

# **Example 5: Isotropic Hardening Plasticity**

## **Governing Equations**

• Elasticity:

$$\sigma_{ij} = \lambda \delta_{ij} \varepsilon_{kk}^{el} + 2\mu \varepsilon_{ij}^{el},$$

or in a Jaumann (corotational) rate form:

$$\dot{\sigma}_{ij}^{J} = \lambda \delta_{ij} \dot{\varepsilon}_{kk}^{el} + 2\mu \dot{\varepsilon}_{ij}^{el}.$$

The Jaumann rate equation is integrated in a corotational framework:

$$\Delta \sigma_{ij}^{J} = \lambda \delta_{ij} \Delta \varepsilon_{kk}^{el} + 2\mu \Delta \varepsilon_{ij}^{el}.$$

- Plasticity:
  - Yield function:

$$\sqrt{\frac{3}{2}S_{ij}S_{ij}} - \sigma_y(\bar{\varepsilon}^{pl}) = 0, \qquad S_{ij} = \sigma_{ij} - \frac{1}{3}\delta_{ij}\sigma_{kk}.$$

- Equivalent plastic strain:

$$\bar{\mathbf{\varepsilon}}^{pl} = \int_{0}^{t} \dot{\bar{\mathbf{\varepsilon}}}^{pl} dt, \qquad \dot{\bar{\mathbf{\varepsilon}}}^{pl} = \sqrt{\frac{2}{3}} \dot{\mathbf{\varepsilon}}_{ij}^{pl} \dot{\mathbf{\varepsilon}}_{ij}^{pl}.$$

– Plastic flow law:

$$\dot{\varepsilon}_{ij}^{pl} = \frac{3}{2} \frac{S_{ij}}{\sigma_y} \dot{\bar{\varepsilon}}^{pl}.$$



### **Integration Procedure**

• We first calculate the von Mises stress based on purely elastic behavior (elastic predictor):

$$\bar{\sigma}^{pr} = \sqrt{\frac{3}{2}S_{ij}^{pr}S_{ij}^{pr}}, \qquad S_{ij}^{pr} = S_{ij}^o + 2\mu\Delta e_{ij}.$$

- If the elastic predictor is larger than the current yield stress, plastic flow occurs. The backward Euler method is used to integrate the equations.
  - After some manipulation we can reduce the problem to a single equation in terms of the incremental equivalent plastic strain:

$$\bar{\sigma}^{pr} - 3\mu\Delta\bar{\varepsilon}^{pl} = \sigma_{v}(\bar{\varepsilon}^{pl}).$$

- This equation is solved with Newton's method.



• After the equation is solved, the following update equations for the stress and the plastic strain can be used:

$$\sigma_{ij} = \eta_{ij}\sigma_y + \frac{1}{3}\delta_{ij}\sigma_{kk}^{pr}, \qquad \Delta \varepsilon_{ij}^{pl} = \frac{3}{2}\eta_{ij}\Delta \bar{\varepsilon}^{pl}$$

$$\eta_{ij} = S_{ij}^{pr}/\bar{\sigma}^{pr}.$$

• In addition, you can readily obtain the consistent Jacobian:

$$\Delta \dot{\sigma}_{ij} = \lambda^* \delta_{ij} \Delta \dot{\epsilon}_{kk} + 2\mu^* \Delta \dot{\epsilon}_{ij} + \left(\frac{h}{1 + h/3\mu} - 3\mu^*\right) \eta_{ij} \eta_{kl} \Delta \dot{\epsilon}_{kl}$$
$$\mu^* = \mu \sigma_y / \bar{\sigma}^{pr}, \quad \lambda^* = k - \frac{2}{3}\mu^*, \quad h = d\sigma_y / d\bar{\epsilon}^{pl}.$$

 A detailed discussion about the isotropic plasticity integration algorithm can be found in Section 4.2.2 of the ABAQUS Theory Manual.

The appropriate coding is shown on the following pages.



