

MEC-E8006 Fatigue of Structures

Lecture 1: Fatigue phenomena

#### **Learning outcomes**

#### After the lecture, you

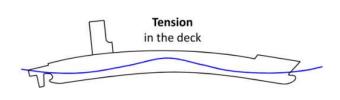
- <u>understand</u> fatigue phenomena
- <u>understand</u> the fatigue process and microstructure effect
- <u>know</u> fractography and analysis of fracture surfaces

#### **Contents of Lecture**

- Importance of fatigue design
- Brief history of fatigue research
- Fatigue definition
- Fatigue damage process
- Fractography and fracture surface analysis

- Many structures and vehicles are affected by time varying loading
  - For instance, a ship has about 10<sup>8</sup>
     load cycles during its lifetime

A 300 m long 4500TEU Container vessel

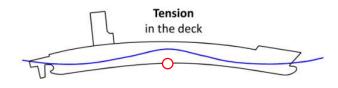


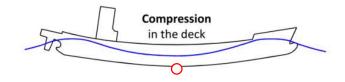


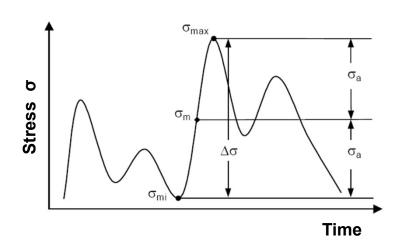




- Many structures and vehicles are affected by time varying loading
  - For instance, a ship has about 10<sup>8</sup>
     load cycles during its lifetime







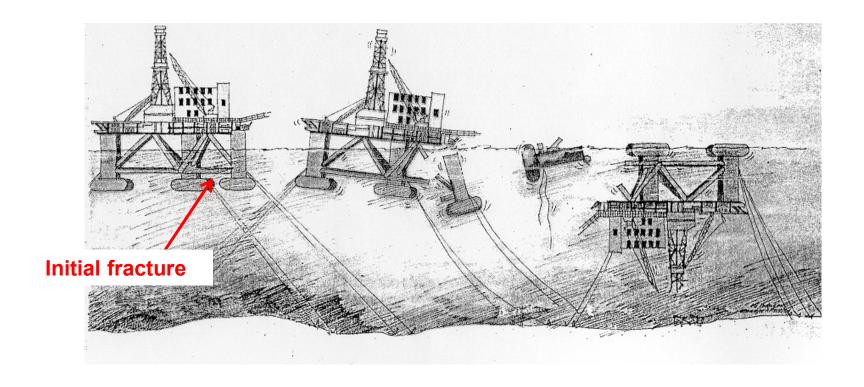
#### Stress range

$$\Delta \sigma = \sigma_{\text{max}} - \sigma_{\text{min}}$$

#### Mean stress

$$\sigma_{\rm m} = (\sigma_{\rm max} + \sigma_{\rm min}) / 2$$

# Failure of the Alexander L. Kielland platform, Norway, 1980

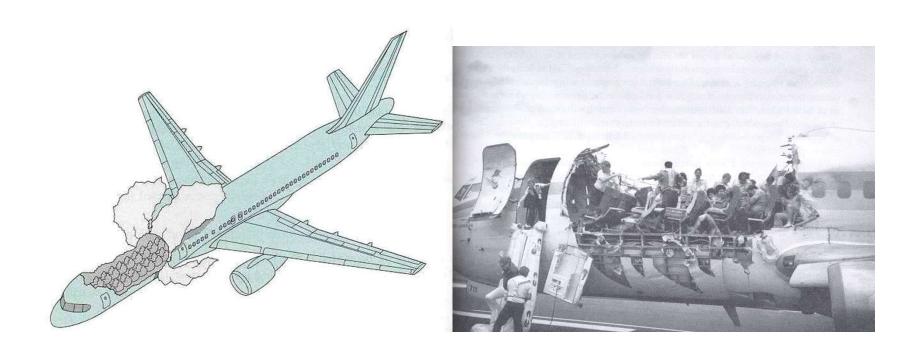


Failure of the Alexander L. Kielland platform,

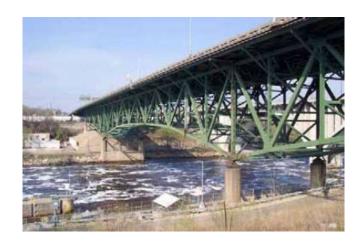
*Norway, 1980* 



#### Aloha Airlines Flight 243, Hawaii 1988



**Bridge 9430, USA, 2007** 



**View 2001** 



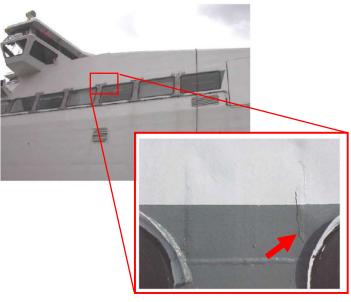
I-35W Mississippi River bridge August 1, 2007



