



Small Satellite Research Laboratory

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UNIVERSITY OF GEORGIA

How UGA is Helping Develop the Next Generation of
Systems Engineers for the State of Georgia

Dr. David L Cotten

Principal Investigator – SSRL

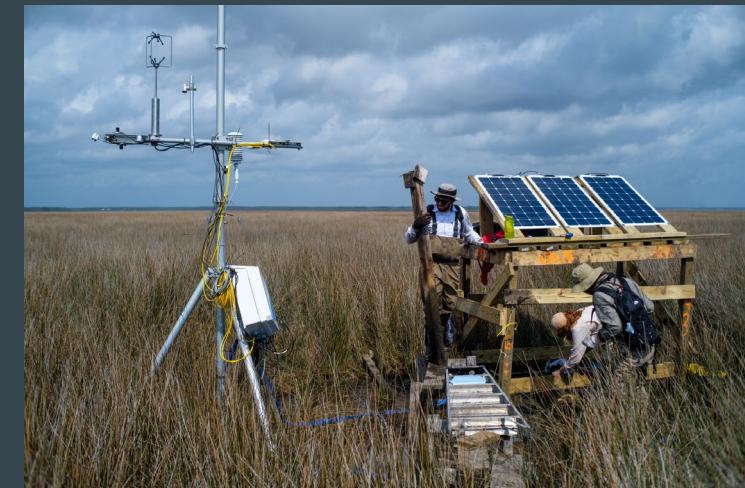
Assistant Research Scientist – Center for Geospatial Research
Department of Geography

Background

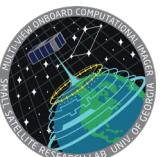
- Physics BS from Louisiana State University: 2005
- Physics and Astronomy PhD from University of Georgia: 2011
- Post Docs in Micro Meteorology and Environmental Physics: 2011-2014
 - Eddy Covariance method for carbon storage in terrestrial ecosystems
- Research Scientist (Center for Geospatial Research – Geography Department) : 2015 - Present
 - Combining Eddy Covariance and Remote Sensing to measure carbon storage as a function of tides in wetland ecosystems



View from the boat at the Bayou Sauvage, LA eddy covariance site.



Eddy Covariance flux tower in Grand Bay, MS.

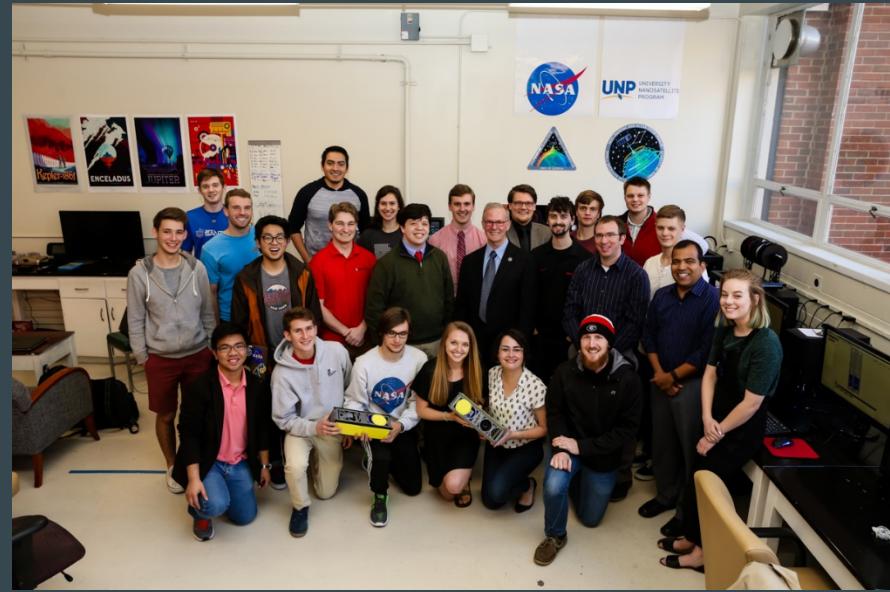


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Small Satellite Research Laboratory

- Founded in 2016
 - 2 Cube satellites under development
- Over 40 undergraduates work in the lab
- 12 faculty advisors, mentors, and researchers from all over campus



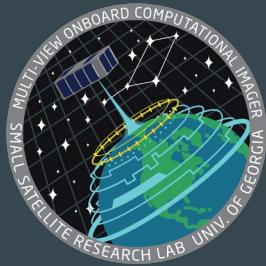
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Our Missions

MOCI

- *Multi-view Onboard Computational Imager*
- Funded through the Air Force (AFRL)
- Use images to create 3D point clouds
- Create high performance computational units for use in space



SPOC

- *SPectral Ocean Color Satellite*
- Funded through NASA
- Measure coastal, oceanic, and environmental productivity
- Adjustable multi spectral payload



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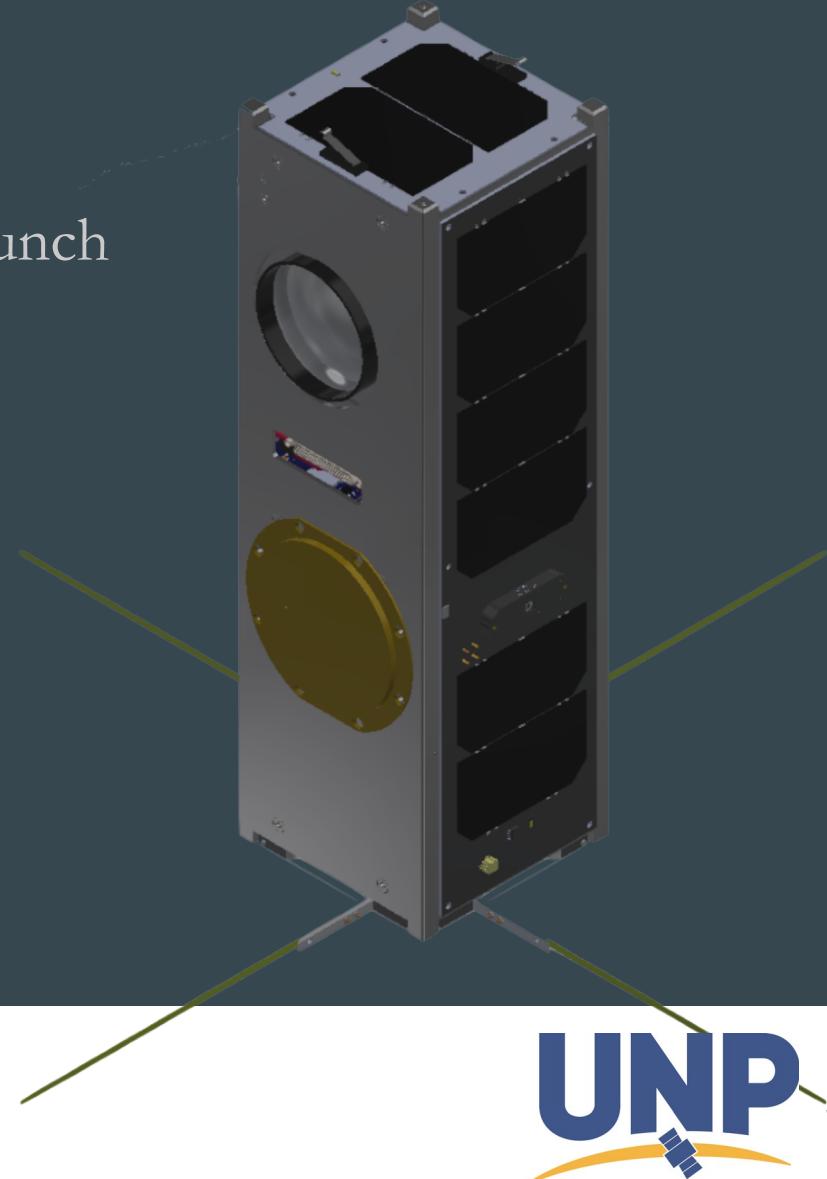


MOCI: Multi-view Onboard Computational Imager

- Part of the 9th University Nanosatellite Program
- Competed against 9 schools for a chance to full build and launch
 - UGA WON!
- Was a 3U
 - Now a 6U
- Launch in late 2020



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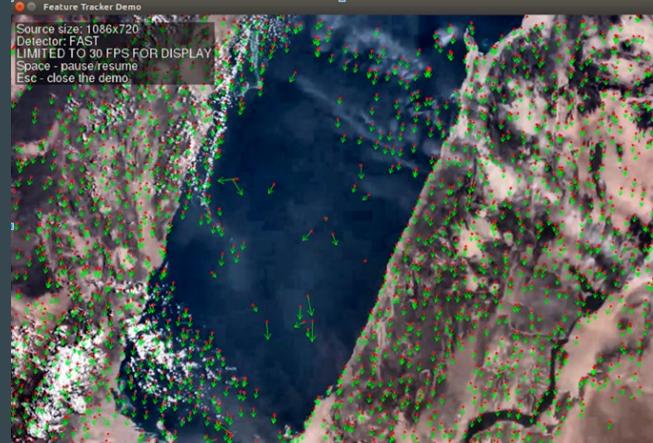


MOCI: Multi-view Onboard Computational Imager

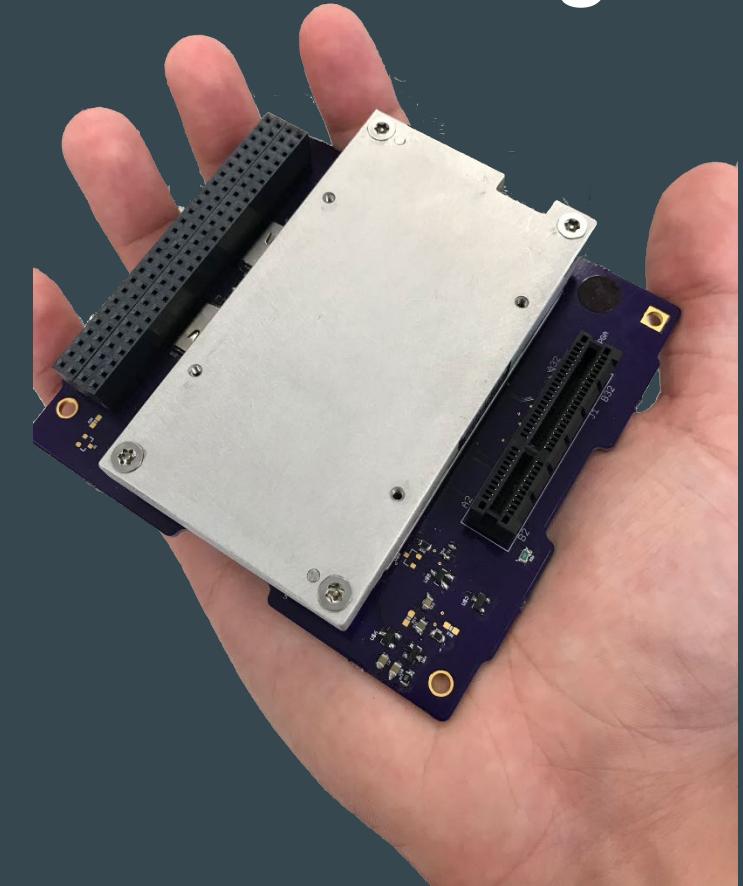
- Use GPU's to increase on board processing abilities
- Utilize computer vision and neural networks to identify and track Earth based objects from an orbiting platform



MODIS image of Lake Erie 20160516



Simulation of the tracking of objects over the red sea from an ISS nadir video.



The UGA SSRL's miniaturized high performance computation unit developed around the Nvidia TX2i for the MOCI satellite

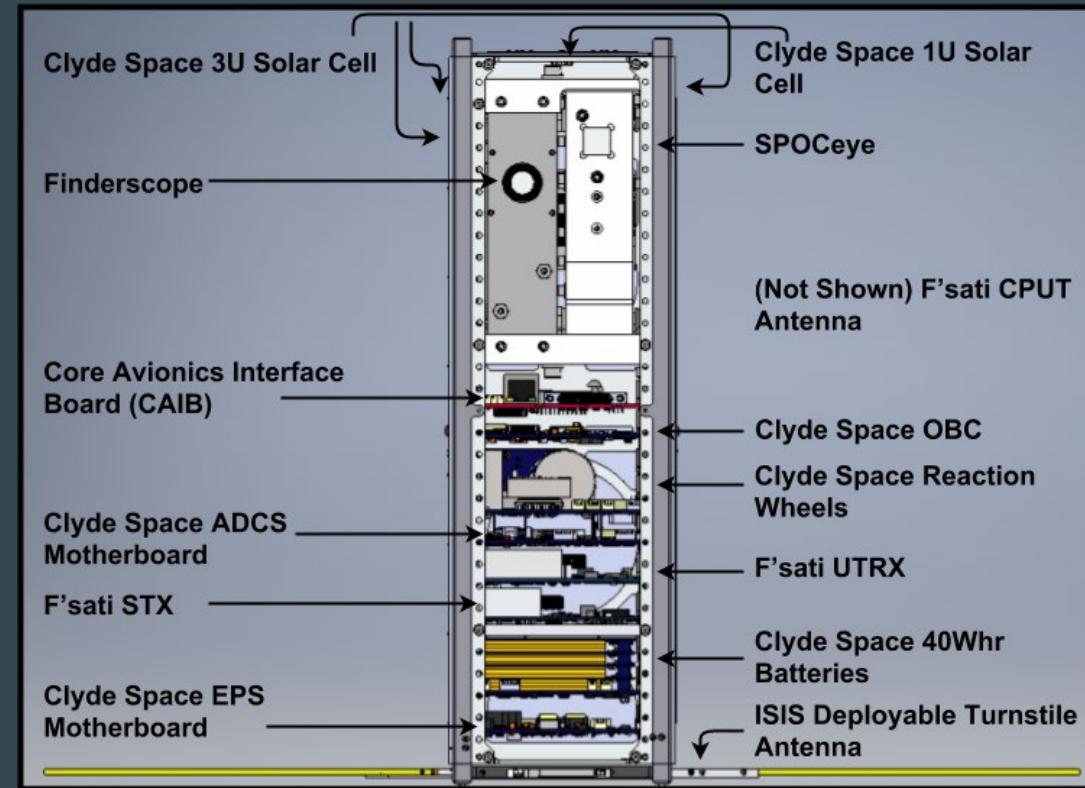
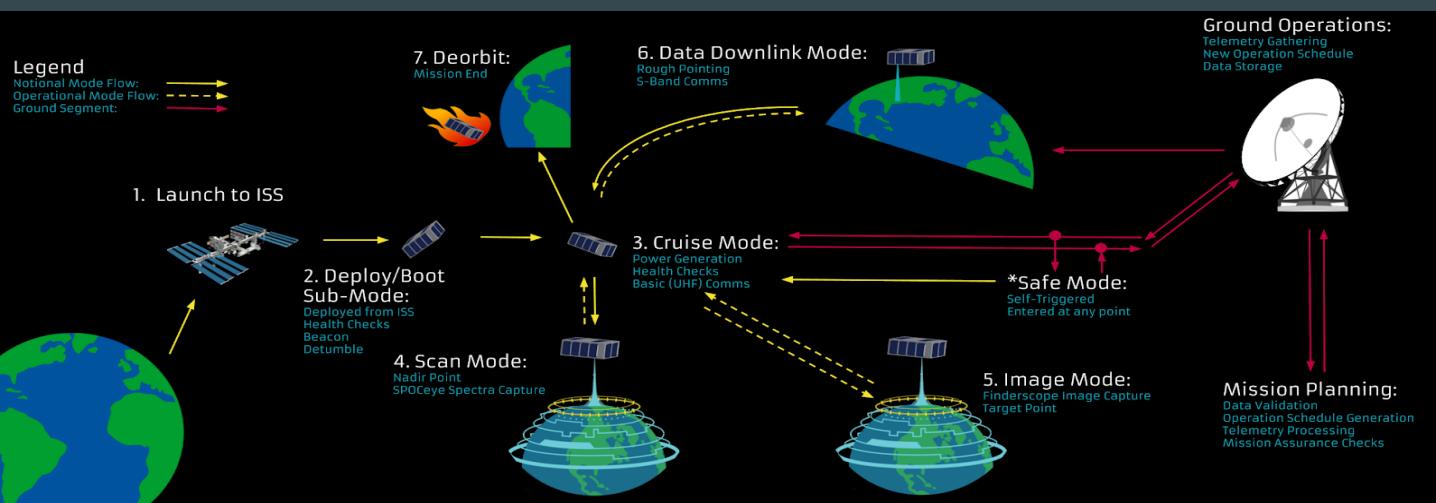


