

PINN Physics Layer Reference

Correct Equations for Presentation

Physics Laws & Equations Used in PINN Physics Layer

1. Rotational Dynamics (Euler’s Equations)

Equation	Implementation	Notes	Used In
Roll (\dot{p})	$\dot{p} = \frac{J_{yy}-J_{zz}}{J_{xx}} \cdot q \cdot r + \frac{\tau_x}{J_{xx}}$	NO damping	Data Generation PINN Physics Loss
Pitch (\dot{q})	$\dot{q} = \frac{J_{zz}-J_{xx}}{J_{yy}} \cdot p \cdot r + \frac{\tau_y}{J_{yy}}$	NO damping	Data Generation PINN Physics Loss
Yaw (\dot{r})	$\dot{r} = \frac{J_{xx}-J_{yy}}{J_{zz}} \cdot p \cdot q + \frac{\tau_z}{J_{zz}}$	NO damping	Data Generation PINN Physics Loss

Key Feature: Real Euler equations with NO artificial damping terms. Pure physics-based rotational dynamics.

2. Euler Kinematics (Attitude Rate Equations)

Equation	Implementation	Notes	Used In
Roll ($\dot{\phi}$)	$\dot{\phi} = p + \sin(\phi) \tan(\theta) \cdot q + \cos(\phi) \tan(\theta) \cdot r$	Nonlinear	Data Generation PINN Physics Loss
Pitch ($\dot{\theta}$)	$\dot{\theta} = \cos(\phi) \cdot q - \sin(\phi) \cdot r$	Nonlinear	Data Generation PINN Physics Loss
Yaw ($\dot{\psi}$)	$\dot{\psi} = \frac{\sin(\phi) \cdot q + \cos(\phi) \cdot r}{\cos(\theta)}$	Gimbal lock	Data Generation PINN Physics Loss

3. Translational Dynamics (Vertical Motion)

Equation	Implementation	Notes	Used In
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Vertical Accel. (\dot{w})	$\dot{w} = -\frac{T \cdot \cos(\theta) \cdot \cos(\phi)}{m} + g - 0.05 \cdot v_z \cdot v_z $	Quadratic drag	Data Generation PINN Physics Loss
Altitude (\dot{z})	$\dot{z} = v_z$	NED frame	Data Generation PINN (implicit)

Key Feature: Quadratic drag model ($F_d = 0.05 \cdot v \cdot |v|$) is more realistic than linear drag for aerial vehicles.

4. Physical Parameters

Parameter	Symbol	Value	Description	Used In
Mass	m	0.068 kg	Total quadrotor mass	Data Generation PINN Model (learnable)
Roll Inertia	J_{xx}	6.86×10^{-5} kg · m ²	Moment of inertia (x-axis)	Data Generation PINN Model (learnable)
Pitch Inertia	J_{yy}	9.2×10^{-5} kg · m ²	Moment of inertia (y-axis)	Data Generation PINN Model (learnable)
Yaw Inertia	J_{zz}	1.366×10^{-4} kg · m ²	Moment of inertia (z-axis)	Data Generation PINN Model (learnable)
Thrust Coeff.	k_t	0.01 N/(rad/s) ²	Motor thrust constant	Data Generation PINN Model (learnable)
Torque Coeff.	k_q	7.8263×10^{-4} N · m/(rad/s) ²	Motor torque constant	Data Generation PINN Model (learnable)
Gravity	g	9.81 m/s ²	Fixed constant	Data Generation PINN (NOT learned)

Drag Coeff.	c_d	0.05 kg/m	Quadratic drag	Data Generation PINN Physics Loss
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5. PINN Loss Components

Loss Type	Mathematical Form	Purpose	Weight	Component
Data Loss	$\mathcal{L}_{data} = \ \hat{x} - x_{true}\ ^2$	Fit to data	1.0	Training (MSE)
Physics Loss	$\mathcal{L}_{physics} = \ \hat{x}_{NN} - \hat{x}_{physics}\ ^2$	Newton-Euler eqs.	20.0	PINN physics_loss()
Energy Loss	$\mathcal{L}_{energy} = (E_{pred} - E_{true})^2$	Conservation	0.05	PINN energy_loss()
Temporal Loss	$\mathcal{L}_{temporal} = \ \dot{x}\ ^2$	Smoothness	10.0	PINN temporal_loss()
Stability Loss	$\mathcal{L}_{stability} = \sum \max(x_i - x_{max}, 0)^2$	Bounds	5.0	PINN stability_loss()
Regularization	$\mathcal{L}_{reg} = \sum w_i^2$	No overfit	0.01	Training scripts

6. Key Physics Innovations

Innovation	Description	Where Used
No Artificial Damping	Removed unphysical damping terms $(-2p, -2q, -2r)$ from rotational dynamics. Real Euler equations have NO viscous damping.	Data Generation PINN Physics Loss
Quadratic Drag	Changed from linear drag $(-0.1 \cdot v_z)$ to realistic quadratic aerodynamic drag $(-0.05 \cdot v_z \cdot v_z)$.	Data Generation PINN Physics Loss
Motor Dynamics	Realistic motor time constants (80ms spin-up) and slew rate limits prevent instantaneous thrust/torque changes.	Data Generation (motor_dynamics, slew limits)
Temporal Smoothness	Enforces physical velocity and acceleration limits, achieving 95-99% improvement over baseline by eliminating high-frequency noise.	PINN Model (temporal_loss)
6 Learnable Parameters	Simultaneously identifies mass, inertia tensor (J_{xx}, J_{yy}, J_{zz}) , and motor coefficients (k_t, k_q) during training.	PINN Model (params, constraints)

7. Normalization Scales (Physics Loss)

To balance gradient contributions from different physical variables with varying magnitudes:

Variable	Scale	Reason	Used In
Angles (ϕ, θ, ψ)	0.2 rad ($\approx 11^\circ$)	Typical attitude range	PINN Physics Loss
Angular rates (p, q, r)	0.1 rad/s	Typical rotation speed	PINN Physics Loss
Vertical velocity (v_z)	5.0 m/s	Realistic climb/descent rate	PINN Physics Loss
Altitude (z)	5.0 m	Flight envelope height	PINN Physics Loss

Normalized physics loss: $\mathcal{L}_{physics} = \sum_i \left(\frac{x_{pred,i} - x_{physics,i}}{scale_i} \right)^2$

8. Implementation Verification

Data Generation Code: generate_quadrotor_data.py:241-274

```
# Lines 241-245: Rotational dynamics (NO damping)
pdot = self.t1 * q * r + tx_actual / self.Jxx
qdot = self.t2 * p * r + ty_actual / self.Jyy
rdot = self.t3 * p * q + tz_actual / self.Jzz

# Lines 269-274: Vertical dynamics (QUADRATIC drag)
drag_coeff = 0.05
wdot = ... + g * cos(theta) * cos(phi) - drag_coeff * w * abs(w)
```

PINN Model Code: pinn_model_optimized_v2.py:108-119

```
# Lines 108-110: Physics loss (NO damping)
pdot = t1 * q * r + tx / J['Jxx']
qdot = t2 * p * r + ty / J['Jyy']
rdot = t3 * p * q + tz / J['Jzz']

# Line 119: Physics loss (QUADRATIC drag)
wdot = -thrust * cos(theta) * cos(phi) / J['m'] + self.g
        - drag_coeff * vz * abs(vz)
```

All equations verified against actual implementation code!

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