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Expand of Finite Element Toolbox

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Introduction

1.1 Motivation

The finite element method (FEM) is a numerical technique for finding approximate solutions to boundary value problems for partial differential equations. FEM contains linear FEM and nonlinear FEM. Linear analysis follows the equation $K \cdot D = F$, which K stands for stiffness matrix, D stands for displacement and F stands for force. It means that the correlation of force and displacement is linear. This relation between force and displacement is only valid for material that has elastic linear property. But in real situation, elastic material such as steel has also nonlinear region. (see the Figure 1.1). The property is elastic linear, until at yield point, the steel is yielding and become non-linear. Non-linear cases include geometry non-linear and material non-linear. The example in Figure 1.1 just describes a geometry non-linear case and it is also the main topic in this thesis. This thesis will only focus on nonlinear FEM. In engineering and science in general, FEM helps tremendously in producing strength visualization and it is also possible to lead engineer for minimizing weight, materials and costs. The accuracy of solution are only limited by the quality of model and by the available computational power. Since the computational power has been improved enormously, FEM software offers a wide range of simulation of complex model designs and analysis of a system.

AMfe is a nonlinear finite element code for structural application at chair of applied mechanics in Technical University of Munich. AMfe Toolbox is developed in Python and Fortran. Python is a high-level, interpreted and dynamically-typed programming language. There are many numerical packages built on Python, such as Numpy, Scipy, Pandas. These packages provide high-performance, easy to use data structures, which just match for a FEM developing work. And because Python is a high-level programming language, it makes less time to develop the code, but on the other hand, it is slow for repeated execution of low-level task. Each Python operation comes with a small type-checking overhead, and with many repeated small operation, this overhead becomes significant. For that reason, the part of code are rewritten in Fortran. Fortran is a general-purpose, imperative programming language that is especially suited to numeric computation and scientific computing. So this combination obtain the advantage from both of them-easy to develop in Python and fast numeric computation in Fortran. The aim of AMfe Toolbox is to solve and analyse FEM problems, especially structural mechanical problems. It contains several

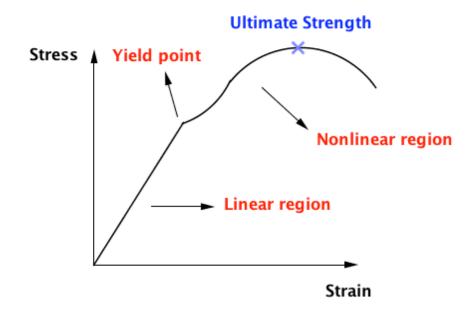


Figure 1.1: Stress-strain curve for steel

modules with different functions to solve problems step by step. A simple structure of AMfe are depicted in Figure 1.3. Displacement, stress and strain are three important factors for structural mechanical engineering. The function for calculation of nodal displacements is completed in AMfe Toolbox. Stress and strain calculation are still empty. Stress and strain calculations are of interest because in structural mechanical analysis and design for geometry the stresses are often very important to the engineer. For that reason, functions for solving of stress and strain will be added in AMfe Toolbox. When stress is able to export, a research to increase the accuracy of results is worthy of consideration. To improve the accuracy of the results, stress recovery is a approach to extrapolate the element solution to nodal solution. The goal is to get as much accuracy from the computed displacements while keeping the computational effort reasonable. When the most important functions are done and works well, it is time to check if AMfe Toolbox export the reasonable results. The results of stress and strain from AMfe Toolbox will be compared with a commercial FEM software - ANSYS. The compare of each element type will be recorded.

1.2 Structure of Study

This study can be divided into three parts. Figure 1.2 shows the flow chart of structure in this study. The first part is marked with green box. The main job for this part is to export data of meshed geometry by using ANSYS Parametric Design Language(APDL) code and import them into AMfe Toolbox. This procedure called Pre-Processing which is the first step in solving a

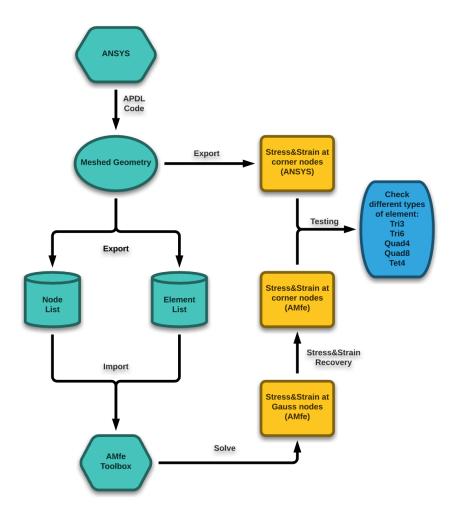


Figure 1.2: Flow chart of structure

problem in Finite Element Analysis. This step also ensures the both analysis with same set of elements and nodes. After that it is the processing step, which is marked with yellow box. AMfe Toolbox has several modules with different functions. These modules combine with each other and provides all the calculation function for processing. This study is focusing on calculating stress and strain. The last part marked with blue colour is to check the accuracy of stress and strain with using different types of elements.