# The most fractures are due to fatigue

- Structural optimization aiming for a light and cost-efficient structure
- Increasing demand for the utilization of new materials such as high strength steel
- Fatigue design has special challenges
  - Large and complex structure
  - For instance, ship hull includes thousands of cut steel parts and hundreds kilometers of weld seams

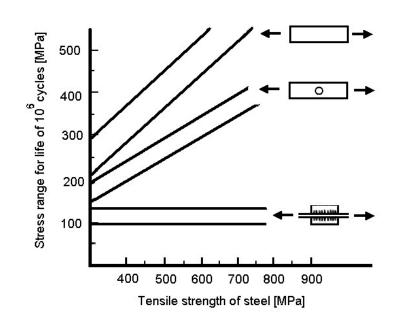




#### Many affecting factors

- Load history
  - Stress range, maximum stress, load frequency
- Geometry effect
  - Production technology
- Material
  - Steel, aluminum, etc.
- Environment
  - Corrosion, temperature, etc.





#### Many affecting factors

#### "Internal" factors

(at the end of the manufacturing process)

- Material
- Dimension
- Surface finishing
- Shape
- Surface treatments

#### "External" factors

(due to operating conditions and environments)

- Average stress
- Loading type (e.g. tension VS bending)
- Temperature and environment
- Load history
- Load variations

### Brief history of fatigue research

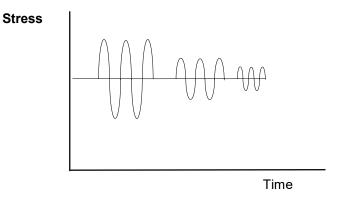
- 1828: Wilhelm Albert tested mine host under cyclic loading
- 1850s: August Wöhler systematic fatigue testing, begin of development of fatigue design
- 1900s: James Alfred Ewing revealed the origin of fatigue failure in microscopic cracks
- 1910: O. H. Basquin introduced a log-log relationship for S-N curves, based on Wöhler's tests
- 1945: A. M. Miner formulated Palmgren's (1924) linear damage hypothesis for fatigue design
- 1960s: L. F. Coffin and S. S. Manson found the relationship between plastic strain amplitude and fatigue life
- 1960s: P. C. Paris showed that fatigue crack growth can be described using fracture mechanics (stress intensity factor)
- 1968: T. Endo and M. Matsuishi created the rainflow-counting algorithm to characterize random loadings
- 1970: W. Elber revealed the importance of crack closure on fatigue crack growth and introduced effective stress intensity concept

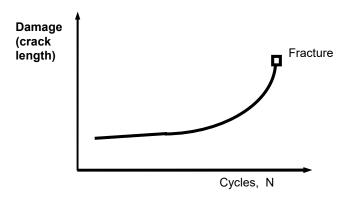


#### **Fatigue definition**

Fatigue - the process of progressive localized permanent structural change occurring in a material subjected to conditions that produce <u>fluctuating stresses</u> and strains at some point or points and that may culminate in <u>cracks</u> or <u>complete fracture</u> after a sufficient number of fluctuations.

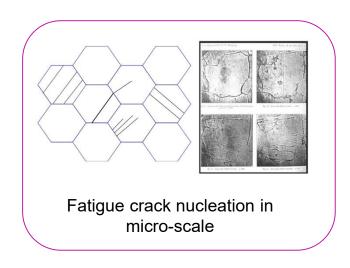
(Am. Soc. for Testing and Materials (ASTM) definition)

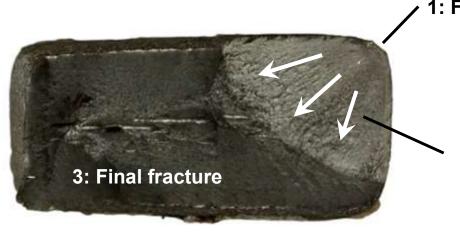




### **Fatigue process**

 Fatigue crack initiates from the high stress location and propagates under cyclic loading until final fracture

