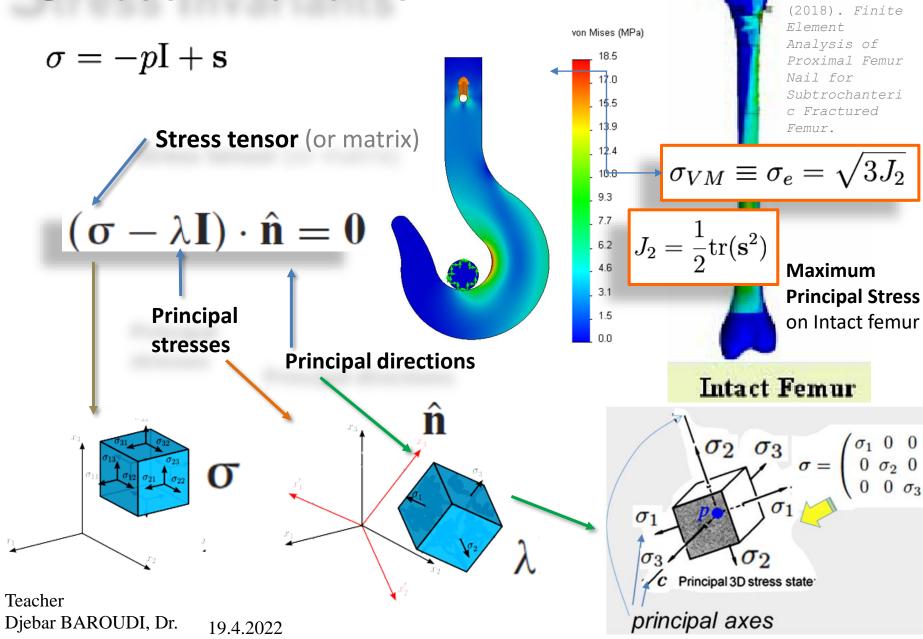
# Stress Invariants



Ref.

Srinivasan,

n et al.

Sowmianarayana

## Elements of solid mechanics – Stress Invariants

Limited to the necessary minimum to follow this course

We summarize necessary basic concepts for solid mechanics

- We use primarily uses Cartesian coordinates
- In parallel also the tensor notation is used; this makes it easy to read: vectors and tensors are represented either by their components or by their symbols

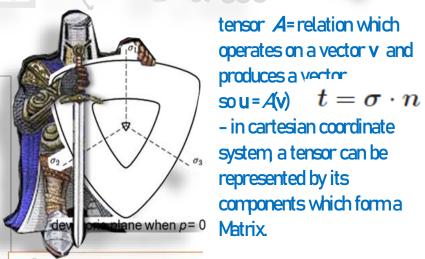
Ex.) Drucker-Prager yield criterion

$$\sqrt{J_2} = A + B I_1$$

Ottosen criterion

$$A\frac{J_2}{\sigma_c} + \Lambda\sqrt{J_2} + BI_1 - \sigma_c = 0,$$

What are these symbols J2 and I1 standing for?



#### failure criterion

(concrete)

**Readings**:

Chapter 1 from: D. Gross and T. Seelig, Fracture Mechanics: With an Introduction to Micromechanics.

or

Reddy; Chapters 3 (kinematics of continua) and 4 (stress measures)

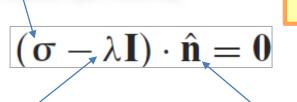
... or read from any other source ...
https://en.wikipedia.org/wiki/Cauchy stress tensor

 $A\frac{J_2}{I} + \Lambda\sqrt{J_2} + BI_1 - \sigma_c = 0,$ 

# Principal Stresses and Principal Planes Why?

Stress tensor (or matrix)

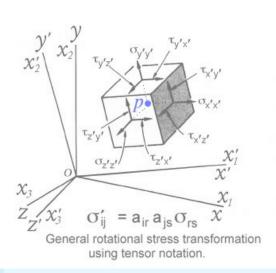
- Determination of maximum normal and shear stresses at a point is of considerable interest
- Many failure criteria are expressed using stress/strain Invariants (= scalars)
- The general laws of physics are <u>independent</u> coordinate systems So, they have <u>Invariance</u> & have symmetry properties

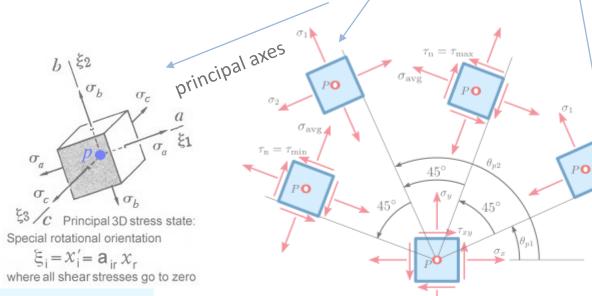


principal stresses

principal directions

The principal axes: solely normal stresses and no shear stresses appear in sections perpendicular to these axes





