

# Fracture mechanics

## Seminar 1: The stress intensity factor



Aalto-yliopisto  
Aalto-universitetet  
Aalto University

Luc St-Pierre

April 26, 2023

# Schedule

## No traditional lectures:

- Go through the material at your own pace. Recordings will be available via MyCourses. No lectures on Tuesdays 14.15-16.00.

**Seminar:** Wednesdays, 14.15-16.00, Otakaari 4, room 216.

- I will summarise the theory and introduce a few examples.

**Calculation hours:** Thursdays, 14.15-16.00, Otakaari 4, room 216.

- I will be available to help you with the weekly assignment.

# Evaluation

- **5 Assignments (40%)**
  - Your mark will be based on your 4 best assignments.
  - 4 sets of problems and 1 computer exercise.
    - *Upload your assignment via MyCourses.*
- **Exam (60%)**
  - Thursday June 8, 9.00-12.00.
  - In-person, room 215, Otakaari 4.
  - You need to pass the exam to pass the course.

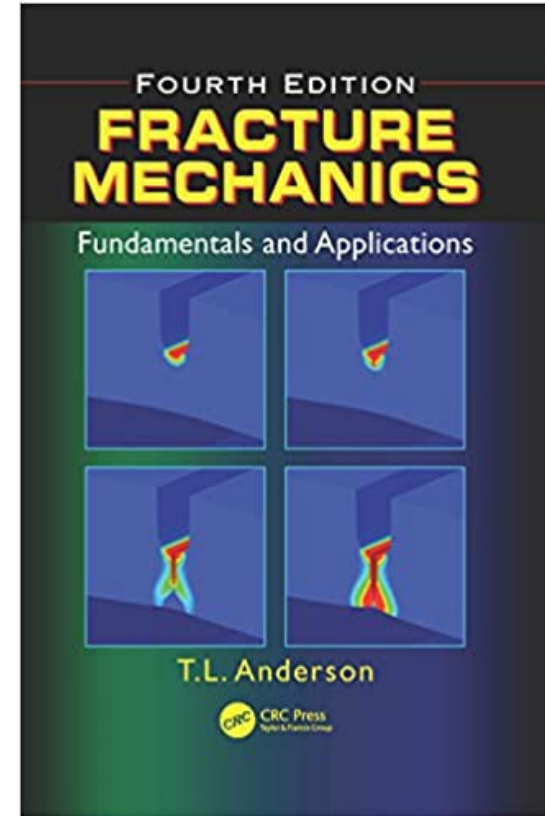
Grade	Final mark %
5	≥90
4	80-89
3	70-79
2	60-69
1	50-59
0 – Fail	≤49

# Material

Lecture notes will be available on MyCourses.

Consult the textbook if you need additional information:

- T.L. Anderson, Fracture Mechanics: fundamentals and applications, 4<sup>th</sup> edition, 2017.



# E-books available

- M. Janssen; J. Zuidema; R.J.H. Wanhill; *Fracture mechanics*, Spon press, 2004.
- A.T. Zehnder; *Fracture mechanics*, Springer, 2012.
- N. Perez; *Fracture mechanics*, Springer, 2017.
- E.E. Gdoutos; *Fracture mechanics: an introduction*, Springer, 2020.

# Learning outcomes

After this week, you should be able to:

- Understand what is the stress intensity factor, and how it is derived.
- Use the stress intensity factor, and the principle of superposition, to solve design problems.

# Why is there a crack?

Components are not designed with cracks, but cracks form and grow under cyclic loading (fatigue). Other factors can lead to cracks:

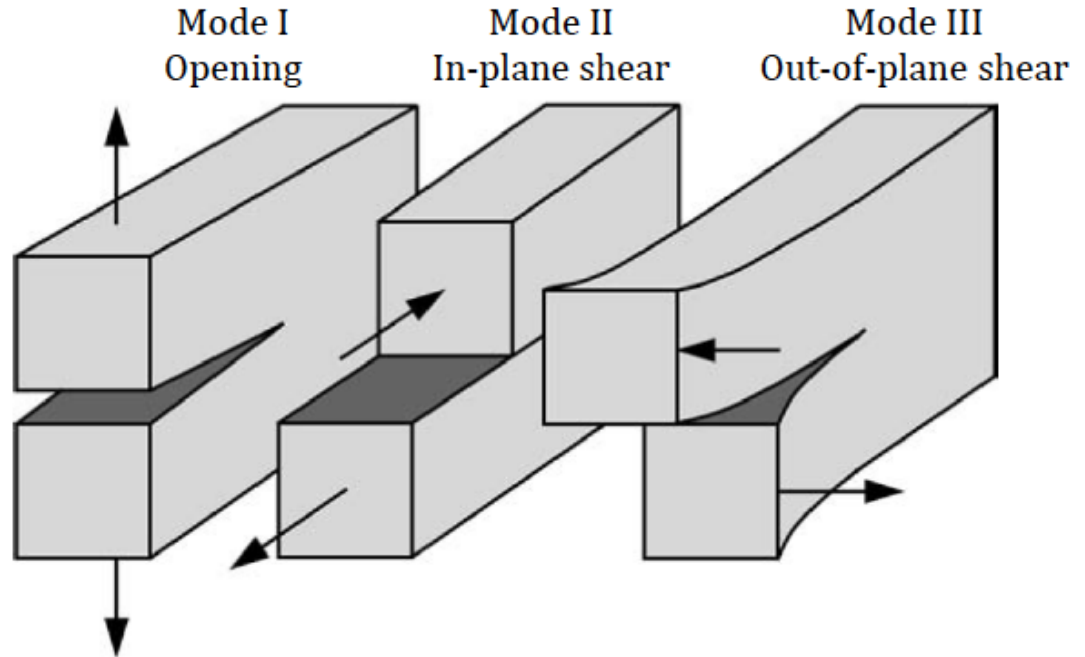
- Thermal stresses;
- Harsh chemical environment;
- Manufacturing process.



# Three modes of loading

A crack can be loaded:

- in a single mode or
- a combination (modes I and II for example).

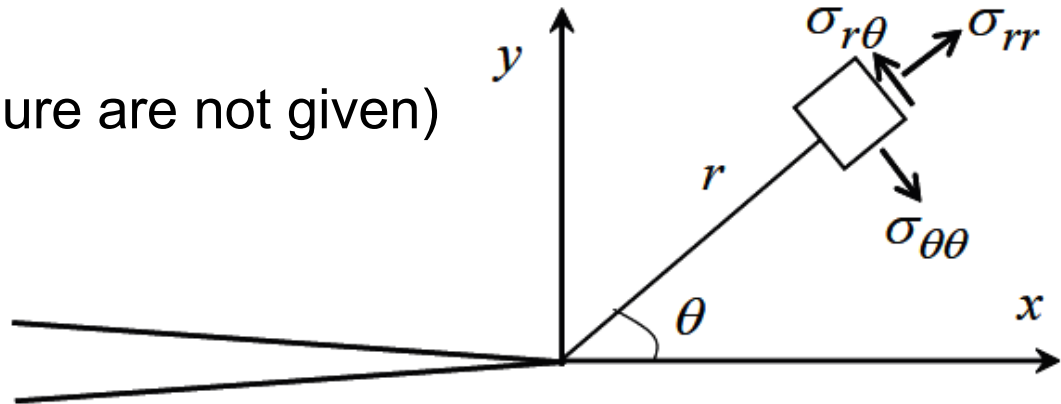




# Stress field close to a crack tip

The analytical solution of Williams (1957) assumes that:

- Linear elastic, isotropic material;
- Sharp crack loaded in mode I;
- Infinitely large plate
- (the external forces/pressure are not given)

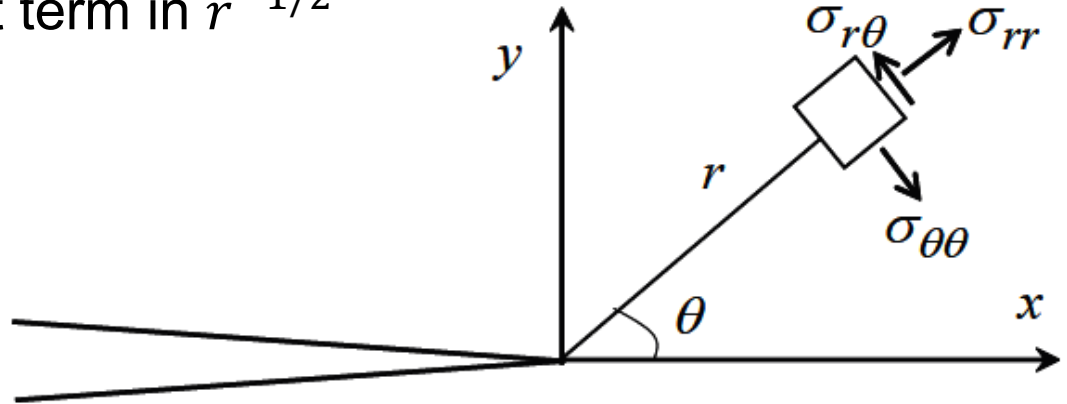


# Stress field close to a crack tip

Main steps to derive the stress field:

1. Apply boundary conditions;
2. Finite strain energy;
3. Keep the most important term in  $r^{-1/2}$

$$\left\{ \begin{array}{l} \sigma_{\theta\theta}(\theta = \pi) = \sigma_{r\theta}(\theta = \pi) = 0 \\ \sigma_{\theta\theta}(\theta) = \sigma_{\theta\theta}(-\theta) \end{array} \right.$$



# Stress field close to a crack tip

The stress field at the crack tip in mode I is given by these equations.

It depends on a **single constant**: the stress intensity factor  $K_I$ .

$$\sigma_{\theta\theta} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{3}{4} \cos \frac{\theta}{2} + \frac{1}{4} \cos \frac{3\theta}{2} \right)$$

$$\sigma_{r\theta} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{1}{4} \sin \frac{\theta}{2} + \frac{1}{4} \sin \frac{3\theta}{2} \right)$$

$$\sigma_{rr} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{5}{4} \cos \frac{\theta}{2} - \frac{1}{4} \cos \frac{3\theta}{2} \right)$$

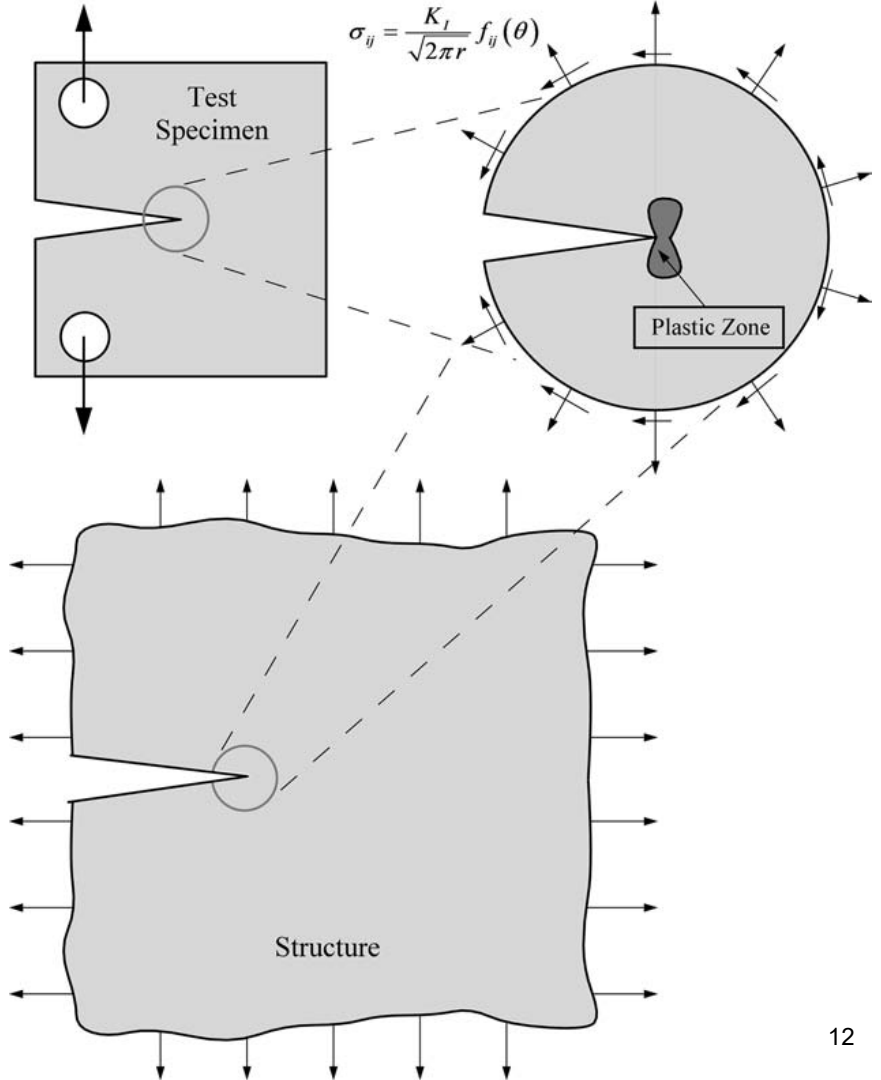
# Local stress field

The stress field close to the crack tip is the same:

- In a large structure or
- In a small test specimen.

This is crucial for  $K_I$  to predict fracture.

How can we get  $K_I$ ?



# Where can I get $K_I$ ?

To evaluate the stress intensity factor  $K_I$ , you need to know:

- The geometry of the component, and
- The external loads.

Solutions are available in the datasheet and textbooks. It is also possible to evaluate  $K_I$  analytically or numerically.

# What can I do with $K_I$ ?

## Predict Fracture!

A crack will propagate (fracture) when  $K_I$  reaches a critical value, which is the material's fracture toughness  $K_{Ic}$ .

Material	$K_{Ic}$ (MPa $\sqrt{m}$ )
Low carbon steel alloys	40-80
Aluminum alloys	22-35
Titanium alloys	14-120
Wood (best orientation)	5-9
PMMA	0.7-1.6
Glass	0.6-0.8
Concrete	0.35-0.45

# $K_{Ic}$ is a material property

## Fracture

- The fracture toughness  $K_{Ic}$  is a material property.
- The stress intensity factor  $K_I$  represents the loading intensity.
- Fracture occurs when:  $K_I = K_{Ic}$ .

## Yielding

- The yield strength  $\sigma_y$  is a material property.
- The von Mises stress  $\sigma_{vm}$  represents the loading intensity.
- Yielding occurs when:  $\sigma_{vm} = \sigma_y$ .