#### **Coding for Isotropic Mises Plasticity**

```
LOCAL ARRAYS
C
    EELAS - ELASTIC STRAINS
    EPLAS - PLASTIC STRAINS
C
C
    FLOW
            - DIRECTION OF PLASTIC FLOW
     DIMENSION EELAS(6), EPLAS(6), FLOW(6), HARD(3)
C
      PARAMETER (ZERO=0.D0, ONE=1.D0, TWO=2.D0, THREE=3.D0, SIX=6.D0,
                ENUMAX=.4999D0, NEWTON=10, TOLER=1.0D-6)
C
C
    UMAT FOR ISOTROPIC ELASTICITY AND ISOTROPIC MISES PLASTICITY
    CANNOT BE USED FOR PLANE STRESS
C
    PROPS(1) - E
C
    PROPS(2) - NU
C
    PROPS (3..) - SYIELD AN HARDENING DATA
C
    CALLS UHARD FOR CURVE OF YIELD STRESS VS. PLASTIC STRAIN
C
```



```
C
C
     ELASTIC PROPERTIES
C
      EMOD=PROPS (1)
      ENU=MIN(PROPS(2), ENUMAX)
      EBULK3=EMOD/(ONE-TWO*ENU)
      EG2=EMOD/(ONE+ENU)
      EG=EG2/TWO
      EG3=THREE*EG
      ELAM= (EBULK3-EG2) / THREE
C
C
     ELASTIC STIFFNESS
C
      DO K1=1, NDI
        DO K2=1, NDI
          DDSDDE(K2, K1)=ELAM
        END DO
        DDSDDE(K1, K1)=EG2+ELAM
      END DO
      DO K1=NDI+1, NTENS
        DDSDDE(K1, K1)=EG
      END DO
```



```
RECOVER ELASTIC AND PLASTIC STRAINS AND ROTATE FORWARD
C
     ALSO RECOVER EQUIVALENT PLASTIC STRAIN
C
C
                                1), DROT, EELAS, 2, NDI, NSHR)
      CALL ROTSIG(STATEV(
      CALL ROTSIG(STATEV(NTENS+1), DROT, EPLAS, 2, NDI, NSHR)
      EQPLAS=STATEV (1+2*NTENS)
C
C
     CALCULATE PREDICTOR STRESS AND ELASTIC STRAIN
C
      DO K1=1, NTENS
        DO K2=1, NTENS
          STRESS(K2) = STRESS(K2) + DDSDDE(K2, K1) * DSTRAN(K1)
        END DO
        EELAS (K1) = EELAS (K1) + DSTRAN (K1)
      END DO
C
C
     CALCULATE EQUIVALENT VON MISES STRESS
C
      SMISES=(STRESS(1)-STRESS(2))**2+(STRESS(2)-STRESS(3))**2
     1
                                      + (STRESS(3)-STRESS(1)) **2
      DO K1=NDI+1,NTENS
        SMISES=SMISES+SIX*STRESS(K1)**2
      END DO
      SMISES=SQRT(SMISES/TWO)
```



```
C
C
     GET YIELD STRESS FROM THE SPECIFIED HARDENING CURVE
C
      NVALUE=NPROPS/2-1
      CALL UHARD (SYIELO, HARD, EQPLAS, EQPLASRT, TIME, DTIME, TEMP,
           DTEMP, NOEL, NPT, LAYER, KSPT, KSTEP, KINC, CMNAME, NSTATV,
     1
           STATEV, NUMFIELDV, PREDEF, DPRED, NVALUE, PROPS (3))
C
     DETERMINE IF ACTIVELY YIELDING
C
C
      IF (SMISES.GT.(ONE+TOLER)*SYIELO) THEN
C
C
       ACTIVELY YIELDING
C
       SEPARATE THE HYDROSTATIC FROM THE DEVIATORIC STRESS
C
       CALCULATE THE FLOW DIRECTION
C
        SHYDRO= (STRESS(1)+STRESS(2)+STRESS(3))/THREE
        DO K1=1,NDI
          FLOW(K1) = (STRESS(K1) - SHYDRO) / SMISES
        END DO
        DO K1=NDI+1, NTENS
          FLOW(K1) = STRESS(K1) / SMISES
        END DO
```



```
C
C
       SOLVE FOR EQUIVALENT VON MISES STRESS
C
       AND EQUIVALENT PLASTIC STRAIN INCREMENT USING NEWTON ITERATION
C
        SYIELD=SYIEL0
        DEQPL=ZERO
        DO KEWTON=1, NEWTON
          RHS=SMISES-EG3*DEQPL-SYIELD
          DEQPL=DEQPL+RHS/(EG3+HARD(1))
           CALL UHARD (SYIELD, HARD, EQPLAS+DEQPL, EQPLASRT, TIME, DTIME, TEMP,
           DTEMP, NOEL, NPT, LAYER, KSPT, KSTEP, KINC, CMNAME, NSTATV,
     1
           STATEV, NUMFIELDV, PREDEF, DPRED, NVALUE, PROPS (3))
          IF(ABS(RHS).LT.TOLER*SYIEL0) GOTO 10
        END DO
C
       WRITE WARNING MESSAGE TO THE .MSG FILE
C
C
        WRITE(7,2) NEWTON
          FORMAT(//,30X,'***WARNING - PLASTICITY ALGORITHM DID NOT ',
    2
                         'CONVERGE AFTER ', 13,' ITERATIONS')
     1
        CONTINUE
   10
```



```
C
C
       UPDATE STRESS, ELASTIC AND PLASTIC STRAINS AND
C
       EQUIVALENT PLASTIC STRAIN
C
        DO K1=1, NDI
           STRESS(K1)=FLOW(K1)*SYIELD+SHYDRO