Example: the speed of light is an invariant

#### **Invariants**

#### Mechanics

$$\mathbf{A} = \boldsymbol{\sigma}$$

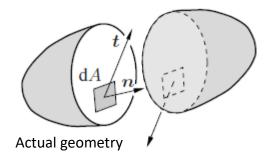
• 2<sup>nd</sup>-order symmetric tensors (matrices) have always three groups independent invariants which, for our purposes, in mechanics, are defined as

$$I_1^A = \operatorname{tr} \mathbf{A},$$
  $A = \boldsymbol{\sigma}$   $I_2^A = \frac{1}{2} \operatorname{tr} (\mathbf{A}^{\operatorname{d}})^2,$  The stress deviator  $\mathbf{S} = \boldsymbol{\sigma}^{\operatorname{d}}$   $\mathbf{S} = \mathbf{A}^{\operatorname{d}} = \boldsymbol{\sigma}^{\operatorname{d}} \equiv \mathbf{S}$   $\mathbf{T} = (\mathbf{S}^{\operatorname{d}})^2,$   $\mathbf{T} = (\mathbf{S}^{\operatorname{d}})^2,$ 

• In plasticity, one needs to compute gradients of functions of invariants with respect to the stress components, for instance, the gradient for an arbitrary isotropic yield function (Isotropic plasticity theory) is as

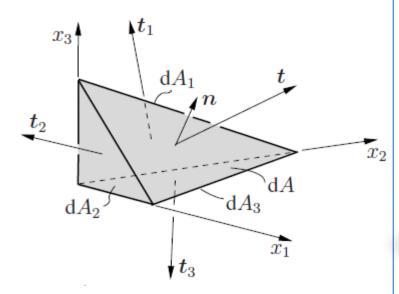
$$\frac{\partial f(I_1, J_2, J_3)}{\partial \boldsymbol{\sigma}} = \frac{\partial f}{\partial I_1} \mathbf{I} + \frac{\partial f}{\partial J_2} \mathbf{S} + \frac{\partial f}{\partial J_3} \mathbf{T} \quad \text{, where } \quad \text{Hill tensor } \mathbf{T} = (\mathbf{S}^2)^{\frac{1}{4}}$$

## **Stress**

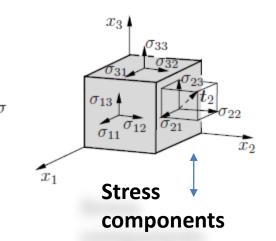


#### **Stress vector:**

$$t = \lim_{\Delta A \to 0} \frac{\Delta F}{\Delta A} = \frac{\mathrm{d}F}{\mathrm{d}A}$$



Stress state - Cauchy's tetrahedron



$$m{\sigma} = egin{pmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \ \sigma_{21} & \sigma_{22} & \sigma_{23} \ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{pmatrix}$$

stress components in Cartesian rectangular coordinates

**Cauchy**'s stress tensor (2nd order symm. tensor)



#### Cauchy's stress theorem:

$$t = \sigma \cdot n$$

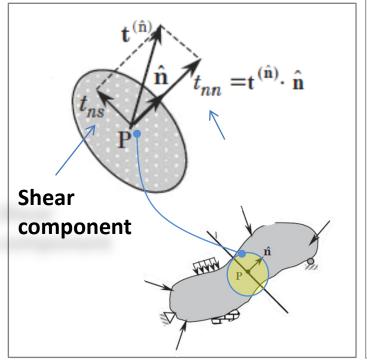
$$\vec{\mathbf{t}}(\vec{x}, \vec{n}, t) = \vec{n} \cdot \mathbf{\sigma}(\vec{x}, t) \leftrightarrow \mathbf{t} = \mathbf{\sigma}^{\mathrm{T}} \mathbf{n}$$

#### Shear- and normal components of the stress vector

## Cauchy's stress theorem:

$$\mathbf{t}(\hat{\mathbf{n}}) = \hat{\mathbf{n}} \cdot \mathbf{\sigma} \\ = \mathbf{\sigma}^{\mathrm{T}} \cdot \hat{\mathbf{n}}$$

$$\underline{t}(\underline{\mathbf{n}}) = \underline{\underline{\sigma}}^{\mathrm{T}} \underline{\mathbf{n}} \longleftrightarrow \mathbf{t}_{i} = \sigma_{ji} n_{j}$$