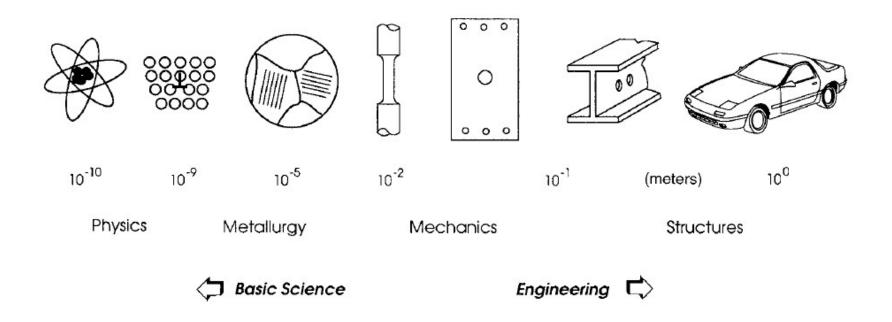




1: Fatigue crack initiation

2: Fatigue crack propagation

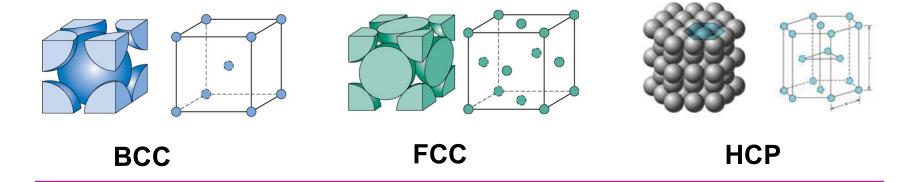
# Size scales and disciplines in engineering material



## **Fatigue of materials**

#### Crystal structures of metals

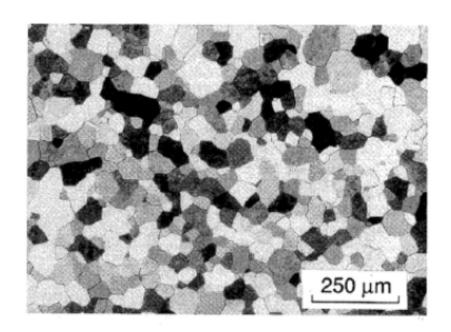
| Material | Lattice | Emax[111] (GPa) | Emin[100] (GPa) |
|----------|---------|-----------------|-----------------|
| a-Fe     | ВСС     | 284.5           | 132.4           |
| Al       | FCC     | 75.5            | 62.8            |
| Cu       |         | 190.3           | 66.7            |
| Mg       | НСР     |                 |                 |
| a-Ti     |         |                 |                 |

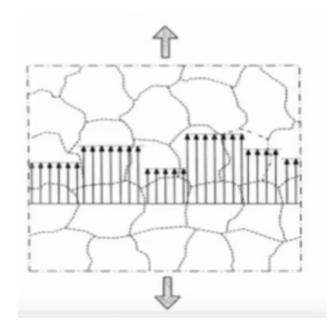




### **Crystal grain structure**

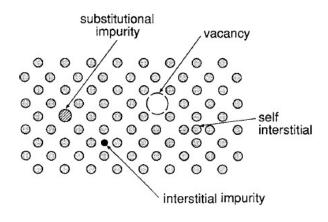
Grain size and shape Variation of the crystal orientation from grain to grain



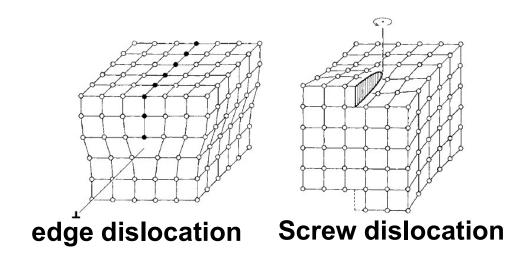


### **Defects in crystals**

#### Point defect



#### Line defect or dislocation

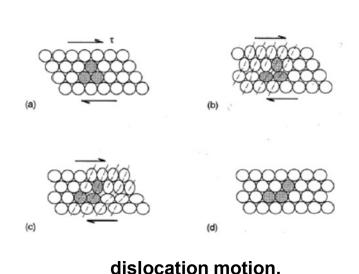


https://www.youtube.com/watch?v=-t6btGjGKYU

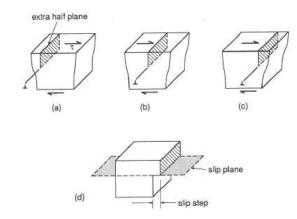
edge dislocation

https://www.youtube.com/watch?v=TxJOP3hA6To screw dislocation

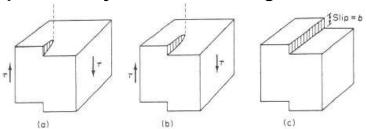
# Plastic deformation by dislocation motion



The plane in which the dislocation line moves is called slip plane.