# Principle of superposition: rules

**YES:** you can add stress intensity factors for the same mode of loading:

$$K_I^{[\text{total}]} = K_I^{[\text{A}]} + K_I^{[\text{B}]} + \dots$$

**NO:** you can't add stress intensity factors for different modes:

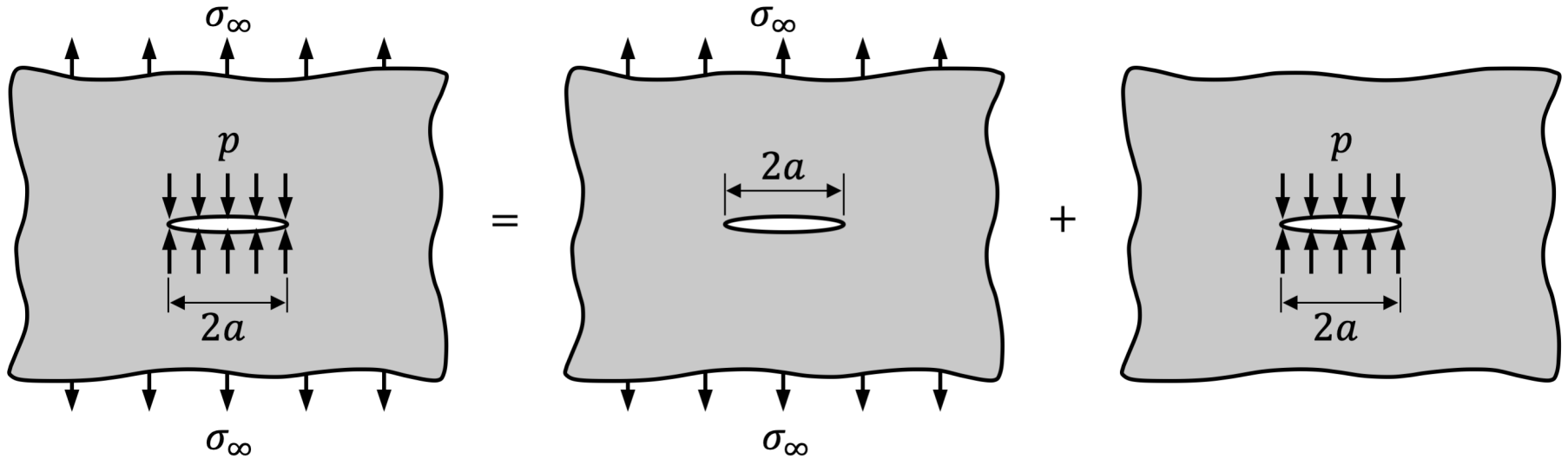
$$K^{[\text{total}]} \neq K_I^{[\text{A}]} + K_{II}^{[\text{B}]} + K_{III}^{[\text{C}]} + \dots$$

**YES:** you can add stress components for different modes:

$$\sigma_{ij}^{[\text{total}]} = \sigma_{ij}^{[\text{modeI}]} + \sigma_{ij}^{[\text{modeII}]} + \sigma_{ij}^{[\text{modeIII}]}$$



# Principle of superposition



$$K_I^{[total]} = K_I^{[A]} + K_I^{[B]}$$

This one will be negative as  $p$  is closing the crack.

# In summary

The stress intensity factor:

- Quantifies the stress field at the crack tip,
- Follows the principle of superposition for a given mode,
- Can be used to predict fracture.

Next week, we will tackle fracture with an approach based on **energy** instead of **stress**.

# Fracture mechanics

## Seminar 2



Aalto-yliopisto  
Aalto-universitetet  
Aalto University

Luc St-Pierre

May 3, 2023

# Learning outcomes

After this week, you should be able to:

- Use the energy release rate to predict fracture.
- Evaluate if crack growth will be stable or unstable.
- Calculate the amount of stable crack growth using an R-curve.

# Energy release rate

Griffith (1920) studied fracture using an energy approach:

Change in work done  
by external forces



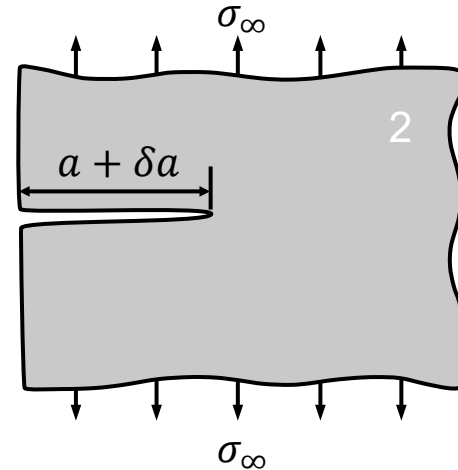
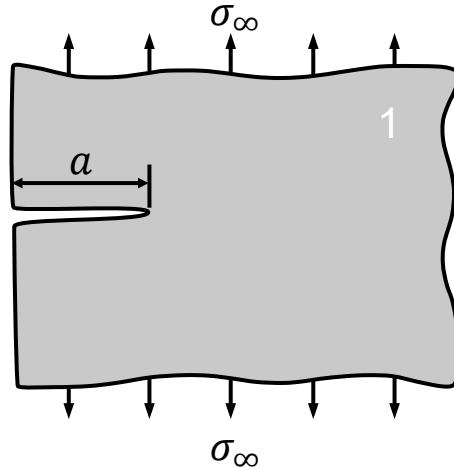
$$\delta W = \delta U + G\delta A$$



Change in strain energy



Change in crack area (m<sup>2</sup>)  
Energy release rate (J/m<sup>2</sup>)



# Energy release rate

Griffith (1920) studied fracture using an energy approach:

Change in work done  
by external forces



$$\delta W = \delta U + G\delta A$$



Change in strain energy



Energy release rate (J/m<sup>2</sup>)

Change in crack area (m<sup>2</sup>)

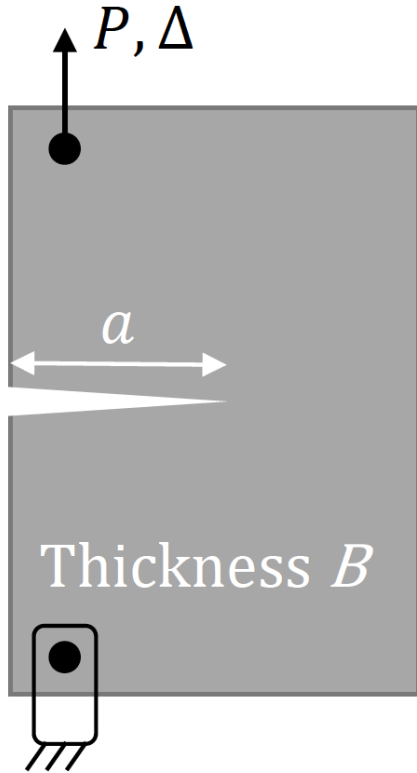
Rearranging gives:

$$G = \frac{\delta W - \delta U}{\delta A}$$

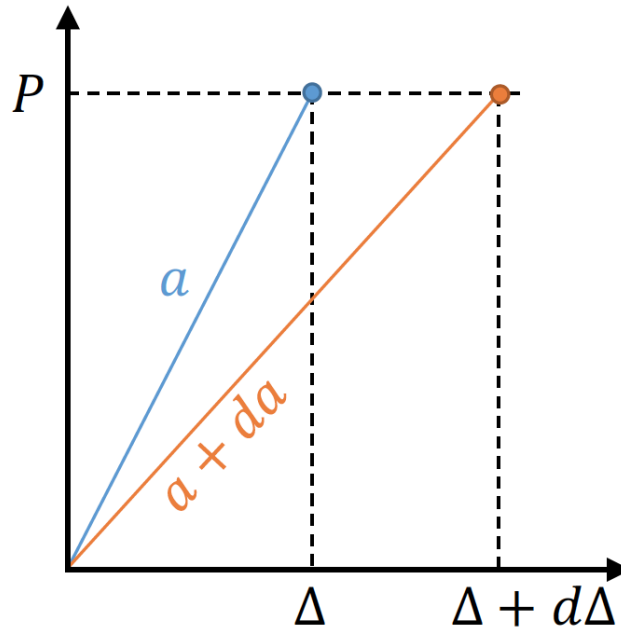
And in differential form:

$$G = -\frac{\partial \Pi}{\partial A} = -\frac{\partial}{\partial A} (U - W)$$

# Introducing the compliance



Constant applied load  $P$



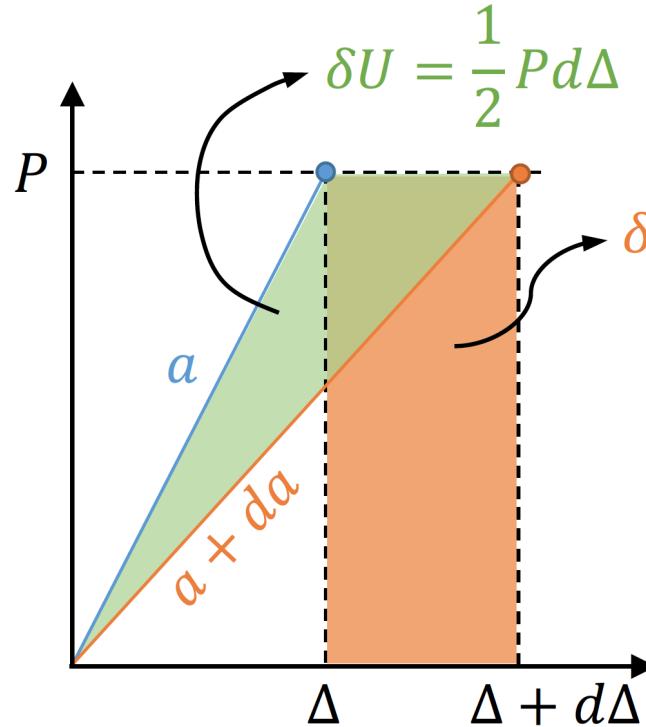
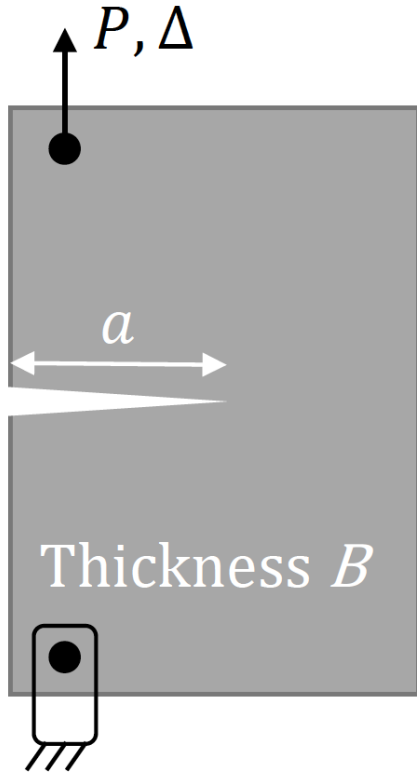
The compliance is:

$$C = \frac{\Delta}{P}$$

If  $P$  is constant, this becomes:

$$d\Delta = P dC$$

# Energy release rate – Load control



We found:  $d\Delta = P dC$

The energy release rate is:

$$\begin{aligned}
 G &= \frac{\delta W - \delta U}{\delta A} \\
 &= \frac{1}{B da} \left( \frac{1}{2} P d\Delta \right) \\
 &= \frac{P^2}{2B} \frac{dC}{da}
 \end{aligned}$$



# Energy release rate

The energy release rate is:

$$G = -\frac{\partial}{\partial A} (U - W) = \frac{P^2}{2B} \frac{dC}{da}$$

↑

Practical and easier to use

Most general definition, suitable to all cases.

Fracture will occur when:

$$G = G_c$$

Where  $G_c$  is a material property called toughness or critical energy release rate, with units of J/m<sup>2</sup>.

# Relation between $K$ and $G$

For an isotropic linear elastic material, the energy release rate  $G$  for a mode I crack is related to the stress intensity factor  $K_I$  according to:

$$K_I^2 = \frac{E}{1 - \nu^2} G$$

*For plane strain*

$$K_I^2 = EG$$

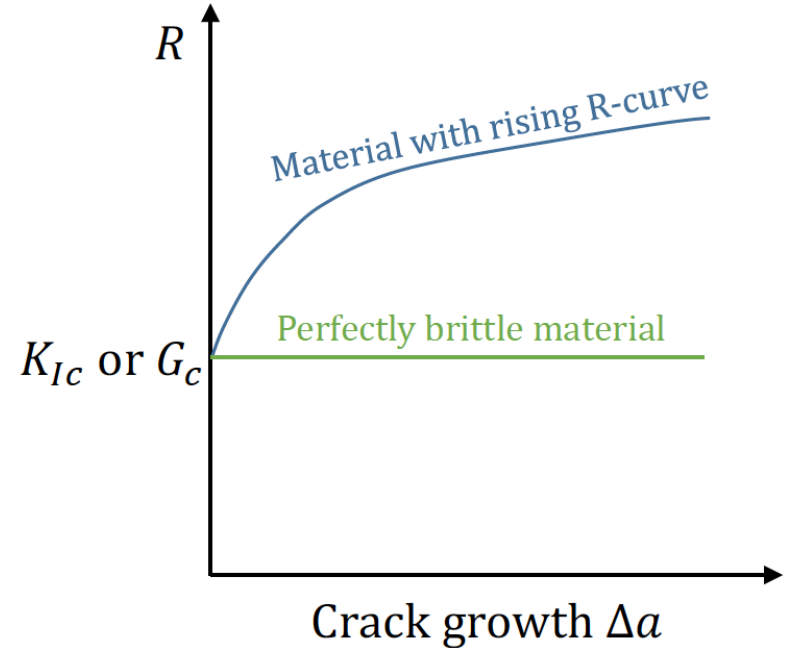
*For plane stress*

# Resistance curve

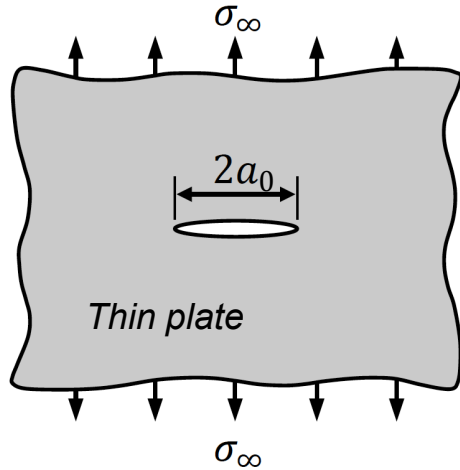
For tough materials, the fracture toughness increases with crack growth.

