
- (4) $+\frac{GMm}{R}$
- 94. A satellite is orbiting earth at a distance r. Variations of its kinetic energy, potential energy and total energy, is shown in the figure. Of the three curves shown in figure, identify the type of mechanical energy they represent.



- (1) 1 Potential, 2 Kinetic, 3 Total
- (2) 1 Total, 2 Kinetic, 3 Potential
- (3) 1 Kinetic, 2 Total, 3 Potential
- (4) 1 Potential, 2 Total, 3 Kinetic
- The mean distance of mars from sun is 1.5 times 95. that of earth from sun. What is approximately the number of years required by mars to make one revolution about sun?
 - (1) 2.35 years
- (2) 1.85 years
- (3) 3.65 years
- (4) 2.75 years

EX	ERC	ISE-I	(Conc	eptua	al Que	stions	5)						ANS	WER	KEY
Que.	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Ans.	3	1	4	4	1	3	2	2	4	4	1	3	2	4	4
Que.	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30
Ans.	2	3	3	4	3	3	2	4	3	2	1	1	1	2	1
Que.	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45
Ans.	1	2	2	3	1	4	3	1	3	1	2	1	2	4	1
Que.	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60
Ans.	4	4	4	4	1	4	1	1	2	2	1	1	2	3	3
Que.	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75
Ans.	2	1	3	2	1	3	1	3	4	3	1	1	4	3	2
Que.	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90
Ans.	3	2	3	3	2	1	1	2	4	1	4	3	1	4	2
Que.	91	92	93	94	95										
Ans.	3	2	1	3	2										

EXERCISE-II (Previous Year Questions)

AIPMT 2007

- 1. Two satellites of earth, S_1 and S_2 , are moving in the same orbit. The mass of S_1 is four times the mass of S₂. Which one of the following statements is true?
 - (1) The kinetic energies of the two satellites are equal
 - (2) The time period of S_1 is four times that of S_2
 - (3) The potential energies of earth and satellite in the two cases are equal
 - (4) S_1 and S_2 are moving with the same speed

AIPMT 2009

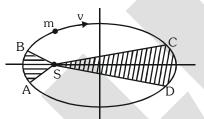
2. The figure shows elliptical orbit of a planet m about the sun S. The shaded area SCD is twice the shaded area SAB. If t_1 is the time for the planet to move from C to D and t2 is the time to move from A to B then :-

(1)
$$t_1 = t_2$$

(2)
$$t_1 < t_2$$

(3)
$$t_1 = 4t_2$$

(4)
$$t_1 = 2t_2$$



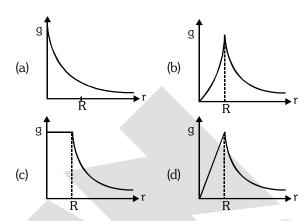
AIPMT 2010

- 3. The radii of circular orbits of two satellites A and B of the earth, are 4R and R, respectively. If the speed of satellite A is 3V, then the speed of satellite B will be :-
 - (1) 3V/2
- (2) 3V/4
- (3) 6V
- (4) 12V
- 4. A particle of mass M is situated at the centre of a spherical shell of same mass and radius a. The gravitational potential at a point situated at $\frac{a}{2}$ distance from the centre, will be :-

(1)
$$-\frac{4GM}{a}$$
 (2) $-\frac{3GM}{a}$ (3) $-\frac{2GM}{a}$ (4) $-\frac{GM}{a}$

AIPMT/NEET & AIIMS (2006-2018

5. The dependence of acceleration due to gravity 'g' on the distance 'r' from the centre of the earth, assumed to be a sphere of radius R of uniform density, is as shown in figure below:-



The correct figure is :-

- (1) (a)
- (2) (b)
- (3) (c)
- (4) (d)
- 6. The additional kinetic energy to be provided to a satellite of mass m revolving around a planet of mass M, to transfer it from a circular orbit of radius R_1 to another of radius $R_2(R_2 > R_1)$ is :-

$$(1) \ \mathsf{GmM}\bigg(\frac{1}{\mathsf{R}_1} - \frac{1}{\mathsf{R}_2}\bigg)$$

(1)
$$GmM\left(\frac{1}{R_1} - \frac{1}{R_2}\right)$$
 (2) $2GmM\left(\frac{1}{R_1} - \frac{1}{R_2}\right)$

