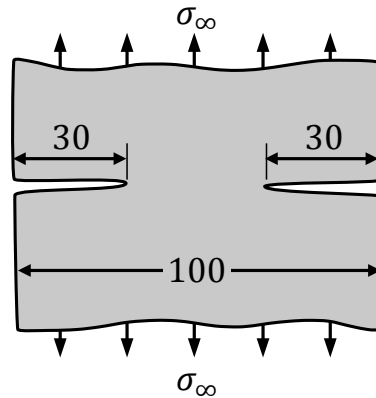


A? Question

You are responsible of designing a new product made from a ductile polymer. The material testing division of your company provided you with a tensile test and a fracture toughness test, conducted on a thin plate with the geometry below. They found that $E = 3 \text{ GPa}$, $\sigma_Y = 40 \text{ MPa}$, and $K_{Ic} = 15 \text{ MPa}\sqrt{\text{m}}$. Can you trust this value of K_{Ic} ?



All dimensions in mm

A! Solution

The plastic zone size under plane stress is given by:

$$d_p = \frac{1}{\pi} \left(\frac{K_{Ic}}{\sigma_Y} \right)^2 = \frac{1}{\pi} \left(\frac{15}{40} \right)^2 = 44.7 \text{ mm}$$

This is larger than the crack length $a = 30 \text{ mm}$ and the unbroken ligament (40 mm). Therefore, the plastic zone size is too large to use LEFM. The consequence is that you shouldn't trust this value of K_{Ic} , the J-integral should be used instead. (It would be possible to use LEFM if $d_p \leq a/10$).

A? Question

Demonstrate that the shape of the plastic zone in mode III is a circle of radius:

$$r_y = \frac{3}{2\pi} \left(\frac{K_{III}}{\sigma_Y} \right)^2,$$

centered at the crack tip. (Hint: search for an expression of the von Mises yielding criterion that includes all 6 components of the stress tensor, instead of only the 3 principal stresses).

A! Solution

The stress field at the crack tip in mode III is (taken from the datasheet):

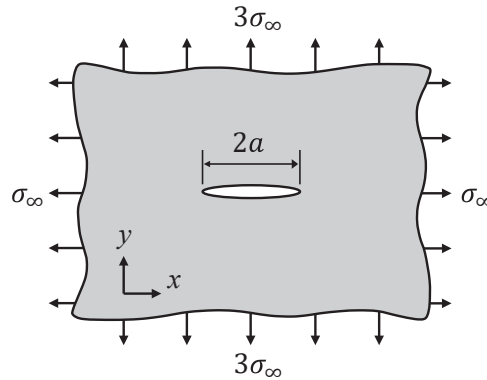
$$\sigma_{xz} = -\frac{K_{III}}{\sqrt{2\pi r}} \sin \frac{\theta}{2} \quad \sigma_{yz} = \frac{K_{III}}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \quad \sigma_{xx} = \sigma_{yy} = \sigma_{zz} = \sigma_{xy} = 0$$

Substituting this stress field in the von Mises criterion gives:

$$\begin{aligned} \frac{1}{\sqrt{2}} [(\sigma_{xx} - \sigma_{yy})^2 + (\sigma_{yy} - \sigma_{zz})^2 + (\sigma_{zz} - \sigma_{xx})^2 + 6(\sigma_{xy}^2 + \sigma_{xz}^2 + \sigma_{yz}^2)]^{1/2} &= \sigma_Y \\ \Rightarrow \frac{1}{\sqrt{2}} \left[\frac{6K_{III}^2}{2\pi r} \left(\sin^2 \frac{\theta}{2} + \cos^2 \frac{\theta}{2} \right) \right]^{1/2} &= \sigma_Y \\ \Rightarrow \frac{1}{\sqrt{2}} \left[\frac{6K_{III}^2}{2\pi r} \right]^{1/2} &= \sigma_Y \\ \Rightarrow \frac{6K_{III}^2}{2\pi r} &= 2\sigma_Y^2 \\ \Rightarrow r &= \frac{3}{2\pi} \left(\frac{K_{III}}{\sigma_Y} \right)^2 \end{aligned}$$

