

Advanced Dynamics - Assignment Report

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Contents

1	Assignment 1	3
1.1	4
1.1.1	4
1.1.2	5
1.1.3	5
1.2	6
2	Assignment 2	10
2.1	Analysis	11
3	Assignment 3	12
3.1	12
3.2	12
3.3	12
3.4	12
3.5	Plugging in the values:	13
3.6	Equilibrium State	14
4	Assignment 4	15

1 Assignment 1

1 WHEEL ROLLING WITHOUT SLIPPING ON A 2D TRACK

A thin wheel of radius R rolls without slipping on a track on the $x_1 - x_2$ plane, defined by $x_2 = f(x_1)$. The wheel plane stays vertical and tangent to such track at the contact point P . Denote with α the angle the disk plane forms with the x_2 axis, and with ϕ the rotation of the disk about its axis \mathbf{e}_ϕ . The position of the center of the disk C is indicated by x_1^C , x_2^C and x_3^C . Assume a set of generalized coordinate $\mathbf{q} = [x_1^C \ x_2^C \ x_3^C \ \alpha \ \phi]$.

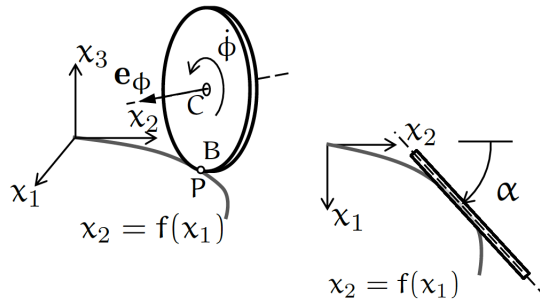


Figure 1.1: Wheel rolling without slipping on a track.

1. State all the constraints acting on the disk.
2. Determine whether the constraints are holonomic or non-holonomic.

Figure 1: Task 1.1

1.1

1.1.1

The list of constraints looks as follows:

1. The wheel always stays vertical (plane parallel to x_3)
This is a holonomic constraint: $f = \theta = 0$ where θ denotes the angle between the disk and x_3
2. x_2 follows a fixed trajectory, given x_1 (and vice versa):
 $f_2 : x_2 = f(x_1) \Rightarrow x_2 - f(x_1) = 0 \Rightarrow$ holonomic.
3. α is the angle between the trajectory and the x_2 axis:
 $\alpha = \frac{\pi}{2} - \frac{\partial f(x_1)}{\partial x_1}$ or written differently:
 $f(\alpha, x_1) = \alpha - \frac{\pi}{2} + \frac{\partial f(x_1)}{\partial x_1} = 0 \Rightarrow$ holonomic
4. Rolling without slipping:
 $v^B = 0 \Rightarrow v_C + \omega \times R_{CB} = 0$ with $\omega = \dot{\alpha}\mathbf{e}_3 + \dot{\phi}\mathbf{e}_\phi$
Seems to be non-holonomic at first glance
5. The disk does not leave the ground:
 $x_3^C - R = 0$ aka the x_3 component of the center of mass is R.
This is holonomic as well

So far we have a 3D system (6 DoF) and 4 holonomic constraints and 1 non-holonomic constraint.

4. Check for integrability:

$$v^B = 0 \Rightarrow v_C + \omega \times R_{CB} = 0 \quad (1)$$

Plugging in $\dot{x}_1^C, \dot{x}_2^C, \omega = \dot{\alpha}\mathbf{e}_3 + \dot{\phi}\mathbf{e}_\phi$ and $R_{CB} = [0, 0, -R]^T$:

$$\dot{x}_1^C \mathbf{e}_1 + \dot{x}_2^C \mathbf{e}_2 - R\dot{\phi} \cos \alpha \mathbf{e}_2 - R\dot{\phi} \sin \alpha \mathbf{e}_1 = 0 \quad (2)$$

Considering the part in \mathbf{e}_1 direction:

$$\dot{x}_1^C - R\dot{\phi} \sin \alpha = 0 \quad (3)$$

As α is not dependent on time it can be easily seen that 3 is integrable:

$$\begin{aligned} \int \dot{x}_1^C dt &= \int R\dot{\phi} \sin \alpha dt \\ \Rightarrow x_1^C &= R\phi \sin \alpha \end{aligned} \quad (4)$$

1.1.2

See subsubsection 1.1.1

1.1.3

3. Determine the degrees of freedom of the system.

Figure 2: Task 1.1.3

As we have a 3D body with 6 generalized coordinates (here 5 are given, already considering constraint 1) and 5 holonomic constraints. We get a total of $6 - 5 = 1$ degree of freedom. That could for instance be the rotation of the wheel around \mathbf{e}_ϕ while all the other generalized coordinates follow accordingly.

