Thermodynamic Properties of Air and Mixtures of Nitrogen, Argon, and Oxygen From 60 to 2000 K at Pressures to 2000 MPa

Eric W. Lemmona)

Physical and Chemical Properties Division, National Institute of Standards and Technology, 325 Broadway, Boulder, Colorado 80303

Richard T Jacobsen and Steven G. Penoncello

Center for Applied Thermodynamic Studies, College of Engineering, University of Idaho, Moscow, Idaho 83844

Daniel G. Friend

Physical and Chemical Properties Division, National Institute of Standards and Technology, 325 Broadway, Boulder, Colorado 80303

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A thermodynamic property formulation for standard dry air based upon available experimental $p-\rho-T$, heat capacity, speed of sound, and vapor-liquid equilibrium data is presented. This formulation is valid for liquid, vapor, and supercritical air at temperatures from the solidification point on the bubble-point curve (59.75 K) to 2000 K at pressures up to 2000 MPa. In the absence of reliable experimental data for air above 873 K and 70 MPa, air properties were predicted from nitrogen data in this region. These values were included in the determination of the formulation to extend the range of validity. Experimental shock tube measurements on air give an indication of the extrapolation behavior of the equation of state up to temperatures and pressures of 5000 K and 28 GPa. The available measurements of thermodynamic properties of air are summarized and analyzed. Separate ancillary equations for the calculation of dew and bubble-point pressures and densities of air are presented. In the range from the solidification point to 873 K at pressures to 70 MPa, the estimated uncertainty of density values calculated with the equation of state is 0.1%. The estimated uncertainty of calculated speed of sound values is 0.2% and that for calculated heat capacities is 1%. At temperatures above 873 K and 70 MPa, the estimated uncertainty of calculated density values is 0.5% increasing to 1.0% at 2000 K and 2000 MPa. In addition to the equation of state for standard air, a mixture model explicit in Helmholtz energy has been developed which is capable of calculating the thermodynamic properties of mixtures containing nitrogen, argon, and oxygen. This model is valid for temperatures from the solidification point on the bubble-point curve to 1000 K at pressures up to 100 MPa over all compositions. The Helmholtz energy of the mixture is the sum of the ideal gas contribution, the real gas contribution, and the contribution from mixing. The contribution from mixing is given by a single generalized equation which is applied to all mixtures used in this work. The independent variables are the reduced density and reduced temperature. The model may be used to calculate the thermodynamic properties of mixtures at various compositions including dew and bubble-point properties and critical points. It incorporates the most accurate published equation of state for each pure fluid. The mixture model may be used to calculate the properties of mixtures generally within the experimental accuracies of the available measured properties. The estimated uncertainty of calculated properties is 0.1% in density, 0.2% in the speed of sound, and 1% in heat capacities. Calculated dew and bubble-point pressures are generally accurate to within 1%. © 2000 American Institute of Physics. [S0047-2689(00)00103-3]

Key words: air, argon, density, equation of state, mixtures, nitrogen, oxygen, pressure, thermodynamic properties, vapor-liquid equilibrium.

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a) Electronic mail: ericl@boulder.nist.gov

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| | Contents | | | experimental $p-\rho-1$ data of Romberg (19/1) | 340 |
|----|---|------------|-----|---|------|
| 1. | Introduction | 336 | 8. | Summary and statistical analysis of data for | |
| | 1.1. Background | 337 | | mixtures of nitrogen, argon, and oxygen | 341 |
| | 1.2. Prior Thermodynamic Property Formulations | | 9. | Maxcondentherm, maxcondenbar, and critical | |
| | for Air | 337 | | point for air | 342 |
| 2. | Experimental Data | 337 | 10. | Coefficients for the dew and bubble-point | |
| 3. | Vapor-Liquid Equilibria for Air | 342 | | pressure and density equations for air | 342 |
| | 3.1. Ancillary Equations | 342 | 11. | Averaged molar composition of standard air at | |
| | 3.2. Freezing Liquid Line | 343 | | specified temperatures | 343 |
| 4. | Equation of State for Air | 343 | | Coefficients for the ideal gas expressions | 345 |
| | 4.1. Effects of Dissociation. | 343 | 13. | Coefficients and exponents for the equation of | |
| | 4.2. Calculation of Air Properties from | 244 | | state for air | 346 |
| | Experimental Data for Nitrogen | 344 | 14. | Pure fluid equations of state used in the mixture | 2.40 |
| | 4.3. Fitting Procedures. | 344 | 1.5 | model | 349 |
| | 4.4. Ideal Gas Heat Capacity | 345 | | Parameters of the mixture model | 350 |
| | 4.5. Ideal Gas Helmholtz Energy | 345 | | Different types of VLE calculations | 351 |
| | 4.6. Equation of State for Air4.7. Calculation of Thermodynamic Properties | 346 347 | 1/. | Regions of stated uncertainty of the mixture | 260 |
| | 4.8. Hugoniot Curve. | 348 | Α 1 | model | 360 |
| 5. | Mixture Model for the Nitrogen–Argon–Oxygen | 340 | A1. | Thermodynamic properties of air on the dew and | 262 |
| ٥. | System | 349 | ۸.2 | bubble lines | 363 |
| | 5.1. Mixture Model | 349 | AZ. | Thermodynamic properties of air | 366 |
| | 5.2. Vapor–Liquid Equilibrium (VLE) Properties. | 350 | | List of Figures | |
| | 5.3. Critical Locus | 351 | 1 | List of Figures | |
| 6. | Comparisons of Calculated Properties to | 331 | 1. | Low temperature $p-\rho-T$ data for air. Solid lines represent the dew and bubble-point | |
| ٠. | Experimental Data | 352 | | curves for air | 338 |
| | 6.1. Comparisons of the Ancillary Equations for | 002 | 2 | High temperature $p-\rho-T$ data for air | 338 |
| | Air with Experimental Data | 352 | 3. | Isochoric and isobaric heat capacities and speed | 330 |
| | 6.2. Comparisons of the Equation of State for | | ٥. | of sound data for air. Solid lines represent the | |
| | Air and the Mixture Model with | | | dew and bubble-point curves for air | 340 |
| | Experimental and Calculated Data for Air | 354 | Δ | Derivation of second virial coefficient data from | 540 |
| | 6.3. Comparisons of the Mixture Model with | | т. | the $p-\rho-T$ data of Romberg (1971) | 340 |
| | Experimental Single Phase Data | 355 | 5. | Derivation of second virial coefficient data from | 510 |
| | 6.4. Comparisons of the Mixture Model with | | ٠. | the $p-\rho-T$ data of Michels <i>et al.</i> (1954a) | |
| | Experimental VLE Data | 356 | | (1954b) | 340 |
| 7. | Estimated Uncertainty of Calculated Properties | 357 | 6. | Critical region phase boundaries for air | 342 |
| | 7.1. Characteristic Curves of Air | 357 | 7. | Calculated Hugoniot curve for air | 348 |
| | 7.2. Uncertainty of the Equation of State for Air | | 8. | Critical lines for nitrogen-argon, nitrogen- | |
| | and of the Mixture Model | 359 | | oxygen, and argon-oxygen binary mixtures | 351 |
| 8. | References | 360 | 9. | Comparisons of dew and bubble-point properties | |
| 9. | Appendix—Tables of Properties of Air | 362 | | calculated with the ancillary equations to | |
| | 9.1 Representative Tables of Thermodynamic | 2.52 | | experimental and calculated data for air | 352 |
| | Properties of Air | 362 | 10. | Comparisons of densities calculated with the | |
| | | | | equation of state to experimental data for air | 353 |
| | List of Tables | | 11. | Comparisons of pressures calculated with the | |
| 1. | Summary of measured compositions of air | 336 | | equation of state to experimental data for air | |
| 2. | Composition of air using nitrogen, argon, | | | in the critical region | 354 |
| | oxygen, and carbon dioxide as constituents | 336 | 12. | Comparisons of isochoric heat capacities | |
| 3. | Composition of air using nitrogen, argon, and | | | calculated with the equation of state to | |
| | oxygen as constituents | 337 | | experimental data for air | 354 |
| 4. | Summary of prior formulations for the | 225 | 13. | Comparisons of isobaric heat capacities calculated | |
| _ | thermodynamic properties of air | 337 | | with the equation of state to experimental data | 255 |
| 5. | Summary and statistical analysis of saturation | 220 | 1.4 | for air | 355 |
| _ | data for air. | 338 | 14. | Comparisons of speeds of sound calculated with | 257 |
| 6. | Summary and statistical analysis of experimental | 220 | 1.5 | the equation of state to experimental data for air | 356 |
| 7 | data for air | 339 | 15. | Comparisons of second virial coefficients | |
| / | Second vidal coefficients derived from the | | | Carcinaled with the editation of state to | |

| 16. | experimental data for air | 356 | 21. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for the nitrogen—oxygen binary mixture | 358 |
|-----|--|-----|---|-----|
| 17 | data for air | 356 | 22. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for | |
| 17. | equation of state to calculated $p-\rho-T$ data | | | 358 |
| | estimated from nitrogen data | 357 | 23. Comparisons of bubble-point pressures calculated | |
| 18. | Comparisons of densities calculated with the | | with the mixture model to experimental data for | |
| | mixture model to experimental data for the | 257 | 76 | 359 |
| 10 | nitrogen–argon binary mixture | 357 | 24. Isochoric heat capacity versus temperature | 250 |
| 19. | Comparisons of densities calculated with the | | | 359 |
| | mixture model to experimental data for the nitrogen—oxygen and argon—oxygen binary | | 25. Isobaric heat capacity versus temperature diagram for air | 359 |
| | mixtures | 357 | 26. Speed of sound versus temperature diagram for | 337 |
| 20. | Comparisons of bubble-point pressures calculated | 337 | | 359 |
| 20. | with the mixture model to experimental data for | | | 359 |
| | the nitrogen-argon binary mixture | 358 | | 360 |

List of Symbols

| Symbol | Physical quantity | <u>Unit</u> |
|----------------------|---|----------------------|
| \overline{A} | Helmholtz energy | J |
| a | Molar Helmholtz energy | J/mol |
| B | Second virial coefficient | dm ³ /mol |
| B_s | Adiabatic bulk modulus | MPa |
| C | Third virial coefficient | $(dm^3/mol)^2$ |
| c_p | Isobaric heat capacity | $J/(mol \cdot K)$ |
| c_v^r | Isochoric heat capacity | $J/(mol \cdot K)$ |
| f | Fugacity | MPa |
| F | Mixture coefficient | |
| g | Gibbs energy | J/mol |
| h | Enthalpy | J/mol |
| k | Isentropic expansion coefficient | |
| K_T | Isothermal bulk modulus | MPa |
| k_T | Isothermal expansion coefficient | |
| M | Molar mass | g/mol |
| N | Coefficients of equations | |
| n | Number of moles | mol |
| p | Presssure | MPa |
| R | Universal gas constant (8.314510±0.000070) | $J/(mol \cdot K)$ |
| S | Entropy | $J/(mol \cdot K)$ |
| T | Temperature | K |
| и | Internal energy | J/mol |
| v | Molar volume | dm ³ /mol |
| V | Total volume | dm^3 |
| W | Speed of sound | m/s |
| X | Mixture composition array | |
| \boldsymbol{x} | Mole fraction (liquid mole fraction for VLE | |
| | calculations) | |
| у | Vapor mole fraction for VLE calculations | |
| Z | Compressibility factor | |
| α | Reduced Helmholtz energy | |
| $oldsymbol{eta}$ | Volume expansivity | 1/K |
| \boldsymbol{eta}_s | Adiabatic compressibility | 1/MPa |
| δ | Reduced density, $\delta = \rho/\rho_c$ or $\delta = \rho/\rho_j$ | |
| γ | Heat capacity ratio | |
| | | |

| κ | Isothermal compressibility | 1/MPa |
|-----------|---|----------------------|
| μ | Chemical potential | J/mol |
| μ_J | Joule-Thomson coefficient | K/MPa |
| ρ | Density | mol/dm^3 |
| au | Reduced temperature, $\tau = T_c/T$ or $\tau = T_j/T$ | |
| ω | Acentric factor | |
| ξ | Mixture coefficient for reduced density | dm ³ /mol |
| ζ | Mixture coefficient for reduced temperature | K |
| φ | Constant in melting line equation | |

| | Superscripts | i | Property of a component in the mixture |
|------------------|---------------------------|------|---|
| 0 | Ideal gas property | j | Maxcondentherm property |
| \boldsymbol{E} | Excess-like property | nbp | Normal boiling point property |
| f | Freezing property | nbpl | Normal boiling point liquid property |
| idmix | Ideal solution property | nbpv | Normal boiling point vapor property |
| r | Residual property | 0 | Initial state point for shock tube calculations |
| , | Liquid phase property | p | Maxcondenbar property |
| " | Vapor phase property | r | Reduced property |
| | Subscripts | red | Reducing property |
| 0 | Reference state property | S | Solidification point property |
| c | Critical point property | tp | Triple point property |
| calc | Calculated point property | tpl | Triple point liquid property |
| exp | Experimental value | tpv | Triple point vapor property |

Fixed Points for Air

| Symbol | Quantity | <u>Value</u> ^b |
|--|--|----------------------------|
| T_{j} | Maxcondentherm temperature | 132.6312 K |
| p_j | Maxcondentherm pressure | 3.78502 MPa |
| $ ho_j$ | Maxcondentherm density | $10.4477 \; mol/dm^3$ |
| T_p | Maxcondenbar temperature | 132.6035 K |
| p_p^r | Maxcondenbar pressure | 3.7891 MPa |
| $ ho_p^r$ | Maxcondenbar density | $11.0948 \; mol/dm^3$ |
| $T_c^{'}$ | Critical temperature | 132.5306 K |
| p_c | Critical pressure | 3.7860 MPa |
| $ ho_c$ | Critical density | $11.8308 \ mol/dm^3$ |
| T_s | Solidification point temperature | 59.75 K |
| p_s | Solidification point pressure | 0.005265 MPa |
| ρ_s | Solidification point density | 33.067 mol/dm^3 |
| $T_{\rm nbp}$ | Normal boiling point temperature ^a | 78.903 K |
| $ ho_{ m nbpl}$ | Normal boiling point density (liquid) ^a | $30.216~\mathrm{mol/dm^3}$ |
| M | Molar mass | 28.9586 ± 0.0002 g/mol |
| T_0 | Reference temperature | 298.15 K |
| | Reference pressure | 0.101 325 MPa |
| h_0^0 | Reference enthalpy at T_0 | 8649.34 J/mol |
| $egin{array}{c} p_0 \ h_0^0 \ s_0^0 \end{array}$ | Reference entropy at T_0 and p_0 | 194.0 J/(mol·K) |
| 957 11 111 | | 0.404.00# 3.50 |

^aNormal boiling point properties are calculated on the bubble-point curve at 0.101 325 MPa.

Fixed Points for Nitrogen

| Symbol | Quantity | <u>Value</u> ^a |
|-------------------|--------------------------|----------------------------|
| T_c | Critical temperature | 126.192 K |
| p_c | Critical pressure | 3.3958 MPa |
| $ ho_c$ | Critical density | 11.1839 mol/dm^3 |
| T_{tp} | Triple point temperature | 63.151 K |

^bUncertainties are reported at the 2σ confidence level; those for the critical parameters are given in Table 9.

| p_{tp} | Triple point pressure | 0.012 523 MPa |
|--|---------------------------------------|----------------------------|
| $ ho_{ m tpv}^{^{1}}$ | Triple point density (vapor) | $0.02407\mathrm{mol/dm^3}$ |
| $ ho_{ m tpl}$ | Triple point density (liquid) | 30.957 mol/dm^3 |
| $T_{\rm nbp}$ | Normal boiling point temperature | 77.355 K |
| $ ho_{ m nbpv}$ | Normal boiling point density (vapor) | $0.1646\ mol/dm^3$ |
| $ ho_{ m nbpl}$ | Normal boiling point density (liquid) | 28.775 mol/dm^3 |
| ω | Acentric factor | 0.037 |
| M | Molar mass | 28.013 48 g/mol |
| T_0 | Reference temperature | 298.15 K |
| | Reference pressure | 0.101 325 MPa |
| h_0^0 | Reference enthalpy at T_0 | 8670.0 J/mol |
| $egin{array}{c} p_{0} \\ h^{0}_{0} \\ s^{0}_{0} \end{array}$ | Reference entropy at T_0 and p_0 | 191.5 J/(mol·K) |

a Reference entropy at T_0 and p_0 191.5 J/(mol·K) a Span *et al.* (2000) report the constants given here and their associated uncertainties; temperatures for nitrogen are given on ITS-90.

Fixed Points for Argon

| Symbol | Quantity | <u>Value</u> ^a |
|--|---------------------------------------|----------------------------|
| T_c | Critical temperature | 150.687 K |
| p_c | Critical pressure | 4.863 MPa |
| $ ho_c$ | Critical density | 13.407 mol/dm^3 |
| T_{tp} | Triple point temperature | 83.8058 K |
| p_{tp} | Triple point pressure | 0.068891 MPa |
| $ ho_{ m tpv}$ | Triple point density (vapor) | 0.1015 mol/dm^3 |
| $ ho_{ m tpl}$ | Triple point density (liquid) | 35.465 mol/dm^3 |
| $T_{\rm nbp}$ | Normal boiling point temperature | 87.302 K |
| $ ho_{ m nbpv}$ | Normal boiling point density (vapor) | 0.1445 mol/dm^3 |
| $ ho_{ m nbpl}$ | Normal boiling point density (liquid) | $34.930~\mathrm{mol/dm^3}$ |
| ω | Acentric factor | -0.002 |
| M | Molar mass | 39.948 g/mol |
| T_0 | Reference temperature | 298.15 K |
| | Reference pressure | 0.101 325 MPa |
| h_0^0 | Reference enthalpy at T_0 | 6197.0 J/mol |
| $egin{array}{c} p_0 \ h_0^0 \ s_0^0 \end{array}$ | Reference entropy at T_0 and p_0 | 154.737 J/(mol·K) |

 s_0° Reference entropy at I_0 and p_0 154./3/ J/(mol·K) ^aTegeler *et al.* (1999) report the constants given here and their associated uncertainties; temperatures for argon are given on ITS-90.

Fixed Points for Oxygen

| Symbol | Quantity | <u>Value</u> ^a | Value (ITS-90) |
|--------------------------------|---------------------------------------|------------------------------|----------------|
| T_c | Critical temperature | 154.581 K | 154.595 K |
| p_c | Critical pressure | 5.043 MPa | |
| $ ho_c$ | Critical density | 13.63 mol/dm^3 | |
| T_{tp} | Triple point temperature | 54.361 K | 54.359 K |
| $p_{\mathrm{tp}}^{\mathrm{T}}$ | Triple point pressure | 0.000 146 3 MPa | |
| $ ho_{ m tpv}$ | Triple point density (vapor) | $0.0003237\mathrm{mol/dm^3}$ | |
| $ ho_{ m tpl}$ | Triple point density (liquid) | $40.816\ \mathrm{mol/dm^3}$ | |
| $T_{\rm nbp}$ | Normal boiling point temperature | 90.188 K | 90.196 K |
| $ ho_{ m nbpv}$ | Normal boiling point density (vapor) | $0.1396~\mathrm{mol/dm^3}$ | |
| $ ho_{ m nbpl}$ | Normal boiling point density (liquid) | 35.663 mol/dm^3 | |
| ω | Acentric factor | 0.022 | |
| M | Molar mass | 31.9988 g/mol | |
| T_0 | Reference temperature | 298.15 K | |
| p_0 | Reference pressure | 0.101 325 MPa | |
| h_0^0 | Reference enthalpy at T_0 | 8680.0 J/mol | |
| h_0^0 s_0^0 | Reference entropy at T_0 and p_0 | 205.043 J/(mol·K) | |

^aSchmidt and Wagner (1985) report the constants given here and their associated uncertainties; temperatures for oxygen are given on IPTS-68 consistent with the equation of state, and on ITS-90 in the last column.

1. Introduction

1.1. Background

The measurement of experimental data and the development of equations for the thermophysical properties of air and mixtures of nitrogen, argon, and oxygen has been a continuing project at the Center for Applied Thermodynamic Studies at the University of Idaho and the National Institute of Standards and Technology for more than 10 years. The experimental measurements on air are summarized by Haynes et al. (1998). Equations of state for air as a pseudopure fluid were published by Jacobsen et al. (1990a), (1992). In addition, an extended corresponding states model for calculating the thermodynamic properties of nitrogenargon-oxygen mixtures was published by Clarke et al. (1994), and a Lagrangian interpolation model for calculating vapor-liquid equilibrium properties for this system was published by Lemmon et al. (1992). A preliminary equation of state for air valid to temperatures of 2000 K and pressures to 2000 MPa was published by Panasiti et al. (1999), and a new model for mixtures of nitrogen, argon, and oxygen was reported by Lemmon and Jacobsen (1998), (1999). The equations of state for air of Jacobsen et al. (1990a), (1992) were reported on the International Practical Temperature Scale of 1968 (IPTS-68). Version 1.0 of the NIST standard reference database for air [Lemmon (1998)] was based on the equation of state of Jacobsen et al. (1992). The new equation of state and mixture model presented here are based on the International Temperature Scale of 1990 (ITS-90).

Atmospheric air is a mixture of fluids including nitrogen, oxygen, argon, carbon dioxide, water vapor, and other trace elements. The standard air considered in this report is dry and contains no carbon dioxide or trace elements. The composition of air used here was reported by Olien (1987) based on the work of Jones (1978). It is consistent with that of the U.S. Standard Atmosphere (1976). Other compositions are given by Giacomo (1982) and Waxman and Davis (1978). The mole fractions of nitrogen, oxygen, argon, and carbon dioxide in air from each of the references listed above are given in Table 1.

In the analysis of Olien (1987), the values of Waxman and Davis (1978) were excluded because they are based on a purified, rather than natural, sample. The difference between the U.S. standard atmosphere (1976) and values from Jones (1978) is that Jones used a more accurate value for the composition of argon. The differences between Giacomo (1982)

TABLE 1. Summary of measured compositions of air

| | U.S. Std. Atmosphere (1976) | Jones (1978) | Waxman and Davis (1978) | Giacomo (1982) |
|--|-------------------------------------|----------------------------------|----------------------------------|----------------------------------|
| $egin{array}{c} \mathbf{N}_2 \ \mathbf{O}_2 \end{array}$ | 0.780 840 0.209 476 | 0.781 02 0.209 46 | 0.781 20 0.209 20 | 0.781 01 0.209 39 |
| $\begin{array}{c} \text{Ar} \\ \text{CO}_2 \end{array}$ | 0.009 340 0.000 314 0.999 970 | 0.009 16 0.000 33 0.999 97 | 0.009 30 0.000 32 1.000 02 | 0.009 17 0.000 40 0.999 97 |

of BIPM (International Bureau of Weights and Measures) and the U.S. standard atmosphere are greater than those for values from Jones. Although the argon value given by Giacomo is essentially the same as that given by Jones, the carbon dioxide value given by Giacomo is for a laboratory setting in which human respiration generally increases carbon dioxide concentration and decreases oxygen concentration. Therefore, the compositions given by Giacomo were not used. Rather, the compositions given by Jones, truncated to four significant digits as indicated in Table 2, were used. In this work, the concentration of carbon dioxide is assumed negligible. The normalized values based upon this assumption are given in Table 3. An analysis of experimental measurements on air at different compositions than that reported here showed that the change caused by the difference in composition was less than the experimental error in the measurements, and hence, no effort was made to transform measurements at different air compositions to the composition reported here.

A property formulation is the set of equations used to calculate properties of a fluid at specified thermodynamic states defined by an appropriate number of independent variables. The term "fundamental equation" is often used in the literature to refer to empirical descriptions of one of four fundamental relations: internal energy, enthalpy, Gibbs energy, and Helmholtz energy. The formulations for fluid properties are often explicit in Helmholtz energy with independent variables of temperature and density. Pure fluids obey the Maxwell criterion (equal pressures and Gibbs energies at constant temperatures during phase changes) so that all thermodynamic properties including the vapor-liquid equilibrium may be calculated from the full expression for the Helmholtz energy without additional ancillary equations. In this work, the general term "equation of state" rather than the term "fundamental equation" is used to refer to the empirical models developed for calculating fluid properties.

Since air is not a pure fluid, general phase equilibrium calculations require consideration of phase compositions. This type of model is discussed in Sec. 5 for arbitrary mixtures of nitrogen, argon, and oxygen. For standard air at a fixed composition, two separate equations are required in addition to the equation of state to represent the dew and bubble-point pressures as functions of temperature. Densities along the dew and bubble-point curves are calculated by the solution of the equation of state and the dew or bubble-point pressure equations given in Sec. 3 at a given temperature. Ancillary equations for the dew and bubble-point densities

TABLE 2. Composition of air using nitrogen, argon, oxygen, and carbon dioxide as constituents

| Component | Mole fraction |
|----------------|---------------|
| N ₂ | 0.7810 |
| O_2 | 0.2095 |
| Ar | 0.0092 |
| CO_2 | 0.0003 |
| _ | 1.0000 |

TABLE 3. Composition of air using nitrogen, argon, and oxygen as constituents

| Component | Mole fraction | |
|-----------|-------------------------|--|
| N_2 | 0.7812 | |
| O_2 | 0.2096 | |
| Ar | $\frac{0.0092}{1.0000}$ | |

are also given and can be used as estimating functions for iterative calculations of derived thermodynamic properties.

1.2. Prior Thermodynamic Property Formulations for Air

Several previous thermodynamic property formulations for air are referenced in Table 4. The most recent prior widerange formulations are those of Jacobsen and co-workers [Jacobsen et al. (1990a); Jacobsen et al. (1992); Panasiti et al. (1999)], the first of which is based upon liquid heat capacities predicted using extended corresponding states methods and the second of which used preliminary measurements of the isochoric heat capacity from the National Institute of Standards and Technology (NIST), subsequently published by Magee (1994). The third formulation was a preliminary equation extended to temperatures and pressures of 2000 K and 2000 MPa using predicted properties above 870 K and 70 MPa calculated by corresponding states methods from experimental nitrogen data.

2. Experimental Data

The estimated uncertainty of reference equations of state for pure fluids or mixtures is generally based upon comparisons to experimental data. These data are used to determine the coefficients of the equation and to evaluate the behavior of the equation of state over the fluid surface. Data types used in this work include $p-\rho-T$, speed of sound, isochoric and isobaric heat capacities, and vapor-liquid equilibrium. The equation of state reported here was developed using least squares fitting methods described in detail in Sec. 4.3. For reference quality equations, extensive comparisons to all data types are required. In the tables included in this section, columns are included which indicate the results of statistical analyses of the comparisons of calculated values to experimental data. The details of this analysis are given in Sec. 6.

The available saturation data are summarized in Table 5 and discussed in Sec. 6.1. Data used in the fit of the ancillary equations are shown as bolded entries in the table. The values reported by Blanke (1977) were obtained by graphical methods from his $p-\rho-T$ data; several of the near critical temperature data were used in the fit of ancillary equations. Values calculated using the Leung–Griffiths model for air as reported by Jacobsen *et al.* (1990b) are also summarized in this table. Several of these values close to the critical point were used in the fit.

TABLE 4. Summary of prior formulations for the thermodynamic properties of air

| Author | Year | Temperature range (K) | High pressure limit (MPa) |
|-----------------------------------|------|-----------------------|------------------------------|
| Hilsenrath (1955) | 1955 | 50-3000 | 10 |
| Michels et al. (1955) | 1955 | 102-348 | 122 |
| Din (1962) | 1962 | 90-450 | 122 |
| Baehr and Schwier (1961) | 1961 | 60-1250 | 450 |
| Vasserman and Rabinovich (1970) | 1970 | $75-160^{a}$ | 50 |
| Vasserman et al. (1971) | 1971 | $75-1300^{b}$ | 100 |
| Sychev et al. (1987) ^c | 1978 | 70-1500 | 100 |
| Jacobsen et al. (1990a) | 1990 | 60-873 | 70 |
| Jacobsen et al. (1992) | 1992 | 60-873 | 70 |
| Panasiti et al. (1999) | 1999 | 60-2000 | 2000 |

aLiquid states only.

Experimental $p-\rho-T$ data for air are illustrated in Figs. 1 and 2 and summarized in Table 6. The solid lines represent the dew and bubble-point curves. Data used in the fit of the equation of state are shown as bolded entries in the table. All available experimental data were considered in preliminary analysis, but in the final regression of the coefficients of the equation of state, only the most accurate data and data in regions where no other data were available were fitted. The air $p-\rho-T$ data sets used in the development of the equation of state for air and for the mixture model were those of Blanke (1977), Howley et al. (1994), Kozlov (1968), Michels et al. (1954a), (1954b), and Romberg (1971). The isochoric and isobaric heat capacity data are illustrated in Fig. 3 and summarized in Table 6. The speed of sound data of Younglove and Frederick (1992), Ewing and Goodwin (1993) and of Van Itterbeek and de Rop (1955) are also illustrated in Fig. 3 and summarized in Table 6. These speed of sound data and the isochoric heat capacity data of Magee (1994) were used in the fit. The critical region data of Chashkin et al. (1966) are not included in this table since their measurements were concerned with investigating the effect of impurities on the behavior of the isochoric heat capacity in the critical region, and the sample had a 1.2% impurity content [see Sychev et al. (1987)]. None of the isobaric heat capacity data summarized in Table 6 were used to develop the equation of state or mixture model because of the age of the data and the availability of other more accurate derived data (isochoric heat capacity and speed of sound) in similar regions.

A summary of sources of second virial coefficients, including those of Romberg (1971) and Michels *et al.* (1954a), (1954b), are given in Table 6. The values of Michels *et al.* were used in the fit of the equation of state. In addition to the values reported by Romberg, values of the second virial coefficients were redetermined graphically from his reported $p-\rho-T$ data. These values are given in Table 7. At the higher isotherms, data below 0.5 mol/dm³ were not used in this graphical redetermination of the values of the second

^bVapor states only.

^cEnglish translation of original work published in 1978.

TABLE 5. Summary and statistical analysis of saturation data for air

| Author | No. of points | Temp. | Pressure range (MPa) | $\mathrm{AAD^a}$ | AAD^b |
|--|---------------------|---|---|---|---|
| Bubble-point pressure | | | | | |
| Blanke (1977) | 9 | 60-129 | 0.01-3.2 | 2.506 | 2.766 |
| Jacobsen et al. (1990b) | 18 | 120-133 | 2.10-3.8 | 0.049 | 0.207 |
| Kuenen and Clark (1917) | 14 | 123-133 | 2.54 - 3.8 | 0.663 | 0.839 |
| Michels et al. (1954a) | 3 | 118-132 | 1.96 - 3.7 | 0.069 | 0.212 |
| Overall | 44 | 60-133 | 0.01 - 3.8 | 0.748 | 1.040 |
| Dew-point pressure | | | | | |
| Blanke (1977) | 11 | 67-132 | 0.01-3.7 | 0.395 | 1.407 |
| Jacobsen et al. (1990b) | 12 | 122-133 | 2.18 - 3.8 | 0.048 | 0.293 |
| Kuenen and Clark (1917) | 18 | 123-133 | 2.40 - 3.8 | 0.587 | 0.548 |
| Michels et al. (1954a) | 6 | 118-133 | 1.83-3.8 | 0.413 | 0.515 |
| Overall | 47 | 67-133 | 0.01 - 3.8 | 0.382 | 0.680 |
| | No. | | Density | | |
| | of points | Temp. range (K) | range (mol/dm ³) | $\mathrm{AAD}^{\mathrm{a}}$ | AAD^{c} |
| Bubble-point density | | | | | |
| Blanke (1977) | 9 | 60-129 | 17.71-33.0 | 0.144 | 0.152 |
| Jacobsen et al. (1990b) | | | | | |
| | 18 | 120-133 | 10.45-21.6 | 0.786 | 0.745 |
| Kuenen and Clark (1917) | 18 11 | 120-133 127-133 | 10.45-21.6 10.70-18.1 | 0.786 6.042 | 0.745 6.039 |
| Kuenen and Clark (1917) Michels et al. (1954a) | | | | | |
| * * * | 11 | 127–133 | 10.70-18.1 | 6.042 | 6.039 |
| Michels et al. (1954a) | 11 10 | 127–133 118–132 | 10.70–18.1 13.67–20.5 | 6.042 3.003 | 6.039 3.034 |
| Michels et al. (1954a) Overall | 11 10 | 127–133 118–132 | 10.70–18.1 13.67–20.5 | 6.042 3.003 | 6.039 3.034 |
| Michels et al. (1954a) Overall Dew-point density | 11 10 48 | 127–133 118–132 60–133 | 10.70–18.1 13.67–20.5 10.45–33.0 | 6.042 3.003 2.332 | 6.039 3.034 2.324 |
| Michels et al. (1954a) Overall Dew-point density Blanke (1977) | 11 10 48 | 127–133 118–132 60–133 | 10.70–18.1 13.67–20.5 10.45–33.0 0.02–8.9 | 6.042 3.003 2.332 | 6.039 3.034 2.324 0.639 |
| Michels et al. (1954a) Overall Dew-point density Blanke (1977) Jacobsen et al. (1990b) | 11 10 48 | 127–133 118–132 60–133 67–132 122–133 | 10.70–18.1 13.67–20.5 10.45–33.0 0.02–8.9 3.40–10.4 | 6.042 3.003 2.332 0.638 1.028 | 6.039 3.034 2.324 0.639 1.036 |

^aComparisons of the ancillary equations with experimental data.

^cComparisons of the equation of state for air with experimental data.

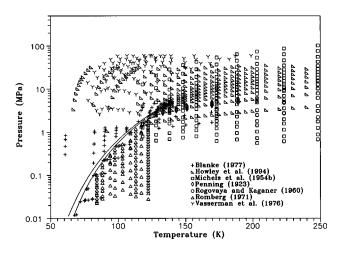


Fig. 1. Low temperature $p-\rho-T$ data for air. Solid lines represent the dew and bubble-point curves for air.

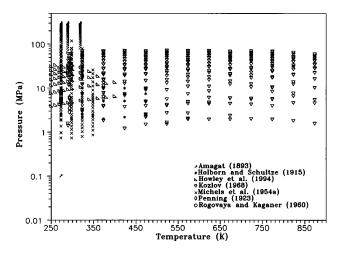


Fig. 2. High temperature $p-\rho-T$ data for air.

^bComparisons of the mixture model with experimental data.