A plot of  $K_{Ic}$  versus  $\Delta a$  is called a resistance curve (R-curve).

A R-curve is a material property.

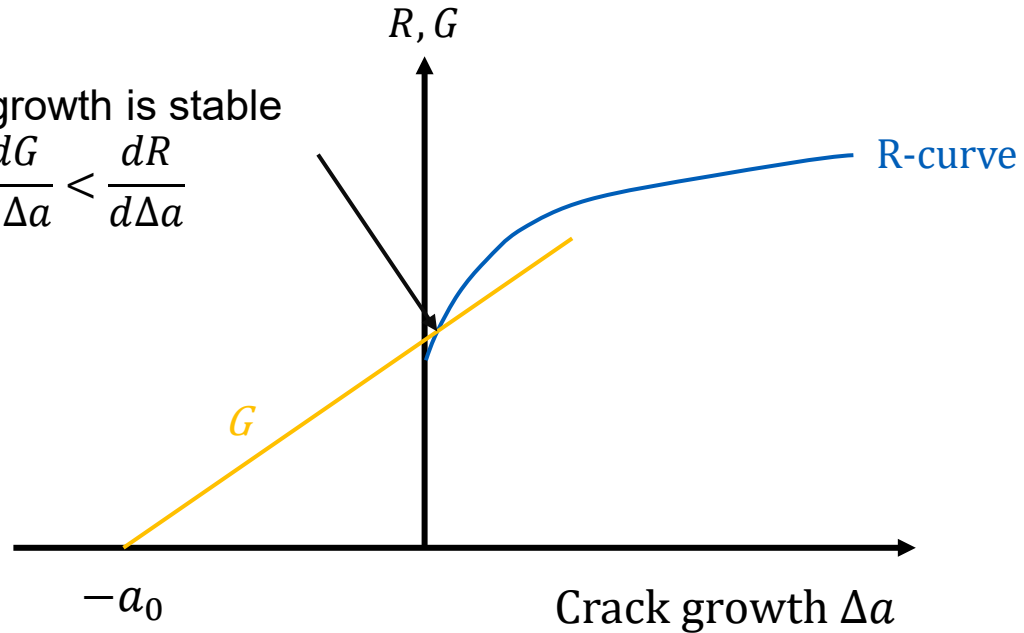


# Stable or unstable crack growth

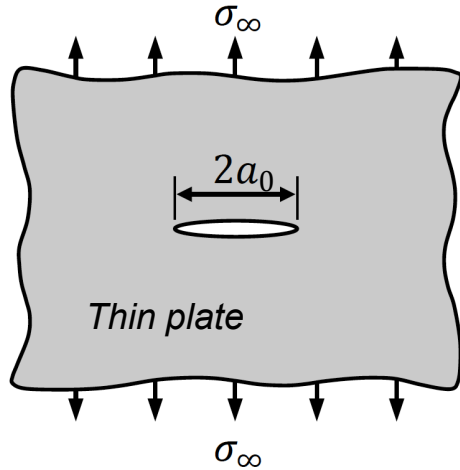


$$G = \frac{K_I^2}{E} = \frac{\sigma_\infty^2 \pi a}{E} = \frac{\sigma_\infty^2 \pi (a_0 + \Delta a)}{E}$$

Crack growth is stable  
 $\frac{dG}{d\Delta a} < \frac{dR}{d\Delta a}$



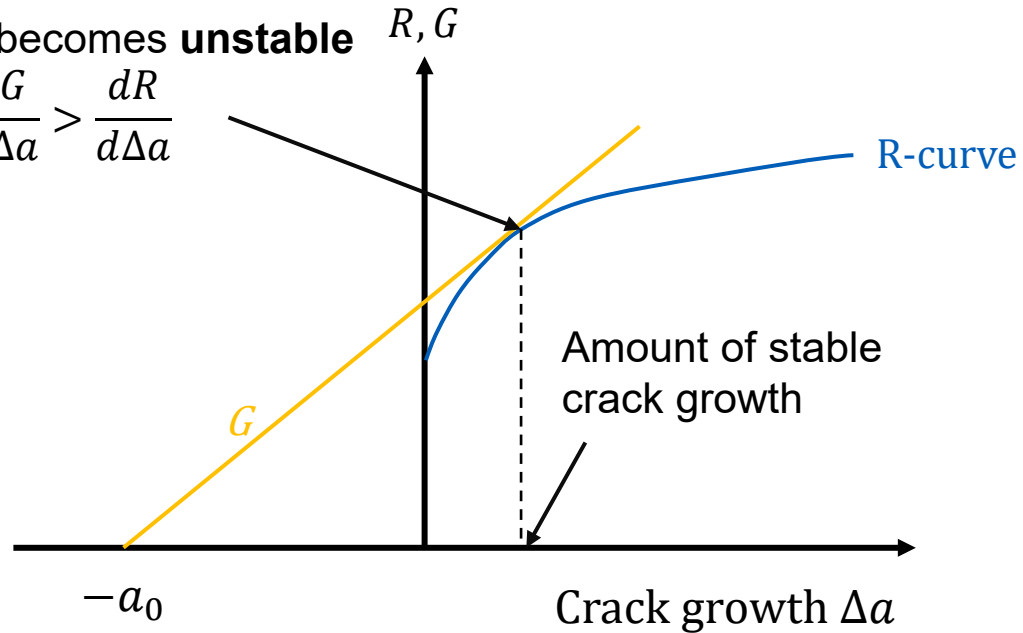
# Stable or unstable crack growth



$$G = \frac{K_I^2}{E} = \frac{\sigma_\infty^2 \pi a}{E} = \frac{\sigma_\infty^2 \pi (a_0 + \Delta a)}{E}$$

Crack growth becomes **unstable**

$$\frac{dG}{d\Delta a} > \frac{dR}{d\Delta a}$$



# Stable or unstable crack growth

Crack growth will be stable when:

$$G = R \quad \text{and} \quad \frac{dG}{da} \leq \frac{dR}{da}$$

Otherwise, crack growth will be unstable when:

$$G \geq R \quad \text{and} \quad \frac{dG}{da} > \frac{dR}{da}$$

Note: the denominator can be  $da$  or  $d\Delta a$  since:  $da = d(a_0 + \Delta a) = d\Delta a$

# In summary

We covered how to:

- Find the energy release rate and use it to predict fracture,
- Evaluate if crack growth will be stable or unstable,
- Calculate the amount of stable crack growth using a R-curve.

Next week, we will study fracture under mixed-mode loading.

# Fracture mechanics

## Seminar 3: Mixed-mode loading



Aalto-yliopisto  
Aalto-universitetet  
Aalto University

Luc St-Pierre

May 10, 2023



# Learning outcomes

After this week, you will be able to:

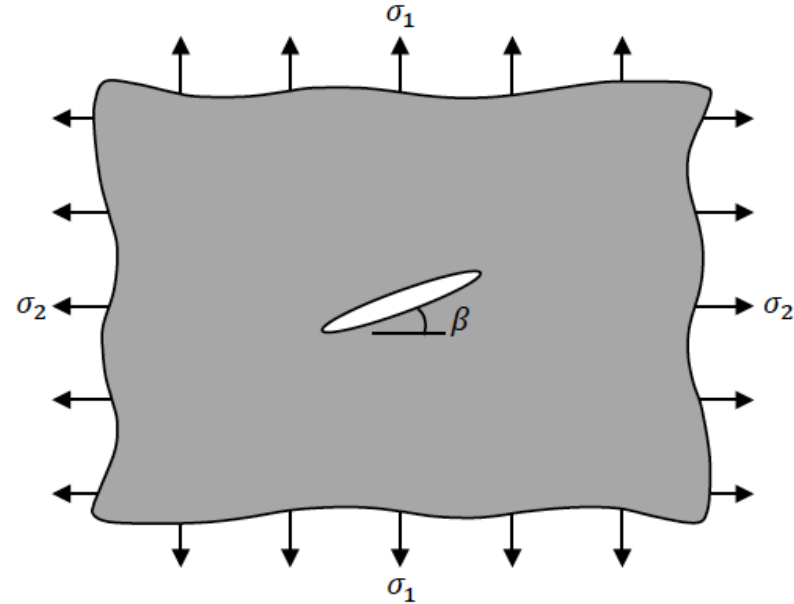
- Calculate the stress intensity factors  $K_I$  and  $K_{II}$  for mixed-mode loading.
- Predict when fracture will occur.
- Find the direction of crack propagation.

# Mixed mode loading

Cracks are often loaded in a combination of mode I and II.

This raises three questions:

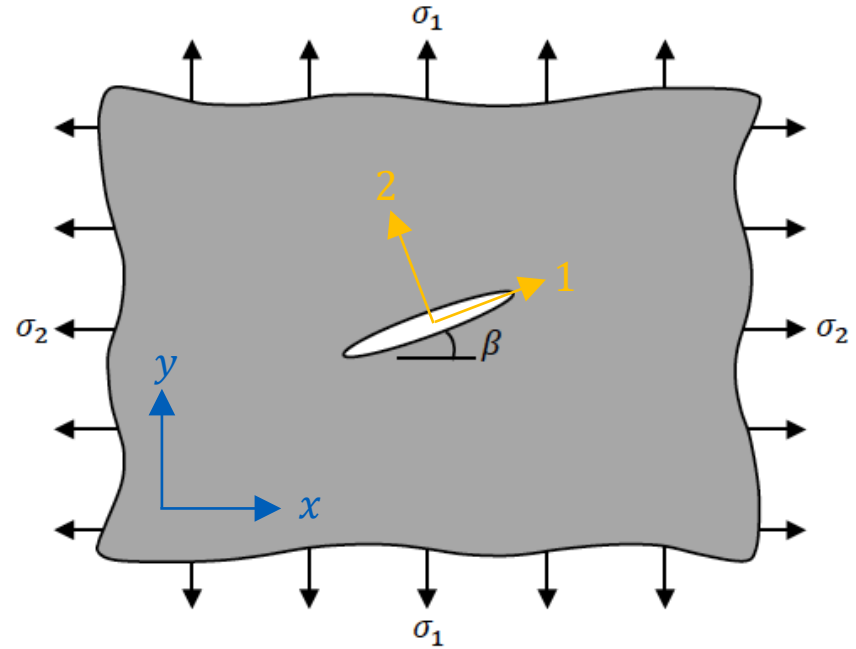
1. How can we compute  $K_I$  and  $K_{II}$ ?
2. When will the crack propagate?
3. In which direction will the crack grow?



# Computing $K_I$ and $K_{II}$

Under mixed-mode loading, we can use Mohr's circle to express the stress components in a reference frame aligned with the crack plane.

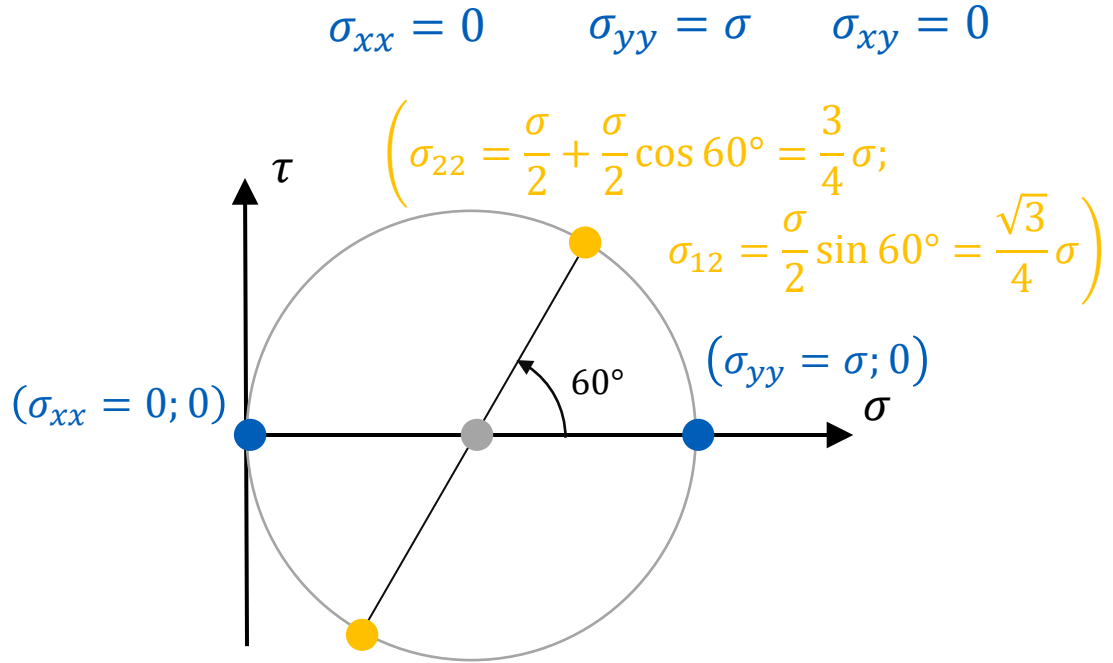
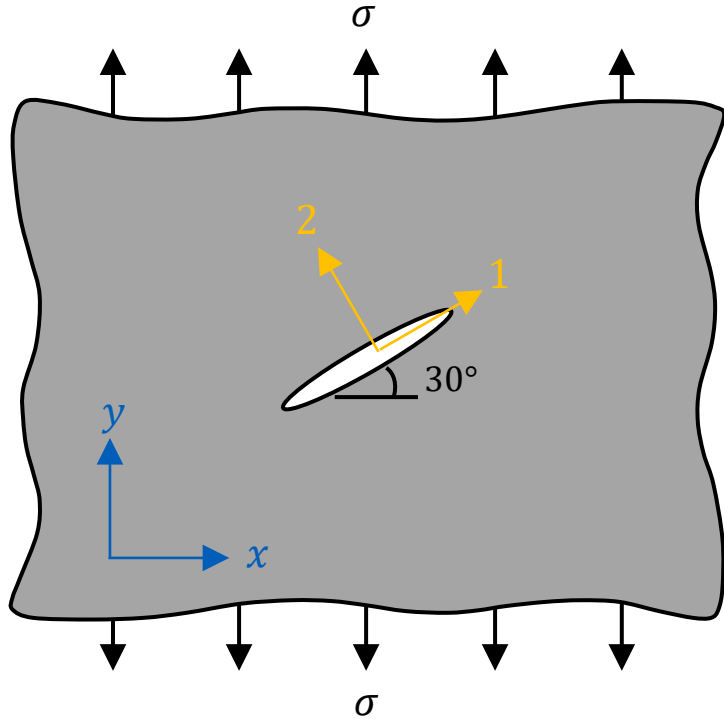
Let's review the Mohr's circle.