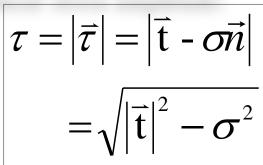


$$|\vec{\mathsf{t}}(P,\vec{n}) = \sigma \vec{n} + \vec{\tau}|$$

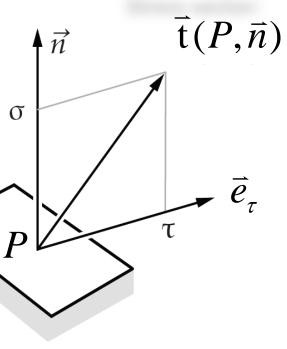
#### **Normal stress:**

$$\sigma = \vec{t} \cdot \vec{n}$$

#### **Shear stress amplitude:**



#### **Stress vector:**



#### Mohr's stress representation in plane - $(\tau, \sigma)$

$$x^2 + y^2 = R^2$$

$$\left|\tau^2 + \sigma^2 = \left|\vec{t}(P, \vec{n})\right|^2 \quad \left||\vec{n}|| = 1\right|$$

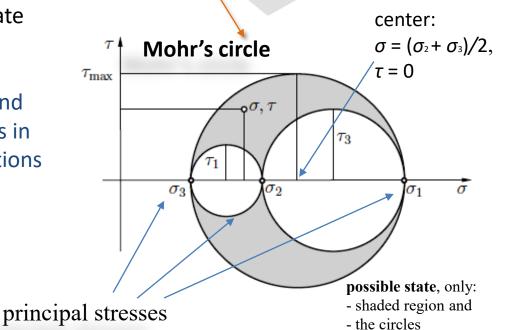
$$\|\vec{n}\| = 1$$

## Mohr's circle – the equation for points $(\tau, \sigma)$

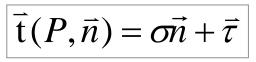
with the material point **P** is fixed and the directions of the plan-section across **P** vary through the outer normal **n** director-cosines.

- a graphical visualization of the stress state by *Mohr's circles* (1835-1918)
- a representation of **normal stresses**  $\sigma$  and corresponding **shear stresses τ** as points in a  $\sigma$ - $\tau$ --diagram for all possible cross sections through the material point P and for all directions

$$|\underline{\mathbf{t}}(\underline{\mathbf{n}}) = \underline{\underline{\mathbf{g}}}^{\mathrm{T}} \underline{\mathbf{n}} \longleftrightarrow \mathbf{t}_{i} = \sigma_{ji} n_{j}$$



 $\sigma = \vec{t} \cdot \vec{n}$ 



 $\overline{\mathsf{t}}(P, \vec{n})$ 

#### **Principal Stresses and Principal Planes**

Why?



 Determination of maximum normal stresses and shear stresses at a point is of considerable interest

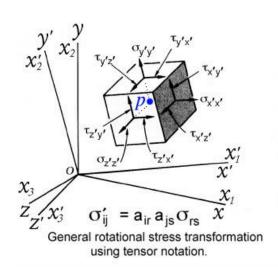
Stress tensor (or matrix)

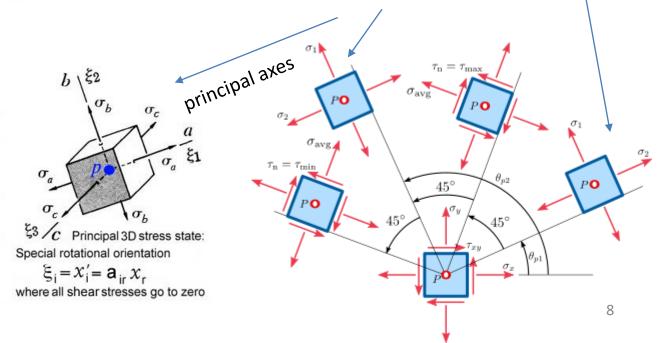


principal stresses principal directions

Many failure criteria are expressed using stress/strain Invariants

The principal axes: solely normal stresses and no shear stresses appear in sections perpendicular to these axes





## **Principal Stresses and Principal Planes**

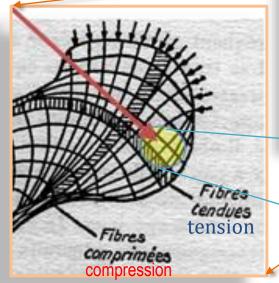
For **any arbitrary state of stress**, we can find a **set of** orthogonal planes on which only normal stresses act and the shearing stresses are zero

- Called Principal Planes and the normal stresses acting on these planes are Principal Stresses denoted as
- They are ordered such that :  $|\sigma_1 > \sigma_2 > \sigma_3|$

