

# PINN Physics Layer Reference

Correct Equations for Presentation

## Physics Laws & Equations Used in PINN Physics Layer

### 1. Rotational Dynamics (Euler's Equations)

Equation	Implementation	Notes
Roll Angular Acceleration	$\dot{p} = \frac{J_{yy}-J_{zz}}{J_{xx}} \cdot q \cdot r + \frac{\tau_x}{J_{xx}}$	Cross-coupling, NO damping
Pitch Angular Acceleration	$\dot{q} = \frac{J_{zz}-J_{xx}}{J_{yy}} \cdot p \cdot r + \frac{\tau_y}{J_{yy}}$	Cross-coupling, NO damping
Yaw Angular Acceleration	$\dot{r} = \frac{J_{xx}-J_{yy}}{J_{zz}} \cdot p \cdot q + \frac{\tau_z}{J_{zz}}$	Cross-coupling, NO damping

**Key Feature:** Real Euler equations with NO artificial damping terms. Pure physics-based rotational dynamics.

### 2. Euler Kinematics (Attitude Rate Equations)

Equation	Implementation	Notes
Roll Rate	$\dot{\phi} = p + \sin(\phi) \tan(\theta) \cdot q + \cos(\phi) \tan(\theta) \cdot r$	Nonlinear coupling
Pitch Rate	$\dot{\theta} = \cos(\phi) \cdot q - \sin(\phi) \cdot r$	Nonlinear coupling
Yaw Rate	$\dot{\psi} = \frac{\sin(\phi) \cdot q + \cos(\phi) \cdot r}{\cos(\theta)}$	Singularity at $\theta = \pm 90^\circ$

### 3. Translational Dynamics (Vertical Motion)

Equation	Implementation	Notes
Vertical Acceleration	$\dot{w} = -\frac{T \cdot \cos(\theta) \cdot \cos(\phi)}{m} + g - 0.05 \cdot v_z \cdot  v_z $	Quadratic aerodynamic drag
Altitude Rate	$\dot{z} = v_z$	Simple integration (NED frame)

**Key Feature:** Quadratic drag model ( $F_d = 0.05 \cdot v \cdot |v|$ ) is more realistic than linear drag for aerial vehicles.

## 4. Physical Parameters

Parameter	Symbol	Value	Description
Mass	$m$	0.068 kg	Total quadrotor mass
Roll Inertia	$J_{xx}$	$6.86 \times 10^{-5}$ kg · m <sup>2</sup>	Moment of inertia (x-axis)
Pitch Inertia	$J_{yy}$	$9.2 \times 10^{-5}$ kg · m <sup>2</sup>	Moment of inertia (y-axis)
Yaw Inertia	$J_{zz}$	$1.366 \times 10^{-4}$ kg · m <sup>2</sup>	Moment of inertia (z-axis)
Thrust Coef-ficient	$k_t$	0.01 N/(rad/s) <sup>2</sup>	Motor thrust constant
Torque Coef-ficient	$k_q$	$7.8263 \times 10^{-4}$ N · m/(rad/s) <sup>2</sup>	Motor torque constant
Gravity	$g$	9.81 m/s <sup>2</sup>	Fixed constant (NOT learned)
Drag Coeffi-cient	$c_d$	0.05 kg/m	Quadratic drag coefficient

## 5. PINN Loss Components

Loss Type	Mathematical Form	Purpose	Weight
Data Loss	$\mathcal{L}_{data} = \ \hat{x} - x_{true}\ ^2$	Fit predictions to data	1.0
Physics Loss	$\mathcal{L}_{physics} = \ \hat{x}_{NN} - \hat{x}_{physics}\ ^2$	Enforce Newton-Euler equations	20.0
Energy Loss	$\mathcal{L}_{energy} = (E_{pred} - E_{true})^2$	Energy conservation	0.05
Temporal Loss	$\mathcal{L}_{temporal} = \ \dot{x}\ ^2$	Smooth state transitions	10.0
Stability Loss	$\mathcal{L}_{stability} = \sum \max( x_i  - x_{max}, 0)^2$	Bounded predictions	5.0
Regularization	$\mathcal{L}_{reg} = \sum w_i^2$	Prevent overfitting	0.01

## 6. Key Physics Innovations

Innovation	Description
No Artificial Damping	Removed unphysical damping terms ( $-2p, -2q, -2r$ ) from rotational dynamics. Real Euler equations have NO viscous damping.

<b>Quadratic Drag</b>	Changed from linear drag ( $-0.1 \cdot v_z$ ) to realistic quadratic aerodynamic drag ( $-0.05 \cdot v_z \cdot  v_z $ ).
<b>Motor Dynamics</b>	Realistic motor time constants (80ms spin-up) and slew rate limits prevent instantaneous thrust/torque changes.
<b>Temporal Smoothness</b>	Enforces physical velocity and acceleration limits, achieving 95-99% improvement over baseline by eliminating high-frequency noise.
<b>6 Learnable Parameters</b>	Simultaneously identifies mass, inertia tensor ( $J_{xx}, J_{yy}, J_{zz}$ ), and motor coefficients ( $k_t, k_q$ ) during training.

## 7. Normalization Scales (Physics Loss)

To balance gradient contributions from different physical variables with varying magnitudes:

Variable	Normalization Scale	Reason
Angles ( $\phi, \theta, \psi$ )	0.2 rad ( $\approx 11\ddot{r}$ )	Typical attitude range
Angular rates ( $p, q, r$ )	0.1 rad/s	Typical rotation speed
Vertical velocity ( $v_z$ )	5.0 m/s	Realistic climb/descent rate
Altitude ( $z$ )	5.0 m	Flight envelope height

$$\text{Normalized physics loss: } \mathcal{L}_{physics} = \sum_i \left( \frac{x_{pred,i} - x_{physics,i}}{\text{scale}_i} \right)^2$$

# Complete Physics Equations Reference

Physics-Informed Neural Networks for Quadrotor Dynamics

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