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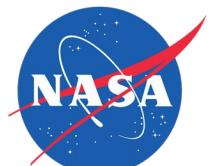
UNP 6

SPOC: Spectral Ocean Color Satellite

- 3U CubeSat
- Adjustable Multispectral Imager
 - 16 bands
 - $\sim 433 \text{ nm} - \sim 822\text{nm}$
 - 130m resolution

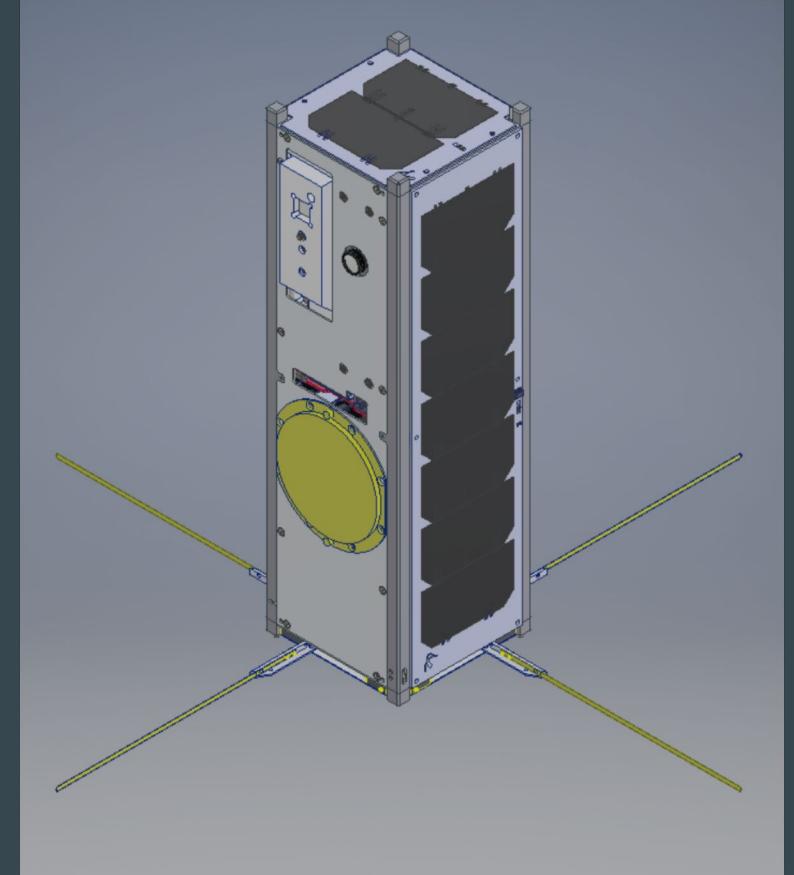


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SPOC: Spectral Ocean Color Satellite

- Delivery July 2019
 - Expected launch October 2019
- ISS orbit
 - ~450km altitude
 - 51.6° inclination



CAD of final SPOC design



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Difficulties at Non Technical School

- Finding appropriate facilities and expertise in satellite integration
- More expensive to start a program



Old lab space used for storage



New lab space with 7 desktops, cleanroom, and ESD area

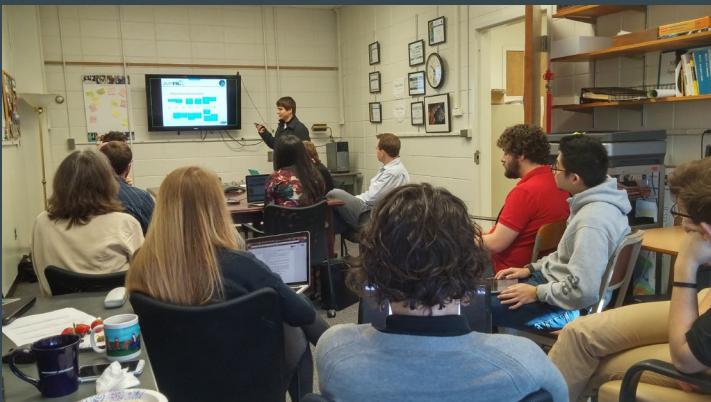


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Difficulties at Non Technical School

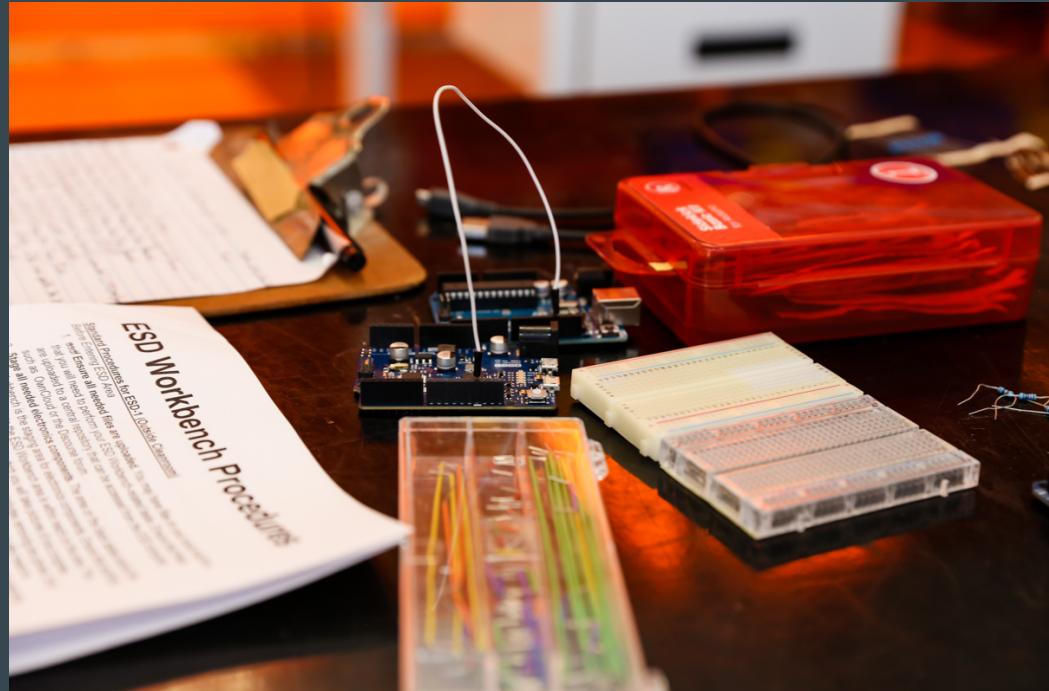
- Missing a strong engineering mentoring component
- Having to rely heavily on independent research
- Getting departments to work together



MOCI PDR, January 2017



Students working on deliverables, March 2017



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Advantages at a Non Technical School

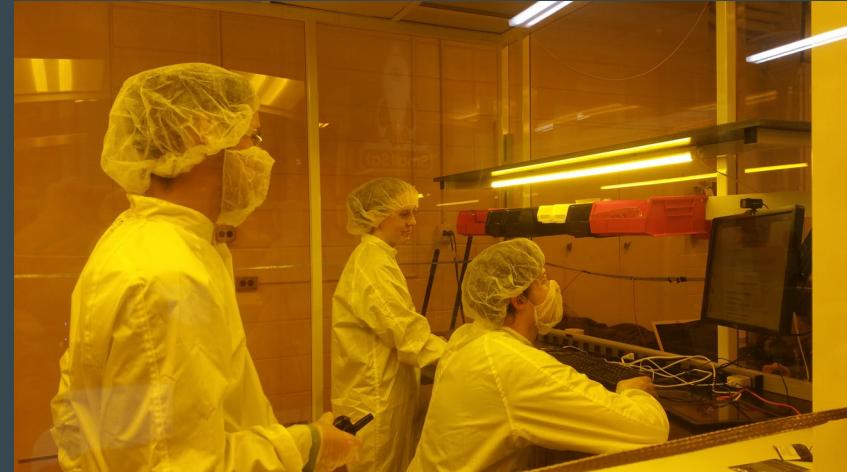
- Students directly involved in designing/choosing equipment and components
- Entire student team is involved in every aspect of the project including: Proposal writing, reviews, outreach, etc.



Team meetings discussing ConOps



Group mechanical analysis



Students performing acceptance testing

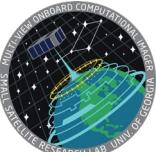
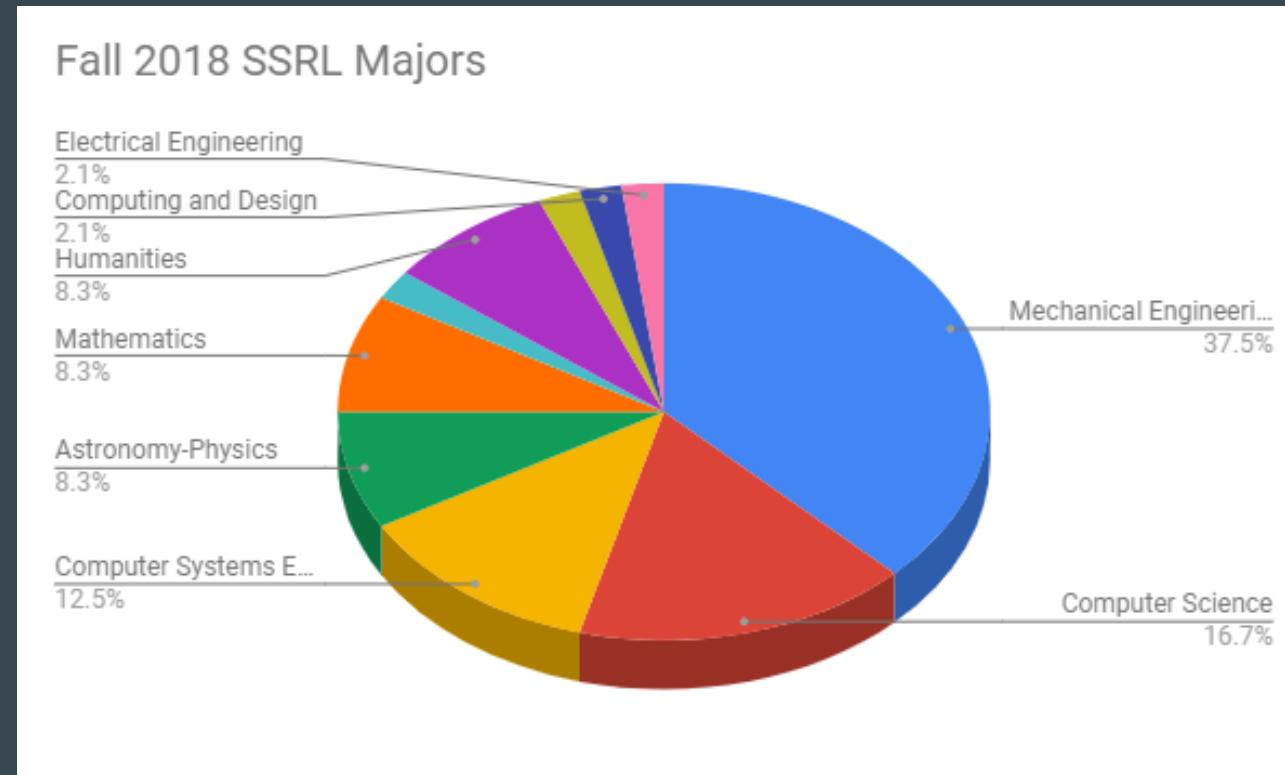


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Advantages at a Non Technical School

- “Outside the box” approaches to problem solving
- Numerous non-STEM majors involved in space based projects
- Easier to attract highly motivated and intelligent students from all majors not just engineering, less intimidating



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Advantages at a Non Technical School

- Students involved in immersive/experiential learning opportunities that cannot be found else where around campus
- Using gained knowledge to create new courses, provide internships for high school students, and create outreach material



Physics and
Astronomy

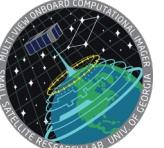
Mathematics/
Geography

Mechanical Engineering

SSRL students learning soldering techniques.