# Mohr's circle in three steps

1. Write the stress components in the global reference frame.
  - Three components:  $\sigma_{xx}$ ;  $\sigma_{yy}$ ;  $\sigma_{xy}$ .
  - Be careful with signs: tension/compression.
2. Draw the circle using two points on  $\sigma$  vs  $\tau$  axes:
  - First point:  $(\sigma_{xx}; -\sigma_{xy})$  and second point:  $(\sigma_{yy}; \sigma_{xy})$ .
3. Rotate clockwise by  $2\theta$  to find  $\sigma_{11}$ ;  $\sigma_{22}$ ;  $\sigma_{12}$ .

# Mohr's circle: example



$$K_I = \sigma_{22} \sqrt{\pi a} = \frac{3}{4} \sigma \sqrt{\pi a}$$

$$K_{II} = \sigma_{12} \sqrt{\pi a} = \frac{\sqrt{3}}{4} \sigma \sqrt{\pi a}$$

# How to predict fracture

**NO:** you cannot add stress intensity factors for different modes:

$$K^{[\text{total}]} \neq K_I^{[A]} + K_{II}^{[B]} + K_{III}^{[C]} + \dots$$

**YES:** you can add the energy release rate of different modes:

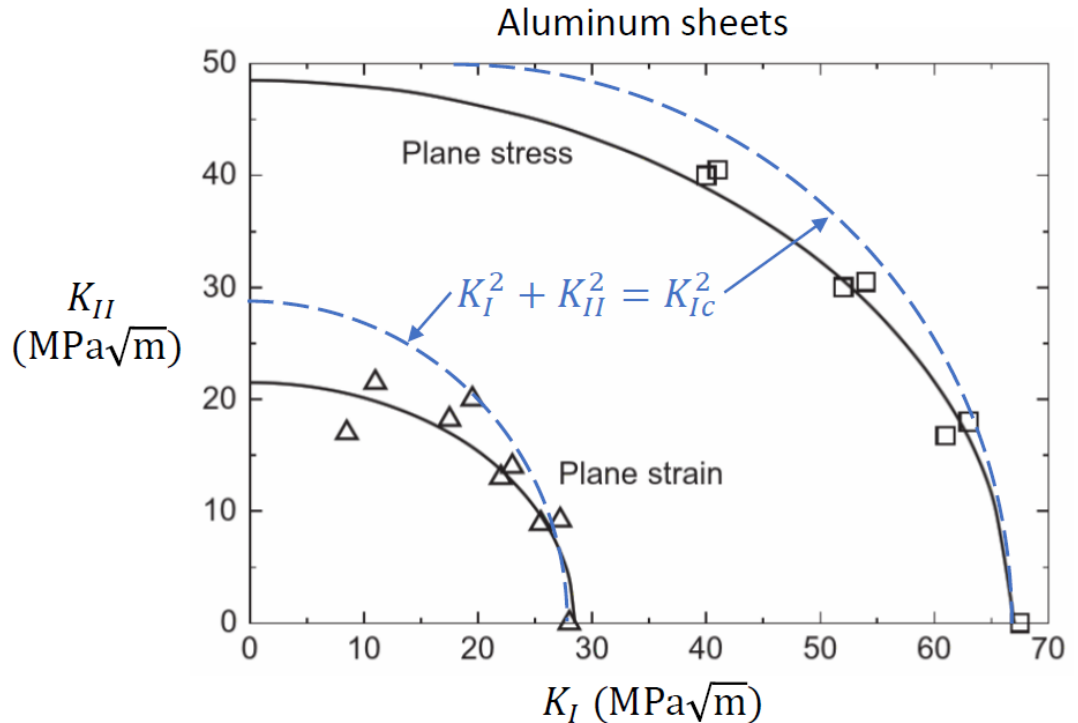
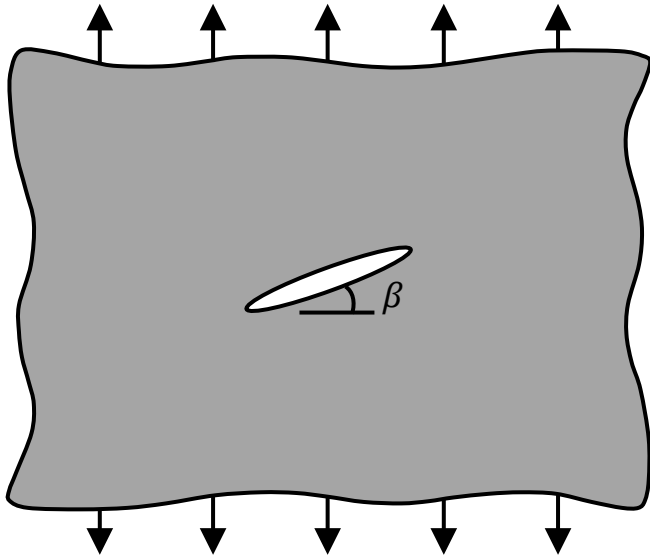
$$G = \frac{K_I^2}{E'} + \frac{K_{II}^2}{E'}$$

Where  $E' = E$  in plane stress and  $E' = E/(1 - \nu^2)$ . A simple fracture criterion is to set  $G = G_c = K_{Ic}^2/E'$ .

# Fracture envelope

Multiple experiments are needed to form a fracture envelope.

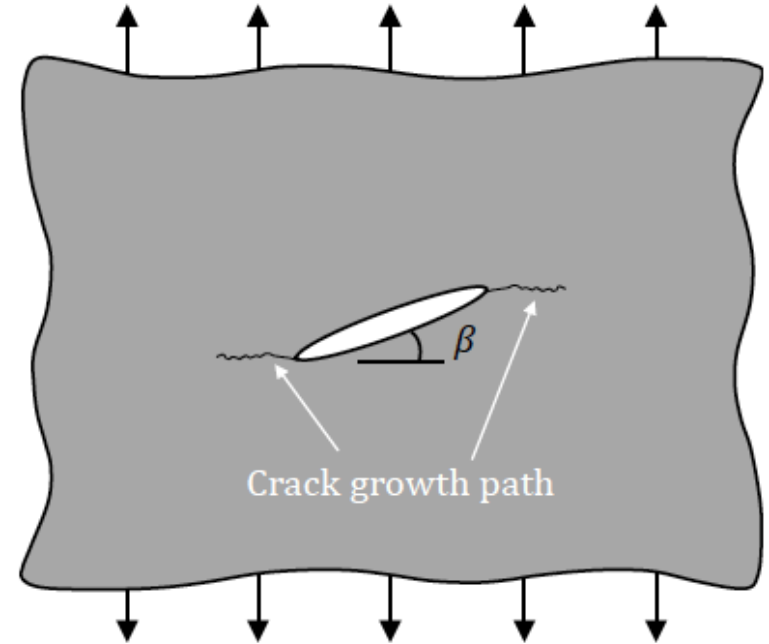
Vary  $\beta$  to change  $K_I/K_{II}$



# Angle of crack propagation

Cracks generally grow in the local mode I direction. This is where:

- $\sigma_{\theta\theta}$  is maximum,
- which is the same direction where  $\sigma_{r\theta} = 0$ .





# Stress field under mixed-mode loading

Add stress components for modes I and II to get the total stress field close to the crack tip:

$$\sigma_{rr} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{5}{4} \cos \frac{\theta}{2} - \frac{1}{4} \cos \frac{3\theta}{2} \right) + \frac{K_{II}}{\sqrt{2\pi r}} \left( -\frac{5}{4} \sin \frac{\theta}{2} + \frac{3}{4} \sin \frac{3\theta}{2} \right)$$

$$\sigma_{\theta\theta} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{3}{4} \cos \frac{\theta}{2} + \frac{1}{4} \cos \frac{3\theta}{2} \right) - \frac{K_{II}}{\sqrt{2\pi r}} \left( \frac{3}{4} \sin \frac{\theta}{2} + \frac{3}{4} \sin \frac{3\theta}{2} \right)$$

$$\sigma_{r\theta} = \underbrace{\frac{K_I}{\sqrt{2\pi r}} \left( \frac{1}{4} \sin \frac{\theta}{2} + \frac{1}{4} \sin \frac{3\theta}{2} \right)}_{\sigma_{ij} \text{ for mode I (from datasheet)}} + \underbrace{\frac{K_{II}}{\sqrt{2\pi r}} \left( \frac{1}{4} \cos \frac{\theta}{2} + \frac{3}{4} \cos \frac{3\theta}{2} \right)}_{\sigma_{ij} \text{ for mode II (from datasheet)}}$$

# Angle of crack propagation

First, set  $\sigma_{r\theta} = 0$ , which gives:

$$0 = K_I \left( \sin \frac{\theta}{2} + \sin \frac{3\theta}{2} \right) + K_{II} \left( \cos \frac{\theta}{2} + 3 \cos \frac{3\theta}{2} \right)$$

And solve for  $\theta$ . You should find multiple values for  $\theta$ .

Second, the crack will propagate in the direction  $\theta$  corresponding to the maximum  $\sigma_{\theta\theta}$ .

$$\sigma_{\theta\theta} = \frac{K_I}{\sqrt{2\pi r}} \left( \frac{3}{4} \cos \frac{\theta}{2} + \frac{1}{4} \cos \frac{3\theta}{2} \right) - \frac{K_{II}}{\sqrt{2\pi r}} \left( \frac{3}{4} \sin \frac{\theta}{2} + \frac{3}{4} \sin \frac{3\theta}{2} \right)$$

# In summary

Under mixed-mode loading:

- find the stress intensity factors  $K_I$  and  $K_{II}$  using Mohr's circle,
- use the energy release rate  $G$  to predict when fracture will occur,
- the crack will propagate in the direction of  $\max(\sigma_{\theta\theta})$ , which corresponds to  $\sigma_{r\theta} = 0$ .

This concludes the part on Linear Elastic Fracture Mechanics (LEFM).  
Next week, we will introduce plasticity in our analysis of fracture.

# Fracture mechanics

## Seminar 4: Plastic zone size



Aalto-yliopisto  
Aalto-universitetet  
Aalto University

Luc St-Pierre

May 17, 2023

# Learning outcomes

After this week, you should be able to:

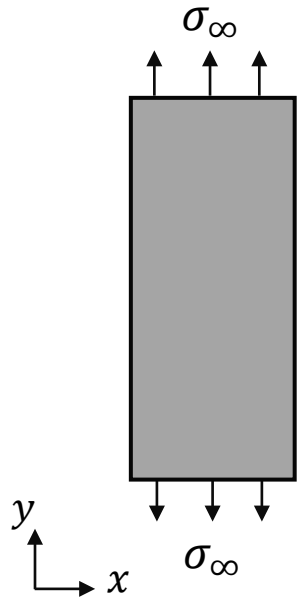
- Evaluate the plastic zone size,
- Assess when it is adequate to use LEFM,
- Design to prevent both fracture and yielding.

# Crack tip plasticity

- So far, we have studied fracture assuming a linear elastic material.
  - That is Linear Elastic Fracture Mechanics (LEFM).
- Many materials have plasticity (metals) or inelastic deformation (polymers).
- Is LEFM applicable when we have plasticity?
  - Yes, if the size of the plastic zone is small.

# Yielding criterion

## Uniaxial loading



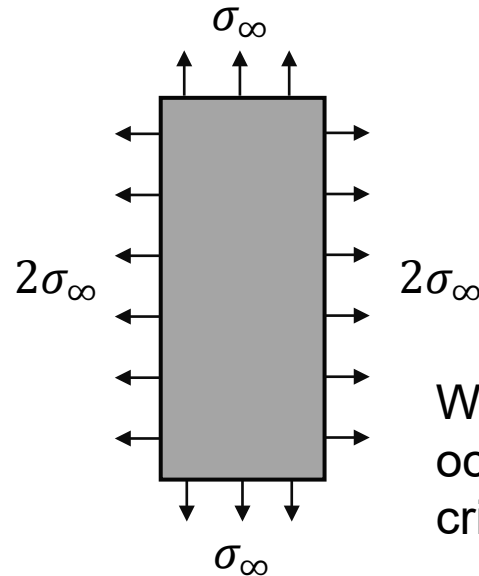
$$\sigma_{xx} = 0$$

$$\sigma_{yy} = \sigma_{\infty}$$

$$\sigma_{xy} = 0$$

Yielding will occur  
when  $\sigma_{\infty} = \sigma_Y$

## Multi-axial loading



$$\sigma_{xx} = 2\sigma_{\infty}$$

$$\sigma_{yy} = \sigma_{\infty}$$

$$\sigma_{xy} = 0$$

When will yielding will  
occur? A yielding  
criterion is needed.

# Yielding criterion

The von Mises yielding criterion can be written as:

$$\frac{1}{2} \left[ (\sigma_{xx} - \sigma_{yy})^2 + (\sigma_{yy} - \sigma_{zz})^2 + (\sigma_{zz} - \sigma_{xx})^2 + 6(\sigma_{xy}^2 + \sigma_{yz}^2 + \sigma_{xz}^2) \right] = \sigma_Y^2$$

If there are no shear stresses then we have three principal stresses  $\sigma_1, \sigma_2, \sigma_3$ , and this becomes:

$$\frac{1}{2} [(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_1 - \sigma_3)^2] = \sigma_Y^2$$



# Plastic zone size: Irwin's approach

# The Irwin approach

Irwin proposed a simple estimate of the plastic zone size. His approach:

- Used the LEFM stress field,
- Considered only stresses on the crack plane,  $\theta = 0$ , and looked for when these stresses would exceed the yield strength  $\sigma_Y$ .