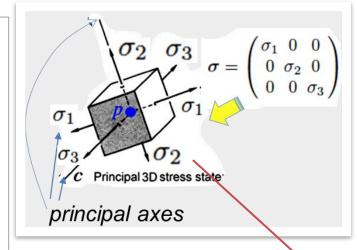
$$|\sigma_1 > \sigma_2 > \sigma_3|$$

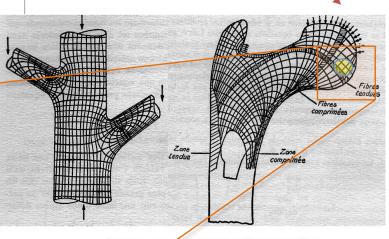
**Maximum shear stresses** are in planes forming an angle of 45 deg. with Principal Planes

$$\tau_{\text{max}} = \frac{\sigma_1 - \sigma_3}{2}$$

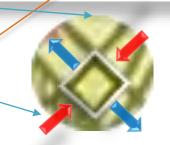


The same holds for strains





Isostatiques des neuds du bois et d'une tête de fémur

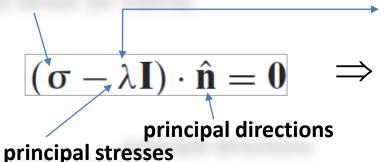


## **Stress Invariants 1(2)**

#### **Invariants**

# $egin{align} &\mathrm{I}_A = \mathrm{tr}(\mathbf{A}) \ &\mathrm{II}_A = rac{1}{2} \left( (\mathrm{tr}\mathbf{A})^2 - \mathrm{tr}(\mathbf{A}\mathbf{A}) ight) \ &\mathrm{III}_A = \det(\mathbf{A}) \ \end{aligned}$

Stress tensor (or matrix)



$$\Rightarrow \sigma^3 - I_\sigma \sigma^2 - III_\sigma \sigma - IIII_\sigma = 0$$

Solutions give the principal stresses  $\sigma_1, \sigma_2, \sigma_3$ Each principal stress corresponds to a principal direction

Also denoted as:  $I_1, I_2, I_3$ 

$$I_1 \uparrow \mid I_{\sigma} = \sigma_{ii} = \sigma_{11} + \sigma_{22} + \sigma_{33}$$
,

$$II_{\sigma} = (\sigma_{ij}\sigma_{ij} - \sigma_{ii}\sigma_{jj})/2$$

$$= -(\sigma_{11}\sigma_{22} + \sigma_{22}\sigma_{33} + \sigma_{33}\sigma_{11}) + \sigma_{12}^2 + \sigma_{23}^2 + \sigma_{31}^2,$$

$$I_3$$
  $III_{\sigma} = \det(\sigma_{ij}) = \begin{vmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{vmatrix}$ .

The same formula are valid for strain invariants too

principal axes

## Stress Invariants in terms of principle stresses

$$\begin{array}{c} (\boldsymbol{\sigma} - \lambda \mathbf{I}) \cdot \hat{\mathbf{n}} = \mathbf{0} \\ \text{principal directions} \\ \text{principal stresses} \end{array} \Rightarrow \begin{array}{c} \sigma^3 - I_\sigma \, \sigma^2 - III_\sigma \, \sigma - IIII_\sigma = 0 \\ \text{Solutions give the principal stresses} \\ \text{Each principal stress corresponds to a principal} \end{array}$$

$$\sigma^3 - I_\sigma \,\sigma^2 - II_\sigma \,\sigma - III_\sigma = 0$$

Solutions give the principal stresses  $\sigma_1, \sigma_2, \sigma_3$ Each principal stress corresponds to a principal direction

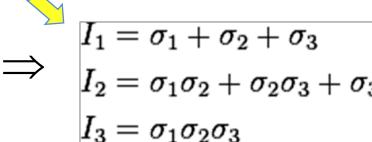
Also denoted as: 
$$I_1, I_2, I_3$$

$$I_1 \equiv \operatorname{tr}(\mathbf{\sigma}) = \sigma_1 + \sigma_2 + \sigma_3$$

$$I_2 \equiv II_{\sigma} = \frac{1}{2} \left[ \operatorname{tr}(\sigma)^2 - \operatorname{tr}(\sigma^2) \right] \implies I_1 = \sigma_1 + \sigma_2 + \sigma_3$$

$$I_2 = \sigma_1 \sigma_2 + \sigma_2 \sigma_3 + \sigma_3 \sigma_1$$

$$\sigma = \begin{pmatrix} \sigma_1 & 0 & 0 \\ 0 & \sigma_2 & 0 \\ 0 & 0 & \sigma_3 \end{pmatrix}^{\square}$$



$$I_3 \equiv III_{\boldsymbol{\sigma}} = \det(\boldsymbol{\sigma})$$

Most tensors used in engineering are symmetric 3×3. For this case the invariants can be calculated as:

$$egin{aligned} \mathbf{I}_A &= \mathbf{tr}(\mathbf{A}) \ &\mathbf{II}_A &= rac{1}{2} \left( (\mathbf{tr}\mathbf{A})^2 - \mathbf{tr}(\mathbf{A}\mathbf{A}) 
ight) \ &\mathbf{III}_A &= \det(\mathbf{A}) \end{aligned}$$