Workshop for middle school students to learn about
CubeSat's and simple circuits



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Statistics

- Over 70 undergraduates have participated in the lab
- 130 applicants per semester



Team Meeting March 2016



Team Meeting December 2016



Team Meeting February 2018



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Statistics - Outreach

- 8 high school interns
- Over 250 K-12 students reached this year



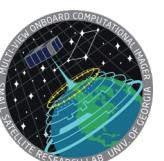
Women in Technology Workshop hosted by SSRL, 2017



NSF LISSIL B community outreach with the help of SSRL,
2017



Middle School Students get a lab tour and holding a 3D model of our
satellite, 2018

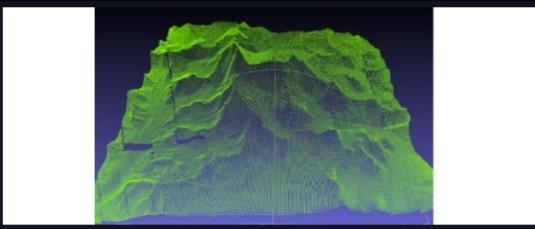


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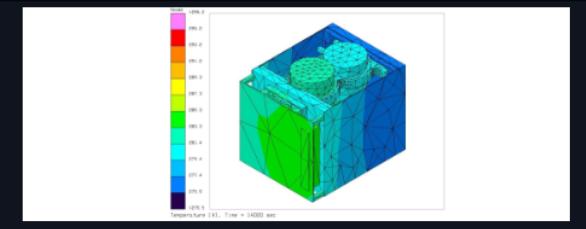


Statistics - Student Development

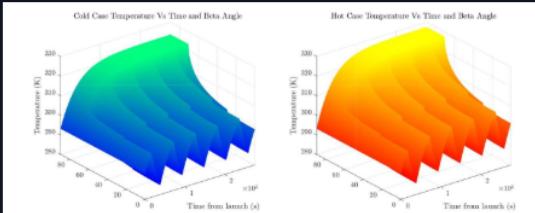
- Posters/Presentations
 - 2016 – 4
 - 2017 – 19
 - 14 unique authors
 - 2018 – 25
 - 26 unique authors



Caleb Adams and David L. Cotten PhD
A Near Real Time Space Based Computer Vision System for Accurate Terrain Mapping
32nd Annual AIAA/USU Conference on Small Satellites, August 2018



Casper Versteeg and David L. Cotten PhD
Thermal Management and Design of High Heat Small Satellite Payloads
32nd Annual AIAA/USU Conference on Small Satellites, August 2018



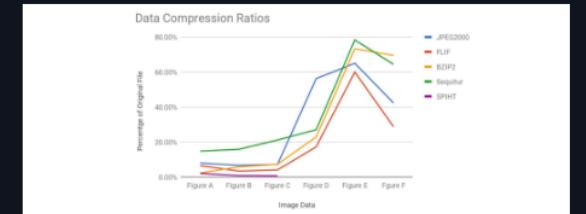
Casper Versteeg and David L. Cotten PhD
Preliminary Thermal Analysis of Small Satellites
AIAA Region II Student Conference, April 5th 2018



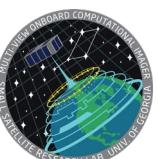
Nicholas Heavner and David L. Cotten PhD
The Multi-view Onboard Computational Imager (MOCI) Mission: A 3U Sized Supercomputer Floating in Space
2018 NASA Laboratory for Astrophysics Workshop, April 9th, 2018



Megan Arrogeti and David L. Cotten PhD
Pseudo Invariant and Coastal Target Feasibility
UGA Center for Undergraduate Research Symposium, April 9th, 2018



Ethan Barnes and David L. Cotten PhD
Compression and Encryption Methods for Small Satellite Communications
UGA Center for Undergraduate Research Symposium, April 9th, 2018



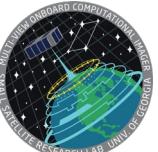
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Statistics - Job placement

- Internships
 - 2017 – 4
 - 2018 – 15
 - 2019 – 2
- Jobs
 - 12 jobs aerospace related around the country



Location of internships in 2017 and 2018



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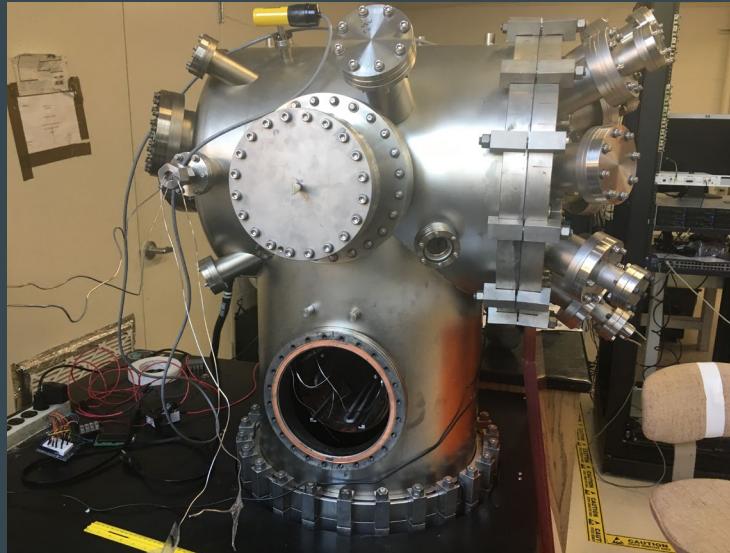


Future Goals

- Work more with Georgia industry
 - Try to get students to stay in Georgia
 - Provide new source of employees
 - Help grow the UGA SSRL



Students meeting with UGA Board Members



Nearly complete T vac chamber



Nearly complete Sband, UHF, and VHF ground station



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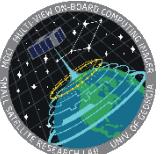
Acknowledgments



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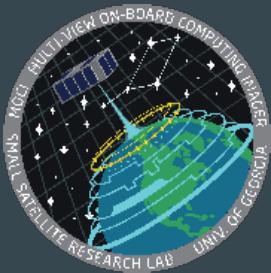
College of Engineering
School of Electrical and Computer Engineering
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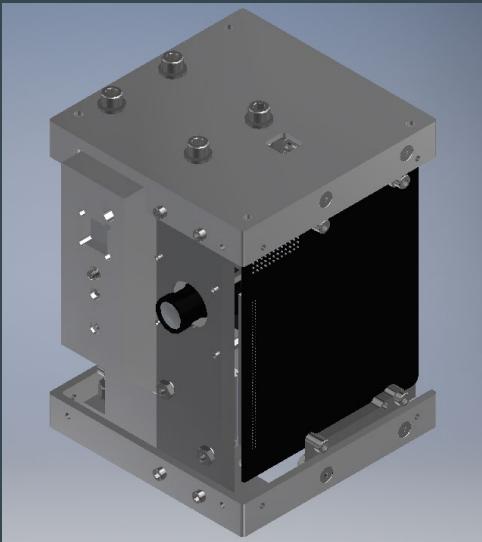


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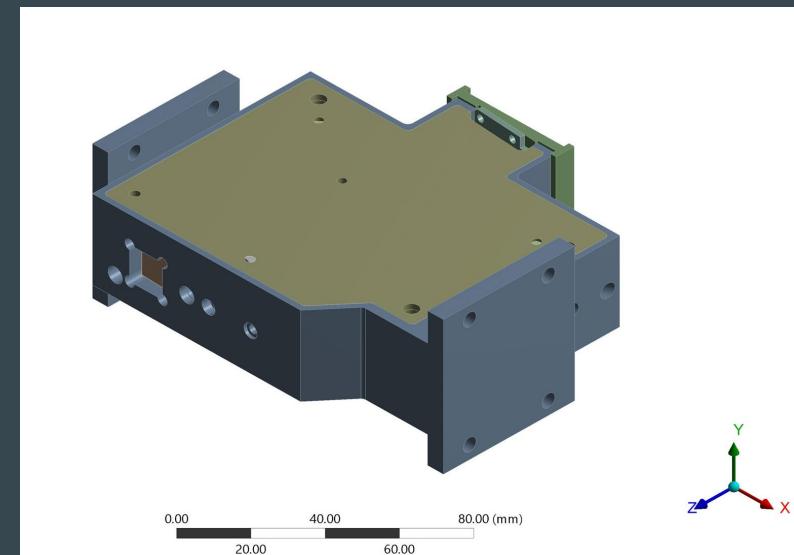


SPOC: Payload Housing

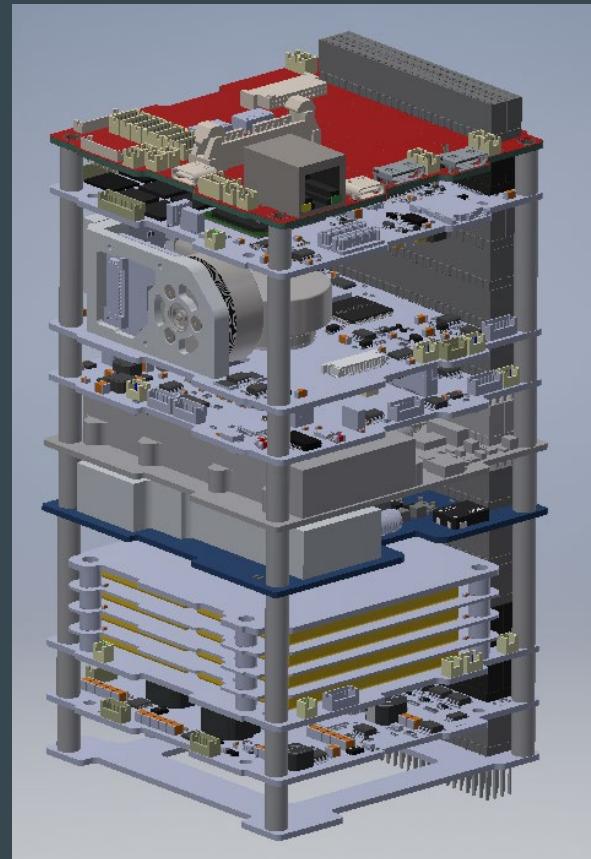
- Three main components:
 - SPOCeye (Camera)
 - Finder scope 1km GSD
 - Electronics/housing



SPOC Payload housing



UGA "shoe box" to hold optical design



UGA SPOC core avionic stack with SSRL boards

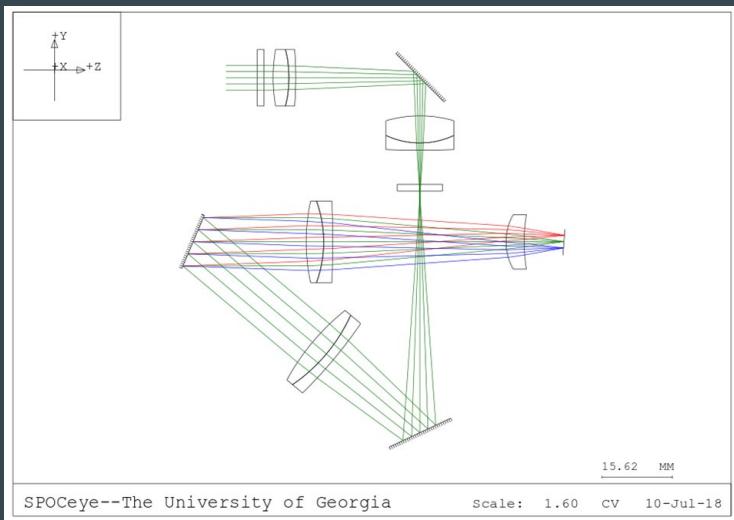


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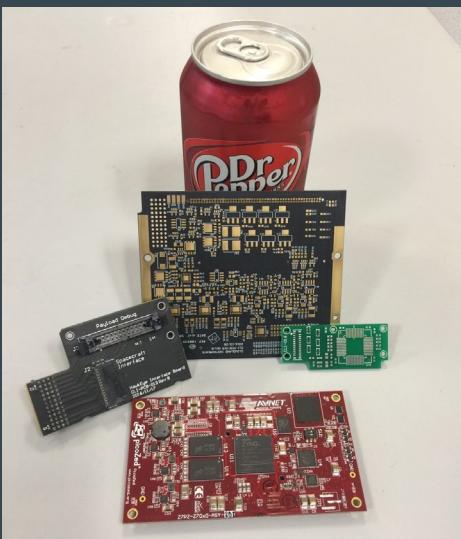


SPOC: Payload Specifics

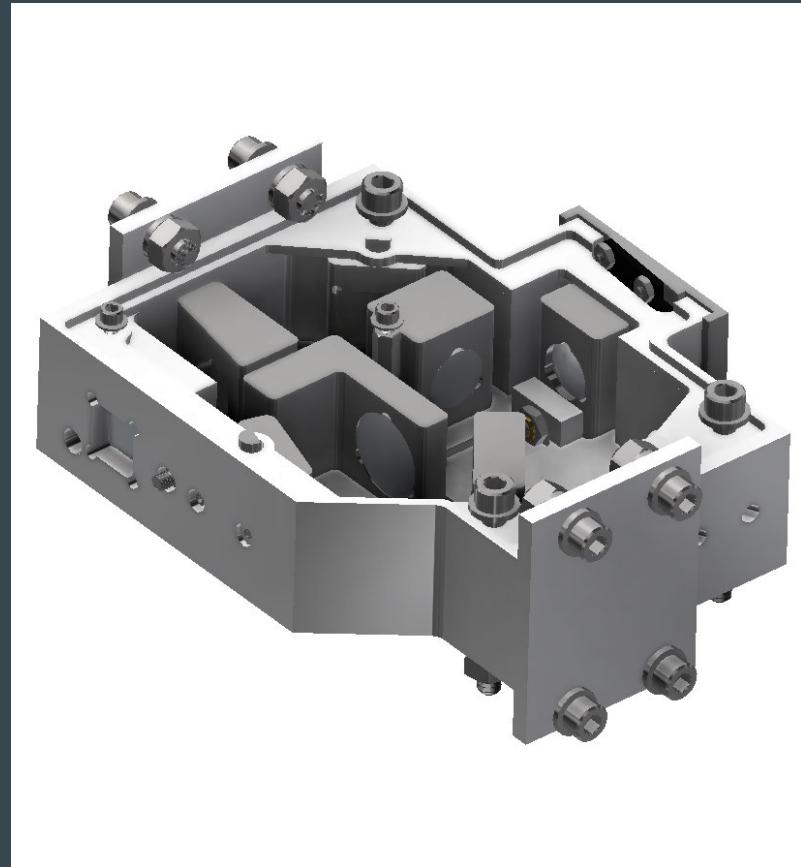
- Optical layout designed by Cloudland Instruments
 - Focus on coastal wetlands status and near-coastal ocean productivity
 - Diffraction style hyperspectral sensor from 432 nm - 866 nm