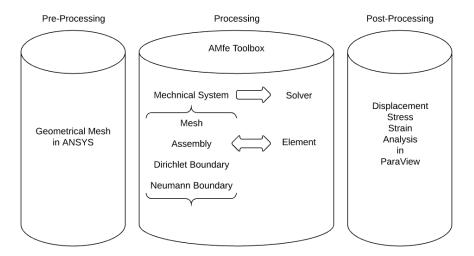


Figure 1.3: Overview of AMfe structure

Calculation of Element Strains and Stresses

2.1 Pre-Processing

The purpose of Pre-Processing is to build a numerical model and export as two lists- node list and element list. The data of model are ultimately imported into AMfe Toolbox. ANSYS Parametric Design Language (APDL) code is the tool to create a meshed geometry in ANSYS. A simple meshed geometry are shown in Figure 2.1. Task step for building model contains a series of related path. An example of a path step is:

Geometry \Rightarrow Element Type \Rightarrow Material \Rightarrow Mesh \Rightarrow Boundary Condition

After building the model, numerical data can be exported as node list and element list. Node list contains coordinate all the nodes of geometry. Element list are formed with the attributes of each elements and the index of node with belongs to element. The element attributes are assigned to different parts of model by 'pointing' to the appropriate entries in the element list. The pointers are simply a set of reference numbers that include the element index (ELEM), material number (MAT), element type number (TYP), real constant set number (REAL), a coordinate system number (ESY) and section ID number (SEC). Connecting with the example in Figure 2.1, both lists are depicted in Table 2.1 and Table 2.2. The numerical information of both lists are captured with regular expression approach and stored in AMfe Toolbox. For different data it is necessary to store in specific data type. Coordinate of each node and the index of nodes from elements are stored as Numpy array. NumPy is the fundamental package for scientific computing with Python. The advantage of Numpy for storage of node and element information depends on its efficient multi-dimensional container of generic data. It contains a powerful Ndimensional array object and sophisticated (broadcasting) functions. Moreover, it is possible for integrating Fortran code. General attributes of element can be stored as a efficient form-Pandas DataFrames. Pandas is a powerful data analysis toolkit, which provides fast, flexible, and expressive data structures designed to make working with relational or labeled data both easy and intuitive. Pandas DataFrames is a two-dimensional size-mutable, potentially heterogeneous tabular data structure with labeled axes (rows and columns). Arithmetic operations align on both row and column labels. This format can be thought of as a dictionary-like container for Series objects.

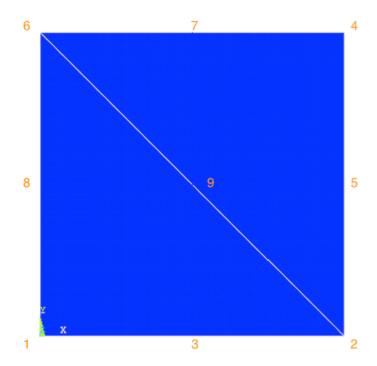


Figure 2.1: A simple meshed geometry

Table 2.1: Node list exported from ANSYS

NODE	X	Y	Z	THXY	THYZ	THZX
1	0.0000	0.0000	0.0000	0.00	0.00	0.00
2	50.000	0.0000	0.0000	0.00	0.00	0.00
3	25.000	0.0000	0.0000	0.00	0.00	0.00
4	50.000	50.000	0.0000	0.00	0.00	0.00
5	50.000	25.000	0.0000	0.00	0.00	0.00
6	0.0000	50.000	0.0000	0.00	0.00	0.00
7	25.000	50.000	0.0000	0.00	0.00	0.00
8	0.0000	25.000	0.0000	0.00	0.00	0.00
9	25.000	25.000	0.0000	0.00	0.00	0.00

Table 2.2: Element list exported from ANSYS

ELEM	MAT	TYP	REL	ESY	SEC	Node	Node	Node	Node	Node	Node
1	1	1	1	0	1	6	2	4	9	5	7
2	1	1	1	0	1	6	1	2	8	3	9

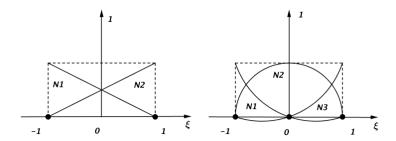


Figure 2.2: Linear shape function and quadratic shape function.