## **Deviatoric stress**

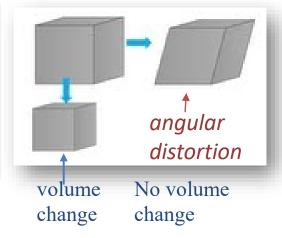
#### Jännitysdeviaattori

$$\begin{bmatrix} \sigma_{11} & \sigma_{12} & \sigma_{13} \\ \sigma_{21} & \sigma_{22} & \sigma_{23} \\ \sigma_{31} & \sigma_{32} & \sigma_{33} \end{bmatrix} = \begin{bmatrix} \sigma_m & 0 & 0 \\ 0 & \sigma_m & 0 \\ 0 & 0 & \sigma_m \end{bmatrix} + \begin{bmatrix} s_{11} & s_{12} & s_{13} \\ s_{21} & s_{22} & s_{23} \\ s_{31} & s_{32} & s_{33} \end{bmatrix}$$

• additive decomposition of stress tensor – very useful

$$\sigma_{ij} = \frac{\sigma_{kk}}{3} \, \delta_{ij} + s_{ij}$$
 or  $\sigma = \sigma_m \, I + s$ .

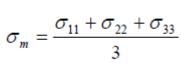
Why split ting Volumetric and Ceviatoric (shearing)?



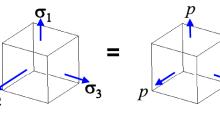
Mean normal stress  $-p \equiv \sigma_m$  or also called *hydrostatic stress state* 

**Deviatoric stress** S characterizes deviation of the stress state from a

hydrostatic state



Expressed in principle stresses:



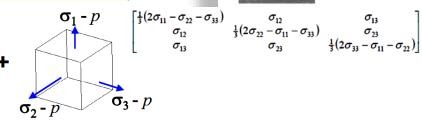
$$\sigma_{2} \qquad \sigma_{3} \qquad p \qquad p$$

$$\sigma_{1} \neq \sigma_{2} \neq \sigma_{3}$$

$$\sigma_{3} \qquad p \qquad p$$

$$hydrostatic stress$$

$$volume change only$$



deviatoric stress - jännitysdeviaattori angular distortion only

## **Stress Invariants 2(2)**

Also denoted as:  $J_1, J_2, J_3$ 

Invariants of the deviatoric Stress:

 $\sigma = -pI + s$ 

 $J_1 = \operatorname{tr}(\mathbf{s}) = 0$  $J_2 = \frac{1}{2}\operatorname{tr}(\mathbf{s}^2)$ deviatoric Stress  $J_3 = \det(\mathbf{s}) = \frac{1}{2} \operatorname{tr}(\mathbf{s}^3)$ 

$$egin{array}{c|c} egin{array}{c|c} egin{array}{c|c} egin{array}{c} I_s = 0 \end{array}, & rac{1}{2} \mathrm{tr}(s^2) \end{array}$$

$$egin{align} J_1 & I_s = 0 \;, & rac{1}{2} ext{tr}(s^2) \ J_2 & II_s = rac{1}{2} s_{ij} s_{ij} = rac{1}{6} \left[ (\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 
ight] \ &= rac{1}{6} \left[ (\sigma_{11} - \sigma_{22})^2 + (\sigma_{22} - \sigma_{23})^2 + (\sigma_{23} - \sigma_{11})^2 
ight] + \sigma_{12}^2 \ \end{array}$$

$$J_3 = rac{1}{6}[(\sigma_{11} - \sigma_{22})^2 + (\sigma_{22} - \sigma_{33})^2 + (\sigma_{33} - \sigma_{11})^2] + \sigma_{12}^2 + \sigma_{23}^2 + \sigma_{31}^2$$

$$\tau_{oct} = \sqrt{\frac{2}{3}J_2} \qquad II_s = \frac{3}{2}\tau_{oct}^2.$$
 For instance, the *equivalent stress* (von Mises stress) – *vertailujännitys* –

