# **Quadrotor Control**



Aircraft Dynamics
UNIVERSITY OF COLORADO BOULDER

$$A = \begin{bmatrix} -1.3256 & -0.5415 & -0.4313 \\ -0.5415 & -1.4162 & -0.0523 \\ -0.4313 & -0.0523 & -0.6582 \end{bmatrix}$$

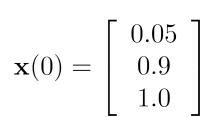
$$\mathbf{v}_i = \begin{bmatrix} 0.6985 \\ 0.6703 \\ 0.2506 \end{bmatrix}, \begin{bmatrix} 0.4916 \\ -0.7040 \\ 0.5126 \end{bmatrix}, \begin{bmatrix} -0.5200 \\ 0.2348 \\ 0.8212 \end{bmatrix}$$

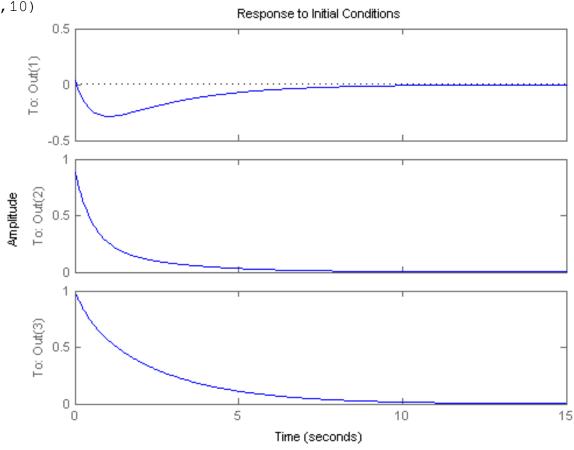
$$\lambda_i = \{-2, -1, -.4\}$$

For a state space model the eigenvalues  $\lambda_i$  of the matrix A are the poles of the system.

#### Matlab:

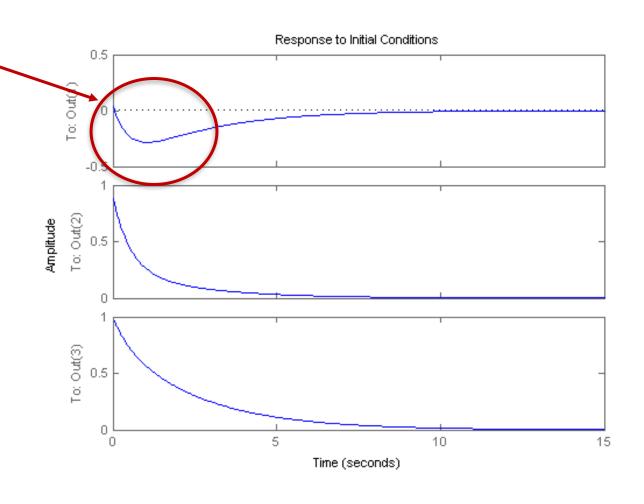
$$sys = ss(A, eye(3), eye(3), 0)$$
  
initial(sys, [0.05; 0.9; 1.0], 10)





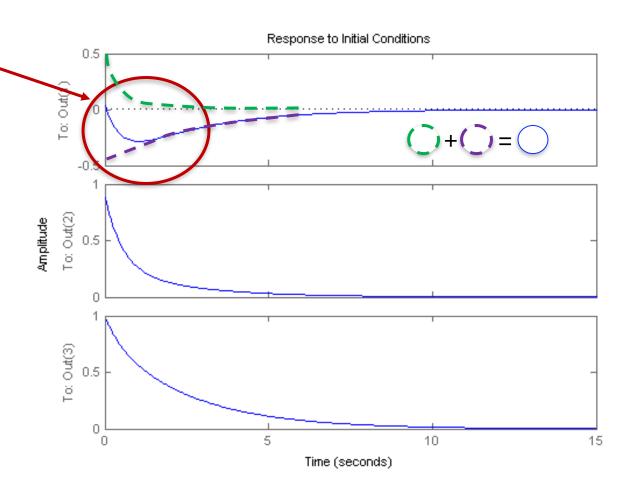
What is happening here if all poles are negative and real (exponential decay)