Cloudland Instruments compact optical design



Cloudland Instruments board and sensor for SPOCeye



Optical housing with adjustable lens mounts

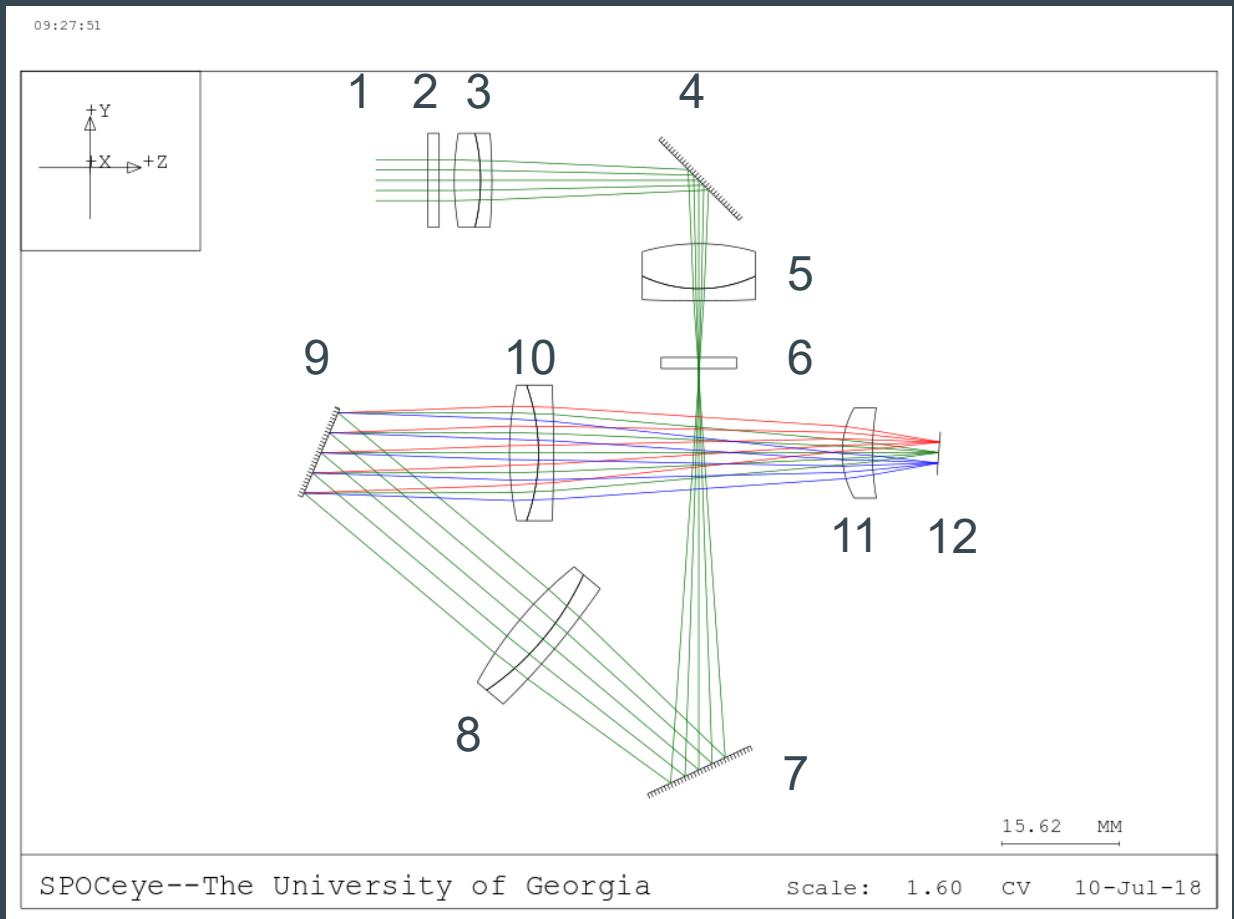


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SPOC: Payload Optics

1. Polarization scrambler
2. Longpass filter (10.5 mm diameter)
3. Primary telescopic lens
4. First mirror fold
5. Secondary telescoping lens
6. Custom optical slit
7. Second mirror fold
8. Collimator lens
9. Diffraction grating fold
10. First camera lens
11. Focusing camera lens
12. CMOS Sensor

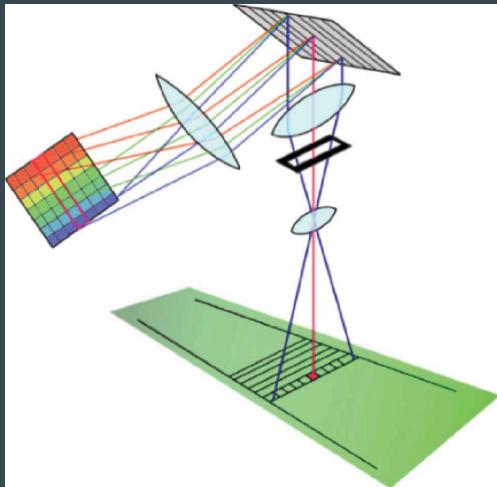


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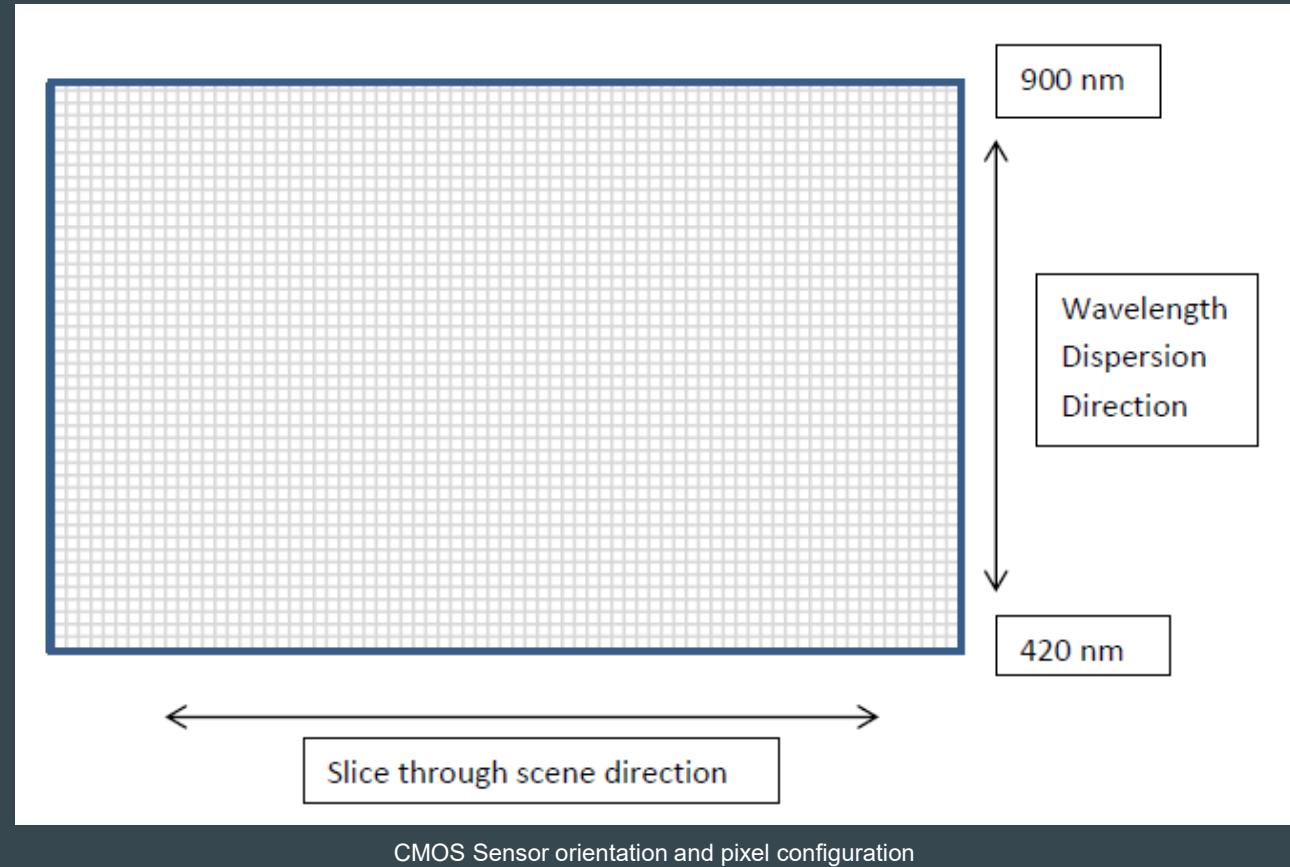


SPOC: Sensor

- 752 x 480 pixel CMOS array
- 98km x 130m Capture
- 1.042 nm per pixel
- 130x130 m pixel for 18 ms exposure



Example of SPOCs push broom scan orientation

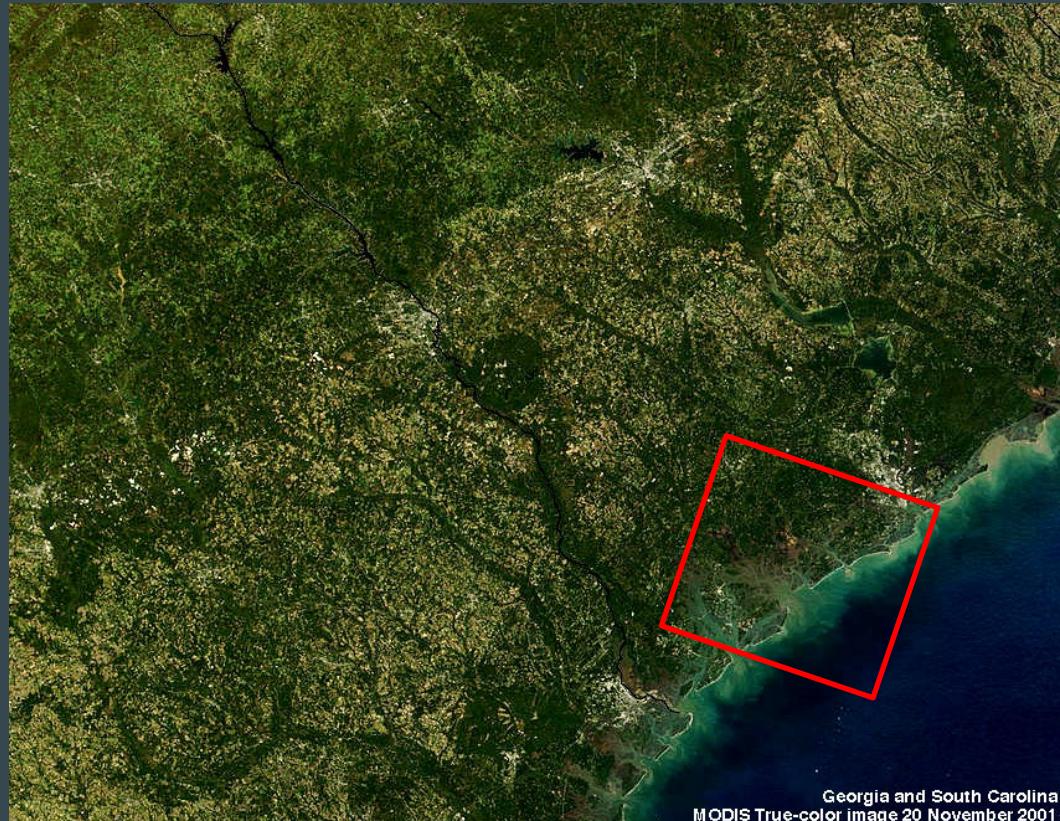


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SPOC: Payload Problem

- Hyperspectral Sensor from 433 nm - 866 nm
- 750 frames $\sim 9,500 \text{ km}^2$ area
- Data Size?
 - Data downlink rates = 1Mb/s
 - 19 minutes of data for a year of science data
 $840,606 \text{ km}^2$
- SNR?
 - Is the signal to noise useful at that resolution?



