A Boundary Value Problem Approach to Autonomous Rendezvous in Industrial Human-Drone Interaction

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$\begin{array}{c} \text{Appendix A. BACKGROUND ON INDIRECT} \\ \text{METHOD} \end{array}$

Let $[t_0,t_f]$ denote the finite time horizon, where $t_0 \in \mathbb{R}$ is the initial time, $t_f \in \mathbb{R}$ is the final time, and $t \in [t_0,t_f]$ denotes the continuous time variable. Let \mathcal{U} denote the set of admissible control functions $u:[t_0,t_f] \to U \subseteq \mathbb{R}^m$, where U is a compact subset of the control space, and let \mathcal{S} denote the set of admissible state trajectories $s:[t_0,t_f] \to \mathbb{R}^n$ with initial condition $s(t_0)=s_0$. The mapping $f:\mathbb{R}^n \times \mathbb{R}^m \times \mathbb{R} \to \mathbb{R}^n$ defines the system dynamics, the mapping $\xi:\mathbb{R}^n \to \mathbb{R}^q$ defines the terminal constraints at time t_f , the function $L:\mathbb{R}^n \times \mathbb{R}^m \times \mathbb{R} \to \mathbb{R}$ defines the running cost, and the function $\Phi:\mathbb{R}^n \to \mathbb{R}$ defines the terminal cost.

The optimal control problem, defined over the finite time horizon $[t_0, t_f]$, consists in determining $u^*(\cdot) \in \mathcal{U}$ and the corresponding optimal state trajectory $s^*(\cdot) \in \mathcal{S}$ that minimize the performance index

$$J[u(\cdot)] = \Phi(s(t_f)) + \int_{t_0}^{t_f} L(s(t), u(t), t) dt, \quad (A.1)$$

subject to the dynamical system

$$\dot{s}(t) = f(s(t), u(t), t), \qquad s(t_0) = s_0,$$
 (A.2)

and the terminal equality constraints

$$\xi(s(t_f)) = 0. \tag{A.3}$$

According to the *Pontryagin Maximum Principle (PMP)* formulated in Kopp (1962), the Hamiltonian is introduced

$$H(s(t), u(t), \lambda(t), t) = L(s(t), u(t), t) + \lambda^{\top} f(s(t), u(t), t), \tag{A.4}$$

where $\lambda(t) \in \mathbb{R}^n$ is the costate vector associated with the state dynamics. The necessary conditions of optimality consist of the state dynamics

$$\dot{s}(t) = \frac{\partial H}{\partial \lambda}(s(t), u(t), \lambda(t), t), \qquad s(t_0) = s_0, \quad (A.5)$$

the costate dynamics

$$\dot{\lambda}(t) = -\frac{\partial H}{\partial s}(s(t), u(t), \lambda(t), t), \tag{A.6}$$

and the Hamiltonian minimization condition

$$u^{\star}(t) = \arg\min_{u(t) \in U} H(s(t), u(t), \lambda(t), t). \tag{A.7}$$

For fixed final time and terminal constraints, the terminal costate satisfies

$$\lambda(t_f) = \frac{\partial \Phi}{\partial s}(s(t_f)) + \left(\frac{\partial \xi}{\partial s}(s(t_f))\right)^{\top} \nu, \tag{A.8}$$

where $\nu \in \mathbb{R}^q$ is the multiplier vector associated with the terminal equality constraints. For free terminal state components, indexed by $i \in \mathcal{I} := \{1, \dots, I\}$, the corresponding costate components satisfy

$$\lambda_i(t_f) = 0, \quad \forall i \in \mathcal{I}.$$
 (A.9)

When the final time t_f is free, the Hamiltonian satisfies the additional transversality condition

$$H(s(t_f), u(t_f), \lambda(t_f), t_f) = 0.$$
 (A.10)

The coupled system composed of the state and costate equations, together with the Hamiltonian minimization and transversality conditions, defines a two-point Boundary Value Problem (BVP) (Avvakumov and Kiselev (2000)) in the unknown trajectories $s:[t_0,t_f]\to\mathbb{R}^n$, $\lambda:[t_0,t_f]\to\mathbb{R}^n$, and $u:[t_0,t_f]\to U$. The boundary specification is mixed in nature, with initial conditions imposed on the state vector at time t_0 and terminal conditions imposed on the costate vector and multipliers at time t_f . Hence, the BVP formulation constitutes the foundation of the indirect method (Wang (2009)), reformulating the optimal control problem as a differential system where the necessary conditions imposed at both the initial and final times.

