

Dynamics of a rocket, no gravity or other external forces

Numerical values used for graphs and comparisons

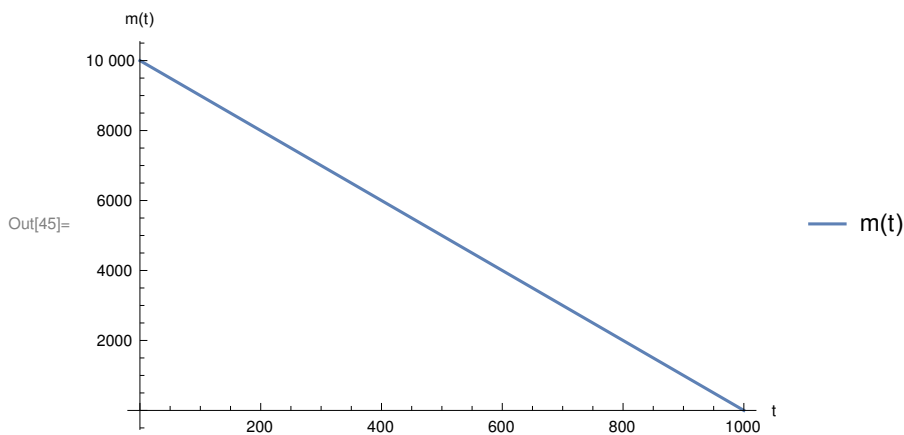
```
In[43]:= params = {m0 → 10 000, R → 10, fthrust → 500, tmax → 500, v0 → 10}
```

```
Out[43]= {m0 → 10 000, R → 10, fthrust → 500, tmax → 500, v0 → 10}
```

Mass vs time, $m(t)$, with $R=dm/dt$ =mass flow rate (exhausted mass per unit of time; R is assumed constant)

```
In[44]:= m[t_] := m0 - R t
```

```
In[45]:= Plot[m[t] /. params, {t, 0, 1000}, PlotLegends → {"m(t)"}, AxesLabel → {"t", "m(t)"}]
```



Ideal rocket equation (Tsiolkovsky rocket equation): $\Delta v = v_e \ln(m_0 / m_f)$, where $v_e = \text{thrust} / R$

(* note: in WolframMathematica, Log[...] is the natural logarithm *)

```
deltav[ve_, m0_, mf_] := ve Log[m0 / mf]
```

```
In[47]:= deltax[ $\frac{fthrust}{R}$ , m[0], m[tmax]]
```

```
Out[47]=  $\frac{fthrust \text{Log}\left[\frac{m_0}{m_0 - R t_{max}}\right]}{R}$ 
```

```
In[48]:= deltax1 = deltax[ $\frac{fthrust}{R}$ , m[0], m[tmax]] /. params
```

```
Out[48]= 50 Log[2]
```

```
In[49]:= N[deltax1]
```

```
Out[49]= 34.6574
```

WARNING !!!! Newton's 2nd law, $dp/dt=f_{thrust}$, with $p=m(t)v(t)$ CANNOT BE DIRECTLY APPLIED !!!!

