



UNIVERSITÀ  
DI TRENTO  
Department of  
Industrial Engineering

# Master Thesis

7<sup>th</sup> Update

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# Impact Model

- Assumptions:
  - although the generalized velocities change substantially, the generalized coordinates of the system remain the same over the impact duration
  - at the contact point between the end-effector and the target there are forces but not moments. Impact occurs at a single point, which is unable to transmit a local moment

# Impact Model

$$\begin{cases} M\ddot{p} + C = u + J^T f_I \\ M_o \ddot{\psi} + C_o = -J_o^T f_I \end{cases}$$

where  $J^T f_I$  and  $J_o^T f_I$  are the torques and forces applied to the VMS and object's generalized coordinates respectively

$$f_I = - (J_o^T)^+ (M_o \ddot{\psi}_o + C_o)$$

$$M(\dot{p}_f - \dot{p}_i) + J^T (J_o^+)^T M_o (\dot{\psi}_f - \dot{\psi}_i) = 0$$

# Impact Model

$$J\dot{p}_f = J_O\dot{\psi}_f \Rightarrow \dot{\psi}_f = J_O^+ J\dot{p}_f$$

$$\dot{p}_f = G^{-1}H$$

$$\text{where } \begin{cases} G = M + J^T(J_O^+)^T M_O J_O^+ J \\ H = M\dot{p}_i + J^T(J_O^+)^T M_O \dot{\psi}_i \end{cases}$$

Differentiate  $\dot{\psi}_f = J_O^+ J\dot{p}_f$  and substitute in  $M_O\ddot{\psi} + C_O = -J_O^T f_I$

# Impact Model

Final VMS dynamics with payload at the EE position:

$$M'\ddot{p} + C' = u$$

$$\text{where } \begin{cases} M' = M + J^T(J_O^T)^+ M_O J_O^+ J \\ C' = C + J^T(J_O^T)^+ M_O \frac{\partial J_O^+}{\partial t} J \dot{p} + J^T(J_O^T)^+ M_O J^+ \frac{\partial J}{\partial t} \dot{p} + J^T(J_O^T)^+ C_O \end{cases}$$

# Controlled based motion

Equation of motion can be rewritten:

$$\begin{bmatrix} M'_{tt} & M'_{tr} \\ M'_{rt} & M'_{rr} \end{bmatrix} \begin{bmatrix} \ddot{p}_t \\ \ddot{p}_r \end{bmatrix} + \begin{bmatrix} C'_t \\ C'_r \end{bmatrix} = \begin{bmatrix} \overline{0} \\ u \end{bmatrix}$$

$$\ddot{p}_t = -M_{tt}^{-1'}(M'_{tr}\ddot{p}_r + C'_t)$$

$$\ddot{p}_r \tilde{M} + \tilde{C} = u$$

# Controlled based motion

$$\ddot{p}_r \tilde{M} + \tilde{C} = u$$

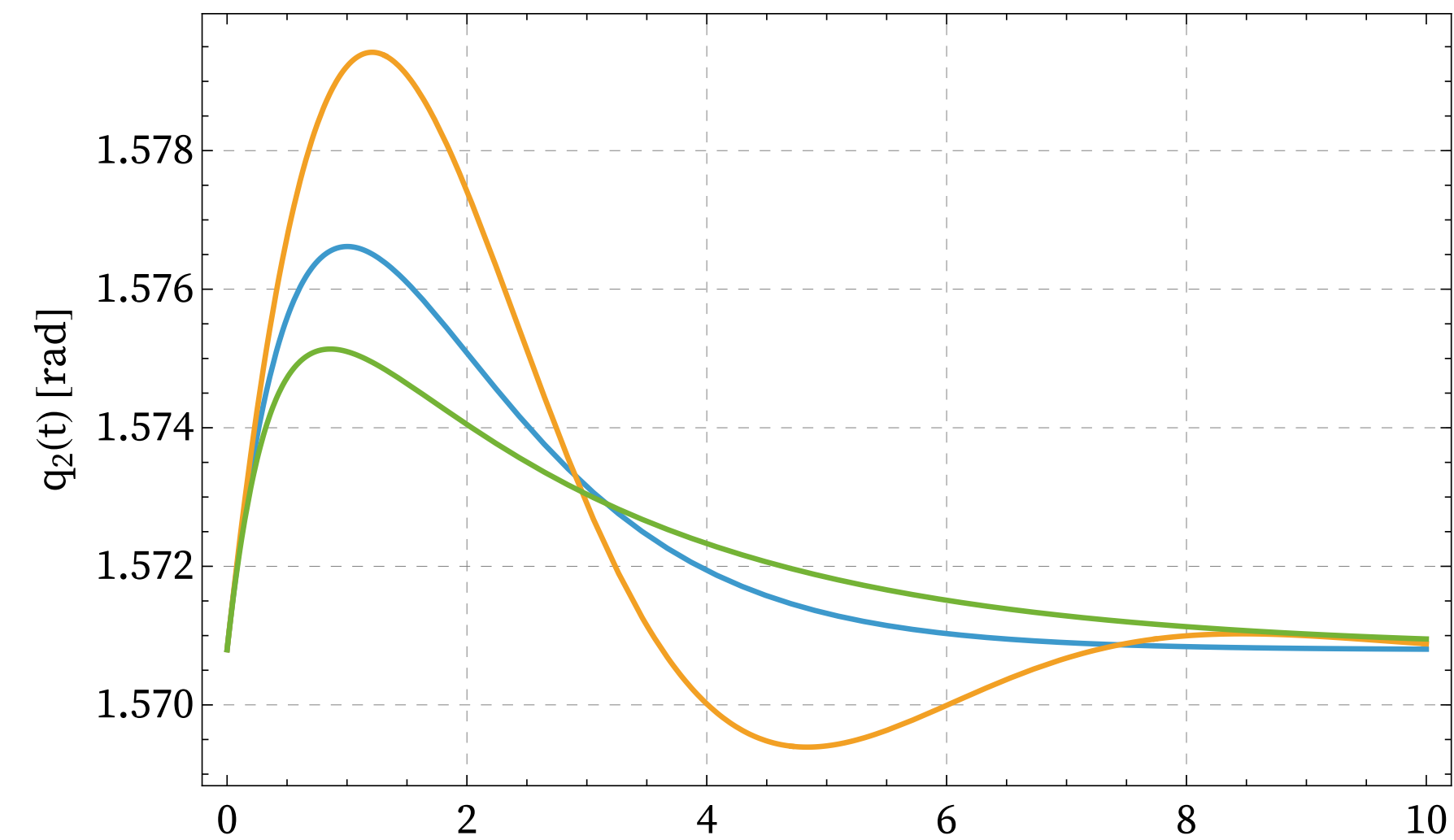
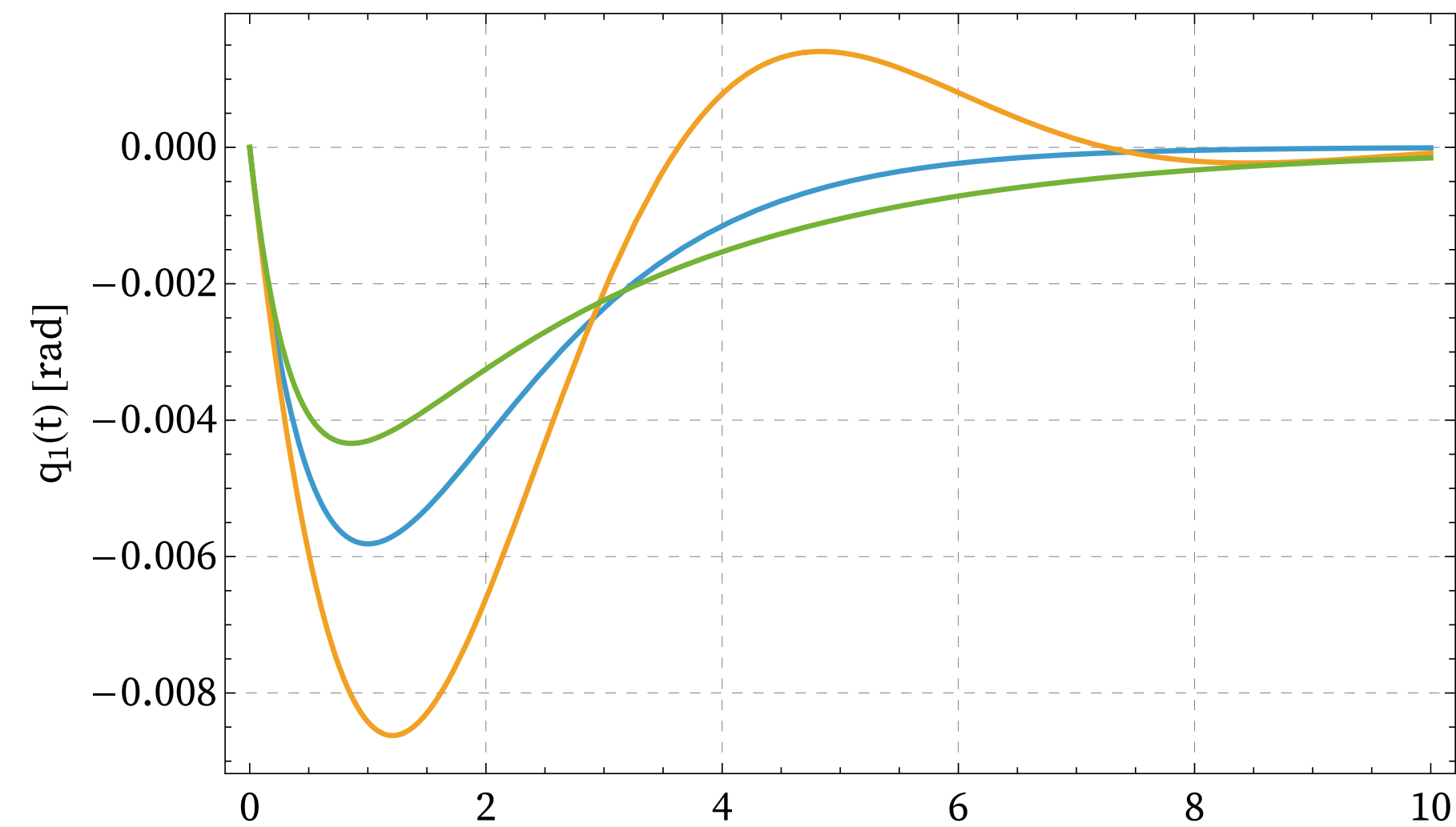
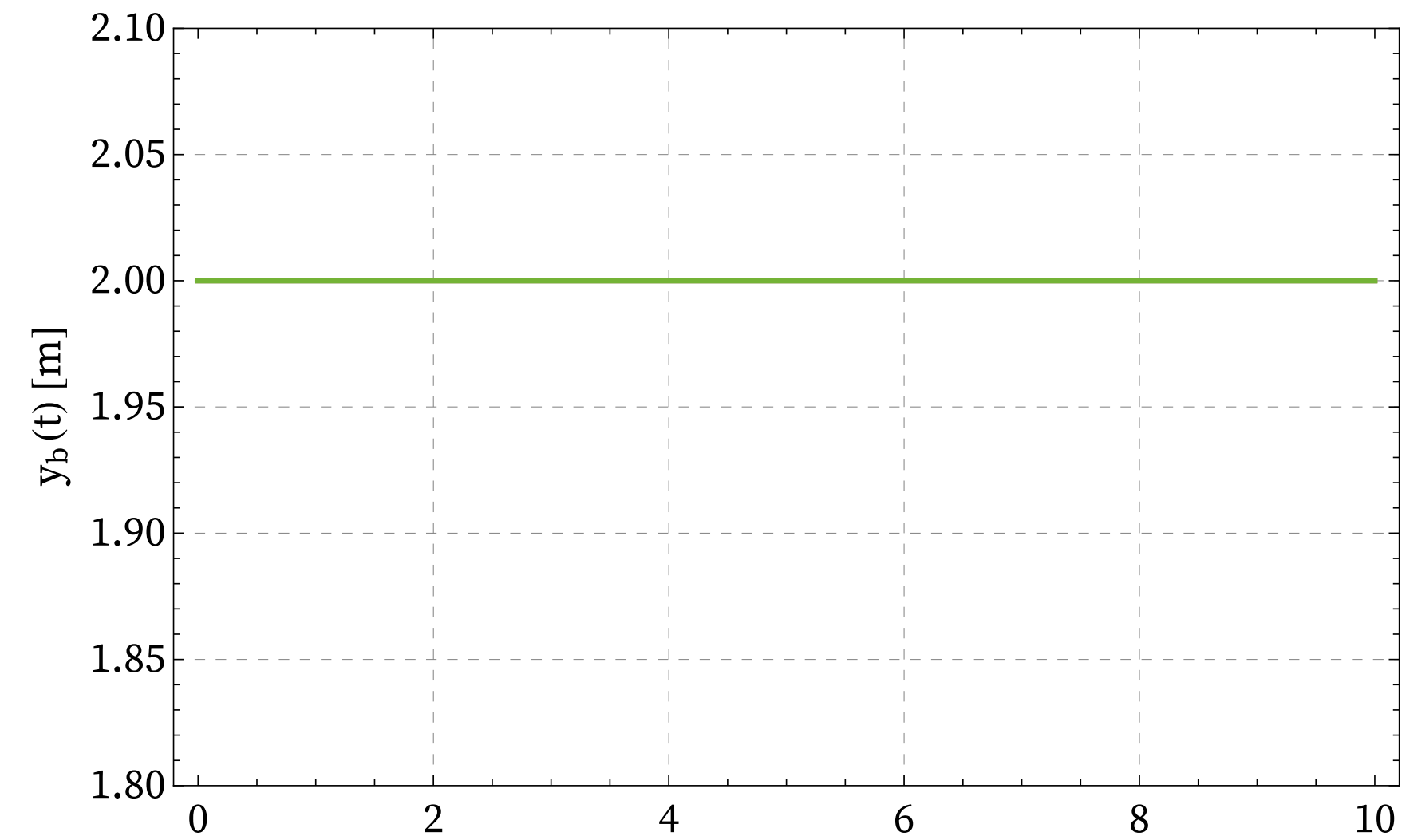
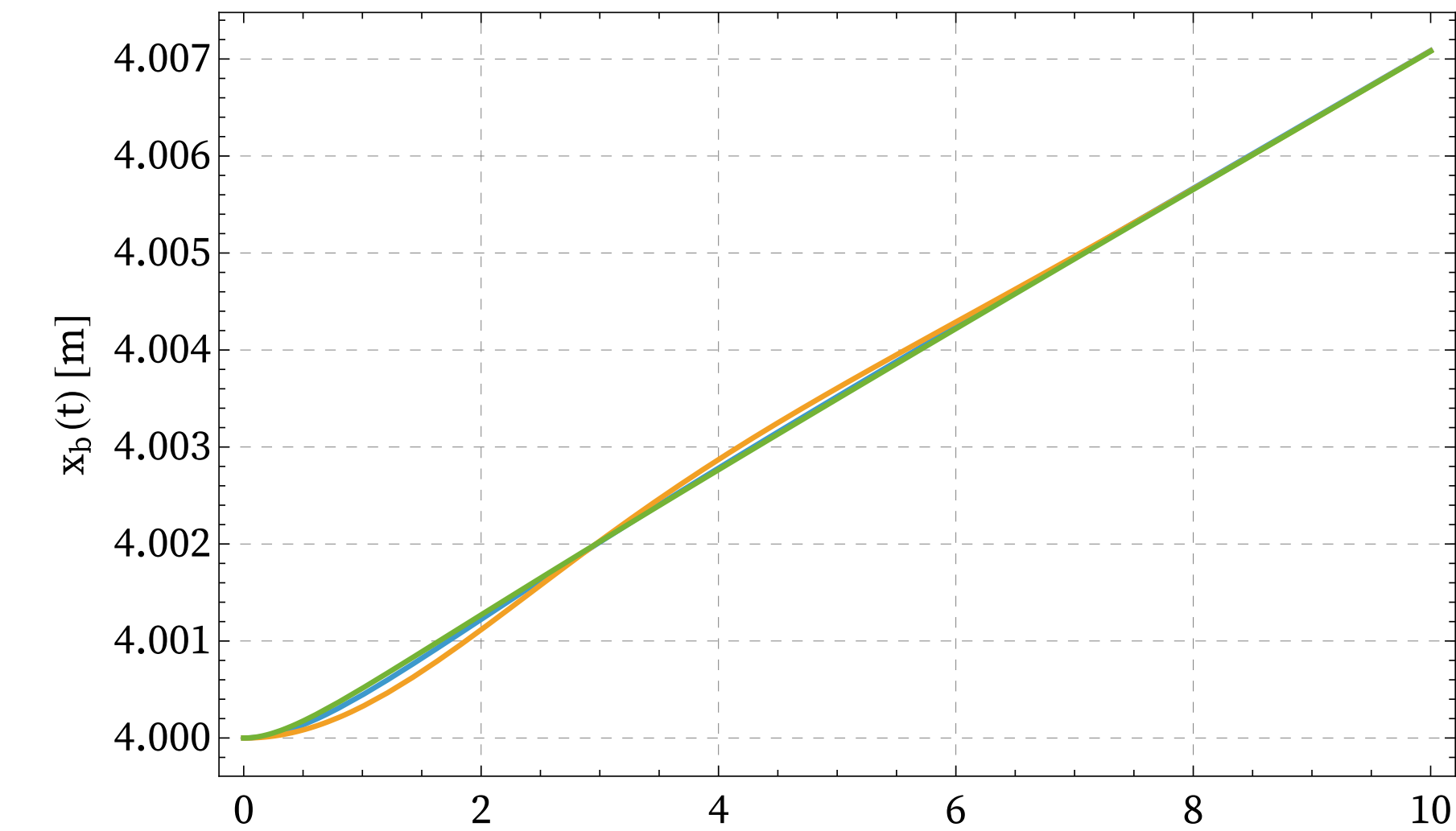
$$u = \tilde{M}[K_d(\dot{q}_d - \dot{q}) + K_p(q_d - q)] + \tilde{C}$$

