

## Step-by-Step Guide to Generating an Ephemeris SSP, July 2005

1. Get the classical orbital elements (a,e,T,i,Ω,ω) and determine M at ephemeris time:

Using  $k=0.01720209894$  and  $\mu=1.000000$  for an asteroid orbiting the sun, with time units in days:

$$M = n(t - T) \quad \text{where:} \quad n = k \sqrt{\frac{\mu}{a^3}}$$

2. Get E from M inverting the Kepler equation,  $M = E - e \sin E$  using the Newton method as follows: Your first guess for E is M. You then use  $f = M - (E - e \sin E)$  you find  $f'$ , and so your second guess of E would be:

$$E_1 = E_0 - \frac{f(E_0)}{f'(E_0)} = E_0 - \left( \frac{M - (E_0 - e \sin(E_0))}{e \cos(E_0) - 1} \right). \text{ Repeat this step several times}$$

for  $E_2$  etc. until the M you get in Kepler's equation differs little from the M you had to begin with.

3. Get the Cartesian coordinates of the position vector:

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix} = \begin{pmatrix} a \cos E - ae \\ a\sqrt{1-e^2} \sin E \\ 0 \end{pmatrix}$$

That's the vector describing the asteroid position in orbit coordinates with the x axis coinciding with the perihelion.

4. Now start spinning that vector. First rotation is by  $(-\omega)$ , then by  $(i)$  and then by  $(\Omega)$  to get the  $\mathbf{r}$  vector in ecliptic coordinates:

$$\begin{pmatrix} x_1 \\ y_1 \\ z_1 \end{pmatrix} = \begin{pmatrix} \cos(\Omega) & -\sin(\Omega) & 0 \\ \sin(\Omega) & \cos(\Omega) & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(i) & -\sin(i) \\ 0 & \sin(i) & \cos(i) \end{pmatrix} \begin{pmatrix} \cos(\omega) & -\sin(\omega) & 0 \\ \sin(\omega) & \cos(\omega) & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} x \\ y \\ z \end{pmatrix}$$

5. Convert  $\mathbf{r}$  to equatorial by rotating the coordinates by the tilt of earth relative to the ecliptic  $\epsilon$ :

$$\begin{pmatrix} x_2 \\ y_2 \\ z_2 \end{pmatrix} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(\epsilon) & -\sin(\epsilon) \\ 0 & \sin(\epsilon) & \cos(\epsilon) \end{pmatrix} \begin{pmatrix} x_1 \\ y_1 \\ z_1 \end{pmatrix}$$

6. Determine the earth-sun vector from the JPL site
7. Determine the range vector  $\vec{\rho}$  from  $\vec{\rho} = \vec{r} + \vec{R}$ , and also  $\hat{\rho} = \vec{\rho}/|\vec{\rho}|$ .
8. Solve for RA and Dec from :  $\sin(\delta) = \hat{\rho}_z$ , and  $\cos(\alpha) = \hat{\rho}_x / \cos(\delta)$