2.2 Shape Function

2.2.1 Quadrilateral Elements

To get the approximate solution (position, displacement, stress and strain) in structural mechanical field is the goal of AMfe Code. The approximation is basically depended on shape function N. The standard approach for the definition of shape functions is to chose them as simple polynomials which are associated to nodes. One element includes n nodes and n shape functions, where the i-th shape function takes the value 1 at the i-th node in reference coordinates, then the j-th shape function can be written like this:

if
$$j = i$$
 then $N_j(\hat{\xi}_i) = 1$
if $j \neq i$ then $N_j(\hat{\xi}_i) = 0$

for
$$i, j = 1,...,n$$
.

In Figure 2.2, linear, quadratic shape function are shown together with corresponding node positions. When it comes to a element with three nodes, as shown in the right of Figure 2.2. For each shape functions, there are three restrictions:

$$\begin{split} N_1\left(\hat{\xi_1}=-1\right) &= 1,\, N_1\left(\hat{\xi_2}=0\right) = 0,\, N_1\left(\hat{\xi_3}=1\right) = 0;\\ N_2\left(\hat{\xi_1}=-1\right) &= 0,\, N_2\left(\hat{\xi_2}=0\right) = 1,\, N_2\left(\hat{\xi_3}=1\right) = 0;\\ N_3\left(\hat{\xi_1}=-1\right) &= 0,\, N_3\left(\hat{\xi_2}=0\right) = 0,\, N_3\left(\hat{\xi_3}=1\right) = 1 \end{split}$$

In general quadratic polynomial, the shape functions have three coefficients, and it can be constructed as follows:

$$p_{quad}\left(\xi\right)=c_{0}+c_{1}\xi+c_{2}\xi^{2}$$

Lagrange polynomials are a general class of polynomials with the node association property: a Lagrange polynomial l_k^n-1 (in one coordinate ξ) of order n-1 passes through n nodes with coordinates $\bar{\xi}^j$ (j=1,...,n) of which the single node k evaluates unity $\left(l_k^{n-1}\left(\bar{\xi}^k\right)=1\right)$ and every other node results in zero $\left(l_k^{n-1}\left(\bar{\xi}^k\right)=0\right)$ for all $j\neq k$.

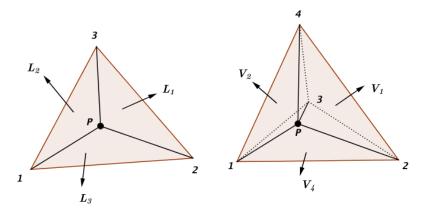


Figure 2.3: triangular and tetradral elemnets.

$$l_{k}^{n-1}(\xi) = \prod_{j=1, j \neq k}^{n} \frac{\xi - \bar{\xi}^{1}}{\xi^{k} - \bar{\xi}^{j}} = \frac{(\xi - \bar{\xi}^{1}) \cdots (\xi - \bar{\xi}^{k-1})(\xi - \bar{\xi}^{k+1}) \cdots (\xi - \bar{\xi}^{n})}{(\bar{\xi}^{k} - \bar{\xi}^{1}) \cdots (\bar{\xi}^{k} - \bar{\xi}^{k-1})(\bar{\xi}^{k} - \bar{\xi}^{k+1}) \cdots (\bar{\xi}^{k} - \bar{\xi}^{n})}$$

$$k = 1, 2, \dots, n$$

For instance linear, i.e. order 1, Lagrange polynomials are achieved with $n-1=1 \Rightarrow n=2$ nodes. Using $\bar{\xi}^1=-1$ and $\bar{\xi}^2=+1$ results in: $l_1^1(\xi)=-1/2(\xi-1)$ and $l_2^1(\xi)=1/2(\xi+1)$. Then the linear shape functions can be simply constructed as $N_1=l_1^1=1/2(1-\xi)$ and $N_2=l_2^1=1/2(1+\xi)$.

2.2.2 Triangular and Tetrahedral Elements

Due to the higher flexibility of triangles and tetrahedra for meshing complex geometries, they are often preferred over quadrilaterals or hexahedral. The shape functions are given in an analog way but are expressed in area and volume coordinates, respectively. Figure ?? shows the geometries and shape functions. These coordinates represent area and volume fractions, a visualization is given below:

$$L_1 = \frac{\text{area}P23}{\text{area}123}, \quad V_1 = \frac{\text{area}P234}{\text{area}1234}$$

$$\sum L_i = 1 \qquad \sum V_i = 1$$

2.3 Non- linear Element Formulation

2.3.1 Discretization of the Displacement Field

After meshing of the geometry, displacement field u can be discretized by inserting shape function N. Firstly, we introduce the nodal displacement d^e to save the displacement at each node of the

considered element for each spatial direction. In elastic materials, through the elastic constitutive equations can be derived that stresses σ are directly related to strains ε at each point:

$$\sigma = E\varepsilon \tag{2.1}$$

It follows that the stress calculation procedure begins with strain calculation, and that the accuracy of stresses depends on that of strains. In the following sections, we focus our attention on two-dimensional isoparametric elements, as the calculation of strains, stresses and axial forces in bar elements is strightforward. Suppose that we have solved the master stiffness equations.