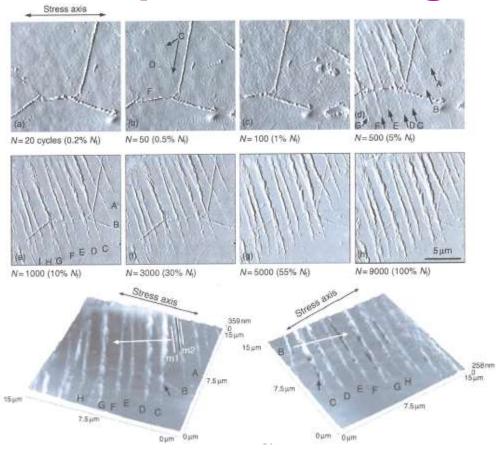


Slip caused by the motion of edge dislocation.



Slip caused by the motion of screw dislocation.

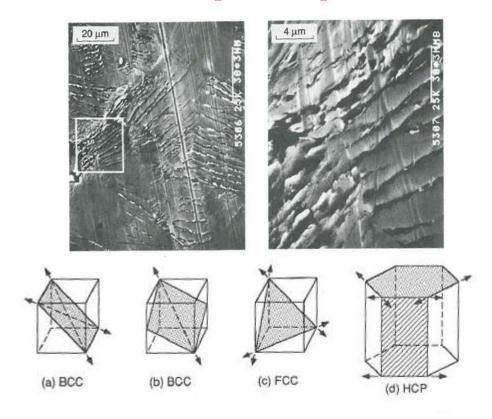
### Process of slip band damage



Slip occurs in metals within individual grain by dislocation moving along crystallographic planes. Slip is localization of plastic strain.

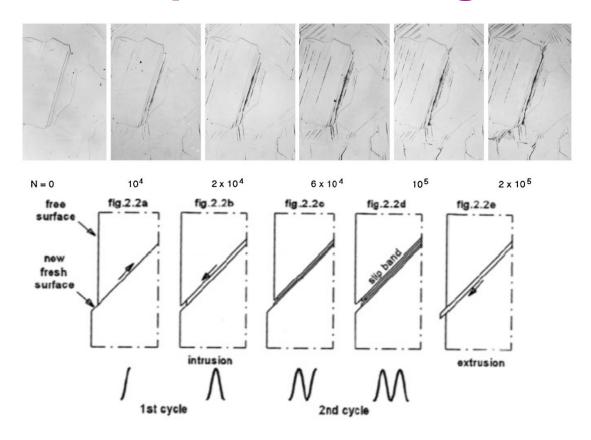


### Slip band and slip step

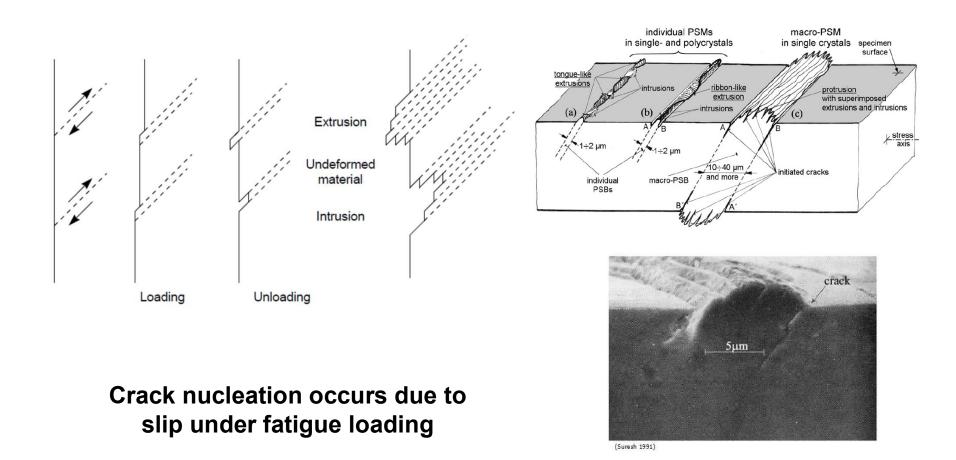


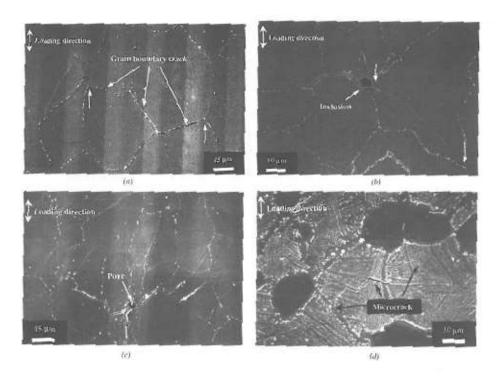
For a given crystal structure, slip is easier on certain planes and within these planes in certain direction. These prefered planes are called as closed-pack planes.