          EPLAS(K1) = EPLAS(K1) + THREE / TWO*FLOW(K1) * DEQPL
          EELAS (K1) = EELAS (K1) - THREE/TWO*FLOW(K1) *DEQPL
        END DO
        DO K1=NDI+1, NTENS
          STRESS(K1)=FLOW(K1)*SYIELD
          EPLAS(K1) = EPLAS(K1) + THREE * FLOW(K1) * DEQPL
          EELAS (K1) = EELAS (K1) - THREE * FLOW (K1) * DEQPL
        END DO
        EQPLAS=EQPLAS+DEQPL
C
C
       CALCULATE PLASTIC DISSIPATION
C
        SPD=DEQPL*(SYIELO+SYIELD)/TWO
```



```
C
C
       FORMULATE THE JACOBIAN (MATERIAL TANGENT)
C
       FIRST CALCULATE EFFECTIVE MODULI
C
        EFFG=EG*SYIELD/SMISES
        EFFG2=TWO*EFFG
        EFFG3=THREE/TWO*EFFG2
        EFFLAM=(EBULK3-EFFG2)/THREE
        EFFHRD=EG3*HARD(1)/(EG3+HARD(1))-EFFG3
        DO K1=1, NDI
          DO K2=1, NDI
            DDSDDE(K2, K1)=EFFLAM
          END DO
          DDSDDE(K1, K1) = EFFG2 + EFFLAM
        END DO
        DO K1=NDI+1, NTENS
          DDSDDE(K1, K1) = EFFG
        END DO
        DO K1=1, NTENS
          DO K2=1, NTENS
            DDSDDE(K2, K1) = DDSDDE(K2, K1) + EFFHRD*FLOW(K2) *FLOW(K1)
          END DO
        END DO
      ENDIF
```



```
C
C
     STORE ELASTIC AND (EQUIVALENT) PLASTIC STRAINS
C
     IN STATE VARIABLE ARRAY
C
      DO K1=1, NTENS
        STATEV (K1) = EELAS (K1)
        STATEV (K1+NTENS) = EPLAS (K1)
      END DO
      STATEV (1+2*NTENS) = EQPLAS
C
      RETURN
      END
      SUBROUTINE UHARD (SYIELD, HARD, EQPLAS, EQPLASRT, TIME, DTIME, TEMP,
           DTEMP, NOEL, NPT, LAYER, KSPT, KSTEP, KINC,
     1
            CMNAME, NSTATV, STATEV, NUMFIELDV,
            PREDEF, DPRED, NVALUE, TABLE)
     3
      INCLUDE 'ABA PARAM.INC'
      CHARACTER*80 CMNAME
      DIMENSION HARD(3), STATEV(NSTATV), TIME(*),
                 PREDEF (NUMFIELDV), DPRED(*)
     1
```



```
C
      DIMENSION TABLE (2, NVALUE)
C
      PARAMETER (ZERO=0.D0)
C
     SET YIELD STRESS TO LAST VALUE OF TABLE, HARDENING TO ZERO
C
C
      SYIELD=TABLE(1, NVALUE)
      HARD(1) = ZERO
     IF MORE THAN ONE ENTRY, SEARCH TABLE
C
C
      IF (NVALUE.GT.1) THEN
        DO K1=1, NVALUE-1
          EQPL1=TABLE(2,K1+1)
          IF (EQPLAS.LT.EQPL1) THEN
            EQPL0=TABLE(2, K1)
            IF (EQPL1.LE.EQPL0) THEN
              WRITE(7, 1)
              FORMAT(//, 30X, '***ERROR - PLASTIC STRAIN MUST BE ',
    1
                               'ENTERED IN ASCENDING ORDER')
     1
              CALL XIT
            ENDIF
```



```
C
C
           CURRENT YIELD STRESS AND HARDENING
C
            DEQPL=EQPL1-EQPL0
            SYIEL0=TABLE(1, K1)
            SYIEL1=TABLE(1, K1+1)
            DSYIEL=SYIEL1-SYIEL0
            HARD(1) = DSYIEL/DEQPL
            SYIELD=SYIEL0+(EQPLAS-EQPL0)*HARD(1)
            GOTO 10
          ENDIF
        END DO
   10
        CONTINUE
      ENDIF
      RETURN
      END
```



#### Remarks

- This **UMAT** yields exactly the same results as the \*PLASTIC option with ISOTROPIC hardening.
  - This result is also true for large-strain calculations. The necessary rotations of stress and strain are taken care of by ABAQUS.
  - The rotation of elastic and plastic strain, prior to integration, is accomplished by the calls to **ROTSIG**.



- The routine calls user subroutine **UHARD** to recover a piecewise linear hardening curve.
  - It is straightforward to replace the piecewise linear curve by an analytic description.
  - A local Newton iteration is used to determine the current yield stress and hardening modulus.
  - If the data are not given in ascending order of strain, the routine
     xIT is called, which closes all files and terminates execution.