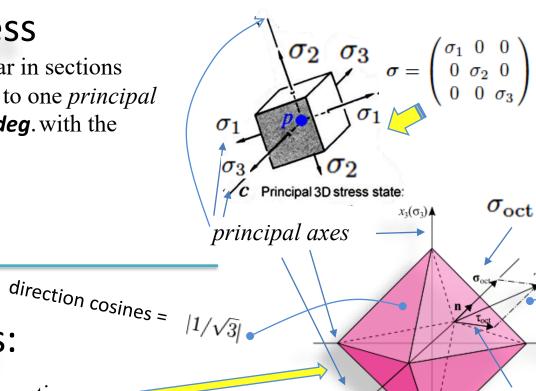
commonly used in solid mechanics is ..... = coefficient the maximum shear stress on the octahedral plane)

(which is up to a constant

 $\sigma_e \propto au_{oct}$ 

## Maximum shear stress

**Extreme shear** stresses appear in sections with normal is perpendicular to one principal axis and forms angles of 45 deg. with the remaining two axes



## ctahedral Stresses:

octopod

Normal and shear stress in cross sections whose normal forms an equal angle with all 3 principal axes:

$$\sigma_{\rm oct} = \frac{\sigma_1 + \sigma_2 + \sigma_3}{3} = \frac{\sigma_{ii}}{3} = \frac{I_\sigma}{3} \ , \qquad \text{Mean normal stress } -p \equiv \sigma_m \text{ or also called } \frac{hydrostatic stress state}{stress state}$$
 
$$\tau_{\rm oct} = \frac{1}{3} \sqrt{(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2} \ . \qquad \qquad n \equiv n_{\rm okt} = \frac{1}{\sqrt{3}}$$

Octa = eight = 8; we have 8 such planes

$$(3-\sigma_1)^2$$
.
$$n \equiv n$$

$$m{n} \equiv m{n}_{
m okt} = rac{1}{\sqrt{3}} \left[ egin{array}{c} 1 \ 1 \ 1 \end{array} 
ight],$$

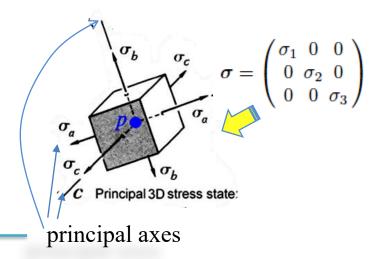
 $x_2(\sigma_2)$ 

 $au_{
m oct}$ 

$$\sigma_n = \mathbf{t} \cdot \mathbf{n} = t_i n_i = \sigma_{ij} n_i n_j.$$

#### Maximum shear stress:

• Extreme shear stresses appear in sections with normal is perpendicular to one principal axis and forms angles of 45 deg. with the remaining two axes



#### Octahedral stresses:

direction cosines =

 Normal and shear stress in cross sections whose normal forms an equal angle with all 3 principal axes:

$$\sigma_{\text{oct}} = \frac{\sigma_1 + \sigma_2 + \sigma_3}{3} = \frac{\sigma_{ii}}{3} = \frac{I_{\sigma}}{3} ,$$

$$\tau_{\text{oct}} = \frac{1}{3} \sqrt{(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2} .$$

Can be related to some type of weighted mean ~'max. shear stress'

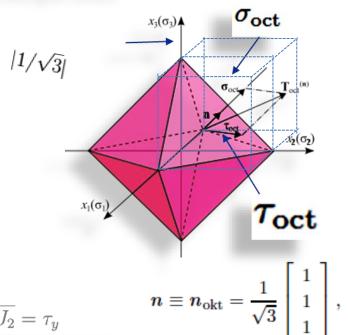
$$\tau_{oct} = \sqrt{2/3} \cdot \tau_y, \quad \tau_y = \sigma_y/\sqrt{3} \text{ von-Mises yield condition:} \quad \sqrt{J_2} = \tau_y$$

Hints:

$$\sigma_n = t \cdot n = t_i n_i = \sigma_{ij} n_i n_j$$

Exercise: show these results:

$$\sigma_{
m oct} = rac{I_{\sigma}}{3}$$
  $au_{
m oct} = \sqrt{rac{2}{3}J_2}$ 



## Invariants in general

 $\mathbf{A} = \sigma$ 

• 2<sup>nd</sup>-order symmetric tensors (matrices) have always three groups independent invariants which, for our purposes, in mechanics, are defined as

$$I_1^A = \operatorname{tr} \mathbf{A},$$
  $A = \boldsymbol{\sigma}$   $J_2^A = \frac{1}{2} \operatorname{tr} (\mathbf{A}^{\operatorname{d}})^2,$  The stress deviator  $\mathbf{S} = \boldsymbol{\sigma}^{\operatorname{d}}$   $\mathbf{S} = \mathbf{A}^{\operatorname{d}} = \boldsymbol{\sigma}^{\operatorname{d}} \equiv \mathbf{S}$   $\mathbf{T} = (\mathbf{S}^{\operatorname{d}})^2,$   $\mathbf{T} = (\mathbf{S}^{\operatorname{d}})^2,$   $\mathbf{T} = (\mathbf{S}^{\operatorname{d}})^2,$ 