$$\mathbf{x}(0) = \begin{bmatrix} 0.05 \\ 0.9 \\ 1.0 \end{bmatrix}$$

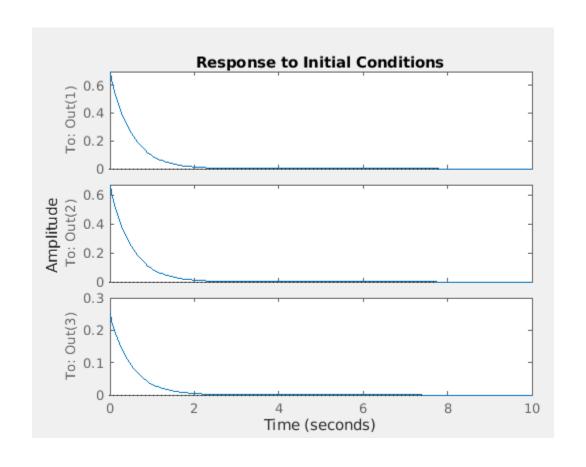


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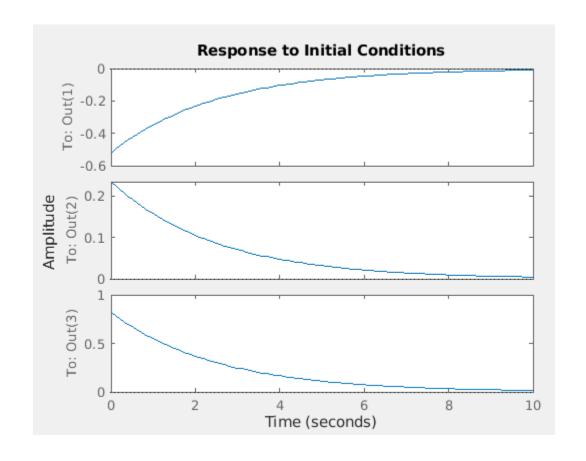
$$\mathbf{x}(0) = \begin{bmatrix} 0.05 \\ 0.9 \\ 1.0 \end{bmatrix}$$



$$\mathbf{x}(0) = \mathbf{v}_1 = \begin{bmatrix} 0.6985 \\ 0.6703 \\ 0.2506 \end{bmatrix}$$



$$\mathbf{x}(0) = \mathbf{v}_3 = \begin{bmatrix} -0.5200 \\ 0.2348 \\ 0.8212 \end{bmatrix}$$



$$A = \begin{bmatrix} -0.0129 & -0.0059 & 0 & -0.0507 \\ -0.104 & -0.8185 & 1 & 0 \\ 0.2867 & -12.6811 & -1.424 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix}$$

$$\mathbf{v}_{i} = \begin{bmatrix} 0.0032 \mp 0.0021i \\ -0.0222 \mp 0.2608i \\ 0.9321 \\ -0.0755 \mp 0.2389i \end{bmatrix}, \begin{bmatrix} -0.0473 \pm 0.5548i \\ -0.0001 \pm 0.0056i \\ -0.0054 \pm 0.0622i \\ 0.8288 \end{bmatrix}$$

$$\lambda_i = \{-1.1212 \pm 3.5472i, -0.0065 \pm 0.0752i\}$$

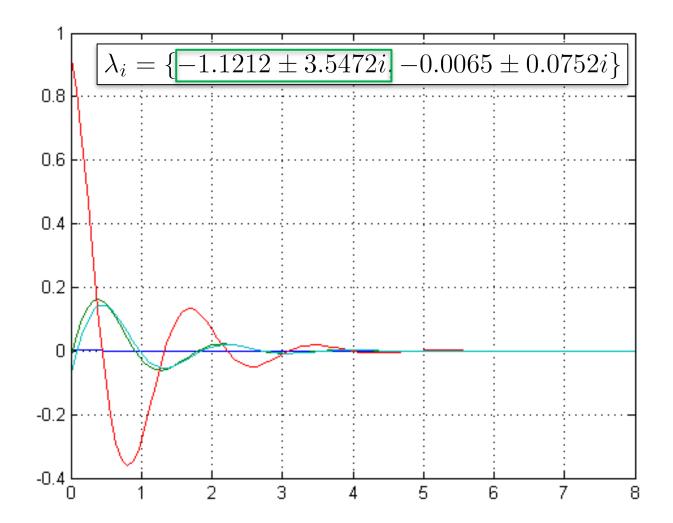
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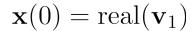
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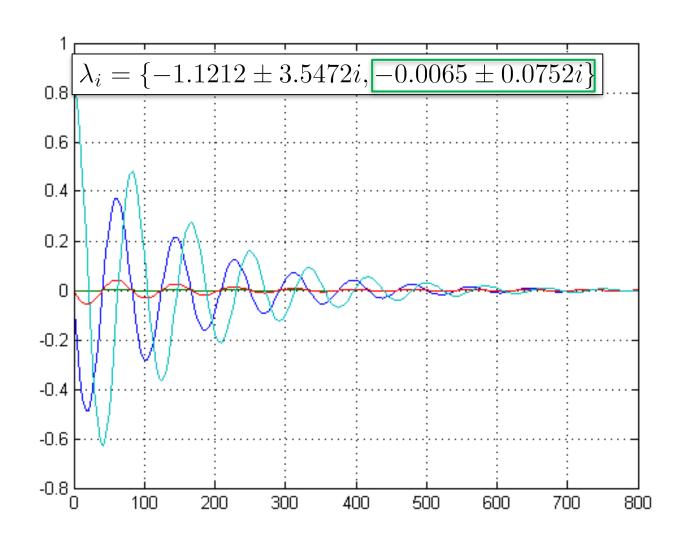
$$\lambda_i = -\omega_n \zeta + \omega_n \sqrt{1 - \zeta^2} i = -\sigma + \omega_d i$$





$$\mathbf{x}(0) = \text{real}(\mathbf{v}_3)$$

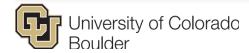
Oscillations with (very) different natural frequency and damping ratio, different relative magnitude and phases



#### Linear Control Design Process

- 1) Derive Equations of Motion
- 2) Linearize and Separate Equations
- 3) Design Control Architecture
- 4) Choose Gain Values
- 5) Test in Linear Simulation
- 6) Test in Nonlinear Simulation

- 1) P(I)D Tuning
- 2) Pole Assignment
- 3) Root Locus
- 4) Optimal Control (LQR)



#### **Pole Locations**