$$Ku = f (2.2)$$

for the node displacements u. To calculate strains and stresses we perform a loop over all defined elements. Let ε be the element index of a specific two-dimensional isoparametric element encountered during this loop, and $u(\varepsilon)$ the vector of computed element node displacements. The strains at any point in the element can be related to these displacements as

$$\varepsilon = Bu(\varepsilon) \tag{2.3}$$

where B is the strain-displacement matrix assembled with the x and y derivatives of the element shape functions evaluated at the point where we are calculating strains. The corresponding stresses are given by

$$\sigma = E\varepsilon = EBu \tag{2.4}$$

In the applications it is of interest to evaluate and report these stresses at the element nodal points located on the corners and possibly midpoints of the element. These are called element nodal point stresses. It is important to realize that the stresses computed at the same nodal point from adjacent elements will not generally be the same, since stresses are not required to be continuous in displacement- assumed finite elements. This suggests some form of stress averaging can be used to improve the stress accuracy, and indeed this is part of the stress recovery technique further. The results from this averaging procedure are called nodal point stresses.

2.4 Gauss integration

Numerical integration plays a role in finite element. Gauss integration, as a very efficient approach compared to other numerical integration, integrate a function $f(\xi)$ on the spatial domain Ω is replaced by a summation of certain function values at the so-called Gauss points $\tilde{\xi}$ which are each multiplied by a scalar (weight) ω . In total $numgp_1 \times numggp_2 \times numgp_3$ Gauss points are used:

$$\iiint_{E} f(\xi) d\xi_{1} d\xi_{2} d\xi_{3} = \sum_{i=1}^{numgp_{1}} \sum_{j=1}^{numgp_{2}} \sum_{k=1}^{numgp_{3}} f\left(\tilde{\xi}_{1}^{i} \tilde{\xi}_{2}^{j} \tilde{\xi}_{3}^{k}\right) \cdot w_{1}^{i} w_{2}^{j} w_{3}^{k}$$
(2.5)

In which we used corresponding weights for each of the three spatial directions. This allows to use Gauss points coordinates and weights tabulated for the one-dimensional case.

The location and weights of these points can be given analytically in Table 2.3.

numgp	i	ξ^i	ω^i
1	1	0	2
2	1	$-1/\sqrt{3}$	1
2	2	$+1/\sqrt{3}$	1
	1	$-\sqrt{3/5}$	5/9
3	2	0	8/9
	3	$+\sqrt{3/5}$	5/9
	1	$-\sqrt{\left(15+\sqrt{120}\right)/35}$	$\left(18 - \sqrt{30} \middle/ 36\right)$
4	2	$-\sqrt{\left(15-\sqrt{120}\right)/35}$	$\left(18 + \sqrt{30} \middle/ 36\right)$
	3	$+\sqrt{\left(15-\sqrt{120}\right)/35}$	$\left(18 + \sqrt{30} \middle/ 36\right)$
	4	$+\sqrt{\left(15+\sqrt{120}\right)/35}$	$\left(18 - \sqrt{30} \middle/ 36\right)$
	1	$-1/3\sqrt{5+2\sqrt{10/7}}$	$(332 - 13\sqrt{70})/900$
5	2	$-1/3\sqrt{5-2\sqrt{10/7}}$	$(332 + 13\sqrt{70})/900$
	3	0	128/255
	4	$+1/3\sqrt{5-2\sqrt{10/7}}$	$(332 + 13\sqrt{70})/900$
	5	$+1/3\sqrt{5+2\sqrt{10/7}}$	$(332 - 13\sqrt{70})/900$

Table 2.3: Gauss-Legendre points and weights

2.5 Extrapolation from Gauss Points

2.5.1 Quad4

This will be explained for the four-node bilinear quadrilateral. The normal Gauss integration rule for element stiffness evaluation is 2×2 , as illustrated in Figure 2.4. The stresses are calculated at the Gauss points, which are identified as k_1^G , k_2^G , k_3^G and k_4^G in Figure 2.4. Point k_i^G is closest to node k_i^E so it is seen that Gauss point numbering essentially follows element node numbering in the counterclockwise sense. The natural coordinates of these points are listed in Table 2.4. The stresses are evaluated at these Gauss points by passing these natural coordinates to the shape function subroutine. Then each stress component is "carried" to the corner nodes k_1^E through k_4^E through a bilinear extrapolation based on the computed values at k_1^G through k_4^E . To understand the extrapolation procedure more clearly it is convenient to consider the region bounded by the Gauss points as an "internal element" or "Gauss element". This interpretation is depicted in Figure 2.4. The Gauss element, denoted by (G), is also a four-node quadrilateral. The coordinate of node for element and Gauss element can be represented as k_i^G and k_i^E , respectively.