Georgia and South Carolina
MODIS True-color image 20 November 2001

140,000 km² area, red insert represents ~9,500km²



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SPOC: Payload Solution

- “Smart” Multi Spectral Imager
 - 16 bands
 - 433 - 866 nm
 - Minimum 130 m spatial and 4nm spectral resolution
- Data rates
 - Area covered increased to 5,500,000km² per year
- SNR
 - Increased to >100 for most bands

Wavelength (nm)	Bandwidth (nm)	SNR per 20 pixels
443	20	181
490	20	185
510	20	171
555	20	157
670	20	139
750	20	83
865	20	63

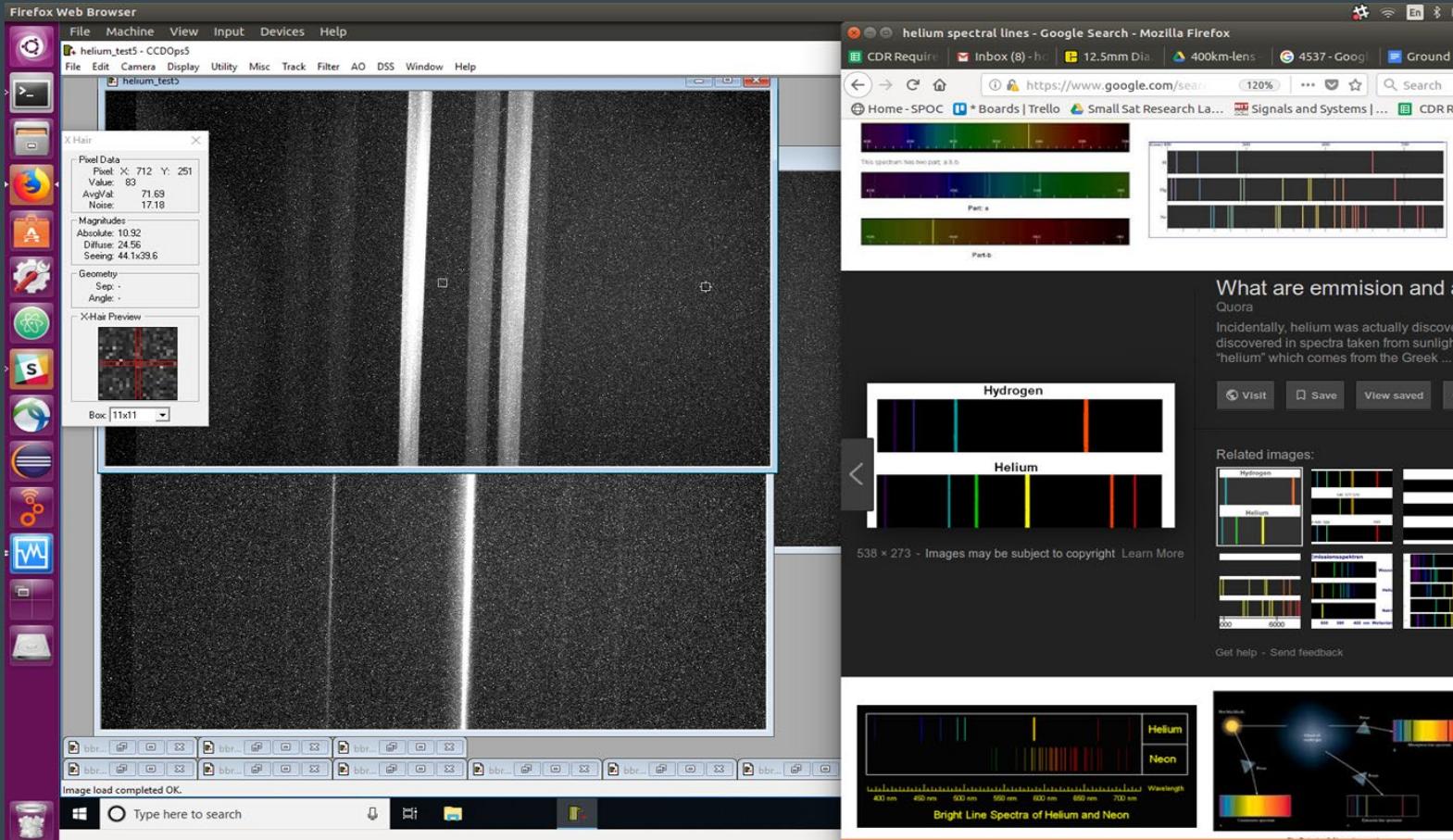
Estimated SNR of corresponding SeaWiFS bands



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Test Results



Optical test bench setup

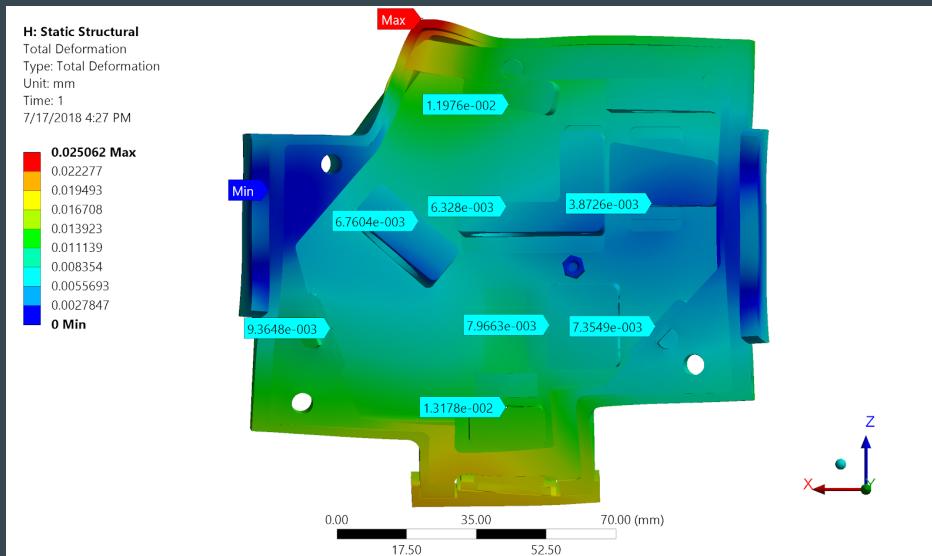


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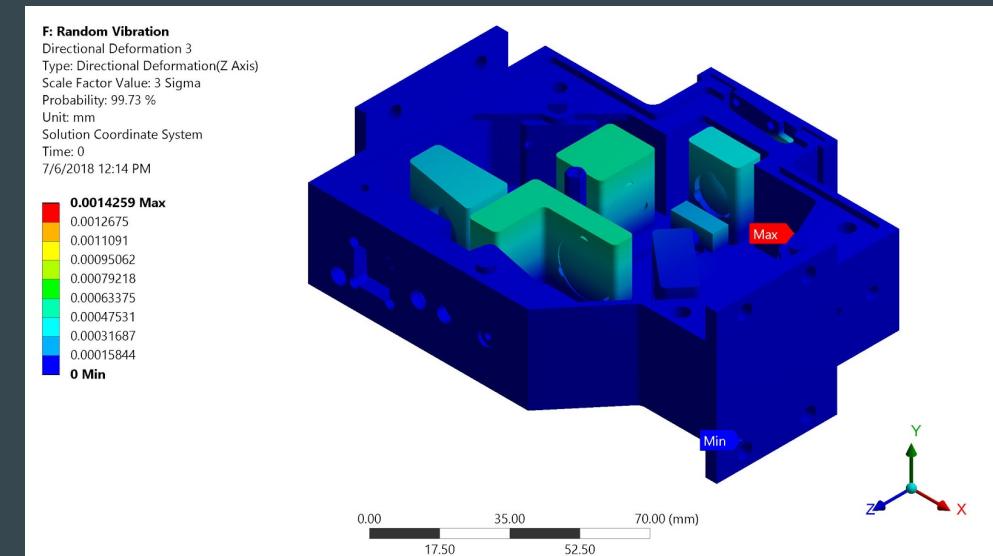


Lab Based Calibration

- Engineering model is nearing completion
- Flight model is currently being machined
- Working with Brandon Zimmerman at NASA Goddard for lab based calibration and characterization
- Lock down mounts once optimal configurations are met



SPOC payload housing thermal simulations



SPOC payload housing vibration simulations



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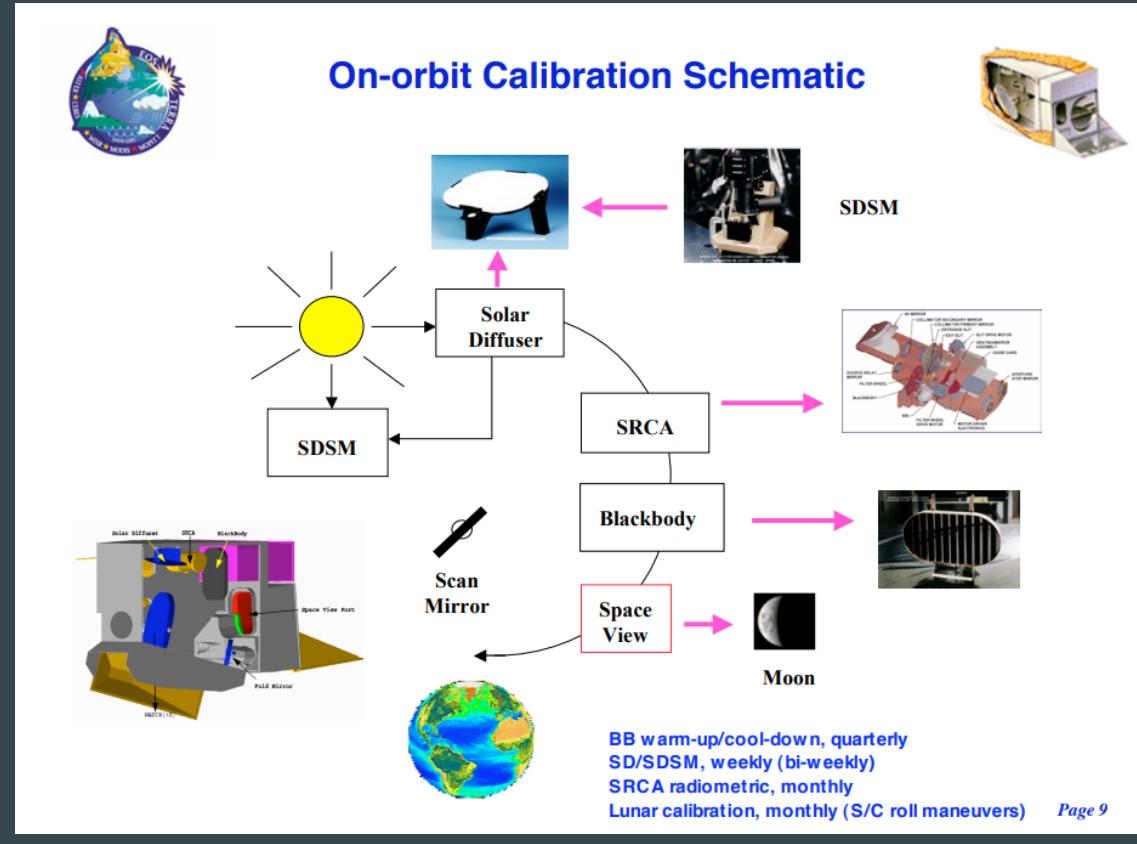
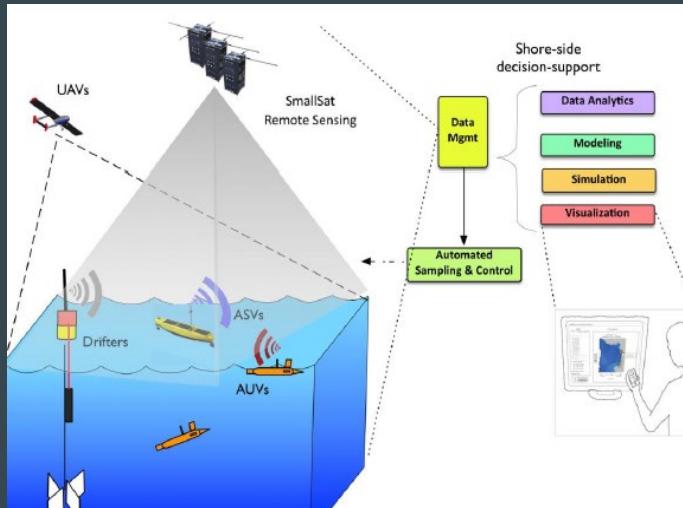


On orbit calibration

- Pseudo invariant targets
- Vicarious calibration with MODIS, SeaHawk, and Norwegian University of Science and Technology (NTNU) hyperspectral imaging small sat mission



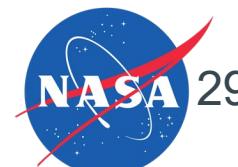
Hawkeye payload for SeaHawk



MODIS On-orbit Calibration Methodologies (Xiong and Barnes CEOS-IVOS at ESA / ESTEC, Noordwijk, Netherlands (12-14 October, 2004))



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SPOC: Data

- Primary targets
 - Sapelo Island LTER
 - Grand Bay NERR
- Intense field campaigns synced with over pass times



MONITORING-PAM chlorophyll fluorometer at
Grand Bay site



Bayou Sauvage research site



Sapelo Island LTER eddy covariance tower



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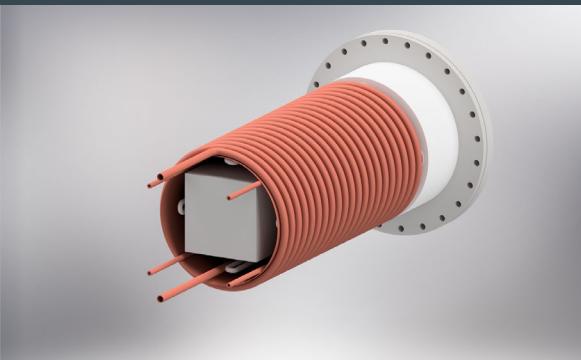