2 TWO BARS LINKAGE

Two bars AB and BC of equal length L are hinged at point B, and move in the plane spanned by the unit vectors \mathbf{e}_1 and \mathbf{e}_2 . The velocity \mathbf{v}_C of point C is required to be directed towards point A at all times, as shown. Show that such constraint is non-holonomic. Use x_1^B, x_2^B, θ_1 and θ_2 as generalized coordinates.

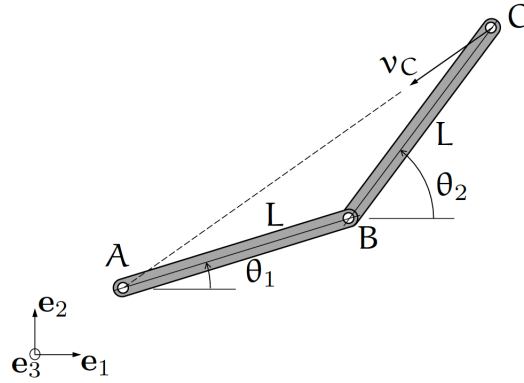


Figure 2.1: A two bar linkage in 2D. The velocity \mathbf{v}_C must be directed towards A at all times.

Figure 3: Task 1.1

1.2

I want to show that the constraint that \mathbf{v}_C always points in the direction of \mathbf{AC} is not integrable. I would like to express this constraint as:

$$\mathbf{v}_C \cdot \mathbf{AC}_p = 0 \quad (5)$$

where \mathbf{AC}_p is a vector perpendicular to \mathbf{AC}

The velocity of point C can be found either by expressing the position of C and derivating or using the velocity transfer formula from B to C which yields:

$$\mathbf{v}_C = \begin{pmatrix} \dot{x}_1^B - L \sin(\theta_2) \dot{\theta}_2 \\ \dot{x}_2^B + L \cos(\theta_2) \dot{\theta}_2 \end{pmatrix} \quad (6)$$

And \mathbf{AC} is simply:

$$\mathbf{AC} = \begin{pmatrix} L (\cos \theta_1 + \cos \theta_2) \\ L (\sin \theta_1 + \sin \theta_2) \end{pmatrix} \quad (7)$$

A simple vector that is perpendicular to AC can be gotten by switching the e_1 and e_2 entries and switching the sign of one of them:

$$AC_p = \begin{pmatrix} L(\sin \theta_1 + \sin \theta_2) \\ -L(\cos \theta_1 + \cos \theta_2) \end{pmatrix} \quad (8)$$

$$\begin{aligned} \Rightarrow v_C * AC_p &= \begin{pmatrix} \dot{x}_1^B - L \sin(\theta_2) \dot{\theta}_2 \\ \dot{x}_2^B + L \cos(\theta_2) \dot{\theta}_2 \end{pmatrix} * \begin{pmatrix} L(\sin \theta_1 + \sin \theta_2) \\ -L(\cos \theta_1 + \cos \theta_2) \end{pmatrix} = \\ & (L(\sin(\theta_1) + \sin(\theta_2))) \dot{x}_1^B + (-L(\cos(\theta_1) + \cos(\theta_2))) \dot{x}_2^B - \\ & L^2 \dot{\theta}_2 \cos(\theta_2) (\cos(\theta_1) + \cos(\theta_2)) - L^2 \dot{\theta}_2 \sin(\theta_2) (\sin(\theta_1) + \sin(\theta_2)) = 0 \end{aligned} \quad (9)$$