(3)
$$\frac{1}{2}$$
GmM $\left(\frac{1}{R_1} - \frac{1}{R_2}\right)$ (4) GmM $\left(\frac{1}{R_1^2} - \frac{1}{R_2^2}\right)$

(4)
$$GmM \left(\frac{1}{R_1^2} - \frac{1}{R_2^2} \right)$$

AIPMT 2011

7. A planet moving along an elliptical orbit is closest to the sun at a distance r₁ and farthest away at a distance of r_2 . If v_1 and v_2 are the linear velocities at these points respectively, then the

ratio
$$\frac{v_1}{v_2}$$
 is :-

- (1) $(r_1/r_2)^2$ (2) r_2/r_1 (3) $(r_2/r_1)^2$ (4) r_1/r_2

AIPMT Pre. 2012

- 8. A spherical planet has a mass M_p and diameter D_p . A particle of mass m falling freely near the surface of this planet will experience an aceleration due to gravity, equal to :-
 - (1) GM_p/D_p^2
- (2) $4GM_{\rm p}m/D_{\rm p}^{2}$
- (3) $4GM_{p}/D_{p}^{2}$
- (4) $GM_{p}m/D_{p}^{2}$

- 9. A geostationary satellite is orbiting the earth at a height of 5R above that surface of the earth, R being the radius of the earth. The time period of another satellite in hours at a height of 2R from the surface of the earth is:-
 - (1) $6\sqrt{2}$
- (2) $6/\sqrt{2}$

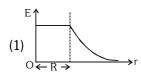
(3) 5

- (4) 10
- 10. The height at which the weight of a body becomes $1/16^{th}$, its weight on the surface of earth (radius R), is :-
 - (1) 3R
- (2) 4R

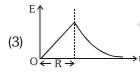
- (3) 5R
- (4) 15R

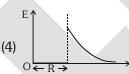
AIPMT Mains 2012

11. Which one of the following plots represents the variation of gravitational field on a particle with distance r due to a thin spherical shell of radius R? (r is measured from the centre of the spherical shell)









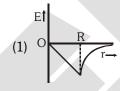
- 12. If v_e is escape velocity and v_0 is orbital velocity of a satellite for orbit close to the earth's surface, then these are related by :
 - (1) $v_e = \sqrt{2v_0}$
- $(2) \quad \mathbf{v}_e = \sqrt{2} \mathbf{v}_0$
- (3) $v_0 = \sqrt{2}v_e$
- $(4) v_0 = v_e$

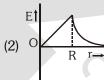
AIIMS 2012

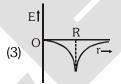
- **13.** A particle is thrown with escape velocity v_e from the surface of earth. Calculate its velocity at height 3 R :-
 - $(1) \simeq 9.25 \text{ km/s}$
- $(2) \approx 5.6 \text{ km/s}$
- $(3) \simeq 11.2 \text{ km/s}$
- $(4) \simeq 4.3 \text{ km/s}$

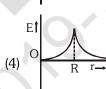
AIPMT 2014

- 14. A black hole is an object whose gravitational field is so strong that even light cannot escape from it. To what approximate radius would earth (mass = 5.98×10^{24} kg) have to be compressed to be a black hole?
 - (1) 10⁻⁹ m
- (2) 10⁻⁶ m
- (3) 10⁻² m
- (4) 100 m
- **15.** Dependence of intensity of gravitational field (E) of earth with distance (r) from centre of earth is correctly represented by :-









AIPMT 2015

16. Kepler's third law states that square of period of revolution (T) of a planet around the sun, is proportional to third power of average distance r between sun and planet

i.e.
$$T^2 = Kr^3$$

here K is constant.