SSRL Facilities

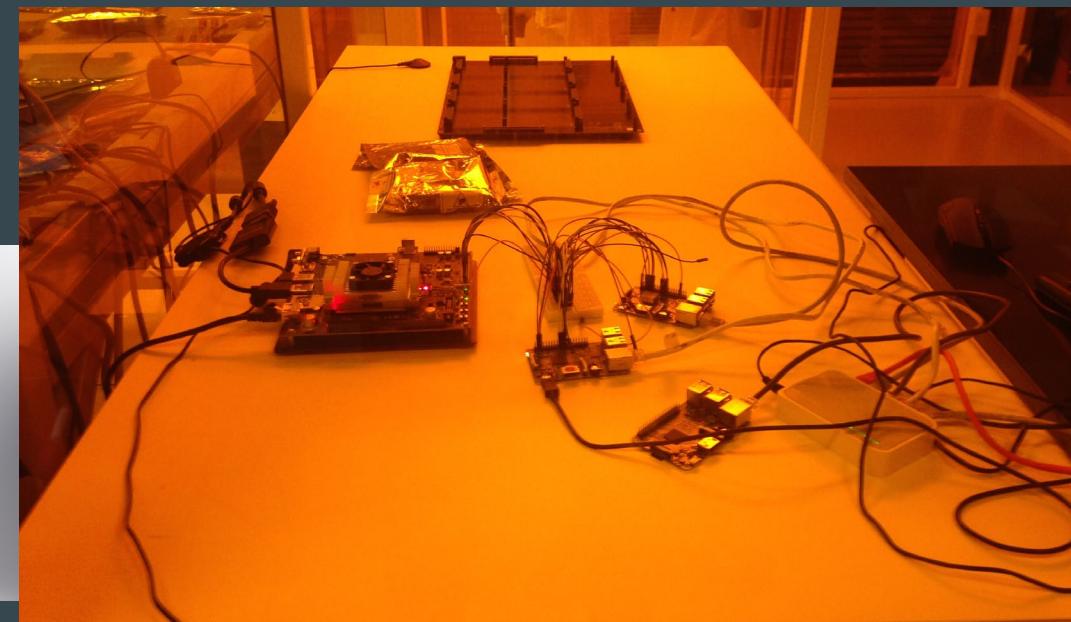
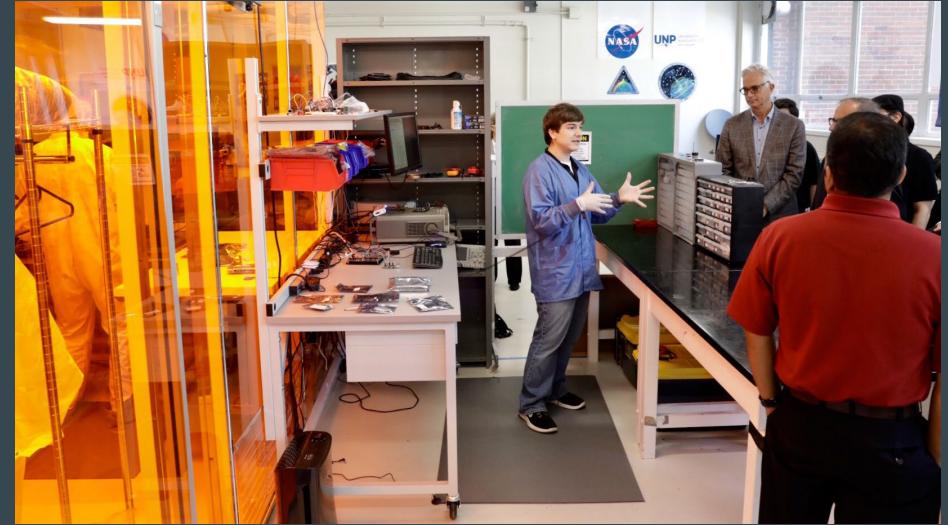
- Clean room
 - ISO 6 and 7
- T-Vac Chamber
 - 10^{-6} Torr
 - -100 K – 150 K
- Ground Station



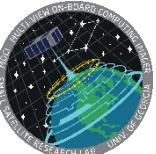
T-Vac Chamber from an old MBE



Thermal shroud for CubeSats



Engineering level components undergoing testing

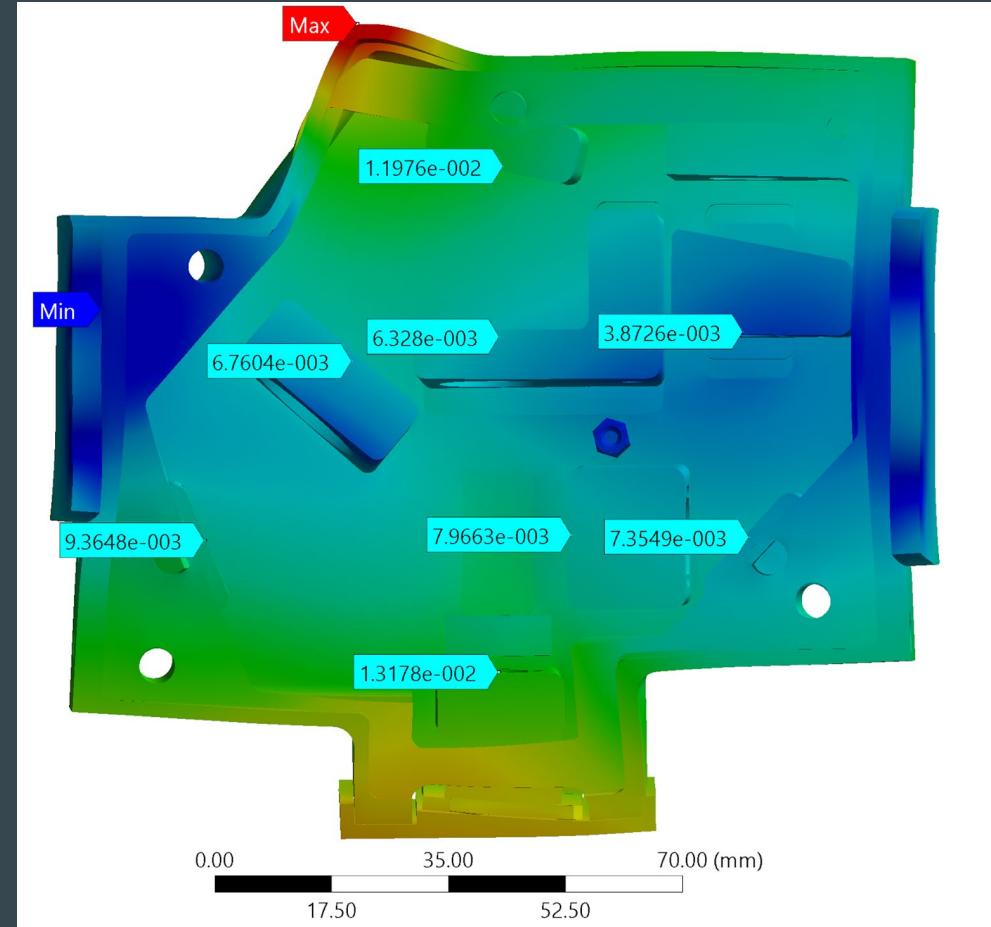


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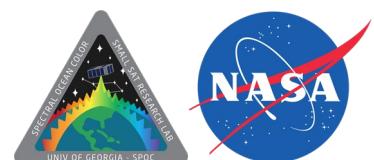
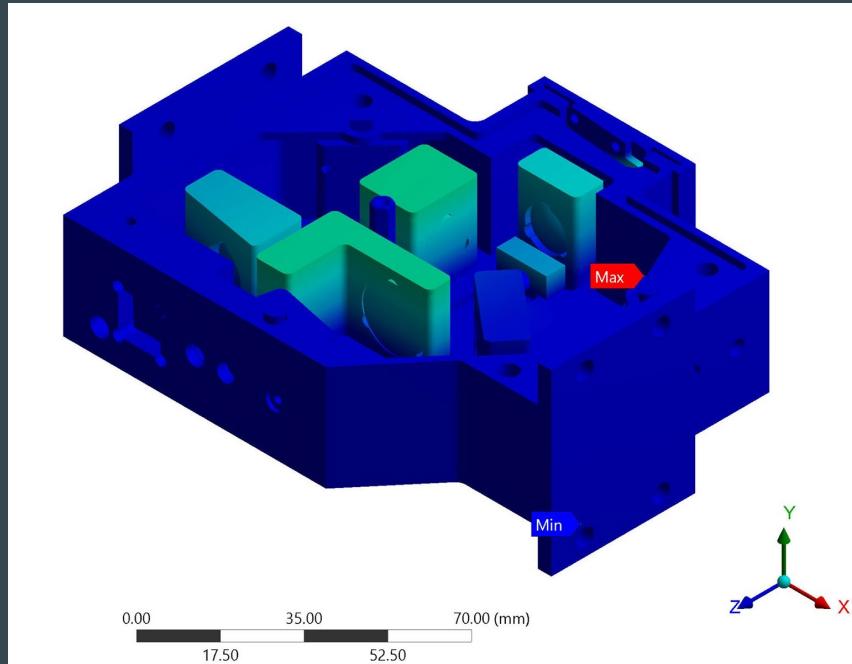
Payload Thermal Simulation

- ANSYS 18.2 Thermal Simulation of Camera Assembly to determine relative displacement and total stress of optics in a heat soaked environment
 - -10° C - 30° C
- Calculated total displacement of optics subassemblies to ensure minimal movement
- Results present possible movement of optic focal points while in orbit



Payload Structural Simulation

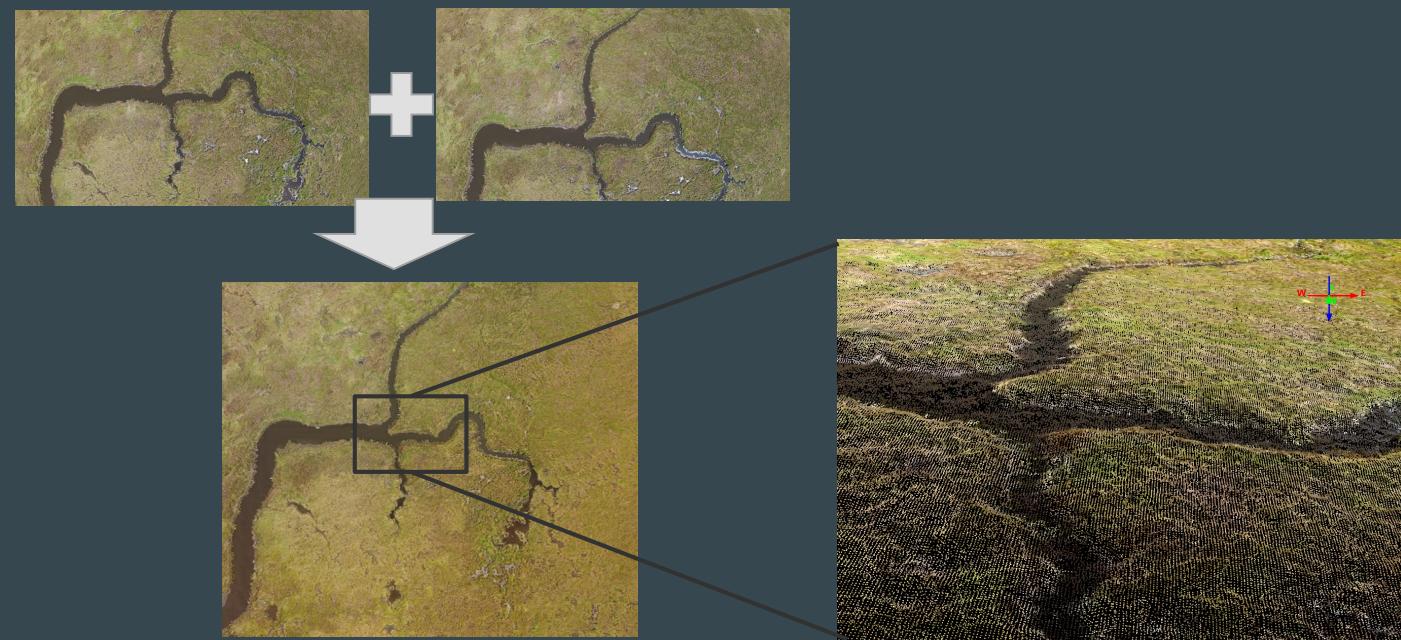
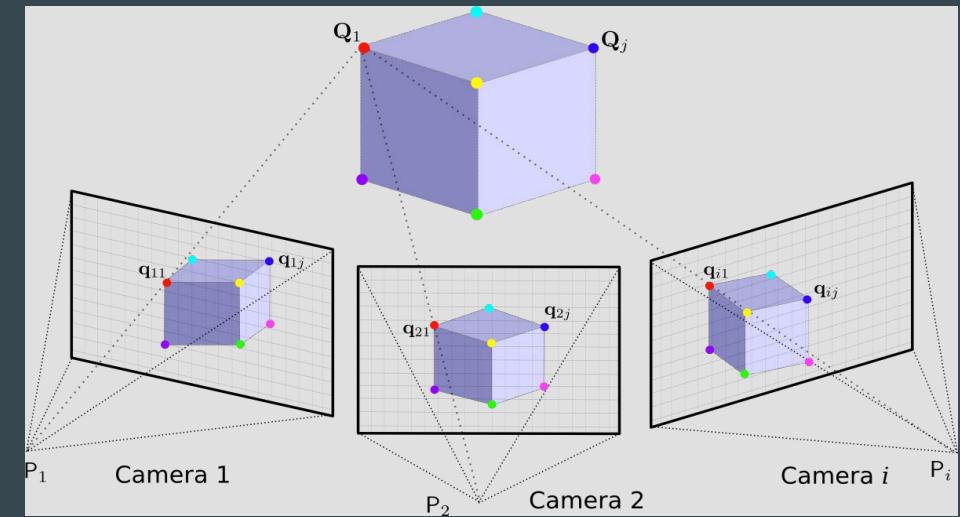
- Modal analysis, inertial loading, response spectrum analysis, and a random vibration analysis.
- Analysis executed on Camera Assembly and individual SPOCeye subassemblies
- Camera Assembly first mode: 2022.0 Hz
- Inertial acceleration of 42Gs, or 412.02 m s^{-2}
 - Calculated FOS of ~32.905 from results
- Response Spectrum maximum displacement of 0.52341 mm
- Random Vibration maximum deformation is 9.0161E-3 mm with a statistical probability of 99.73%.
 - Occurs on Camera Assembly in +YY orientation



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MOCI: Structure from Motion

- Input is a set of 2D images, output is a 3D structure
- Generates a point cloud from multiple images from multiple angles
- Sort of like saying “cloud computing” or “Big Data”, it’s really just a buzz word with lots of complicated parts



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Credit: Caleb Adams



MOCI: Simulating SfM

- Takes in the .json as an arg
- Runs through workflow
 - Workflow can be stopped at discrete steps
- Outputs:
 - DEM/DSM
 - Feature Set
 - Dense/Sparse Point Cloud
 - GeoTiff (Raster)

`./main.py config.json`

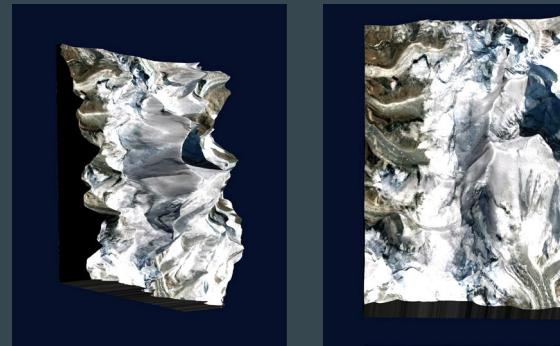
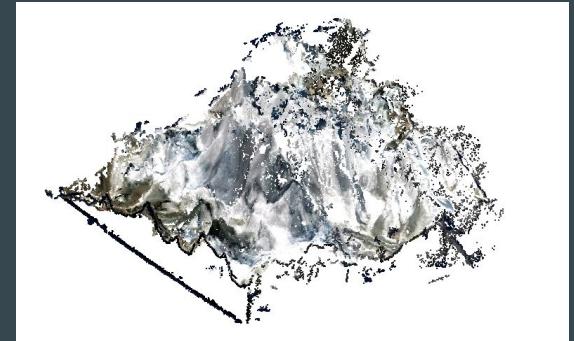
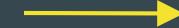
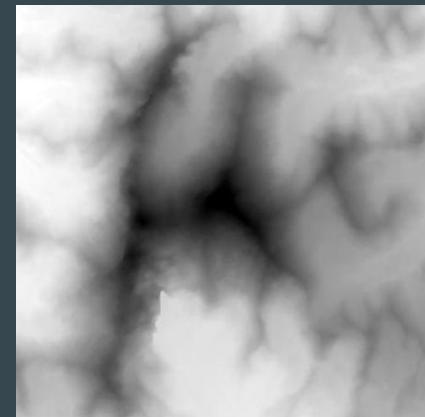