THERMODYNAMIC PROPERTIES OF AIR

TABLE 6. Summary and statistical analysis of experimental data for air

| A . # | No. of | Temp. | Pressure range | Density range | AAD ^a | AADa (mixtur |
|---------------------------------|-----------|-----------|-------------------|------------------------|------------------|-----------------|
| Author | points | range (K) | (MPa) | (mol/dm ³) | (air EOS) | model) |
| $p-\rho-T$ | | | | | | |
| Amagat (1893) | 78 | 273-318 | 0.10 - 304.0 | 0.04-31.0 | 0.220 | 0.246 |
| Blanke (1977) | 110 | 61 - 170 | 0.01-4.9 | 0.02 - 33.0 | 0.350 | 0.312 |
| Holborn and Schultze (1915) | 42 | 273-473 | 1.98-10.0 | 0.61-4.5 | 0.019 | 0.031 |
| Howley et al. (1994) | 286 | 67-400 | 1.06 - 35.2 | 1.95 - 32.2 | 0.050 | 0.044 |
| Kozlov (1968) | 348 | 288-873 | 1.21 - 72.2 | 0.21-17.2 | 0.072 | 0.069 |
| Michels et al. (1954a) | 157 | 273-348 | 0.73 - 228.0 | 0.30 - 28.7 | 0.022 | 0.030 |
| Michels et al. (1954b) | 199 | 118-248 | 0.56 - 102.0 | 0.33 - 25.2 | 0.147 | 0.152 |
| Penning (1923) | 62 | 128-293 | 2.52 - 6.20 | 1.20 - 4.5 | 0.094 | 0.100 |
| Rogovaya and Kaganer (1960) | 10 | 273 | 4.71 - 8.6 | 2.12 - 3.9 | 0.091 | 0.102 |
| Romberg (1971) | 124 | 84-122 | 0.02 - 2.0 | 0.03 - 2.7 | 0.028 | 0.045 |
| Vasserman et al. (1976) | 109 | 77-199 | 2.62 - 59.8 | 22.82-31.4 | 0.078 | 0.080 |
| Overall | 1525 | 61 - 873 | 0.01 - 304.0 | 0.02 - 33.0 | 0.097 | 0.097 |
| c_v | | | | | | |
| Eucken and Hauck (1928) | 16 | 138-165 | 6.72-19.6 | 18.08-19.0 | 6.571 | 6.063 |
| Henry (1931) | 3 | 288-623 | 0.10 | 0.02 | 0.579 | 0.580 |
| Magee (1994) | 227 | 66-299 | 1.71-34.6 | 2.04-33.0 | 0.373 | 0.493 |
| Overall | 246 | 66-623 | 0.10 - 34.6 | 0.02 - 33.0 | 0.779 | 0.856 |
| c_p | | | | | | |
| Bridgeman (1929) | 51 | 273-553 | 2.03-22.3 | 0.44-8.6 | 0.920 | 0.916 |
| Eucken (1913) | 6 | 271-480 | 0.10 | 0.03 | 0.531 | 0.532 |
| Holborn and Jakob (1917) | 7 | 333 | 0.10-29.4 | 0.04-9.5 | 0.434 | 0.405 |
| Jakob (1923) | 47 | 194-523 | 0.10-29.4 | 0.02 - 14.6 | 1.083 | 1.083 |
| Nesselmann (1925) | 19 | 194-523 | 4.90-19.6 | 1.11-14.6 | 1.845 | 1.810 |
| Poferl <i>et al.</i> (1959) | 46 | 300-2499 | 0.30-1.0 | 0.01-0.4 | 3.656 | 3.511 |
| Overall | 176 | 194-2499 | 0.10-29.4 | 0.01-14.6 | 1.746 | 1.702 |
| w | | | | | | |
| Abbey and Barlow (1948) | 6 | 293 | < 0.1 | < 0.04 | 0.388 | 0.389 |
| Colwell and Gibson (1941) | 7 | 273 | < 0.1 | < 0.04 | 0.100 | 0.100 |
| Colwell <i>et al.</i> (1938) | 9 | 297-299 | 0.10 | 0.04 | 0.063 | 0.062 |
| Ewing and Goodwin (1993) | 13 | 255 | 0.03-6.9 | 0.01 - 3.4 | 0.018 | 0.006 |
| Hardy et al. (1942) | 7 | 273-297 | 0.10 | 0.04 | 1.858 | 1.861 |
| King and Partington (1930) | 11 | 1138-1440 | 0.10 | 0.01 | 1.941 | 1.946 |
| Quigley (1945) | 29 | 92-259 | 0.10 | 0.05 - 0.1 | 0.313 | 0.306 |
| Shilling and Partington (1928) | 28 | 273-1572 | 0.10 | 0.01 | 1.405 | 1.407 |
| Tucker (1943) | 6 | 292-347 | 0.07-0.1 | 0.02 | 2.353 | 2.351 |
| Van Itterbeek and de Rop (1955) | 44 | 229-313 | 0.10-1.3 | 0.04 - 0.7 | 0.216 | 0.211 |
| Younglove and Frederick (1992) | 169 | 90-300 | 0.34-13.8 | 0.25-29.7 | 0.216 | 0.105 |
| Overall | 329 | 90-1572 | <13.8 | <29.7 | 0.446 | 0.387 |
| B | | | | | | |
| Andersen (1950) | 6 | 273-473 | | | 1.008 | 0.732 |
| Friedman (1957) | 5 | 150-273 | | | 0.819 | 0.558 |
| Hilsenrath (1955) | 61 | 50-1501 | | | 3.779 | 2.719 |
| Levelt Sengers et al. (1972) | 54 | 100-1400 | | | 0.674 | 0.472 |
| Michels et al. (1954a) | 4 | 273-348 | | | 0.220 | 0.357 |
| Michels et al. (1954b) | 10 | 118-248 | | | 0.127 | 0.159 |
| Romberg (1971) | 39 | 84–473 | | | 0.240 | 1.130 |
| Overall | 179 | 50-1501 | | | 1.588 | 1.357 |

^aAAD—Average absolute percent deviation in density for $p-\rho-T$ and average absolute difference in B (cm³/mol) for the second virial coefficients.

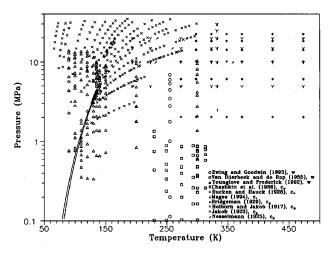


Fig. 3. Isochoric and isobaric heat capacities and speed of sound data for air. Solid lines represent the dew and bubble-point curves for air.

virial coefficients since in many apparatus, such low density data are subject to local adsorption by the walls of the apparatus or to higher uncertainties in the measurement of extremely low pressures. No value of the second virial coefficient was determined for the 84 K isotherm. The $p-\rho-T$ data used to generate the second virial coefficients are shown in Fig. 4. The solid lines show isotherms calculated from the equation of state presented here and the solid curve represents the dew-point curve. The y intercept (zero density) represents the second virial coefficient at a given temperature, and the third virial coefficients can be taken from the slope of each line at zero density. The values of the second virial coefficient calculated from the equation of state are in good agreement with those presented in Table 7 and shown as circles in the figure. Values of the second virial coefficients and $p-\rho-T$ data from Michels et al. are shown in a similar fashion in Fig. 5. The circles at zero density represent the second virial coefficients reported by Michels.

Binary $p-\rho-T$ and vapor-liquid equilibrium data for mixtures containing nitrogen, argon, and oxygen are summarized in Table 8. The composition ranges listed in these tables show the composition of the first component listed. All available experimental data were considered in the preliminary analysis, but the final regression of the coefficients of the mixture model used the data sets indicated by the

Table 7. Second virial coefficients derived from the experimental $p-\rho-T$ data of Romberg (1971)

| Temperature (K/ITS-90) | Graphical redetermination of the second virial coefficient (dm³/mol) |
|------------------------|--|
| 88.27 | -0.219 |
| 94.06 | -0.194 |
| 97.56 | -0.179 |
| 103.24 | -0.160 |
| 109.10 | -0.143 |
| 113.05 | -0.134 |
| 117.12 | -0.125 |
| 122.22 | -0.115 |

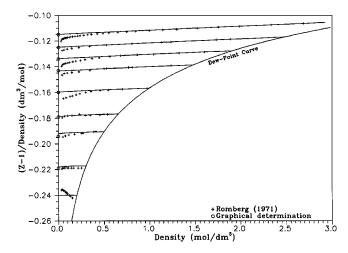


Fig. 4. Derivation of second virial coefficient data from the $p-\rho-T$ data of Romberg (1971).

bolded entries in Table 8. Data of Crain and Sonntag (1966), Palavra (1979), and Kosov and Brovanov (1979) were used for the nitrogen-argon mixture, and data of Pool et al. (1962) were used for the nitrogen-oxygen and argonoxygen mixtures. In addition to the limited data for the nitrogen-oxygen system, the following data sets for air were also used in the mixture modeling: Blanke (1977), Howley et al. (1994), Michels et al. (1954a), (1954b), Romberg (1971), Magee (1994), Ewing and Goodwin (1993), and Younglove and Frederick (1992). The largest impact from the addition of these data sets was on the nitrogen-oxygen parameters, however, the argon-oxygen parameters were also influenced to some degree by the air data. These data had little impact on the nitrogen-argon parameters, as there were sufficient binary data for this system to model the parameters well. The VLE data used in the model development included the nitrogen-argon data of Hiza et al. (1999), the nitrogen-oxygen data of Duncan and Staveley (1966), and the argon-oxygen data of Burn and Din (1962) and of Wilson et al. (1965). Comparisons with ternary VLE data indi-

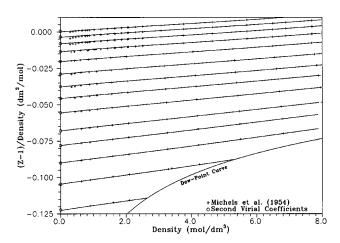


Fig. 5. Second virial coefficient data and $p-\rho-T$ data of Michels *et al.* (1954a), (1954b).

Table 8. Summary and statistical analysis of data for mixtures of nitrogen, argon, and oxygen

| | No. of | Pressure range | Density range | Temp. | Comp. | |
|---|---|--|---|---|---|--|
| Author | points | (MPa) | (mol/dm ³) | range (K) | (mol) | AAD^a |
| N_2 -Ar — $p-\rho-T$ | | | | | | |
| Crain and Sonntag (1966) Holst and Hamburger (1916) Kosov and Brovanov (1979) Maslennikova et al. (1979) Massengill and Miller (1973) Palavra (1979) Pool et al. (1962) Ricardo et al. (1992) Townsend (1956) Zandbergen and Beenakker (1967) Overall N ₂ -Ar — VLE | 264 41 201 88 6 203 24 21 144 55 1047 | 0.19-52.5 0.01-0.2 5.93-58.8 100800. 0.27-1.3 0.95-24.9 0.10-0.2 7.41-145. 0.18-13.9 0.76-9.6 0.01-800. | 0.08-26.6 0.01-0.2 2.46-18.8 17.5-41.1 25.7-30.2 25.8-32.9 28.3-33.9 25.8-34.7 0.07-5.7 0.55-13.1 0.01-41.1 | 143–273 74–90 293–353 298–423 90–113 94–106 84 119 298–323 171–293 74–423 | 0.20-0.80 0.31-0.80 0.16-0.81 0.26-0.75 0.50 0.32-0.70 0.18-0.89 0.52 0.16-0.84 0.19-0.74 0.16-0.89 | 0.209 0.193 0.363 0.320 0.409 0.021 0.502 0.160 0.054 0.097 |
| Elshayal and Lu (1975) Fastovskii and Petrovskii (1956) Hiza et al. (1999) Jin et al. (1993) Lewis and Staveley (1975) Miller et al. (1973) Narinskii (1966) Pool et al. (1962) Sprow and Prausnitz (1966) Thorpe (1968) Wilson et al. (1965) Overall N_2 - O_2 — p - ρ - T | 7 56 44 13 8 14 98 12 17 68 176 513 | 0.39-0.7 0.12-0.4 0.08-0.8 1.48-2.8 0.07-0.2 0.84-1.5 0.13-2.3 0.07-0.2 0.08-0.2 0.13-1.1 0.10-2.6 0.07-2.8 | | 100 79-103 85-100 123 85 112 80-120 84 84 81-115 72-134 72-134 | 0.14-0.90 0.05-0.90 0.10-0.89 0.05-0.97 0.23-0.81 0.11-0.85 0.02-0.96 0.06-0.88 0.06-0.91 0.10-0.91 0.05-1.00 0.02-1.00 | 0.468 1.182 0.379 0.555 0.840 0.326 0.484 0.420 0.460 0.830 0.805 0.708 |
| Blagoi and Rudenko (1958) Knaap et al. (1961) Kuenen et al. (1922) Pool et al. (1962) Overall N ₂ -O ₂ — VLE | 19 7 153 7 186 | 2.93-6.0 0.09-0.2 0.09-6.0 | 30.7-38.0 29.6-34.8 1.22-19.4 29.8-34.1 1.22-38.0 | 67–79 77 135–293 84 67–293 | 0.11-0.80 0.29-0.91 0.25-0.50 0.25-0.74 0.11-0.91 | 0.415 0.097 0.877 0.088 0.771 |
| Armstrong et al. (1955) Cockett (1957) Din (1960) Dodge and Dunbar (1927) Duncan and Staveley (1966) Hiza et al. (1999) Pool et al. (1962) Thorogood and Haselden (1963) Wilson et al. (1965) Yorizane et al. (1978) Overall Ar-O ₂ — p-ρ-T | 70 62 108 49 11 65 11 13 138 20 547 | 0.003-0.1 0.12-0.1 0.11-1.0 0.06-3.0 0.002-0.01 0.001-0.8 0.05-0.2 0.10 0.10-2.6 0.05-0.1 0.001-3.0 | | 65-78 81-91 79-116 77-125 63 63-100 84 88-90 78-136 80 63-136 | 0.03-0.94 0.07-0.81 0.10-0.89 0.05-0.91 0.10-0.80 0.07-0.88 0.10-0.90 0.002-0.08 0.05-0.99 0.11-0.85 0.002-0.99 | 0.870 1.287 0.788 0.903 1.524 0.689 0.310 1.320 0.540 2.489 0.865 |
| Blagoi and Rudenko (1958) Knaap et al. (1961) Pool et al. (1962) Saji and Okuda (1965) Overall Ar-O ₂ — VLE | 36 6 15 20 77 | 0.05 _b -0.1 0.05-0.1 | 34.8–38.2 34.7–35.2 34.7–36.3 34.9–36.1 34.7–38.2 | 70–89 90 84–90 85–87 70–90 | 0.10-0.87 0.28-0.79 0.17-0.86 0.25-0.92 0.10-0.92 | 0.274 0.052 0.025 0.244 0.200 |
| Bourbo and Ischkin (1936) Burn and Din (1962) Clark et al. (1954) Fastovskii and Petrovskii (1955) Hiza et al. (1999) Narinskii (1957) Parikh and Zollweg (1997) Pool et al. (1962) Wang (1960) Wilson et al. (1965) Yorizane et al. (1978) Overall N ₂ -Ar-O ₂ — VLE | 27 140 55 24 16 55 24 24 35 200 58 568 | 0.07-0.2 0.06-1.0 0.10-0.7 0.12-0.2 0.16-0.3 0.11-1.2 0.12-0.9 0.05-0.1 0.12-0.2 0.10-2.6 0.10 0.05-2.6 | | 87-95 85-118 90-110 89-96 95-100 90-120 92-115 84-90 90-96 87-139 89-92 84-139 | $\begin{array}{c} 0.04 - 0.86 \\ 0.10 - 0.91 \\ 0.10 - 0.90 \\ 0.21 - 0.83 \\ 0.11 - 0.79 \\ 0.03 - 0.96 \\ 0.01 - 0.92 \\ 0.10 - 0.90 \\ 0.02 - 1.00 \\ 0.003 - 0.98 \\ < 0.76 \\ < 1.00 \\ \end{array}$ | 0.710 0.232 0.313 0.774 0.805 0.237 1.149 0.168 1.393 0.403 0.857 0.493 |
| Fastovskii and Petrovskii (1957) Funada et al. (1982) Narinskii (1969) Weishaupt (1948) Wilson et al. (1965) Overall | 14 60 115 41 1427 1657 | 0.1 0.1 0.13-2.2 0.1 0.10-2.6 0.10-2.6 | | 81–88 90 82–120 81–92 78–136 78–136 | $\begin{array}{c} 0.13 - 0.68 \\ < 0.004 \\ 0.04 - 0.86 \\ 0.01 - 0.89 \\ 0.001 - 0.99 \\ < 0.99 \end{array}$ | 0.555 0.278 0.489 1.613 0.561 0.572 |

^aAAD—Average absolute deviation in density for $p-\rho-T$ and average absolute deviation in bubble-point pressure for VLE.

^bNo pressure reported, bubble-point pressure assumed.

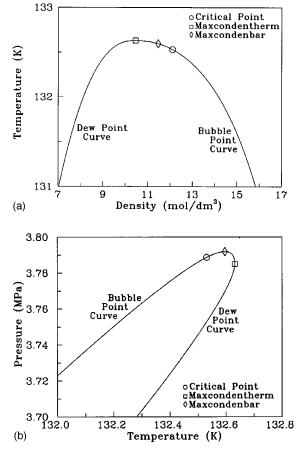


Fig. 6. Critical region phase boundaries for air.

cated that the binary mixture parameters were sufficient for modeling the ternary mixture interactions.

3. Vapor-Liquid Equilibria for Air

The maxcondentherm (state point of maximum temperature along the saturation line), maxcondenbar (state point of maximum pressure along the saturation line), and critical point for air used in this work were reported by Jacobsen *et al.* (1990b). These values were determined using a modified Leung–Griffiths model to represent the high-pressure VLE data for ternary systems of nitrogen, argon, and oxygen. The values of the properties at the maxcondentherm (T_j, p_j, ρ_j) were used as reducing parameters for the equa-

TABLE 10. Coefficients for the dew and bubble-point pressure and density equations for air

| i | N_i | i | N_i |
|---|-------------------------------|---|---------------------|
| | Bubble-point pressure, | | Dew-point pressure, |
| | Eq. (1) | | Eq. (1) |
| 1 | 0.226 072 4 | 1 | -0.1567266 |
| 2 | -7.080499 | 2 | -5.539635 |
| 3 | 5.700 283 | 5 | 0.756 721 2 |
| 4 | $-12.440\ 17$ | 8 | -3.514322 |
| 5 | 17.819 26 | | |
| 6 | -10.813 64 | | |
| | Bubble-point density , | | Dew-point density, |
| | Eq. (2) | | Eq. (3) |
| 1 | 44.3413 | 1 | -2.0466 |
| 2 | -240.073 | 2 | -4.7520 |
| 3 | 285.139 | 3 | -13.259 |
| 4 | -88.3366 | 4 | -47.652 |
| 5 | $-0.892\ 181$ | | |

tion of state reported here so that the dew-point pressure and density equations will not be double-valued at any given temperature above the critical temperature as illustrated in Fig. 6; thus, for these temperature dependent equations, the dew-point line is represented by the bubble-point equation between the maxcondentherm point and the critical point. The fixed point properties for air used in this work are given in Table 9.

3.1. Ancillary Equations

The ancillary equations were developed using linear and nonlinear regression algorithms with the experimental database discussed in Sec. 2 and calculated data discussed in Sec. 6.1. The equation for the dew and bubble-point pressures is

$$\ln\left(\frac{p}{p_i}\right) = \left(\frac{T_j}{T}\right) \sum_{i=1}^8 N_i \theta^{i/2}, \tag{1}$$

where the N_i are the coefficients given in Table 10 for either the dew or bubble-point pressure equation and $\theta = 1 - T/T_j$. Values of N_i are zero for the coefficients not listed in the table. Densities along the dew and bubble lines can be calculated from the Helmholtz energy formulation given in Sec. 4 at any temperature and the corresponding pressure calculated from Eq. (1). Ancillary equations for the densities

TABLE 9. Maxcondentherm, maxcondenbar, and critical point for air

| | Pressure ^a (MPa) | Temperature ^{a,c} (K) | Density (mol/dm³) |
|------------------------------------|---------------------------------|-----------------------------------|----------------------|
| Maxcondentherm ^b | $3.785\ 02\pm0.004\pm\Delta p$ | $132.6312 \pm 0.002 \pm \Delta T$ | 10.4477±0.05 |
| Maxcondenbar | $3.7891 \pm 0.002 \pm \Delta p$ | $132.6035 \pm 0.004 \pm \Delta T$ | 11.0948 ± 0.05 |
| Critical point | $3.7860 \pm \Delta p$ | $132.5306 \pm \Delta T$ | 11.8308 ± 0.05 |

 $^{^{}a}\Delta p$ = 0.02 MPa, ΔT = 0.02 K; the uncertainties in the maxcondentherm and maxcondenbar pressure and temperature comprise components related to the value associated with the critical parameters and smaller components related to the uncertainty in the distance from the critical point.

^bValues used for the reducing parameters in the equation of state.

^cTemperature on ITS-90.

along the saturation lines are given here for convenience in certain calculations such as estimation routines. The equation for the bubble-point density is

$$\frac{\rho}{\rho_j} - 1 = N_1 \,\theta^{0.65} + N_2 \,\theta^{0.85} + N_3 \,\theta^{0.95} + N_4 \,\theta^{1.1} + N_5 \ln \frac{T}{T_j}, \tag{2}$$

and the equation for the dew-point density is

$$\ln\left(\frac{\rho}{\rho_i}\right) = N_1 \,\theta^{0.41} + N_2 \,\theta^{1.0} + N_3 \,\theta^{2.8} + N_4 \,\theta^{6.5}, \tag{3}$$

where the coefficients are also given in Table 10. In these equations, T_j , p_j , and ρ_j are values at the maxcondentherm. Comparisons of these equations to experimental data are given in Sec. 6.1. These ancillary equations are valid for temperatures from the solidification point (59.75 K) to the maxcondentherm (132.6312 K). Values for the dew-point pressure and density between the critical point and the maxcondentherm, shown in Fig. 6, are calculated by the bubble-point equations.

3.2. Freezing Liquid Line

A freezing liquid line for air was estimated using the melting curves for pure nitrogen, argon, and oxygen due to the lack of experimental information for the mixture. The solidification temperature T_s of air along the bubble line given by Blanke (1973) is 59.75 K. The solidification pressure p_s corresponding to this temperature was calculated from the bubble-point pressure equation, Eq. (1), as 0.005 265 MPa. The freezing liquid line for air was assumed to have the same general pressure dependence as the freezing liquid line for pure nitrogen. This can be expressed as

$$p_{r_{\text{Air}}}^f = \varphi p_{r_{\text{N}_2}}^f, \tag{4}$$

where the reduced freezing pressures are defined as

$$p_r^f = \frac{p^f(T_r)}{p_{tp}} - 1 (5)$$

for nitrogen, argon, or oxygen, and

$$p_r^f = \frac{p^f(T_r)}{p_s} - 1$$
(6)

for air. In these equations, $p^f(T_r)$ is the freezing liquid pressure at the reduced temperature, T_r is the reduced temperature $(T/T_{\rm tp} \text{ or } T/T_s)$, and the subscript tp indicates the triple point for nitrogen, argon, or oxygen. The value of φ was determined at a reduced temperature of 2 from

$$\begin{split} x_{\mathrm{N}_{2}} \ln p_{r_{\mathrm{N}_{2}}}^{f}(2) + x_{\mathrm{Ar}} \ln p_{r_{\mathrm{Ar}}}^{f}(2) + x_{\mathrm{O}_{2}} \ln p_{r_{\mathrm{O}_{2}}}^{f}(2) \\ = & \ln p_{r_{\mathrm{Air}}}^{f}(2) = \ln \varphi p_{r_{\mathrm{N}_{1}}}^{f}(2), \end{split} \tag{7}$$

where the mole fractions x are taken from the composition of air used in this work, and the pure fluid melting pressures were taken from Span *et al.* (2000), Tegeler *et al.* (1999), and Schmidt and Wagner (1985). The value φ calculated

TABLE 11. Averaged molar composition of standard air at specified temperatures

| | | Comp | osition | |
|------|--------|--------|---------|--------|
| T(K) | N_2 | Ar | O_2 | NO |
| 1500 | 0.7804 | 0.0093 | 0.2089 | 0.0014 |
| 1600 | 0.7802 | 0.0093 | 0.2083 | 0.0022 |
| 1700 | 0.7797 | 0.0093 | 0.2078 | 0.0032 |
| 1800 | 0.7790 | 0.0094 | 0.2071 | 0.0045 |
| 1900 | 0.7780 | 0.0094 | 0.2063 | 0.0063 |
| 2000 | 0.7768 | 0.0094 | 0.2056 | 0.0082 |

from this equation is 2.773 234. Although values of the freezing liquid line for air could have been calculated from a generalization of Eq. (7), Eq. (4) was selected instead since its extrapolation behavior to very high temperatures was known. The resulting equation for the freezing liquid pressure of air is

$$\frac{p}{p_s} - 1 = 35\,493.5 \left[\left(\frac{T}{T_s} \right)^{1.789\,63} - 1 \right],\tag{8}$$

derived using the melting pressure equation of nitrogen [Span et al. (2000)].

4. Equation of State for Air

4.1. Effects of Dissociation

The details included in this and the following section are taken from Panasiti (1996) with some modifications. At high temperatures, air and its constituents dissociate, and the composition changes to include various atomic and ionic species and new compounds including oxides of nitrogen. Hilsenrath and Klein (1965) report the extent of dissociation in the temperature range from 1500 to 15000 K. They considered the equilibrium between atoms, molecules, and ions for the elements nitrogen, oxygen, argon, carbon, and neon. In all, 28 species were considered by Hilsenrath and Klein. For the temperature range of current interest, the amount of nitric oxide (NO) in equilibrium at 2000 K formed from the dissociation of oxygen and nitrogen was large enough that it might be expected to cause a noticeable change in the thermodynamic properties of the mixture. No other constituents were considered in the current study due to their small concentrations. Table 11 lists averaged values of composition, derived from Hilsenrath and Klein, for a mixture of nitrogen, argon, oxygen, and nitric oxide at various temperatures. Although Hilsenrath and Klein consider a pressure dependence, the effect on calculated properties is small, and the composition of air is assumed to be constant over all pressures at a given temperature.

Several methods were considered to determine the effects of dissociation on the thermodynamic properties of air. These methods include calculations using cubic equations, extended corresponding states predictions [Clarke *et al.* (1994); Lemmon (1996)], and mixture equations of state of the form developed by Bender (1973). Use of extended cor-

responding states models with accurate pure fluid information is one of the most accurate methods for obtaining mixture properties. However, the equation of state for oxygen [Schmidt and Wagner (1985)] has limited temperature and pressure ranges (less than 300 K and 82 MPa), and the accuracy of properties calculated beyond these limits is uncertain. In addition, there is no published equation of state for nitric oxide. Therefore, the Peng–Robinson (1976) equation of state was used with standard mixing rules to determine the effect of dissociation of air and its constituents. The acentric factors ω required in the Peng-Robinson model and the critical parameters for nitrogen, argon, and oxygen are listed in the fixed point tables. The critical temperature of nitric oxide is 180.15 K, the critical pressure is 6.48 MPa, and the acentric factor is 0.6. The values of the critical parameters and the acentric factor for nitric oxide were taken from Prasad and Viswanath (1974).

Although cubic equations generally cannot be used to calculate values of density with high accuracy over a wide range of temperature and pressure, the purpose here was to compare the differences in density between two mixtures: the three-component N2-Ar-O2 model of fixed composition for air used in this work, and a four-component N₂-Ar-O₂-NO model which accounted for dissociation by the addition of the NO component. Errors associated with the calculation of density differences should be much smaller than those calculated for the overall value of density. The highest deviation at 2000 K and 2000 MPa is approximately 0.1%, which is less than the expected uncertainty of experimental measurements at these high temperatures and pressures. The differences in calculated heat capacities between the two mixtures is approximately 0.8%, also within the estimated uncertainty. Based on this analysis, it was concluded that dissociation effects are small enough at temperatures up to 2000 K and pressures to 2000 MPa that they were not explicitly considered in the development of the new equation of state for air.

4.2. Calculation of Air Properties from Experimental Data for Nitrogen

A modified extended corresponding states technique was used to predict properties of air corresponding to states defined by measurements of nitrogen properties. Experimental $p-\rho-T$ measurements for nitrogen extend to 1800 K and 2220 MPa. Air properties were calculated from high-pressure and high-temperature nitrogen data using the assumption that the compressibility factors are equal

$$Z_{\text{Air}}(T_{\text{Air}}, \rho_{\text{Air}}) = Z_{\text{N}_2}(T_{\text{N}_2}, \rho_{\text{N}_2})$$
 (9)

at corresponding states defined by

$$T_{\text{Air}} = (1.038128 + 0.000054933T_{\text{N}_2})T_{\text{N}_2}$$
 (10)

and

$$\rho_{Air} = 1.043492 \rho_{N_2}, \tag{11}$$

for T in kelvin. The pressure was calculated from Eq. (9) for air using nitrogen data

$$p_{\text{Air}} = \frac{T_{\text{Air}}}{T_{\text{N}_2}} \frac{\rho_{\text{Air}}}{\rho_{\text{N}_2}} p_{\text{N}_2}.$$
 (12)

The numerical constants in Eqs. (10) and (11) were obtained by fitting selected $p-\rho-T$ nitrogen data of Jaeschke and Hinze (1991), Klimeck et al. (1998), Michels et al. (1936), and Nowak et al. (1997) in the range from 200 to 520 K at pressures to 67 MPa to a preliminary version of the air equation of state. The data of Nowak et al. represent the most reliable high temperature, high pressure data available for nitrogen. This fitting procedure minimized deviations between air properties estimated from nitrogen data and values calculated from the preliminary equation of state for air. The average absolute deviations for values calculated from the nitrogen equation of state of Span et al. (2000) were 0.01% for the data of Jaeschke and Hinze, 0.002% for the data of Klimeck et al., 0.02% for the data of Michels et al., and 0.001% for the data of Nowak et al. In comparison, the average absolute deviations for values calculated from the air equation of state as compared with the transformed data were 0.02% for the data of Jaeschke and Hinze and of Klimeck et al., 0.04% for the data of Michels et al., and 0.03% for the data of Nowak et al. Using Eqs. (10)-(12), high temperature, high pressure $p-\rho-T$ nitrogen data of Robertson and Babb (1969) and Saurel (1958) were transformed to predict state points for air and were used in subsequent fitting of the equation of state for air. The data of Robertson and Babb were used in the final regression of the coefficients. The estimated uncertainty of calculated air properties is 1.0% in density at the highest temperatures and pressures of 2000 K and 2000 MPa.

4.3. Fitting Procedures

The equation of state was developed using selected experimental pressure—density—temperature $(p-\rho-T)$ data, isochoric heat capacity data, speed of sound data, and second virial coefficients as indicated in Table 6 and discussed in Sec. 2. The units adopted for this work are megapascals for pressure, moles per cubic decimeter for density, kelvins for temperature, and joules for energy. Units of the experimental data were converted as necessary from those of the original publications to these units. All temperatures were converted to the International Temperature Scale of 1990 (ITS-90) based on temperature differences given by Preston-Thomas (1990).

The functional forms of preliminary equations for the residual part of the Helmholtz energy were optimized with the algorithm developed by Wagner (1974). The fitting process was used to select an optimal set of terms for representing the selected data from a large bank of terms of the form given in Eq. (25) below. The exponents i_k , j_k , and l_k were selected in this process. The exponents j_k in Eq. (25) are generally considered to be greater than or equal to zero so that only ideal gas terms contribute as T goes to infinity, and

 i_k and l_k are integers greater than zero. Each data point used in the least-squares determination of the coefficients of the equation of state was assigned a weighting factor. The weights used in the fitting process were calculated using the error propagation formula (sometimes called the theorem of propagation of variance). The functions for weighting were calculated by making use of a preliminary equation of state for the partial derivatives required for estimating variances by the error propagation formula. In several instances, the error propagation weights were modified by the assignment of arbitrary multiplicative factors to increase or lessen the effect of a particular data set on the overall representation of the surface.

The isobaric heat capacity and speed of sound data are not linearly related to the coefficients in the Helmholtz energy formulation and must first be linearized before being used in the fitting process. The isobaric heat capacity is linearized by calculating the density at the experimental pressure and temperature and reducing the isobaric heat capacities to an equivalent isochoric heat capacity using

$$c_{v} = c_{p} - R \frac{\left(1 + \delta \frac{\partial \alpha^{r}}{\partial \delta} - \delta \tau \frac{\partial^{2} \alpha^{r}}{\partial \delta \partial \tau}\right)^{2}}{1 + 2\delta \frac{\partial \alpha^{r}}{\partial \delta} + \delta^{2} \frac{\partial^{2} \alpha^{r}}{\partial \delta^{2}}}.$$
 (13)

The speed of sound is linearized by calculating the density and the ratio of the heat capacities γ from a preliminary equation of state. To improve the representation of the speed of sound data and of the available shock tube data for air, the final functional form was developed using nonlinear regression techniques. Using a combination of linear and nonlinear techniques, a functional form was obtained which yielded the best representation of the selected experimental data. Details of the nonlinear fitting procedures are given by Lemmon and Jacobsen (2000).

4.4. Ideal Gas Heat Capacity

The ideal gas heat capacity for air is given by the mole fraction average of the ideal gas heat capacities for nitrogen, argon, and oxygen

$$\frac{c_p^0(T)}{R} = x_{N_2} \left(\frac{c_p^0(T)}{R} \right)_{N_2} + x_{Ar} \left(\frac{c_p^0(T)}{R} \right)_{Ar} + x_{O_2} \left(\frac{c_p^0(T)}{R} \right)_{O_2}.$$
(14)

The ideal gas heat capacity for nitrogen taken from Span et al. (2000) is

$$\frac{c_p^0}{R} = \sum_{i=1}^4 N_i T^{i-1} + \frac{N_5 u^2 e^u}{(e^u - 1)^2},\tag{15}$$

where $u = N_6/T$. The ideal gas heat capacity for argon is

$$\frac{c_p^0}{R} = \frac{5}{2}. (16)$$

The ideal gas heat capacity for oxygen taken from Schmidt and Wagner (1985) is

TABLE 12. Coefficients for the ideal gas expressions

| i | N_{i} | i | N_i |
|----|----------------------------------|----|--------------------------------|
| | c_p^0 of nitrogen, Eq. (15) | | c_p^0 of oxygen, Eq. (17) |
| 1 | 3.5 | 1 | 3.500 42 |
| 2 | 0.3066469×10^{-5} | 2 | 0.166961×10^{-7} |
| 3 | $0.470\ 124\ 0 \times 10^{-8}$ | 3 | 1.067 78 |
| 4 | $-0.3987984 \times 10^{-12}$ | 4 | 1.012 58 |
| 5 | 1.012 941 | 5 | 0.944 365 |
| 6 | 3364.011 | 6 | 2242.45 |
| | | 7 | 11 580.4 |
| | c_n^0 of air, | | α^0 of air, |
| | Eq. (18) | | Eq. (24) |
| 1 | 3.490 888 032 | 1 | $0.605719400 \times 10^{-7}$ |
| 2 | $2.395525583 \times 10^{-6}$ | 2 | $-0.210274769\times10^{-4}$ |
| 3 | $7.172\ 111\ 248 \times 10^{-9}$ | 3 | $-0.158860716 \times 10^{-3}$ |
| 4 | $-3.115413101\times10^{-13}$ | 4 | -13.841928076 |
| 5 | 0.223 806 688 | 5 | 17.275 266 575 |
| 6 | 0.791 309 509 | 6 | $-0.195363420\times10^{-3}$ |
| 7 | 0.212 236 768 | 7 | 2.490 888 032 |
| 8 | 0.197 938 904 | 8 | 0.791 309 509 |
| 9 | 3364.011 | 9 | 0.212 236 768 |
| 10 | 2242.45 | 10 | -0.197938904 |
| 11 | 11 580.4 | 11 | 25.363 65 |
| | | 12 | 16.907 41 |
| | | 13 | 87.312 79 |

$$\frac{c_p^0}{R} = N_1 + N_2 T^2 + \frac{N_3}{T^{1.5}} + \frac{N_4 v^2 e^v}{(e^v - 1)^2} + \frac{(2/3)N_5 w^2 e^{-w}}{[(2/3)e^{-w} + 1]^2},$$
(17)

where $v = N_6/T$ and $w = N_7/T$. The coefficients for Eqs. (15) and (17) are given in Table 12.