• In plasticity, one needs to compute gradients of functions of invariants, for instance, the gradient for an arbitrary isotropic yield function (Isotropic plasticity theory) is as

$$\frac{\partial f(I_1, J_2, J_3)}{\partial \boldsymbol{\sigma}} = \frac{\partial f}{\partial I_1} \mathbf{I} + \frac{\partial f}{\partial J_2} \mathbf{S} + \frac{\partial f}{\partial J_3} \mathbf{T} \quad \text{, where } \quad \text{Hill tensor } \mathbf{T} = (\mathbf{S}^2)^{\frac{1}{6}}$$

Lode invariants

cylindrical coordinates

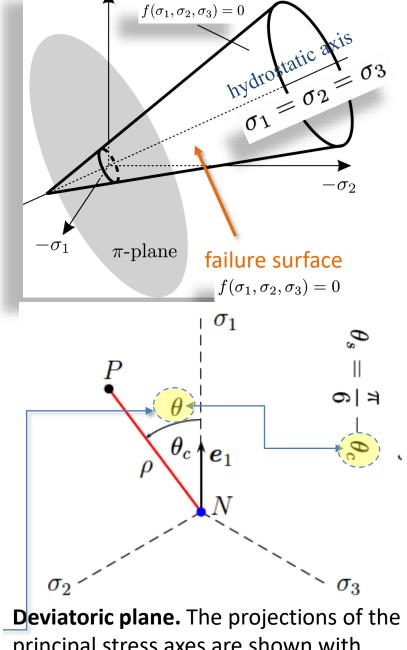
$$(r:=rho, \theta, z:=ksi)$$
 $\sigma_3$ 
 $\sigma_3$ 
 $\sigma_3$ 
 $\sigma_3$ 
 $\sigma_3$ 

Principal stress space.

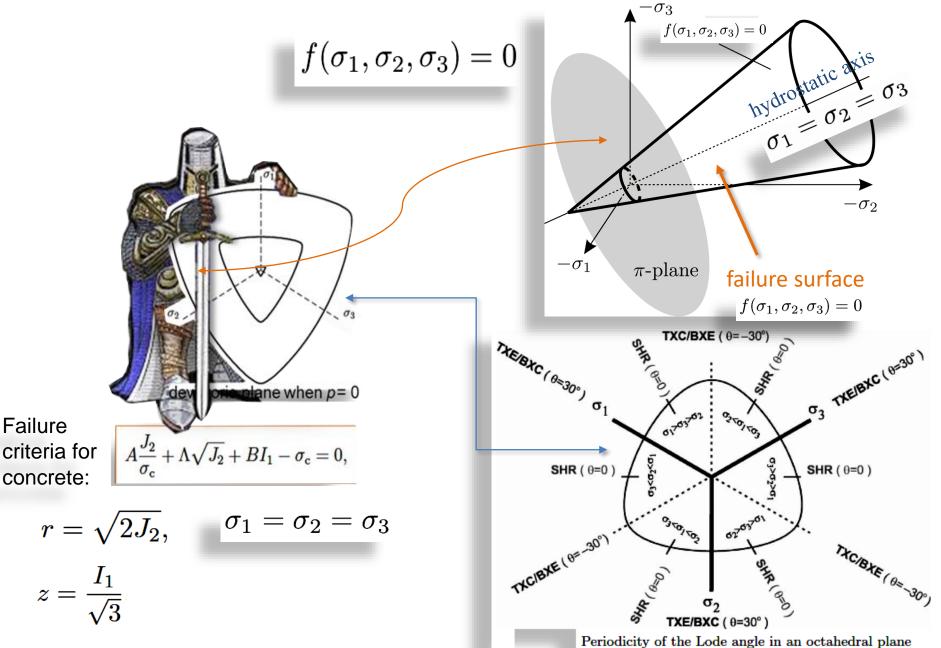
**Lode** invariants is an other alternative invariant triplet that is more useful than principal stresses for geometrical visualization of isotropic yield surfaces and is the cylindrical coordinates  $(r, \theta, z)$ representation of the above representation, where the z-coordinate points along the hydrostatic axis

$$r=\sqrt{2J_2},$$
  $\sigma_1=\sigma_2=\sigma_3$   $z=rac{I_1}{\sqrt{3}}$ 

$$\sin(3 heta_s) = -\sin(3ar{ heta}_s) = \cos(3ar{ heta_c}) = rac{J_3}{2}igg(rac{3}{J_2}igg)^{3/2}$$



principal stress axes are shown with dashed line (ref. Reijo's lecture Notes)