A? Question

A thin aluminium plate, with $\sigma_{ys} = 320 \text{ MPa}$ and $K_{Ic} = 30 \text{ MPa}\sqrt{\text{m}}$, is subjected to bi-axial loading as shown below. Plot the maximum allowable stress σ_∞ , as a function of a , to avoid both yielding and fracture.

**A! Solution**

The global stress field for this plate is:

$$\sigma_{xx} = \sigma_\infty, \quad \sigma_{yy} = 3\sigma_\infty, \quad \sigma_{zz} = \sigma_{xy} = \sigma_{xz} = \sigma_{yz} = 0.$$

Substituting this stress field in the von Mises criterion gives us the load that will cause yielding:

$$\begin{aligned} \frac{1}{\sqrt{2}} [(\sigma_{xx} - \sigma_{yy})^2 + (\sigma_{yy} - \sigma_{zz})^2 + (\sigma_{zz} - \sigma_{xx})^2 + 6(\sigma_{xy}^2 + \sigma_{xz}^2 + \sigma_{yz}^2)]^{1/2} &= \sigma_{ys} \\ \Rightarrow \frac{1}{\sqrt{2}} [4\sigma_\infty^2 + 9\sigma_\infty^2 + \sigma_\infty^2]^{1/2} &= \sigma_{ys} \\ \Rightarrow \sigma_\infty = \frac{\sigma_{ys}}{\sqrt{7}} &\quad \text{yielding} \end{aligned}$$

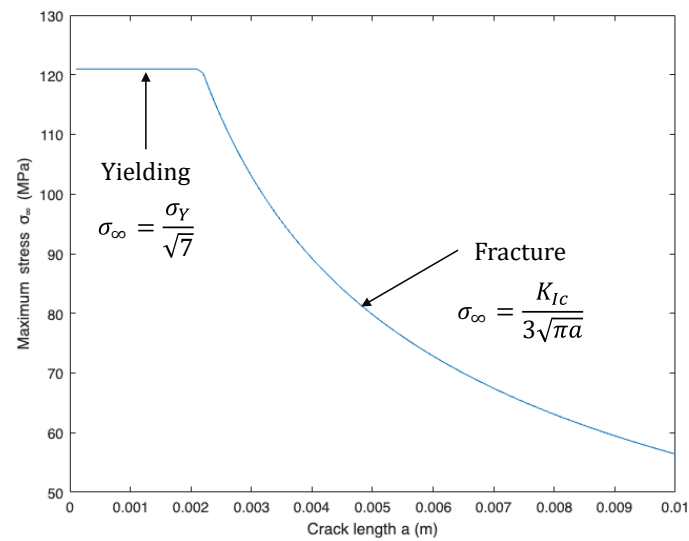
Otherwise, fracture will occur when $K_I = K_{Ic}$, which gives:

$$\begin{aligned} 3\sigma_\infty\sqrt{\pi a} &= K_{Ic} \\ \Rightarrow \sigma_\infty &= \frac{K_{Ic}}{3\sqrt{\pi a}} \quad \text{fracture} \end{aligned}$$

The maximum allowable stress σ_∞ to avoid both yielding and fracture is therefore:

$$\sigma_\infty = \min \left(\frac{\sigma_{ys}}{\sqrt{7}}; \frac{K_{Ic}}{3\sqrt{\pi a}} \right)$$

This is plotted below as a function of the crack length a .



A? Question

Is the transition flaw size a material property?

A! Solution

The transition flaw size is not a material property. The example above shows that the transition flaw size depends on the geometry of the structure. A material property, such as the yield strength or fracture toughness, is independent of geometry.