# The Irwin approach

From the datasheet, the mode I stress field for  $\theta = 0$  is:

$$\sigma_{yy}(\theta = 0) = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \left( 1 + \sin \frac{\theta}{2} \sin \frac{3\theta}{2} \right) = \frac{K_I}{\sqrt{2\pi r}}$$

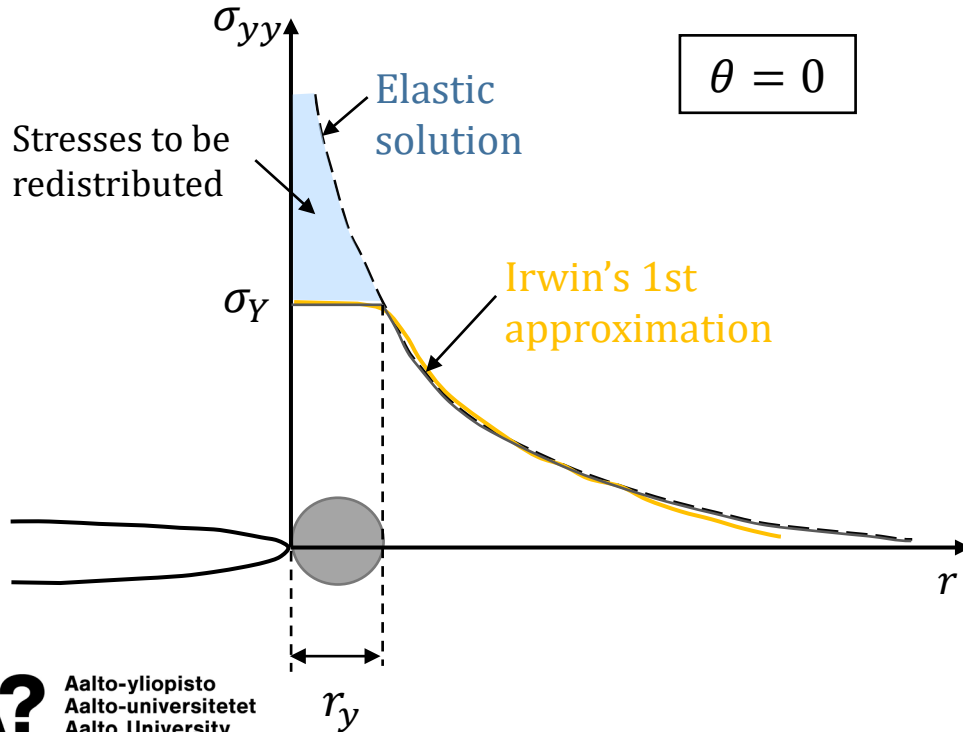
$$\sigma_{xx}(\theta = 0) = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \left( 1 - \sin \frac{\theta}{2} \sin \frac{3\theta}{2} \right) = \frac{K_I}{\sqrt{2\pi r}}$$

$$\sigma_{xy}(\theta = 0) = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \sin \frac{\theta}{2} \cos \frac{3\theta}{2} = 0$$

$$\sigma_{zz}(\theta = 0) = \begin{cases} 0 & \text{for plane stress} \\ \nu(\sigma_{xx} + \sigma_{yy}) = \frac{2\nu K_I}{\sqrt{2\pi r}} & \text{for plane strain} \end{cases}$$

Next, substitute these expressions in the von Mises yielding criterion and solve for  $r$  to get the size of the plastic zone

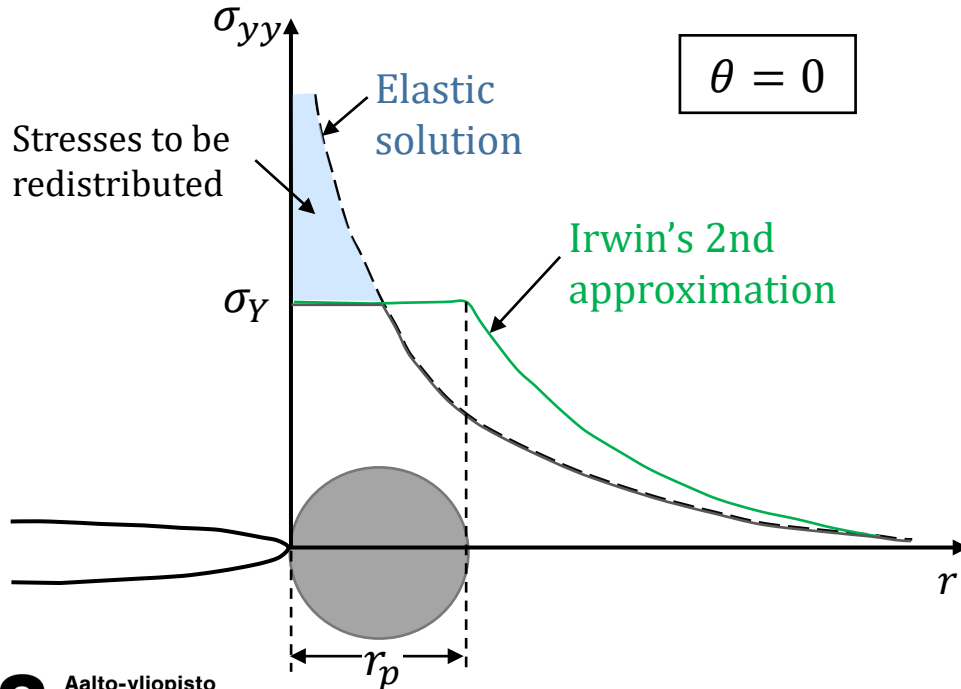
# The Irwin approach



Plane stress

$$r_y = \frac{1}{2\pi} \left( \frac{K_I}{\sigma_Y} \right)^2$$

# The Irwin approach



Plane stress

$$r_p = \frac{1}{\pi} \left( \frac{K_I}{\sigma_Y} \right)^2$$

Plane strain

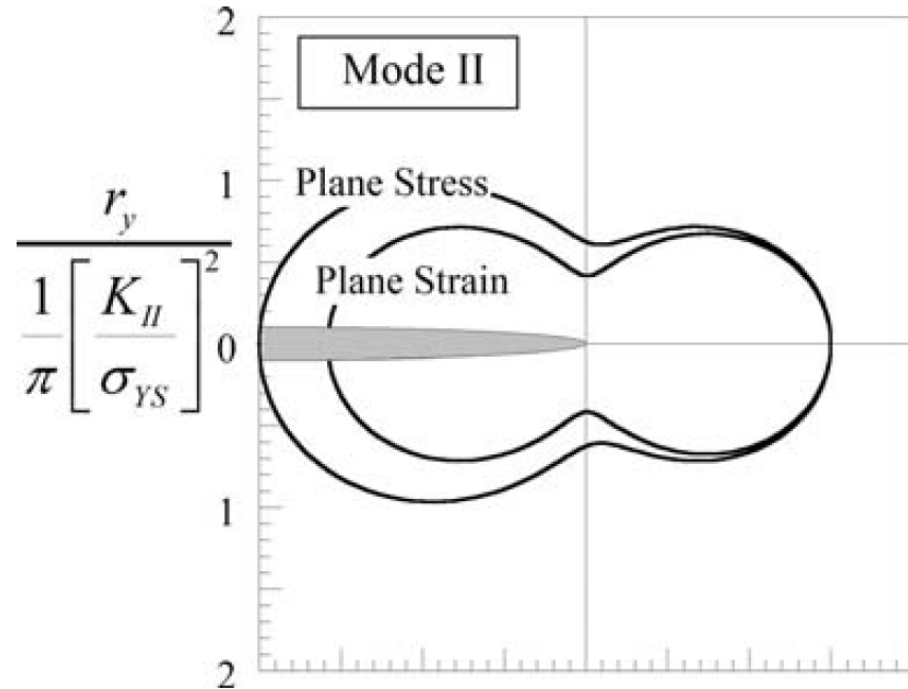
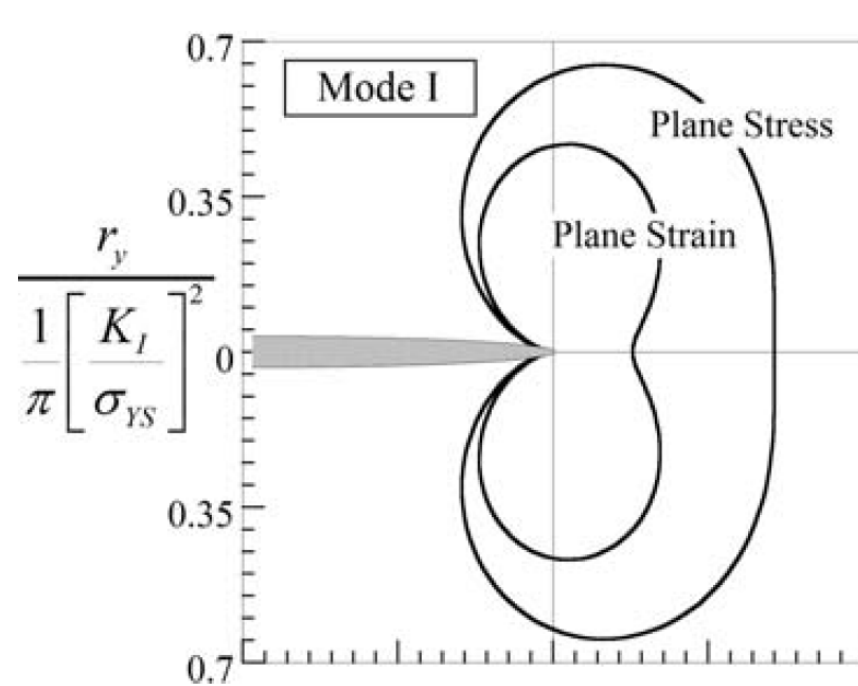
$$r_p = \frac{1}{3\pi} \left( \frac{K_I}{\sigma_Y} \right)^2$$

# Plastic zone shape

The Irwin approach gives a scalar and not the shape of the plastic zone size. To find its shape, you need to:

- For a given mode, select the stress field from the datasheet and substitute it in the von Mises yielding criterion.
- Solve for  $r$  as a function of  $\theta$  to find the shape of the plastic zone.

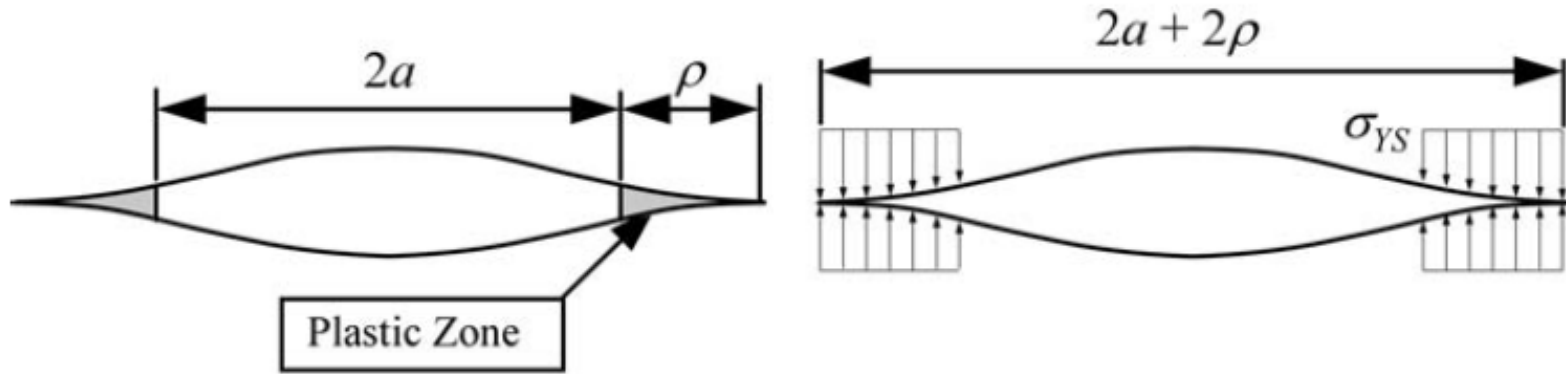
# Plastic zone shapes



# Plastic zone size: The strip-yield model

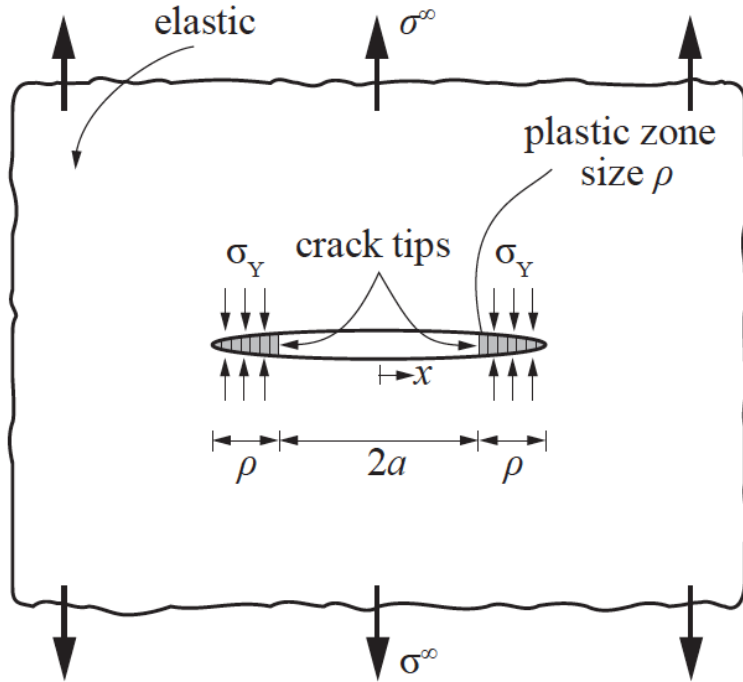


# Strip-yield model



The strip-yield model replaces the physical crack of length  $2a$  by a fictitious crack of length  $2(a + \rho)$ . A closing stress  $\sigma_Y$  is keeping a portion  $\rho$  closed.