If the masses of sun and planet are M and m respectively then as per Newton's law of gravitation force of attraction between them is

 $F = \frac{GMm}{r^2}, \ \text{here} \ G \ \text{is gravitational constant}.$ The relation between G and K is described as :

- (1) GMK = $4\pi^2$
- (2) K = G
- $(3) K = \frac{1}{G}$
- (4) $GK = 4\pi^2$

Re-AIPMT 2015

- **17.** A satellite S is moving in an elliptical orbit around the earth. The mass of the satellite is very small compared to the mass of the earth. Then,
 - (1) the acceleration of S is always directed towards the centre of the earth.
 - (2) the angular momentum of S about the centre of the earth changes in direction, but its magnitude remains constant.
 - (3) the total mechanical energy of S varies periodically with time.
 - (4) the linear momentum of S remains constant in magnitude.
- **18.** A remote sensing satellite of earth revolves in a circular orbit at a height of 0.25×10^6 m above the surface of earth. If earth's radius is 6.38×10^6 m and g=9.8 m/s², then the orbital speed of the satellite is :
 - (1) 6.67 km/s
- (2) 7.76 km/s
- (3) 8.56 km/s
- (4) 9.13 km/s

AIIMS 2015

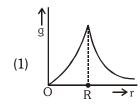
- **19.** Find the velocity of a satellite at height 80 km from earth, if the radius of earth is 6400 km
 - (1) 7 km/s
- (2) 8 km/s
- (3) 7.84 km/s
- (4) 11.2 km/s

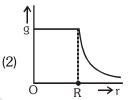
NEET-I 2016

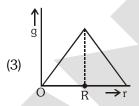
- **20.** At what height from the surface of earth the gravitation potential and the value of g are -5.4×10^7 J/kg and 6.0 m/s² respectively? Take the radius of earth as 6400 km:
 - (1) 2600 km
- (2) 1600 km
- (3) 1400 km
- (4) 2000 km
- **21.** The ratio of escape velocity at earth (v_e) to the escape velocity at a planet (v_p) whose radius and mean density are twice as that of earth is:-
 - (1) 1 : 2
- (2) $1: 2\sqrt{2}$
- (3) 1 : 4
- (4) $1: \sqrt{2}$

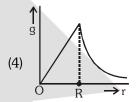
NEET-II 2016

22. Starting from the centre of the earth having radius R, the variation of g (acceleration due to gravity) is shown by :-









- **23.** A satellite of mass m is orbiting the earth (of radius R) at a height h from its surface. The total energy of the satellite in terms of g_0 , the value of acceleration due to gravity at the earth's surface, is :-
 - (1) $\frac{2mg_0R^2}{R+h}$
- (2) $-\frac{2mg_0R^2}{R+h}$
- (3) $\frac{mg_0R^2}{2(R+h)}$
- (4) $-\frac{mg_0R^2}{2(R+h)}$

AIIMS 2016

- **24.** Find out energy required (in Giga Joule) to escape a space shuttle of 1000 kg mass from surface of earth. (R_o = 6400 km)
 - (1) 20
- (2)64
- (3) 50
- (4) 28

NEET(UG) 2017

- **25.** The acceleration due to gravity at a height 1 km above the earth is the same as at a depth d below the surface of earth. Then:
 - (1) d = 1 km
 - (2) $d = \frac{3}{2} km$
 - (3) d = 2 km
 - (4) $d = \frac{1}{2} \text{ km}$

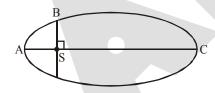
- Two astronauts are floating in gravitational free space after having lost contact with their spaceship. The two will:-
 - (1) Move towards each other.
 - (2) Move away from each other.
 - (3) Will become stationary
 - (4) Keep floating at the same distance between them.

AIIMS 2017

- **27.** Find the angular momentum of satellite of mass 400 kg moving around the earth in radius 4×10^7 m.
 - (1) $50 \times 10^{12} \text{ kgm}^2/\text{s}$
- (2) $512 \times 10^4 \text{ kgm}^2/\text{s}$
- (3) $5 \times 10^{12} \text{ kgm}^2/\text{s}$
- (4) $512 \times 10^6 \text{ kgm}^2/\text{s}$

NEET(UG) 2018

- 28. If the mass of the Sun were ten times smaller and the universal gravitational constant were ten time larger in magnitude, which of the following is **not** correct?
 - (1) Raindrops will fall faster
 - (2) Walking on the ground would become more
 - (3) Time period of a simple pendulum on the Earth would decrease
 - (4) 'g' on the Earth will not change
- **29**. The kinetic energies of a planet in an elliptical orbit about the Sun, at positions A, B and C are KA, KB and K_C respectively. AC is the major axis and SB is perpendicular to AC at the position of the Sun S as shown in the figure. Then