$$\ddot{e} + K_d \dot{e} + K_p e = 0$$

$$\text{with } e = q - q_d, \omega_n = \sqrt{K_{p_i}} \text{ and } \xi = \frac{K_{d_i}}{2\sqrt{K_{p_i}}}$$

$$\xi = 1 \quad \Rightarrow \quad K_d = 2\sqrt{K_p}$$

# Controlled based motion





# Linearized equations

Initial equation:  $M'\ddot{p} + C' = u$

Linearize equations:

$M_{lin}\ddot{q} + C_{lin}\dot{q} + K_{lin}q = cost$ , constant given by desired position

Eigenvalue problem:

$$\{K_{lin} - \omega^2 M_{lin}\} \{u_1, u_2\}^T$$

$M_{lin}$  and  $K_{lin}$  don't depend on  $q_{1,0}$ :

$$M_{lin}(q_{2,0}), K_{lin}(q_{2,0}, m_O)$$

# Matrices

- When we keep the same initial conditions for  $M_{lin}$  and  $K_{lin}$ , if

$$m_O = m \Rightarrow M_{lin} = K_{lin} \forall q_{2,0}:$$

```
= Chop@FullSimplify[Mcoupled] /. q200 -> Pi/3 // MatrixForm
```

```
/MatrixForm=
( 295 620.  142 135. )
( 142 135.  94 235.9 )
```

```
= Chop@FullSimplify[Mcoupled] /. q200 -> Pi/3 // MatrixForm
```

```
/MatrixForm=
( 295 620.  142 135. )
( 142 135.  94 235.9 )
```

```
= Chop@FullSimplify[Kcoupled /. m0 -> 2000 /. q200 -> Pi/3 /. q100 -> 0] // MatrixForm
```

```
/MatrixForm=
( 203 850.  96 579.5 )
( 96 579.5  63 865.6 )
```

```
= Chop@FullSimplify[Kcoupled /. m0 -> 3000 /. q200 -> Pi/3 /. q100 -> 0] // MatrixForm
```

```
/MatrixForm=
( 295 620.  142 135. )
( 142 135.  94 235.9 )
```

- When we set  $q_{2,0} = \pi/2$ , the second row has the same values:

```
Chop@FullSimplify[Mcoupled] /. q200 -> q20 // MatrixForm
```

```
MatrixForm=
( 200 479.  94 235.9 )
( 94 235.9  94 235.9 )
```

```
Chop@FullSimplify[Mcoupled] /. q200 -> Pi/2 // MatrixForm
```

```
MatrixForm=
( 200 479.  94 235.9 )
( 94 235.9  94 235.9 )
```

```
Chop@FullSimplify[Kcoupled /. m0 -> 4000 /. q200 -> Pi/2 /. q100 -> q10] // MatrixForm
```

```
MatrixForm=
( 262 098.  124 606. )
( 124 606.  124 606. )
```

```
Chop@FullSimplify[Kcoupled /. m0 -> 3000 /. q200 -> Pi/2 /. q100 -> 0] // MatrixForm
```

```
MatrixForm=
( 200 479.  94 235.9 )
( 94 235.9  94 235.9 )
```

# Matrices

- When the mass is correct ( $m_O = m \Rightarrow M_{lin} = K_{lin}$ ):

$$\begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix} - \omega^2 \begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \text{ with } m_{i,j} = k_{i,j} = x_{i,j}$$

$$\Rightarrow \det \left( (1 - \omega^2) \begin{bmatrix} x_{11} & x_{12} \\ x_{21} & x_{22} \end{bmatrix} \right) = 0$$

$$\omega_{1,2} = 1$$

$$U = \begin{bmatrix} ind & ind \\ ind & ind \end{bmatrix}$$

# Matrices

- When the mass is incorrect ( $m_O \neq m \Rightarrow M_{lin} \neq K_{lin}$ ) and  $q_{2,0} = \pi/2$ :

$$\begin{bmatrix} k_1 & k_2 \\ k_2 & k_2 \end{bmatrix} - \omega^2 \begin{bmatrix} m_1 & m_2 \\ m_2 & m_2 \end{bmatrix} \text{ with } m_i \neq k_i$$

$$\Rightarrow \det \left( \begin{bmatrix} k_1 - \omega^2 m_1 & k_2 - \omega^2 m_2 \\ k_2 - \omega^2 m_2 & k_2 - \omega^2 m_2 \end{bmatrix} \right) = 0$$

$$\begin{cases} \omega_1 = \sqrt{\frac{k_2}{m_2}} \\ \omega_2 = \sqrt{\frac{k_2 - k_1}{m_2 - m_1}} \end{cases} \Rightarrow \begin{bmatrix} k_1 - \frac{k_2}{m_2} m_1 & 0 \\ 0 & 0 \end{bmatrix} \text{ and } \begin{bmatrix} \frac{k_1 m_2 - k_2 m_1}{m_2 - m_1} & \frac{k_1 m_2 - k_2 m_1}{m_2 - m_1} \\ \frac{k_1 m_2 - k_2 m_1}{m_2 - m_1} & \frac{k_1 m_2 - k_2 m_1}{m_2 - m_1} \end{bmatrix}$$