The ideal gas heat capacity c_p^0 for air, calculated by combining the ideal gas heat capacity equations for nitrogen, argon, and oxygen, according to Eq. (14), is given by the following expression:

$$\frac{c_p^0}{R} = \sum_{i=1}^4 N_i T^{i-1} + \frac{N_5}{T^{1.5}} + \frac{N_6 u^2 e^u}{(e^u - 1)^2} + \frac{N_7 v^2 e^v}{(e^v - 1)^2} + \frac{(2/3)N_8 w^2 e^{-w}}{((2/3)e^{-w} + 1)^2},$$
(18)

where $u=N_9/T$, $v=N_{10}/T$, and $w=N_{11}/T$. The coefficients of this equation are given in Table 12; note that different values for the coefficients N_i are used in Eqs. (15), (17), and (18)

4.5. Ideal Gas Helmholtz Energy

The ideal gas contribution to the Helmholtz energy of air is given by

$$a^{0} = -RT + \sum_{i=1}^{3} x_{i} (h_{i}^{0} - Ts_{i}^{0} + RT \ln x_{i}), \qquad (19)$$

where x_i is the mole fraction of component i in air, and h_i^0 and s_i^0 are the ideal gas enthalpy and entropy of component i at the specified temperature. The ideal gas enthalpy is given by

$$h_i^0 = h_{0i}^0 + \int_{T_0}^T c_{pi}^0 dT, \qquad (20)$$

where h_{0i}^0 is the enthalpy datum value at the reference temperature (T_0) for component i, based upon a zero reference point for the ideal crystal at absolute zero temperature. The ideal gas entropy is given by

$$s_i^0 = s_{0i}^0 + \int_{T_0}^T \frac{c_{pi}^0}{T} dT - R \ln\left(\frac{p^0}{p_0}\right), \tag{21}$$

where s_{0i}^0 is the entropy datum value at the reference temperature and pressure $(T_0 \text{ and } p_0)$ for component i, also based upon a zero reference point of the ideal crystal at absolute zero temperature. The variable p^0 is the ideal gas pressure at the fluid density and temperature. Combining Eqs. (19)-(21) results in the following expression for the ideal gas Helmholtz energy

$$a^{0} = -RT + \sum_{i=1}^{3} x_{i} \left(RT \ln \frac{\rho T}{\rho_{0} T_{0}} + h_{0i}^{0} - Ts_{0i}^{0} + \int_{T_{0}}^{T} c_{pi}^{0} dT - T \int_{T_{0}}^{T} \frac{c_{pi}^{0}}{T} dT + RT \ln x_{i} \right), \quad (22)$$

where p^0/p_0 has been replaced with $\rho T/\rho_0 T_0$. Dividing Eq. (22) by RT and replacing the temperatures with T_{ci}/τ and the densities with $\delta \rho_{ci}$ results in

$$\alpha^{0} = \frac{a^{0}}{RT} = -1 + \sum_{i=1}^{3} x_{i} \left(\ln \frac{\delta \tau_{0}}{\delta_{0} \tau} + \frac{h_{0i}^{0} \tau}{RT_{c}} - \frac{s_{0i}^{0}}{R} - \frac{\tau}{R} \int_{\tau_{0}}^{\tau} \frac{c_{pi}^{0}}{\tau^{2}} d\tau + \frac{1}{R} \int_{\tau_{0}}^{\tau} \frac{c_{pi}^{0}}{\tau} d\tau + \ln x_{i} \right), \tag{23}$$

where $\tau_0 = T_{ci}/T_0$, $\delta_0 = \rho_0/\rho_{ci}$ is the reduced ideal gas density at p_0 and T_0 , T_0 is the reference temperature (298.15 K), and p_0 is the reference pressure (0.101325 MPa). T_{ci} and ρ_{ci} are the critical parameters of the pure fluids. Values of h_{0i}^0 and s_{0i}^0 for nitrogen, argon, and oxygen are taken from Cox *et al.* (1989) and are given in the fixed point tables.

The following expression for α^0 is obtained by combining Eqs. (18) and (23):

$$\alpha^{0} = \ln \delta + \sum_{i=1}^{5} N_{i} \tau^{i-4} + N_{6} \tau^{1.5} + N_{7} \ln \tau$$

$$+ N_{8} \ln[1 - \exp(-N_{11}\tau)] + N_{9} \ln[1 - \exp(-N_{12}\tau)]$$

$$+ N_{10} \ln[2/3 + \exp(N_{13}\tau)], \qquad (24)$$

where $\delta = \rho/\rho_j$ and $\tau = T_j/T$, as given by Eq. (25) in the Sec. 4.6, and T_j and ρ_j are parameters at the maxcondentherm of air. The coefficients of this equation are given in Table 12.

4.6. Equation of State for Air

The equation of state for air was developed using experimental data for pressure-density-temperature $(p-\rho-T)$, isochoric heat capacity, speed of sound, and second virial

TABLE 13. Coefficients and exponents for the equation of state for air

| k | N_k | i_k | j_k | l_k |
|----|--|-------|-------|-------|
| 1 | 0.118 160 747 229 | 1 | 0 | 0 |
| 2 | 0.713 116 392 079 | 1 | 0.33 | 0 |
| 3 | $-0.161824192067 \times 10^{1}$ | 1 | 1.01 | 0 |
| 4 | $0.714\ 140\ 178\ 971 \times 10^{-1}$ | 2 | 0 | 0 |
| 5 | $-0.865421396646 \times 10^{-1}$ | 3 | 0 | 0 |
| 6 | 0.134 211 176 704 | 3 | 0.15 | 0 |
| 7 | $0.112626704218\times10^{-1}$ | 4 | 0 | 0 |
| 8 | $-0.420533228842\times10^{-1}$ | 4 | 0.2 | 0 |
| 9 | $0.349008431982\times10^{-1}$ | 4 | 0.35 | 0 |
| 10 | $0.164957183186\times10^{-3}$ | 6 | 1.35 | 0 |
| 11 | $-0.101\ 365\ 037\ 912$ | 1 | 1.6 | 1 |
| 12 | -0.173813690970 | 3 | 0.8 | 1 |
| 13 | $-0.472\ 103\ 183\ 731 \times 10^{-1}$ | 5 | 0.95 | 1 |
| 14 | $-0.122523554253 \times 10^{-1}$ | 6 | 1.25 | 1 |
| 15 | -0.146629609713 | 1 | 3.6 | 2 |
| 16 | $-0.316055879821\times10^{-1}$ | 3 | 6 | 2 |
| 17 | $0.233594806142\times10^{-3}$ | 11 | 3.25 | 2 |
| 18 | $0.148287891978 \times 10^{-1}$ | 1 | 3.5 | 3 |
| 19 | $-0.938782884667\times10^{-2}$ | 3 | 15 | 3 |

coefficients, along with calculated air properties at high temperatures and pressures as described in Sec. 4.2. Values of $p-\rho-T$ on the dew and bubble-point curves calculated using Eqs. (1), (2), and (3) were used to define the vapor-liquid equilibrium boundaries. To expand the range of the equation beyond 2000 K and 2000 MPa, data published by Nellis *et al.* (1991), determined using a shock tube apparatus, were included in the fitting process as discussed in Sec. 4.8 below. The equation of state for air used in this work is explicit in the nondimensional Helmholtz energy

$$\alpha(\delta, \tau) = \frac{a(\rho, T)}{RT} = \alpha^{0}(\delta, \tau) + \alpha^{r}(\delta, \tau), \qquad (25)$$

where α^0 is the ideal-gas contribution to the Helmholtz energy given in the previous section, α^r is the residual contribution to the Helmholtz energy, $\delta = \rho/\rho_j$ is the reduced density, $\tau = T_j/T$ is the reciprocal reduced temperature, ρ_j is the density at the maxcondentherm, and T_j is the temperature at the maxcondentherm.

The residual Helmholtz energy contribution to the equation of state is given by

$$\alpha^{r}(\delta,\tau) = \sum_{k=1}^{10} N_{k} \delta^{i_{k}} \tau^{j_{k}} + \sum_{k=11}^{19} N_{k} \delta^{i_{k}} \tau^{j_{k}} \exp(-\delta^{l_{k}}).$$
(26)

The coefficients N_k of the equation of state are given in Table 13. The values of i_k , j_k , and l_k were determined from the fitting procedures described in Sec. 4.3. For the nonexponential terms in Eq. (26), i_k ranges from 1 to 6 and j_k ranges from 0 to 1.35. The thermodynamic properties can be

calculated from the Helmholtz energy using differentiation with respect to density and temperature as described in Sec. 4.7.

4.7. Calculation of Thermodynamic Properties

The functions used for calculating pressure, compressibility factor, internal energy, enthalpy, entropy, Gibbs energy, isochoric heat capacity, isobaric heat capacity, and the speed of sound from Eq. (25) are given as Eqs. (27)–(34). These functions were used in calculating the tables of thermodynamic properties of air given in the Appendix. The densities for the dew and bubble-point states are determined by the simultaneous solution of the dew or bubble-point pressure equation (1) and the equation of state for a given temperature. The derived properties for saturation states are calculated as functions of temperature and density using the standard thermodynamic relations given in Eqs. (27)–(34).

$$Z = \frac{p}{\rho RT} = 1 + \delta \left(\frac{\partial \alpha^r}{\partial \delta} \right)_{\tau}$$
 (27)

$$\frac{u}{RT} = \tau \left[\left(\frac{\partial \alpha^0}{\partial \tau} \right)_{\delta} + \left(\frac{\partial \alpha^r}{\partial \tau} \right)_{\delta} \right] \tag{28}$$

$$\frac{h}{RT} = \tau \left[\left(\frac{\partial \alpha^0}{\partial \tau} \right)_{\delta} + \left(\frac{\partial \alpha^r}{\partial \tau} \right)_{\delta} \right] + \delta \left(\frac{\partial \alpha^r}{\partial \delta} \right)_{\tau} + 1 \qquad (29)$$

$$\frac{s}{R} = \tau \left[\left(\frac{\partial \alpha^0}{\partial \tau} \right)_{\delta} + \left(\frac{\partial \alpha^r}{\partial \tau} \right)_{\delta} \right] - \alpha^0 - \alpha^r \tag{30}$$

$$\frac{g}{RT} = 1 + \alpha^0 + \alpha^r + \delta \left(\frac{\partial \alpha^r}{\partial \delta} \right)_{\tau}$$
 (31)

$$\frac{c_v}{R} = -\tau^2 \left[\left(\frac{\partial^2 \alpha^0}{\partial \tau^2} \right)_{\delta} + \left(\frac{\partial^2 \alpha^r}{\partial \tau^2} \right)_{\delta} \right]$$
(32)

$$\frac{c_p}{R} = \frac{c_v}{R} + \frac{\left[1 + \delta \left(\frac{\partial \alpha^r}{\partial \delta}\right)_{\tau} - \delta \tau \left(\frac{\partial^2 \alpha^r}{\partial \delta \partial \tau}\right)\right]^2}{\left[1 + 2\delta \left(\frac{\partial \alpha^r}{\partial \delta}\right)_{\tau} + \delta^2 \left(\frac{\partial^2 \alpha^r}{\partial \delta^2}\right)_{\tau}\right]}$$
(33)

$$\frac{w^2M}{RT} = \frac{c_p}{c_v} \left[1 + 2\delta \left(\frac{\partial \alpha^r}{\partial \delta} \right) + \delta^2 \left(\frac{\partial^2 \alpha^r}{\partial \delta^2} \right) \right]$$
(34)

Equations for the second and third virial coefficients are given in Eqs. (35)–(36).

$$B(T) = \frac{1}{\rho_j} \left(\frac{\partial \alpha^r}{\partial \delta} \Big|_{\tau} \right)_{\delta = 0}$$
 (35)

$$C(T) = \frac{1}{\rho_j^2} \left(\frac{\partial^2 \alpha^r}{\partial \delta^2} \Big|_{\tau} \right)_{\delta = 0}$$
 (36)

Other derived properties, given in Eqs. (37)–(47), include the first derivative of pressure with respect to density $(\partial p/\partial \rho)_T$, second derivative of pressure with respect to density $(\partial^2 p/\partial \rho)_T$, first derivative of pressure with respect to temperature $(\partial p/\partial T)_\rho$, Joule–Thomson coefficient (μ_J) , isentropic expansion coefficient (k), isothermal expansion coefficient (k_T) , volume expansivity (β) , adiabatic compressibility (β_s) , adiabatic bulk modulus (B_s) , isothermal compressibility (κ) , and isothermal bulk modulus (K_T) .

$$\left(\frac{\partial p}{\partial \rho}\right)_{T} = RT \left[1 + 2\delta \left(\frac{\partial \alpha^{r}}{\partial \delta}\right)_{T} + \delta^{2} \left(\frac{\partial^{2} \alpha^{r}}{\partial \delta^{2}}\right)_{T}\right]$$
(37)

$$\left(\frac{\partial^{2} p}{\partial \rho^{2}}\right)_{T} = \frac{RT}{\rho} \left[2 \delta \left(\frac{\partial \alpha^{r}}{\partial \delta}\right)_{\tau} + 4 \delta^{2} \left(\frac{\partial^{2} \alpha^{r}}{\partial \delta^{2}}\right)_{\tau} + \delta^{3} \left(\frac{\partial^{3} \alpha^{r}}{\partial \delta^{3}}\right)_{\tau} \right]$$
(38)

$$\left(\frac{\partial p}{\partial T}\right)_{\rho} = R\rho \left[1 + \delta \left(\frac{\partial \alpha^{r}}{\partial \delta}\right)_{\tau} - \delta \tau \left(\frac{\partial^{2} \alpha^{r}}{\partial \delta \partial \tau}\right)\right] \tag{39}$$

$$\mu_{J} = \left(\frac{\partial T}{\partial p}\right)_{h} = \frac{T\beta - 1}{\rho c_{p}} \tag{40}$$

$$k = -\frac{v}{p} \left(\frac{\partial p}{\partial v} \right)_{s} = \frac{w^{2} \rho M}{p} \tag{41}$$

$$k_T = -\frac{v}{p} \left(\frac{\partial p}{\partial v} \right)_T = \frac{\rho}{p} \left(\frac{\partial p}{\partial \rho} \right)_T \tag{42}$$

$$\beta = \frac{1}{v} \left(\frac{\partial v}{\partial T} \right)_{p} = \frac{1}{\rho} \left(\frac{\partial p}{\partial T} \right)_{\rho} \left(\frac{\partial \rho}{\partial p} \right)_{T} \tag{43}$$

$$\beta_s = \frac{1}{kp} = -\frac{1}{v} \left(\frac{\partial v}{\partial p} \right)_s \tag{44}$$

$$B_s = kp = -v \left(\frac{\partial p}{\partial v} \right)_s \tag{45}$$

$$\kappa = \frac{1}{k_T p} = -\frac{1}{v} \left(\frac{\partial v}{\partial p} \right)_T \tag{46}$$

$$K_T = k_T p = -v \left(\frac{\partial p}{\partial v} \right)_T \tag{47}$$

The derivatives of the ideal gas Helmholtz energy, Eq. (24), are given in Eqs. (48)–(49).

$$\frac{\partial \alpha^{0}}{\partial \tau} \bigg|_{\delta} = \sum_{i=1}^{5} (i-4)N_{i}\tau^{i-5} + 1.5N_{6}\tau^{0.5} + N_{7}/\tau + \frac{N_{8}N_{11}}{e^{N_{11}\tau} - 1} + \frac{N_{9}N_{12}}{e^{N_{12}\tau} - 1} + \frac{N_{10}N_{13}}{(2/3)e^{-N_{13}\tau} + 1}$$

$$\frac{\partial^{2}\alpha^{0}}{\partial^{2}\tau} \bigg|_{\delta} = \sum_{i=1}^{5} (i-4)(i-5)N_{i}\tau^{i-6} + 0.75N_{6}\tau^{-0.5} - N_{7}/\tau^{2} - \frac{N_{8}N_{11}^{2}e^{N_{11}\tau}}{(e^{N_{11}\tau} - 1)^{2}} - \frac{N_{9}N_{12}^{2}e^{N_{12}\tau}}{(e^{N_{12}\tau} - 1)^{2}} + \frac{(2/3)N_{10}N_{13}^{2}e^{-N_{13}\tau}}{[(2/3)e^{-N_{13}\tau} + 1]^{2}}$$
(49)

The derivatives of the residual Helmholtz energy, Eq. (26), are given in Eqs. (50)–(55).

$$\begin{split} \frac{\partial \alpha^{r}}{\partial \delta} \bigg|_{\tau} &= \sum_{k=1}^{10} i_{k} N_{k} \delta^{i_{k}-1} \tau^{j_{k}} + \sum_{k=11}^{19} N_{k} \delta^{i_{k}-1} \tau^{j_{k}} \\ &\times \exp(-\delta^{l_{k}}) (i_{k} - l_{k} \delta^{l_{k}}) \\ \frac{\partial^{2} \alpha^{r}}{\partial \delta^{2}} \bigg|_{\tau} &= \sum_{k=1}^{10} i_{k} (i_{k} - 1) N_{k} \delta^{i_{k}-2} \tau^{j_{k}} + \sum_{k=11}^{19} N_{k} \delta^{i_{k}-2} \tau^{j_{k}} \\ &\times \exp(-\delta^{l_{k}}) [(i_{k} - l_{k} \delta^{l_{k}}) (i_{k} - 1 - l_{k} \delta^{l_{k}}) - l_{k}^{2} \delta^{l_{k}}] \end{split}$$

$$(50)$$

$$\frac{\partial^{3} \alpha^{r}}{\partial \delta^{3}} \bigg|_{\tau} = \sum_{k=1}^{10} i_{k} (i_{k} - 1) (i_{k} - 2) N_{k} \delta^{i_{k} - 3} \tau^{j_{k}}$$

$$+ \sum_{k=11}^{19} N_{k} \delta^{i_{k} - 3} \tau^{j_{k}} \exp(-\delta^{l_{k}}) \{ i_{k} (i_{k} - 1) (i_{k} - 2) \}$$

$$+ \delta^{l_{k}} [-2l_{k} + 6i_{k} l_{k} - 3i_{k}^{2} l_{k} - 3i_{k} l_{k}^{2} + 3l_{k}^{2} - l_{k}^{3}]$$

$$+ (\delta^{l_{k}})^{2} [3i_{k} l_{k}^{2} - 3l_{k}^{2} + 3l_{k}^{3}] - l_{k}^{3} (\delta^{l_{k}})^{3} \}$$

$$\frac{\partial \alpha^{r}}{\partial \tau} \bigg|_{\delta} = \sum_{k=1}^{10} j_{k} N_{k} \delta^{i_{k}} \tau^{j_{k} - 1} + \sum_{k=11}^{19} j_{k} N_{k} \delta^{i_{k}} \tau^{j_{k} - 1} \exp(-\delta^{l_{k}})$$

$$(53)$$

$$\frac{\partial^{2} \alpha^{r}}{\partial \tau^{2}} \bigg|_{\delta} = \sum_{k=1}^{10} j_{k} (j_{k} - 1) N_{k} \delta^{i_{k}} \tau^{j_{k} - 2} + \sum_{k=11}^{19} j_{k} (j_{k} - 1) \times N_{k} \delta^{i_{k}} \tau^{j_{k} - 2} \exp(-\delta^{l_{k}}) \tag{54}$$

$$\frac{\partial^{2} \alpha^{r}}{\partial \tau \partial \delta} = \sum_{k=1}^{10} i_{k} j_{k} N_{k} \delta^{i_{k}-1} \tau^{j_{k}-1} + \sum_{k=11}^{19} j_{k} N_{k} \delta^{i_{k}-1} \tau^{j_{k}-1}$$

$$\times \exp(-\delta^{l_{k}}) (i_{k} - l_{k} \delta^{l_{k}})$$
(55)

4.8. Hugoniot Curve

Data measured with a shock tube apparatus [Nellis *et al.*, (1991)] were included in the optimization of the equation of state for air to improve the extrapolation behavior beyond 2000 K and 2000 MPa. One method used to demonstrate the

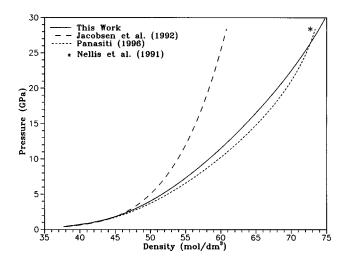


Fig. 7. Calculated Hugoniot curve for air.

extrapolation behavior of an equation of state is the examination of the Hugoniot curve. The conservation relations for fluid properties before and after the shock wave taken from Nellis *et al.* and the equation of state presented here were used to generate the Hugoniot curve for air. The Hugoniot curve is the locus of states accessible to a fluid after a shock wave which occur at a specified initial state. The equations for the conservation of mass, momentum, and energy across the shock wave are

$$p - p_0 = \rho_0(w_s - w_0)(w_p - w_0), \tag{56}$$

$$v = v_0 \left[1 - \frac{(w_p - w_0)}{(w_s - w_0)} \right], \tag{57}$$

and

$$u - u_0 = 0.5(p + p_0)(v_0 - v),$$
 (58)

where p_0 is the initial pressure, ρ_0 is the initial mass density, u_0 is the initial molar internal energy, v_0 is the initial molar volume, w_0 is the initial velocity of the material ahead of the shock front, p is the final shock pressure, v is the final molar volume, and v is the final molar internal energy. The velocity of the shock wave is v0 and the velocity of the material downstream from the shock front is v0. Combining Eqs. (56) and (57) results in

$$v = v_0 \left(1 - \frac{p - p_0}{\rho_0 w_s^2} \right). \tag{59}$$

Rearrangement of Eq. (58) results in

$$v = \frac{-2(u - u_0)}{p + p_0} + v_0.$$
(60)

To calculate a point on the Hugoniot curve for air for a specified upstream state (p_0, T_0) and downstream pressure (p), Eqs. (59) and (60) are solved simultaneously by iterating on the unknown value for the shock wave velocity w_s . The calculated molar volume from Eq. (59) is used with the known value of p to calculate the temperature and internal energy q from the air equation of state. Figure 7 shows the

TABLE 14. Pure fluid equations of state used in the mixture model

| Fluid | Author | Temperature range (K) | Maximum pressure (MPa) |
|-----------------|---|--------------------------|---------------------------|
| Nitrogen | Span et al. (2000) | 63.151-1000 | 2200 |
| Argon Oxygen | Tegeler <i>et al.</i> (1999) Schmidt and Wagner (1985) | 83.806–700 54.361–300 | 1000 82 |

Hugoniot curve for air calculated using this process for an initial state on the bubble line at 77.1 K as specified by Nellis *et al.* The data point on the plot is the lowest measured shock tube point from Nellis *et al.* The reported shock tube points above 30 GPa were not included because, according to Nellis *et al.*, nitrogen spontaneously dissociates above 30 GPa.

As shown in Fig. 7, the Hugoniot curve approaches the shock tube point of Nellis *et al.* (1991) indicating that the extrapolation behavior of the equation of state presented here is reasonable. However, the extrapolation behavior of the formulation should not be based solely on Fig. 7. To further assess the extrapolation behavior of the formulation, other techniques, such as determining the isothermal behavior at extreme pressures and densities, were used. These techniques are described in Sec. 7.1.

5. Mixture Model for the Nitrogen-Argon-Oxygen System

5.1. Mixture Model

The model used in this work to calculate the thermodynamic properties of nitrogen-argon-oxygen mixtures is based on the Helmholtz energy of the mixture using a corresponding states theory that was originally developed by Lemmon (1996) and Lemmon and Jacobsen (1998), (1999). The Helmholtz energy of the mixture is calculated as the sum of an ideal gas contribution, a real gas contribution, and a contribution from mixing. The Helmholtz energy for an ideal solution (the first two contributions) is determined at the reduced density and temperature of the mixture using accurate pure fluid equations of state for the mixture components. The contribution from mixing, a modified excess function, is given by a single generalized empirical equation which is applied to all mixtures considered. Reducing parameters, which are dependent on the mole fraction, are used to modify values of density and temperature. The model may be used to calculate thermodynamic properties of mixtures at various compositions, including dew and bubble-point properties and critical points. It incorporates the most accurate published equation of state for each pure fluid as given in Table 14. Additional information concerning the model is given by Lemmon and Tillner-Roth (1999).

An excess property of a mixture is defined as the actual mixture property at a given condition minus the value for an ideal solution at the same condition. In most other work dealing with excess properties, the mixing condition is defined at constant pressure and temperature. Because the independent variables for the pure fluid Helmholtz energy equations are

density and temperature, properties are calculated here at the density and temperature of the mixture. Since this model deals with the entire fluid surface, reduced values of density and temperature are used rather than the physical values to ensure that properties of the constituents are calculated for the same phase as the mixture. While this approach is arbitrary and different from the usual excess property format, it results in an accurate representation of the phase boundaries for pure fluids and their mixtures.

The formulation for nitrogen—argon—oxygen mixtures given here is a modification of the generalized model developed by Lemmon (1996) for a wide range of fluids including hydrocarbons and cryogens. It preserves the general nature of the previous work, but is more accurate at low temperatures for the calculation of nitrogen—argon—oxygen mixture properties than the prior formulation. By restricting the application to a particular class of fluids, the model was designed to represent the unique characteristics of this system. The model represents available measured data in all parts of the thermodynamic surface within their estimated experimental accuracy.

An advantage of the approach used here is that the behavior of the Helmholtz energy contribution from mixing is the same for the nitrogen—oxygen, nitrogen—argon, and argon—oxygen binary systems. Relatively simple scaling factors are used to determine its magnitude for each pair. This generalization makes it possible to extend the limits of the model for the argon—oxygen mixture, for which there are few experimental single-phase data. In addition, all vapor and liquid thermodynamic properties, including energy, entropy, heat capacity, sound speed, and the mixture critical temperature, pressure, and density, can be calculated using this approach.

Preliminary equations for the mixture model based on the Helmholtz energy have incorporated equations with 7–10 terms in the excess contribution [Lemmon and Jacobsen, (1998), (1999)]. Many of these terms were included to account for deficiencies in the pure fluid equations, especially in the extrapolation behavior to high temperatures and pressures, and at temperatures below the triple points of the pure fluids. The equation presented here uses only two terms and is more predictive in nature, and results calculated for the argon—oxygen system should generally be more accurate than the preliminary models.

The Helmholtz energy for mixtures of nitrogen, argon, and oxygen can be calculated using

$$a = a^{\text{idmix}} + a^E. \tag{61}$$

The Helmholtz energy for the ideal mixture is

$$a^{\text{idmix}} = \sum_{i=1}^{3} x_i [a_i^0(\rho, T) + a_i^r(\delta, \tau) + RT \ln x_i], \quad (62)$$

In these equations, ρ and T are the mixture density and temperature, a_i^0 is the ideal gas Helmholtz energy for component i, and $a_i^r (= \alpha_i^r RT)$ is the pure-fluid residual Helmholtz energy of component i evaluated at a reduced density and temperature defined below. Equations for the ideal gas Helm-

TABLE 15. Parameters of the mixture model

| | F_{ij} | ζ_{ij} (K) | $\xi_{ij}(\mathrm{dm^3/mol})$ |
|-----------------|-----------|------------------|-------------------------------|
| Nitrogen-Argon | 1.121 527 | -1.237 713 | -0.000 760 31 |
| Nitrogen-Oxygen | 1. | -0.856350 | -0.00041847 |
| Argon-Oxygen | 0.597 203 | -2.115126 | 0.000 412 32 |

holtz energy and residual Helmholtz energy for the pure fluids are given in the references shown in Table 14.

The excess contribution to the Helmholtz energy from mixing used in this work is

$$\frac{a^{E}}{RT} = \alpha^{E}(\delta, \tau, \mathbf{x})$$

$$= \left\{ \sum_{i=1}^{2} \sum_{j=i+1}^{3} x_{i}x_{j}F_{ij} \right\} [-0.00195245\delta^{2}\tau^{-1.4}$$

$$+0.00871334\delta^{2}\tau^{1.5}], \tag{63}$$

where the coefficients and exponents were obtained from nonlinear regression of experimental mixture data. Values of F_{ij} are given in Table 15. All single phase thermodynamic properties can be calculated from the Helmholtz energy as described in Sec. 4.7 using the relations

$$\alpha^{0} = \sum_{i=1}^{3} x_{i} \left[\frac{a_{i}^{0}(\rho, T)}{RT} + \ln x_{i} \right]$$
 (64)

and

$$\alpha^{r} = \sum_{i=1}^{3} x_{i} \alpha_{i}^{r}(\delta, \tau) + \alpha^{E}(\delta, \tau, \mathbf{x}), \tag{65}$$

where the derivatives are taken at constant composition. Calculations of two-phase properties are described in Sec. 5.2. The reduced values of density and temperature for the mixture used in Eqs. (63) and (65) are

$$\delta = \rho/\rho_{\rm red} \tag{66}$$

and

$$\tau = T_{\rm red}/T,\tag{67}$$

where ρ and T are the mixture density and temperature, and $\rho_{\rm red}$ and $T_{\rm red}$ are the reducing values

$$\rho_{\text{red}} = \left(\sum_{i=1}^{3} \frac{x_i}{\rho_{c_i}} + \sum_{i=1}^{2} \sum_{j=i+1}^{3} x_i x_j \xi_{ij}\right)^{-1}$$
 (68)

and

$$T_{\text{red}} = \sum_{i=1}^{3} x_i T_{c_i} + \sum_{i=1}^{2} \sum_{j=i+1}^{3} x_i x_j \zeta_{ij}.$$
 (69)

The parameters ζ_{ij} and ξ_{ij} are used to define the shapes of the reducing temperature lines and reducing density lines. These reducing parameters are not the same as the critical parameters of the mixture and are determined simultaneously in the nonlinear fit of experimental data with the other pa-

rameters of the mixture model. The generalized factors and mixture parameters F_{ij} , ζ_{ij} and ξ_{ij} are given in Table 15.

If equations for the ideal gas Helmholtz energy in the nondimensional form $\alpha_i^0(\delta,\tau)$, similar to Eq. (24), are used rather than equations in the dimensional form $a_i^0(\rho,T)$ as indicated by Eq. (64), the reducing variables for δ and τ in the ideal gas equation are

$$\delta = \rho/\rho_c. \tag{70}$$

and

$$\tau = T_{c.}/T, \tag{71}$$

rather than the reducing values defined by Eqs. (66) and (67). This only applies to the ideal gas part of the equation, not to the residual Helmholtz energy. The residual and excess terms $\alpha_i^r(\delta,\tau)$ and $\alpha^E(\delta,\tau,\mathbf{x})$ in Eq. (65) must be evaluated at the reduced state point of the mixture defined by Eqs. (66) and (67). This complication is avoided though the use of dimensional equations of the form given in Eq. (22) or dimensionless equations of the form given in Eq. (23) where the critical properties cancel out of the equation. Equations of the form given in Eq. (24) are derived from dimensional equations, and the critical parameters of the pure fluids are built into the coefficients of the equations.

5.2. Vapor-Liquid Equilibrium (VLE) Properties

In a two-phase nonreacting mixture, the thermodynamic constraints for vapor-liquid equilibrium (VLE) are

$$T' = T'' = T, \tag{72}$$

$$p' = p'' = p, \tag{73}$$

and

$$\mu_i' = \mu_i'', \quad i = 1, 2, \dots, q,$$
 (74)

where the superscripts $^{\prime}$ and $^{\prime\prime}$ refer to the liquid and vapor phases, respectively, and q is the number of components in the mixture. Equation (74) is identical to equating the fugacities of the liquid and vapor phases for each component in the mixture

$$f_i' = f_i'' \tag{75}$$

The chemical potential of component i in a mixture is

$$\mu_i(\rho, T) = \left(\frac{\partial A}{\partial n_i}\right)_{T, V, n_i} = \mu_i^c(T) + RT \ln(f_i), \quad (76)$$

where $\mu_i^c(T)$ is a function of temperature only and the notation n_j indicates that all mole numbers are held constant except n_i . The chemical potential in an ideal gas mixture is

$$\mu_{i}^{0} = \left(\frac{\partial A^{0}}{\partial n_{i}}\right)_{T,V,n_{i}} = \mu_{i}^{c}(T) + RT \ln(f_{i}^{0}), \tag{77}$$

where f_i^0 is the ideal gas partial pressure of constituent i, $x_i p^0 = x_i \rho RT$. Subtracting these equations results in

TABLE 16. Different types of VLE calculations

| Calculation type | Specified quantities | Calculated quantities |
|------------------|----------------------|-----------------------|
| BUBL p | T and the x_i | p and the y_i |
| DEW p | T and the y_i | p and the x_i |
| BUBL T | p and the x_i | T and the y_i |
| DEW T | p and the y_i | T and the x_i |
| FLASH | T and p | x_i and y_i |

$$f_i = x_i \rho RT \exp\left(\frac{\partial (n\alpha^r)}{\partial n_i}\right)_{T,V,n_j},$$
 (78)

where α^r was defined in Eq. (65). The partial derivative at constant temperature, total volume (not molar volume), and mole number of all constituents except i is generally evaluated numerically.

Five common VLE calculations are listed in Table 16. The VLE conditions for these five calculations can be determined iteratively. The x_i represent the mole fractions in the liquid phase, and the y_i represent the mole fractions in the vapor phase. Details of the iterative processes can be found in Van Ness and Abbott (1982) or Smith and Van Ness (1975).

5.3. Critical Locus

In the development of the mixture model, equations for the reducing parameters were given for the calculation of the reduced density and temperature δ and τ . These parameters are equal to the critical parameters in the pure fluid limits. For a mixture, the reducing parameters are not the same as the critical parameters, but are empirical functions designed to minimize the deviations between experimental data and the mixture model.