### Process of slip band damage



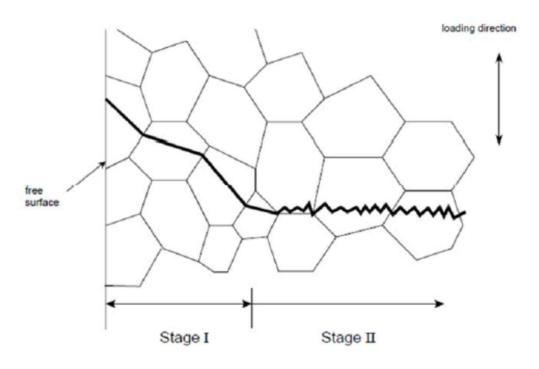
Slip bands are regions where there is intense deformation due to shear motion between crystal planes. The plastic deformation is often concentrated in slip bands.





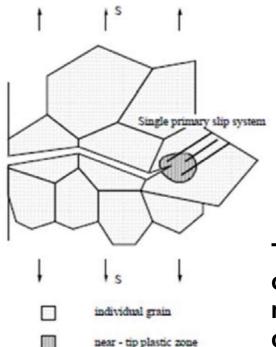
Fatigue micro crack nucleates due to two main mechanisms:

- 1. Along slip bands
- 2. At or near material discontinuities like second phase, voids, inclusions, grain boundaries or pores



- Stage I: Fatigue cracks that nucleate in local shear bands initially tend to grow in a plane of maximum shear stress range.
- Stage II: Crack grows in a zig zag manner controlled primary by maximum tensile stress-range.

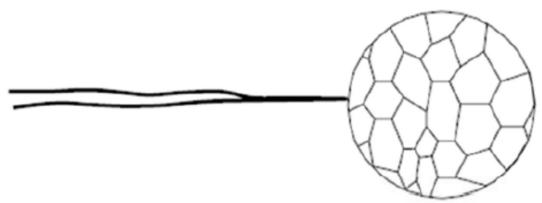
#### Stage I





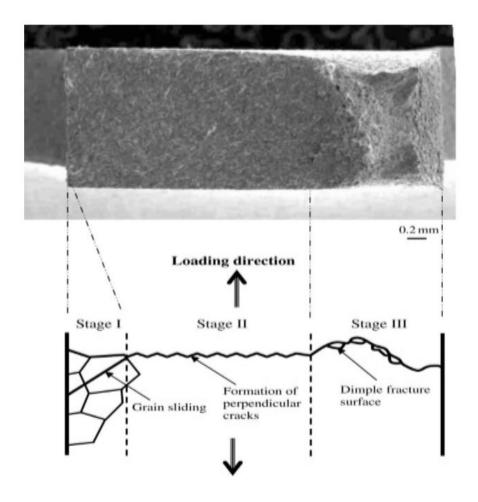
This growth is quite small usually on the order of several grains and thus, growth of microcracks strongly influenced by the slip characteristics of material, material grain size and the extent of plasticity near the crack tip.

#### Stage II



- Plastic zone size is much larger than the material microstructure so that the microstructure does not play such important role.
- The high plastic stress and strain ahead of crack tip due to notch root plasticity or crack tip plasticity directly enhanced driving force.

#### Different stages of fatigue crack progress

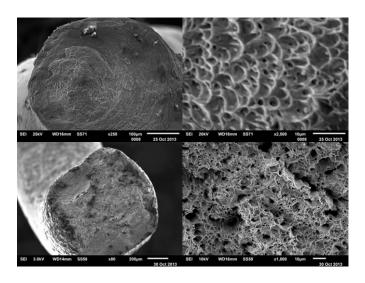


An electro microscopic analysis of fractured surfaces reveal a wide range of fatigue crack growth mechanism.

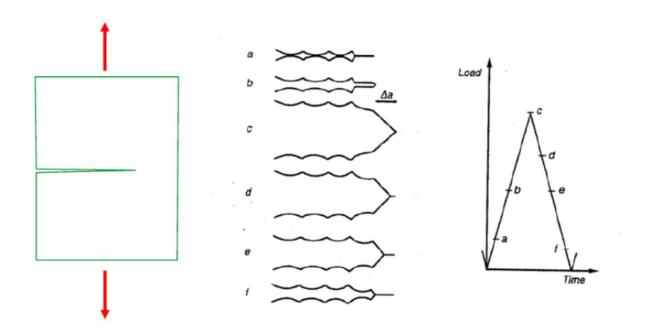


Three more common models of fatigue crack growth mechanism:

- 1. Striation formation
- 2. Microvoid coalescence
- 3. Microcleavage

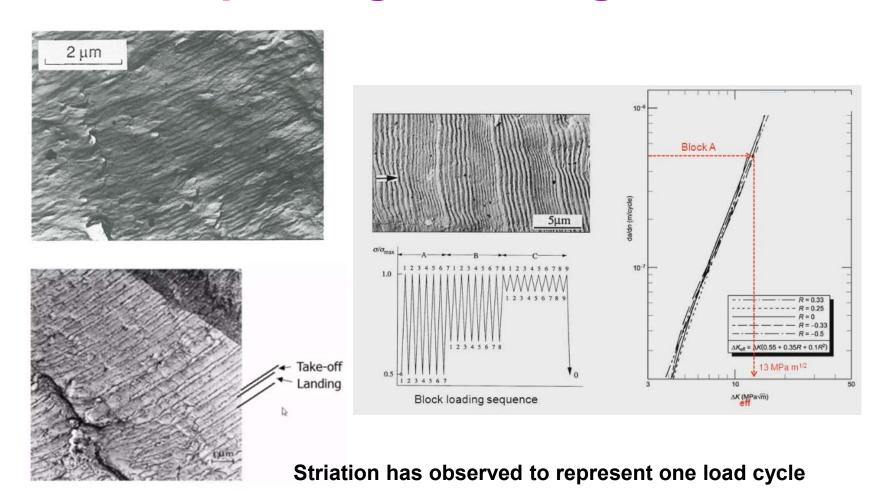


#### **Striation**



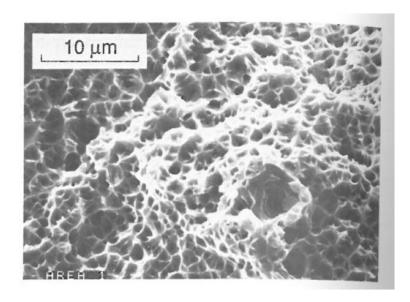
Striation is the progress of the crack on each cycle.

Striation forming by a plastic crack tip blunting mechanism during the loading and unloading portion of the fatigue cycle.



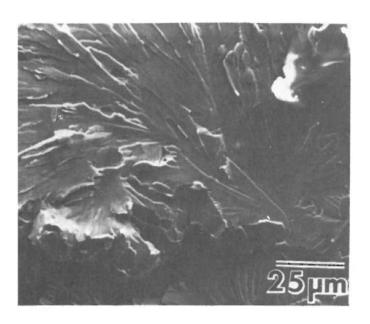


#### Microvoid coalescence

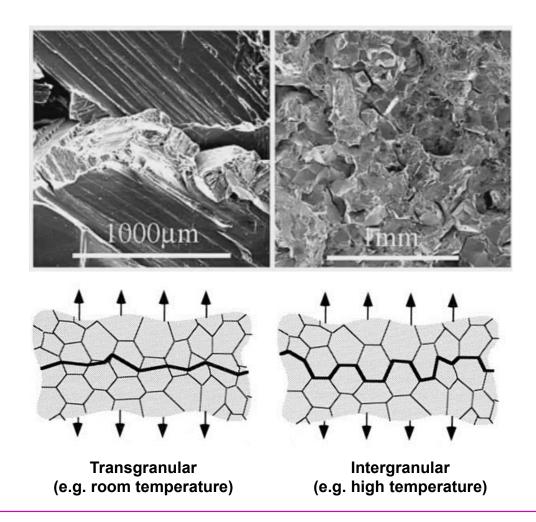


- This a high-energy microscopic process which is observed in ductile material that there is little restriction on dislocation movement and slip is abundant.
- Microvoids grow during plastic flow of the material.
- Shear result in elongated dimples.

#### Micro-cleavage



- This is low-energy crack growth process.
- It involves fractured along specific crystallographic planes where atomic bond are weak.
- We may have more than 1 cleavage plane, with different activation energy; e.g. single crystal silicon;
- Brittle of fast fracture happen without plastic deformation.

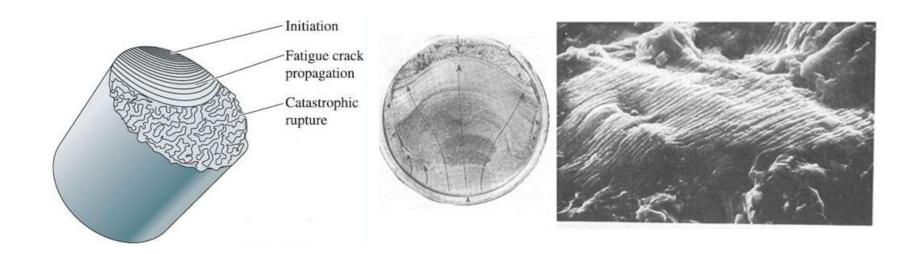




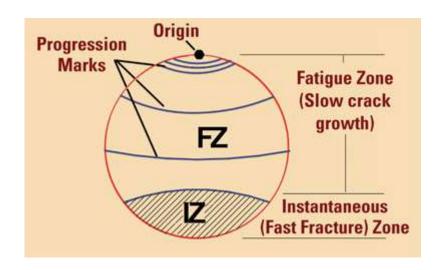
#### Fatigue surface, general characteristics