- (1) $K_A < K_B < K_C$ (2) $K_A > K_B > K_C$
- (3) $K_B < K_A < K_C$ (4) $K_B > K_A > K_C$

AIIMS 2018

- **30**. A satellite of mass 2000 kg is revolving in elliptical path around earth. If mean altitude at perigee and apogee are 1600 km and 5600 km respectively, then the energy required to send the satellite in this orbit will be (Radius of earth is 6400 km)
 - (1) $7.2 \times 10^{10} \,\mathrm{J}$
- (2) $7.8 \times 10^{10} \,\mathrm{J}$
- (3) $8.5 \times 10^{10} \text{ J}$
- (4) $5.6 \times 10^{10} \,\mathrm{J}$
- 31. A satellite of mass 1500 kg is revolving around earth. The radius of earth is 6400 km. Minimum and maximum distance of satellite from surface of earth are 1200 km and 2400 km respectively. The approximate time period of satellite is :-
 - (1) 2 h
- (2) 11.5 h
- (3) 19.2 h
- (4) 4.07 h
- A geo synchronous satellite is revolving around the **32**. earth. What will be its orbital radius? (Radius of earth = 6400 km
 - (1) 42440 km
- (2) 36000 km
- (3) 32000 km
- (4) 38400 km
- **33**. A satellite is revolving around the earth such that its minimum and maximum distance from earth's surface are 800 km and 2400 km respectively. Calculate velocity of satellite when its distance from earth is minimum

(given radius of earth = 6400 km)

- (1) $7.91 \times 10^3 \,\text{m/s}$
- (2) 7.91×10^4 m/s
- (3) $6 \times 10^3 \,\text{m/s}$
- $(4) 9 \times 10^3 \text{ m/s}$

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EX	EXERCISE-II (Previous Years Questions) ANSWER KEY														KEY
Que.	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Ans.	4	4	3	2	4	3	2	3	1	1	4	2	2	3	1
Que.	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30
Ans.	1	1	2	3	1	2	4	4	2	3	4	1	4	2	3
Que.	31	32	33												
Anc	1	1	1												

EXERCISE-III (Analytical Questions)

1. Masses and radii of earth and moon are M_1 , M_2 and R₁, R₂ respectively. The distance between their centre is 'd'. The minimum velocity given to mass 'M' from the mid point of line joining their centre so that it will escape:

(1)
$$\sqrt{\frac{4G(M_1 + M_2)}{d}}$$

$$(1) \ \sqrt{\frac{4G\big(M_1+M_2\big)}{d}} \qquad \qquad (2) \ \sqrt{\frac{4G}{d} \frac{M_1 M_2}{\big(M_1+M_2\big)}}$$

(3)
$$\sqrt{\frac{2G}{d} \left(\frac{M_1 + M_2}{M_1 M_2} \right)}$$
 (4) $\sqrt{\frac{2G}{d} (M_1 + M_2)}$

(4)
$$\sqrt{\frac{2G}{d}(M_1 + M_2)}$$

2. If R is the average radius of earth, ω is its angular velocity about its axis and g is the gravitational acceleration on the surface of earth then the cube of the radius of orbit of a geostationary satellite will be equal to :-

(1)
$$\frac{R^2g}{\omega}$$

(2)
$$\frac{R^2\omega^2}{\sigma}$$

(3)
$$\frac{Rg}{\omega^2}$$

(1)
$$\frac{R^2g}{\omega}$$
 (2) $\frac{R^2\omega^2}{g}$ (3) $\frac{Rg}{\omega^2}$ (4) $\frac{R^2g}{\omega^2}$

- 3. A geostationary satellite is orbiting the earth at a height of 6R from the earth's surface (R is the earth's radius). What is the period of rotation of another satellite at a height of 2.5 R from the earth surface?
 - (1) $6\sqrt{2}$ hours
- (2) 10 hours
- (3) $\frac{5\sqrt{5}}{\sqrt{3}}$ hours (4) none of the above
- 4. A planet revolves around the sun in an elliptical orbit. If $\nu_{_{\scriptscriptstyle D}}$ and $\nu_{_{\scriptscriptstyle a}}$ are the velocities of the planet at the perigee and apogee respectively, then the eccentricity of the elliptical orbit is given by:
 - $(1) \frac{v_p}{v}$