The criteria for the critical point of a binary mixture are

$$\left(\frac{\partial^2 g}{\partial x^2}\right)_{p,T} = \left(\frac{\partial^3 g}{\partial x^3}\right)_{p,T} = 0,\tag{79}$$

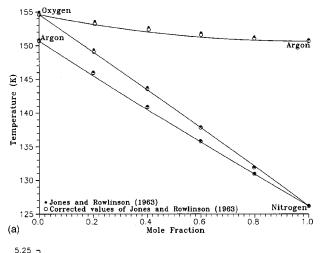
where g is the molar Gibbs energy of the mixture. Near the critical point, this equation can be expanded in terms of the numerical differences

$$\Delta \left(\frac{\partial^2 g}{\partial x^2} \right)_{p,T} \approx \frac{\partial}{\partial T} \left(\frac{\partial^2 g}{\partial x^2} \right)_{p,T} \Delta T + \frac{\partial}{\partial p} \left(\frac{\partial^2 g}{\partial x^2} \right)_{p,T} \Delta p \tag{80}$$

and

$$\Delta \left(\frac{\partial^3 g}{\partial x^3} \right)_{p,T} \approx \frac{\partial}{\partial T} \left(\frac{\partial^3 g}{\partial x^3} \right)_{p,T} \Delta T + \frac{\partial}{\partial p} \left(\frac{\partial^3 g}{\partial x^3} \right)_{p,T} \Delta p, \tag{81}$$

where ΔT and Δp are the differences from the critical point temperature and pressure in an iterative algorithm for finding the critical point, and the left sides of these equations represent the deviations from the zero value at the critical point indicated in Eq. (79). Solving these equations for ΔT and Δp results in



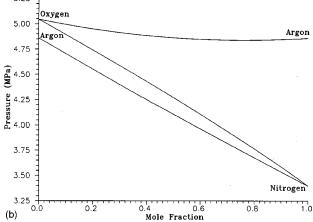


Fig. 8. Critical lines for nitrogen-argon, nitrogen-oxygen, and argon-oxygen binary mixtures.

$$\Delta T \approx \frac{\left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T} \frac{\partial}{\partial p} \left(\frac{\partial^{3} g}{\partial x^{3}}\right)_{p,T} - \left(\frac{\partial^{3} g}{\partial x^{3}}\right)_{p,T} \frac{\partial}{\partial p} \left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T}}{\frac{\partial}{\partial T} \left(\frac{\partial^{3} g}{\partial x^{3}}\right)_{p,T} \frac{\partial}{\partial p} \left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T} - \frac{\partial}{\partial T} \left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T} \frac{\partial}{\partial p} \left(\frac{\partial^{3} g}{\partial x^{3}}\right)_{p,T}} \tag{82}$$

and

$$\Delta p \approx \frac{-\left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T} - \frac{\partial}{\partial T} \left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T} \Delta T}{\frac{\partial}{\partial p} \left(\frac{\partial^{2} g}{\partial x^{2}}\right)_{p,T}}, \quad (83)$$

from which an estimate of the location of the critical point can be obtained. The reducing parameters can be used as initial estimates for the critical pressure and temperature and the values of T and p are repeatedly incremented according to Eqs. (82) and (83) until the second and third derivatives of the Gibbs energy are both simultaneously near zero.

The critical lines for the nitrogen-argon, nitrogen-oxygen, and argon-oxygen binary mixtures are shown in Fig. 8 along with experimental values of the critical temperature from Jones and Rowlinson (1963). The value of T_c for

oxygen reported by Jones and Rowlinson is 0.3% different from the value given by Schmidt and Wagner (1985). Likewise, the value for argon is 0.2% different from that given by Tegeler *et al.* (1999). To account for these differences, the experimental values of Jones and Rowlinson were corrected for each (i,j) binary pair using

$$T_c = T_{c,\text{Jones}} + x_i (T_{c,i} - T_{c,i,\text{Jones}}) + x_j (T_{c,j} - T_{c,j,\text{Jones}}).$$
 (84)

The adjusted values are also shown in Fig. 8.

6. Comparisons of Calculated Properties to Experimental Data

The uncertainty of the equation of state and the mixture model was determined by statistical comparisons of property values calculated with the equation of state or mixture model to experimental data. These statistics are based on the percent deviation in any property X defined as

$$\% \Delta X = 100 \left(\frac{X_{\text{exp}} - X_{\text{calc}}}{X_{\text{exp}}} \right), \tag{85}$$

where $X_{\rm exp}$ is the measured or predicted data, and $X_{\rm calc}$ is computed from the equation of state. Using this definition, the average absolute deviation (AAD) is defined as

$$AAD = \frac{1}{n} \sum_{i=1}^{n} |\% \Delta X_{i}|,$$
 (86)

where n is the number of data points. For the second virial coefficient, the difference in B (in cm³/mol) is used rather than the percent difference since the percent difference can become very large near B=0.

6.1. Comparisons of the Ancillary Equations for Air with Experimental Data

Comparisons of dew and bubble-point properties calculated using Eqs. (1), (2), and (3) to the experimental data of Blanke (1977), Michels *et al.* (1954a), Kuenen and Clark (1917) and the calculated data reported by Jacobsen *et al.* (1990b) are shown in Fig. 9 and the average absolute deviations are given in Table 6.

For the bubble-point pressure equation, the Blanke data below 110 K are not in agreement with values calculated from the mixture model given in Sec. 5, with differences approaching 15% in pressure at the lowest temperature. The Blanke values are the only available low-temperature data for the properties of air on the phase boundaries, however, they were determined graphically from $p-\rho-T$ data. Because of the large deviations mentioned above, they were not used to develop the ancillary equations where they conflicted with values calculated from the mixture model. To maintain consistency between the mixture model and the ancillary equation reported here, the bubble-point pressures from the mixture model were used for temperatures below 100 K in developing the ancillary equations. At higher temperatures,

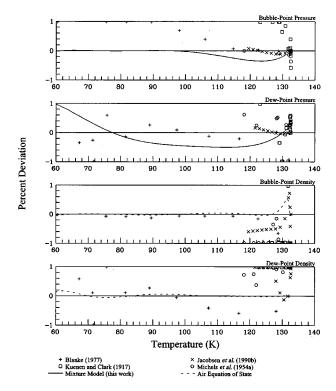


Fig. 9. Comparisons of dew and bubble-point properties calculated with the ancillary equations to experimental and calculated data for air.

the data of Michels *et al.* and Blanke between 120 and 130 K and the Leung-Griffiths calculations from Jacobsen *et al.* between 131.8 and 132.5 K were used.

For the dew-point pressure equation, values below 66 K from the mixture model were used in the absence of experimental data. Most of the data of Blanke are represented within 0.3%. The Leung–Griffiths calculations from Jacobsen *et al.* (1990b) between 130.4 and 132.6 K were fitted in the critical region. For both the dew and bubble-point pressures, the first term of Eq. (1) ensures that the first derivative of pressure with respect to temperature is infinite at the maxcondentherm, as shown in Fig. 6.

Below 118 K, the data of Blanke are the only available saturated density data for air. For the bubble-point densities, both the air equation of state and the mixture model of this work agree with the density data of Blanke to within 0.1%. The single-phase liquid surface is represented well by accurate experimental data, and the uncertainty of density values calculated from the equation of state should be less than 0.1% even at the lowest temperatures. For the dew-point densities, the average absolute deviation of the data of Blanke was 0.6%. These data have not been represented well by previous equations of state for air [Jacobsen et al. (1990a); Jacobsen et al. (1992); Panasiti et al. (1999)] nor with preliminary equations developed using second virial coefficients, and these data show deviations of up to 2% from all of these equations. Calculated density values from the equation of state were used in the development of the dew and bubble-point ancillary equations for consistency.

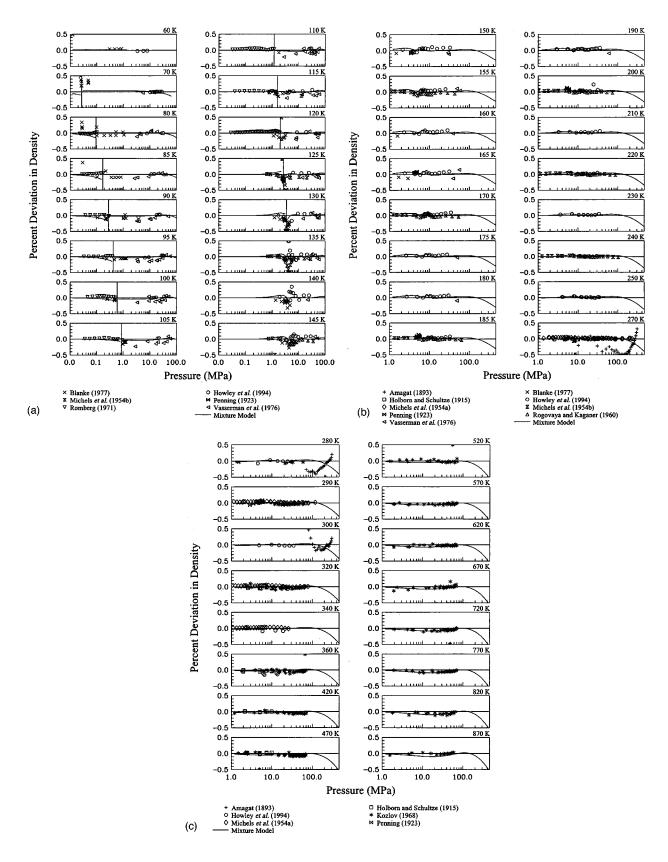


Fig. 10. Comparisons of densities calculated with the equation of state to experimental data for air.

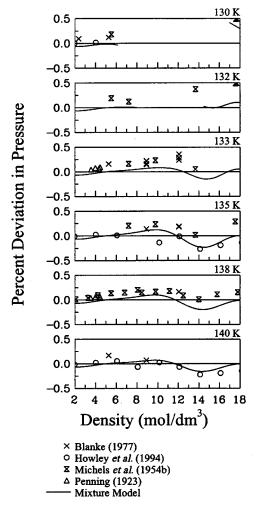


Fig. 11. Comparisons of pressures calculated with the equation of state to experimental data for air in the critical region.

6.2. Comparisons of the Equation of State for Air and the Mixture Model with Experimental and Calculated Data for Air

Table 6 shows the statistical analysis of comparisons to experimental data sets for $p-\rho-T$, isochoric and isobaric heat capacities, speed of sound, and second virial coefficients for air. Comparisons to all available experimental data are given in Figs. 10–15. The vertical lines in these figures show the locations of the phase boundaries at the indicated temperature. Deviations between the mixture model at the composition of standard air and the air equation of state are shown as solid lines at the indicated isotherms or isochores. These comparisons are quite useful in determining the consistency of data sets and the uncertainty of the equation of state, since the mixture model has very few adjustable parameters, and only a very limited set of data was used to determine the parameters. Equations of state can be overfit due to the large number of adjustable parameters. For the mixture model, this is not a problem, and deviations between

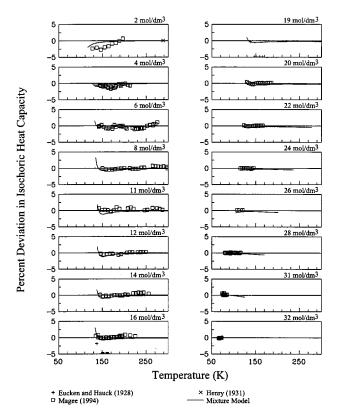


Fig. 12. Comparisons of isochoric heat capacities calculated with the equation of state to experimental data for air.

experimental data, the equation of state, and the mixture model are useful in showing various trends and inconsistencies.

Figure 10 shows comparisons of densities calculated with the equation of state to experimental $p-\rho-T$ data. Several inconsistencies can be seen in the data used in the fit. At 290 K, the data of Michels et al. (1954a) and of Kozlov (1968) differ by an average of 0.1% in density. This offset is also seen between the data of Kozlov and the equation of state throughout most of the data of Kozlov. The data of Howley et al. (1994) are generally consistent with the data of Michels et al., except at temperatures below 200 K. At 155 K, these two data sets differ by 0.1%. Although used in the fit, the scatter in the data of Blanke is within 0.2% in the liquid, and within 0.5% in the vapor. Differences between the overlapping vapor phase data of Romberg (1971) and the equation of state are less than 0.1%, as compared with 0.5% for the data of Blanke. There are no overlapping data in the liquid. Despite the age of the data of Amagat (1893), the equation of state (which was not fit to these data) agrees on average to within 0.3% at pressures between 200 and 300 MPa and temperatures from 270 to 300 K. The only other data in a comparable range are the data of Michels et al. at 270 K up to 228 MPa. The equation of state agrees within 0.1% with these data. Differences between the mixture model and the Amagat data are larger at the highest pressures, however, these pressures are well beyond the limits of the oxygen equation of state.

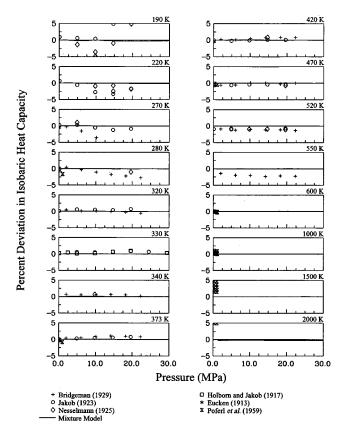


Fig. 13. Comparisons of isobaric heat capacities calculated with the equation of state to experimental data for air.

The deviations of data in the critical region between 130 and 140 K are shown in Fig. 11. Since density deviations tend to be large in the critical region for any substance (because $\partial \rho/\partial p$ at constant temperature approaches infinity at the critical point), this figure shows comparisons of pressures calculated with the equation of state to experimental $p-\rho-T$ data. The scatter between different data sets is within $\pm 0.3\%$ in pressure throughout most of the critical region, with deviations tending to be positive for the data of Blanke (1977) and Michels *et al.* (1954b), and negative or near zero for comparisons with the data of Howley *et al.* (1994). Deviations between the mixture model and the equation of state tend to be less than that with the experimental data, with the mixture model agreeing best with the data of Howley *et al.*

Comparisons of values calculated from the equation of state are shown for isochoric heat capacities in Fig. 12, isobaric heat capacities in Fig. 13, and the speed of sound in Fig. 14. The equation of state agrees very well with the c_v data of Magee (1994) except at the lowest density isochore, 2 mol/dm³, where the uncertainty of the isochoric heat capacity data tend to be highest. Comparisons with the mixture model are quite good, except for the upturns in the vapor phase near the saturation boundaries. Deviations with the isobaric heat capacity data in the vapor phase are generally within 2%, however, both the uncertainty and the scatter in these data are higher than that for the isochoric heat capacity data. Deviations between the equation of state and the mix-

ture model are nearly negligible, even at 2000 K. Comparisons between the equation of state and the speed of sound data of Ewing and Goodwin (1993) and the data of Young-love and Frederick (1992) are very good at temperatures above 200 K and below 120 K. Near the saturation boundaries and in the critical regions, the deviations tend to be somewhat higher but generally within 1%. The mixture model follows the data of Younglove and Frederick more closely than does the equation of state in these regions.

The differences in the second virial coefficients are shown in Fig. 15. Differences between the data of Romberg (1971) at low temperatures with the equation of state tend to be positive, whereas differences between the graphically determined data from Romberg given here in Table 7 tend to be more scattered, generally within $\pm 1 \text{ cm}^3/\text{mol}$ (about 0.5% at 90 K) except at the lowest temperature. This scatter is due to the difficulty of determining the second virial coefficient as the uncertainty of $p-\rho-T$ data increases at low pressures and temperatures as shown in Fig. 4. The second virial coefficients calculated from the mixture model are not as accurate as those calculated from the air equation of state. Values of the second virial coefficient for oxygen calculated from the equation of state of Schmidt and Wagner (1985) show a minimum at 75 K, and calculated values become positive at temperatures below 58 K. This low temperature behavior is unrealistic and reduces the accuracy of virial coefficients calculated using the mixture model in spite of the fact that the calculated values from Schmidt and Wagner are within the uncertainty of available experimental data.

Figure 16 shows calculated dew and bubble-line densities in the critical region reported by Jacobsen *et al.* (1990b) determined using a Leung–Griffiths model for ternary systems. As shown in Fig. 16, the equation of state developed in this work represents the calculated data of Jacobsen *et al.* (1990b) more accurately than the previous equation of Jacobsen *et al.* (1992). Densities were calculated from each equation of state at the temperature and bubble or dew-point pressure of the equation. This pressure was calculated from the respective ancillary equation for each equation of state. Dew and bubble-point states at the air composition calculated from the mixture model as described in Sec. 5.2. are also shown in the figure.

Figure 17 shows the percent deviation in density between values calculated with the equation of state for air and the properties of air predicted from nitrogen data by the methods described in Sec. 4.2 and used in the development of the air equation of state. At temperatures above 1000 K where deviations between the nitrogen equation of state and the experimental data for nitrogen exceed 2%, the equation of state for air mimics the trends set by the equation of state for nitrogen [Span *et al.* (2000)] as shown by the calculated data points in Fig. 17.

6.3. Comparisons of the Mixture Model with Experimental Single Phase Data

Summary comparisons of values calculated using the mixture model to $p-\rho-T$ data for mixtures of nitrogen, argon,

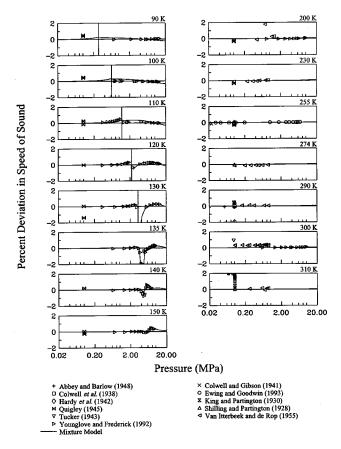


Fig. 14. Comparisons of speeds of sound calculated with the equation of state to experimental data for air.

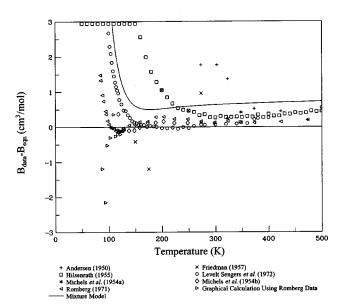


Fig. 15. Comparisons of second virial coefficients calculated with the equation of state to experimental data for air.

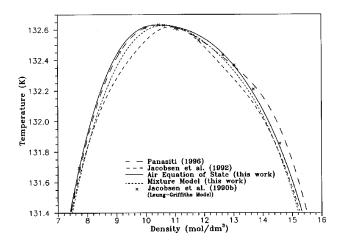


Fig. 16. Comparisons of critical region phase boundaries calculated with the equation of state to predicted data for air.

and oxygen are given in Table 8; this table indicates the temperature range and composition range for the first component listed for each data set. The bubble-point pressures, calculated from the model, were assumed for several data sets for the nitrogen—oxygen and argon—oxygen systems where the pressures were not included in the published data. Comparisons of densities calculated from the mixture model to experimental data are shown in Fig. 18 for the nitrogen—argon binary mixture and in Fig. 19 for the nitrogen—oxygen and argon—oxygen binary mixtures.

The Maslennikova *et al.* (1979) data for the nitrogenargon mixture extend to 800 MPa and deviations are on average within 0.3%. From this and because of the generalized nature of the model, calculated densities should be within 1.0% up to 800 MPa for the nitrogen—oxygen and argon—oxygen systems where no data exist. The magnitudes of the deviations for most of the nitrogen—argon data are similar to those for the extended corresponding states models of Clarke *et al.* (1994) and version 9.08 of the NIST14 database [Friend (1992)].

Very few data exist for the nitrogen-oxygen and argon-oxygen mixtures. For the nitrogen-oxygen mixture, air data and the data of Pool *et al.* (1962) were used to determine the parameters for the mixture model. Although the temperature range of experimental $p-\rho-T$ data for the argon-oxygen system is only 70-90 K with pressures up to 0.2 MPa in the liquid phase, the model should be valid for all temperatures and pressures for argon-oxygen mixtures within the ranges of experimental data for the nitrogen-oxygen and nitrogen-argon systems due to the predictive nature of the model. From these comparisons, we estimate that the uncertainty of $p-\rho-T$ calculations in the extended range for argon-oxygen mixtures is within 0.5%. This estimate is based additionally on comparisons to the large amount of VLE data available over the entire two-phase range for this mixture.

6.4. Comparisons of the Mixture Model with Experimental VLE Data

Comparisons of bubble-point pressures calculated with the mixture model to experimental data are shown in Fig. 20 for

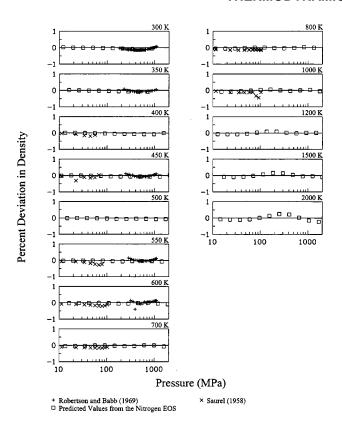


Fig. 17. Comparisons of densities calculated with the equation of state to calculated $p-\rho-T$ data estimated from nitrogen data.

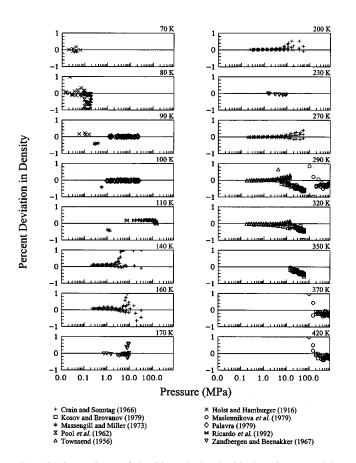


Fig. 18. Comparisons of densities calculated with the mixture model to experimental data for the nitrogen-argon binary mixture.

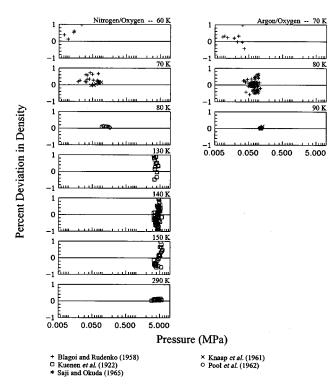


Fig. 19. Comparisons of densities calculated with the mixture model to experimental data for the nitrogen-oxygen and argon-oxygen binary mixtures.

the nitrogen—argon binary mixture, Fig. 21 for the nitrogen—oxygen binary mixture, Fig. 22 for the argon—oxygen binary mixture, and Fig. 23 for the nitrogen—argon—oxygen ternary mixture. The average deviations in bubble-point pressure for the VLE data of Wilson *et al.* (1965) for the nitrogen—argon, nitrogen—oxygen, argon—oxygen, and nitrogen—argon—oxygen mixtures are within 0.8% and are nearly the same as those for the model developed by Lemmon (1996). However, the deviations between the mixture model and the lower temperature nitrogen—oxygen data of Duncan and Staveley (1966) and Armstrong *et al.* (1955) are substantially lower in the new model reported here, and the average deviations for these data sets are about 1.6% and 0.8%, respectively, about 50% smaller than those for the model of Lemmon (1996).

7. Estimated Uncertainty of Calculated Properties

7.1. Characteristic Curves of Air

Plots of constant property lines on various thermodynamic coordinates are useful in assessing the behavior of the equation of state in regions where there are no accurate experimental results for the corresponding property. The equation of state for air developed here was used to produce plots of temperature against isochoric heat capacity (Fig. 24), isobaric heat capacity (Fig. 25), and speed of sound (Fig. 26). As mentioned in Sec. 4.8, analytical methods in addition to calculating the Hugoniot curve are needed to determine the extrapolation behavior of an equation of state. Figure 27

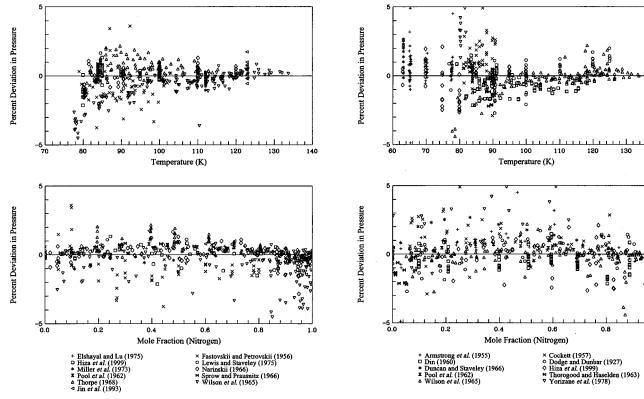


Fig. 20. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for the nitrogen—argon binary mixture.

Fig. 21. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for the nitrogen—oxygen binary mixture.

shows a pressure versus density plot along isotherms. This plot indicates that the equation of state presented here exhibits reasonable extrapolation behavior at high pressures and densities.

Plots of certain "ideal curves" are useful in assessing the behavior of an equation of state [Deiters and de Reuck (1997), Span and Wagner (1997), Span (2000)]. The characteristic curves considered in this work are the Boyle curve, given by the equation

$$\left(\frac{\partial Z}{\partial v}\right)_T = 0,\tag{87}$$

the Joule-Thomson inversion curve,

$$\left(\frac{\partial Z}{\partial T}\right)_{p} = 0, \tag{88}$$

or

$$\left(\frac{\partial T}{\partial p}\right)_h = 0,\tag{89}$$

and the Joule inversion curve

$$\left(\frac{\partial Z}{\partial T}\right)_{v} = 0. \tag{90}$$

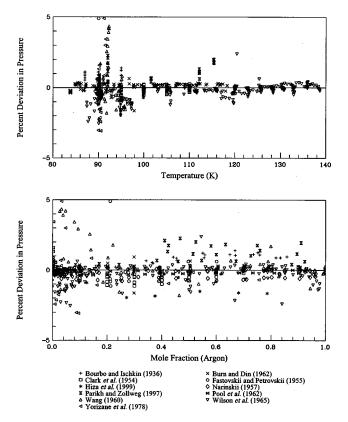


Fig. 22. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for the argon-oxygen binary mixture.

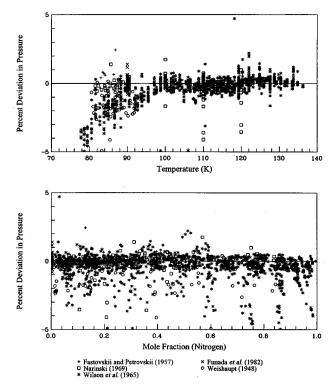


Fig. 23. Comparisons of bubble-point pressures calculated with the mixture model to experimental data for the nitrogen-argon-oxygen ternary mixture.

Figure 28 illustrates these characteristic curves for the equation of state for air. Although the curves in Fig. 28 do not provide numerical information, reasonable shapes of these curves as shown indicate qualitatively correct extrapolation behavior of the equation of state.

7.2. Uncertainty of the Equation of State for Air and of the Mixture Model

The uncertainties of the models presented here have been estimated by comparing calculated results with experimental data, as summarized in Sec. 6, and assessing the data them-

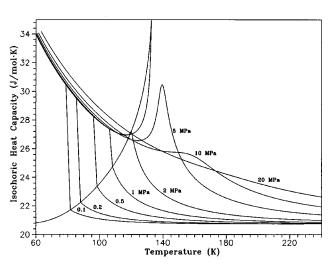


Fig. 24. Isochoric heat capacity versus temperature diagram for air.

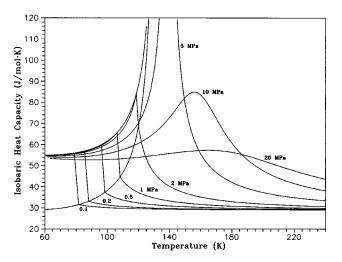


Fig. 25. Isobaric heat capacity versus temperature diagram for air.

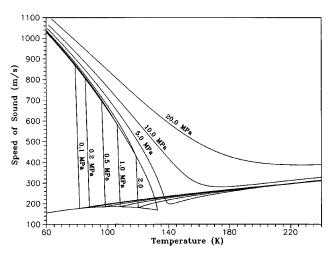


Fig. 26. Speed of sound versus temperature diagram for air.

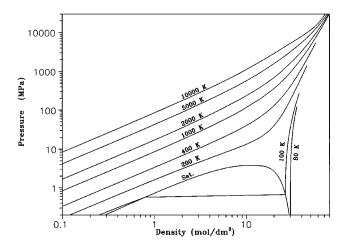


Fig. 27. Pressure versus density diagram for air.

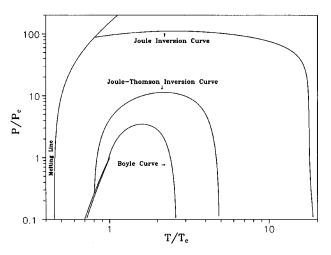


Fig. 28. Characteristic curves for air.

selves according to the quoted experimental uncertainties and the mutual consistency of data obtained from different sources. A general knowledge of the behavior of thermodynamic surfaces and substantial experience with other correlating equations for well-studied fluids have also been applied in the estimation of uncertainties. The uncertainties represent expanded combined uncertainties where a normal distribution of errors is assumed and a coverage factor of 2 has been applied (equivalent to a confidence level of about 95%). This means that if a uniform distribution of state points is considered within the range specified below, no more than 2% of values for a given property will deviate from the physical value by more than the specified uncertainty.

For the equation of state of standard air, the region for which sufficient data are available to establish reliable uncertainty estimates ranges from the solidification point to 873 K at pressures up to 70 MPa, except in the critical region from 130 to 134 K and 9 to 13 mol/dm³. The estimated uncertainty of density values calculated using the equation of state for air is 0.1%. The estimated uncertainty of speed of sound values is within 0.2% based on comparisons with the data of Younglove and Frederick (1992), Van Itterbeek and de Rop (1955), and Ewing and Goodwin (1993). The estimated uncertainty is 1% for calculated values of heat capacity based upon comparisons to the experimental data of Magee (1994). In the critical region defined above, the uncertainty in pressure calculations is estimated to be 0.3%. Outside the range of the primary experimental data for air (T>870 K, p)>70 MPa) and at temperatures less than 2000 K and pressures less than 2000 MPa, the estimated uncertainty in predicted densities is 1.0%. Without experimental data, the uncertainty of extrapolated properties cannot be verified.

The uncertainty of the mixture model reported here for arbitrary compositions of nitrogen, oxygen, and argon is within 0.1% in density, 0.2% for the speed of sound, and 1% in heat capacity for both binary and ternary mixtures in the same range given for the air equation of state. The uncertainty of calculated dew and bubble-point pressures is within

TABLE 17. Regions of stated uncertainty of the mixture model

| Mixture | Temperature range (K) | Maximum pressure (MPa) |
|-------------------------|-----------------------|---------------------------|
| Nitrogen-Argon | 70-420 | 800 |
| Nitrogen-Oxygen and Air | 60-870 | 100 |
| Argon-Oxygen | 70-90 | 0.2 |
| | $70-400^{a}$ | 100^{a} |

^aNo data are available to verify this range, however, the uncertainty of the equation should be at least 0.5% in density for the range stated due to the predictive nature of the model.

1%. The mixtures and ranges for which calculated properties have been verified by experimental data to have deviations within these limits are listed in Table 17. In regions where there are no binary mixture data, the uncertainty is estimated to be of the same magnitude. However, these estimates cannot be verified until experimental data are available to support these estimates.

Because of the generalized and predictive nature of the mixture model, calculated densities of mixtures of nitrogen, argon, and oxygen extrapolated to temperatures up to 1000 K and pressures to 100 MPa have an estimated uncertainty of 0.5%. Although the equation of state for oxygen by Schmidt and Wagner (1985) is valid only to temperatures of 300 K, the original work of Lemmon (1996) demonstrated that the oxygen equation extrapolates to temperatures up to 1000 K within 0.5% in density through comparisons with the Kozlov (1968) data. As new measurements become available, they will refine the uncertainty estimates in regions not covered by experimental data, and will enable continued evaluation and optimization of the mixture model.

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9. Appendix: Tables of Properties of Air

9.1. Representative Tables of Thermodynamic Properties of Air

Properties of air along the dew and bubble-point curves were calculated using pressures from the appropriate ancillary equation at the specified temperature. The density was calculated using the air equation of state for the input temperature and pressure. Dew and bubble-point entries for the isobar tables were calculated using pressure as the input to Eq. (1) to determine the corresponding temperatures. The densities for these entries were calculated using the temperature and pressure as input variables in the equation of state.