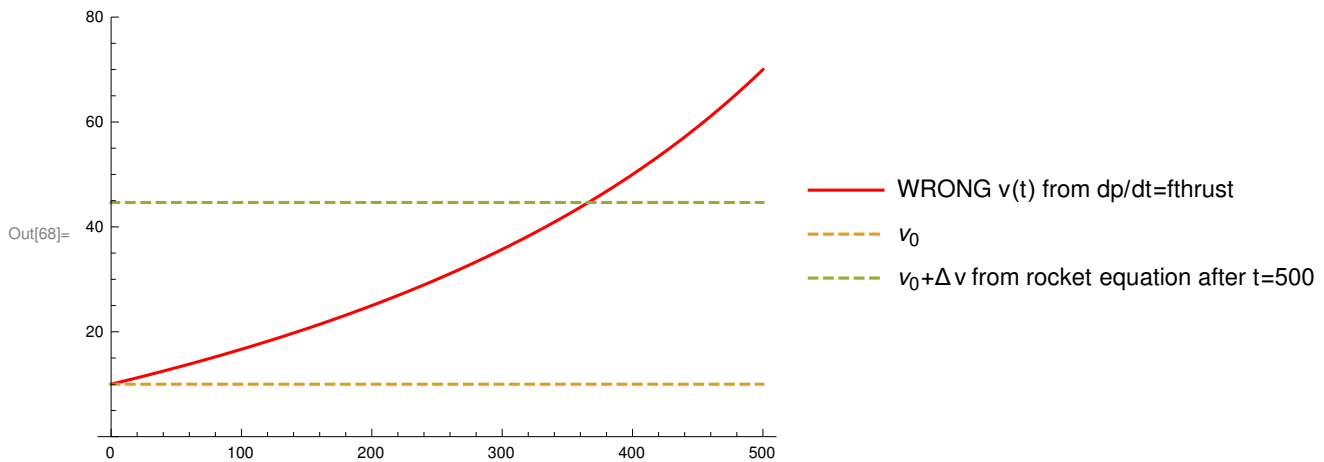
```
In[50]:= (* to see that, let's try to solve dp/dt=fthrust : *)
sol1 = DSolve[{D[ m[t] * v[t], t] == fthrust, v[0] == v0}, v[t], t] // FullSimplify
```

```
Out[50]= {{v[t] ->  $\frac{f_{thrust} t + m_0 v_0}{m_0 - R t}$ }}
```

```
In[51]:= (* for our paramtrs v(t) is: *)
v[t] /. sol1 /. params
```

```
Out[51]=  $\left\{ \frac{100\,000 + 500 t}{10\,000 - 10 t} \right\}$ 
```

```
In[68]:= (* v(t) IS NOT CORRECT!!!! since v(tmax) is not equal Δv known from the rocket equation !!! *)
Plot[{v[t] /. sol1 /. params, (v0 /. params), (v0 /. params)+deltav1},
{t, 0, tmax /. params}, PlotLegends ->
{"WRONG v(t) from dp/dt=fthrust", "v0", "v0+Δv from rocket equation after t=500"},
PlotStyle -> {Red, Dashed, Dashed}, PlotRange -> {0, 80}]
```



(* let's compare with Newton's 2nd law, $dp/dt=F_{net}$, for $F_{net}=f_0=const$ and $m=const$
correct: $v(t) = a t + v_0$, where $a=f_0/m=const$, but not useful for a rocket *)

```
In[53]:= DSolve[{D[ m v[t], t] == f0, v[0] == v0}, v[t], t] // FullSimplify
```

```
Out[53]= {{v[t] ->  $\frac{f_0 t}{m} + v_0$ }}
```

How to find correct $v(t)$?

(*

$$a_x(t) = dv_x/dt \quad a_x(t)=F_{net,x} / m(t) \quad \text{where} \quad F_{net,x}=F_{thrust}=(-v_e)_x R=v_e R \quad \text{note: minus because } v_e \text{ vector opposite to } F_{thrust}$$

and we need to integrate in $[0;t]$ both sides of

$$\frac{dv(t)}{dt} = \frac{R v_e}{m(t)}$$

where $m(t) = m_0 - R t$, and $v_e=F_{thrust}/R$

*)

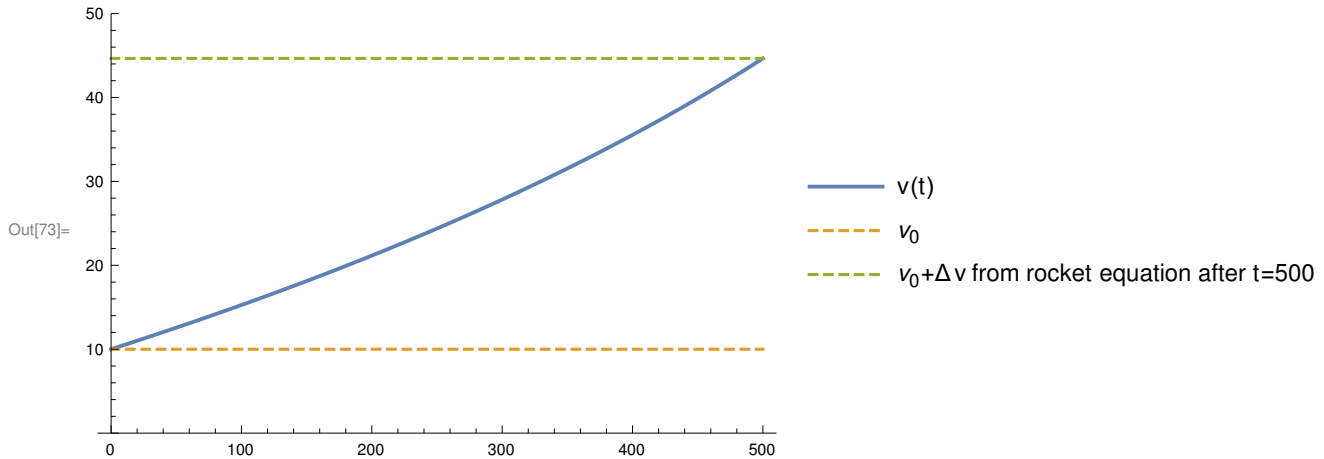
$$(* \text{ hint: } *) \int \frac{1}{m_0 - R t} dt$$

$$\text{Out[59]= } -\frac{\text{Log}[m_0 - R t]}{R}$$

In[70]:= (* result: *)

$$v[t_] := v_0 + \frac{f_{\text{thrust}}}{R} \text{Log}\left[\frac{m_0}{m[t]}\right]$$