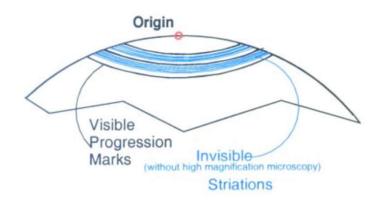
Fatigue failure surfaces have three characteristics features:

- A surface or near-surface defect as the origin of crack
- Striation corresponding to slow, intermittent crack growth
- Dull, fibrous brittle fracture surface (rapid growth)



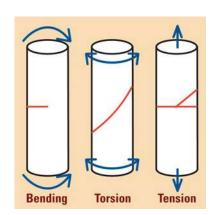
### Fatigue surface general characteristics





Beach mark or progression mark shows the progress of crack at various stage such as start up or shut down forces. Progression mark depicts the variation of applied stress ranges.

#### **Example**



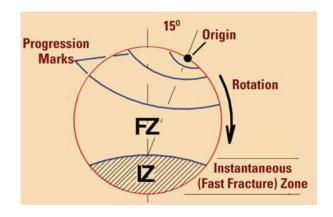
#### Plane Bending, Reversed Bending, and Rotational Bending Fatigue Failures Low Stress with Low **High Stress with Low** Low Stress with High Stress Concentration Stress Concentration Stress Concentration Multiple Origins The ratchet marks on either (Primary, Secondary Origin side of the primary origin Progression and Tertiary) Progression Marks Marks grow in slightly different Fatigue Fatigue directions, indicating the racture began at this origin Plane Bending Ratchet Marks Note how the corners of the progression marks turn down-Overload ward because the high stress Overload (Fast Fracture) Zone (Fast Fracture) Zone concentration accelerates the crack growth near the surface Origin #1 The fact that the fatigue zone Reversed (two-way) on one side is larger than the Ratchet Marks #1 Fatigue Zone #1 Fatigue Zone fatigue zone on the other does #1 Fatigue Zone Bending not necessarily indicate that the stress is also higher. A #2 Fatigue Zone better measure is a comparison of the progression marks (if they #2 Fatigue Zor #2 Fatigue Zon Multiple Origins Origin IZ Bisector No readily obvious origin but Ratchet Marks multiple origins around the **Rotational Bending** exterior The high total stress at the exterior of the shaft causes failures to begin On low stress rotating bending at many locations. This high total failures the direction of rotation stress may just be the result of the can be seen by the distortion of load on the part or it may be caused **Multiple Origins** by a high stress concentration acting Overload on a moderate or low load. Zone **Multiple Origins** Plane Bending of Again, the ratchet marks to Plate or Bar Stock the side of the primary origin grow in slightly different Overload Zone Overload Zone

Note that in most cases tension failures will appear very similar to either plane bending or reversed bending failures. One exception is that they will more frequently start on the interior of the part because the entire cross section of the piece is being stressed.



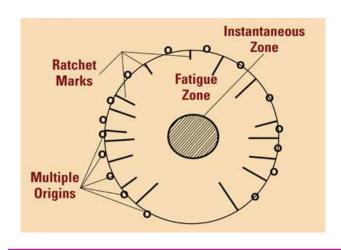
N. W. Sachs, Understanding the Surface Features of Fatigue Fractures: How They Describe the Failure Cause and the Failure History, JFAPBC (2005) 2:11-15.

### **Example**



Single origin



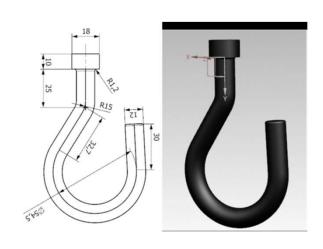


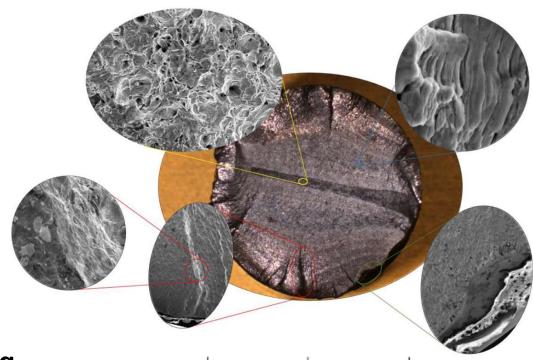
Multiple origins



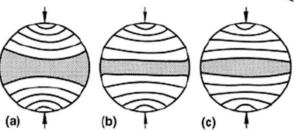


## **Example**





#### **Reverse bending loading**





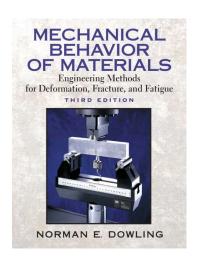
### Readings – Course material

#### Course book

Mechanical Behavior of Materials Engineering Methods for Deformation, Fracture, and Fatigue, Norman E. Dowling