# Strip-yield model



At the fictitious crack tip, we have:

$$K_I^{(tot)} = K_I^{(\sigma^\infty)} + K_I^{(\sigma_Y)} = 0$$

where

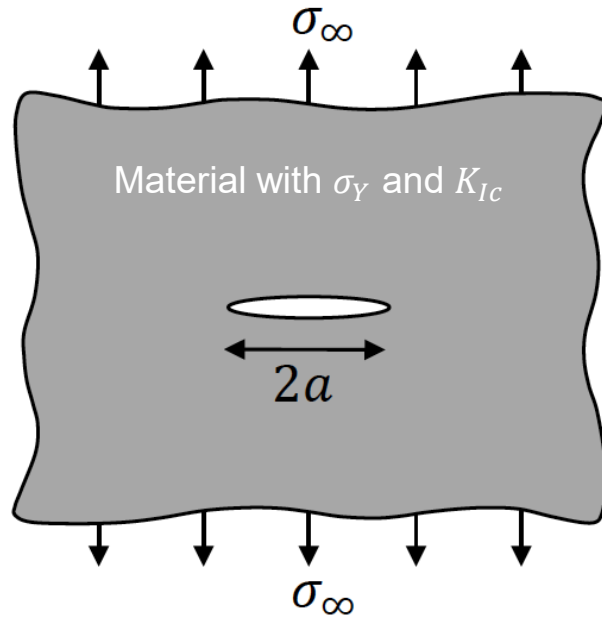
$$\begin{cases} K_I^{(\sigma^\infty)} = \sigma^\infty \sqrt{\pi(a + \rho)} \\ K_I^{(\sigma_Y)} = -2\sigma_Y \sqrt{\frac{a + \rho}{\pi}} \cos^{-1} \left( \frac{a}{a + \rho} \right) \end{cases}$$

After a bit of algebra, we get:  $\rho = \frac{\pi}{8} \left( \frac{K_I}{\sigma_Y} \right)^2$

(20% higher than Irwin's approach)

# Failure mechanisms: Yielding vs Fracture

# Yielding vs Fracture



Fracture will occur when:

$$\sigma_\infty = \frac{K_{Ic}}{\sqrt{\pi a}}$$

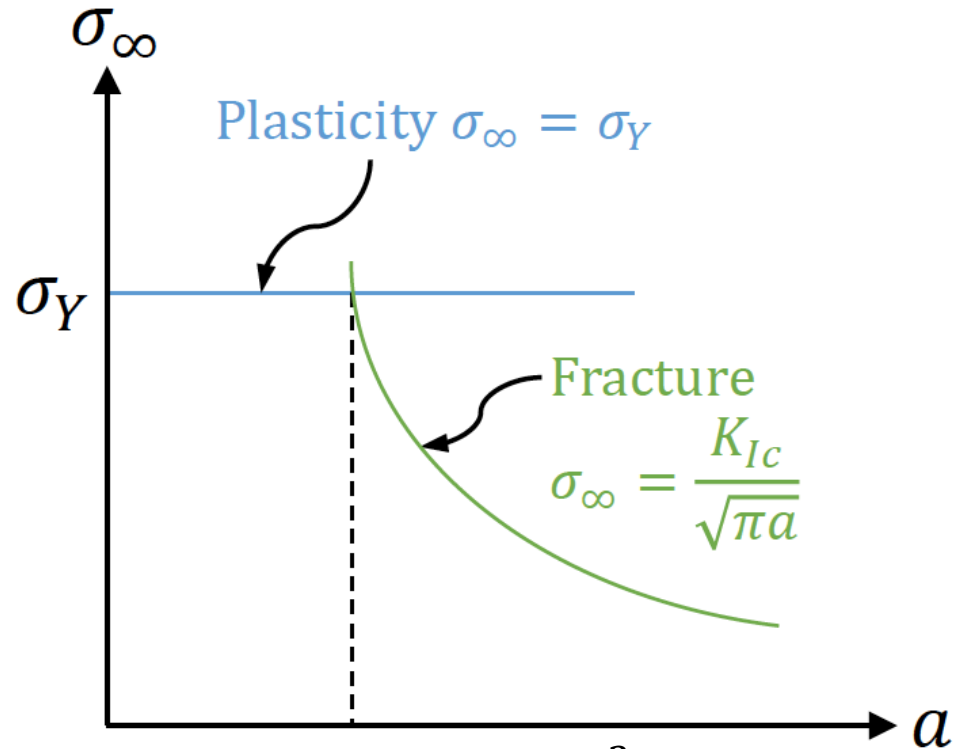
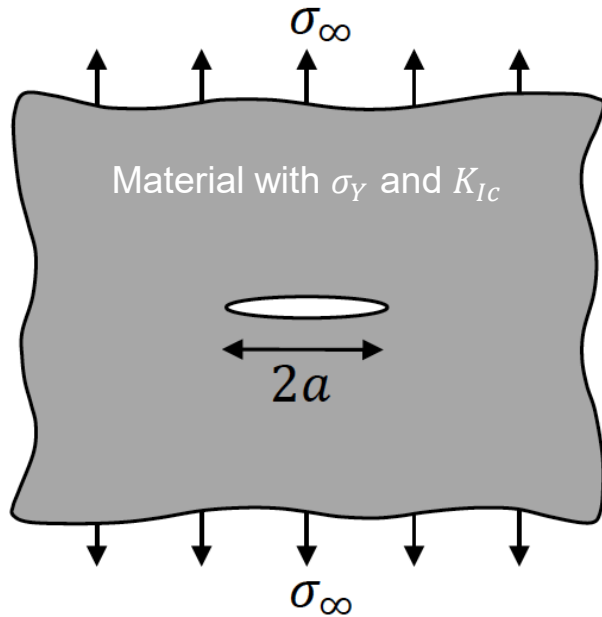
Otherwise, for a short crack (or  $a = 0$ ), yielding will occur when:

$$\sigma_\infty = \sigma_Y$$

The maximum allowable stress is:

$$\sigma_\infty = \min \left( \sigma_Y ; \frac{K_{Ic}}{\sqrt{\pi a}} \right)$$

# Yielding vs Fracture



$$a_T = \frac{1}{\pi} \left( \frac{K_{Ic}}{\sigma_Y} \right)^2$$

# In summary

We covered how to:

- Estimate the plastic zone size  $r_p$ ,
- Assess if LEFM is applicable. It is when  $r_p < a/10$ ,
- Design to prevent both yielding and fracture.

Next week, we will cover the J-integral; fracture mechanisms and testing.

# Fracture mechanics

## Seminar 5: J-integral, testing, and more



Aalto-yliopisto  
Aalto-universitetet  
Aalto University

Luc St-Pierre

May 24, 2023

# Learning outcomes

After this week, you should be able to:

- Understand and use the  $J$ -integral,
- Explain how to measure the fracture toughness,
- Describe the main fracture mechanisms in metals and composites.

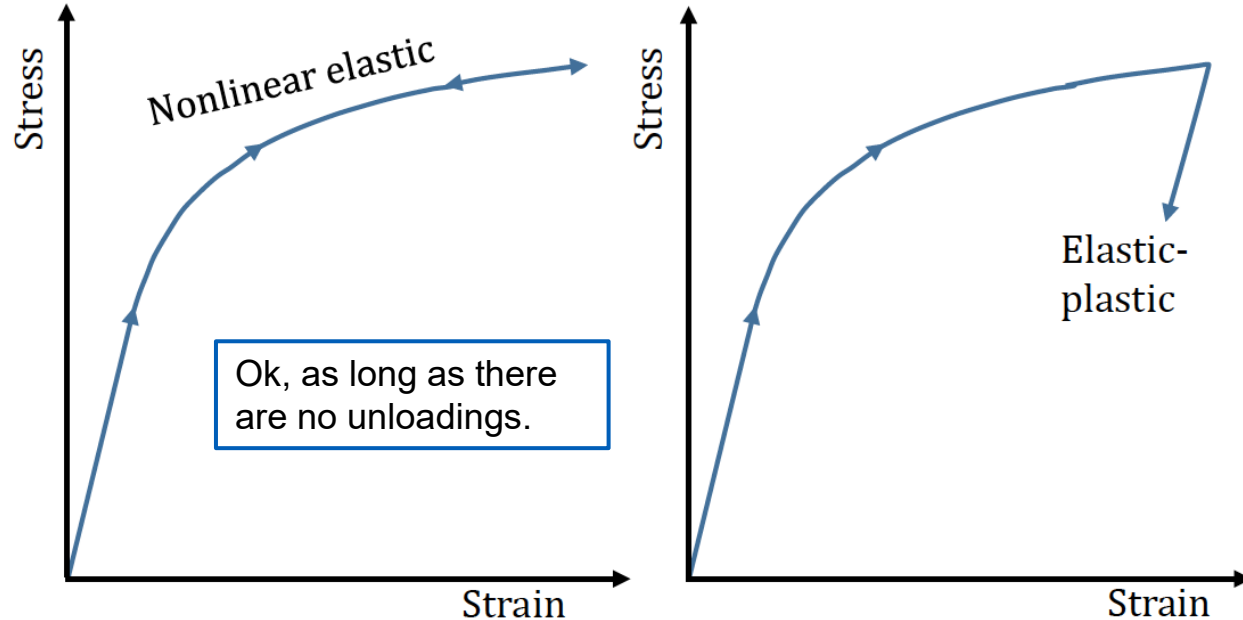


# Elastic-Plastic Fracture Mechanics

- Last week, we saw how to estimate the size of the plastic zone at the crack tip.
  - If the plastic zone size is small ( $r_p < a/10$ ), you can use LEFM.
- What can we do if the plastic zone size is large?
  - Use the  $J$ -integral.
  - Fracture will occur when:  $J = J_{Ic}$

# $J$ -integral: material model

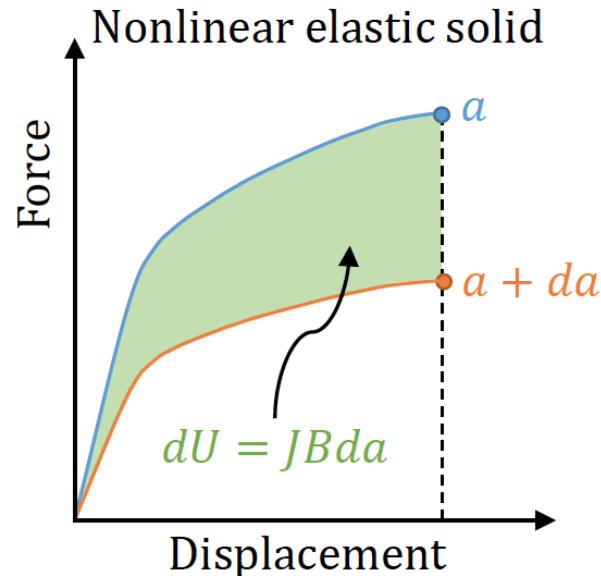
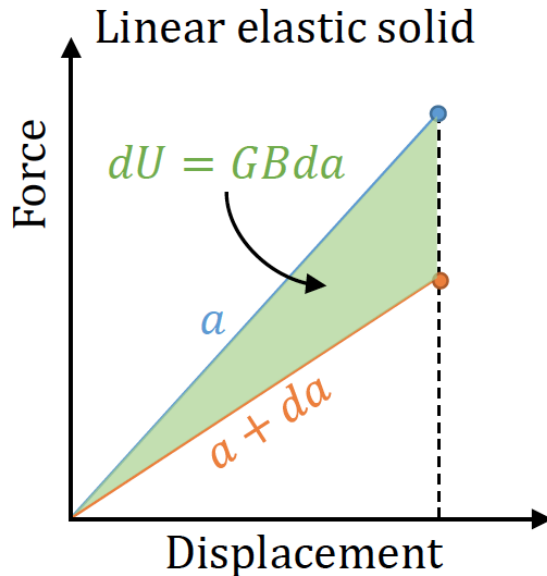
- The  $J$ -integral is developed for a non-linear elastic material.
- This is different from the elastic-plastic behavior of most metals.



# J-integral: definition

The  $J$ -integral is defined just like the energy release rate  $G$ :

$$J = -\frac{d\Pi}{dA} \quad \text{where} \quad \Pi = U - W$$



If the material is linear elastic then  $J = G$ .

# J-integral and the stress field

Assuming that the stress-strain curve of the material follows the Ramberg-Osgood equation:

$$\varepsilon = \frac{\sigma}{E} + K \left( \frac{\sigma}{E} \right)^n$$

Hutchinson, Rice and Rosengren showed that stresses at the crack tip scale as:

$$\sigma_{ij} \propto \left( \frac{J}{r} \right)^{\frac{1}{n+1}}$$

# $J$ as a contour integral

$$J = \int_{\Gamma} \left( w dy - t_i \frac{\partial u_i}{\partial x} ds \right)$$

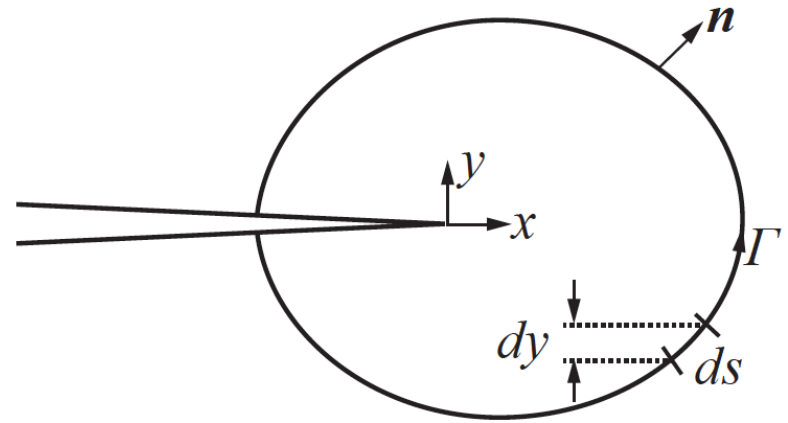
Where,

Strain energy:  $w = \int_0^{\varepsilon_{ij}} \sigma_{ij} d\varepsilon_{ij}$

Traction vector:  $t_i = \sigma_{ij} n_j$

Vector normal to contour:  $n_j$

Displacement vector:  $u_i$



The  $J$ -integral is contour independent.

The  $J$ -integral can be calculated easily in a finite element analysis.

# Example problem

$$J = \int_{\Gamma} \left( w dy - t_i \frac{\partial u_i}{\partial x} ds \right)$$

Where,

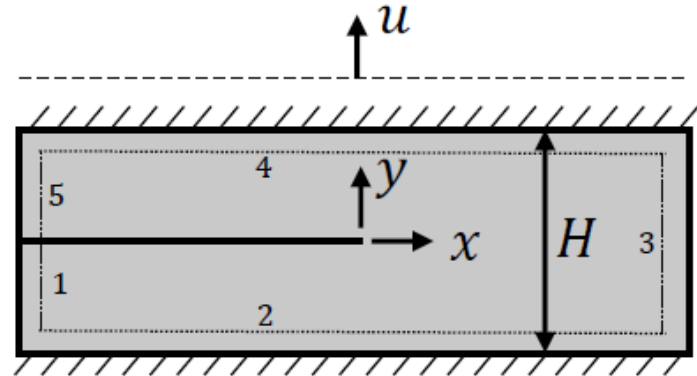
Strain energy:  $w = \int_0^{\varepsilon_{ij}} \sigma_{ij} d\varepsilon_{ij}$

Traction vector:  $t_i = \sigma_{ij} n_j$

Vector normal to contour:  $n_j$

Displacement vector:  $u_i$

Determine the  $J$ -integral for the infinitely wide strip below. Assume that the material is linear elastic, isotropic, and under plane stress.