(2)
$$\frac{v_{a} - v_{p}}{v_{a} + v_{p}}$$

- (3) $\frac{v_p + v_a}{v_p v_a}$
- (4) $\frac{V_p V_a}{V_p + V_a}$

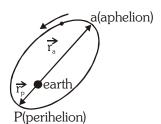
Check Your Understanding

- 5. Two solid spherical planets of equal radii R having masses 4M and 9M their centre are separated by a distance 6R. A projectile of mass m is sent from the planet of mass 4 M towards the heavier planet. What is the distance r of the point from the lighter planet where the gravitational force on the projectile is zero?
 - (1) 1.4 R
- (2) 1.8 R
- (3) 1.5 R
- (4) 2.4 R
- 6. Read the following statements:
 - S₁: An object shall weigh more at pole than at equator when weighed by using a physical balance.
 - S_2 : It shall weigh the same at pole and equator when weighed by using a physical balance.
 - S₃: It shall weigh the same at pole and equator when weighed by using a spring balance.
 - S_4 : It shall weigh more at the pole than at equator when weighed using a spring balance.

Which of the above statements is /are correct?

- (1) S_1 and S_2
- (2) S_1 and S_4
- (3) S_2 and S_2
- (4) S_2 and S_4 .
- 7. A small planet is revolving around a very massive star in a circular orbit of radius R with a period of revolution T. If the gravitational force between the planet and the star were proportional to $R^{-5/2}$, then T would be proportional to
 - (1) $R^{3/2}$
 - (2) $R^{3/5}$
 - (3) $R^{7/2}$
 - $(4) R^{7/4}$

8. Consider a satellite orbiting the earth as shown in the figure below. Let L_a and L_p represent the angular momentum of the satellite about the earth when at aphelion and perihelion respectively. Consider the following relations.



- (i) $\vec{L}_a = \vec{L}_P$
- (ii) $\vec{L}_a = -\vec{L}_P$
- (iii) $\vec{r}_a \times \vec{L}_a = \vec{r}_P \times \vec{L}_P$

Which of the above relations is/are true?

- (1) (i) only
- (2) (ii) only
- (3) (iii) only
- (4) (i) and (iii)

- **9.** A particle falls on earth:
 - (i) from infinity, (ii) from a height 10 times the radius of earth. The ratio of the velocities gained on reaching at the earth's surface is :
 - (1) $\sqrt{11} : \sqrt{10}$
- (2) $\sqrt{10} : \sqrt{11}$
- (3) 10 : 11
- (4) 11 : 10

EXERCISE-III ((Analytical	Questions)
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ANSWER KEY

Que.	1	2	3	4	5	6	7	8	9
Ans.	1	4	1	4	4	4	4	1	1

Directions for Assertion & Reason questions

These questions consist of two statements each, printed as Assertion and Reason. While answering these Questions you are required to choose any one of the following four responses.

- **(A)** If both Assertion & Reason are True & the Reason is a correct explanation of the Assertion.
- **(B)** If both Assertion & Reason are True but Reason is not a correct explanation of the Assertion.
- **(C)** If Assertion is True but the Reason is False.
- **(D)** If both Assertion & Reason are false.
- **1. Assertion**: The difference in the value of acceleration due to gravity at pole and equator is proportional to square of angular velocity of earth.

Reason: The value of acceleration due to gravity is minimum at the equator and maximum at the pole.

[AIIMS 2009]

- (1) A
- (2) B
- (3) C
- (4) D
- **2. Assertion**: Two satellites A & B are in the same orbit around the earth, B being behind A. B cannot overtake A by increasing its speed.

Reason: It will then go into a different orbit.

- (1) A
- (2) B
- (3) C
- (4) D
- **3. Assertion**: A geostationary satellite rotates in a direction from west to east.

Reason: At midnight when the sun is directly below, it pulls on an object in the same direction as the pull of the earth on that object. At noon when the sun is directly above, its pulls on an object in a direction opposite to the pull of the earth.

- (1) A
- (2) B
- (3) C
- (4) D
- **4. Assertion**: The acceleration of a particle near the earth surface differs slightly from the gravitational acceleration $a_{\sigma} = GM/R^2$.

Reason: The earth is not a uniform sphere and because the earth rotates.

- (1) A
- (2) B
- (3) C
- (4) D
- **5. Assertion**: Kepler's law of areas is equivalent to the law of conservation of angular momentum.