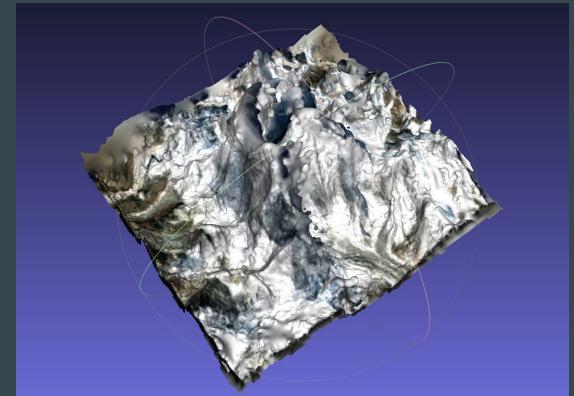
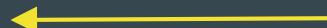
image acquisition



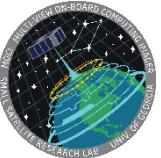
point cloud generation



GeoTiff Generation (Rasterization)



surface reconstruction

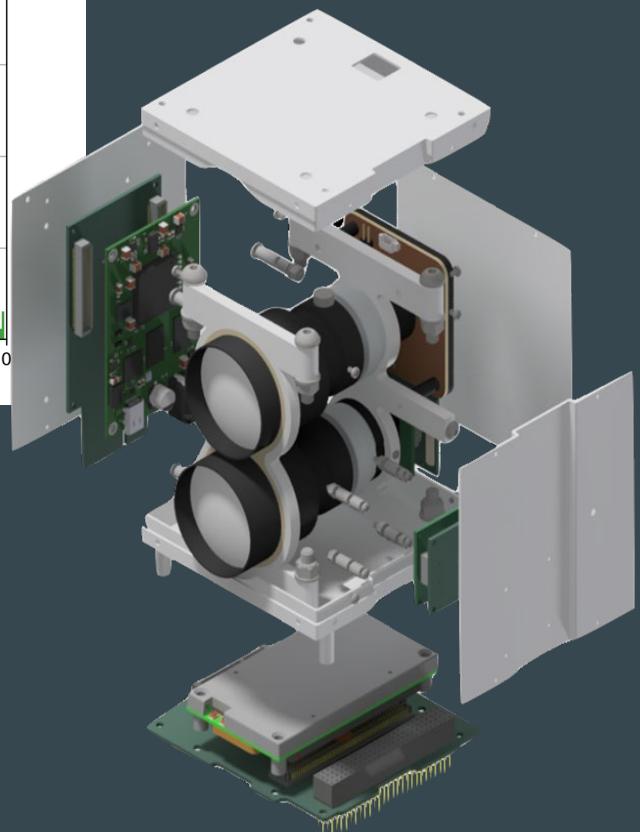
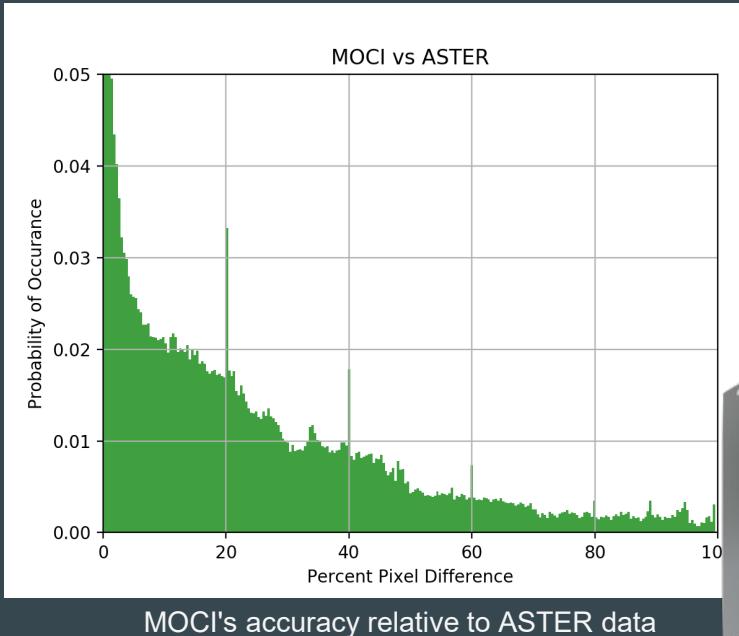


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Credit: Caleb Adams



Current Results

- After 500+ tests...
- Testing with a custom 6.4 m GSD camera
- Comparing DSM/DEM's with ASTER data (15m GSD data)
- Allows for better reconstructions!
- Simulations defined, finalized, confirmed our hardware!
 - Custom 6.4m GSD camera
 - Integrated FPGA (Opal Kelly XEM7310)
 - Integrated GPU (TX2)



The MOCI 1U bionicle payload

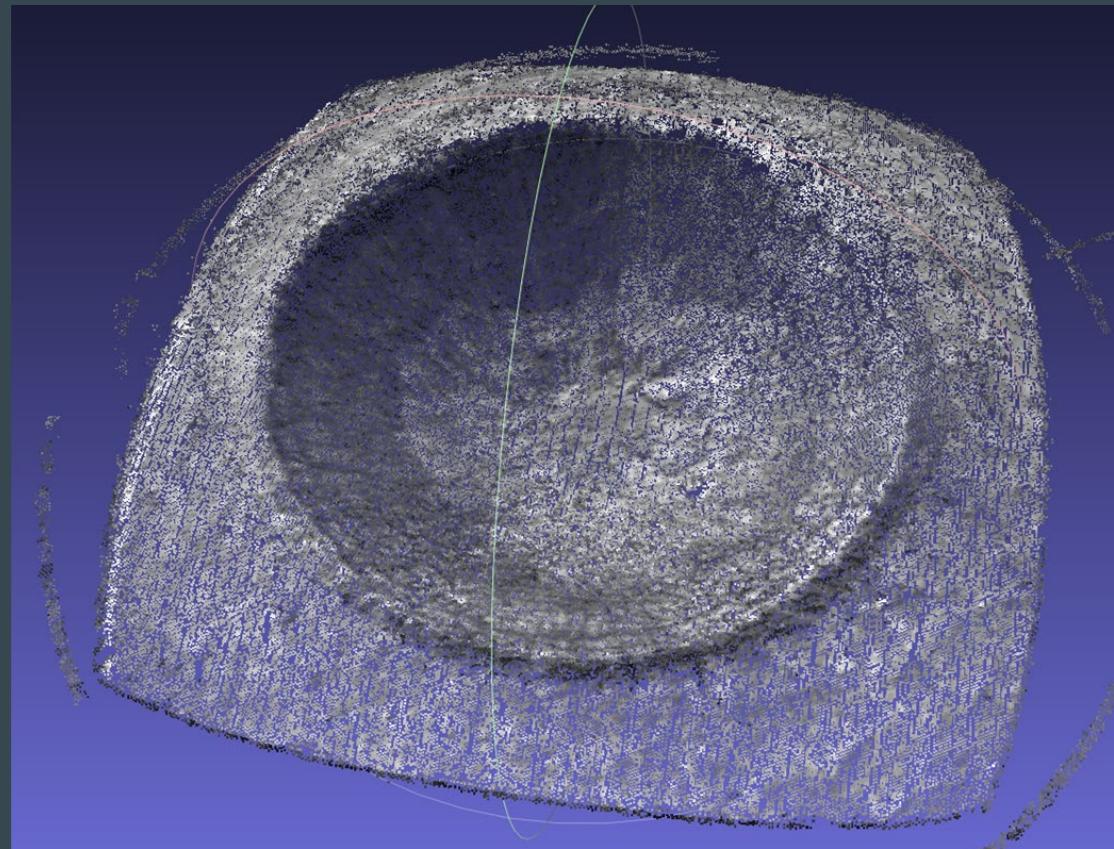


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Credit: Caleb Adams



MOCI Optimization & Future Plans

- Now that we have proven feasibility...
- Use constrained geometry advantageously
- AI & neural nets to get better workflows
- Can be used on other satellites
 - Onboard image analysis
 - Autonomous space operations



Dense point cloud of Lenne crater on the moon, using MOCI & simulations

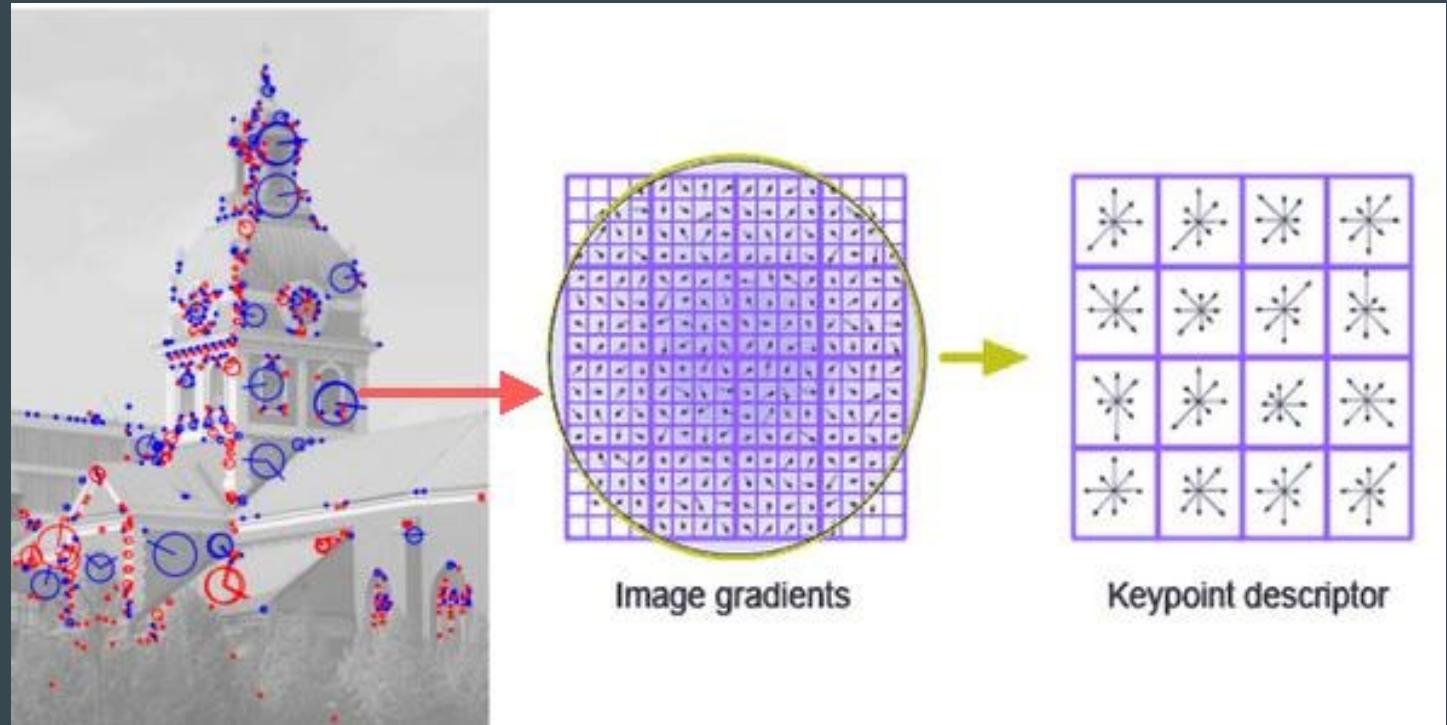


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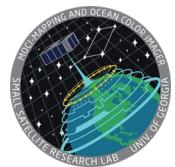


Simulating SfM - Feature Extraction

- Scale Invariant Feature Transform (SIFT) Algorithm
 - Implemented on an FPGA
 - Also CUDA implementation
- Features then need to be matched (we're glossing over that step here, may be added in the future)



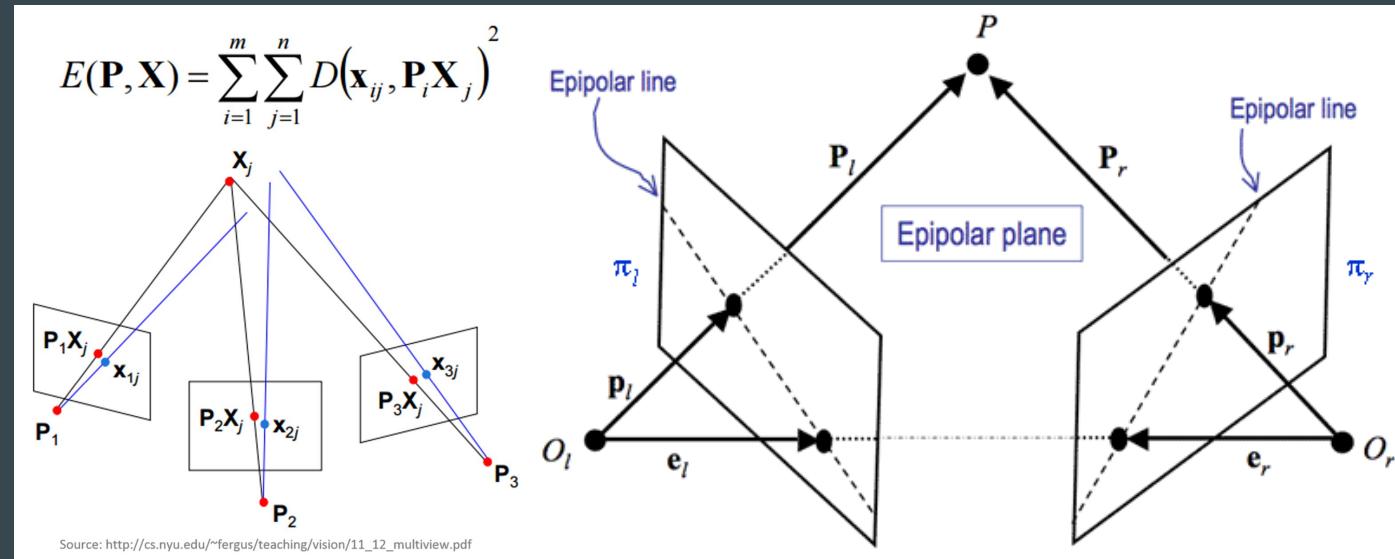
<http://www.codeproject.com/KB/recipes/619039/SIFT.JPG>