Its quadrilateral (natural) coordinates are denoted by ξ' and η' . These are linked to ξ' and η' by the simple relations from Gauss-Legendre quadrature in Table 2.4.

$$k_i^G = \frac{k_i^E}{\sqrt{3}}, \quad k_i^E = k_i^G \sqrt{3}$$
 (2.6)

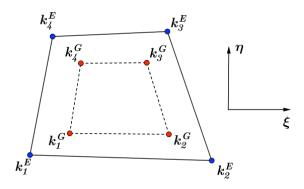


Figure 2.4: Quad4 in element coordinate and Gauss element coordinate.

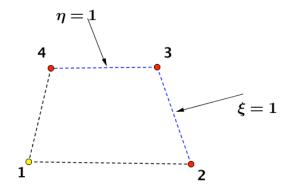


Figure 2.5: Equation of side opposite corner 1 for Quad4.

The element geometry and natural coordinates are shown in Figure 2.4. Only one type of node (corner) and associated shape function is present. Consider node 1 as typical. Inspection of Figure 2.5 suggests trying

$$N_1^e = c_1 L_{2-3} L_{3-4} (2.7)$$

This plainly vanishes over nodes 2, 3 and 4, and can be normalized to unity at node 1 by adjusting c_1 . By construction it vanishes over the sides 2-3 and 3-4 that do not belong to 1. The equation

Corner	ξ	η	ξ'	η'	Gauss node	ξ	η	ξ'	η'
1	-1	-1	$-\sqrt{3}$	$-\sqrt{3}$	1'	$1/\sqrt{3}$	$-1/\sqrt{3}$	-1	-1
2	+1	-1	$+\sqrt{3}$	$-\sqrt{3}$	2'	$1/\sqrt{3}$	$1/\sqrt{3}$	+1	-1
3	+1	+1	$+\sqrt{3}$	$+\sqrt{3}$	3'	$1/\sqrt{3}$	$1/\sqrt{3}$	+1	+1
4	-1	+1	$-\sqrt{3}$	$+\sqrt{3}$	4'	$-1/\sqrt{3}$	$1/\sqrt{3}$	-1	+1

Table 2.4: Natural Coordinate of Quad4

of side 2-3 is $\xi = 1$, or $\xi - 1 = 0$. The equation of side 3-4 is $\eta = 1$, or $\eta - 1 = 0$. Replacing in Equation (2.7) yields

$$N_1^e(\xi,\eta) = c_1(\xi-1)(\eta-1) = c_1(1-\xi)(1-\eta)$$
(2.8)

To find c_1 , evaluate at node 1, the natural coordinates of which are $\xi = \eta = -1$:

$$N_1^e(-1, -1) = c_1 \times 2 \times 2 = 4c_1 = 1 \tag{2.9}$$

Hence $c_1 = \frac{1}{4}$ and the shape functions is

$$N_1^e = \frac{1}{4} (1 - \xi) (1 - \eta) \tag{2.10}$$

For the other three nodes the procedure is the same, traversing the element cyclically. It can be verified that the general expression of the shape functions for this element is

$$N_i^e = \frac{1}{4} \left(1 + \xi_i \xi \right) \left(1 + \left(\eta_i \eta \right) \right) \tag{2.11}$$

Following this general expression, the shape functions of Node 2, 3 and 4 are demonstrated as

$$N_2^e = \frac{1}{4} (1 + \xi) (1 - \eta) \tag{2.12}$$

$$N_3^e = \frac{1}{4} (1 + \xi) (1 + \eta) \tag{2.13}$$

$$N_4^e = \frac{1}{4} (1 - \xi) (1 + \eta) \tag{2.14}$$

2.5.2 Quadrangle with quadratic shape function and eight nodes: Quad8

The Nine-Node biquadratic quadrilateral has three types of shape functions, which are associated with corner nodes, midside nodes and center node, respectively. The element coordinate and Gauss element coordinate are illustrated in Figure 2.6

The lines whose product is used to construct three types of shape functions are illustrated in Figure for nodes 1, 5 and 9, respectively. The technique has been sufficiently illustrated in previous

examples. Here the summary of calculation for nodes 1, 5 and 9, which are taken as representatives of the three types: The three types of shape function are represented in Figure 2.7.

$$N_1^e = c_1 L_{2-3} L_{3-4} L_{5-7} L_{6-8} = c_1 (\xi - 1) (\eta - 1) \xi \eta$$
 (2.15)

$$N_5^e = c_5 L_{2-3} L_{1-4} L_{6-8} L_{3-4} = c_5 (\xi - 1) (\xi + 1) \eta (\eta - 1) = c_5 (1 - \xi^2) \eta (1 - \eta)$$
 (2.16)

$$N_9^e = c_9 L_{1-2} L_{2-3} L_{3-4} L_{4-1} = c_9 \left(\xi - 1\right) \left(\eta - 1\right) \left(\xi + 1\right) \left(\eta + 1\right) = c_9 \left(1 - \xi^2\right) \left(1 - \eta^2\right) \ (2.17)$$