Reason: For planetery motion $\frac{dA}{dt} = \frac{L}{2m} = constant$

- (1) A
- (2) B
- (3) C
- (4) D

6. Assertion : For circular orbits, the law of periods is $T^2 \propto r^3$, where M is the mass of sun and r is the radius of orbit.

Reason: The square of the period T of any planet about the sun is proportional to the cube of the semi-major axis a of the orbit.

- (1) A
- (2) B
- (3) C
- (4) D
- **7. Assertion**: If rotation of earth about its own axis is suddenly stops then acceleration due to gravity will increase at all places on the earth.

Reason: At height h from the surface of earth,

acceleration due to gravity is $g_h = g \left(1 - \frac{2h}{R_e} \right)$.

- (1) A
- (2) B
- (3) C
- (4) D
- **8. Assertion**: If the earth stops rotating about its axis, the value of the weight of the body at equator will decrease.

Reaon: The centripetal force does not act on the body at the equator.

- (1) A
- (2) B
- (3) C
- (4) D
- **9. Assertion :**Gravitational potential is maximum at infinite.

Reason: Gravitational potential is the amount of work done to shifting a unit mass from infinity to a given point in gravitational attraction force field.

- (1) A
- (2) B
- (3) C
- (4) D

10. Assertion: A person feels weightlessness in an artificial satellite of the earth. However a person on the moon (natural satellite) feels his weight.

Reason: Artificial satellite is a freely falling body and on the moon surface, the weight is mainly due to moon's gravitational attraction.

- (1) A
- (2) B
- (3) C
- (4) D
- **11. Assertion**: A satellite can not be set on a stable orbit in a plane not passing through the earth's centre.

Reason: For the orbital motion of satellite, the centripetal force is provided by gravitational pull of earth on satellite.

- (1) A
- (2) B
- (3) C
- (4) D
- **12. Assertion:** The time period of revolution of a satellite close to surface of earth is smaller than that revolving far from surface of earth

Reason: The square of time period of revolution of a satellite is directly proportional to cube of its orbital radius.

- (1) A
- (2) B
- (3) C
- (4) D
- **13. Assertion:** The gravitational field of moon is much less than that of earth.

Reason: Gravitational field of a given mass (M) depends upon M/r^2 , which is smaller for moon.

- (1) A
- (2) B
- (3) C
- (4) D
- **14. Assertion :** In keplar's law conservation of angular momentum does not take place.

Reason: Gravitation force acting between earth and the sun is perpendicular to line joining them.

- (1) A
- (2) B
- (3) C
- (4) D
- **15. Assertion:** On decreasing the distance between two masses potential energy increases.

Reason: Potential energy $U = \frac{GMm}{r}$

- (1) A
- (2) B
- (3) C
- (4) D

16. Assertion: A body falling freely under the force of gravity has constant acceleration (9.81 m/sec²).

Reason: Earth attracts every body towards its centre by the same force.

- (1) A
- (2) B
- (3) C
- (4) D
- **17. Assertion**: If an earth satellite moves to a lower orbit, there is some dissipation of energy but the satellite speed increases.

Reason: The speed of satellite is a constant quantity.

- (1) A
- (2) B
- (3) C
- (4) D
- **18. Assertion :** A body kept inside a spherical shell does not experience any gravitational force.

Reason: The body inside a spherical shell is protected from the gravitational attraction of bodies outside the shell.

- (1) A
- (2) B
- (3) C
- (4) D
- **19. Assertion :** If value of acceleration due to gravity of a planet is lesser than earth then escape velocity from planet is lesser too.

Reason: Escape velocity just depends only on acceleration due to gravity.

- (1) A
- (2) B
- (3) C
- (4) D
- **20**. **Assertion:** If gravitational force is proportional to $\frac{1}{R^3}$ then Keplar's second law would not be affected

Reason: For Keplar's second law, central nature of force is the only condition.

- (1) A
- (2) B
- (3) C
- (4) D
- **21.** Assertion: Angular momentum of a satellite remains conserved. [AIIMS 2018]

Reason: Conservation of linear momentum leads to conservation of angular momentum.

- (1) A
- (2) B
- (3) C
- (4) D

E	EXERCISE-IV (Assertion & Reason)												ANS	WER	KEY
Que.	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Ans.	2	1	2	1	1	1	4	4	1	1	1	1	1	4	4
Que.	16	17	18	19	20	21									
Ans.	4	3	4	4	1	3									