# Matrices

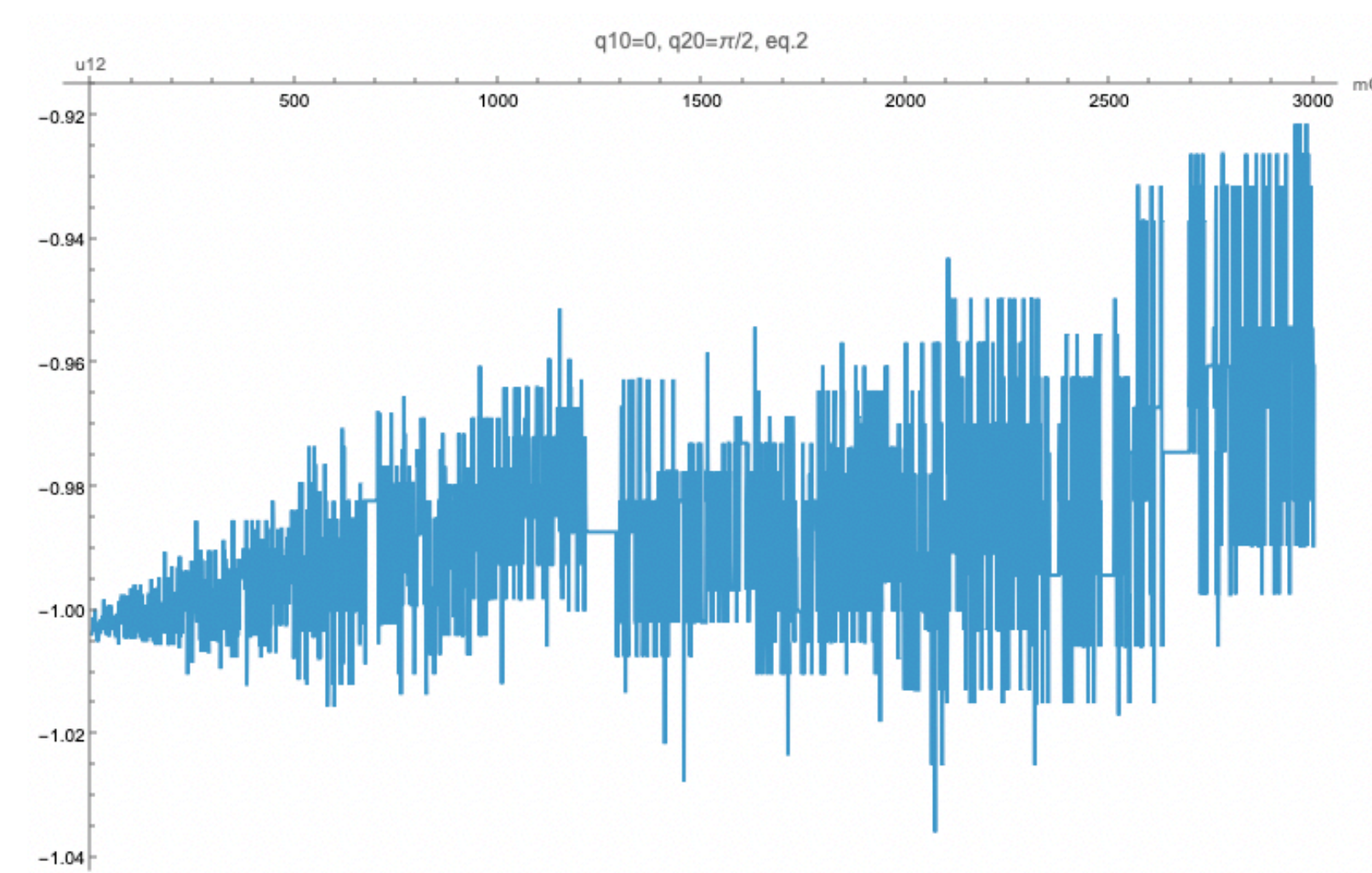
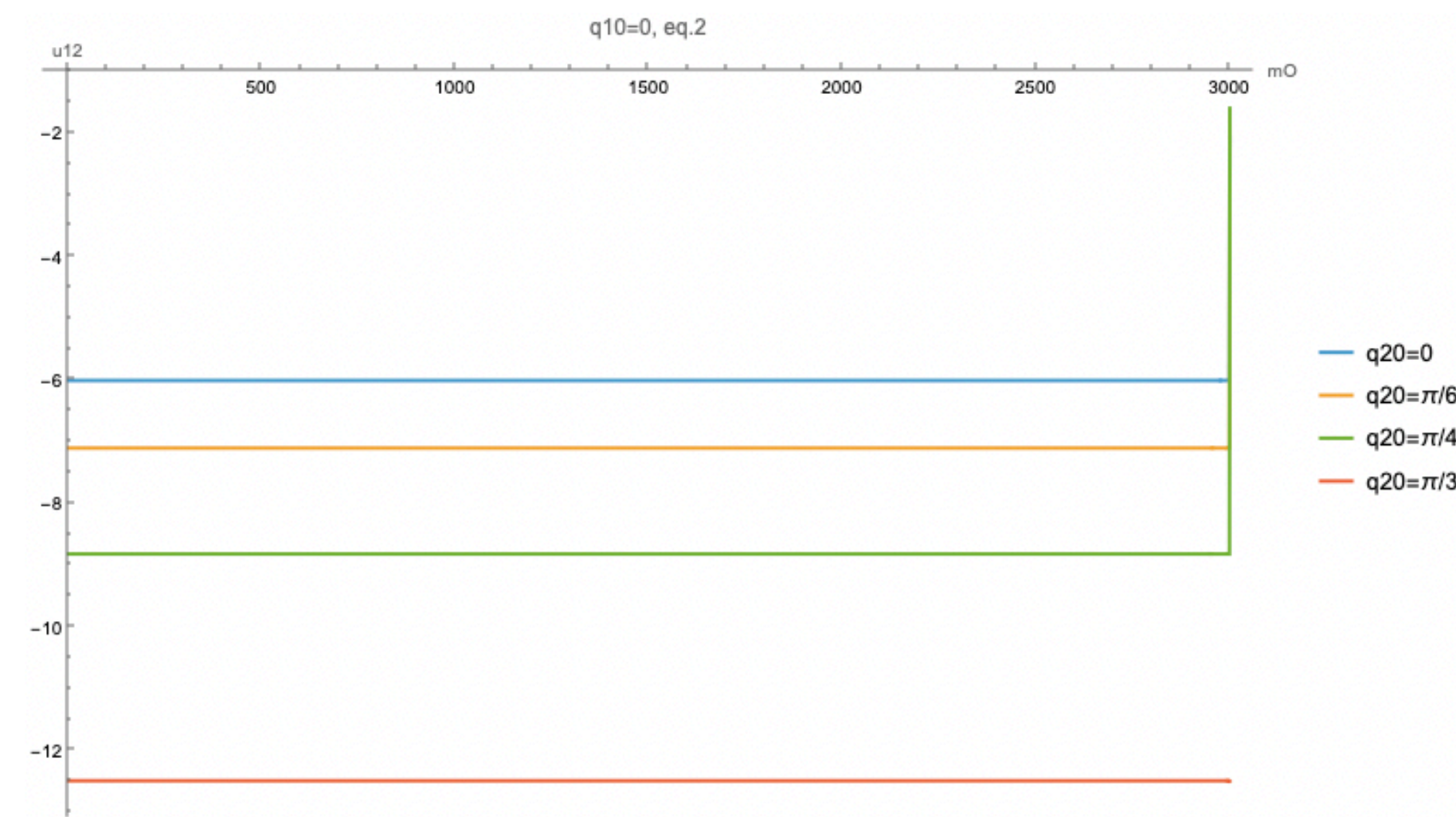
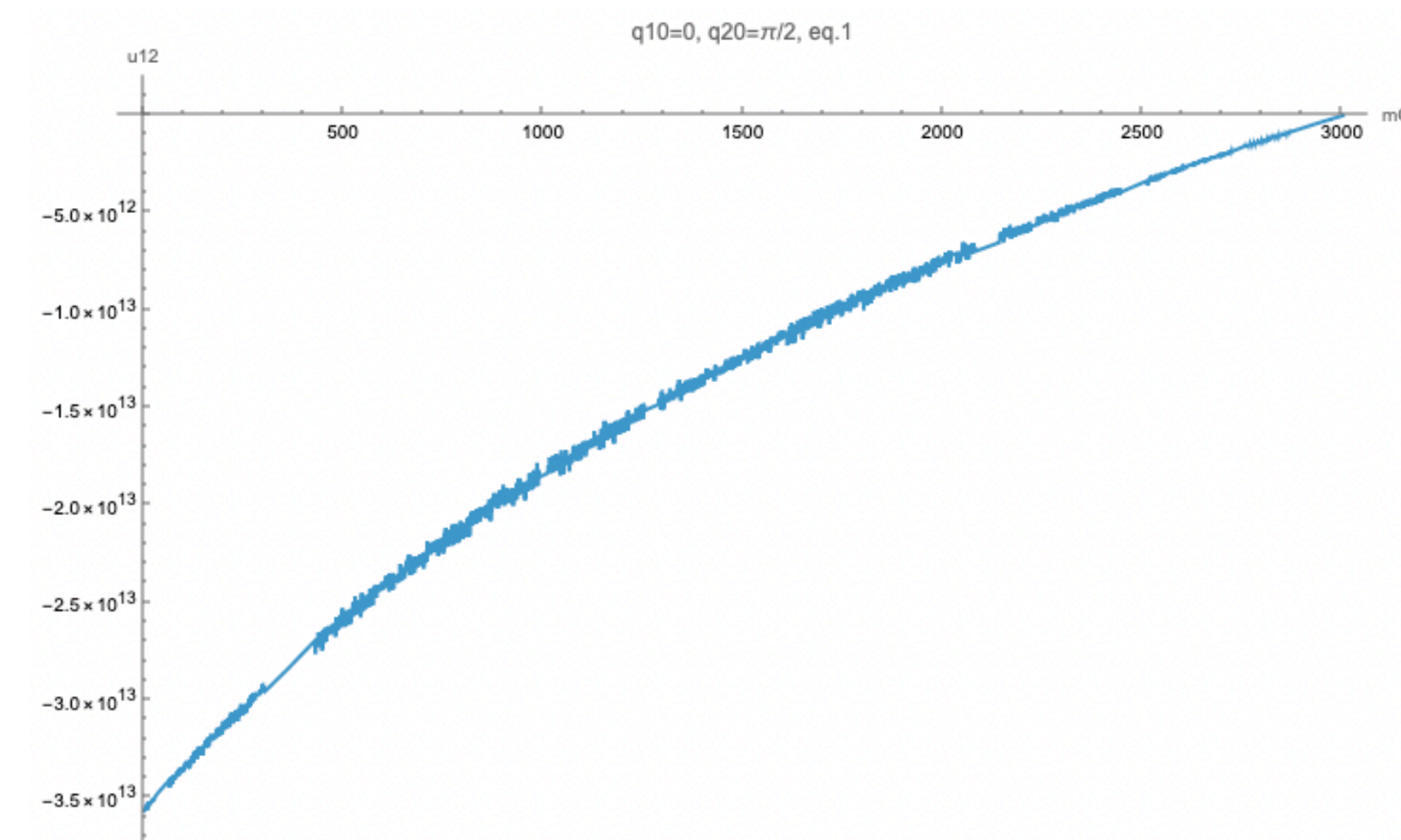
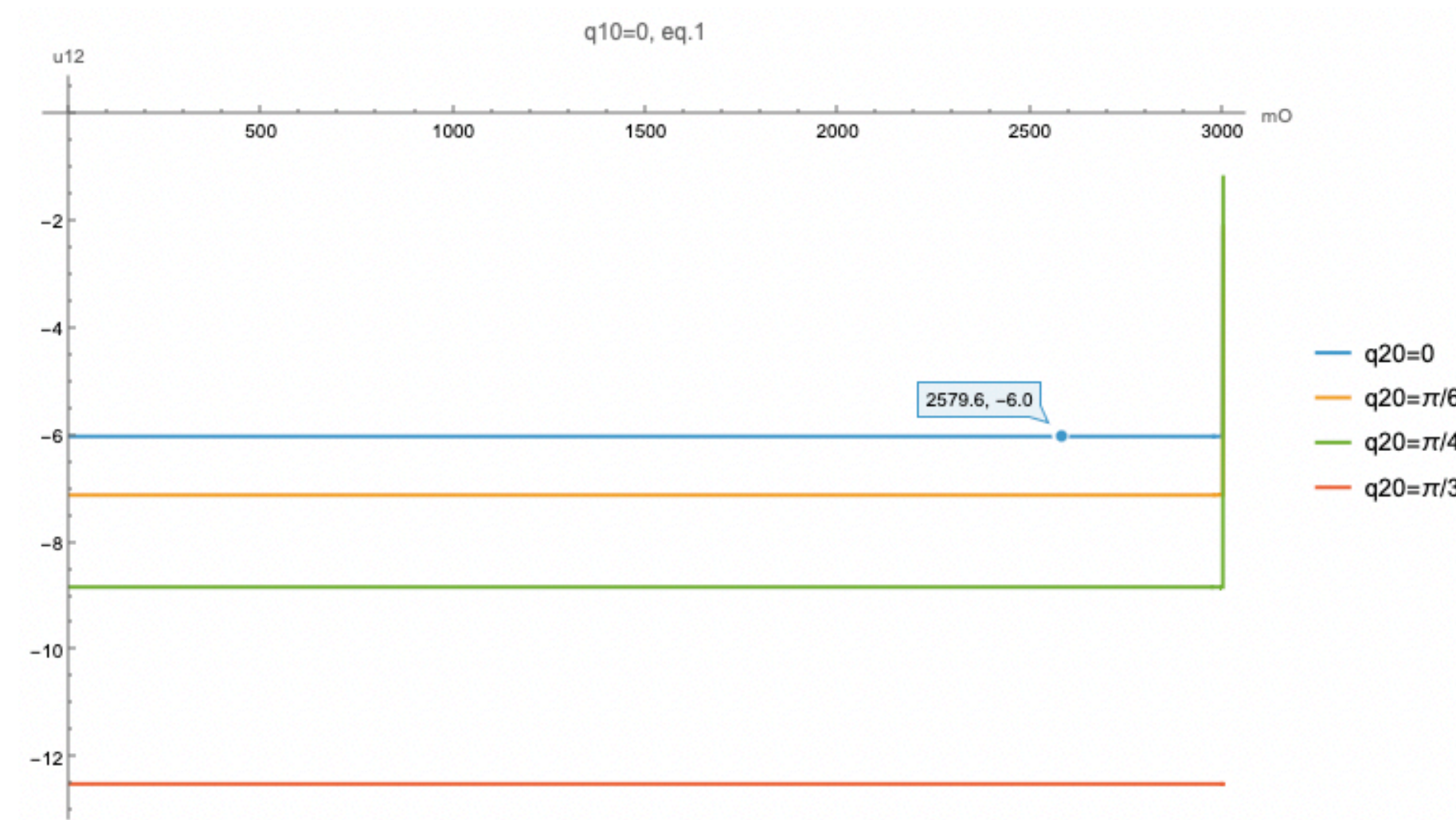
- When the mass is incorrect ( $m_O \neq m \Rightarrow M_{lin} \neq K_{lin}$ ) and  $q_{2,0} = \pi/2$ :