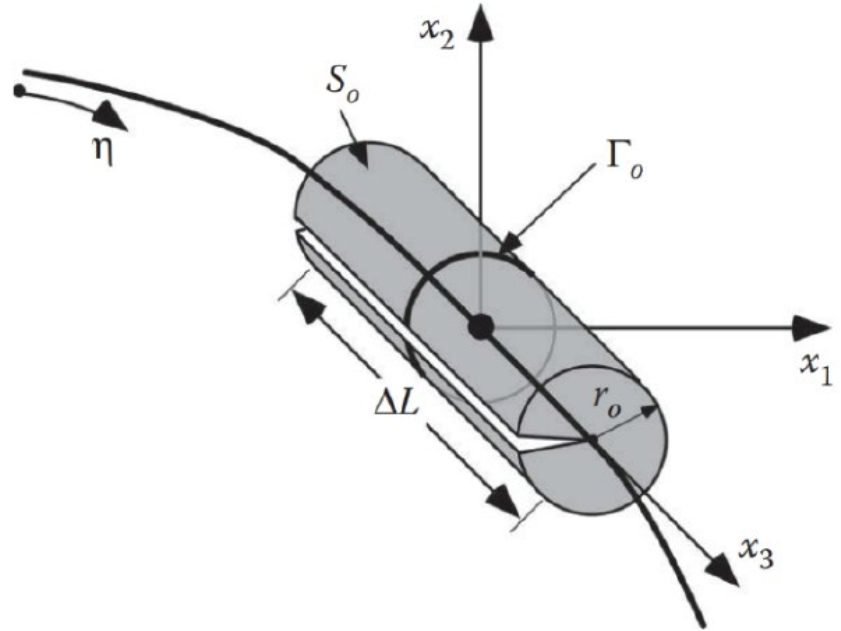


# Computational implementation

# Contour integral

Most Finite Element packages can compute the  $J$ -integral.

- Its definition has been extended to 3D cracks.
- The software may be able to convert  $J$  to  $K_I, K_{II}, K_{III}$ .

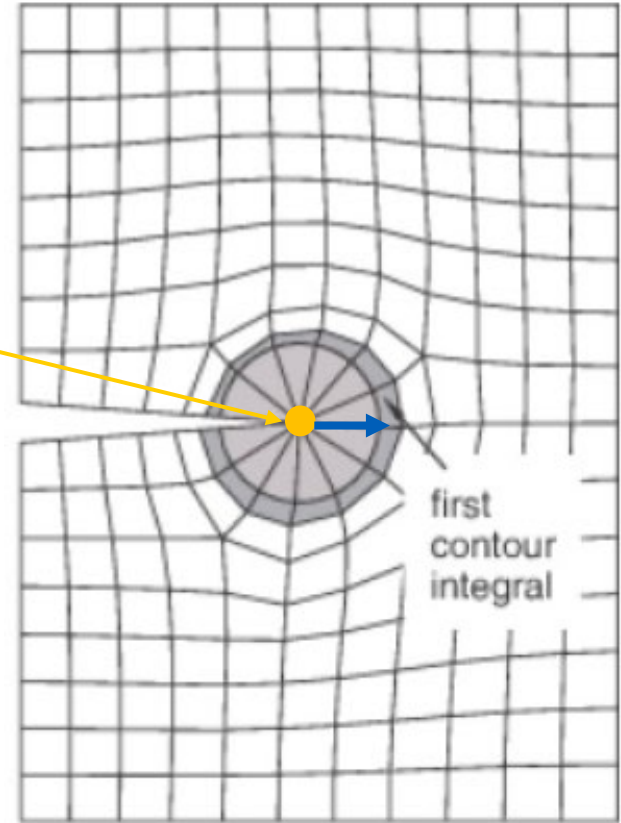




# Contour integral

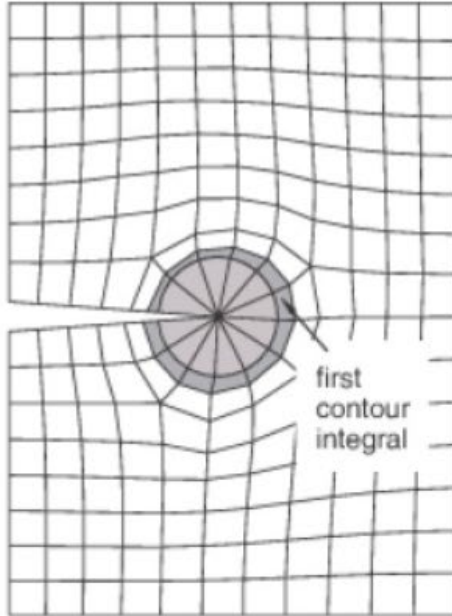
To compute the contour integral, you need to provide:

1. A crack tip (2D) or crack front (3D),
2. The direction of crack propagation (shown here in **blue**),
3. The number of contours.

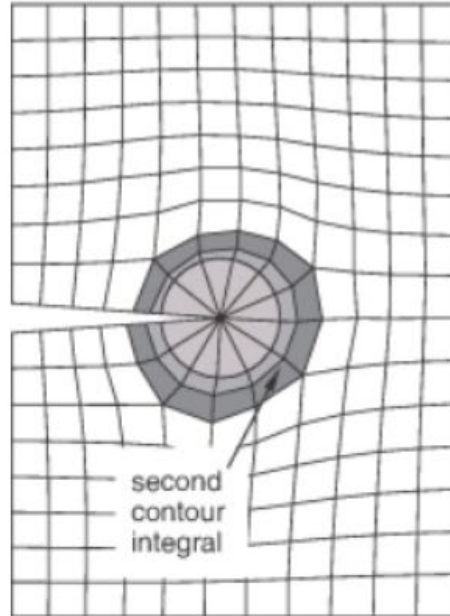


# Number of contours

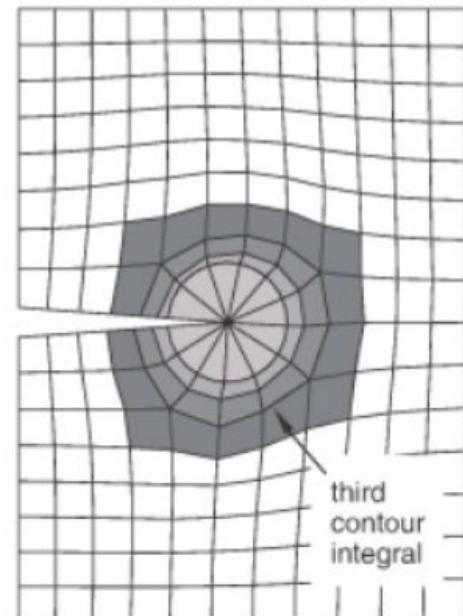
1 contour



2 contours



3 contours



# Contour integral

- The  $J$ -integral should converge to a certain value after a few contours.
  - How many contours? This is highly dependent on the mesh size and on the problem.
- **Warning:** results may diverge if you request more contours than there are elements!

# Fracture testing

# Fracture testing

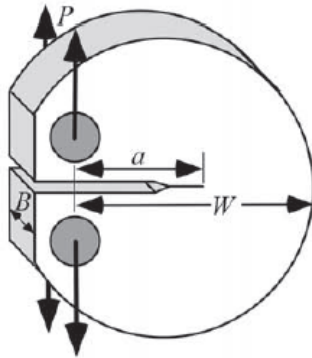
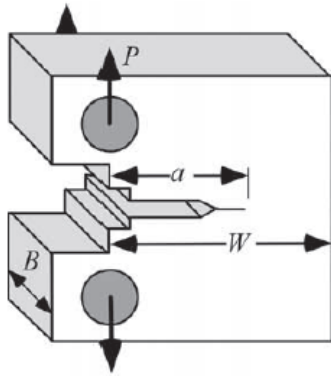
Measuring the fracture toughness is complex. Consult the relevant standard, e.g. ASTM E1820.

There are two testing methods:

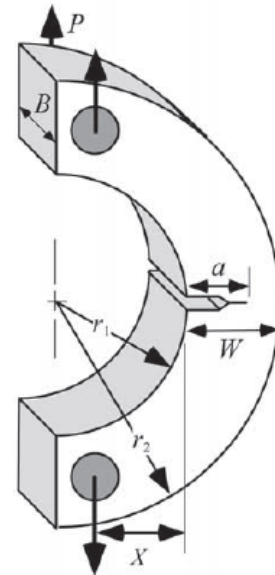
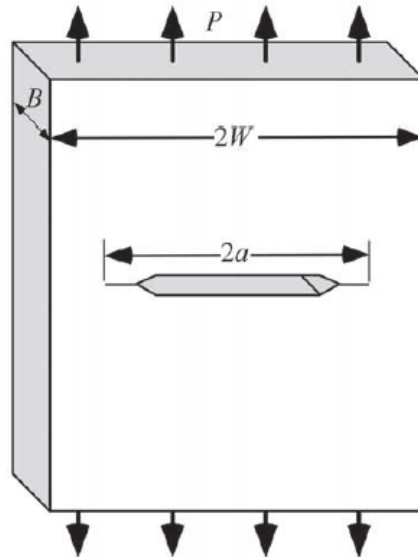
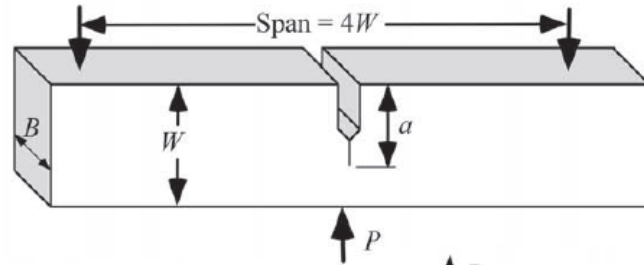
1. To measure the fracture toughness  $K_{Ic}$
2. Measure the R-curve using the  $J$ -integral.

# Specimen geometries

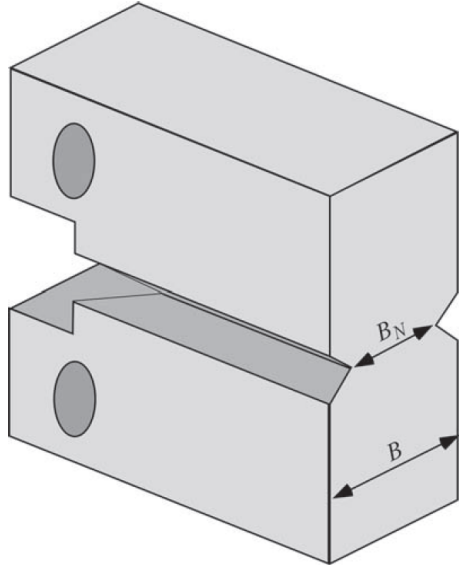
Compact Tension (CT)



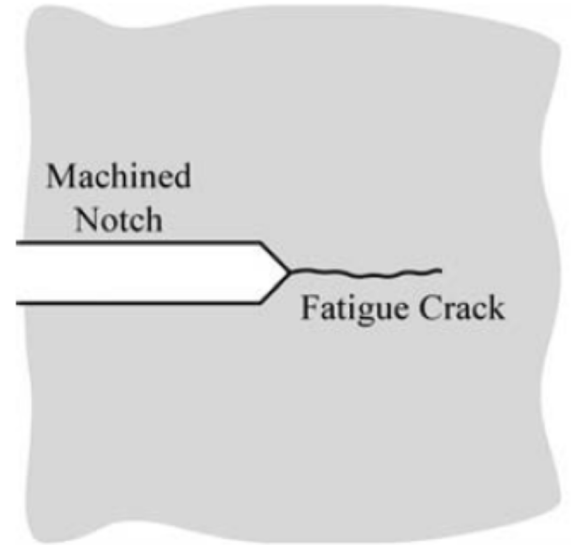
Single-Edge-Notched Bend (SENB)



# Side grooves and precrack

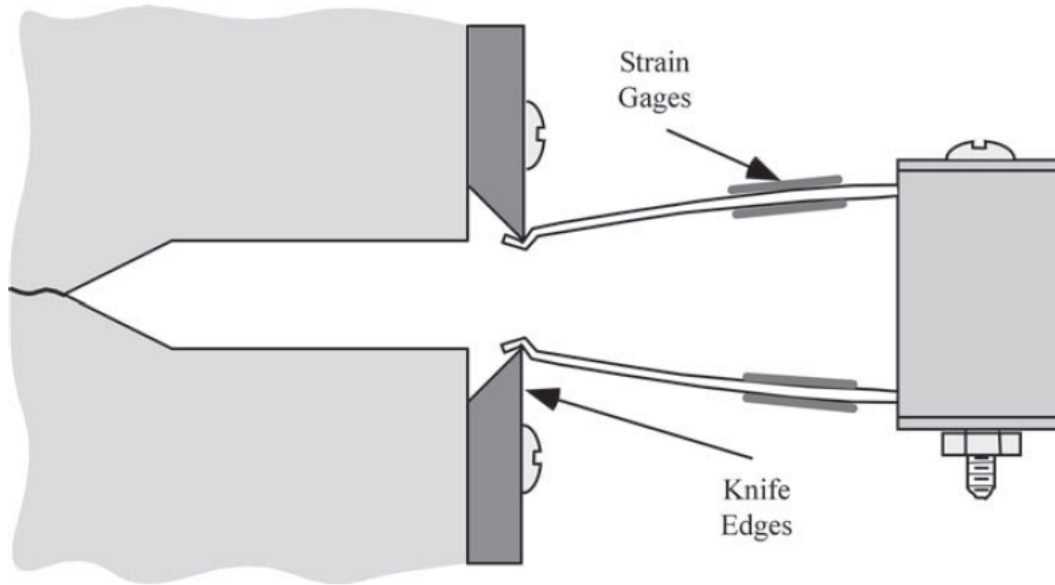


Side grooves help to propagate a straight crack.



For metals, fatigue is the only way to produce a sharp crack.