Imposing the normalization conditions the results are

$$c_1 = \frac{1}{4}, \quad c_5 = -\frac{1}{2}, c_9 = 1$$

By following this approach all the shape functions can be calculated

$$N_1 = \frac{1}{4} (\xi - 1) (\eta - 1) \cdot \xi \cdot \eta \tag{2.18}$$

$$N_2 = \frac{1}{4} (\xi + 1) (\eta - 1) \cdot \xi \cdot \eta \tag{2.19}$$

$$N_3 = \frac{1}{4} (\xi + 1) (\eta + 1) \cdot \xi \cdot \eta \tag{2.20}$$

$$N_4 = \frac{1}{4} (\xi - 1) (\eta + 1) \cdot \xi \cdot \eta \tag{2.21}$$

$$N_5 = \frac{1}{2} (1 + \xi) (1 + \xi) (\eta - 1) \cdot \eta \tag{2.22}$$

$$N_6 = \frac{1}{2} (1 + \eta) (1 - \eta) (\xi + 1) \cdot \xi \tag{2.23}$$

$$N_7 = \frac{1}{2} (1 + \xi) (1 + \xi) (\eta + 1) \cdot \eta \tag{2.24}$$

$$N_8 = \frac{1}{2} (1 + \eta) (1 - \eta) (\xi - 1) \cdot \xi \tag{2.25}$$

$$N_9 = (1+\xi)(1-\xi)(1+\eta)(1-\eta) \tag{2.26}$$

2.5.3 Triangle with three nodes: Tri3

The geometry of the 3-node triangle shown in Figure 2.8 is specified by the location of its three corner nodes on the $\{x, y\}$ plane

The shape function for triangular element has a different form, which compares with quadrilateral elements. The three shape functions have the simply own coordinates - the triangular

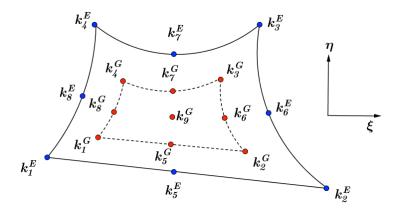


Figure 2.6: Quad8 in element coordinate and Gauss element coordinate.

coordinates: $N_i = \xi_i$ for i = 1, 2, 3. The shape function can be derived from a method as follows:

The equation of the triangle side opposite to node i is $L_{j-k} = \xi_i = 0$, where j and k are the cyclic permutations of i. Here symbol L_{j-k} denotes the left hand side of the homogeneous equation of the natural coordinate line that passes through node points j and k. See Figure 2.9 for i = 1, j = 2 and k = 3. Hence the obvious suppose is:

$$N_i^e = c_i L_i (2.27)$$

To find c_1 , evaluate $N_1^e\left(\xi_1,\xi_2,\xi_3\right)$ at node 1. The triangular coordinates of this node are $\xi_1=1$, $\xi_2=\xi_3=0$. $N_1^e\left(1,0,0\right)=c_1\times 1=1$. so $c_1=1$ and analogically $c_2=1$, $c_3=1$

$$N_1^e = L_1, \quad N_2^e = L_2, \quad N_3^e = L_3$$
 (2.28)

2.5.4 Second order triangle with six nodes: Tri6

The geometry of the six-node quadratic is shown in Figure XXX. Inspection reveals two types of nodes: corners(1, 2 and 3) and midside nodes(4, 5 and 6). For both cases it is necessary to product the two linear functions in the triangular coordinates because the shape function should be quadratic. These functions are illustrated in Figures XXX for corner node 1 and midside node 4, respectively.

For corner node 1, inspection of Figure 2.11 at top left side suggests trying

$$N_1^e = c_1 L_{2-3} L_{4-6} (2.29)$$

 N_1^e will vanish over 2-5-3 and 4-6. This makes the function zero at node 2,3,4,5,6, as is obvious upon inspection of Figure XXX, while being nonzero at 1. The value can be adjusted to be unity if c_1 is appropriately chosen. The equations of the lines that appear (XXX equation upper) are

$$L_{2-3}: \xi_1 = 0, \quad L_{4-6}: \xi_1 - \frac{1}{2} = 0$$
 (2.30)

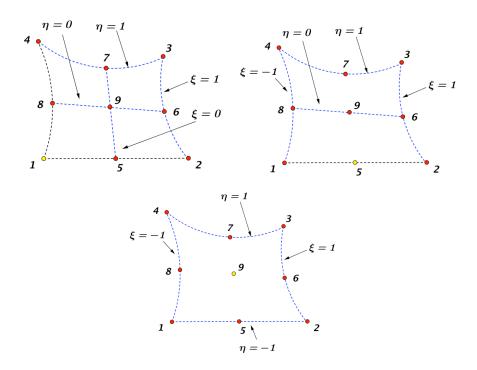


Figure 2.7: Equation of side opposite corner 1 for Quad8.