After applying trigonometry:

$$\begin{aligned} & (L(\sin(\theta_1) + \sin(\theta_2))) \dot{x}_1^B + (-L(\cos(\theta_1) + \cos(\theta_2))) \dot{x}_2^B - \\ & L^2 \dot{\theta}_2 (\cos(\theta_2) \cos(\theta_1) + \sin(\theta_2) \sin(\theta_1) + 1) = 0 \end{aligned} \quad (10)$$

To simplify the notation I will refer to $\sin \theta_1$ as s_1 and $\sin \theta_2$ as s_2 :

$$L \dot{x}_1^B (s_1 + s_2) - L \dot{x}_2^B (c_1 + c_2) - L^2 \dot{\theta}_2^2 (c_2 c_1 + s_2 s_1 + 1) = 0 \quad (11)$$

Dividing by L:

$$\dot{x}_1^B (s_1 + s_2) - \dot{x}_2^B (c_1 + c_2) - L \dot{\theta}_2^2 (c_2 c_1 + s_2 s_1 + 1) = 0 \quad (12)$$

With the generalized coordinates $q = [x_1^B, x_2^B, \theta_1, \theta_2]$

Writing down the coefficients of the non-holonomic constraint:

$$\begin{aligned} a_1 &= s_1 + s_2 \\ a_2 &= -(c_1 + c_2) \\ a_3 &= 0 \\ a_4 &= -L(1 + s_1 s_2 + c_1 c_2) \\ b &= 0 \end{aligned} \quad (13)$$

To check for the exact velocity form we want to proof that there can't exist a $C(q) \neq 0$ for which holds:

$$\frac{\partial(Cb)}{\partial q_i} = \frac{\partial(Ca_1)}{\partial t} \quad \text{and} \quad \frac{\partial(Ca_i)}{\partial q_k} = \frac{\partial(Ca_k)}{\partial q_i} \quad \text{for all the gen. coord.} \quad (14)$$

q1-q3:

$$\begin{aligned} \frac{\partial(Ca_1)}{\partial \theta_1} = 0 &\Rightarrow C * c_1 + \frac{\partial C}{\partial \theta_1}(s_1 + s_2) = 0 \\ \Rightarrow \frac{c_1}{s_1 + s_2} + \frac{1}{C} \frac{dC}{d\theta_1} = 0 &\Rightarrow \int \frac{1}{C} dC = - \int \frac{c_1}{s_1 + s_2} d\theta_1 \\ \Rightarrow \ln C = \frac{D}{s_1 + s_2} + D &\Rightarrow C = \frac{D}{s_1 + s_2} \quad \text{where the integration const. D was updated} \\ \Rightarrow C = \frac{D(x_1^B, x_2^B, \theta_2)}{s_1 + s_2} & \end{aligned} \quad (15)$$

q1 - q4:

$$\frac{\partial Ca_1}{\partial \theta_2} = \underbrace{\frac{\partial Ca_4}{\partial x_1^B}}_0 = 0 \Rightarrow \frac{\partial D}{\partial \theta_2} = 0 \Rightarrow C = \frac{D(x_1^B, x_2^B)}{s_1 + s_2} \quad (16)$$

q1 - q2:

$$\begin{aligned} \frac{\partial Ca_1}{\partial x_2^B} = \frac{\partial Ca_2}{\partial x_1^B} &\Rightarrow \frac{\partial C}{\partial x_2^B}(s_1 + s_2) = \frac{\partial C}{\partial x_1^B}(c_1 + c_2) \Rightarrow \frac{\partial C}{\partial x_2^B} = \frac{\partial C}{\partial x_1^B} \frac{c_1 + c_2}{s_1 + s_2} \\ \text{Remembering that } D = D(x_1^B, x_2^B) &\Rightarrow D = \text{const.} \\ \Rightarrow C = \frac{D}{s_1 + s_2} &\text{ where D const.} \end{aligned} \quad (17)$$

q2 - q3:

$$\begin{aligned} \frac{\partial Ca_1}{\partial x_2^B} = 0 &\Rightarrow D \frac{\partial \frac{(c_1 + c_2)}{s_1 + s_2}}{\partial \theta_1} = 0 \\ D \frac{\partial \frac{c_1}{s_1 + s_2}}{\partial \theta_1} &= - \frac{Ds_1}{s_1 + s_2} - \frac{Dc_1^2}{(s_1 + s_2)^2} = -D \frac{s_1(s_1 + s_2) - c_1^2}{(s_1 + s_2)^2} \end{aligned} \quad (18)$$

$$D \frac{\partial \frac{c_2}{s_1 + s_2}}{\partial \theta_1} = -D \frac{c_1 c_2}{(s_1 + s_2)^2} \quad (19)$$

Which leads to:

$$-D \frac{c_1^2 + s_1^2 + s_1 s_2 + c_1 c_2}{(s_1 + s_2)^2} \stackrel{!}{=} 0 \quad (20)$$

Which is a contradiction. Therefore this is a non-holonomic constraint.