TABLE A1. Thermodynamic properties of air on the dew and bubble lines

| Temperature | Pressure | Density (mol/dm³) | Enthalpy (J/mol) | Entropy J/(mol-K) | c_v J/(mol-K) | c_p J/(mol-K) | Speed of sound (m/s) |
|--------------------|----------------------|----------------------|---------------------|----------------------|-----------------|-----------------|----------------------------|
| (K) | (MPa) | | | | | | |
| 59.75 ^a | 0.005 265 | 33.067 | -4713.1 | 70.902 | 34.01 | 55.06 | 1030.3 |
| 59.75 ^a | 0.002 43 | 0.004 91 | 1721.6 | 182.97 | 20.80 | 29.22 | 154.8 |
| 60 | 0.005 55 | 33.031 | -4699.3 | 71.131 | 33.95 | 55.06 | 1028.3 |
| 60 | 0.002 58 | 0.005 19 | 1728.6 | 182.59 | 20.81 | 29.23 | 155.1 |
| 61 | 0.006 80 | 32.888 | -4644.3 | 72.041 | 33.73 | 55.06 | 1020.3 |
| 61 | 0.003 27 | 0.006 47 | 1756.7 | 181.09 | 20.82 | 29.26 | 156.4 |
| 62 | 0.008 27 | 32.745 | -4589.2 | 72.937 | 33.51 | 55.06 | 1012.2 |
| 62 | 0.004 11 | 0.008 00 | 1784.7 | 179.66 | 20.84 | 29.30 | 157.6 |
| 63 | 0.009 99 | 32.601 | -4534.1 | 73.817 | 33.30 | 55.07 | 1004.0 |
| 63 | 0.005 12 | 0.009 82 | 1812.5 | 178.29 | 20.86 | 29.35 | 158.8 |
| 64 | 0.012 00 | 32.457 | -4478.9 | 74.684 | 33.09 | 55.08 | 995.8 |
| 64 | 0.006 33 | 0.011 95 | 1840.1 | 176.97 | 20.89 | 29.40 | 160.0 |
| 65 | 0.014 32 | 32.312 | -4423.8 | 75.538 | 32.88 | 55.10 | 987.5 |
| 65 | 0.007 76 | 0.014 44 | 1867.5 -4368.6 | 175.71 76.379 | 20.91 | 29.46 | 161.2 |
| 66 66 | 0.016 99 0.009 44 | 32.166 0.017 33 | -4368.6 1894.7 | 76.379 174.50 | 32.68 20.94 | 55.12 29.52 | 979.1 162.3 |
| 67 | 0.020 04 | 32.020 | -4313.4 | 77.208 | 32.49 | 55.15 | 970.7 |
| 67 | 0.020 04 | 0.020 66 | -4313.4 1921.7 | 173.34 | 20.97 | 29.59 | 163.4 |
| 68 | 0.011 42 | 31.873 | -4258.2 | 78.025 | 32.29 | 55.18 | 962.2 |
| 68 | 0.023 32 | 0.024 48 | 1948.5 | 172.22 | 21.00 | 29.66 | 164.5 |
| 69 | 0.013 / 1 | 31.725 | -4202.9 | 78.830 | 32.11 | 55.22 | 953.7 |
| 69 | 0.016 37 | 0.028 84 | 1974.9 | 171.15 | 21.04 | 29.75 | 165.6 |
| 70 | 0.031 91 | 31.576 | -4147.6 | 79.624 | 31.92 | 55.27 | 945.1 |
| 70 | 0.031 71 | 0.033 79 | 2001.1 | 170.12 | 21.07 | 29.84 | 166.7 |
| 71 | 0.036 91 | 31.427 | -4092.2 | 80.408 | 31.74 | 55.32 | 936.4 |
| 71 | 0.022 93 | 0.039 38 | 2027.0 | 169.13 | 21.11 | 29.93 | 167.7 |
| 72 | 0.042 50 | 31.277 | -4036.7 | 81.181 | 31.56 | 55.38 | 927.7 |
| 72 | 0.026 92 | 0.045 66 | 2052.6 | 168.17 | 21.16 | 30.04 | 168.7 |
| 73 | 0.048 73 | 31.126 | -3981.2 | 81.944 | 31.39 | 55.44 | 918.9 |
| 73 | 0.031 44 | 0.052 70 | 2077.8 | 167.25 | 21.20 | 30.15 | 169.7 |
| 74 | 0.055 66 | 30.974 | -3925.6 | 82.698 | 31.22 | 55.51 | 910.0 |
| 74 | 0.036 55 | 0.060 55 | 2102.7 | 166.36 | 21.25 | 30.28 | 170.6 |
| 75 | 0.063 33 | 30.821 | -3869.8 | 83.442 | 31.05 | 55.59 | 901.1 |
| 75 | 0.042 28 | 0.069 27 | 2127.2 | 165.50 | 21.30 | 30.41 | 171.6 |
| 76 | 0.071 79 | 30.668 | -3814.0 | 84.178 | 30.89 | 55.68 | 892.1 |
| 76 | 0.048 70 | 0.078 92 | 2151.3 | 164.67 | 21.36 | 30.55 | 172.5 |
| 77 | 0.081 09 | 30.513 | -3758.1 | 84.905 | 30.73 | 55.78 | 883.0 |
| 77 | 0.055 86 | 0.089 56 | 2175.0 | 163.87 | 21.41 | 30.70 | 173.4 |
| 78 | 0.091 29 | 30.357 | -3702.1 | 85.624 | 30.57 | 55.88 | 873.9 |
| 78 | 0.063 81 | 0.101 27 | 2198.3 | 163.09 | 21.47 | 30.86 | 174.2 |
| 79 | 0.102 45 | 30.200 | -3645.9 | 86.334 | 30.41 | 56.00 | 864.7 |
| 79 | 0.072 61 | 0.114 10 | 2221.2 | 162.34 | 21.54 | 31.04 | 175.1 |
| 80 | 0.114 62 | 30.042 | -3589.6 | 87.037 | 30.26 | 56.12 | 855.4 |
| 80 | 0.082 32 | 0.128 13 | 2243.6 | 161.61 | 21.60 | 31.22 | 175.8 |
| 81 | 0.127 85 | 29.883 | -3533.2 | 87.733 | 30.11 | 56.26 | 846.1 |
| 81 | 0.093 00 | 0.143 43 | 2265.5 | 160.91 | 21.67 | 31.41 | 176.6 |
| 82 | 0.142 21 | 29.722 | -3476.6 | 88.421 | 29.97 | 56.40 | 836.7 |
| 82 | 0.104 71 | 0.160 06 | 2286.9 | 160.22 | 21.75 | 31.62 | 177.4 |
| 83 | 0.157 75 | 29.560 | -3419.8 | 89.103 | 29.83 | 56.56 | 827.2 |
| 83 | 0.117 51 | 0.178 11 | 2307.8 | 159.56 | 21.82 | 31.84 | 178.1 |
| 84 | 0.174 53 | 29.397 | -3362.9 | 89.778 | 29.69 | 56.72 | 817.6 |
| 84 | 0.131 47 | 0.197 65 | 2328.1 | 158.91 | 21.90 | 32.07 | 178.8 |
| 85 85 | 0.192 62 | 29.232 | -3305.8 | 90.447 | 29.55 | 56.90 32.32 | 808.0 |
| 85 86 | 0.146 65 | 0.218 75 | 2347.9 | 158.28 | 21.98 | 32.32 | 179.4 |
| 86 86 | 0.212 07 | 29.066 | -3248.4 | 91.110 157.67 | 29.42 | 57.09 32.58 | 798.2 |
| 86 87 | 0.163 12 0.232 95 | 0.241 50 28.898 | 2367.1 -3190.9 | 157.67 | 22.07 | 32.58 57.30 | 180.0 788.4 |
| 87 87 | 0.232 95 | 28.898 0.265 98 | - 3190.9 2385.8 | 91.767 157.07 | 29.29 22.16 | 57.30 32.85 | /88.4 180.6 |
| | 0.180 94 0.255 31 | | | | | | |
| 88 88 | 0.255 31 0.200 18 | 28.729 0.292 28 | -3133.1 2403.8 | 92.418 | 29.16 22.25 | 57.52 | 778.6 181.2 |
| 88 89 | 0.200 18 | 0.292 28 28.558 | -3075.1 | 156.49 93.065 | 22.25 29.03 | 33.14 57.76 | 768.6 |
| 89 89 | 0.279 22 0.220 91 | 0.320 48 | - 3075.1 2421.2 | 93.065 155.92 | 29.03 22.34 | 33.45 | 181.7 |
| | 0.440 91 | U.JZU 40 | 4441.4 | 133.94 | 44.34 | 1141 | 101/ |

TABLE A1. Thermodynamic properties of air on the dew and bubble lines—Continued

| Tommo t | D | D! | E | Fortuna | _ | _ | Speed |
|-----------------|-------------------|--------------------------------|--------------------|----------------------|--------------------------------|--------------------------------|-------------------|
| Temperature (K) | Pressure (MPa) | Density (mol/dm ³) | Enthalpy (J/mol) | Entropy J/(mol-K) | $\frac{c_v}{J/(\text{mol-K})}$ | $\frac{c_p}{J/(\text{mol-K})}$ | of sound (m/s) |
| 90 | 0.243 20 | 0.350 68 | 2437.9 | 155.36 | 22.44 | 33.77 | 182.2 |
| 91 | 0.331 96 | 28.210 | -2958.2 | 94.342 | 28.79 | 58.28 | 748.4 |
| 91 | 0.267 12 | 0.382 98 | 2454.0 | 154.82 | 22.54 | 34.11 | 182.6 |
| 92 | 0.360 91 | 28.033 | -2899.4 | 94.974 | 28.68 | 58.57 | 738.2 |
| 92 | 0.292 73 | 0.417 47 | 2469.3 | 154.29 | 22.64 | 34.47 | 183.1 |
| 93 | 0.391 66 | 27.854 | -2840.2 | 95.602 | 28.56 | 58.87 | 727.9 |
| 93 | 0.320 11 | 0.454 26 | 2484.0 | 153.77 | 22.74 | 34.85 | 183.5 |
| 94 | 0.424 29 | 27.673 | -2780.8 | 96.225 | 28.45 | 59.20 | 717.5 |
| 94 | 0.349 34 | 0.493 45 | 2497.9 | 153.26 | 22.85 | 35.26 | 183.8 |
| 95 | 0.458 86 | 27.489 | -2720.9 | 96.845 | 28.35 | 59.55 | 707.0 |
| 95 | 0.380 47 | 0.535 17 | 2511.0 | 152.75 | 22.96 | 35.68 | 184.2 |
| 96 | 0.495 43 | 27.304 | -2660.8 | 97.461 | 28.24 | 59.93 | 696.5 |
| 96 | 0.413 59 | 0.579 53 | 2523.3 | 152.26 | 23.08 | 36.13 | 184.5 |
| 97 | 0.534 08 | 27.115 | -2600.2 | 98.074 | 28.14 | 60.33 | 685.8 |
| 97 | 0.448 78 | 0.626 67 | 2534.8 | 151.78 | 23.20 | 36.61 | 184.7 |
| 98 | 0.574 86 | 26.924 | -2539.2 | 98.684 | 28.04 | 60.76 | 675.0 |
| 98 | 0.486 09 | 0.67671 | 2545.4 | 151.30 | 23.32 | 37.12 | 184.9 |
| 99 | 0.617 86 | 26.730 | -2477.8 | 99.291 | 27.95 | 61.22 | 664.2 |
| 99 | 0.525 62 | 0.729 80 | 2555.2 | 150.83 | 23.44 | 37.65 | 185.1 |
| 100 | 0.663 13 | 26.533 | -2416.0 | 99.896 | 27.86 | 61.71 | 653.3 |
| 100 | 0.567 42 | 0.786 09 | 2564.0 | 150.36 | 23.57 | 38.23 | 185.3 |
| 101 | 0.710 74 | 26.333 | -2353.6 | 100.50 | 27.77 | 62.23 | 642.2 |
| 101 | 0.611 59 | 0.845 75 | 2571.9 | 149.90 | 23.70 | 38.83 | 185.4 |
| 102 | 0.760 77 | 26.130 | -2290.8 | 101.10 | 27.68 | 62.80 | 631.1 |
| 102 | 0.658 20 | 0.908 95 | 2578.8 | 149.44 | 23.83 | 39.48 | 185.5 |
| 103 | 0.813 29 | 25.923 | - 2227.4 | 101.70 | 27.60 | 63.40 | 619.8 |
| 103 | 0.707 32 | 0.975 87 | 2584.6 | 148.99 | 23.97 | 40.18 | 185.6 |
| 103 | 0.868 36 | 25.713 | -2163.4 | 102.29 | 27.53 | 64.05 | 608.5 |
| 104 | 0.759 03 | 1.0467 | 2589.4 | 148.54 | 24.11 | 40.92 | 185.6 |
| 105 | 0.926 06 | 25.499 | -2098.9 | 102.89 | 27.45 | 64.75 | 597.1 |
| 105 | 0.926 06 | 1.1217 | - 2098.9 2593.0 | 148.10 | 24.26 | 41.71 | 185.5 |
| 106 | 0.813 41 | 25.281 | - 2033.7 | 103.49 | 27.38 | 65.51 | 585.5 |
| 106 | 0.870 55 | 1.2011 | 2595.5 | 147.65 | 24.41 | 42.57 | 185.5 |
| 106 | 1.049 61 | 25.058 | 2393.3 1967.8 | 104.08 | 27.32 | 42.57 66.32 | 573.9 |
| | | | | 104.08 | | | |
| 107 | 0.930 52 | 1.2852 | 2596.7 | | 24.56 | 43.49 | 185.4 |
| 108 | 1.115 61 | 24.831 | - 1901.2 | 104.68 | 27.26 | 67.21 | 562.1 |
| 108 | 0.993 40 | 1.3742 | 2596.6 | 146.77 | 24.72 | 44.49 | 185.2 |
| 109 | 1.184 53 | 24.598 | - 1833.8 | 105.27 | 27.20 | 68.16 | 550.2 |
| 109 | 1.059 28 | 1.4684 | 2595.1 | 146.33 | 24.89 | 45.57 | 185.1 |
| 110 | 1.256 42 | 24.361 | - 1765.6 | 105.87 | 27.15 | 69.20 | 538.2 |
| 110 | 1.128 24 | 1.5682 | 2592.2 | 145.89 | 25.06 | 46.75 | 184.9 |
| 111 | 1.331 38 | 24.118 | - 1696.5 | 106.47 | 27.10 | 70.34 | 526.1 |
| 111 | 1.200 36 | 1.6740 | 2587.7 | 145.45 | 25.23 | 48.04 | 184.6 |
| 112 | 1.409 47 | 23.868 | - 1626.4 | 107.06 | 27.06 | 71.58 | 513.9 |
| 112 | 1.275 74 | 1.7862 | 2581.7 | 145.00 | 25.42 | 49.45 | 184.3 |
| 113 | 1.490 77 | 23.613 | - 1555.4 | 107.67 | 27.03 | 72.95 | 501.5 |
| 113 | 1.354 45 | 1.9053 | 2573.8 | 144.55 | 25.61 | 51.01 | 184.0 |
| 114 | 1.575 34 | 23.350 | -1483.2 | 108.27 | 27.00 | 74.46 | 489.0 |
| 114 | 1.436 60 | 2.0318 | 2564.2 | 144.10 | 25.81 | 52.73 | 183.6 |
| 115 | 1.663 27 | 23.080 | -1409.9 | 108.88 | 26.98 | 76.13 | 476.3 |
| 115 | 1.522 26 | 2.1664 | 2552.5 | 143.64 | 26.01 | 54.64 | 183.2 |
| 116 | 1.754 62 | 22.801 | -1335.2 | 109.49 | 26.97 | 78.00 | 463.5 |
| 116 | 1.611 54 | 2.3097 | 2538.7 | 143.18 | 26.23 | 56.79 | 182.7 |
| 117 | 1.849 47 | 22.514 | -1259.2 | 110.11 | 26.96 | 80.09 | 450.5 |
| 117 | 1.704 52 | 2.4625 | 2522.7 | 142.70 | 26.46 | 59.21 | 182.2 |
| 118 | 1.947 89 | 22.217 | -1181.6 | 110.73 | 26.97 | 82.46 | 437.3 |
| 118 | 1.801 32 | 2.6259 | 2504.0 | 142.22 | 26.70 | 61.96 | 181.7 |
| 119 | 2.049 95 | 21.908 | -1102.4 | 111.36 | 26.98 | 85.16 | 423.9 |
| 119 | 1.902 02 | 2.8009 | 2482.7 | 141.73 | 26.96 | 65.10 | 181.1 |
| 120 | 2.155 73 | 21.588 | -1021.2 | 112.00 | 27.01 | 88.28 | 410.2 |
| 120 | 2.006 74 | 2.9889 | 2458.3 | 141.22 | 27.23 | 68.74 | 180.4 |
| 121 | 2.265 29 | 21.253 | -937.90 | 112.65 | 27.05 | 91.92 | 396.3 |
| | | | | | | | |

TABLE A1. Thermodynamic properties of air on the dew and bubble lines—Continued

| Temperature (K) | Pressure (MPa) | Density (mol/dm³) | Enthalpy (J/mol) | Entropy J/(mol-K) | c_v J/(mol-K) | c_p J/(mol-K) | Speed of sound (m/s) |
|-----------------------|-------------------|----------------------|---------------------|----------------------|-----------------|-----------------|----------------------------|
| 122 | 2.378 71 | 20.903 | - 852.17 | 113.31 | 27.11 | 96.23 | 382.0 |
| 122 | 2.228 71 | 3.4103 | 2399.1 | 140.16 | 27.82 | 78.02 | 179.1 |
| 123 | 2.496 04 | 20.534 | -763.63 | 113.98 | 27.19 | 101.4 | 367.4 |
| 123 | 2.346 20 | 3.6481 | 2363.3 | 139.59 | 28.16 | 84.05 | 178.3 |
| 124 | 2.617 34 | 20.144 | -671.82 | 114.68 | 27.30 | 107.8 | 352.3 |
| 124 | 2.468 23 | 3.9078 | 2322.6 | 139.00 | 28.52 | 91.43 | 177.5 |
| 125 | 2.742 67 | 19.727 | -576.07 | 115.40 | 27.44 | 115.9 | 336.7 |
| 125 | 2.594 95 | 4.1934 | 2276.3 | 138.38 | 28.91 | 100.6 | 176.7 |
| 126 | 2.872 07 | 19.278 | -475.47 | 116.15 | 27.62 | 126.5 | 320.4 |
| 126 | 2.726 56 | 4.5101 | 2223.2 | 137.71 | 29.34 | 112.4 | 175.8 |
| 127 | 3.005 54 | 18.788 | -368.72 | 116.93 | 27.85 | 140.9 | 303.2 |
| 127 | 2.863 31 | 4.8653 | 2161.9 | 137.00 | 29.83 | 128.0 | 174.9 |
| 128 | 3.143 06 | 18.242 | -253.75 | 117.78 | 28.17 | 161.9 | 285.0 |
| 128 | 3.005 51 | 5.2697 | 2090.3 | 136.22 | 30.37 | 149.6 | 174.0 |
| 129 | 3.284 48 | 17.616 | -127.06 | 118.70 | 28.61 | 195.2 | 265.4 |
| 129 | 3.153 61 | 5.7405 | 2004.9 | 135.34 | 30.99 | 181.3 | 173.0 |
| 130 | 3.429 47 | 16.863 | 18.109 | 119.76 | 29.24 | 256.2 | 243.7 |
| 130 | 3.308 35 | 6.3074 | 1899.9 | 134.33 | 31.73 | 232.6 | 171.9 |
| 131 | 3.576 98 | 15.869 | 198.35 | 121.07 | 30.27 | 401.5 | 219.1 |
| 131 | 3.471 16 | 7.0343 | 1763.0 | 133.10 | 32.62 | 329.9 | 170.8 |
| 132 | 3.722 84 | 14.198 | 478.83 | 123.13 | 32.34 | 1015. | 189.1 |
| 132 | 3.646 25 | 8.1273 | 1553.9 | 131.33 | 33.81 | 598.0 | 169.4 |
| 132.6312 ^b | 3.785 02 | 10.4477 | 1111.3 | 127.87 | 35.26 | 2196. | 168.0 |

^aSolidification point. ^bMaxcondentherm.