Section: 2.3-2.4

Section: 9.5

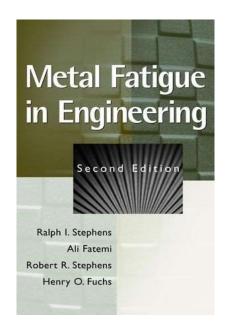


#### Optional reading materials given in MyCourses webpages

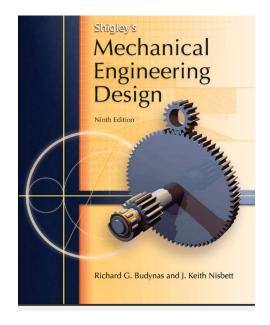
- Sachs, N.W. 2005. Fracture features: Understanding the Surface Features of Fatigue Fractures: How They Describe the Failure Cause and the Failure History, Journal of Failure Analysis and Prevention, 2:11-15.
- W. Schlitz, W. 1996. A history of fatigue, Engineering Fracture Mechanics, 54:263-300.



### Readings – Other references



**Chapter 1 and 3** 



**Chapter 6: 6-1 and 6-2** 



#### **Assignment 1-1: Fatigue Mechanics**

Select an article form Engineering Failure Analysis journal and write a summary of the article. The summary report should be short (max. A4) and it should include the description of the following issues:

- Analyzed structure and failure location
- Fatigue mechanics (crack initiation, propagation, final failure)
- Possible reasons for the failure and main affecting factors

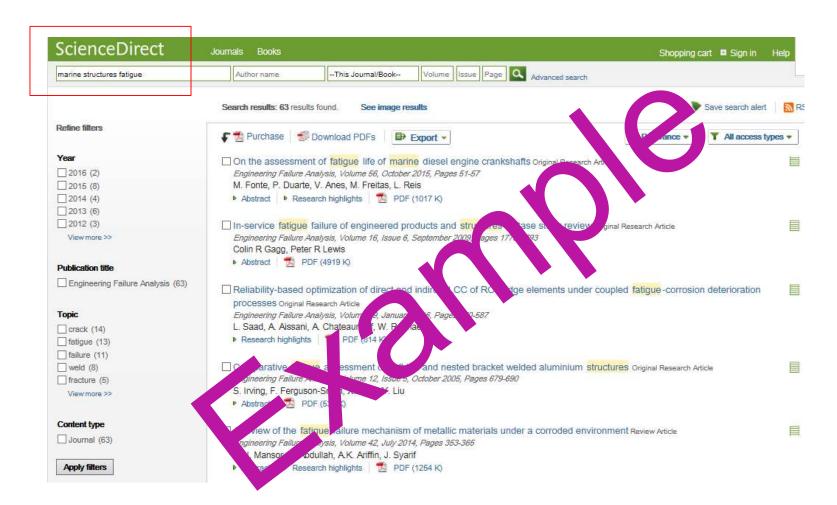
www: <a href="http://www.journals.elsevier.com/engineering-failure-analysis/">http://www.journals.elsevier.com/engineering-failure-analysis/</a>

#### Report delivery

In PDF-format using to MyCourses page (<a href="https://mycourses.aalto.fi">https://mycourses.aalto.fi</a>)



#### **Assignment 1-1: Fatigue Mechanics**





#### **Assignment 1-1: Fatigue Mechanics**

Engineering Failure Analysis 56 (2015) 51-57



Contents lists available at ScienceDirect

#### **Engineering Failure Analysis**

journal homepage: www.elsevier.com/locate/engfailanal



#### On the assessment of fatigue life of marine diesel engine crankshafts



<sup>a</sup> Escola Superior Náutica (ENIDH), Portugal

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Keywords: Marine diesel engine Crankshafts failure Fatigue strength assessment Fatigue life

#### ABSTRACT

The fatigue strength and its co ct asses ortant role in design and maintenance of marine crank perational safety and reliability. g on cra Crankshafts are under a nating ben and rotating bending combined e responsible for fatigue failure. The mostly with torsion on main rnals, whi ercial man epends on the main engine in service of its design crankshaft design strictly follows the rules in particula of classification societies at study provides an overview on the assessment of fatigue life of marine e afts and its maintenance taking into account the ving in the t decades, considering that accurate estimation of fatigue life inportant to en safety of components and its reliability. An example of a ouilt crankshaft failere is also presented and the probable root case of damage, and the end some fi remarks are presented.

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(a)



b IDMEC, Instituto Superior Técnico, Universidade de Lisboa, Portugal