$$\begin{bmatrix} k_1 - \frac{k_2}{m_2}m_1 & 0 \\ 0 & 0 \end{bmatrix} \text{ and } \begin{bmatrix} \frac{k_1m_2 - k_2m_1}{m_2 - m_1} & \frac{k_1m_2 - k_2m_1}{m_2 - m_1} \\ \frac{k_1m_2 - k_2m_1}{m_2 - m_1} & \frac{k_1m_2 - k_2m_1}{m_2 - m_1} \end{bmatrix}$$

$$U = \begin{bmatrix} 0 & 1 \\ ind & -1 \end{bmatrix} \text{ or } U = \begin{bmatrix} ind & 1 \\ ind & -1 \end{bmatrix}$$

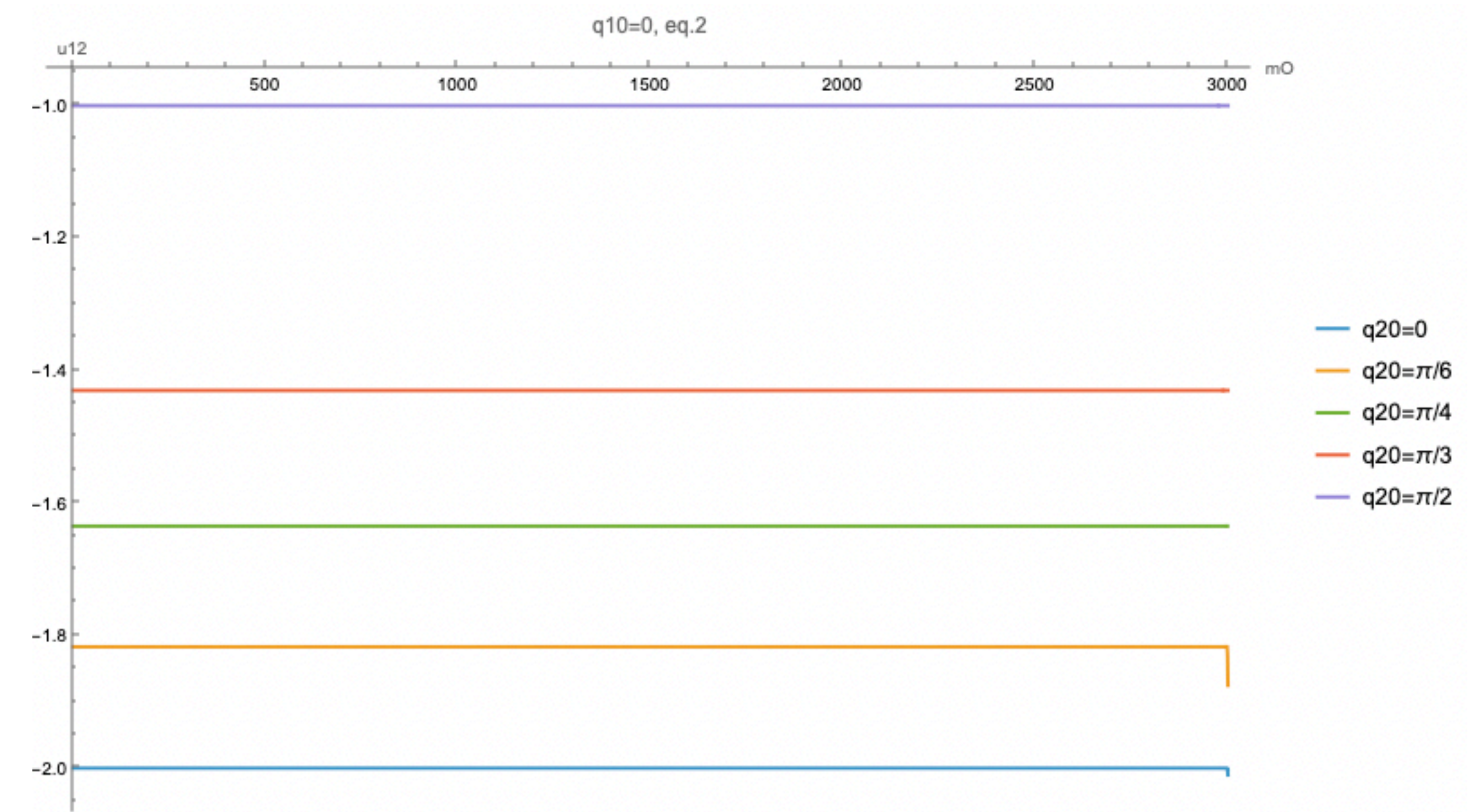
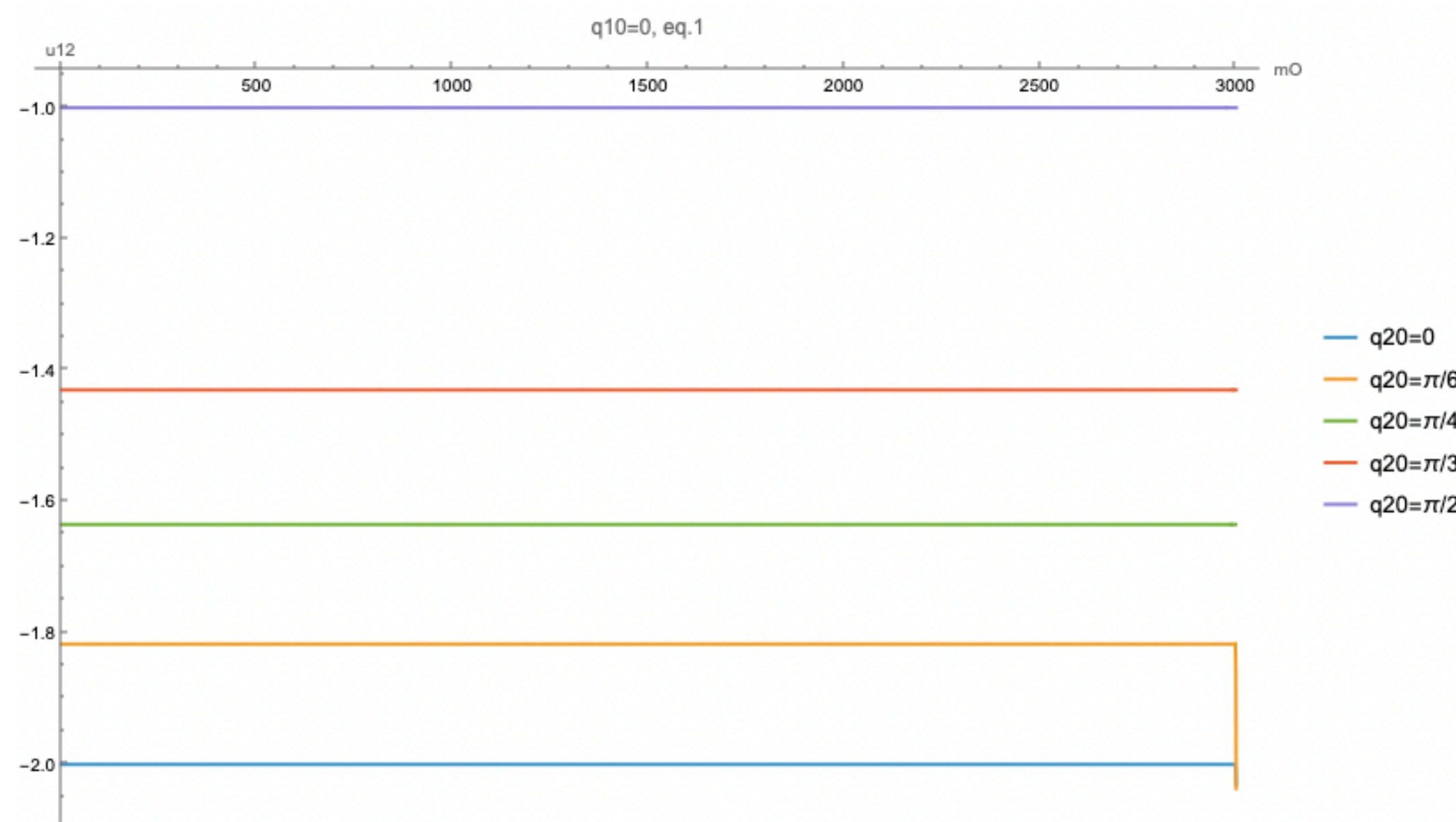