Replacing into (eq upper upper)

$$N_1^e = c_1 \xi_1 \left(\xi_1 - \frac{1}{2} \right) \tag{2.31}$$

Same as triangle with three nodes, $N_1^e(1,0,0) = c_1 \times 1 \times \frac{1}{2} = 1$. Then $c_1 = 2$ can be calculated and finally

$$N_1^e = 2\xi_1 \left(\xi_1 - \frac{1}{2}\right) = \xi_1 \left(2\xi_1 - 1\right) \tag{2.32}$$

For midside node 4, inspection of Figure XXX suggests trying

$$N_4^e = c_4 L_{2-3} L_{1-3} (2.33)$$

The equation of sides L_{2-3} and L_{1-3} are $\xi_1=0$ and $\xi_2=0$, respectively. Therefore $N_4^e\left(\xi_1,\xi_2,\xi_3\right)=c_4\xi_1\xi_2$. To find c_4 , evaluate this function at node 4, the triangular coordinates of which are $\xi_1=\xi_2=\frac{1}{2},\xi_3=0$. Then $N_4^e\left(\frac{1}{2},\frac{1}{2},0\right)=c_4\times\frac{1}{2}\times\frac{1}{2}=1$. Hence $c_4=4$, which the shape function gives

$$N_A^e = 4\xi_1 \xi_2 \tag{2.34}$$

The rest shape function can be calculated by using same approach.

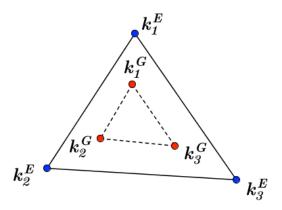


Figure 2.8: Tri3 in element coordinate and Gauss element coordinate.

2.5.5 Tetraeder element with four nodes: Tet4

2.6 Coordinate Transformations

Quantities that are closely linked with the element geometry are best expressed in triangular coordinates. On the other hand, quantities such as displacements, strains and stresses are usually expressed in the Cartesian system. Thus we need transformation equations through which it is possible to pass from one coordinate system to the other. Cartesian and triangular coordinates are linked by the relation.

$$\begin{bmatrix} 1 \\ x \\ y \end{bmatrix} = \begin{bmatrix} 1 & 1 & 1 \\ x_1 & x_2 & x_2 \\ y_1 & y_2 & y_2 \end{bmatrix} \begin{bmatrix} \xi_1 \\ \xi_2 \\ \xi_3 \end{bmatrix}$$
 (2.35)

$$\begin{bmatrix} \xi_1 \\ \xi_2 \\ \xi_3 \end{bmatrix} = \frac{1}{2A} \begin{bmatrix} x_2 y_3 - x_3 y_2 & y_2 - y_3 & x_3 - x_2 \\ x_3 y_1 - x_1 y_3 & y_3 - y_1 & x_1 - x_3 \\ x_1 y_2 - x_2 y_1 & y_1 - y_2 & x_2 - x_1 \end{bmatrix} \begin{bmatrix} 1 \\ x \\ y \end{bmatrix} = \frac{1}{2A} \begin{bmatrix} 2A_{23} & y_{23} & x_{32} \\ 2A_{31} & y_{31} & x_{13} \\ 2A_{12} & y_{12} & x_{21} \end{bmatrix} \begin{bmatrix} 1 \\ x \\ y \end{bmatrix}$$
(2.36)

bibid

Corner	ξ	η	ξ'	η'	Gauss node	ξ	η	ξ'	η'
1	-1	-1	-5/3	-5/3	1'	-2/3	-2/3	-1	-1
2	+1	-1	+7/3	-5/3	2'	+1/3	-2/3	+1	-1
3	-1	+1	-5/3	+7/3	3'	-2/3	+1/3	-1	+1

Table 2.5: Natural Coordinate of Tri3

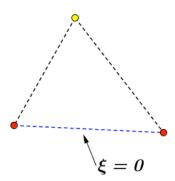


Figure 2.9: Equation of side opposite corner 1 for Tri3.

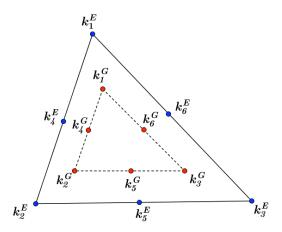


Figure 2.10: Tri6 in element coordinate and Gauss element coordinate.