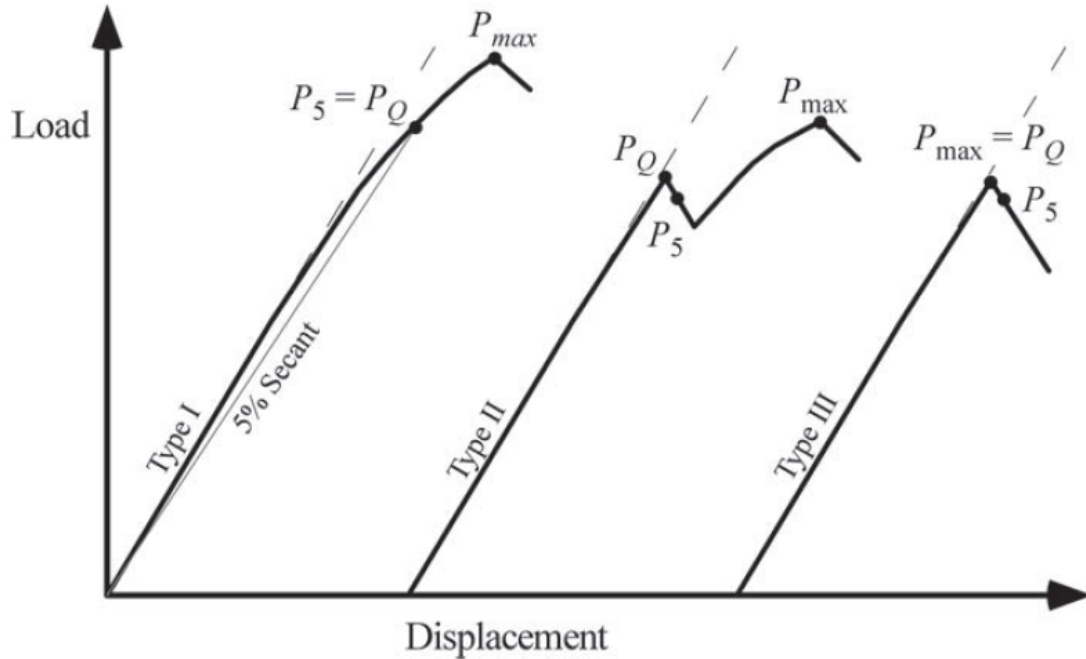
# Instrumentation



- Displacement is measured at the crack mouth by a clip-gauge.
- Force is measured by the testing machine.



# Method 1: $K_{Ic}$



Calculate the stress intensity factor with:

$$K_Q = \frac{P_Q}{B\sqrt{W}} f\left(\frac{a}{W}\right)$$

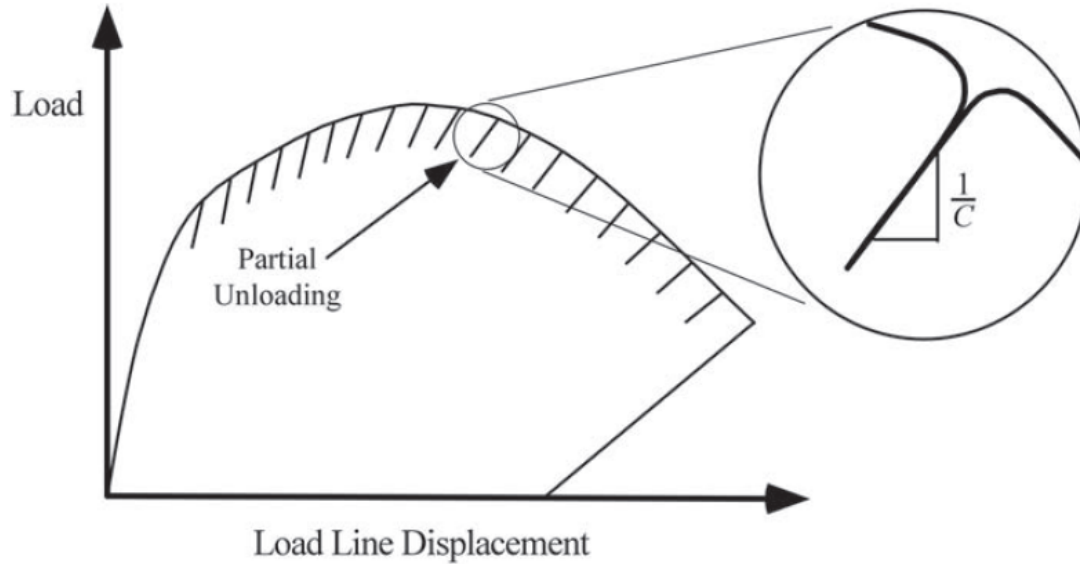
A valid test should respect these conditions:

$$0.45 \leq \frac{a}{W} \leq 0.55$$

$$P_{max} \leq 1.10P_Q$$

$$a, (W - a), B \geq 2.5 \left( \frac{K_{Ic}}{\sigma_Y} \right)^2$$

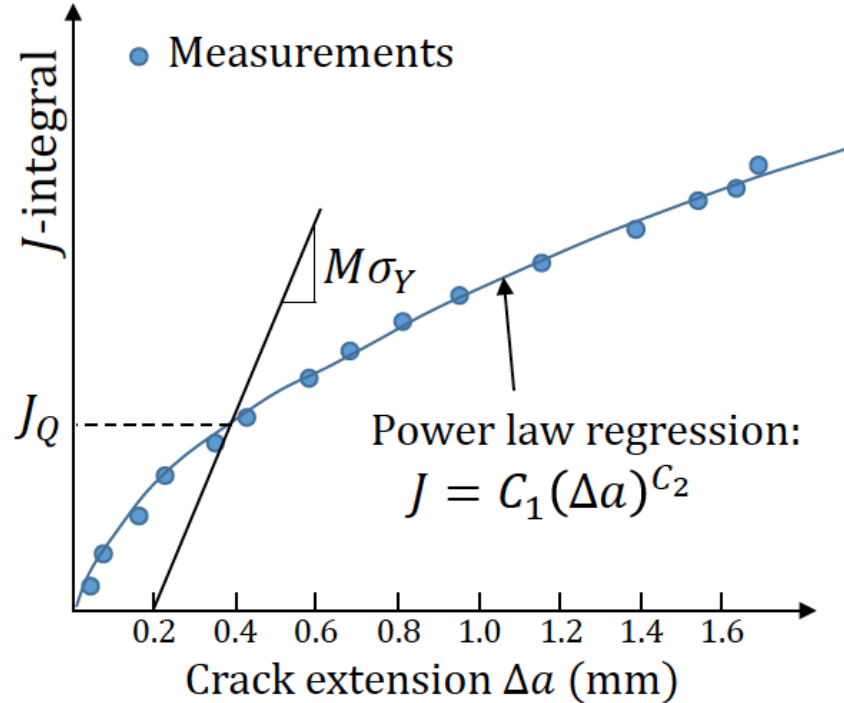
# Method 2: R-curve



Use the compliance  $C$  to calculate the crack length  $a$  during the test.

Calculate  $J$  as a function of  $a$  (for each partial unloading).

# Method 2: R-curve



The value  $J_Q = J_{Ic}$  if:

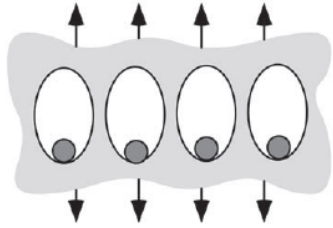
$$B, b_0 \geq \frac{25J_Q}{\sigma_Y}$$

If this is satisfied, you can get:

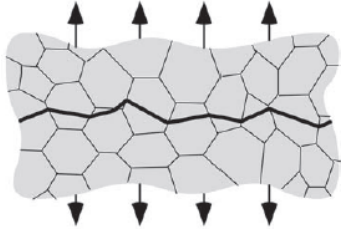
$$K_{Ic} = \sqrt{\frac{EJ_{Ic}}{1 - \nu^2}}$$

# Fracture mechanics

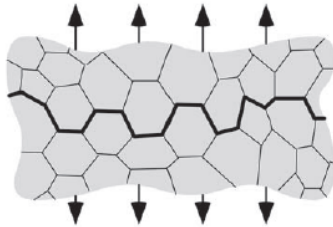
# Fracture mechanisms in metals



1. Ductile fracture,

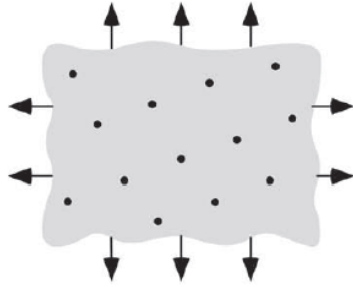


2. Cleavage,

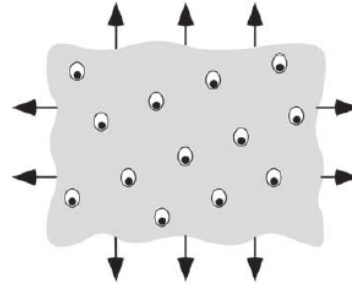


3. Intergranular fracture

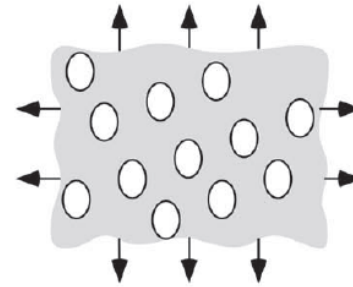
# Ductile fracture



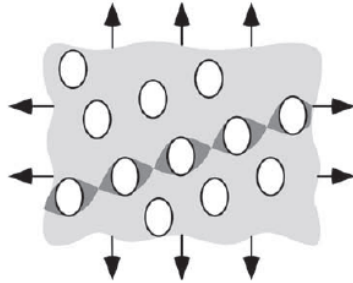
1. Inclusions in a ductile matrix



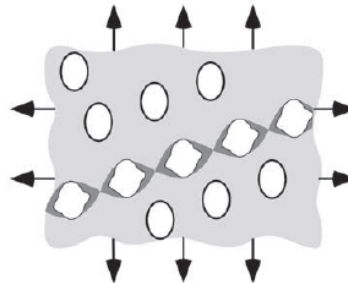
2. Void nucleation



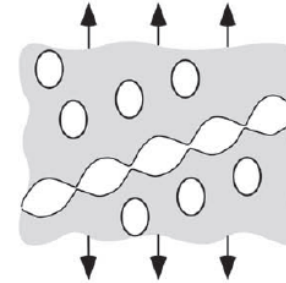
3. Void growth



4. Strain localisation between voids



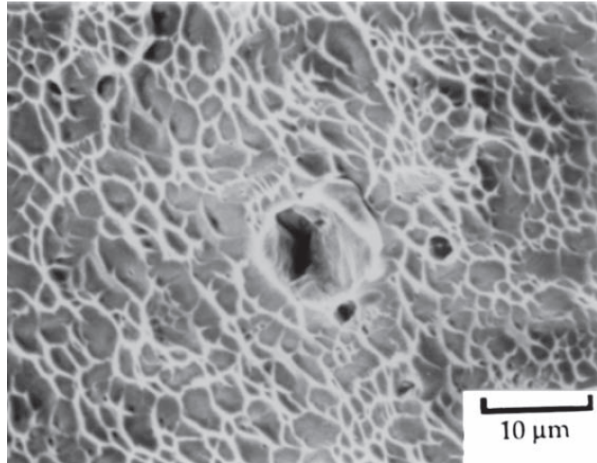
5. Necking between voids



6. Void coalescence and fracture

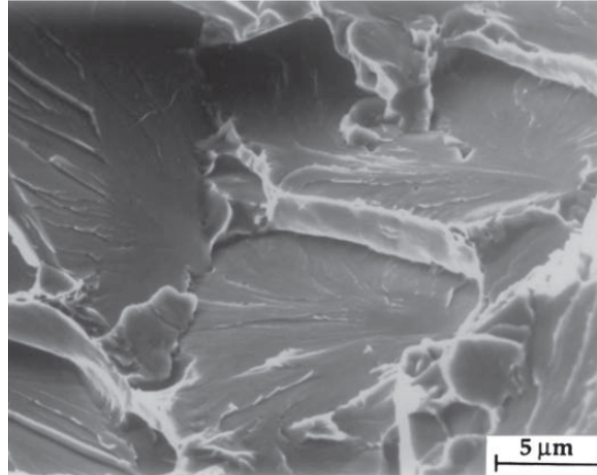
# Fractography

**Ductile fracture**



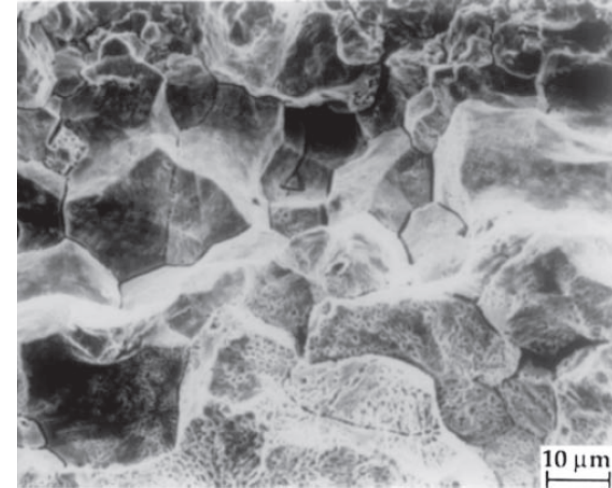
Most metals at room temperature

**Cleavage**



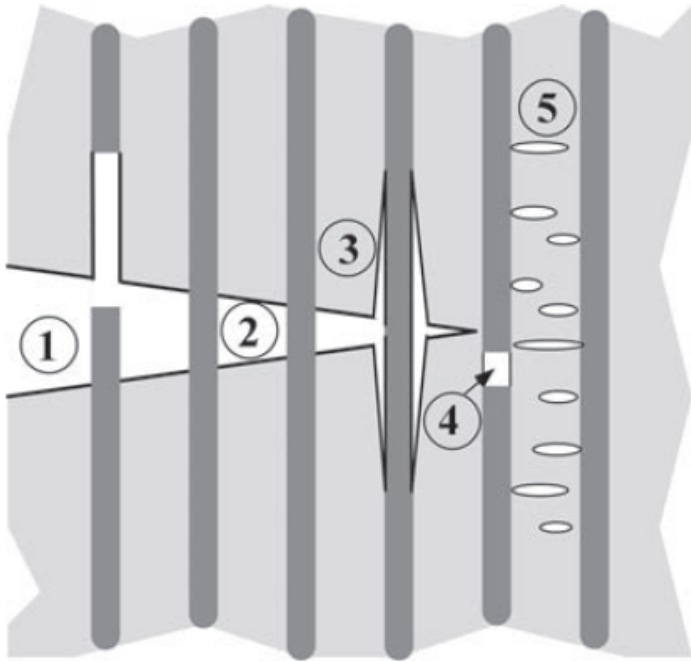
Metals at low temperatures

**Intergranular fracture**



Metals in harsh environments

# Fracture mechanisms in composites



1. Fibre pull-out,
2. Fibre bridging,
3. Fibre/matrix debonding,
4. Fibre failure,
5. Matrix cracking.



# In summary

We covered:

- How to use the  $J$ -integral, and how it is implemented in FEM.
- The procedure to measure fracture toughness,
- What are the main fracture mechanisms.