2 Assignment 2

2.1 Hamilton's Principle

This assignment asks to get the equations of motion using the Hamilton Principle:

$$\int_{t_1}^{t_2} \delta(T - V + W) dt = 0 \quad (21)$$

Using the theory discussed in the lecture we can split the term into different contributions::

Remembering the result of the Hamilton principle from the lecture:

$$\left[\frac{\partial L}{\partial v} - \frac{\partial}{\partial x} \left(\frac{\partial L}{\partial v'} \right) + \frac{\partial^2}{\partial x^2} \left(\frac{\partial L}{\partial v''} \right) - \frac{\partial}{\partial t} \left(\frac{\partial L}{\partial \dot{v}} \right) + \frac{\partial}{\partial x} \left(\frac{\partial}{\partial t} \left(\frac{\partial L}{\partial \dot{v}'} \right) \right) \right] + f(x, t) + \sum F_i D(x - x_i) = 0, \quad x \in [0; l]$$

Boundary conditions:

$$\left[\frac{\partial L}{\partial v'} - \frac{\partial}{\partial x} \left(\frac{\partial L}{\partial v''} \right) - \frac{\partial}{\partial t} \left(\frac{\partial L}{\partial \dot{v}'} \right) \right] \delta v = 0, \quad x = 0, l$$

$$\frac{\partial L}{\partial v''} \delta v' = 0, \quad x = 0, l$$

Figure 4: Hamilton Formulas for a continuous system

We consider w our variable instead of v .

$$\frac{\partial L}{\partial w} = 0, \quad \frac{\partial}{\partial x} \frac{\partial L}{\partial w'} = 0, \quad (22)$$

$$\frac{\partial^2}{\partial x^2} \frac{\partial L}{\partial w''} = -E \frac{\partial^2}{\partial x^2} I(x), \quad \frac{\partial}{\partial t} \frac{\partial L}{\partial \dot{w}} = 2\delta A(x) \ddot{w}, \quad (23)$$

$$\frac{\partial}{\partial x} \left(\frac{\partial}{\partial t} \left(\frac{\partial L}{\partial \dot{w}'} \right) \right) = 2\rho I(x) \ddot{w}'' + 2\rho I(x)' \dot{w}' \quad (24)$$

$$f(x, t) = p(x, t) \quad (25)$$

$$\frac{\partial L}{\partial w'} = 0, \quad \frac{\partial}{\partial x} \left(\frac{\partial L}{\partial w''} \right) = -EI(x)' \quad (26)$$

$$\frac{\partial}{\partial t} \left(\frac{\partial L}{\partial \dot{w}'} \right) = 2\rho I(x) \ddot{w}' \quad (27)$$

Therefore the green part comes to:

$$-E \frac{\partial^2}{\partial x^2} I(x) - 2\rho A(x) \ddot{w} + 2\rho I(x) \ddot{w}'' + 2\rho I(x)' \ddot{w}' + p(x, t) = 0 \quad (28)$$

$$-E \frac{\partial^2}{\partial x^2} I(x) - 2\rho A(x) \ddot{w} + 2\rho I(x) \ddot{w}'' + 2\rho I(x)' \ddot{w}' + p(x, t) = 0 \quad (29)$$

While for the red we get:

$$\left[EI(x)' - 2\rho I(x) \ddot{w}' \right] \delta w = 0, \quad x = 0, L \quad (30)$$

