# Potential Flow

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# Introduction

This document contains the derivation of a mathematical model used to solve for the airflow around a given aircraft geometry. It assumes potential flow and is accordingly limited in its applicability.

## 1 Problem statement

Given a finite time-invariant parametrized closed surface  $S \subset \mathbb{R}^3$  and sufficient far field boundary conditions, find the pressure distribution on S caused by and adiabatic inviscid irrotational flow given by

$$\mathbf{u} = \nabla \phi \tag{1}$$

where  $\mathbf{u} = \mathbf{u}(\mathbf{x}, t)$  is the flow velocity vector and  $\phi = \phi(\mathbf{x}, t)$  is the velocity potential at some position vector  $\mathbf{x} \in \mathbb{R}^3$  and time coordinate  $t \in \mathbb{R}$ .

#### 1.1 Far field boundary conditions

Given a Cartesian coordinate system in  $\mathbb{R}^3$  centered near S with a position vector  $\mathbf{x}$ , we write the far field boundary conditions as

$$\lim_{|\mathbf{x}| \to \infty} p(\mathbf{x}) = p_{\infty} = \text{const.}$$
 (2a)

$$\lim_{|\mathbf{x}| \to \infty} \rho(\mathbf{x}) = \rho_{\infty} = \text{const.}$$
 (2b)

$$\lim_{|\mathbf{x}| \to \infty} \mathbf{u}(\mathbf{x}) = \mathbf{u}_{\infty} = \text{const.}$$
 (2c)

#### 1.2 Tangential flow

The tangential flow constraint requires that, for any normal vector  $\mathbf{n} = \mathbf{n}(\mathbf{x})$  to the surface S,

$$\mathbf{n} \cdot \mathbf{u} = \mathbf{n} \cdot \nabla \phi = 0 \qquad ; \qquad \forall \mathbf{x} \in S \tag{3}$$

## 2 Solution

### 2.1 Helmholtz decomposition

Using Helmholtz decomposition, we get

$$\phi(\mathbf{x},t) = \frac{1}{4\pi} \int_{\partial V_i} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \phi(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dS' - \frac{1}{4\pi} \int_{V_i} \frac{\nabla'^2 \phi(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dV' \quad ; \quad \forall \mathbf{x} \in V_i \quad (4)$$

for any compact domain  $V_i \subset \mathbb{R}^3$  with a piecewise smooth boundary  $\partial V_i$ , where the primed symbols above are with respect to the new  $\mathbf{x}'$  dummy variables and  $\mathbf{n}$  is the outward pointing unit normal vector to  $\partial V_i$ .

To extend this result to exterior domains, we define  $V_e = V - (V_i - \partial V_i)$  for some compact domain  $V \subset \mathbb{R}^3$  having a piecewise smooth boundary  $\partial V$  and then take the limit as  $V \to \mathbb{R}^3$ .

We first let  $V_e = V_{e1} \cup V_{e2}$ , where  $V_{e1} \cap V_{e2}$  is a single piecewise smooth surface containing exactly one hole and satisfying  $(V_{e1} \cap V_{e2}) \cap \partial V_i \neq \emptyset$ . Applying (4) to both  $V_{e1}$  and  $V_{e2}$  followed by summing the results leads to

$$\phi(\mathbf{x},t) = -\frac{1}{4\pi} \int_{\partial V_i} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \phi(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dS' + \frac{1}{4\pi} \int_{\partial V} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \phi(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dS' - \frac{1}{4\pi} \int_{V_e} \frac{\nabla'^2 \phi(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dV' \quad ; \quad \forall \mathbf{x} \in V_e$$

We then note that applying (2c) on  $\partial V$  as  $V \to \mathbb{R}^3$  leads to

$$\begin{split} \frac{1}{4\pi} \int_{\partial V} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \phi(\mathbf{x}',t)}{|\mathbf{x}-\mathbf{x}'|} dS' &= -\frac{1}{4\pi} \int_{\partial V_i} \mathbf{n}(\mathbf{x}') \cdot \frac{(\mathbf{u}_{\infty} - \mathbf{u}_{\infty})}{|\mathbf{x}-\mathbf{x}'|} dS' + \\ &+ \frac{1}{4\pi} \int_{\partial V} \mathbf{n}(\mathbf{x}') \cdot \frac{\mathbf{u}_{\infty}}{|\mathbf{x}-\mathbf{x}'|} dS' - \frac{1}{4\pi} \int_{V} \frac{\nabla' \cdot \mathbf{u}_{\infty}}{|\mathbf{x}-\mathbf{x}'|} dV' \end{split}$$

since the divergence of a constant vector field is always zero, meaning that

$$\phi(\mathbf{x},t) - \mathbf{u}_{\infty} \cdot \mathbf{x} = -\frac{1}{4\pi} \int_{\partial V_i} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \left(\phi(\mathbf{x}',t) - \mathbf{u}_{\infty} \cdot \mathbf{x}'\right)}{|\mathbf{x} - \mathbf{x}'|} dS' - \frac{1}{4\pi} \int_{V_o} \frac{\nabla'^2 \left(\phi(\mathbf{x}',t) - \mathbf{u}_{\infty} \cdot \mathbf{x}'\right)}{|\mathbf{x} - \mathbf{x}'|} dV' \quad ; \quad \forall \mathbf{x} \in V_e \quad (5)$$

## 2.2 Incompressible flow

For an incompressible flow,

$$\nabla \cdot \mathbf{u} = \nabla \cdot \nabla \phi = \nabla^2 \phi = 0 \tag{6}$$

turning (5) into

$$\phi(\mathbf{x},t) - \mathbf{u}_{\infty} \cdot \mathbf{x} = -\frac{1}{4\pi} \int_{\partial V_i} \mathbf{n}(\mathbf{x}') \cdot \frac{\nabla' \left(\phi(\mathbf{x}',t) - \mathbf{u}_{\infty} \cdot \mathbf{x}'\right)}{|\mathbf{x} - \mathbf{x}'|} dS' \quad ; \quad \forall \mathbf{x} \in V_e$$

Defining the source strength  $\sigma(\mathbf{x}, t) = \mathbf{n}(\mathbf{x}, t) \cdot \nabla \left( \phi(\mathbf{x}, t) - \mathbf{u}_{\infty} \cdot \mathbf{x} \right)$  and noting that  $\partial V_i = S$ , we write this as

$$\phi(\mathbf{x},t) = \mathbf{u}_{\infty} \cdot \mathbf{x} - \frac{1}{4\pi} \int_{S} \frac{\sigma(\mathbf{x}',t)}{|\mathbf{x} - \mathbf{x}'|} dS' \qquad ; \qquad \forall \mathbf{x} \in V_{e}$$

Applying (3) to the above equation and simplifying leads to

$$\mathbf{n}(\mathbf{x}) \cdot \int_{S} \sigma(\mathbf{x}', t) \frac{\mathbf{x}' - \mathbf{x}}{|\mathbf{x}' - \mathbf{x}|} dS' = 4\pi \mathbf{n}(\mathbf{x}) \cdot \mathbf{u}_{\infty} \implies \frac{\partial \sigma(\mathbf{x}, t)}{\partial t} = 0 \quad ; \quad \forall \mathbf{x} \in S \quad (7)$$