TABLE A2. Thermodynamic properties of air

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|--------------------|------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| | | | 0.101 325 MP | a isobar | | | |
| 59.77 | 33.069 | -4713.1 | -4710.0 | 70.905 | 34.01 | 55.05 | 1030.6 |
| 60 | 33.036 | -4700.3 | -4697.2 | 71.119 | 33.96 | 55.05 | 1028.8 |
| 62 | 32.750 | -4590.2 | -4587.1 | 72.924 | 33.52 | 55.05 | 1012.6 |
| 64 | 32.462 | -4480.1 | -4477.0 | 74.672 | 33.09 | 55.07 | 996.3 |
| 66 | 32.171 | -4369.9 | -4366.8 | 76.367 | 32.69 | 55.11 | 979.6 |
| 68 | 31.878 | -4259.7 | -4256.5 | 78.013 | 32.30 | 55.17 | 962.7 |
| 70 | 31.581 | -4149.3 | -4146.1 | 79.614 | 31.92 | 55.25 | 945.5 |
| 72 | 31.281 | -4038.7 | -4035.5 | 81.172 | 31.56 | 55.37 | 928. |
| 74 | 30.978 | -3927.9 | -3924.6 | 82.691 | 31.22 | 55.51 | 910.4 |
| 76 | 30.670 | -3816.7 | -3813.4 | 84.173 | 30.89 | 55.68 | 892.3 |
| 78 | 30.358 | -3705.2 | -3701.9 | 85.622 | 30.57 | 55.88 | 874.0 |
| 78.90 | 30.215 | -3654.7 | -3651.4 | 86.266 | 30.43 | 55.99 | 865.0 |
| 81.72 | 0.155 27 | 1628.3 | 2280.9 | 160.41 | 21.73 | 31.56 | 177.2 |
| 82 | 0.154 67 | 1634.6 | 2289.8 | 160.52 | 21.71 | 31.52 | 177. |
| 84 | 0.150 53 | 1679.4 | 2352.5 | 161.28 | 21.58 | 31.26 | 180.0 |
| 86 | 0.146 63 | 1723.8 | 2414.8 | 162.01 | 21.48 | 31.04 | 182. |
| 88 | 0.142 95 | 1767.9 | 2476.7 | 162.72 | 21.40 | 30.85 | 184.3 |
| 90 | 0.139 47 | 1811.7 | 2538.2 | 163.41 | 21.33 | 30.69 | 187. |
| 92 94 | 0.136 17 0.133 03 | 1855.4 1898.8 | 2599.5 2660.4 | 164.09 164.74 | 21.27 | 30.55 30.42 | 189.4 191. |
| 96 | 0.130 05 | 1942.1 | 2721.2 | 165.38 | 21.22 21.17 | 30.32 | 193. |
| 98 | 0.130 03 | 1942.1 | 2781.7 | 166.00 | 21.17 | 30.22 | 196. |
| 100 | 0.124 49 | 2028.2 | 2842.1 | 166.61 | 21.13 | 30.13 | 198. |
| 102 | 0.121 90 | 2071.0 | 2902.2 | 167.21 | 21.09 | 30.05 | 200. |
| 104 | 0.121 90 | 2113.8 | 2962.3 | 167.79 | 21.03 | 29.98 | 200. |
| 106 | 0.117 04 | 2156.4 | 3022.2 | 168.36 | 21.03 | 29.92 | 204. |
| 108 | 0.114 76 | 2199.0 | 3082.0 | 168.92 | 20.98 | 29.86 | 206. |
| 110 | 0.112 57 | 2241.5 | 3141.6 | 169.47 | 20.96 | 29.81 | 208. |
| 112 | 0.110 47 | 2284.0 | 3201.2 | 170.01 | 20.94 | 29.76 | 210.: |
| 114 | 0.108 44 | 2326.3 | 3260.7 | 170.53 | 20.93 | 29.72 | 212. |
| 116 | 0.106 49 | 2368.6 | 3320.1 | 171.05 | 20.91 | 29.68 | 214. |
| 118 | 0.104 62 | 2410.9 | 3379.4 | 171.56 | 20.90 | 29.64 | 216.4 |
| 120 | 0.102 81 | 2453.1 | 3438.7 | 172.05 | 20.89 | 29.61 | 218. |
| 122 | 0.101 06 | 2495.3 | 3497.8 | 172.54 | 20.87 | 29.58 | 220. |
| 124 | 0.099 377 | 2537.4 | 3557.0 | 173.02 | 20.86 | 29.55 | 222. |
| 126 | 0.097 748 | 2579.5 | 3616.0 | 173.50 | 20.85 | 29.52 | 223. |
| 128 | 0.096 174 | 2621.5 | 3675.1 | 173.96 | 20.84 | 29.50 | 225. |
| 130 | 0.094 650 | 2663.5 | 3734.0 | 174.42 | 20.84 | 29.48 | 227.0 |
| 132 | 0.093 175 | 2705.5 | 3793.0 | 174.87 | 20.83 | 29.45 | 229. |
| 134 | 0.091 747 | 2747.5 | 3851.9 | 175.31 | 20.82 | 29.43 | 231. |
| 136 | 0.090 363 | 2789.4 | 3910.7 | 175.75 | 20.82 | 29.42 | 232.9 |
| 138 | 0.089 021 | 2831.3 | 3969.5 | 176.18 | 20.81 | 29.40 | 234. |
| 140 | 0.087 718 | 2873.2 | 4028.3 | 176.60 | 20.81 | 29.38 | 236. |
| 142 | 0.086 455 | 2915.1 | 4087.1 | 177.02 | 20.80 | 29.37 | 238. |
| 144 | 0.085 227 | 2956.9 | 4145.8 | 177.43 | 20.80 | 29.35 | 239. |
| 146 | 0.084 035 | 2998.7 | 4204.5 | 177.83 | 20.79 | 29.34 | 241. |
| 148 | 0.082 877 | 3040.5 | 4263.1 | 178.23 | 20.79 | 29.33 | 243. |
| 150 | 0.081 750 | 3082.3 | 4321.8 | 178.62 | 20.78 | 29.32 | 244. |
| 155 | 0.079 065 | 3186.8 | 4468.3 | 179.59 | 20.78 | 29.29 | 249. |
| 160 | 0.076 553 | 3291.1 | 4614.7 | 180.51 | 20.77 | 29.27 | 253. |
| 165 170 | 0.074 198 | 3395.4 3499.6 | 4761.0 4907.2 | 181.41 182.29 | 20.76 20.76 | 29.25 29.23 | 257. 261. |
| 170 | 0.071 985 | | | | | | 261. 264. |
| 180 | 0.069 902 0.067 937 | 3603.8 3707.9 | 5053.3 5199.3 | 183.13 183.96 | 20.75 20.75 | 29.22 29.20 | 264. |
| 185 | 0.067 937 | 3812.0 | 5345.3 | 183.96 184.76 | 20.75 | 29.20 | 208. 272. |
| 190 | 0.064 324 | 3916.0 | 5491.2 | 185.54 | 20.75 | 29.19 | 272. 276. |
| 195 | 0.062 659 | 4020.0 | 5637.1 | 186.29 | 20.75 | 29.17 | 270. 279. |
| 200 | 0.061 079 | 4124.0 | 5782.9 | 187.03 | 20.73 | 29.16 | 283. |
| 210 | 0.058 147 | 4331.9 | 6074.5 | 188.45 | 20.74 | 29.15 | 290. |
| 220 | 0.055 486 | 4539.8 | 6365.9 | 189.81 | 20.74 | 29.14 | 297. |
| 230 | 0.053 480 | 4747.6 | 6657.3 | 191.11 | 20.74 | 29.13 | 304. |
| | 0.000 | ., 17.0 | 0001.0 | | -0.7 | ->.10 | JUT. |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|-------------|------------------------|--------------------|---------------------|-----------|-----------|-----------|-------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 250 | 0.048 793 | 5163.2 | 7239.9 | 193.53 | 20.75 | 29.13 | 317.1 |
| 260 | 0.046 908 | 5371.1 | 7531.1 | 194.68 | 20.76 | 29.13 | 323.4 |
| 270 | 0.045 164 | 5578.9 | 7822.4 | 195.78 | 20.76 | 29.13 | 329.6 |
| 280 | 0.043 546 | 5786.8 | 8113.7 | 196.84 | 20.77 | 29.13 | 335.6 |
| 290 | 0.042 040 | 5994.8 | 8405.1 | 197.86 | 20.78 | 29.14 | 341.5 |
| 300 | 0.040 634 | 6203.0 | 8696.5 | 198.85 | 20.80 | 29.15 | 347.4 |
| 310 | 0.039 320 | 6411.2 | 8988.1 | 199.80 | 20.81 | 29.16 | 353.1 |
| 320 | 0.038 089 | 6619.5 | 9279.8 | 200.73 | 20.83 | 29.18 | 358.7 |
| 330 | 0.036 932 | 6828.1 | 9571.6 | 201.63 | 20.85 | 29.19 | 364.2 |
| 340 | 0.035 844 | 7036.8 | 9863.6 | 202.50 | 20.87 | 29.21 | 369.7 |
| 350 | 0.034 818 | 7245.7 | 10 156.0 | 203.34 | 20.89 | 29.23 | 375.0 |
| 360 | 0.033 850 | 7454.9 | 10 448.0 | 204.17 | 20.92 | 29.26 | 380.3 |
| 370 | 0.032 933 | 7664.3 | 10 741.0 | 204.97 | 20.94 | 29.28 | 385.4 |
| 380 | 0.032 066 | 7874.0 | 11 034.0 | 205.75 | 20.97 | 29.31 | 390.5 |
| 390 | 0.031 243 | 8084.0 | 11 327.0 | 206.51 | 21.01 | 29.34 | 395.5 |
| 400 | 0.030 461 | 8294.3 | 11 621.0 | 207.26 | 21.04 | 29.38 | 400.5 |
| 450 | 0.027 074 | 9351.8 | 13 094.0 | 210.73 | 21.25 | 29.58 | 424.2 |
| 500 | 0.024 365 | 10 421.0 | 14 579.0 | 213.86 | 21.50 | 29.83 | 446.4 |
| 550 | 0.022 149 | 11 503.0 | 16 078.0 | 216.71 | 21.80 | 30.13 | 467.3 |
| 600 | 0.020 303 | 12 602.0 | 17 592.0 | 219.35 | 22.13 | 30.45 | 487.1 |
| 650 | 0.018 742 | 13 717.0 | 19 123.0 | 221.80 | 22.47 | 30.79 | 505.9 |
| 700 | 0.017 403 | 14 849.0 | 20 671.0 | 224.09 | 22.82 | 31.14 | 523.9 |
| 750 | 0.016 243 | 15 999.0 | 22 237.0 | 226.25 | 23.17 | 31.48 | 541.2 |
| 800 | 0.015 228 | 17 166.0 | 23 820.0 | 228.30 | 23.51 | 31.82 | 557.8 |
| 900 | 0.013 536 | 19 549.0 | 27 035.0 | 232.08 | 24.15 | 32.47 | 589.6 |
| 1000 | 0.012 183 | 21 994.0 | 30 311.0 | 235.53 | 24.73 | 33.05 | 619.6 |
| 1100 | 0.011 075 | 24 494.0 | 33 642.0 | 238.71 | 25.25 | 33.57 | 648.1 |
| 1200 | 0.010 153 | 27 042.0 | 37 022.0 | 241.65 | 25.70 | 34.02 | 675.5 |
| 1300 | 0.009 372 | 29 633.0 | 40 444.0 | 244.39 | 26.10 | 34.42 | 701.7 |
| 1400 | 0.008 703 | 32 261.0 | 43 904.0 | 246.95 | 26.45 | 34.77 | 727.0 |
| 1500 | 0.008 122 | 34 922.0 | 47 397.0 | 249.36 | 26.76 | 35.08 | 751.5 |
| 1600 | 0.007 615 | 37 612.0 | 50 919.0 | 251.63 | 27.04 | 35.35 | 775.2 |
| 1700 | 0.007 167 | 40 329.0 | 54 466.0 | 253.79 | 27.29 | 35.60 | 798.2 |
| 1800 | 0.006 769 | 43 069.0 | 58 038.0 | 255.83 | 27.51 | 35.82 | 820.5 |
| 1900 | 0.006 413 | 45 830.0 | 61 630.0 | 257.77 | 27.71 | 36.03 | 842.3 |
| 2000 | 0.006 092 | 48 610.0 | 65 242.0 0.2 MPa | 259.62 | 27.90 | 36.21 | 863.5 |
| 50.70 | 22.072 | 4712.0 | | | 24.01 | FF 05 | 1021.0 |
| 59.78 | 33.072 | -4712.9 | -4706.8 | 70.908 | 34.01 | 55.05 | 1031.0 |
| 60 | 33.041 | -4701.0 | -4695.0 | 71.106 | 33.96 | 55.04 | 1029.3 |
| 62 | 32.756 | -4591.0 | -4584.9 | 72.911 | 33.52 | 55.04 | 1013.2 |
| 64 | 32.468 | -4481.0 | -4474.8 | 74.659 | 33.10 | 55.06 | 996.8 |
| 66 | 32.177 | -4370.9 | -4364.6 | 76.353 | 32.69 | 55.10 | 980.2 |
| 68 | 31.884 | -4260.7 | -4254.4 | 77.999 | 32.30 | 55.16 | 963.3 |
| 70 72 | 31.588 | -4150.3 | -4144.0 -4033.4 | 79.599 | 31.93 | 55.24 | 946.1 |
| 72 | 31.288 | -4039.8 | | 81.157 | 31.57 | 55.35 | 928.7 |
| 74 | 30.985 | -3929.1 | -3922.6 | 82.675 | 31.22 | 55.49 | 911.0 |
| 76 78 | 30.678 | -3818.0 | -3811.5 | 84.157 | 30.89 | 55.65 | 893.0 |
| 78 | 30.366 | -3706.5 | -3700.0 | 85.605 | 30.57 | 55.86 | 874.7 |
| 80 | 30.050 | -3594.7 | -3588.0 | 87.022 | 30.27 | 56.10 | 856.1 |
| 82 | 29.728 | -3482.3 | -3475.5 | 88.411 | 29.97 | 56.38 | 837.2 |
| 84 | 29.400 | -3369.2 | -3362.4 | 89.773 | 29.69 | 56.71 | 817.8 |
| 85.39 | 29.168 | -3290.4 | -3283.6 | 90.705 | 29.50 | 56.98 | 804.2 |
| 87.99 | 0.292 03 | 1718.8 | 2403.6 | 156.49 | 22.25 | 33.14 | 181.2 |
| 88 | 0.291 99 | 1719.0 | 2403.9 | 156.50 | 22.24 | 33.14 | 181.2 |
| 90 | 0.284 13 | 1765.8 | 2469.7 | 157.24 | 22.05 | 32.70 | 183.8 |
| 92 | 0.276 75 | 1812.1 | 2534.8 | 157.95 | 21.90 | 32.33 | 186.3 |
| 94 | 0.269 81 | 1857.9 | 2599.1 | 158.64 | 21.77 | 32.03 | 188.7 |
| 96 | 0.263 27 | 1903.2 | 2662.9 | 159.32 | 21.66 | 31.77 | 191.1 |
| 98 | 0.257 08 | 1948.2 | 2726.2 | 159.97 | 21.57 | 31.54 | 193.5 |
| 100 | 0.251 21 | 1993.0 | 2789.1 | 160.60 | 21.49 | 31.34 | 195.8 |
| 102 | 0.245 64 | 2037.4 | 2851.6 | 161.22 | 21.42 | 31.17 | 198.0 |
| 104 | 0.240 33 | 2081.6 | 2913.8 | 161.83 | 21.36 | 31.01 | 200.3 |
| 104 106 | 0.235 27 | 2125.6 | 2975.7 | 162.42 | 21.30 | 30.87 | 202.4 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|-------------|------------------------|--------------------|------------------|------------------|----------------|----------------|----------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 108 | 0.230 44 | 2169.4 | 3037.3 | 162.99 | 21.26 | 30.75 | 204.6 |
| 110 | 0.225 83 | 2213.0 | 3098.7 | 163.55 | 21.21 | 30.64 | 206.7 |
| 112 | 0.221 41 | 2256.5 | 3159.8 | 164.11 | 21.17 | 30.53 | 208.8 |
| 114 | 0.217 17 | 2299.9 | 3220.8 | 164.65 | 21.14 | 30.44 | 210.9 |
| 116 | 0.213 11 | 2343.1 | 3281.6 | 165.17 | 21.11 | 30.36 | 212.9 |
| 118 | 0.209 20 | 2386.2 | 3342.2 | 165.69 | 21.08 | 30.28 | 214.9 |
| 120 | 0.205 45 | 2429.2 | 3402.7 | 166.20 | 21.06 | 30.21 | 216.9 |
| 122 | 0.201 83 | 2472.2 | 3463.1 | 166.70 | 21.03 | 30.14 | 218.8 |
| 124 | 0.198 35 | 2515.0 | 3523.3 | 167.19 | 21.01 | 30.08 | 220.8 |
| 126 | 0.194 99 | 2557.8 | 3583.4 | 167.67 | 20.99 | 30.03 | 222.7 |
| 128 | 0.191 75 | 2600.4 | 3643.4 | 168.14 | 20.97 | 29.98 | 224.6 |
| 130 | 0.188 63 | 2643.0 | 3703.3 | 168.61 | 20.96 | 29.93 | 226.5 |
| 132 | 0.185 61 | 2685.6 | 3763.2 | 169.06 | 20.94 | 29.89 | 228.3 |
| 134 | 0.182 68 | 2728.1 | 3822.9 | 169.51 | 20.93 | 29.85 | 230.1 |
| 136 | 0.179 86 | 2770.5 | 3882.5 | 169.95 | 20.92 | 29.81 | 232.0 |
| 138 | 0.177 12 | 2812.9 | 3942.1 | 170.39 | 20.90 | 29.77 | 233.8 |
| 140 | 0.174 47 | 2855.3 | 4001.6 | 170.82 | 20.89 | 29.74 | 235.5 |
| 142 | 0.171 90 | 2897.6 | 4061.1 | 171.24 | 20.88 | 29.71 | 237.3 |
| 144 | 0.169 40 | 2939.8 | 4120.5 | 171.65 | 20.87 | 29.68 | 239.1 |
| 146 | 0.166 98 | 2982.1 | 4179.8 | 172.06 | 20.87 | 29.65 | 240.8 |
| 148 | 0.164 63 | 3024.3 | 4239.1 | 172.47 | 20.86 | 29.63 | 242.5 |
| 150 | 0.162 35 | 3066.4 | 4298.3 | 172.86 | 20.85 | 29.60 | 244.2 |
| 155 | 0.156 93 | 3171.7 | 4446.2 | 173.83 | 20.83 | 29.55 | 248.4 |
| 160 | 0.151 86 | 3276.8 | 4593.8 | 174.77 | 20.82 | 29.50 | 252.6 |
| 165 | 0.147 12 | 3381.8 | 4741.2 | 175.68 | 20.81 | 29.46 | 256.6 |
| 170 | 0.142 67 | 3486.6 | 4888.5 | 176.56 | 20.80 | 29.43 | 260.6 |
| 175 | 0.138 49 | 3591.4 | 5035.5 | 177.41 | 20.79 | 29.40 | 264.5 |
| 180 | 0.134 55 | 3696.0 | 5182.4 | 178.24 | 20.78 | 29.37 | 268.4 |
| 185 | 0.134 33 | 3800.6 | 5329.2 | 179.04 | 20.78 | 29.35 | 272.2 |
| 190 | 0.130 83 | 3905.1 | 5475.9 | 179.82 | 20.77 | 29.32 | 275.9 |
| 195 | 0.127 32 | 4009.5 | 5622.5 | 180.59 | 20.77 | 29.32 | 279.6 |
| 200 | 0.124 00 | 4113.9 | 5768.9 | 181.33 | 20.77 | 29.29 | 283.2 |
| 210 | 0.120 84 | 4322.5 | 6061.7 | 182.76 | 20.77 | 29.29 | 290.4 |
| 220 | 0.113 00 | 4522.5 | 6354.2 | 184.12 | 20.76 | 29.26 | 290.4 |
| 230 | 0.109 70 | 4739.3 | 6646.4 | 185.42 | 20.76 | 29.24 | 304.0 |
| 240 | 0.104 87 | 4947.6 | 6938.6 | 186.66 | 20.76 | 29.22 | 310.7 |
| 250 | 0.100 46 | 5155.9 | 7230.6 | 187.85 | 20.76 | 29.21 | 310.7 |
| | | | | | | | |
| 260 270 | 0.092 660 0.089 203 | 5364.1 5572.3 | 7522.5 7814.4 | 189.00 190.10 | 20.77 20.77 | 29.19 29.19 | 323.4 329.6 |
| 280 | | | 8106.3 | | | | |
| | 0.085 996 | 5780.6 | | 191.16 | 20.78 | 29.19 | 335.7 |
| 290 | 0.083 012 | 5988.9 | 8398.1 | 192.18 | 20.79 | 29.19 | 341.6 |
| 300 | 0.080 230 | 6197.2 | 8690.1 | 193.17 | 20.80 | 29.20 | 347.5 |
| 310 | 0.077 629 | 6405.7 | 8982.0 | 194.13 | 20.82 | 29.20 | 353.2 |
| 320 | 0.075 193 | 6614.3 | 9274.1 | 195.06 | 20.83 | 29.21 | 358.8 |
| 330 | 0.072 905 | 6823.0 | 9566.3 | 195.96 | 20.85 | 29.23 | 364.4 |
| 340 | 0.070 753 | 7032.0 | 9858.7 | 196.83 | 20.87 | 29.24 | 369.8 |
| 350 | 0.068 725 | 7241.1 | 10 151.0 | 197.68 | 20.90 | 29.26 | 375.2 |
| 360 | 0.066 810 | 7450.4 | 10 444.0 | 198.50 | 20.92 | 29.29 | 380.4 |
| 370 | 0.064 999 | 7660.0 | 10 737.0 | 199.31 | 20.95 | 29.31 | 385.6 |
| 380 | 0.063 285 | 7869.9 | 11 030.0 | 200.09 | 20.98 | 29.34 | 390.7 |
| 390 | 0.061 658 | 8080.0 | 11 324.0 | 200.85 | 21.01 | 29.37 | 395.7 |
| 400 | 0.060 114 | 8290.5 | 11 618.0 | 201.59 | 21.04 | 29.40 | 400.7 |
| 450 | 0.053 424 | 9348.5 | 13 092.0 | 205.07 | 21.25 | 29.59 | 424.4 |
| 500 | 0.048 077 | 10 418.0 | 14 578.0 | 208.20 | 21.51 | 29.84 | 446.6 |
| 550 | 0.043 704 | 11 501.0 | 16 077.0 | 211.06 | 21.80 | 30.14 | 467.5 |
| 600 | 0.040 061 | 12 600.0 | 17 592.0 | 213.69 | 22.13 | 30.46 | 487.3 |
| 650 | 0.036 980 | 13 715.0 | 19 123.0 | 216.14 | 22.47 | 30.80 | 506.1 |
| 700 | 0.034 338 | 14 847.0 | 20 672.0 | 218.44 | 22.82 | 31.14 | 524.1 |
| 750 | 0.032 049 | 15 997.0 | 22 238.0 | 220.60 | 23.17 | 31.49 | 541.4 |
| 800 | 0.030 047 | 17 164.0 | 23 821.0 | 222.64 | 23.51 | 31.83 | 558.1 |
| 900 | 0.026 709 | 19 548.0 | 27 036.0 | 226.43 | 24.15 | 32.47 | 589.8 |
| 1000 | 0.024 039 | 21 993.0 | 30 313.0 | 229.88 | 24.73 | 33.05 | 619.8 |
| | | 24 493.0 | 33 644.0 | 233.05 | 25.25 | 33.57 | 648.3 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|--------------|------------------------|--------------------|---------------------|------------------|----------------|----------------|-------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 1200 | 0.020 034 | 27 041.0 | 37 024.0 | 235.99 | 25.70 | 34.02 | 675.7 |
| 1300 | 0.018 494 | 29 632.0 | 40 447.0 | 238.73 | 26.10 | 34.42 | 701.9 |
| 1400 | 0.017 173 | 32 261.0 | 43 906.0 | 241.30 | 26.45 | 34.77 | 727.2 |
| 1500 | 0.016 029 | 34 922.0 | 47 399.0 | 243.71 | 26.76 | 35.08 | 751.7 |
| 1600 | 0.015 027 | 37 612.0 | 50 921.0 | 245.98 | 27.04 | 35.35 | 775.4 |
| 1700 | 0.014 144 | 40 329.0 | 54 469.0 | 248.13 | 27.29 | 35.60 | 798.3 |
| 1800 | 0.013 358 | 43 069.0 | 58 040.0 | 250.17 | 27.51 | 35.82 | 820.7 |
| 1900 | 0.012 656 | 45 830.0 | 61 633.0 | 252.12 | 27.71 | 36.03 | 842.5 |
| 2000 | 0.012 023 | 48 610.0 | 65 245.0 0.5 MPa | 253.97 | 27.90 | 36.21 | 863.7 |
| 59.84 | 33.080 | -4712.3 | -4697.1 | 70.919 | 34.01 | 55.02 | 1032.1 |
| 60 | 33.057 | -4703.4 | -4688.2 | 71.067 | 33.98 | 55.02 | 1030.8 |
| 62 | 32.772 | -4593.5 | -4578.2 | 72.871 | 33.54 | 55.01 | 1014.7 |
| 64 | 32.485 | -4483.6 | -4468.2 | 74.618 | 33.11 | 55.02 | 998.4 |
| 66 | 32.195 | -4373.6 | -4358.1 | 76.311 | 32.71 | 55.06 | 981.9 |
| 68 | 31.903 | -4263.6 | -4247.9 | 77.956 | 32.32 | 55.11 | 965.1 |
| 70 | 31.608 | -4153.5 | -4137.6 | 79.554 | 31.94 | 55.19 | 948.0 |
| 72 | 31.309 | -4043.1 | -4027.2 | 81.111 | 31.59 | 55.30 | 930.7 |
| 74 | 31.007 | -3932.6 | -3916.4 | 82.627 | 31.24 | 55.43 | 913.1 |
| 76 | 30.701 | -3821.7 | -3805.4 | 84.108 | 30.91 | 55.59 | 895.2 |
| 78 | 30.391 | -3710.5 | -3694.1 | 85.554 | 30.59 | 55.79 | 877.0 |
| 80 | 30.076 | -3598.9 | -3582.3 | 86.969 | 30.28 | 56.02 | 858.5 |
| 82 | 29.755 | -3486.8 | -3469.9 | 88.356 | 29.99 | 56.29 | 839.7 |
| 84 | 29.429 | -3374.0 | -3357.1 | 89.716 | 29.70 | 56.61 | 820.5 |
| 86 | 29.096 | -3260.7 | -3243.5 | 91.052 | 29.43 | 56.98 | 801.0 |
| 88 | 28.756 | -3146.5 | -3129.1 | 92.367 | 29.17 | 57.41 | 781.0 |
| 90 | 28.408 | -3031.4 | -3013.8 | 93.663 | 28.92 | 57.91 | 760.6 |
| 92 | 28.051 | -2915.2 | -2897.4 | 94.942 | 28.68 | 58.49 | 739.8 |
| 94 | 27.683 | -2797.8 | -2779.8 | 96.207 | 28.46 | 59.15 | 718.4 |
| 96 | 27.304 | -2679.0 | -2660.7 | 97.460 | 28.24 | 59.92 | 696.5 |
| 96.12 | 27.281 | -2671.8 | -2653.4 | 97.535 | 28.23 | 59.98 | 695.2 |
| 98.36 100 | 0.695 38 | 1830.0 | 2549.0 | 151.13 | 23.36 | 37.31 | 185.0 |
| 100 | 0.678 71 0.659 81 | 1872.9 1924.1 | 2609.6 2681.9 | 151.74 152.45 | 23.07 22.79 | 36.53 35.75 | 187.4 190.3 |
| 104 | 0.642 25 | 1974.2 | 2752.7 | 153.14 | 22.79 | 35.11 | 190.3 |
| 104 | 0.625 86 | 2023.5 | 2822.4 | 153.80 | 22.37 | 34.57 | 195.7 |
| 108 | 0.610 51 | 2072.0 | 2891.0 | 154.45 | 22.21 | 34.10 | 198.2 |
| 110 | 0.596 06 | 2120.0 | 2958.8 | 155.07 | 22.07 | 33.70 | 200.7 |
| 112 | 0.582 43 | 2167.4 | 3025.9 | 155.67 | 21.95 | 33.35 | 203.2 |
| 114 | 0.569 54 | 2214.4 | 3092.3 | 156.26 | 21.85 | 33.04 | 205.5 |
| 116 | 0.557 31 | 2260.9 | 3158.1 | 156.83 | 21.76 | 32.77 | 207.9 |
| 118 | 0.545 69 | 2307.1 | 3223.4 | 157.39 | 21.68 | 32.52 | 210.1 |
| 120 | 0.534 63 | 2352.9 | 3288.2 | 157.93 | 21.61 | 32.30 | 212.4 |
| 122 | 0.524 08 | 2398.5 | 3352.6 | 158.47 | 21.54 | 32.10 | 214.6 |
| 124 | 0.514 00 | 2443.8 | 3416.6 | 158.99 | 21.48 | 31.92 | 216.7 |
| 126 | 0.504 35 | 2488.9 | 3480.3 | 159.50 | 21.43 | 31.75 | 218.8 |
| 128 | 0.495 11 | 2533.8 | 3543.6 | 160.00 | 21.38 | 31.60 | 220.9 |
| 130 | 0.486 25 | 2578.4 | 3606.7 | 160.48 | 21.34 | 31.46 | 223.0 |
| 132 | 0.477 73 | 2622.9 | 3669.5 | 160.96 | 21.30 | 31.34 | 225.0 |
| 134 | 0.469 55 | 2667.2 | 3732.0 | 161.43 | 21.26 | 31.22 | 227.0 |
| 136 | 0.461 67 | 2711.3 | 3794.4 | 161.90 | 21.23 | 31.11 | 229.0 |
| 138 | 0.454 08 | 2755.3 | 3856.5 | 162.35 | 21.20 | 31.01 | 230.9 |
| 140 | 0.446 76 | 2799.2 | 3918.4 | 162.80 | 21.17 | 30.91 | 232.8 |
| 142 | 0.439 69 | 2843.0 | 3980.1 | 163.23 | 21.14 | 30.83 | 234.7 |
| 144 | 0.432 87 | 2886.6 | 4041.7 | 163.66 | 21.12 | 30.75 | 236.6 |
| 146 | 0.426 28 | 2930.2 | 4103.1 | 164.09 | 21.10 | 30.67 | 238.4 |
| 148 | 0.419 89 | 2973.6 | 4164.4 | 164.50 | 21.08 | 30.60 | 240.3 |
| 150 | 0.413 72 | 3017.0 | 4225.5 | 164.91 | 21.06 | 30.53 | 242.1 |
| 155 | 0.399 10 | 3125.0 | 4377.8 | 165.91 | 21.02 | 30.39 | 246.5 |
| 160 | 0.385 55 | 3232.6 | 4529.4 | 166.88 | 20.98 | 30.26 | 250.9 |
| 165 170 | 0.372 95 | 3339.8 | 4680.4 4830.9 | 167.80 168.70 | 20.95 | 30.15 30.05 | 255.1 259.3 |
| | 0.361 19 | 3446.6 | 4830.9 | 108./U | 20.93 | 30.05 | 239.3 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|---------------------|--------------------|-------------------|-----------|-----------|-----------|------------------|
| (K) | (mol/dm^3) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 180 | 0.339 87 | 3659.5 | 5130.6 | 170.42 | 20.89 | 29.90 | 267.4 |
| 185 | 0.330 17 | 3765.6 | 5280.0 | 171.24 | 20.87 | 29.83 | 271.3 |
| 190 | 0.321 04 | 3871.5 | 5429.0 | 172.03 | 20.86 | 29.78 | 275.1 |
| 195 | 0.312 41 | 3977.3 | 5577.7 | 172.80 | 20.85 | 29.72 | 278.9 |
| 200 | 0.304 25 | 4082.9 | 5726.2 | 173.55 | 20.84 | 29.68 | 282.6 |
| 210 | 0.289 19 | 4293.7 | 6022.6 | 175.00 | 20.82 | 29.60 | 289.9 |
| 220 | 0.275 59 | 4504.0 | 6318.3 | 176.38 | 20.81 | 29.54 | 297.0 |
| 230 | 0.263 25 | 4714.1 | 6613.4 | 177.69 | 20.80 | 29.49 | 303.9 |
| 240 | 0.251 99 | 4923.9 | 6908.1 | 178.94 | 20.80 | 29.45 | 310.6 |
| 250 | 0.241 67 | 5133.4 | 7202.4 | 180.14 | 20.80 | 29.43 | 310.0 |
| 260 | 0.232 18 | 5342.9 | 7496.4 | 181.30 | 20.80 | 29.39 | |
| | | | | | | | 323.5 |
| 270 | 0.223 42 | 5552.2 | 7790.1 | 182.41 | 20.80 | 29.37 | 329.8 |
| 280 | 0.215 31 | 5761.4 | 8083.7 | 183.47 | 20.81 | 29.35 | 335.9 |
| 290 | 0.207 77 | 5970.6 | 8377.1 | 184.50 | 20.81 | 29.34 | 341.9 |
| 300 | 0.200 75 | 6179.8 | 8670.5 | 185.50 | 20.82 | 29.33 | 347.8 |
| 310 | 0.194 20 | 6389.1 | 8963.8 | 186.46 | 20.84 | 29.33 | 353.6 |
| 320 | 0.188 06 | 6598.4 | 9257.1 | 187.39 | 20.85 | 29.33 | 359.3 |
| 330 | 0.182 31 | 6807.8 | 9550.4 | 188.29 | 20.87 | 29.34 | 364.8 |
| 340 | 0.176 89 | 7017.3 | 9843.9 | 189.17 | 20.89 | 29.35 | 370.3 |
| 350 | 0.171 80 | 7227.0 | 10 137.0 | 190.02 | 20.91 | 29.36 | 375.7 |
| 360 | 0.166 99 | 7436.9 | 10 431.0 | 190.85 | 20.93 | 29.37 | 381.0 |
| 370 | 0.162 45 | 7647.0 | 10 725.0 | 191.65 | 20.96 | 29.39 | 386.2 |
| 380 | 0.158 15 | 7857.3 | 11 019.0 | 192.44 | 20.99 | 29.41 | 391.3 |
| 390 | 0.154 07 | 8067.9 | 11 313.0 | 193.20 | 21.02 | 29.44 | 396.3 |
| 400 | 0.150 20 | | 11 608.0 | 193.20 | | 29.44 | 401.3 |
| | | 8278.8 | | | 21.06 | | |
| 450 | 0.133 45 | 9338.5 | 13 085.0 | 197.43 | 21.26 | 29.65 | 425.1 |
| 500 | 0.120 07 | 10 409.0 | 14 573.0 | 200.56 | 21.51 | 29.89 | 447.3 |
| 550 | 0.109 14 | 11 494.0 | 16 075.0 | 203.42 | 21.81 | 30.17 | 468.2 |
| 600 | 0.100 04 | 12 593.0 | 17 591.0 | 206.06 | 22.13 | 30.48 | 488.0 |
| 650 | 0.092 347 | 13 709.0 | 19 123.0 | 208.52 | 22.47 | 30.82 | 506.8 |
| 700 | 0.085 752 | 14 842.0 | 20 673.0 | 210.81 | 22.82 | 31.16 | 524.8 |
| 750 | 0.080 038 | 15 993.0 | 22 240.0 | 212.97 | 23.17 | 31.51 | 542.1 |
| 800 | 0.075 038 | 17 160.0 | 23 823.0 | 215.02 | 23.51 | 31.84 | 558.7 |
| 900 | 0.066 707 | 19 545.0 | 27 040.0 | 218.81 | 24.15 | 32.48 | 590.5 |
| 1000 | 0.060 042 | 21 990.0 | 30 318.0 | 222.26 | 24.74 | 33.06 | 620.4 |
| 1100 | 0.054 589 | 24 491.0 | 33 650.0 | 225.43 | 25.25 | 33.57 | 648.9 |
| 1200 | 0.050 045 | 27 039.0 | 37 030.0 | 228.37 | 25.71 | 34.03 | 676.2 |
| 1300 | 0.046 199 | 29 631.0 | 40 453.0 | 231.11 | 26.10 | 34.42 | 702.5 |
| 1400 | 0.040 199 | 32 259.0 | 43 914.0 | 233.68 | 26.46 | 34.77 | 727.8 |
| 1500 | 0.042 902 | 34 921.0 | 47 407.0 | 236.09 | 26.76 | 35.08 | 752.2 |
| | | | | | | | |
| 1600 | 0.037 544 | 37 611.0 | 50 929.0 | 238.36 | 27.04 | 35.36 | 775.9 |
| 1700 | 0.035 338 | 40 328.0 | 54 477.0 | 240.51 | 27.29 | 35.60 | 798.8 |
| 1800 | 0.033 377 | 43 068.0 | 58 049.0 | 242.55 | 27.51 | 35.83 | 821.2 |
| 1900 | 0.031 621 | 45 829.0 | 61 641.0 | 244.50 | 27.71 | 36.03 | 842.9 |
| 2000 | 0.030 042 | 48 610.0 | 65 254.0 | 246.35 | 27.90 | 36.21 | 864.1 |
| | | | 1.0 MPa is | obar | | | |
| 59.93 | 33.094 | -4711.2 | -4681.0 | 70.936 | 34.02 | 54.97 | 1033.8 |
| 60 | 33.083 | -4707.2 | -4677.0 | 71.002 | 34.00 | 54.97 | 1033.3 |
| 62 | 32.800 | -4597.6 | -4567.1 | 72.804 | 33.56 | 54.96 | 1017.3 |
| 64 | 32.514 | -4487.9 | -4457.2 | 74.549 | 33.14 | 54.97 | 1001.1 |
| 66 | 32.225 | -4378.2 | -4347.2 | 76.241 | 32.73 | 55.00 | 984.7 |
| 68 | 31.934 | -4268.5 | -4237.2 | 77.884 | 32.34 | 55.05 | 968.0 |
| 70 | 31.641 | -4158.6 | -4127.0 | 79.480 | 31.97 | 55.12 | 951.1 |
| 70 72 | 31.344 | -4138.6 -4048.6 | -4127.0 -4016.7 | 81.034 | 31.61 | 55.21 | 931.1 |
| | | | | | | | |
| 74 | 31.044 | -3938.4 | -3906.1 | 82.549 | 31.27 | 55.33 | 916.5 |
| 76 70 | 30.740 | -3827.9 | -3795.3 | 84.026 | 30.93 | 55.49 | 898.8 |
| 78 | 30.431 | -3717.0 | -3684.2 | 85.470 | 30.62 | 55.67 | 880.8 |
| 80 | 30.119 | -3605.8 | -3572.6 | 86.882 | 30.31 | 55.89 | 862.5 |
| 82 | 29.801 | -3494.2 | -3460.6 | 88.265 | 30.01 | 56.15 | 843.8 |
| 84 | 29.478 | -3381.9 | -3348.0 | 89.622 | 29.73 | 56.45 | 824.9 |
| 86 | 29.148 | -3269.1 | -3234.8 | 90.954 | 29.46 | 56.80 | 805.6 |
| 88 | 28.812 | -3155.5 | -3120.8 | 92.264 | 29.20 | 57.20 | 786.0 |
| | | -3041.1 | -3005.9 | 93.555 | 28.95 | 57.67 | 765.9 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sour |
|-------------|------------------------|--------------------|----------|-----------|-----------|-----------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 92 | 28.115 | -2925.6 | -2890.1 | 94.828 | 28.71 | 58.21 | 745.4 |
| 94 | 27.752 | -2809.1 | -2773.1 | 96.086 | 28.48 | 58.83 | 724.4 |
| 96 | 27.379 | -2691.2 | -2654.7 | 97.332 | 28.26 | 59.55 | 702.9 |
| 98 | 26.993 | -2571.8 | -2534.8 | 98.568 | 28.06 | 60.38 | 680.9 |
| 100 | 26.593 | -2450.7 | -2413.1 | 99.798 | 27.87 | 61.36 | 658.2 |
| 102 | 26.177 | -2327.5 | -2289.3 | 101.02 | 27.69 | 62.50 | 634.9 |
| 104 | 25.742 | -2201.8 | -2162.9 | 102.25 | 27.53 | 63.85 | 610.8 |
| 106 | 25.284 | -2073.2 | -2033.7 | 103.48 | 27.38 | 65.48 | 585.8 |
| 106.22 | 25.232 | -2059.0 | -2019.3 | 103.62 | 27.37 | 65.68 | 583.0 |
| 108.10 | 1.3836 | 1873.7 | 2596.5 | 146.73 | 24.74 | 44.60 | 185.2 |
| 110 | 1.3382 | 1931.9 | 2679.2 | 147.49 | 24.21 | 42.69 | 188.7 |
| 112 | 1.2951 | 1990.8 | 2763.0 | 148.24 | 23.78 | 41.11 | 192.1 |
| 114 | 1.2559 | 2047.6 | 2843.9 | 148.96 | 23.43 | 39.85 | 195.3 |
| 116 | 1.2200 | 2102.9 | 2922.5 | 149.64 | 23.15 | 38.81 | 198.3 |
| 118 | 1.1870 | 2156.7 | 2999.2 | 150.30 | 22.91 | 37.94 | 201.2 |
| 120 | 1.1563 | 2209.5 | 3074.3 | 150.93 | 22.71 | 37.20 | 204.0 |
| 122 | 1.1276 | 2261.3 | 3148.1 | 151.54 | 22.54 | 36.56 | 206.8 |
| 124 | 1.1009 | 2312.2 | 3220.6 | 152.13 | 22.39 | 36.00 | 209.4 |
| 126 | 1.0757 | 2362.5 | 3292.1 | 152.70 | 22.26 | 35.52 | 211.9 |
| 128 | 1.0519 | 2412.1 | 3362.7 | 153.25 | 22.15 | 35.08 | 214.4 |
| 130 | 1.0295 | 2461.1 | 3432.5 | 153.79 | 22.05 | 34.69 | 216.8 |
| 132 | 1.0082 | 2509.7 | 3501.5 | 154.32 | 21.95 | 34.34 | 219.2 |
| 134 | 0.988 04 | 2557.8 | 3569.9 | 154.84 | 21.87 | 34.03 | 221.5 |
| 136 | 0.968 81 | 2605.5 | 3637.7 | 155.34 | 21.80 | 33.74 | 223.8 |
| 138 | 0.950 48 | 2652.8 | 3704.9 | 155.83 | 21.73 | 33.48 | 226.0 |
| 140 | 0.932 97 | 2699.8 | 3771.6 | 156.31 | 21.66 | 33.24 | 228.2 |
| 142 | 0.916 22 | 2746.4 | 3837.9 | 156.78 | 21.61 | 33.02 | 230.3 |
| 144 | 0.900 18 | 2792.8 | 3903.7 | 157.24 | 21.55 | 32.82 | 232.4 |
| 146 | 0.884 78 | 2838.9 | 3969.2 | 157.69 | 21.51 | 32.64 | 234.5 |
| 148 | 0.869 99 | 2884.8 | 4034.3 | 158.13 | 21.46 | 32.46 | 236.5 |
| 150 | 0.855 77 | 2930.5 | 4099.0 | 158.57 | 21.42 | 32.30 | 238.5 |
| 155 | 0.822 47 | 3043.8 | 4259.6 | 159.62 | 21.33 | 31.95 | 243.4 |
| 160 | 0.792 01 | 3156.0 | 4418.6 | 160.63 | 21.26 | 31.66 | 248.1 |
| 165 | 0.763 99 | 3267.4 | 4576.3 | 161.60 | 21.20 | 31.41 | 252.7 |
| 170 | 0.738 11 | 3377.9 | 4732.8 | 162.53 | 21.14 | 31.19 | 257.1 |
| 175 | 0.714 11 | 3487.9 | 4888.2 | 163.44 | 21.10 | 31.00 | 261.5 |
| 180 | 0.691 76 | 3597.2 | 5042.8 | 164.31 | 21.06 | 30.84 | 265.7 |
| 185 | 0.670 90 | 3706.1 | 5196.6 | 165.15 | 21.03 | 30.70 | 269.8 |
| 190 | 0.651 36 | 3814.6 | 5349.8 | 165.97 | 21.00 | 30.57 | 273.9 |
| 195 | 0.633 01 | 3922.6 | 5502.4 | 166.76 | 20.98 | 30.46 | 277.9 |
| 200 | 0.615 74 | 4030.4 | 5654.4 | 167.53 | 20.95 | 30.36 | 281.7 |
| 210 | 0.584 06 | 4245.0 | 5957.2 | 169.01 | 20.92 | 30.19 | 289.3 |
| 220 | 0.555 66 | 4458.7 | 6258.4 | 170.41 | 20.90 | 30.06 | 296.6 |
| 230 | 0.530 03 | 4671.7 | 6558.4 | 171.74 | 20.88 | 29.95 | 303. |
| 240 | 0.506 76 | 4884.0 | 6857.4 | 173.01 | 20.86 | 29.85 | 310.6 |
| 250 | 0.485 53 | 5095.9 | 7155.5 | 174.23 | 20.85 | 29.78 | 317.3 |
| 260 | 0.466 07 | 5307.4 | 7453.0 | 175.40 | 20.85 | 29.72 | 323.8 |
| 270 | 0.448 16 | 5518.5 | 7749.9 | 176.52 | 20.85 | 29.66 | 330.2 |
| 280 | 0.431 62 | 5729.5 | 8046.3 | 177.60 | 20.85 | 29.62 | 336.4 |
| 290 | 0.416 30 | 5940.2 | 8342.3 | 178.64 | 20.85 | 29.59 | 342.5 |
| 300 | 0.402 05 | 6150.8 | 8638.1 | 179.64 | 20.86 | 29.56 | 348.4 |
| 310 | 0.388 76 | 6361.4 | 8933.6 | 180.61 | 20.87 | 29.54 | 354 |
| 320 | 0.376 35 | 6571.9 | 9229.0 | 181.54 | 20.88 | 29.53 | 360. |
| 330 | 0.364 72 | 6782.4 | 9524.2 | 182.45 | 20.90 | 29.52 | 365.0 |
| 340 | 0.353 80 | 6992.9 | 9819.4 | 183.33 | 20.91 | 29.52 | 371. |
| 350 | 0.343 53 | 7203.6 | 10 115.0 | 184.19 | 20.93 | 29.52 | 376.0 |
| 360 | 0.333 85 | 7414.4 | 10 410.0 | 185.02 | 20.96 | 29.52 | 381. |
| 370 | 0.324 71 | 7625.3 | 10 705.0 | 185.83 | 20.98 | 29.53 | 387. |
| 380 | 0.316 06 | 7836.4 | 11 000.0 | 186.62 | 21.01 | 29.54 | 392 |
| 390 | 0.307 87 | 8047.7 | 11 296.0 | 187.39 | 21.04 | 29.56 | 397.3 |
| 400 | 0.300 09 | 8259.3 | 11 592.0 | 188.13 | 21.07 | 29.58 | 402 |
| 450 | 0.266 51 | 9322.0 | 13 074.0 | 191.63 | 21.27 | 29.73 | 426.2 |
| 500 | 0.239 74 | 10 395.0 | 14 566.0 | 194.77 | 21.53 | 29.95 | 448.5 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|-------------|---------------------|----------------------|----------------------|------------------|----------------|----------------|----------------|
| (K) | (mol/dm^3) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 550 | 0.217 90 | 11 481.0 | 16 070.0 | 197.64 | 21.82 | 30.22 | 469.4 |
| 600 | 0.199 72 | 12 582.0 | 17 589.0 | 200.28 | 22.14 | 30.53 | 489.2 |
| 650 | 0.184 35 | 13 699.0 | 19 124.0 | 202.74 | 22.48 | 30.86 | 508.0 |
| 700 | 0.171 19 | 14 834.0 | 20 675.0 | 205.04 | 22.83 | 31.19 | 526.0 |
| 750 | 0.159 79 | 15 985.0 | 22 243.0 | 207.20 | 23.18 | 31.53 | 543.2 |
| 800 | 0.149 81 | 17 153.0 | 23 828.0 | 209.25 | 23.52 | 31.87 | 559.9 |
| 900 | 0.133 19 | 19 539.0 | 27 047.0 | 213.04 | 24.16 | 32.50 | 591.5 |
| 1000 | 0.119 90 | 21 986.0 | 30 326.0 | 216.49 | 24.74 | 33.07 | 621.5 |
| 1100 | 0.109 02 | 24 487.0 | 33 659.0 | 219.67 | 25.26 | 33.58 | 650.0 |
| 1200 | 0.099 953 | 27 036.0 | 37 041.0 | 222.61 | 25.71 | 34.03 | 677.2 |
| 1300 | 0.092 279 | 29 628.0 | 40 465.0 | 225.35 | 26.11 | 34.43 | 703.4 |
| 1400 | 0.085 701 | 32 257.0 | 43 925.0 | 227.91 | 26.46 | 34.78 | 728.7 |
| 1500 | 0.079 999 | 34 919.0 | 47 419.0 | 230.32 | 26.77 | 35.09 | 753.1 |
| 1600 | 0.075 008 | 37 610.0 | 50 942.0 | 232.60 | 27.04 | 35.36 | 776.7 |
| 1700 | 0.070 604 | 40 327.0 | 54 490.0 | 234.75 | 27.29 | 35.61 | 799.7 |
| 1800 | 0.066 689 | 43 067.0 | 58 062.0 | 236.79 | 27.51 | 35.83 | 822.0 |
| 1900 | 0.063 185 | 45 829.0 | 61 655.0 | 238.73 | 27.71 | 36.03 | 843.7 |
| 2000 | 0.060 031 | 48 609.0 | 65 268.0 | 240.59 | 27.90 | 36.22 | 864.9 |
| | | .= | 2.0 MPa is | | | | |
| 60.11 | 33.121 | -4709.1 | -4648.7 | 70.970 | 34.02 | 54.88 | 1037.4 |
| 62 | 32.854 | -4605.7 | -4544.8 | 72.673 | 33.61 | 54.86 | 1022.4 |
| 64 | 32.570 | -4496.5 | -4435.1 | 74.414 | 33.18 | 54.86 | 1006.5 |
| 66 | 32.285 | -4387.3 | -4325.3 | 76.103 | 32.78 | 54.88 | 990.3 |
| 68 | 31.997 | -4278.1 | -4215.6 | 77.741 | 32.39 | 54.91 | 973.8 |
| 70 | 31.706 | -4168.8 | -4105.7 | 79.334 | 32.02 | 54.97 | 957.2 |
| 72 | 31.412 | -4059.3 | -3995.7 | 80.884 | 31.66 | 55.05 | 940.3 |
| 74 | 31.116 | -3949.7 | -3885.5 | 82.393 | 31.32 | 55.15 | 923.1 |
| 76 | 30.815 | -3839.9 | -3775.0 | 83.866 | 30.99 | 55.29 | 905.7 |
| 78 | 30.511 | -3729.9 | -3664.3 | 85.304 | 30.67 | 55.45 | 888.1 |
| 80 | 30.203 | -3619.4 | -3553.2 | 86.710 | 30.36 | 55.64 | 870.2 |
| 82 | 29.891 | -3508.6 | -3441.7 | 88.087 | 30.07 | 55.87 | 852.0 |
| 84 | 29.573 | -3397.4 | -3329.7 | 89.436 | 29.78 | 56.13 | 833.5 |
| 86 | 29.249 | -3285.5 | -3217.2 | 90.760 | 29.51 | 56.44 | 814.7 |
| 88 | 28.919 | -3173.1 | -3103.9 | 92.062 | 29.25 | 56.80 | 795.6 |
| 90 | 28.583 | -3059.9 | -2989.9 | 93.343 | 29.00 | 57.21 | 776.1 |
| 92 | 28.238 | -2945.9 | -2875.0 | 94.606 95.852 | 28.76 | 57.69 | 756.2 |
| 94 | 27.885 | -2830.9 | -2759.1 | | 28.53 | 58.23 | 736.0 |
| 96 98 | 27.522 27.149 | -2714.7 -2597.3 | -2642.1 -2523.6 | 97.084 98.305 | 28.31 28.10 | 58.86 59.57 | 715.3 694.1 |
| 100 | 26.763 | - 2397.3 - 2478.4 | - 2323.6 - 2403.7 | 98.505 | 27.90 | 60.41 | 672.5 |
| | | | | | | | |
| 102 104 | 26.364 25.948 | -2357.8 | -2281.9 | 100.72 101.92 | 27.72 | 61.37 | 650.2 627.4 |
| 104 | 25.514 | -2235.2 -2110.2 | -2158.1 -2031.8 | 101.92 | 27.55 27.40 | 62.49 63.82 | 603.9 |
| 108 | | | | | | | 579.6 |
| | 25.058 | - 1982.5 | - 1902.7 | 104.33 | 27.26 | 65.40 | |
| 110 | 24.577 | -1851.4 | -1770.0 | 105.55 | 27.14 | 67.31 | 554.3 |
| 112 | 24.065 23.513 | - 1716.2 | -1633.1 | 106.78 | 27.04 26.97 | 69.66 | 528.0 500.4 |
| 114 | 22.913 | -1576.0 | -1491.0 | 108.04 | | 72.64 | |
| 116 | 22.245 | - 1429.3 | -1342.0 | 109.34 | 26.94 26.96 | 76.56 82.02 | 470.9 439.1 |
| 118 | 22.059 | -1273.6 | -1183.7 | 110.69 | | 83.81 | 439.1 |
| 118.52 | | - 1231.7 | -1141.0 | 111.05 | 26.97 | | |
| 119.94 | 2.9766 | 1788.0 | 2459.9 | 141.25 | 27.21 | 68.49 | 180.5 |
| 120 122 | 2.9709 | 1791.1 1880.8 | 2464.3 2592.2 | 141.29 | 27.17 | 68.18 | 180.7 186.1 |
| | 2.8114 | 1880.8 1961.3 | | 142.35 | 26.12 | 60.34 55.18 | 186.1 |
| 124 | 2.6809 | | 2707.4 | 143.28 | 25.36 24.78 | 55.18 51.49 | |
| 126 | 2.5702 | 2035.7 | 2813.8 | 144.13 | 24.78 | 51.49 | 195.1 |
| 128 | 2.4740 | 2105.5 | 2913.9 | 144.92 | 24.33 | 48.71 | 199.0 |
| 130 | 2.3888 | 2171.9 | 3009.1 | 145.66 | 23.96 | 46.54 | 202.6 |
| 132 | 2.3124 | 2235.5 | 3100.4 | 146.36 | 23.66 | 44.78 | 206.1 |
| 134 | 2.2432 | 2296.8 | 3188.4 | 147.02 | 23.40 | 43.33 | 209.3 |
| 136 | 2.1799 | 2356.4 | 3273.8 | 147.65 | 23.18 | 42.11 | 212.4 |
| 138 | 2.1217 | 2414.3 | 3357.0 | 148.26 | 22.99 | 41.07 | 215.4 |
| 140 | 2.0678 | 2470.9 | 3438.2 | 148.84 | 22.83 | 40.16 | 218.3 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature (K) | Density (mol/dm ³) | Internal energy (J/mol) | Enthalpy (J/mol) | Entropy J/(mol-K) | c_v J/(mol-K) | c_p J/(mol-K) | Speed of sound (m/s) |
|-----------------|--------------------------------|-------------------------------|------------------|----------------------|-----------------|-----------------|----------------------------|
| (K) | (IIIOI/dili [*]) | (J/IIIO1) | (J/IIIOI) | J/(IIIOI-K) | J/(IIIOI-K) | J/(IIIOI-K) | |
| 144 | 1.9707 | 2580.9 | 3595.7 | 149.95 | 22.55 | 38.68 | 223.7 |
| 146 | 1.9267 | 2634.4 | 3672.5 | 150.48 | 22.43 | 38.06 | 226.3 |
| 148 | 1.8853 | 2687.2 | 3748.0 | 151.00 | 22.32 | 37.51 | 228.8 |
| 150 | 1.8462 | 2739.2 | 3822.5 | 151.50 | 22.22 | 37.01 | 231.3 |
| 155 | 1.7572 | 2866.7 | 4004.9 | 152.69 | 22.02 | 35.96 | 237.2 |
| 160 | 1.6785 | 2991.0 | 4182.5 | 153.82 | 21.85 | 35.12 | 242.7 |
| 165 | 1.6082 | 3112.7 | 4356.3 | 154.89 | 21.71 | 34.44 | 248.1 |
| 170 | 1.5448 | 3232.4 | 4527.1 | 155.91 | 21.60 | 33.87 | 253.1 |
| 175 | 1.4872 | 3350.4 | 4695.2 | 156.88 | 21.50 | 33.39 | 258.0 |
| 180 | 1.4345 | 3466.9 | 4861.1 | 157.82 | 21.42 | 32.99 | 262.8 |
| 185 | 1.3861 | 3582.3 | 5025.2 | 158.72 | 21.35 | 32.64 | 267.3 |
| 190 | 1.3413 | 3696.5 | 5187.6 | 159.58 | 21.29 | 32.34 | 271.8 |
| 195 | 1.2997 | 3809.9 | 5348.6 | 160.42 | 21.24 | 32.08 | 276.1 |
| 200 | 1.2610 | 3922.4 | 5508.4 | 161.23 | 21.19 | 31.84 | 280.3 |
| 210 | 1.1909 | 4145.5 | 5824.9 | 162.77 | 21.12 | 31.46 | 288.4 |
| 220 | 1.1291 | 4366.5 | 6137.9 | 164.23 | 21.07 | 31.15 | 296.2 |
| 230 | 1.0739 | 4585.7 | 6448.1 | 165.61 | 21.02 | 30.90 | 303.6 |
| 240 | 1.0243 | 4803.5 | 6756.1 | 166.92 | 20.99 | 30.70 | 310.8 |
| 250 | 0.979 42 | 5020.2 | 7062.2 | 168.17 | 20.97 | 30.53 | 317.8 |
| 260 | 0.938 59 | 5236.0 | 7366.9 | 169.36 | 20.95 | 30.39 | 324.6 |
| 270 | 0.901 25 | 5451.0 | 7670.2 | 170.51 | 20.94 | 30.28 | 331.2 |
| 280 | 0.866 94 | 5665.4 | 7972.4 | 171.61 | 20.93 | 30.18 | 337.6 |
| 290 | 0.835 28 | 5879.3 | 8273.8 | 172.67 | 20.93 | 30.18 | 343.8 |
| 300 | 0.805 97 | 6092.8 | 8574.3 | 173.68 | 20.93 | 30.03 | 349.9 |
| 310 | 0.778 73 | 6306.0 | 8874.3 | 174.67 | 20.93 | 29.97 | 355.9 |
| 320 | 0.778 73 | 6519.0 | 9173.7 | 175.62 | 20.94 | 29.97 | 361.7 |
| 330 | 0.733 36 | 6731.7 | 9472.7 | 176.54 | 20.94 | 29.92 | 367.4 |
| | | | | | | | |
| 340 | 0.707 45 | 6944.4 | 9771.4 | 177.43 | 20.97 | 29.85 | 373.0 |
| 350 | 0.686 60 | 7156.9 | 10 070.0 | 178.30 | 20.98 | 29.83 | 378.5 |
| 360 | 0.666 99 | 7369.5 | 10 368.0 | 179.14 | 21.00 | 29.82 | 383.9 |
| 370 | 0.648 50 | 7582.1 | 10 666.0 | 179.95 | 21.03 | 29.81 | 389.2 |
| 380 | 0.631 04 | 7794.8 | 10 964.0 | 180.75 | 21.05 | 29.80 | 394.4 |
| 390 | 0.614 51 | 8007.6 | 11 262.0 | 181.52 | 21.08 | 29.80 | 399.5 |
| 400 | 0.598 86 | 8220.6 | 11 560.0 | 182.28 | 21.11 | 29.81 | 404.5 |
| 450 | 0.531 41 | 9289.2 | 13 053.0 | 185.79 | 21.30 | 29.91 | 428.5 |
| 500 | 0.477 83 | 10 367.0 | 14 552.0 | 188.95 | 21.55 | 30.09 | 450.8 |
| 550 | 0.434 20 | 11 457.0 | 16 063.0 | 191.83 | 21.84 | 30.33 | 471.8 |
| 600 | 0.397 94 | 12 561.0 | 17 586.0 | 194.48 | 22.16 | 30.62 | 491.5 |
| 650 | 0.367 32 | 13 680.0 | 19 125.0 | 196.94 | 22.50 | 30.93 | 510.3 |
| 700 | 0.34 112 | 14 816.0 | 20 680.0 | 199.25 | 22.84 | 31.26 | 528.3 |
| 750 | 0.31 842 | 15 970.0 | 22 251.0 | 201.42 | 23.19 | 31.58 | 545.5 |
| 800 | 0.298 57 | 17 139.0 | 23 838.0 | 203.47 | 23.53 | 31.91 | 562.1 |
| 900 | 0.265 50 | 19 528.0 | 27 061.0 | 207.26 | 24.17 | 32.53 | 593.7 |
| 1000 | 0.239 05 | 21 976.0 | 30 343.0 | 210.72 | 24.75 | 33.10 | 623.6 |
| 1100 | 0.217 40 | 24 479.0 | 33 679.0 | 213.90 | 25.26 | 33.61 | 652.0 |
| 1200 | 0.199 36 | 27 030.0 | 37 062.0 | 216.84 | 25.72 | 34.05 | 679.2 |
| 1300 | 0.184 08 | 29 623.0 | 40 487.0 | 219.58 | 26.11 | 34.44 | 705.3 |
| 1400 | 0.170 99 | 32 253.0 | 43 949.0 | 222.15 | 26.46 | 34.79 | 730.5 |
| 1500 | 0.159 63 | 34 915.0 | 47 444.0 | 224.56 | 26.77 | 35.10 | 754.8 |
| 1600 | 0.149 69 | 37 607.0 | 50 967.0 | 226.83 | 27.05 | 35.37 | 778.4 |
| 1700 | 0.140 92 | 40 324.0 | 54 517.0 | 228.98 | 27.29 | 35.61 | 801.3 |
| 1800 | 0.133 12 | 43 065.0 | 58 089.0 | 231.03 | 27.52 | 35.83 | 823.6 |
| 1900 | 0.126 14 | 45 827.0 | 61 683.0 | 232.97 | 27.72 | 36.04 | 845.3 |
| 2000 | 0.119 85 | 48 608.0 | 65 296.0 | 234.82 | 27.90 | 36.22 | 866.4 |
| | | | 5.0 MPa is | | | | |
| 60.64 | 33.200 | -4702.6 | -4552.0 | 71.073 | 34.04 | 54.61 | 1047.7 |
| 62 | 33.012 | -4629.1 | - 4477.6 | 72.287 | 33.75 | 54.58 | 1037.3 |
| 64 | 32.736 | -4521.2 | -4368.5 | 74.019 | 33.33 | 54.55 | 1021.9 |
| 66 | 32.458 | -4413.4 | -4259.4 | 75.698 | 32.93 | 54.54 | 1006.3 |
| 68 | 32.178 | -4305.7 | -4150.3 | 77.326 | 32.54 | 54.55 | 990.5 |
| 70 | 31.895 | -4198.0 | -4041.2 | 78.907 | 32.17 | 54.57 | 974.6 |
| , , | | | | | | | |
| 72 | 31.611 | -4090.2 | -3932.0 | 80.445 | 31.82 | 54.61 | 958.5 |