# First frequency





# Second frequency



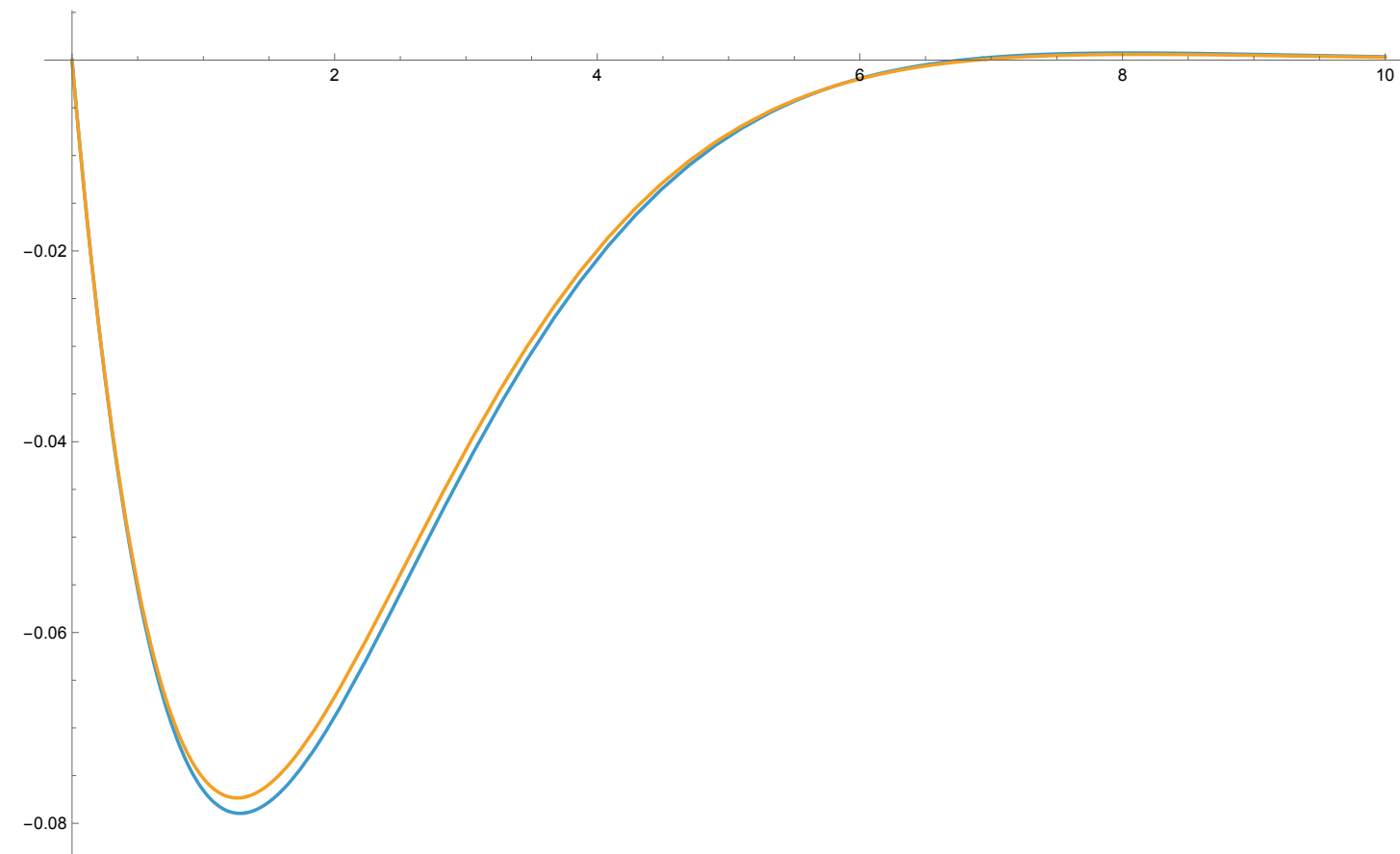
# Real dynamics vs Approximated One

Assumption: we can neglect the other arm's contribution in the linearized equation:

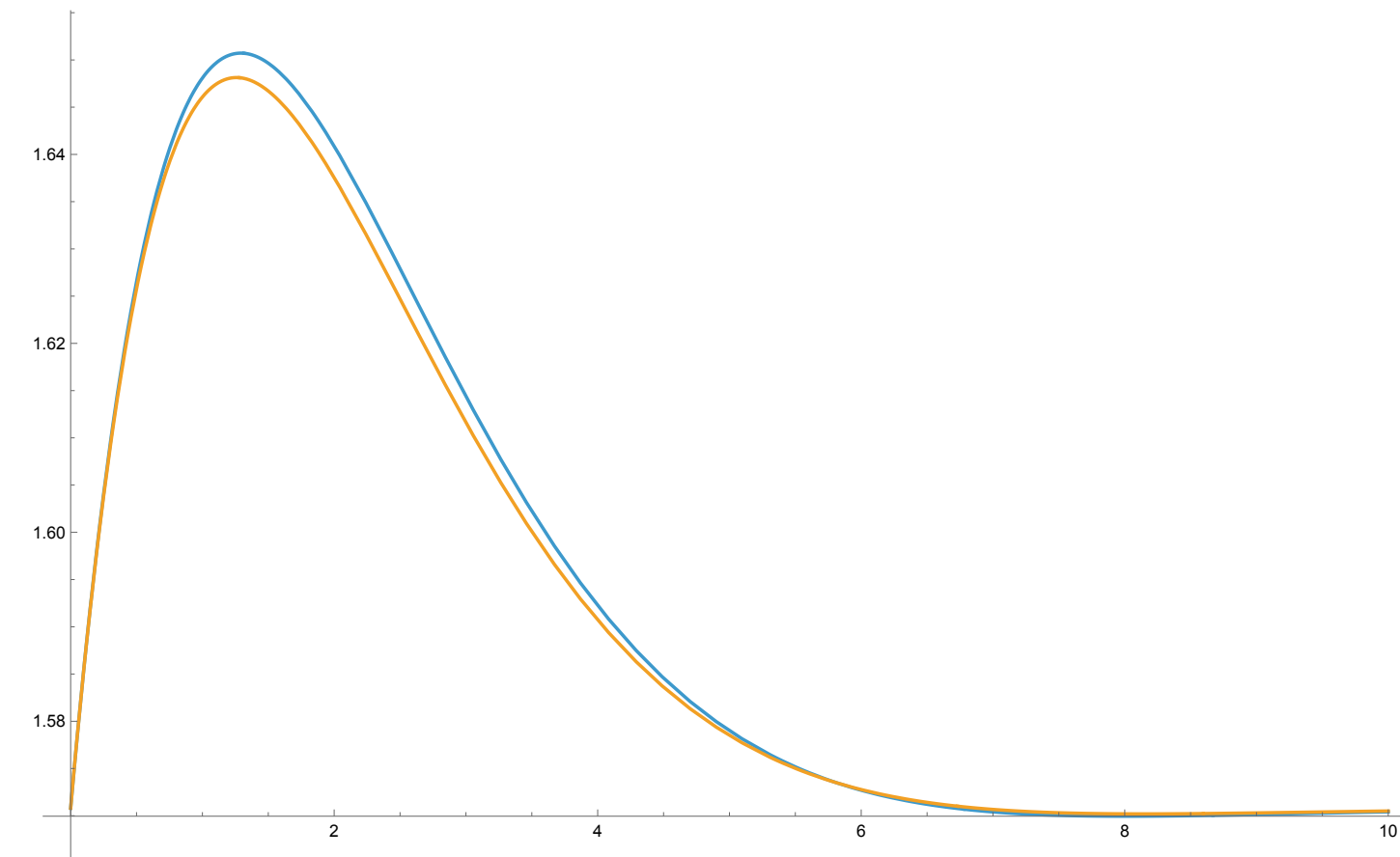
$$\begin{cases} q_2(t) = q_{2,0}, \dot{q}_2(t) = 0, \ddot{q}_2(t) = 0 & \text{for } q_1 \\ q_1(t) = q_{1,0}, \dot{q}_1(t) = 0, \ddot{q}_1(t) = 0 & \text{for } q_2 \end{cases}$$