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Appendix B. STATE-COSTATE DYNAMICS

According to the PMP, the necessary conditions of optimality for a continuous–time system with Hamiltonian $H^{\rm d}(s^{\rm d}(t),u^{\rm d}(t),\lambda^{\rm d}(t),t)$ consist of the canonical equations

$$\dot{s}^{\mathrm{d}}(t) = \frac{\partial H^{\mathrm{d}}}{\partial \lambda^{\mathrm{d}}} \left(s^{\mathrm{d}}(t), u^{\mathrm{d}}(t), \lambda^{\mathrm{d}}(t), t \right), \qquad s^{\mathrm{d}}(t_0) = \mathbf{s}_0^{\mathrm{d}},$$
(B.1)

$$\dot{\lambda}^{\mathrm{d}}(t) = -\frac{\partial H^{\mathrm{d}}}{\partial s^{\mathrm{d}}} \left(s^{\mathrm{d}}(t), u^{\mathrm{d}}(t), \lambda^{\mathrm{d}}(t), t \right), \tag{B.2}$$

together with the Hamiltonian minimization condition

$$u^{\mathbf{d},\star}(t) = \arg\min_{u^{\mathbf{d}}(t) \in \mathcal{U}} H^{\mathbf{d}}(s^{\mathbf{d}}(t), u^{\mathbf{d}}(t), \lambda^{\mathbf{d}}(t), t). \tag{B.3}$$

By substituting the explicit dynamics and cost function of the drone, the extended Hamiltonian reads

$$H^{\mathbf{d}}(s^{\mathbf{d}}(t), u^{\mathbf{d}}(t), t) = L^{\mathbf{d}}(s^{\mathbf{d}}(t), u^{\mathbf{d}}(t), t) + \lambda_{p}^{\mathbf{d}}(t)^{\top} v^{\mathbf{d}}(t) + \lambda_{v}^{\mathbf{d}}(t)^{\top} \left(-\frac{\mathbf{k}^{\mathbf{d}}}{\mathbf{m}^{\mathbf{d}}} v^{\mathbf{d}}(t) + \frac{1}{\mathbf{m}^{\mathbf{d}}} u^{\mathbf{d}}(t) \right),$$
(B.4)

where the running cost is

$$L^{d}(s^{d}(t), u^{d}(t), t) = \|u^{d}(t)\|_{R}^{2} + \|v^{d}(t)\|_{Q}^{2} + (v^{d}(t) - v_{\text{safe}}(t) \hat{v}^{d}(t))^{\top} S(v^{d}(t) - v_{\text{safe}}(t) \hat{v}^{d}(t)).$$
(B.5)

The canonical state equations follow from the partial derivative of the Hamiltonian with respect to the costate variables:

$$\frac{\partial H^{d}}{\partial \lambda^{d}(t)} = \begin{pmatrix} \frac{\partial H^{d}}{\partial \lambda^{d}_{p}(t)} \\ \frac{\partial H^{d}}{\partial \lambda^{d}_{v}(t)} \end{pmatrix} \Longrightarrow \begin{cases} \dot{p}^{d}(t) = v^{d}(t), \\ \dot{v}^{d}(t) = -\frac{k^{d}}{m^{d}} v^{d}(t) + \frac{1}{m^{d}} u^{d}(t). \end{cases}$$
(B.6)

The Hamiltonian minimization condition gives

$$\frac{\partial H^{d}}{\partial u^{d}(t)} = 0 \quad \Rightarrow \quad u^{d,\star}(t) = \operatorname{proj}_{\mathcal{U}}\left(-\frac{1}{2m^{d}} R^{-1} \lambda_{v}^{d}(t)\right), \tag{B.}$$

yielding the explicit state dynamics

$$\dot{p}^{\rm d}(t) = v^{\rm d}(t), \qquad \dot{v}^{\rm d}(t) = -\frac{{\rm k}^{\rm d}}{{\rm m}^{\rm d}} v^{\rm d}(t) + u^{\rm d,\star}(t). \quad (B.8)$$