In[73]:= Plot[{v[t] /. params, (v_0 /. params), (v_0 /. params)+deltav1}, {t, 0, tmax /. params},
PlotLegends → {"v(t)", "v_0", "v_0+Δv from rocket equation after t=500"},
PlotStyle → {Thick, Dashed, Dashed}, PlotRange → {0, 50}]



How to find x(t)?

$$(* v = \frac{dx}{dt} \rightarrow x(t_1) = x_0 + \int_0^{t_1} v(t) dt *)$$

$$\text{In[74]:= } (* \text{ hint: } *) \int \text{Log}\left[\frac{m_0}{m_0 - R t}\right] dt$$

$$\text{Out[74]= } t - \frac{(m_0 - R t) \text{Log}\left[\frac{m_0}{m_0 - R t}\right]}{R}$$

$$\text{In[75]:= } \int_0^{t_1} \text{Log}\left[\frac{m_0}{m_0 - R t}\right] dt$$

$$\text{Out[75]= } \text{ConditionalExpression}\left[t_1 + \left(-\frac{m_0}{R} + t_1\right) \text{Log}\left[\frac{m_0}{m_0 - R t_1}\right], \text{Re}\left[\frac{m_0}{R t_1}\right] > 1 \parallel \text{Re}\left[\frac{m_0}{R t_1}\right] < 0 \parallel \frac{m_0}{R t_1} \notin \mathbb{R}\right]$$

$$\text{In[77]:= } \text{Integrate}\left[\text{Log}\left[\frac{m_0}{m_0 - R t}\right], \{t, 0, t_1\}, \text{Assumptions} \rightarrow \{R > 0, t_1 > 0, m_0 > R t_1\}\right]$$

$$\text{Out[77]= } t_1 + \left(-\frac{m_0}{R} + t_1\right) \text{Log}\left[\frac{m_0}{m_0 - R t_1}\right]$$

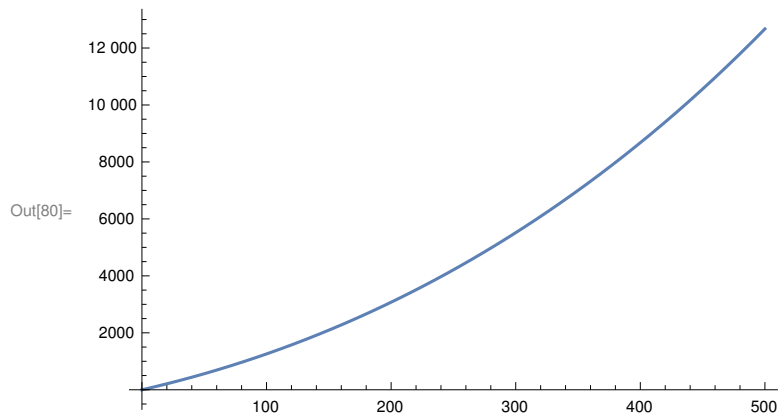
$$\text{In[78]:= } \text{Integrate}[v[t], \{t, 0, t_1\}, \text{Assumptions} \rightarrow \{R > 0, t_1 > 0, m_0 > R t_1\}]$$

$$\text{Out[78]= } t_1 v_0 + \frac{f_{\text{thrust}} \left(t_1 + \left(-\frac{m_0}{R} + t_1\right) \text{Log}\left[\frac{m_0}{m_0 - R t_1}\right]\right)}{R}$$

In[79]:= (* assuming $x_0=0$: *)

$$x[t_]:=t\,v_0+\frac{f_{thrust}\left(t+\left(-\frac{m_0}{R}+t\right)\text{Log}\left[\frac{m_0}{m_0-R\,t}\right]\right)}{R}$$

In[80]:= Plot[{x[t] /. params}, {t, 0, tmax /. params}]



(* let's verify if $x''(t)$ equals $a(t)$ *)

In[82]:= x''[t] // FullSimplify

Out[82]=

$$\frac{f_{thrust}}{m_0 - R\,t}$$

(* OK, $a = F/m(t)$ *)