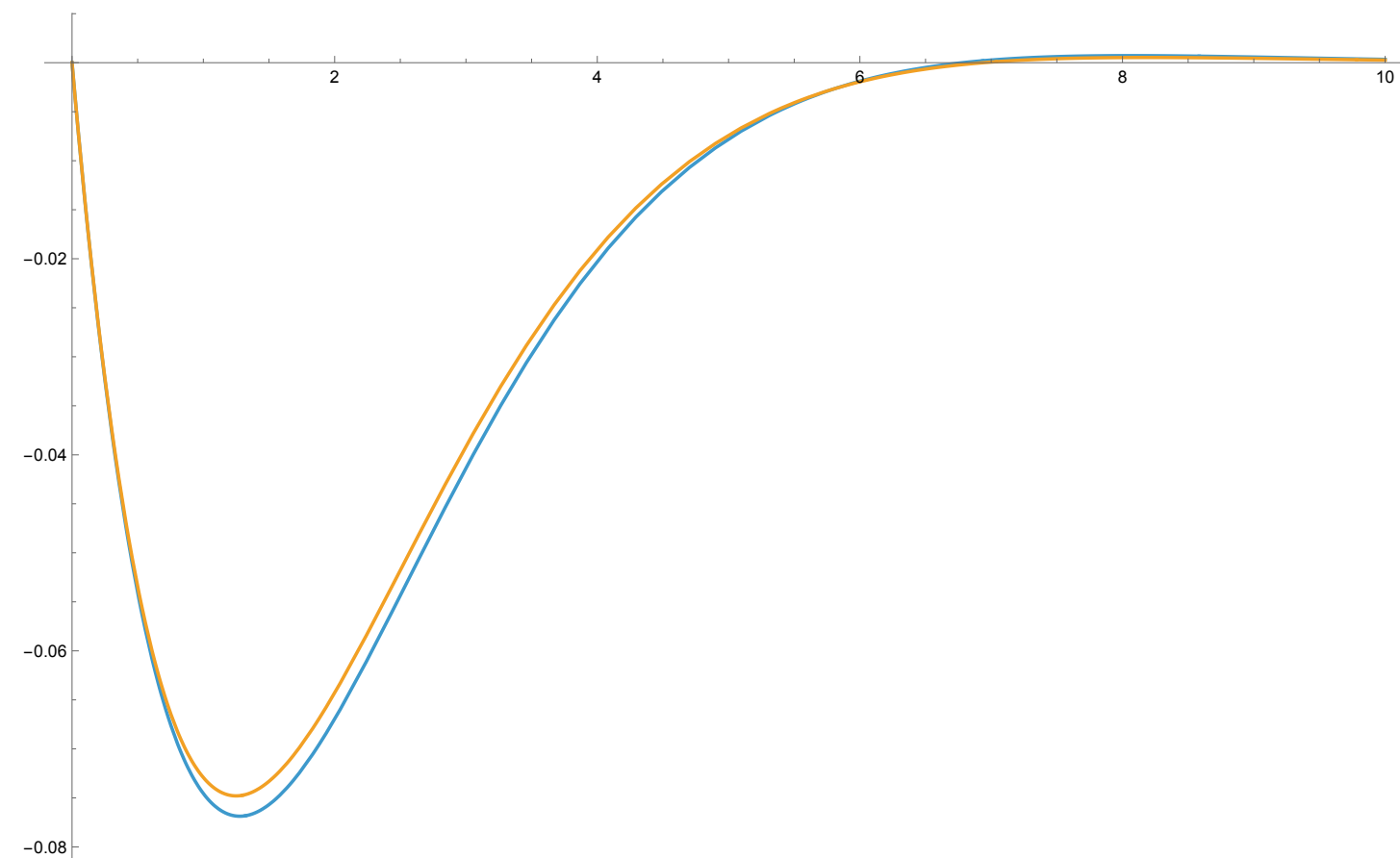
# Real dynamics vs Approximated One



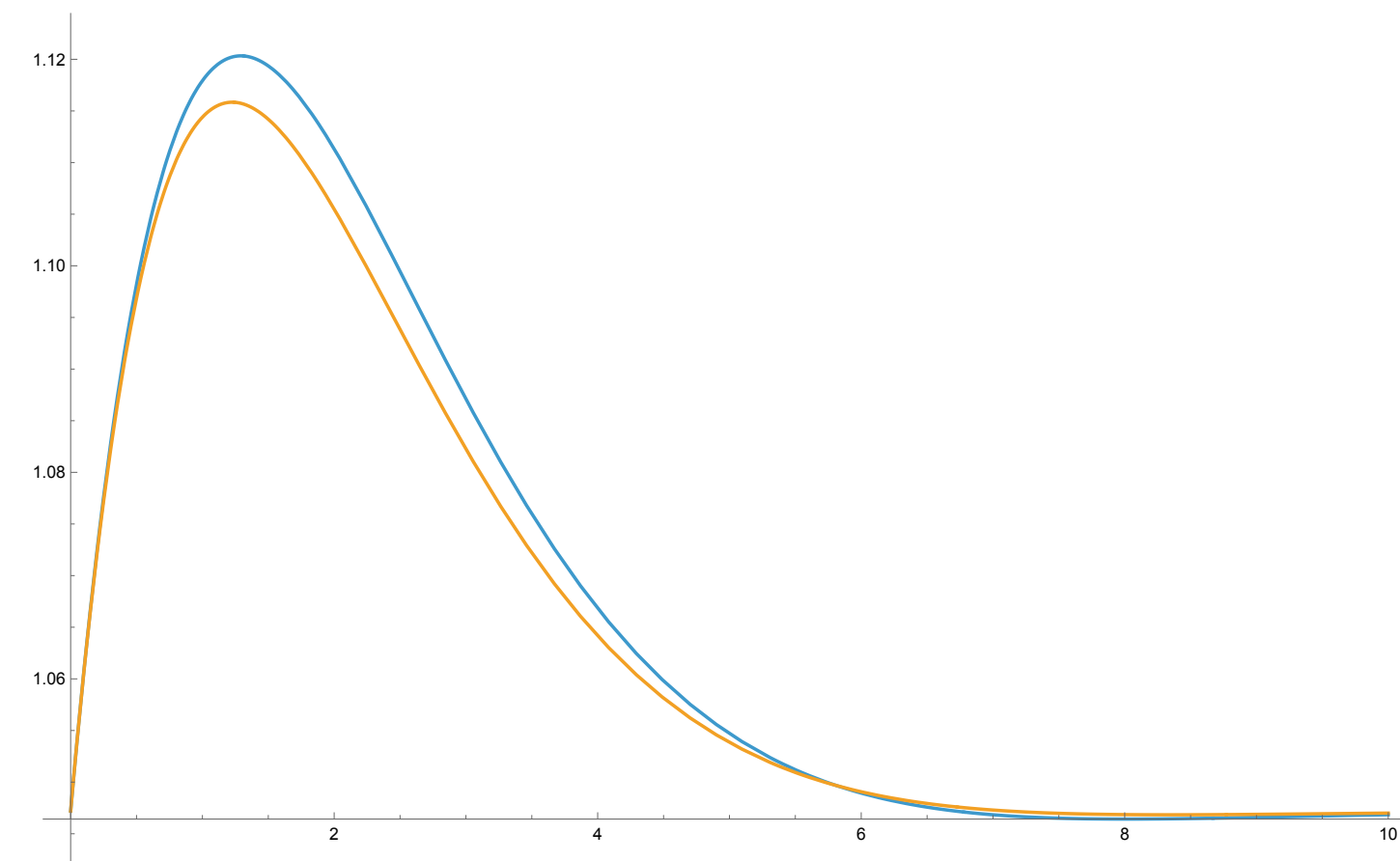
$q_1(t), q_{20} = \pi/2$



$q_2(t), q_{20} = \pi/2$

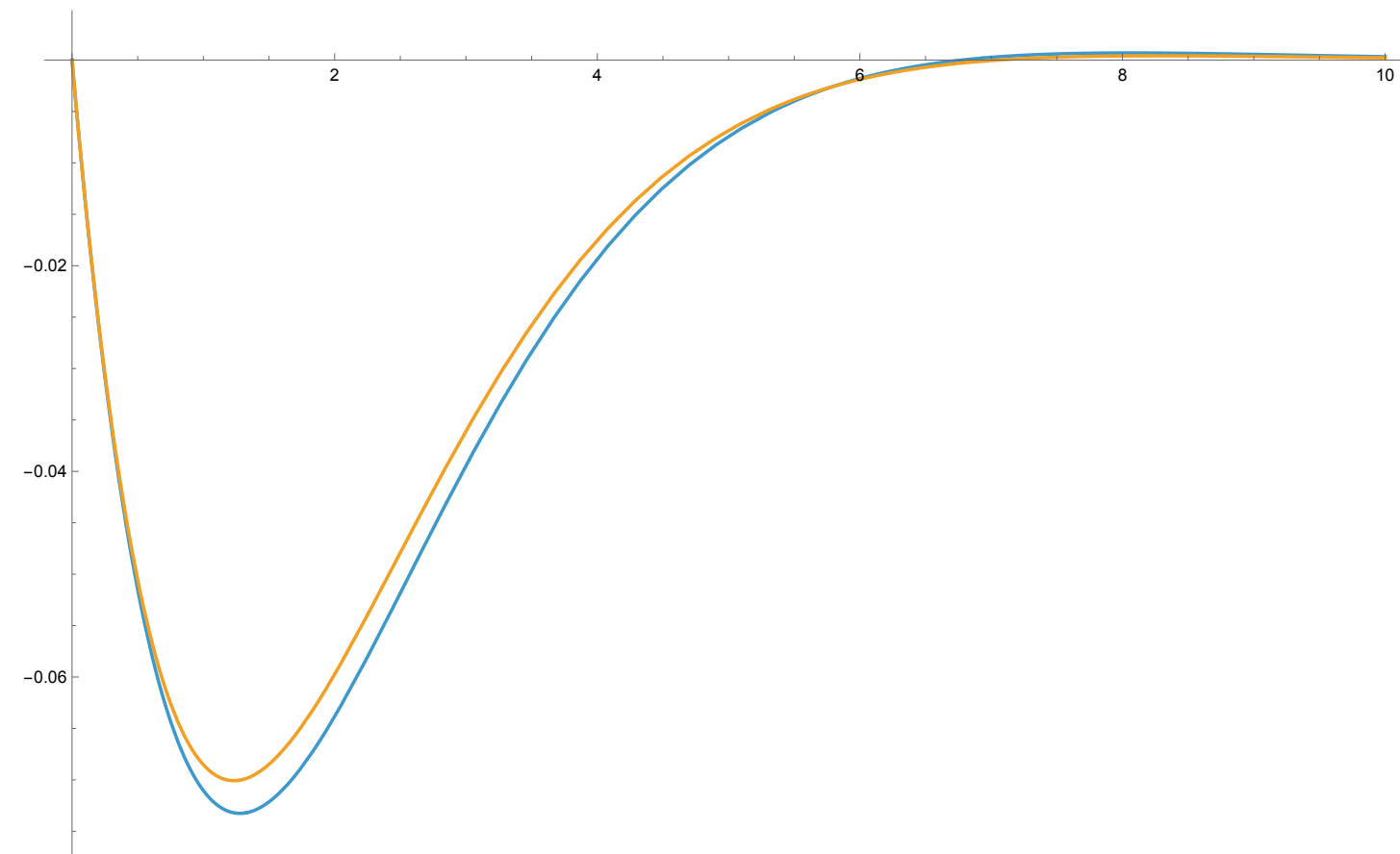


$q_1(t), q_{20} = \pi/3$

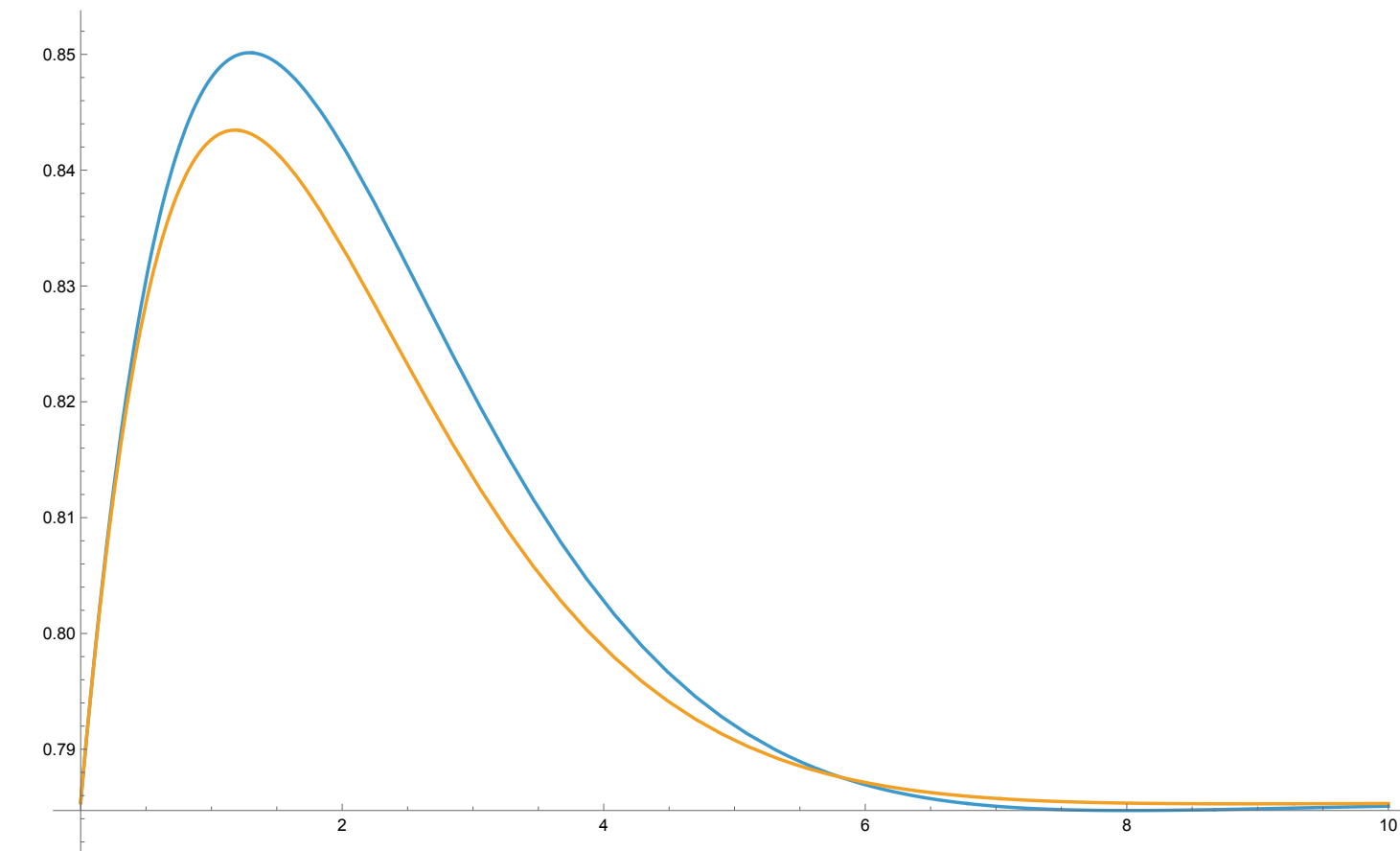


$q_2(t), q_{20} = \pi/3$

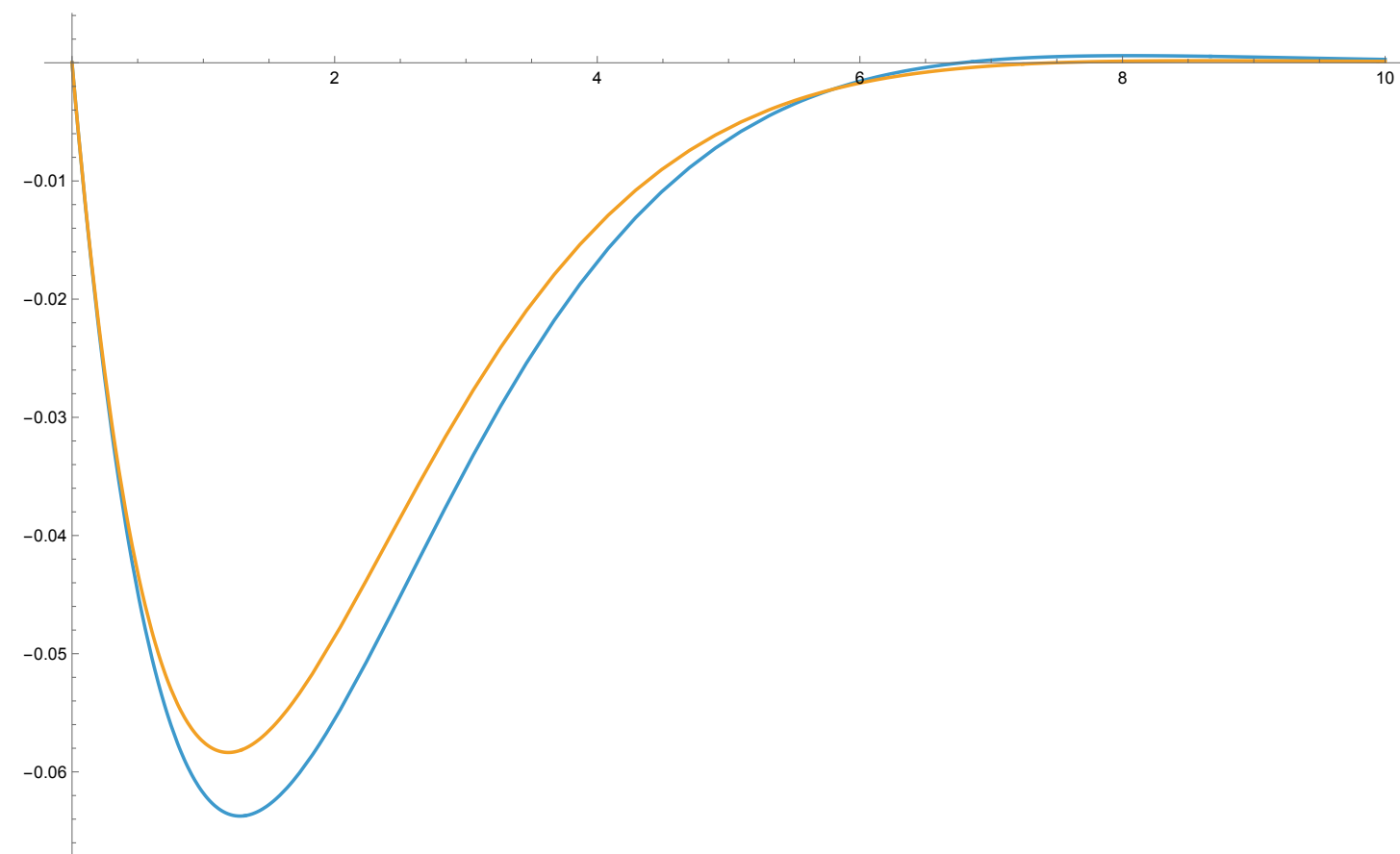