The costate dynamics is obtained from the negative partial derivative of the Hamiltonian with respect to the state variables:

$$\dot{\lambda}^{\mathrm{d}}(t) = -\left(\frac{\partial H^{\mathrm{d}}}{\partial p^{\mathrm{d}}(t)} \frac{\partial H^{\mathrm{d}}}{\partial p^{\mathrm{d}}(t)}\right). \tag{B.9}$$

Since $H^{\rm d}(\cdot)$ depends on the drone position $p^{\rm d}(t)$ only through the safety term $v_{\rm safe}(t) = \tilde{v}_{\rm safe}(p^{\rm d}(t), p^{\rm h}(t))$, its partial derivative with respect to position can be expressed as

$$\frac{\partial H^{\rm d}}{\partial p^{\rm d}(t)} = \frac{\partial L}{\partial v_{\rm safe}(t)} \frac{\partial v_{\rm safe}(t)}{\partial p^{\rm d}(t)}, \quad \frac{\partial v_{\rm safe}(t)}{\partial p^{\rm d}(t)} = \tilde{v}'_{\rm safe}(d(t)) \, \hat{r}(t), \tag{B.10}$$

where the unit relative position vector is defined as $\hat{r}(t) = \frac{p^{\rm d}(t) - {\bf p}^{\rm h}(t)}{\|p^{\rm d}(t) - {\bf p}^{\rm h}(t)\|}$, and points from the human operator toward the drone and $\tilde{v}'_{\rm safe}(d(t))$ is equal to

$$\tilde{v}'_{\text{safe}}(d(t)) = \begin{cases} 0, & d(t) \le d_{\text{stop}}, \\ \frac{v_{\text{cruise}}}{d_{\text{cruise}} - d_{\text{stop}}}, & d_{\text{stop}} < d(t) < d_{\text{cruise}}, \\ 0, & d(t) \ge d_{\text{cruise}}. \end{cases}$$
(B.11)

Let $e(t) = v^{\rm d}(t) - v_{\rm safe}(t) \, \hat{v}^{\rm d}(t)$, where $\hat{v}^{\rm d}(t) = v^{\rm d}(t) / \|v^{\rm d}(t)\|$ represents the instantaneous heading direction of the drone. The derivative of the running cost $L^{\rm d}$ with respect to $v_{\rm safe}(t)$ is then

$$\frac{\partial L}{\partial v_{\text{safe}}(t)} = -2\,\hat{v}^{\text{d}}(t)^{\mathsf{T}} \mathbf{S} \big(v^{\text{d}}(t) - v_{\text{safe}}(t)\,\hat{v}^{\text{d}}(t) \big), \quad (B.12)$$

which, when combined with the chain rule above, gives

$$\frac{\partial H^{\mathbf{d}}}{\partial p^{\mathbf{d}}(t)} = -2 \,\tilde{v}_{\text{safe}}'(d(t)) \,\hat{r}(t) \,\hat{v}^{\mathbf{d}}(t)^{\mathsf{T}} \mathbf{S} \big(v^{\mathbf{d}}(t) - v_{\text{safe}}(t) \,\hat{v}^{\mathbf{d}}(t) \big). \tag{B.13}$$

Thus, the position costate dynamics is defined as

$$\dot{\lambda}_p^{\mathbf{d}}(t) = 2\,\tilde{v}_{\text{safe}}'(d(t))\,\hat{r}(t)\,\hat{v}^{\mathbf{d}}(t)^{\mathsf{T}}\mathbf{S}\big(v^{\mathbf{d}}(t) - v_{\text{safe}}(t)\,\hat{v}^{\mathbf{d}}(t)\big). \tag{B.14}$$