 $\sin(3 heta_s) = -\sin(3ar{ heta}_s) = \cos(3 heta_c) = rac{J_3}{2}igg(rac{3}{J_2}igg)^{3/2}$ 

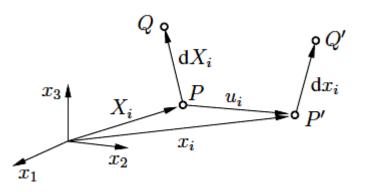
**Failure** 

concrete:

Ref for this figure above: Brannon, R. M. (2007). Elements of Phenomenological Plasticity: geometrical insight, computational algorithms, and applications in shock physics. Shock Wave Science and Technology Reference Library: Solids I, Springer-New York. 2: pp. 189-274

## Strain

#### **Deformation**



$$\begin{split} \mathrm{d}s^2 &= \mathrm{d}x_k \mathrm{d}x_k = \frac{\partial x_k}{\partial X_i} \frac{\partial x_k}{\partial X_j} \, \mathrm{d}X_i \mathrm{d}X_j \ , \\ \mathrm{d}S^2 &= \mathrm{d}X_k \mathrm{d}X_k = \mathrm{d}X_i \mathrm{d}X_j \, \delta_{ij} \ . \end{split}$$

$$\mathrm{d}s^2 = \mathrm{d}x_k \mathrm{d}x_k = \frac{\partial x_k}{\partial X_i} \frac{\partial x_k}{\partial X_j} \, \mathrm{d}X_i \mathrm{d}X_j \; ,$$
 
$$\mathrm{d}S^2 = \mathrm{d}X_k \mathrm{d}X_k = \mathrm{d}X_i \mathrm{d}X_j \, \delta_{ij} \; .$$
 
$$E_{ij} = \frac{1}{2} (\frac{\partial u_i}{\partial X_j} + \frac{\partial u_j}{\partial X_i} + \frac{\partial u_k}{\partial X_i} \frac{\partial u_k}{\partial X_j})$$

Infinitesimal or Engineering strain tensor:

$$\varepsilon_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i}) . \longrightarrow \varepsilon = \begin{pmatrix} \varepsilon_{11} & \varepsilon_{12} & \varepsilon_{13} \\ \varepsilon_{21} & \varepsilon_{22} & \varepsilon_{23} \\ \varepsilon_{31} & \varepsilon_{32} & \varepsilon_{33} \end{pmatrix}$$

$$(\partial u_i / \partial X_j \ll 1)$$
symmetric

Green's strain tensor

## Strain invariants

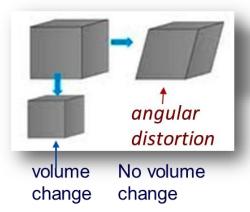
Same formula as for the stress tensor:

$$I_{\varepsilon}$$
,  $II_{\varepsilon}$ ,  $III_{\varepsilon}$ 

$$\varepsilon_{ij} = \frac{1}{2}(u_{i,j} + u_{j,i}) \quad \varepsilon = \begin{pmatrix} \varepsilon_{11} \ \varepsilon_{12} \ \varepsilon_{13} \\ \varepsilon_{21} \ \varepsilon_{22} \ \varepsilon_{23} \\ \varepsilon_{31} \ \varepsilon_{32} \ \varepsilon_{33} \end{pmatrix}$$

$$(\partial u_i / \partial X_j \ll 1)$$

Engineering strains



$$I_{\varepsilon} = \varepsilon_{V} = \varepsilon_{kk} = \varepsilon_{1} + \varepsilon_{2} + \varepsilon_{3}$$
.

Decomposition:

 $Volumetric\ strain\ (relative\ volume\ change)$ 

$$\varepsilon_{ij} = \frac{\varepsilon_{kk}}{3} \, \delta_{ij} + e_{ij} \quad \text{or} \quad \varepsilon = \frac{\varepsilon_V}{3} \, I + e$$

Distortion = Deviator

Most tensors used in engineering are symmetric 3×3.

For this case the invariants can be calculated as:

$$egin{aligned} \mathbf{I}_{A} &= \operatorname{tr}(\mathbf{A}) \ \mathbf{II}_{A} &= rac{1}{2} \left( (\operatorname{tr}\mathbf{A})^2 - \operatorname{tr}(\mathbf{A}\mathbf{A}) 
ight) \ \mathbf{III}_{A} &= \operatorname{det}(\mathbf{A}) \end{aligned}$$

Second Invariant of the deviator:

$$II_e = \frac{1}{2} e_{ij} e_{ij} = \frac{1}{6} \left[ (\varepsilon_1 - \varepsilon_2)^2 + (\varepsilon_2 - \varepsilon_3)^2 + (\varepsilon_3 - \varepsilon_1)^2 \right].$$