# Real dynamics vs Approximated One



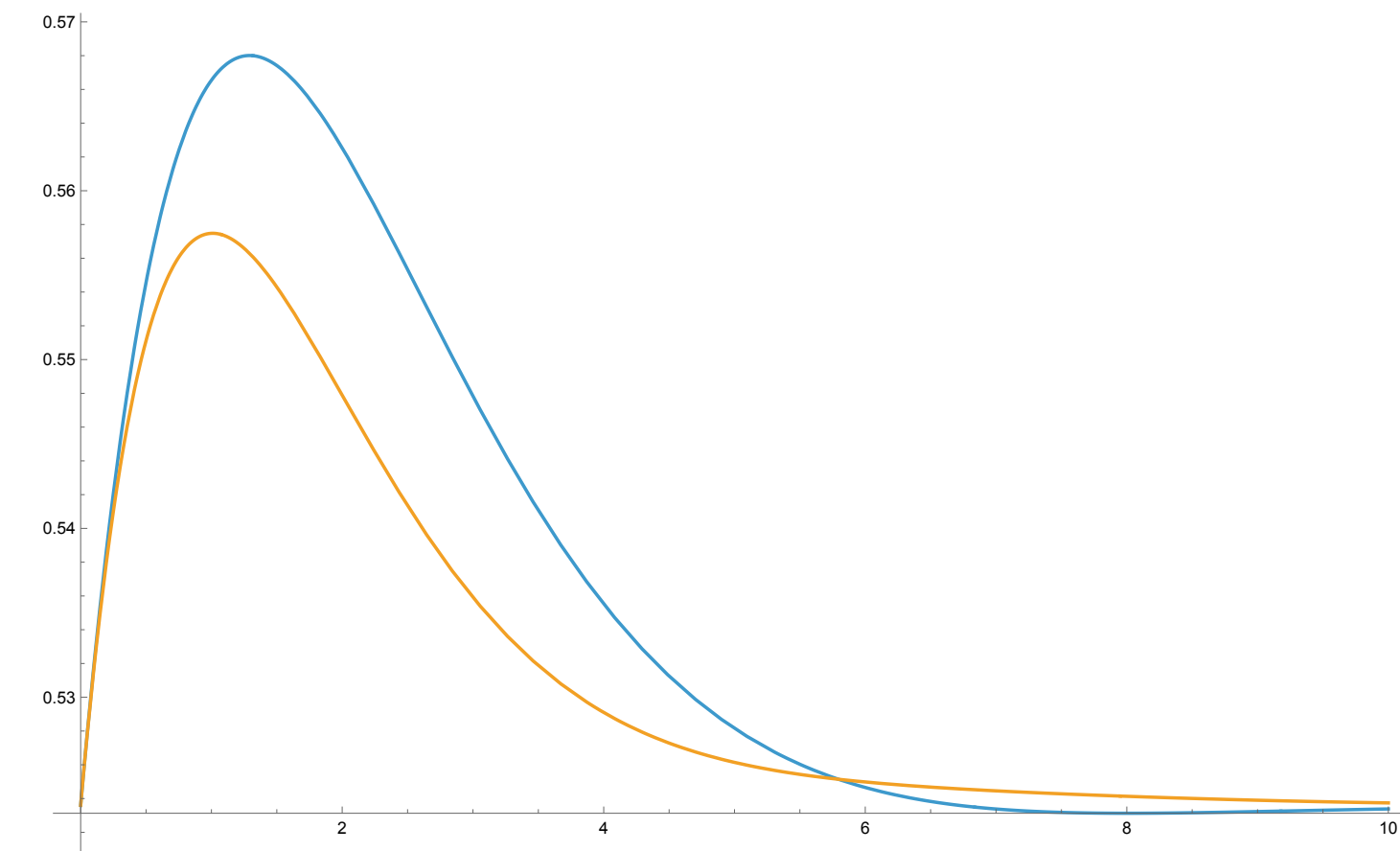
$q_1(t), q_{20} = \pi/4$



$q_2(t), q_{20} = \pi/4$

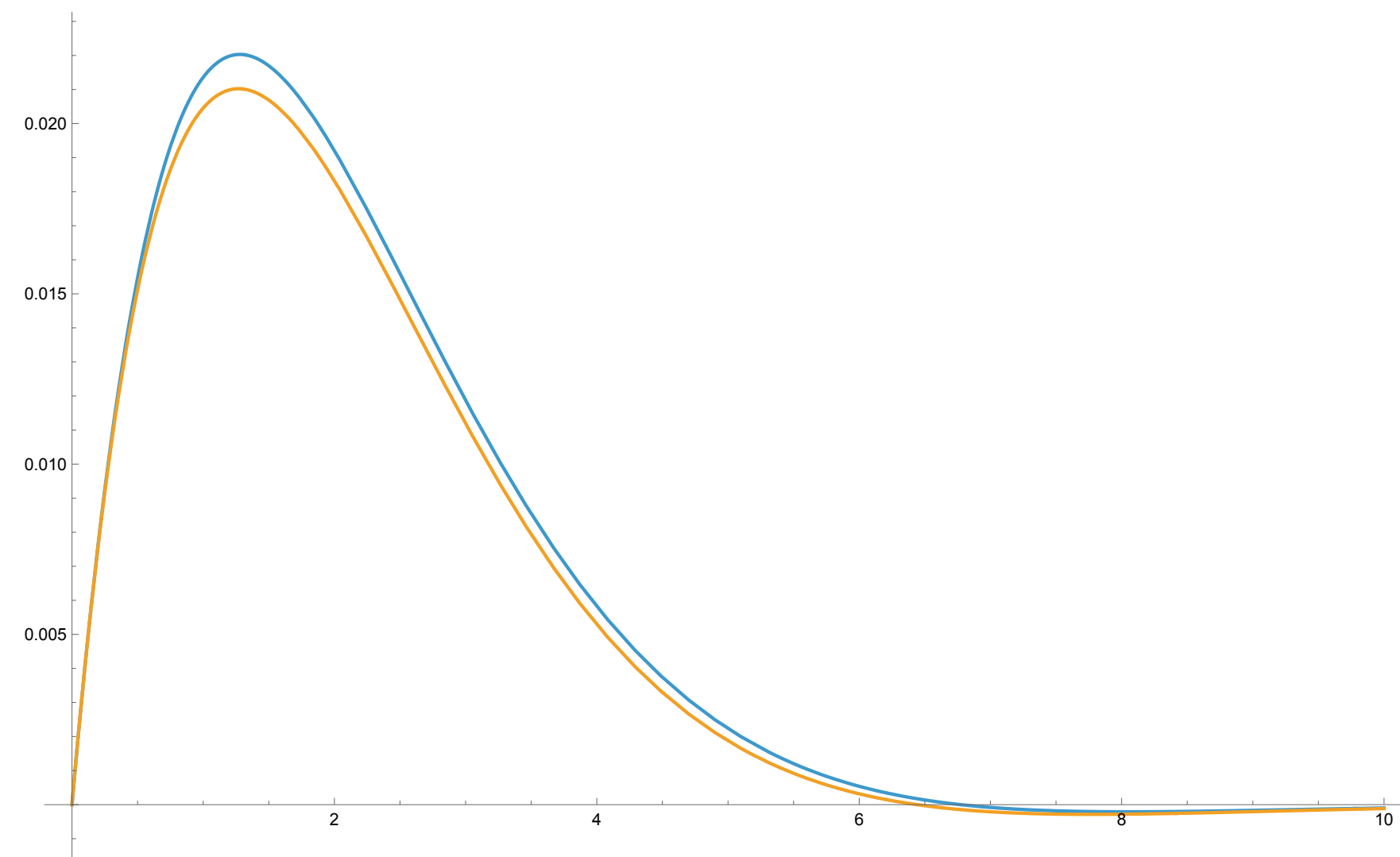


$q_1(t), q_{20} = \pi/6$

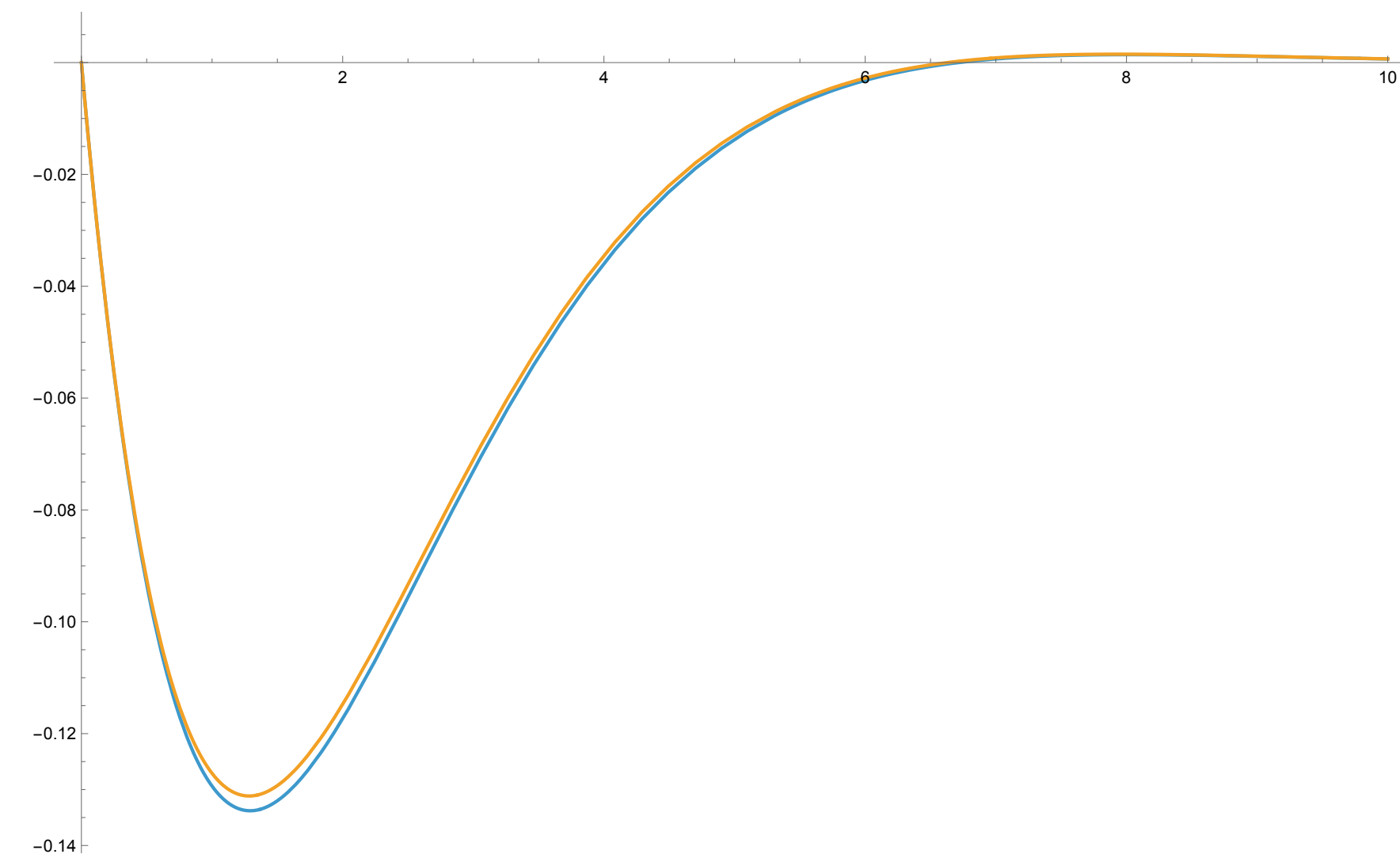


$q_2(t), q_{20} = \pi/6$

# Real dynamics vs Approximated One



$q_1(t), q_{20} = 0$

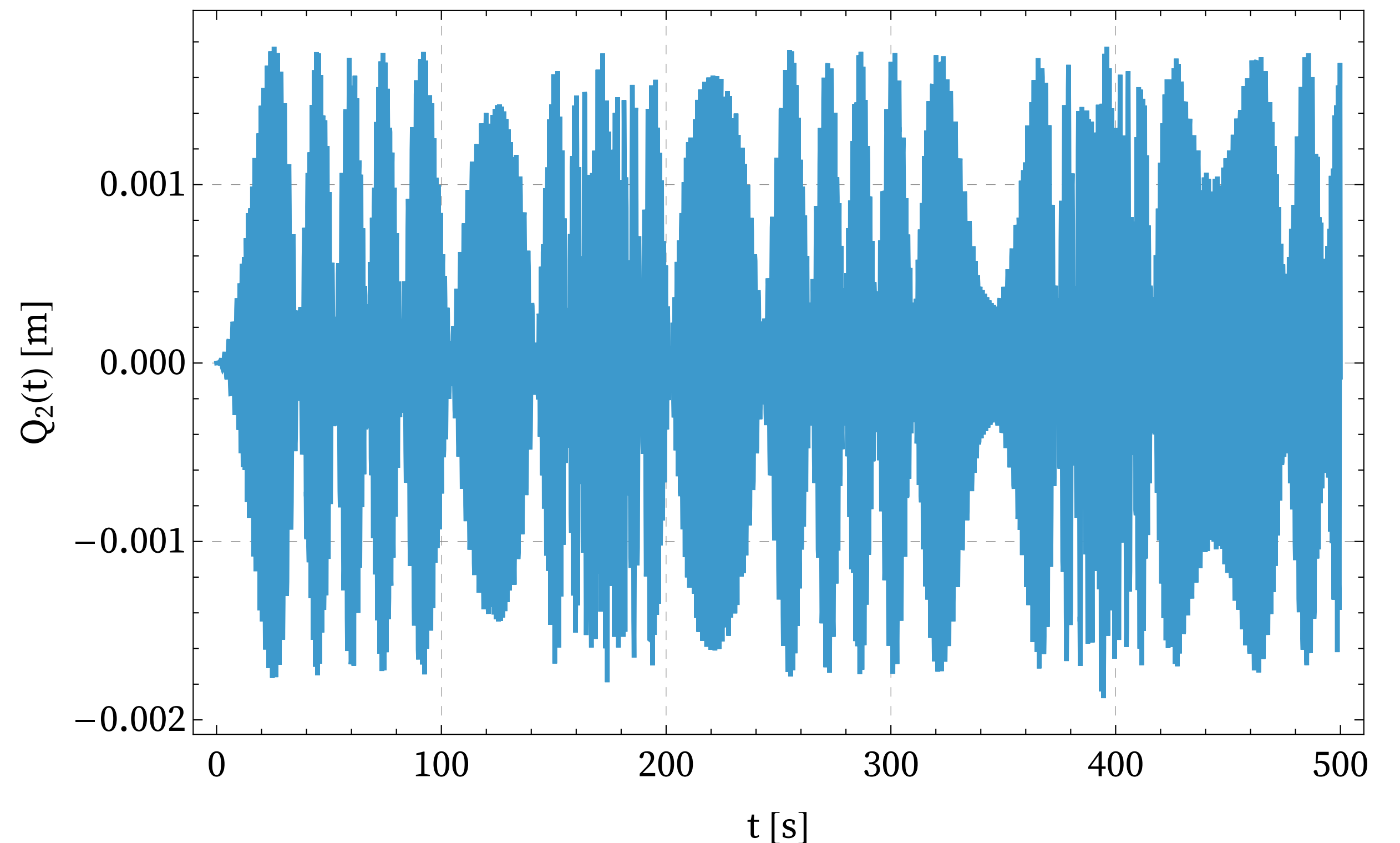


$q_2(t), q_{20} = 0$

If  $m_O = m$ , the plots are the same

# Elastic deformation

- Problems of wrong initial velocities addressed
- Change in amplitude due change in arm's configuration: when the arms are  $90^\circ$ , the beat stops
- Computed closed-loop control on arms



# Controlled Simulation

The arms are always in 90° configuration