TABLE A2. Thermodynamic properties of air—Continued

| Геmperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|------------------|------------------|------------------|----------------|------------------|----------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 76 | 31.034 | -3874.5 | -3713.4 | 83.401 | 31.15 | 54.75 | 925.7 |
| 78 | 30.742 | -3766.4 | -3603.8 | 84.824 | 30.83 | 54.85 | 909.0 |
| 80 | 30.446 | -3658.2 | -3493.9 | 86.214 | 30.52 | 54.98 | 892.1 |
| 82 | 30.146 | -3549.7 | -3383.8 | 87.574 | 30.23 | 55.13 | 875.0 |
| 84 | 29.843 | -3441.0 | -3273.4 | 88.904 | 29.94 | 55.31 | 857.7 |
| 86 | 29.535 | -3331.9 | -3162.6 | 90.208 | 29.67 | 55.52 | 840.2 |
| 88 | 29.223 | -3222.4 | -3051.3 | 91.488 | 29.41 | 55.77 | 822.4 |
| 90 | 28.906 | -3112.4 | -2939.5 | 92.744 | 29.15 | 56.06 | 804.4 |
| 92 | 28.582 | -3002.0 | -2827.0 | 93.980 | 28.91 | 56.38 | 786.2 |
| 94 | 28.253 | -2890.9 | -2713.9 | 95.196 | 28.68 | 56.75 | 767.7 |
| 96 | 27.917 | -2779.1 | -2600.0 | 96.395 | 28.45 | 57.17 | 748.9 |
| 98 | 27.574 | -2666.5 | -2485.2 | 97.579 | 28.24 | 57.64 | 729.9 |
| 100 | 27.222 | -2553.1 | -2369.4 | 98.748 | 28.03 | 58.18 | 710.6 |
| 102 | 26.861 | -2438.6 | -2252.4 | 99.906 | 27.84 | 58.79 | 690.9 |
| 104 | 26.490 | -2322.9 | -2134.2 | 101.05 | 27.66 | 59.48 | 671.0 |
| 106 | 26.108 | -2206.0 | -2014.5 | 102.19 | 27.48 | 60.26 | 650.7 |
| 108 | 25.713 | -2087.5 | -1893.1 | 103.33 | 27.32 | 61.15 | 630.0 |
| 110 | 25.304 | -1967.4 | -1769.8 | 104.46 | 27.17 | 62.17 | 609.0 |
| 112 | 24.879 | -1845.3 | -1644.3 | 105.59 | 27.03 | 63.34 | 587.6 |
| 114 | 24.436 | -1720.9 | -1516.3 | 106.72 | 26.90 | 64.69 | 565.7 |
| 116 | 23.972 | -1594.0 | -1385.4 | 107.86 | 26.79 | 66.26 | 543.3 |
| 118 | 23.483 | -1464.0 | - 1251.1 | 109.01 | 26.70 | 68.12 | 520.4 |
| 120 | 22.966 | -1330.4 | -1112.7 | 110.17 | 26.62 | 70.34 | 496.8 |
| 122 | 22.415 | -1192.4 | -969.37 | 111.36 | 26.57 | 73.05 | 472.5 |
| 124 | 21.822 | -1049.1 | -820.02 | 112.57 | 26.55 | 76.44 | 447.2 |
| 126 | 21.176 | -899.10 | -662.99 | 113.83 | 26.57 | 80.80 | 420.8 |
| 128 | 20.462 | -740.19 | -495.84 | 115.14 | 26.63 | 86.66 | 393.0 |
| 130 | 19.654 | -569.09 | -314.69 | 116.55 | 26.78 | 95.03 | 363.1 |
| 132 | 18.706 | -380.05 | -112.75 | 118.09 | 27.04 | 107.94 | 330.4 |
| 134 | 17.533 | -162.02 | 123.16 | 119.86 | 27.53 | 130.20 | 294.1 |
| 136 | 15.954 | 108.31 | 421.72 | 122.07 | 28.44 | 173.22 | 253.7 |
| 138 | 13.686 | 470.13 | 835.47 | 125.09 | 29.89 | 241.20 | 216.2 |
| 140 | 11.096 | 892.71 | 1343.3 | 128.75 | 30.37 | 246.60 | 199.5 |
| 142 | 9.2688 | 1229.6 | 1769.0 | 131.77 | 29.21 | 178.44 131.40 | 199.2 |
| 144 | 8.1721 | 1462.3 | 2074.2 | 133.90 | 27.96 | | 202.8 |
| 146 | 7.4446 | 1636.9 | 2308.5 | 135.52 | 26.99 | 105.29 | 206.9 |
| 148 | 6.9126 | 1778.7 | 2502.1 | 136.83 | 26.24 | 89.47 | 211.0 |
| 150 155 | 6.4975 5.7485 | 1900.3 2152.4 | 2669.9 | 137.96 | 25.66 | 78.99 63.74 | 214.9 |
| | | | 3022.2 | 140.27 | 24.63 | | 223.8 |
| 160 | 5.2273 | 2362.0 | 3318.5 | 142.15 | 23.95 | 55.49 | 231.8 |
| 165 170 | 4.8317 | 2547.3 | 3582.1 3824.1 | 143.78 | 23.46 | 50.29 46.71 | 239.1 245.9 |
| 170 | 4.5153 4.2532 | 2716.8 2875.2 | 4050.8 | 145.22 146.54 | 23.08 22.78 | 44.09 | 252.2 |
| 180 | 4.0305 | 3025.5 | | 147.75 | 22.78 | | 252.2 |
| 185 | 3.8376 | | 4266.1 | 148.88 | | 42.09 | 263.9 |
| 190 | 3.6681 | 3169.5 3308.6 | 4472.4 4671.7 | 148.88 149.94 | 22.35 22.18 | 40.51 39.24 | 269.3 |
| 195 | 3.5173 | 3443.6 | 4865.2 | 150.95 | 22.18 | | 274.5 |
| 200 | | | | 151.90 | | 38.19 | 274.3 |
| 210 | 3.3819 | 3575.4 3831.0 | 5053.8 | | 21.92 21.72 | 37.31 | |
| 220 | 3.1473 2.9500 | 4078.5 | 5419.6 5773.4 | 153.69 155.34 | 21.72 | 35.92 34.88 | 288.8 297.7 |
| 230 | 2.7806 | 4319.9 | | 156.87 | | | 306.0 |
| 240 | 2.6330 | 4519.9 | 6118.1 6455.6 | 158.30 | 21.46 21.37 | 34.08 33.45 | 314.0 |
| 250 | 2.5028 | 4789.7 | 6787.4 | 159.66 | 21.37 | 32.94 | 321.5 |
| 260 | 2.3868 | 5019.7 | 7114.6 | 160.94 | 21.25 | 32.52 | 328.8 |
| 270 | 2.3808 | 5019.7 5247.4 | 7438.0 | 160.94 | 21.25 | 32.32 32.17 | 328.8 |
| 280 | 2.2825 | 5473.0 | 7438.0 7758.2 | 163.33 | 21.21 | 31.88 | 342.7 |
| 290 | 2.1880 | 5697.0 | 8075.7 | 163.33 164.44 | 21.17 | 31.63 | 342.7 |
| 300 | 2.0232 | 5919.6 | 8390.9 | 165.51 | 21.13 | 31.42 | 355.6 |
| 310 | 1.9507 | 6141.1 | 8704.2 | 166.54 | 21.13 | 31.24 | 361.8 |
| 320 | | | | 160.54 167.53 | | | |
| 320 | 1.8837 | 6361.6 6581.2 | 9015.9 9326.2 | 167.53 | 21.11 21.11 | 31.09 30.96 | 367.9 373.8 |
| 220 | 1.8216 | | | | | | |
| 340 | 1.7637 | 6800.3 | 9635.2 | 169.40 | 21.12 | 30.85 | 379.6 |

THERMODYNAMIC PROPERTIES OF AIR

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|----------------------|----------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 360 | 1.6591 | 7236.8 | 10 250.0 | 171.16 | 21.13 | 30.68 | 390.7 |
| 370 | 1.6117 | 7454.5 | 10 557.0 | 172.00 | 21.15 | 30.61 | 396.1 |
| 380 | 1.5671 | 7672.0 | 10 863.0 | 172.82 | 21.17 | 30.56 | 401.4 |
| 390 | 1.5250 | 7889.3 | 11 168.0 | 173.61 | 21.19 | 30.51 | 406.6 |
| 400 | 1.4852 | 8106.4 | 11 473.0 | 174.38 | 21.22 | 30.47 | 411.7 |
| 450 | 1.3152 | 9192.6 | 12 994.0 | 177.97 | 21.39 | 30.41 | 435.9 |
| 500 | 1.1814 | 10 284.0 | 14 516.0 | 181.17 | 21.62 | 30.48 | 458.3 |
| 550 | 1.0731 | 11 384.0 | 16 044.0 | 184.08 | 21.90 | 30.64 | 479.2 |
| 600 | 0.983 39 | 12 497.0 | 17 581.0 | 186.76 | 22.21 | 30.87 | 498.9 |
| 650 | 0.907 81 | 13 624.0 | 19 131.0 | 189.24 | 22.54 | 31.14 | 517.0 |
| 700 | 0.843 21 | 14 766.0 | 20 696.0 | 191.56 | 22.89 | 31.43 | 535.4 |
| 750 | 0.787 32 | 15 924.0 | 22 275.0 | 193.74 | 23.23 | 31.74 | 552.5 |
| 800 | 0.738 47 | 17 099.0 | 23 869.0 | 195.80 | 23.56 | 32.04 | 569.0 |
| 900 | 0.657 11 | 19 494.0 | 27 103.0 | 199.61 | 24.20 | 32.63 | 600.3 |
| 1000 | 0.592 03 | 21 949.0 | 30 394.0 | 203.07 | 24.77 | 33.18 | 629.9 |
| 1100 | 0.538 74 | 24 456.0 | 33 737.0 | 206.26 | 25.29 | 33.67 | 658. |
| 1200 | 0.494 31 | 27 011.0 | 37 126.0 | 209.21 | 25.74 | 34.10 | 685. |
| 1300 | 0.456 67 | 29 607.0 | 40 556.0 | 211.95 | 26.13 | 34.48 | 711.0 |
| 1400 | 0.424 38 | 32 239.0 | 44 021.0 | 214.52 | 26.48 | 34.82 | 736.0 |
| 1500 | 0.396 36 | 34 904.0 | 47 519.0 | 216.93 | 26.79 | 35.12 | 760. |
| 1600 | 0.371 82 0.350 15 | 37 598.0 | 51 045.0 | 219.21 | 27.06 | 35.39 | 783.0 |
| 1700 | | 40 317.0 | 54 596.0 | 221.36 | 27.30 | 35.63 | 806. |
| 1800 | 0.330 88 | 43 059.0 | 58 171.0 | 223.40 | 27.53 | 35.85 | 828.4 |
| 1900 | 0.313 61 | 45 823.0 | 61 766.0 65 380.0 | 225.35 | 27.73 | 36.05 | 850. |
| 2000 | 0.298 06 | 48 605.0 | 10.0 MPa is | 227.20 | 27.91 | 36.23 | 871. |
| (1.50 | 22.220 | 4601.2 | | | 24.00 | 54.10 | 1064 |
| 61.52 | 33.329 | -4691.3 | -4391.2 | 71.245 | 34.08 | 54.19 | 1064. |
| 62 64 | 33.264 32.998 | -4665.6 -4559.7 | -4364.9 -4256.7 | 71.671 73.390 | 33.98 33.57 | 54.18 54.12 | 1060. 1046. |
| | | | | | | | |
| 66 | 32.731 | -4454.0 | -4148.5 | 75.054 | 33.17 | 54.07 | 1031. |
| 68 70 | 32.462 32.192 | -4348.4 -4243.0 | -4040.4 -3932.4 | 76.668 78.233 | 32.79 32.43 | 54.03 54.00 | 1016. 1001. |
| 70 72 | 31.921 | | - 3932.4 - 3824.4 | 78.233 79.755 | 32.43 | 53.99 | 986. |
| 74 | 31.648 | -4137.6 -4032.4 | - 3824.4 - 3716.4 | 81.234 | 31.74 | 53.99 | 980. 971. |
| 76 76 | 31.373 | - 4032.4 - 3927.1 | -3608.4 | 82.674 | 31.41 | 54.01 | 956. |
| 78 78 | 31.096 | -3821.9 | -3500.3 | 84.077 | 31.09 | 54.04 | 940. |
| 80 | 30.817 | -3716.7 | -3392.2 | 85.446 | 30.79 | 54.09 | 925. |
| 82 | 30.536 | -3611.5 | -3284.0 | 86.782 | 30.50 | 54.15 | 909. |
| 84 | 30.253 | -3506.1 | -3175.6 | 88.088 | 30.21 | 54.24 | 893. |
| 86 | 29.966 | -3400.7 | -3067.0 | 89.366 | 29.94 | 54.34 | 877. |
| 88 | 29.677 | -3295.2 | -2958.2 | 90.616 | 29.68 | 54.47 | 861. |
| 90 | 29.384 | -3189.5 | -2849.1 | 91.842 | 29.42 | 54.61 | 845. |
| 92 | 29.088 | -3083.5 | -2739.7 | 93.044 | 29.18 | 54.78 | 829. |
| 94 | 28.788 | -2977.4 | -2630.0 | 94.224 | 28.94 | 54.97 | 813. |
| 96 | 28.484 | -2870.9 | -2519.8 | 95.384 | 28.71 | 55.19 | 796. |
| 98 | 28.176 | -2764.1 | -2409.2 | 96.524 | 28.49 | 55.44 | 780. |
| 100 | 27.863 | -2656.9 | -2298.1 | 97.647 | 28.28 | 55.72 | 763. |
| 102 | 27.546 | -2549.4 | -2186.3 | 98.753 | 28.08 | 56.02 | 746. |
| 104 | 27.222 | -2441.3 | -2073.9 | 99.844 | 27.89 | 56.36 | 729. |
| 106 | 26.893 | -2332.7 | - 1960.8 | 100.92 | 27.70 | 56.74 | 712. |
| 108 | 26.558 | -2223.5 | - 1846.9 | 101.99 | 27.52 | 57.16 | 695. |
| 110 | 26.215 | -2113.6 | -1732.2 | 103.04 | 27.35 | 57.61 | 678. |
| 112 | 25.866 | -2003.1 | -1616.5 | 104.08 | 27.19 | 58.12 | 661. |
| 114 | 25.509 | - 1891.7 | - 1499.7 | 105.11 | 27.03 | 58.67 | 643. |
| 116 | 25.143 | - 1779.5 | -1381.8 | 106.14 | 26.89 | 59.27 | 626. |
| 118 | 24.768 | -1666.3 | -1262.6 | 107.16 | 26.75 | 59.93 | 608. |
| 120 | 24.383 | -1552.1 | -1142.0 | 108.17 | 26.62 | 60.65 | 591. |
| 122 | 23.988 | -1436.8 | -1019.9 | 109.18 | 26.50 | 61.44 | 573. |
| 124 | 23.582 | -1320.2 | -896.19 | 110.19 | 26.39 | 62.30 | 556. |
| 126 | 23.164 | -1202.4 | -770.65 | 111.19 | 26.28 | 63.25 | 538. |
| 128 | 22.732 | -1083.1 | - 643.14 | 112.20 | 26.19 | 64.28 | 520. |
| 130 | 22.287 | -962.17 | -513.47 | 113.20 | 26.10 | 65.41 | 503. |
| | | | | | | | |