And for lastly for the blue:

$$EI(x) \delta w' = 0, \quad x = 0, L \quad (31)$$

2.2 Analysis

3 Assignment 3

3.1

Lagrange equations:

$$\begin{aligned} \frac{L^2 M \ddot{\phi}}{3} + \frac{L^2 m \ddot{\phi}}{3} - \frac{L^2 m \ddot{\phi} \cos(\beta)^2}{3} + \frac{LMg \cos(\beta) \sin(\phi)}{2} + \frac{L^2 M \Omega^2 \sin(\beta) \sin(\phi)}{2} + \\ \frac{L^2 M \Omega^2 \cos(\beta) \sin(\beta) \sin(\phi)}{4} - \frac{L^2 M \Omega^2 \cos(\beta)^2 \cos(\phi) \sin(\phi)}{6} = 0 \end{aligned} \quad (32)$$

3.2

This can be reformulated for the differential equation of $\ddot{\phi}$:

$$\begin{aligned} \ddot{\phi} \left(\frac{L^2(M+m)}{3} - \frac{L^2 m \cos(\beta)^2}{3} \right) + \sin(\phi) \left(\frac{LMg \cos(\beta)}{2} + \frac{L^2 M \Omega^2 \sin(\beta)}{2} + \frac{L^2 M \Omega^2 \cos(\beta) \sin(\beta)}{4} \right) - \\ \frac{L^2 M \Omega^2 \cos(\beta)^2 \cos(\phi) \sin(\phi)}{6} = 0 \end{aligned} \quad (33)$$

3.3

Equation of motion for the case that the rotation around the vertical bar is not constant:

$$\begin{aligned} \frac{L^2 \ddot{\theta} \left(16M + 16m + 12M \cos(\beta) + 12m \cos(\beta) + 2M \cos(\beta)^2 + 4m \cos(\beta)^2 - 2M \cos(\beta)^2 \cos(\phi)^2 \right)}{12} + \\ \frac{L^2 \ddot{\theta} (12M \cos(\phi) \sin(\beta) + 6M \cos(\beta) \cos(\phi) \sin(\beta))}{12} \end{aligned} \quad (34)$$

3.4

After matlab integration of the work per area we get for the work

$W =$

$$\begin{aligned} \frac{L^4 \left(c \left(\dot{\theta} + \dot{\phi} \sin(\beta) \right)^2 \left(\cos(\phi)^2 - 1 \right) - c \left(\dot{\phi} \cos(\phi) + \dot{\theta} \cos(\phi) \sin(\beta) \right)^2 + c \dot{\phi}^2 \cos(\beta)^2 \left(\cos(\phi)^2 - 1 \right) \right)}{3} \\ - L \left(Lc \left(L\dot{\theta} + L\dot{\theta} \cos(\beta) \right)^2 + \frac{L^3 c \dot{\theta}^2 \cos(\beta)^2}{12} \right) - L^3 c \left(L\dot{\theta} + L\dot{\theta} \cos(\beta) \right) \left(\dot{\phi} \cos(\phi) + \dot{\theta} \cos(\phi) \sin(\beta) \right) \end{aligned} \quad (35)$$

Which results in the equations of motion:

$$\begin{aligned} & \frac{L^2 M \ddot{\phi}}{3} + \frac{2L^4 c (\dot{\phi} + \dot{\theta} \sin(\beta))}{3} + \frac{L^2 m \ddot{\phi}}{3} - \frac{L^2 m \ddot{\phi} \cos(\beta)^2}{3} + \frac{LMg \cos(\beta) \sin(\phi)}{2} + \\ & L^4 c \dot{\theta} \cos(\phi) (\cos(\beta) + 1) + \frac{L^2 M \dot{\theta}^2 \sin(\beta) \sin(\phi)}{2} - \frac{L^2 M \dot{\theta}^2 \cos(\beta)^2 \cos(\phi) \sin(\phi)}{6} + \\ & \frac{L^2 M \dot{\theta}^2 \cos(\beta) \sin(\beta) \sin(\phi)}{4} \end{aligned} \quad \text{for } \phi \quad (36)$$