For the velocity component, the Hamiltonian depends explicitly and implicitly on $v^{\rm d}(t)$ through the running cost $L^{\rm d}$ and the dynamic term. The partial derivative is

$$\frac{\partial H^{\rm d}}{\partial v^{\rm d}(t)} = \frac{\partial L^{\rm d}}{\partial v^{\rm d}(t)} + \lambda_p^{\rm d}(t) - \frac{\mathbf{k}^{\rm d}}{\mathbf{m}^{\rm d}} \, \lambda_v^{\rm d}(t), \tag{B.15}$$

which leads to the costate dynamics

$$\dot{\lambda}_{v}^{\mathrm{d}}(t) = \frac{\mathrm{k}^{\mathrm{d}}}{\mathrm{m}^{\mathrm{d}}} \lambda_{v}^{\mathrm{d}}(t) - \lambda_{p}^{\mathrm{d}}(t) - \frac{\partial L^{\mathrm{d}}}{\partial v^{\mathrm{d}}(t)}.$$
 (B.16)

Considering

$$L^{d} = \|u^{d}(t)\|_{\mathbf{R}}^{2} + \|v^{d}(t)\|_{\mathbf{Q}}^{2} + e(t)^{\mathsf{T}} \mathbf{S}e(t),$$
 (B.17)

the derivative of $L^{\rm d}$ with respect to $v^{\rm d}(t)$ reads

$$\frac{\partial L^{\rm d}}{\partial v^{\rm d}(t)} = 2Q v^{\rm d}(t) + 2S e(t) \frac{\partial e(t)}{\partial v^{\rm d}(t)}. \tag{B.18}$$

Since e(t) depends on $v^{\rm d}(t)$ both directly and through $\hat{v}^{\rm d}(t)$, its derivative is

$$\frac{\partial e(t)}{\partial v^{\mathbf{d}}(t)} = \mathbf{I} - v_{\text{safe}}(t) \frac{\partial \hat{v}^{\mathbf{d}}(t)}{\partial v^{\mathbf{d}}(t)},
\frac{\partial \hat{v}^{\mathbf{d}}(t)}{\partial v^{\mathbf{d}}(t)} = \frac{1}{\|v^{\mathbf{d}}(t)\|} (\mathbf{I} - \hat{v}^{\mathbf{d}}(t) \, \hat{v}^{\mathbf{d}}(t)^{\mathsf{T}}).$$
(B.19)

Assuming that $v^{\rm d}(t)$ varies slowly so that its direction changes negligibly, the operator $({\rm I}-\hat{v}^{\rm d}\hat{v}^{\rm d}^{\rm T})$ acts as a projection onto the plane orthogonal to $\hat{v}^{\rm d}(t)$. Hence,

$$(\mathbf{I} - \hat{v}^{\mathbf{d}} \hat{v}^{\mathbf{d} \top}) \left(v^{\mathbf{d}}(t) - v_{\text{safe}}(t) \, \hat{v}^{\mathbf{d}}(t) \right) \simeq 0, \tag{B.20}$$

which implies that only the component of $v^{\rm d}(t)$ along its heading direction is relevant in the derivative. Under this assumption, (B.20) simplifies to the approximate form

$$\frac{\partial L^{\mathrm{d}}}{\partial v^{\mathrm{d}}(t)} \simeq 2(\mathrm{Q} + \mathrm{S}) v^{\mathrm{d}}(t) - 2\mathrm{S} v_{\mathrm{safe}}(t) \hat{v}^{\mathrm{d}}(t). \tag{B.21}$$

Since $v_{\text{safe}}(t)$ depends only on the drone position and not explicitly on the velocity, no additional coupling term appears in the velocity costate dynamics. Substituting the expression of $\frac{\partial L^{\text{d}}}{\partial v^{\text{d}}(t)}$ into the canonical costate equation yields

$$\dot{\lambda}_v^{\rm d}(t) = \frac{\mathbf{k}^{\rm d}}{\mathbf{m}^{\rm d}}\,\lambda_v^{\rm d}(t) - \lambda_p^{\rm d}(t) - 2(\mathbf{Q} + \mathbf{S})\,v^{\rm d}(t) + 2\,\mathbf{S}\,v_{\rm safe}(t)\,\hat{v}^{\rm d}(t). \tag{B.22} \label{eq:delta_v_$$

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