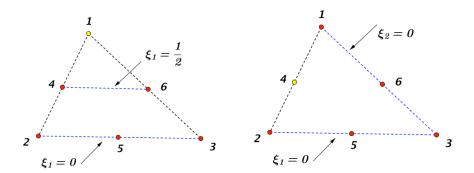


Figure 2.11: Equation of side opposite corner 1 for Tri6.

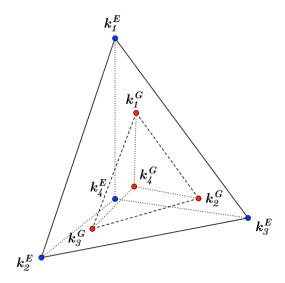


Figure 2.12: Tri4 in element coordinate and Gauss element coordinate.

Table 2.6: Tetrahedral Coordinate of Tet4

Corner node	L1	L2	L3	L4	L1'	L2'	L3'	L4'
1	1	0	0	0	α	β	β	β
2	0	1	0	0	β	α	β	β
3	0	0	1	0	β	β	α	β
4	0	0	0	1	β	β	β	α
Gauss node	L1	L2	L3	L4	L1'	L2'	L3'	L4'
1'	α	β	β	β	1	0	0	0
2'	β	α	β	β	0	1	0	0
3'	β	β	α	β	0	0	0	1
4'	β	β	β	α	0	0	0	1
$\alpha = 0.58541020; \beta = 0.13819660$								

Calculation of Element Strains and Stresses

- 3.1 Shape Function
- 3.1.1 Quadrilateral Elements
- 3.1.2 Triangular and Tetrahedral Elements

Bestimmung des optimalen Tunings k

Zusammenfassung und Ausblick

Averaging-Methode

Die Averaging-Methode oder Methode der Mittelwertbildung (engl. averaging method) wird zur Analyse nichtlinearer Oszillationen eines dynamischen Systems verwendet Mitropolsky: Averaging Die Grundidee der Methode ist die Approximation des Ausgangssystems durch ein gemitteltes System, welches dadurch deutlich einfacher zu untersuchen ist Hagedorn: NichtlinSchwingungen1978 Aus dem Verständnis der Dynamik des gemittelten Systems können Rückschlüsse auf die Dynamik des Ausgangssystems gezogen werden Hagedorn: NichtlinSchwingungen1978 Dabei wird durch Zeitintegration über eine Periode des Ausgangssystems der Einfluss der hochfrequenten Dynamiken herausgefiltert Hagedorn: NichtlinSchwingungen1978 Die wichtigsten Zusammenhänge zur praktischen Anwendung, wie sie im Rahmen der Herleitungen in dieser Arbeit benötigt werden, sind im Folgenden kurz dargelegt, wobei hier die einfachste Form der Mittelwertbildung, die periodische Mittelwertbildung Sanders: AveragingMethods2007 betrachtet wird. Die angegebenen Zusammenhänge geben den Grundgedanken der Averaging-Methode wieder. Weitere Details, Beweise und Verallgemeinerungen sind in Sanders: AveragingMethods2007 zu finden.

Ausgegangen wird von einem System in der Form

$$\vec{\dot{x}} = \epsilon f(x, t) + \mathcal{O}(\epsilon^2), \qquad x(0) = a, \tag{5.1}$$

wobei ϵ ein kleiner Parameter und f ein T-periodisches Vektorfeld in der Zeit t ist.

Vernachlässigung von Termen höherer Ordnung und Mittelwertbildung durch Integration über *t* führt auf die *gemittelte Gleichung*

$$\vec{z} = \epsilon \vec{f}(z), \qquad z(0) = a, \tag{5.2}$$

mit

$$\vec{\bar{f}}(z) = \frac{1}{T} \int_0^T f(z, s) \, \mathrm{d}s. \tag{5.3}$$

Das grundlegende Ergebnis dieser Mittelung ist, dass die Lösungen des gemittelten Systems nahe (von Ordnung ϵ) an den Lösungen des Ausgangssystems in einem Zeitintervall der Ordnung $1/\epsilon$ sind **Sanders:AveragingMethods2007** was sich als

$$\|\mathbf{x}(t) - \mathbf{z}(t)\| \le c\epsilon \quad \text{für} \quad 0 \le t \le \frac{L}{\epsilon}$$
 (5.4)

mit positiven Konstanten c und L schreiben lässt. Unter weiter spezifizierten Anforderungen, welche für reale Systeme in der Regel zutreffen, bleiben die Lösungen des gemittelten Systems

auch für das Zeitintervall $t \in [0, \infty)$ nahe (von Ordnung e) an den Lösungen des Ausgangssystems **Sanders:AveragingMethods2007**

Rotationspendel

Disclaimer

I hereby declare that this thesis is entirely the result o	of my own work except where otherwise
indicated. I have only used the resources given in the li	st of references.
Garching, 30.05.2016	(Signature)