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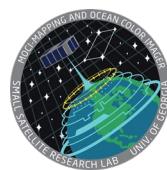
Simulating SfM - Bundle Adjustment

- Addition of the Epipolar Constraint may remove Bundle Adjustment
 - Camera position is known
 - Center of rotation is known
- Given the set of image coordinates x_{ji} find the set of camera matrices, P_i , and the points X_j such that $P_i X_j = x_{ji}$ This is known as project reconstruction
- With known P_i , the epipolar constraint could be used to make a sparse point cloud generation faster than a pure bundle adjustment approach



bundle adjustment

Epipolar constrained reconstruction
<https://www.cs.auckland.ac.nz>



UNP



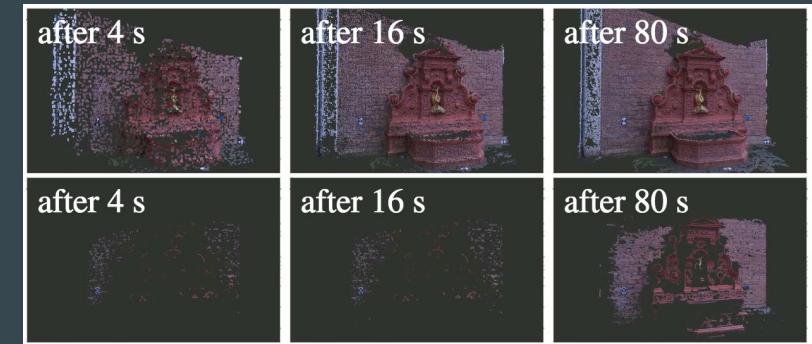
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Simulating SfM - Point Cloud Reconstruction

- Patch-based Multi-View Stereo (pmvs)
- Hierarchical Progressive Multi-View Stereo (hpmvs)
- Runs a dense reconstruction from the sparse reconstruction
 - after a sparse bundle adjustment
 - after an epipolar constrained reconstruction



pmvs - <https://www.di.ens.fr/pmvs/pmvs-1/images/overflow.jpg>



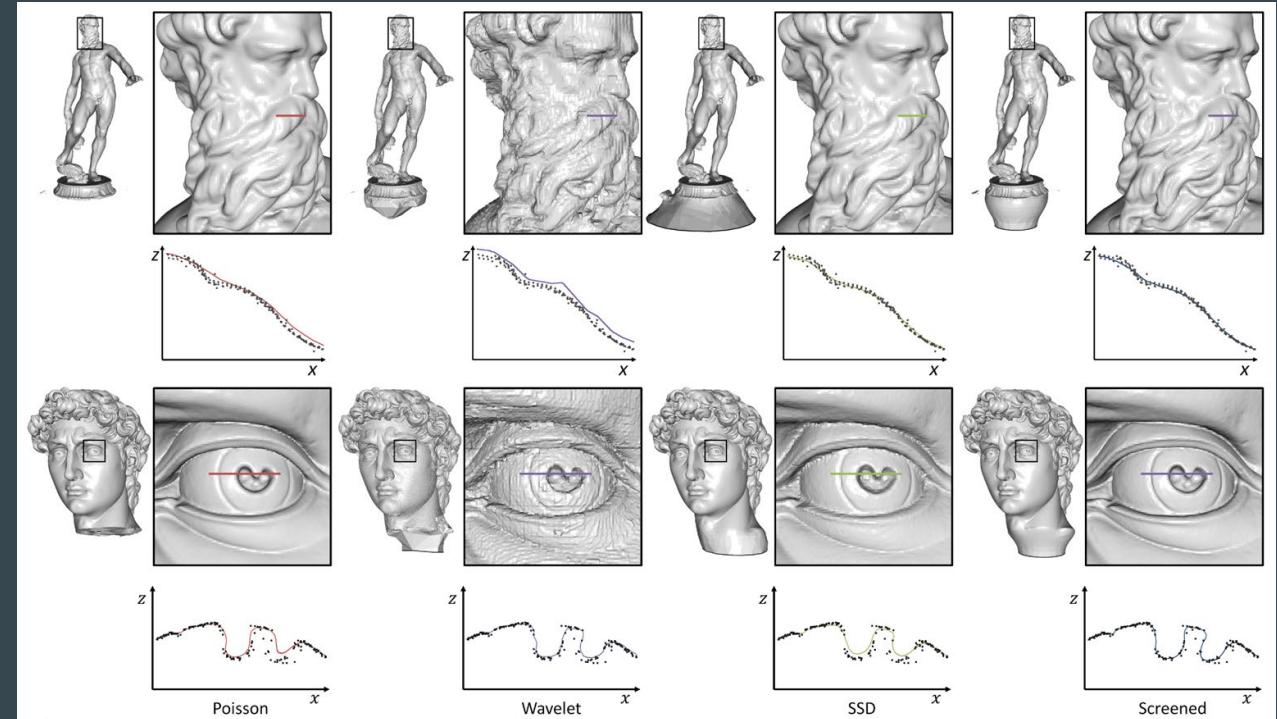
hpmvs - https://www.cv-foundation.org/openaccess/content_cvpr_2016/papers/Locher_Progressive_Prioritized_Multi-View_CVPR_2016_paper.pdf



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Simulating SfM - Surface Reconstruction

- Given a set of oriented points, build a 3D model from those points to approximate the original model
- Screened Poisson Surface Reconstruction
- Texturing post-reconstruction



<http://www.cs.jhu.edu/~misha/MyPapers/ToG13.pdf>

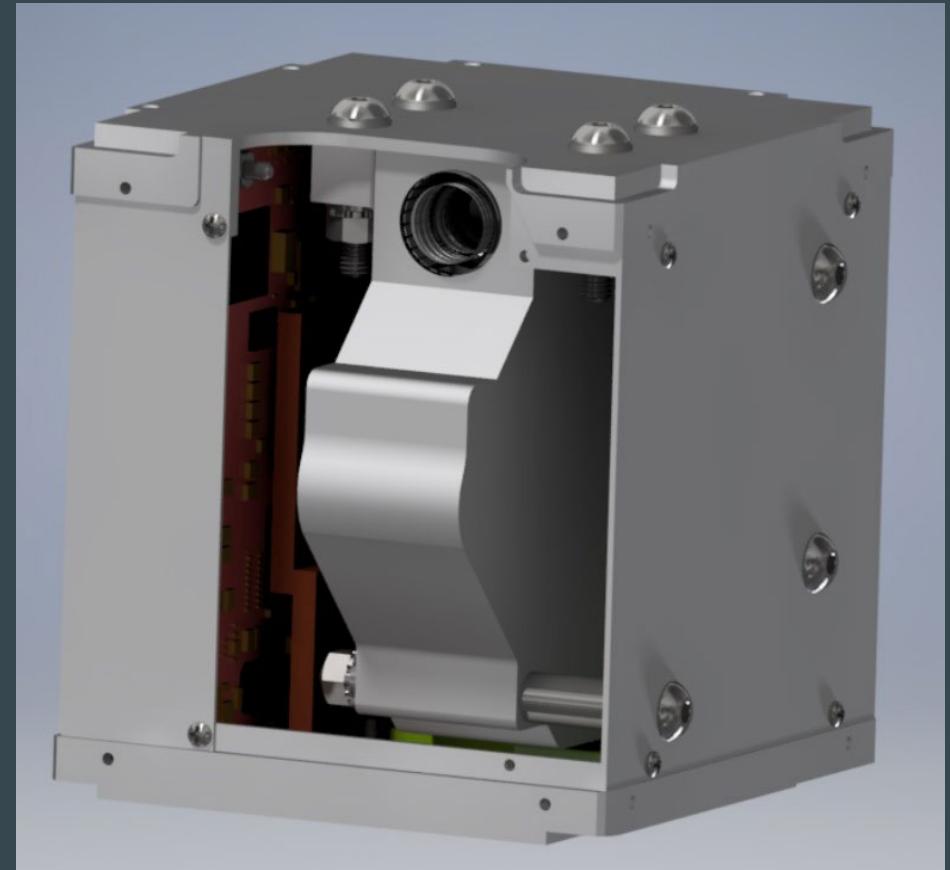


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Satisfaction of Mission Requirements

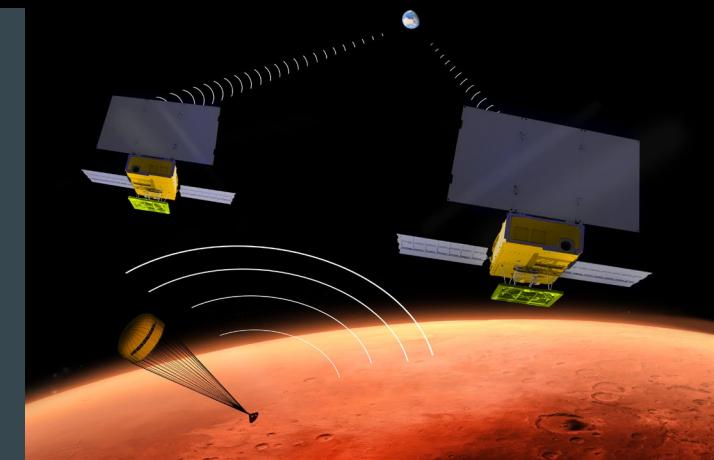
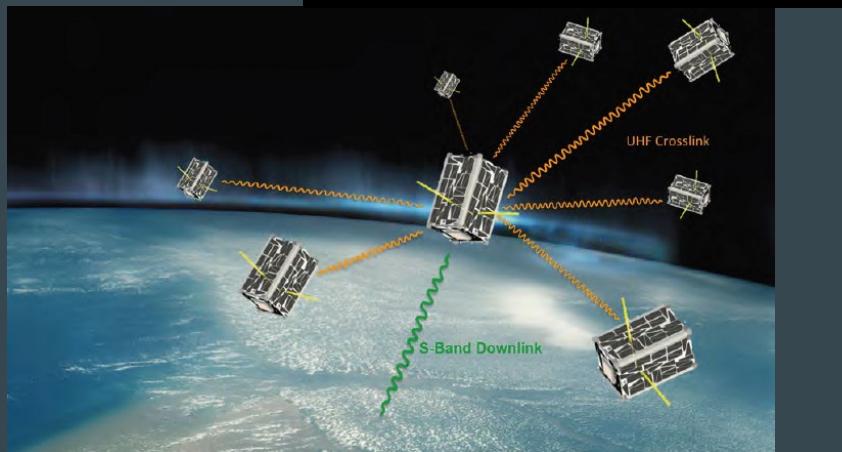
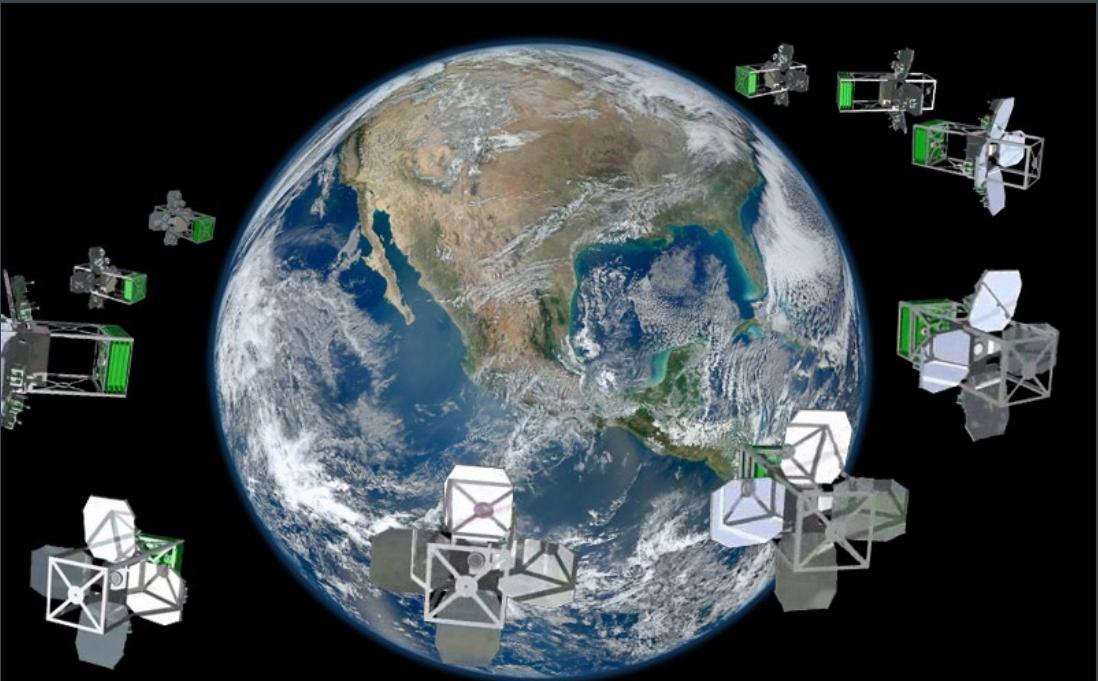
Still in the building and testing phases but have thus far been successful in:

- Optics Design
 - 131 meter ground resolution
 - Images acquired will be between 400 and 850nm
- Mechanical Design
 - Approx. mass (+10% contingency): 0.713 kg
 - 105mm x 96 mm x 97 mm
 - Three adjustable lenses and diffraction grating for ground calibration



Future Goals

- Expand missions beyond Earth Observations
 - Interferometry
 - Planetary atmospheres
- Design missions to accompany traditional satellites



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