And

$$\begin{aligned} & \frac{cL^4}{6} \left(16\dot{\theta} + 24\dot{\theta} \cos(\beta) + 6\dot{\phi} \cos(\phi) + 4\dot{\phi} \sin(\beta) + 13\dot{\theta} \cos(\beta)^2 - 4\dot{\theta} \cos(\beta)^2 \cos(\phi)^2 + 6\dot{\phi} \cos(\beta) \cos(\phi) + 12\dot{\theta} \cos(\beta) \cos(\phi) \sin(\beta) \right) + \\ & \frac{\ddot{\theta}}{12} \left(16M + 16m + 12M \cos(\beta) + 12m \cos(\beta) + 2M \cos(\beta)^2 + 4m \cos(\beta)^2 - 2M \cos(\beta)^2 \cos(\phi)^2 + \right. \\ & \left. 12M \cos(\phi) \sin(\beta) + 6M \cos(\beta) \cos(\phi) \sin(\beta) \right) L^2 \end{aligned} \quad \text{for } \theta \quad (37)$$

3.5 Plugging in the values:

Let $L = 0.25[m]$, $\beta = \frac{\pi}{6}$, $M = 0.5[kg]$, $m = 0.2[kg]$, $g = 9.81[\frac{m}{s^2}]$ and $c = 0.1[\frac{Ns}{m^3}]$.

The equations become:

$$\begin{aligned} & \frac{\dot{\phi}}{3840} + \frac{11\ddot{\phi}}{960} + \frac{\dot{\theta}}{7680} + \frac{\dot{\theta}^2 \sin(\phi)}{128} + \frac{981\sqrt{3} \sin(\phi)}{3200} + \frac{\sqrt{3}\dot{\theta}^2 \sin(\phi)}{512} + \frac{\dot{\theta} \cos(\phi) \left(\frac{\sqrt{3}}{2} + 1 \right)}{2560} \\ & - \frac{\dot{\theta}^2 \cos(\phi) \sin(\phi)}{256} \end{aligned} \quad \text{for } \phi \quad (38)$$

And

$$\begin{aligned} & \frac{\dot{\phi}}{7680} + \frac{103\dot{\theta}}{61440} - \frac{\dot{\theta} \cos(\phi)^2}{5120} + \frac{\sqrt{3}\dot{\theta}}{1280} + \\ & \frac{\ddot{\theta}}{192} \left(3 \cos(\phi) - \frac{3 \cos(\phi)^2}{4} + \frac{21\sqrt{3}}{5} + \frac{3\sqrt{3} \cos(\phi)}{4} + \frac{251}{20} \right) + \\ & \frac{\dot{\phi} \cos(\phi)}{2560} + \frac{\dot{\theta} \cos(\phi)}{2560} + \frac{\sqrt{3}\dot{\phi} \cos(\phi)}{5120} + \frac{\sqrt{3}\dot{\theta} \cos(\phi)}{5120} \end{aligned} \quad \text{for } \theta \quad (39)$$

3.6 Equilibrium State

Reminder- The potential energy looks as follows:

$$\frac{LMg (2 \sin (\beta) - \cos (\beta) \cos (\phi))}{2} \quad (40)$$

Case 1: Ω fixed:

The derivative of the potential energy w.r.t. θ is always 0.

$$\frac{\partial V}{\partial \theta} = 0 \quad (41)$$

The derivative of the potential energy w.r.t. ϕ not however:

$$\frac{\partial V}{\partial \phi} = \frac{LMg \cos (\beta) \sin (\phi)}{2} \quad (42)$$

Equation 12 is only 0 for $\phi = k * \pi$ which makes sense as in this configuration the square has a momentary velocity in horizontal direction which doesn't change the altitude of any body.

Case 2: $\dot{\theta}$ can change:

As V is independent of θ altogether the result is the same for this case.

The equation of motions are after plugging in equilibrium state:

$$\begin{aligned} \phi : \\ 0 &= 0 \end{aligned} \quad (43)$$

$$\begin{aligned} \theta : \\ \frac{L^2 \ddot{\theta}}{12} (16M + 16m + 12M \cos (\beta) + 12M \sin (\beta) + 12m \cos (\beta) + \\ 4m \cos (\beta)^2 + 6M \cos (\beta) \sin (\beta)) &= 0 \end{aligned} \quad (44)$$

As the coefficient of $\ddot{\theta}$ is constant, the angular acceleration of the vertical shaft has to be 0. This results in a constant angular velocity which recovers the case of a constant Ω .

4 Assignment 4