TABLE A2. Thermodynamic properties of air—Continued

| Γemperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|--------------------|---------------|-----------|-----------|-----------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 134 | 21.349 | -715.17 | -246.76 | 115.22 | 25.96 | 68.02 | 468.8 |
| 136 | 20.855 | -588.77 | -109.26 | 116.24 | 25.90 | 69.51 | 451.7 |
| 138 | 20.342 | -460.24 | 31.349 | 117.27 | 25.86 | 71.13 | 434.8 |
| 140 | 19.810 | - 329.46 | 175.34 | 118.30 | 25.82 | 72.88 | 418.1 |
| 142 | 19.257 | - 196.33 | 322.96 | 119.35 | 25.79 | 74.76 | 401.9 |
| 144 | 18.684 | -60.807 | 474.42 | 120.41 | 25.77 | 76.71 | 386.1 |
| 146 | 18.090 | 77.041 | 629.83 | 121.48 | 25.76 | 78.69 | 371.0 |
| 148 | 17.478 | 217.00 | 789.15 | 122.56 | 25.75 | 80.60 | 356.7 |
| 150 | 16.851 | 358.66 | 952.10 | 123.66 | 25.73 | 82.31 | 343.4 |
| 155 | 15.253 | 715.54 | 1371.2 | 126.41 | 25.65 | 84.77 | 316.0 |
| 160 | 13.709 | 1064.0 | 1793.4 | 129.09 | 25.45 | 83.46 | 297.6 |
| 165 | 12.331 | 1388.9 | 2199.9 | 131.59 | 25.11 | 78.69 | 287.3 |
| 170 | 11.170 | 1682.4 | 2577.7 | 133.84 | 24.70 | 72.33 | 282.7 |
| 175 | 10.218 | 1944.8 | 2923.4 | 135.85 | 24.30 | 66.09 | 281.8 |
| 180 | 9.4390 | 2180.7 | 3240.1 | 137.63 | 23.94 | 60.75 | 283.1 |
| 185 | 8.7946 | 2395.5 | 3532.6 | 139.24 | 23.62 | 56.38 | 285.6 |
| 190 | 8.2534 | 2593.7 | 3805.4 | 140.69 | 23.35 | 52.86 | 288.8 |
| 195 | 7.7920 | 2778.9 | 4062.3 | 142.03 | 23.11 | 50.01 | 292.4 |
| 200 | 7.3931 | 2953.7 | 4306.3 | 143.26 | 22.90 | 47.67 | 296.3 |
| 210 | 6.7353 | 3279.3 | 4764.1 | 145.50 | 22.56 | 44.10 | 304.3 |
| 220 | 6.2116 | 3581.7 | 5191.5 | 147.49 | 22.30 | 41.53 | 312.4 |
| 230 | 5.7819 | 3867.3 | 5596.8 | 149.29 | 22.10 | 39.62 | 320.3 |
| 240 | 5.4208 | 4140.6 | 5985.3 | 150.94 | 21.93 | 38.14 | 328.0 |
| 250 | 5.1116 | 4404.4 | 6360.7 | 152.47 | 21.80 | 36.98 | 335.4 |
| 260 | 4.8429 | 4660.8 | 6725.7 | 153.91 | 21.70 | 36.04 | 342.7 |
| 270 | 4.6064 | 4911.3 | 7082.2 | 155.25 | 21.61 | 35.28 | 349.7 |
| 280 | 4.3961 | 5157.0 | 7431.7 | 156.52 | 21.54 | 34.64 | 356.5 |
| 290 | 4.2075 | 5398.7 | 7775.4 | 157.73 | 21.49 | 34.11 | 363. |
| 300 | 4.0370 | 5637.1 | 8114.2 | 158.88 | 21.44 | 33.66 | 369.5 |
| 310 | 3.8819 | 5872.8 | 8448.9 | 159.97 | 21.41 | 33.28 | 375.7 |
| 320 | 3.7400 | 6106.2 | 8780.0 | 161.03 | 21.38 | 32.95 | 381.8 |
| 330 | 3.6096 | 6337.7 | 9108.1 | 162.04 | 21.36 | 32.67 | 387.7 |
| 340 | 3.4891 | 6567.4 | 9433.5 | 163.01 | 21.35 | 32.42 | 393.5 |
| 350 | 3.3774 | 6795.8 | 9756.7 | 163.94 | 21.34 | 32.21 | 399.1 |
| 360 | 3.2735 | 7023.0 | 10 078.0 | 164.85 | 21.34 | 32.03 | 404.6 |
| 370 | 3.1765 | 7249.1 | 10 397.0 | 165.72 | 21.34 | 31.87 | 410.0 |
| 380 | 3.0857 | 7474.4 | 10 715.0 | 166.57 | 21.35 | 31.73 | 415.3 |
| 390 | 3.0004 | 7699.0 | 11 032.0 | 167.39 | 21.37 | 31.61 | 420.5 |
| 400 | 2.9202 | 7923.0 | 11 347.0 | 168.19 | 21.38 | 31.50 | 425.6 |
| 450 | 2.5803 | 9037.7 | 12 913.0 | 171.88 | 21.52 | 31.18 | 449.0 |
| 500 | 2.3157 | 10 151.0 | 14 469.0 | 175.16 | 21.73 | 31.08 | 471.8 |
| 550 | 2.1029 | 11 268.0 | 16 023.0 | 178.12 | 22.00 | 31.12 | 492.5 |
| 600 | 1.9274 | 12 395.0 | 17 583.0 | 180.84 | 22.30 | 31.27 | 511.9 |
| 650 | 1.7800 | 13 533.0 | 19 151.0 | 183.35 | 22.62 | 31.47 | 530 |
| 700 | 1.6542 | 14 685.0 | 20 730.0 | 185.69 | 22.95 | 31.71 | 547.5 |
| 750 | 1.5454 | 15 852.0 | 22 322.0 | 187.88 | 23.29 | 31.97 | 564.0 |
| 800 | 1.4504 | 17 033.0 | 23 928.0 | 189.96 | 23.62 | 32.24 | 580.8 |
| 900 | 1.2922 | 19 441.0 | 27 179.0 | 193.79 | 24.24 | 32.79 | 611.0 |
| 1000 | 1.1655 | 21 904.0 | 30 484.0 | 197.27 | 24.81 | 33.30 | 640. |
| 1100 | 1.0618 | 24 419.0 | 33 837.0 | 200.46 | 25.32 | 33.76 | 668. |
| 1200 | 0.975 11 | 26 980.0 | 37 235.0 | 203.42 | 25.77 | 34.18 | 695.0 |
| 1300 | 0.901 65 | 29 581.0 | 40 672.0 | 206.17 | 26.16 | 34.55 | 720. |
| 1400 | 0.838 54 | 32 218.0 | 44 143.0 | 208.74 | 26.50 | 34.88 | 745. |
| 1500 | 0.783 74 | 34 886.0 | 47 646.0 | 211.16 | 26.81 | 35.17 | 769. |
| 1600 | 0.735 69 | 37 583.0 | 51 176.0 | 213.44 | 27.08 | 35.43 | 792. |
| 1700 | 0.693 21 | 40 305.0 | 54 731.0 | 215.59 | 27.32 | 35.66 | 814. |
| 1800 | 0.655 39 | 43 050.0 | 58 308.0 | 217.64 | 27.54 | 35.88 | 836. |
| 1900 | 0.621 49 | 45 815.0 | 61 906.0 | 219.58 | 27.74 | 36.07 | 857. |
| 2000 | 0.590 94 | 48 600.0 | 65 522.0 | 219.38 | 27.74 | 36.25 | 878. |
| 2000 | 0.370 74 | +o 000.0 | | | 41.73 | 30.23 | 0/0.0 |
| 63.24 | 22 574 | _ 1666 7 | 20.0 MPa isol | | 2/ 10 | 52 10 | 1004 |
| D 3 //I | 33.574 | -4666.7 | -4071.0 | 71.587 | 34.18 | 53.48 | 1094. |
| 64 | 33.480 | -4627.8 | -4030.5 | 72.224 | 34.03 | 53.44 | 1089.3 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|--------------------|------------------|------------------|----------------|----------------|----------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 68 | 32.981 | -4423.5 | -3817.1 | 75.457 | 33.27 | 53.24 | 1062.8 |
| 70 | 32.730 | -4321.8 | -3710.7 | 76.999 | 32.91 | 53.15 | 1049.4 |
| 72 | 32.480 | -4220.3 | -3604.5 | 78.496 | 32.57 | 53.07 | 1036.0 |
| 74 | 32.228 | -4119.0 | -3498.4 | 79.949 | 32.23 | 53.00 | 1022.6 |
| 76 | 31.976 | -4017.9 | -3392.5 | 81.361 | 31.91 | 52.94 | 1009.1 |
| 78 | 31.724 | -3917.1 | -3286.7 | 82.736 | 31.60 | 52.88 | 995.6 |
| 80 | 31.470 | -3816.5 | -3180.9 | 84.074 | 31.30 | 52.84 | 982.1 |
| 82 | 31.216 | -3716.0 | -3075.3 | 85.378 | 31.02 | 52.80 | 968.6 |
| 84 | 30.961 | -3615.7 | -2969.7 | 86.650 | 30.74 | 52.77 | 955.0 |
| 86 | 30.705 | -3515.6 | -2864.2 | 87.892 | 30.47 | 52.76 | 941.4 |
| 88 | 30.447 | -3415.6 | -2758.7 | 89.105 | 30.20 | 52.75 | 927.8 |
| 90 | 30.189 | -3315.7 | -2653.2 | 90.290 | 29.95 | 52.75 | 914.2 |
| 92 | 29.929 | -3215.9 | -2547.7 | 91.450 | 29.71 | 52.76 | 900.5 |
| 94 | 29.667 | -3116.3 | -2442.1 | 92.585 | 29.47 | 52.79 | 886.9 |
| 96 | 29.405 | -3016.7 | -2336.5 | 93.696 | 29.24 | 52.82 | 873.2 |
| 98 | 29.140 | -2917.2 | -2230.9 | 94.786 | 29.02 | 52.86 | 859.6 |
| 100 | 28.874 | -2817.7 | -2125.1 | 95.854 | 28.81 | 52.91 | 846.0 |
| 102 | 28.606 | -2718.3 | -2019.2 | 96.903 | 28.60 | 52.98 | 832.4 |
| 104 | 28.337 | -2619.0 | -1913.2 | 97.932 | 28.40 | 53.05 | 818.8 |
| 106 | 28.065 | -2519.6 | -1807.0 | 98.943 | 28.21 | 53.13 | 805.2 |
| 108 | 27.791 | -2420.3 | -1700.6 | 99.937 | 28.02 | 53.23 | 791.7 |
| 110 | 27.516 | -2320.9 | -1594.1 | 100.91 | 27.84 | 53.33 | 778.2 |
| 112 | 27.238 | -2221.6 | -1487.3 | 101.88 | 27.67 | 53.44 | 764.8 |
| 114 | 26.958 | -2122.2 | -1380.3 | 102.82 | 27.50 | 53.57 | 751.5 |
| 116 | 26.675 | -2022.8 | -1273.0 | 103.76 | 27.34 | 53.70 | 738.2 |
| 118 | 26.390 | -1923.4 | -1165.5 | 104.68 | 27.18 | 53.84 | 725.0 |
| 120 | 26.103 | -1823.9 | -1057.7 | 105.58 | 27.03 | 53.99 | 711.9 |
| 122 | 25.813 | -1724.4 | -949.55 | 106.48 | 26.88 | 54.14 | 699.0 |
| 124 | 25.520 | -1624.8 | -841.10 | 107.36 | 26.74 | 54.31 | 686.1 |
| 126 | 25.225 | -1525.2 | -732.32 | 108.23 | 26.61 | 54.48 | 673.4 |
| 128 | 24.928 | -1425.5 | -623.18 | 109.09 | 26.48 | 54.65 | 660.9 |
| 130 | 24.628 | -1325.8 | -513.69 | 109.94 | 26.35 | 54.84 | 648.5 |
| 132 | 24.325 | -1226.0 | -403.84 | 110.77 | 26.23 | 55.02 | 636.3 |
| 134 | 24.020 | -1126.3 | -293.61 | 111.60 | 26.11 | 55.21 | 624.3 |
| 136 | 23.712 | -1026.4 | -183.00 | 112.42 | 26.00 | 55.40 | 612.5 |
| 138 | 23.402 | -926.63 | -72.016 | 113.23 | 25.89 | 55.59 | 601.0 |
| 140 | 23.090 | -826.82 | 39.349 | 114.03 | 25.79 | 55.78 | 589.6 |
| 142 | 22.776 | -727.03 | 151.09 | 114.83 | 25.69 | 55.96 | 578.5 |
| 144 | 22.460 | -627.28 | 263.20 | 115.61 | 25.59 | 56.15 | 567.7 |
| 146 | 22.142 | -527.60 | 375.68 | 116.39 | 25.50 | 56.33 | 557.2 |
| 148 | 21.822 | -428.01 | 488.50 | 117.15 | 25.41 | 56.50 | 546.9 |
| 150 | 21.501 | -328.54 | 601.66 | 117.91 | 25.32 | 56.66 | 536.9 |
| 155 | 20.693 | -80.620 | 885.87 | 119.78 | 25.12 | 57.01 | 513.3 |
| 160 | 19.884 | 165.72 | 1171.6 | 121.59 | 24.92 | 57.25 | 491.8 |
| 165 | 19.077 | 409.75 | 1458.1 | 123.35 | 24.74 | 57.34 | 472.4 |
| 170 | 18.280 | 650.59 | 1744.7 | 125.07 | 24.56 | 57.24 | 455.3 |
| 175 | 17.500 | 887.32 | 2030.2 | 126.72 | 24.39 | 56.91 | 440.5 |
| 180 | 16.745 | 1119.0 | 2313.4 | 128.32 | 24.22 | 56.35 | 428.0 |
| 185 | 16.021 | 1344.9 | 2593.3 | 129.85 | 24.06 | 55.58 | 417.6 |
| 190 | 15.332 | 1564.4 | 2868.9 | 131.32 | 23.89 | 54.62 | 409.1 |
| 195 | 14.683 | 1777.1 | 3139.3 | 132.72 | 23.74 | 53.52 | 402.4 |
| 200 | 14.074 | 1982.9 | 3403.9 | 134.06 | 23.58 | 52.34 | 397.2 |
| 210 | 12.978 | 2374.1 | 3915.2 | 136.56 | 23.30 | 49.90 | 397.2 |
| 220 | 12.034 | 2740.3 | 4402.3 | 138.83 | 23.04 | 47.56 | 387.6 |
| 230 | 11.221 | 3084.8 | 4867.1 | 140.89 | 22.81 | 45.45 | 387.3 |
| 240 | 10.519 | 3410.9 | | 140.89 | 22.62 | | 388.7 |
| 250 | 9.9093 | | 5312.2 5740.2 | | | 43.61 | 388.7 391.3 |
| | | 3721.9 | 5740.2 | 144.53 | 22.45 | 42.03 | |
| 260 | 9.3749 | 4020.3 | 6153.6 | 146.16 | 22.31 | 40.69 | 394.7 |
| 270 | 8.9032 | 4308.2 | 6554.6 | 147.67 | 22.19 | 39.54 | 398.6 |
| 280 | 8.4837 | 4587.4 | 6944.9 | 149.09 | 22.09 | 38.55 | 402.8 |
| 290 | 8.1078 | 4859.3 | 7326.1 | 150.43 | 22.00 | 37.71 | 407.3 |
| 300 | 7.7690 7.4616 | 5125.0 5385.5 | 7699.4 8065.9 | 151.69 152.89 | 21.93 21.86 | 36.97 36.34 | 412.0 416.7 |
| 310 | | | | | | | |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|----------------------|------------------------|----------------------|----------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 320 | 7.1814 | 5641.5 | 8426.5 | 154.04 | 21.81 | 35.79 | 421.5 |
| 330 | 6.9246 | 5893.6 | 8781.8 | 155.13 | 21.77 | 35.30 | 426.3 |
| 340 | 6.6882 | 6142.3 | 9132.7 | 156.18 | 21.74 | 34.87 | 431.1 |
| 350 | 6.4698 | 6388.2 | 9479.5 | 157.19 | 21.71 | 34.50 | 435.9 |
| 360 | 6.2673 | 6631.6 | 9822.8 | 158.15 | 21.69 | 34.17 | 440.6 |
| 370 | 6.0788 | 6872.8 | 10 163.0 | 159.08 | 21.68 | 33.87 | 445.3 |
| 380 | 5.9028 | 7112.2 | 10 500.0 | 159.98 | 21.67 | 33.61 | 450.0 |
| 390 | 5.7380 | 7349.9 | 10 835.0 | 160.85 | 21.67 | 33.38 | 454.6 |
| 400 | 5.5834 | 7586.1 | 11 168.0 | 161.70 | 21.68 | 33.18 | 459.2 |
| 450 | 4.9320 | 8752.1 | 12 807.0 | 165.56 | 21.76 | 32.46 | 481. |
| 500 | 4.4288 | 9904.1 | 14 420.0 | 168.96 | 21.94 | 32.09 | 501.3 |
| 550 | 4.0260 | 11 053.0 | 16 020.0 | 172.01 | 22.17 | 31.95 | 521. |
| 600 | 3.6950 | 12 204.0 | 17 617.0 | 174.79 | 22.45 | 31.95 | 539.5 |
| 650 | 3.4173 | 13 364.0 | 19 216.0 | 177.35 | 22.76 | 32.04 | 557.0 |
| 700 | 3.1804 | 14 533.0 | 20 822.0 | 179.73 | 23.08 | 32.20 | 573. |
| 750 | 2.9756 | 15 715.0 | 22 436.0 | 181.96 | 23.40 | 32.39 | 589.8 |
| 800 | 2.7966 | 16 910.0 | 24 061.0 | 184.05 | 23.72 | 32.61 | 605. |
| 900 | 2.4982 | 19 339.0 | 27 345.0 | 187.92 | 24.33 | 33.07 | 634. |
| 1000 | 2.2588 | 21 820.0 | 30 674.0 | 191.43 | 24.89 | 33.52 | 662. |
| 1100 | 2.0621 | 24 349.0 | 34 047.0 | 194.64 | 25.39 | 33.94 | 689. |
| 1200 | 1.8975 | 26 921.0 | 37 461.0 | 197.61 | 25.83 | 34.32 | 715.2 |
| 1300 | 1.7576 | 29 532.0 | 40 911.0 | 200.37 | 26.21 | 34.67 | 740.0 |
| 1400 | 1.6372 | 32 177.0 | 44 393.0 | 202.95 | 26.55 | 34.97 | 764.0 |
| 1500 | 1.5323 | 34 852.0 | 47 904.0 | 205.38 | 26.85 | 35.25 | 787. |
| 1600 | 1.4402 | 37 555.0 | 51 442.0 | 207.66 | 27.12 | 35.50 | 809. |
| 1700 | 1.3586 | 40 282.0 | 55 003.0 | 209.82 | 27.36 | 35.72 | 831. |
| 1800 | 1.2858 | 43 032.0 | 58 586.0 | 211.87 | 27.58 | 35.93 | 853.0 |
| 1900 | 1.2205 | 45 802.0 | 62 189.0 | 213.81 | 27.78 | 36.12 | 873.5 |
| 2000 | 1.1615 | 48 590.0 | 65 809.0 | 215.67 | 27.96 | 36.29 | 894.2 |
| 60.21 | 24.220 | 4501.0 | 50.0 MPa is | | 24.51 | 51.01 | 1170.5 |
| 68.21 70 | 34.238 34.049 | -4581.8 -4497.3 | -3121.4 -3028.8 | 72.575 73.915 | 34.51 34.20 | 51.91 51.77 | 1172.1 1163.3 |
| 70 | 33.837 | | - 3028.8 - 2925.4 | 75.371 | 33.87 | 51.60 | 1152.9 |
| 72 74 | 33.626 | -4403.1 -4309.3 | -2923.4 -2822.4 | 76.783 | 33.55 | 51.45 | 1132. |
| 7 4 76 | 33.416 | -4309.3 -4216.0 | -2822.4 -2719.7 | 78.153 | 33.24 | 51.45 | 1132. |
| 78 | 33.206 | -4123.0 | -2617.2 | 79.483 | 32.94 | 51.14 | 1132. |
| 80 | 32.997 | -4030.4 | -2517.2 -2515.1 | 80.776 | 32.65 | 50.99 | 1111. |
| 82 | 32.789 | -3938.2 | -2413.3 | 82.033 | 32.37 | 50.84 | 1101. |
| 84 | 32.581 | -3846.4 | -2311.7 | 83.257 | 32.10 | 50.70 | 1091. |
| 86 | 32.374 | - 3754.9 | -2210.5 | 84.448 | 31.84 | 50.56 | 1081. |
| 88 | 32.167 | -3663.9 | -2109.5 | 85.609 | 31.59 | 50.42 | 1071. |
| 90 | 31.961 | -3573.2 | -2008.8 | 86.740 | 31.34 | 50.29 | 1061. |
| 92 | 31.756 | -3482.9 | - 1908.3 | 87.844 | 31.10 | 50.16 | 1051. |
| 94 | 31.551 | -3392.9 | -1808.1 | 88.922 | 30.87 | 50.10 | 1041. |
| 96 | 31.346 | -3303.3 | -1708.2 | 89.974 | 30.65 | 49.91 | 1031. |
| 98 | 31.142 | -3214.0 | -1608.5 | 91.002 | 30.43 | 49.79 | 1022. |
| 100 | 30.939 | -3125.1 | - 1509.0 | 92.007 | 30.22 | 49.68 | 1012. |
| 102 | 30.736 | -3036.5 | - 1409.8 | 92.989 | 30.01 | 49.56 | 1002. |
| 104 | 30.534 | -2948.3 | -1310.8 | 93.950 | 29.81 | 49.45 | 993. |
| 106 | 30.332 | -2860.4 | -1212.0 | 94.891 | 29.62 | 49.34 | 984. |
| 108 | 30.130 | -2772.9 | -1113.4 | 95.813 | 29.43 | 49.23 | 974. |
| 110 | 29.929 | -2685.6 | - 1015.0 | 96.715 | 29.25 | 49.13 | 965. |
| 112 | 29.729 | -2598.8 | -916.88 | 97.599 | 29.23 | 49.03 | 956. |
| 114 | 29.529 | -2512.2 | -818.93 | 98.466 | 28.90 | 48.93 | 947. |
| 116 | 29.329 | -2426.0 | -721.17 | 99.316 | 28.73 | 48.83 | 938. |
| 118 | 29.130 | -2340.0 | -623.60 | 100.15 | 28.57 | 48.74 | 929. |
| 120 | 28.932 | -2254.4 | -526.23 | 100.13 | 28.41 | 48.64 | 929. |
| 122 | 28.733 | -2169.2 | -429.04 | 101.77 | 28.26 | 48.55 | 920. 912. |
| 124 | 28.536 | -2109.2 -2084.2 | -332.03 | 102.56 | 28.11 | 48.46 | 903. |
| 126 | 28.339 | - 1999.6 | -235.21 | 103.34 | 27.96 | 48.37 | 895. |
| 128 | 28.142 | - 1999.0 - 1915.3 | - 233.21 - 138.56 | 103.34 | 27.82 | 48.28 | 887. |
| | 27.946 | - 1913.3 - 1831.3 | | | 27.68 | | 878. |
| 130 | /. / 94b | | -42.096 | 104.84 | / / hx | 48.19 | ×/× |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|--------------|------------------------|----------------------|----------------------|------------------|----------------|----------------|-------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 134 | 27.555 | -1664.2 | 150.32 | 106.30 | 27.42 | 48.02 | 862.7 |
| 136 | 27.361 | -1581.2 | 246.26 | 107.01 | 27.29 | 47.93 | 854.8 |
| 138 | 27.167 | - 1498.4 | 342.03 | 107.71 | 27.17 | 47.84 | 847.1 |
| 140 | 26.974 | -1416.0 | 437.64 | 108.40 | 27.05 | 47.76 | 839.4 |
| 142 | 26.781 | -1333.9 | 533.07 | 109.08 | 26.94 | 47.67 | 831.9 |
| 144 146 | 26.589 26.398 | - 1252.1 - 1170.7 | 628.33 723.42 | 109.74 110.40 | 26.82 26.71 | 47.59 47.50 | 824.6 817.3 |
| 148 | 26.207 | -1170.7 -1089.6 | 818.33 | 111.04 | 26.60 | 47.42 | 810.2 |
| 150 | 26.017 | - 1008.7 | 913.08 | 111.68 | 26.50 | 47.33 | 803.2 |
| 155 | 25.545 | -808.12 | 1149.2 | 113.23 | 26.25 | 47.11 | 786.3 |
| 160 | 25.079 | -609.54 | 1384.2 | 114.72 | 26.01 | 46.89 | 770.2 |
| 165 | 24.618 | -413.01 | 1618.0 | 116.16 | 25.79 | 46.66 | 754.9 |
| 170 | 24.163 | -218.56 | 1850.8 | 117.55 | 25.59 | 46.42 | 740.5 |
| 175 | 23.714 | -26.212 | 2082.3 | 118.89 | 25.39 | 46.18 | 726.9 |
| 180 | 23.272 | 164.03 | 2312.6 | 120.19 | 25.21 | 45.93 | 714.1 |
| 185 | 22.837 | 352.15 | 2541.6 | 121.44 | 25.04 | 45.68 | 702.1 |
| 190 | 22.410 | 538.13 | 2769.3 | 122.66 | 24.88 | 45.41 | 690.9 |
| 195 | 21.990 | 721.98 | 2995.7 | 123.83 | 24.72 | 45.14 | 680.4 |
| 200 | 21.579 | 903.69 | 3220.7 | 124.97 | 24.58 | 44.87 | 670.6 |
| 210 | 20.783 | 1260.7 | 3666.6 | 127.15 | 24.31 | 44.29 | 653.1 |
| 220 | 20.022 | 1609.4 | 4106.6 4540.6 | 129.20 | 24.07 | 43.70 | 638.1 |
| 230 240 | 19.299 18.613 | 1949.7 2282.1 | 4540.6 4968.5 | 131.13 132.95 | 23.86 23.66 | 43.10 42.49 | 625.4 614.7 |
| 250 | 17.963 | 2606.8 | 5390.3 | 134.67 | 23.49 | 41.88 | 605.8 |
| 260 | 17.350 | 2924.2 | 5806.0 | 136.30 | 23.33 | 41.27 | 598.5 |
| 270 | 16.773 | 3234.7 | 6215.8 | 137.85 | 23.19 | 40.68 | 592.7 |
| 280 | 16.228 | 3538.8 | 6619.8 | 139.32 | 23.06 | 40.12 | 588.0 |
| 290 | 15.716 | 3836.8 | 7018.2 | 140.71 | 22.95 | 39.57 | 584.4 |
| 300 | 15.234 | 4129.2 | 7411.3 | 142.05 | 22.85 | 39.06 | 581.7 |
| 310 | 14.780 | 4416.5 | 7799.4 | 143.32 | 22.76 | 38.57 | 579.7 |
| 320 | 14.353 | 4699.1 | 8182.8 | 144.54 | 22.68 | 38.11 | 578.4 |
| 330 | 13.950 | 4977.3 | 8561.7 | 145.70 | 22.61 | 37.68 | 577.7 |
| 340 | 13.569 | 5251.6 | 8936.4 | 146.82 | 22.55 | 37.28 | 577.4 |
| 350 | 13.210 | 5522.3 | 9307.3 | 147.90 | 22.50 | 36.91 | 577.6 |
| 360 | 12.870 | 5789.7 | 9674.6 | 148.93 | 22.45 | 36.56 | 578.1 |
| 370 380 | 12.549 12.244 | 6054.1 | 10 039.0 10 399.0 | 149.93 150.89 | 22.42 22.39 | 36.24 35.94 | 578.9 580.0 |
| 390 | 11.955 | 6315.8 6575.1 | 10 399.0 | 150.89 | 22.36 | 35.67 | 581.4 |
| 400 | 11.680 | 6832.1 | 11 113.0 | 152.72 | 22.35 | 35.42 | 582.9 |
| 450 | 10.489 | 8090.6 | 12 858.0 | 156.83 | 22.34 | 34.44 | 592.4 |
| 500 | 9.5353 | 9319.0 | 14 563.0 | 160.42 | 22.44 | 33.81 | 604.1 |
| 550 | 8.7536 | 10 531.0 | 16 243.0 | 163.63 | 22.62 | 33.44 | 616.6 |
| 600 | 8.0999 | 11 737.0 | 17 910.0 | 166.53 | 22.85 | 33.25 | 629.5 |
| 650 | 7.5440 | 12 943.0 | 19 571.0 | 169.19 | 23.11 | 33.19 | 642.5 |
| 700 | 7.0646 | 14 153.0 | 21 230.0 | 171.65 | 23.40 | 33.21 | 655.5 |
| 750 | 6.6465 | 15 370.0 | 22 892.0 | 173.94 | 23.69 | 33.29 | 668.3 |
| 800 | 6.2780 | 16 595.0 | 24 559.0 | 176.09 | 23.99 | 33.40 | 681.0 |
| 900 | 5.6573 | 19 076.0 | 27 915.0 | 180.04 | 24.57 | 33.71 | 705.9 |
| 1000 | 5.1533 | 21 599.0 | 31 302.0 | 183.61 | 25.09 | 34.04 | 730.0 |
| 1100 | 4.7351 | 24 163.0 | 34 722.0 | 186.87 | 25.57 | 34.37 | 753.6 |
| 1200 | 4.3818 | 26 764.0 | 38 175.0 | 189.88 | 25.99 | 34.68 | 776.5 |
| 1300 1400 | 4.0790 3.8164 | 29 400.0 | 41 658.0 | 192.66 195.27 | 26.36 26.69 | 34.97 35.23 | 798.8 820.6 |
| 1500 | 3.5862 | 32 066.0 34 760.0 | 45 168.0 48 703.0 | 195.27 197.70 | 26.98 | 35.23 35.47 | 820.6 841.8 |
| 1600 | 3.3828 | 37 479.0 | 52 260.0 | 200.00 | 27.23 | 35.68 | 862.5 |
| 1700 | 3.2015 | 40 221.0 | 55 839.0 | 202.17 | 27.47 | 35.88 | 882.8 |
| 1800 | 3.0389 | 42 983.0 | 59 436.0 | 204.23 | 27.68 | 36.07 | 902.6 |
| 1900 | 2.8923 | 45 764.0 | 63 051.0 | 206.18 | 27.87 | 36.23 | 922.0 |
| 2000 | 2.7593 | 48 562.0 | 66 683.0 | 208.04 | 28.04 | 36.39 | 941.1 |
| | | | 100.0 MPa iso | | | | |
| 75.92 | 35.183 | -4416.8 | -1574.5 | 74.060 | 34.95 | 50.34 | 1281.8 |
| 76 | 35.176 | -4413.2 | -1570.4 | 74.114 | 34.94 | 50.34 | 1281.5 |
| 78 | 35.002 | -4326.8 | -1469.9 | 75.419 | 34.65 | 50.15 | 1273.7 |

TABLE A2. Thermodynamic properties of air—Continued

| Геmperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of soun |
|-------------|------------------------|------------------|------------------|-----------|-----------|----------------|---------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 80 | 34.829 | -4240.9 | -1369.8 | 76.686 | 34.37 | 49.96 | 1265.9 |
| 82 | 34.658 | -4155.4 | -1270.0 | 77.918 | 34.10 | 49.78 | 1258.3 |
| 84 | 34.487 | -4070.3 | -1170.7 | 79.115 | 33.83 | 49.60 | 1250.6 |
| 86 | 34.318 | -3985.6 | -1071.6 | 80.280 | 33.58 | 49.42 | 1243.1 |
| 88 | 34.150 | -3901.3 | -972.99 | 81.414 | 33.33 | 49.24 | 1235.6 |
| 90 | 33.982 | -3817.4 | -874.69 | 82.518 | 33.09 | 49.06 | 1228.3 |
| 92 | 33.816 | -3733.9 | -776.74 | 83.595 | 32.85 | 48.89 | 1220.9 |
| 94 | 33.651 | -3650.8 | -679.13 | 84.644 | 32.63 | 48.72 | 1213.7 |
| 96 | 33.486 | -3568.2 | -581.87 | 85.668 | 32.40 | 48.55 | 1206.5 |
| 98 | 33.323 | -3485.8 | -484.94 | 86.668 | 32.19 | 48.38 | 1199.4 |
| 100 | 33.161 | -3403.9 | -388.34 | 87.644 | 31.98 | 48.22 | 1192.4 |
| 102 | 33.000 | -3322.4 | -292.06 | 88.597 | 31.78 | 48.06 | 1185.4 |
| 104 | 32.839 | -3241.2 | -196.11 | 89.528 | 31.58 | 47.90 | 1178.5 |
| 106 | 32.680 | -3160.4 | -100.48 | 90.439 | 31.39 | 47.74 | 1171.7 |
| 108 | 32.522 | -3080.0 | -5.1555 | 91.330 | 31.20 | 47.58 | 1165.0 |
| 110 | 32.365 | -2999.9 | 89.857 | 92.202 | 31.02 | 47.43 | 1158.3 |
| 112 | 32.208 | -2920.2 | 184.56 | 93.055 | 30.84 | 47.28 | 1151.7 |
| 114 | 32.053 | -2840.9 | 278.97 | 93.891 | 30.66 | 47.13 | 1145.2 |
| 116 | 31.898 | -2761.9 | 373.08 | 94.709 | 30.50 | 46.98 | 1138.7 |
| 118 | 31.745 | -2683.2 | 466.90 | 95.511 | 30.33 | 46.84 | 1132.3 |
| 120 | 31.592 | -2604.9 | 560.44 | 96.297 | 30.17 | 46.69 | 1126.0 |
| 122 | 31.440 | -2527.0 | 653.68 | 97.067 | 30.01 | 46.55 | 1119.8 |
| 124 | 31.289 | -2449.3 | 746.65 | 97.823 | 29.86 | 46.41 | 1113.6 |
| 126 | 31.139 | -2372.0 | 839.34 | 98.565 | 29.71 | 46.28 | 1107.5 |
| 128 | 30.990 | -2295.1 | 931.76 | 99.293 | 29.57 | 46.14 | 1101.5 |
| 130 | 30.842 | -2218.4 | 1023.9 | 100.01 | 29.43 | 46.01 | 1095.5 |
| 132 | 30.695 | -2142.1 | 1115.8 | 100.71 | 29.29 | 45.88 | 1089.7 |
| 134 | 30.548 | -2066.1 | 1207.4 | 101.40 | 29.15 | 45.75 | 1083.9 |
| 136 | 30.403 | - 1990.4 | 1298.8 | 102.07 | 29.02 | 45.62 | 1078.1 |
| 138 | 30.258 | - 1915.0 | 1389.9 | 102.74 | 28.90 | 45.49 | 072.5 |
| 140 | 30.114 | - 1840.0 | 1480.7 | 103.39 | 28.77 | 45.36 | 1066.9 |
| 142 | 29.971 | - 1765.2 | 1571.4 | 104.04 | 28.65 | 45.24 | 1061.4 |
| 144 | 29.829 | - 1690.7 | 1661.7 | 104.67 | 28.53 | 45.12 | 1055.9 |
| 146 | 29.688 | - 1616.6 | 1751.8 | 105.29 | 28.41 | 45.00 | 1050.6 |
| 148 | 29.547 | - 1542.7 | 1841.7 | 105.90 | 28.30 | 44.88 | 1045.3 |
| 150 | 29.408 | - 1469.1 | 1931.3 | 106.50 | 28.19 | 44.76 | 1040. |
| 155 | 29.062 | - 1286.5 | 2154.4 | 107.96 | 27.92 | 44.47 | 1027.3 |
| 160 | 28.722 | -1105.6 | 2376.1 | 109.37 | 27.67 | 44.19 | 1015.0 |
| 165 | 28.387 | -926.45 | 2596.3 | 110.73 | 27.43 | 43.91 | 1003.2 |
| 170 | 28.057 | -748.99 | 2815.2 | 112.03 | 27.20 | 43.65 | 991.7 |
| 175 | 27.732 | -573.18 | 3032.8 | 113.30 | 26.99 | 43.38 | 980.8 |
| 180 | 27.412 | -398.99 | 3249.0 | 114.51 | 26.78 | 43.13 | 970.2 |
| 185 | 27.097 | -226.38 | 3464.1 | 115.69 | 26.59 | 42.88 | 960.0 |
| 190 | 26.787 | -55.313 | 3677.9 | 116.83 | 26.41 | 42.64 | 950.3 |
| 195 | 26.482 | 114.25 | 3890.4 | 117.94 | 26.23 | 42.40 | 941.0 |
| 200 | 26.182 | 282.34 | 4101.8 | 119.01 | 26.07 | 42.16 | 932.0 |
| 210 | 25.596 | 614.24 | 4521.2 | 121.05 | 25.76 | 41.71 | 915.2 |
| 220 | 25.029 | 940.66 | 4936.1 | 122.98 | 25.48 | 41.27 | 899.9 |
| 230 | 24.481 | 1261.9 | 5346.7 | 124.81 | 25.23 | 40.86 | 885.9 |
| 240 | 23.951 | 1578.1 | 5753.2 | 126.54 | 24.99 | 40.45 | 873. |
| 250 | 23.440 | 1889.6 | 6155.8 | 128.18 | 24.78 | 40.07 | 861. |
| 260 | 22.946 22.469 | 2196.6 | 6554.6 | 129.75 | 24.59 | 39.70 | 851. |
| 270 280 | | 2499.3 2798.0 | 6949.8 7341.5 | 131.24 | 24.42 | 39.34 | 841. |
| | 22.009 | | 7341.5 | 132.66 | 24.26 | 39.00 | 833. 825 |
| 290 | 21.566 | 3092.9 | 7729.9 | 134.03 | 24.11 | 38.68 | 825. |
| 300 | 21.138 | 3384.3 3672.2 | 8115.1 | 135.33 | 23.98 | 38.37 38.07 | 818. |
| 310 | 20.725 | 3672.2 | 8497.3 | 136.59 | 23.86 | 38.07 | 812. |
| 320 | 20.327 | 3956.9 4238.6 | 8876.6 | 137.79 | 23.75 | 37.78 37.51 | 806. |
| 330 | 19.942 | 4238.6 | 9253.0 | 138.95 | 23.65 | 37.51 27.26 | 801. |
| 340 | 19.572 | 4517.5 4793.8 | 9626.9 | 140.06 | 23.57 | 37.26 | 797. |
| 250 | | /I /U X X | 9998.2 | 141.14 | 23.49 | 37.01 | 793.3 |
| 350 360 | 19.214 18.869 | 5067.6 | 10 367.0 | 142.18 | 23.42 | 36.78 | 789.9 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|-------------|------------------------|-------------------|------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 380 | 18.214 | 5608.4 | 11 099.0 | 144.16 | 23.30 | 36.36 | 784.1 |
| 390 | 17.904 | 5875.7 | 11 461.0 | 145.10 | 23.25 | 36.16 | 781.8 |
| 400 | 17.604 | 6141.2 | 11 822.0 | 146.01 | 23.21 | 35.98 | 779.8 |
| 450 | 16.245 | 7445.0 | 13 601.0 | 150.20 | 23.10 | 35.22 | 773.7 |
| 500 | 15.089 | 8720.1 | 15 347.0 | 153.88 | 23.12 | 34.69 | 772.4 |
| 550 | 14.095 | 9977.4 | 17 072.0 | 157.17 | 23.22 | 34.34 | 774.3 |
| 600 | 13.231 | 11 225.0 | 18 784.0 | 160.15 | 23.39 | 34.13 | 778.2 |
| 650 | 12.473 | 12 470.0 | 20 487.0 | 162.88 | 23.61 | 34.03 | 783.7 |
| 700 | 11.803 | 13 716.0 | 22 188.0 | 165.40 | 23.86 | 34.01 | 790.1 |
| 750 | 11.206 | 14 966.0 | 23 889.0 | 167.75 | 24.11 | 34.05 | 797.4 |
| 800 | 10.671 | 16 222.0 | 25 593.0 | 169.95 | 24.38 | 34.12 | 805.1 |
| 900 | 9.7481 | 18 757.0 | 29 015.0 | 173.98 | 24.90 | 34.33 | 821.8 |
| 1000 | 8.9805 | 21 326.0 | 32 461.0 | 177.61 | 25.39 | 34.59 | 839.3 |
| 1100 | 8.3307 | 23 929.0 | 35 933.0 | 180.92 | 25.83 | 34.85 | 857.4 |
| 1200 | 7.7726 | 26 565.0 | 39 430.0 | 183.96 | 26.23 | 35.10 | 875.6 |
| 1300 | 7.2877 | 29 230.0 | 42 952.0 | 186.78 | 26.58 | 35.34 | 893.9 |
| 1400 | 6.8619 | 31 923.0 | 46 497.0 | 189.40 | 26.89 | 35.55 | 912.2 |
| 1500 | 6.4847 | 34 641.0 | 50 062.0 | 191.86 | 27.16 | 35.75 | 930.3 |
| 1600 | 6.1481 | 37 381.0 | 53 647.0 | 194.18 | 27.40 | 35.94 | 948.2 |
| 1700 | 5.8456 | 40 142.0 | 57 249.0 | 196.36 | 27.62 | 36.10 | 965.9 |
| 1800 | 5.5723 | 42 921.0 | 60 867.0 | 198.43 | 27.82 | 36.26 | 983.4 |
| 1900 | 5.3239 | 45 717.0 | 64 501.0 | 200.39 | 28.01 | 36.41 | 1000.6 |
| 2000 | 5.0972 | 48 530.0 | 68 148.0 | 202.26 | 28.17 | 36.55 | 1017.7 |
| | | | 200.0 MPa i | | | | |
| 89.73 | 36.712 | -4043.0 | 1404.9 | 76.473 | 35.48 | 48.75 | 1462.1 |
| 90 | 36.693 | -4032.4 | 1418.2 | 76.622 | 35.45 | 48.72 | 1461.4 |
| 92 | 36.561 | -3954.8 | 1515.5 | 77.691 | 35.22 | 48.54 | 1456.1 |
| 94 | 36.430 | -3877.6 | 1612.4 | 78.733 | 35.00 | 48.37 | 1450.9 |
| 96 | 36.299 | -3800.8 | 1708.9 | 79.749 | 34.78 | 48.19 | 1445.8 |
| 98 | 36.170 | -3724.3 | 1805.2 | 80.741 | 34.56 | 48.02 | 1440.7 |
| 100 | 36.042 | -3648.1 | 1901.0 | 81.710 | 34.36 | 47.85 | 1435.7 |
| 102 | 35.914 | -3572.3 | 1996.6 | 82.655 | 34.15 | 47.68 | 1430.7 |
| 104 | 35.788 | -3496.8 | 2091.7 | 83.580 | 33.96 | 47.51 | 1425.8 |
| 106 | 35.662 | -3421.6 | 2186.6 | 84.483 | 33.76 | 47.35 | 1420.9 |
| 108 | 35.538 | -3346.7 | 2281.1 | 85.366 | 33.58 | 47.18 | 1416.1 |
| 110 | 35.414 | -3272.1 | 2375.3 | 86.231 | 33.39 | 47.02 | 1411.4 |
| 112 | 35.291 | -3197.9 | 2469.2 | 87.076 | 33.21 | 46.86 | 1406.6 |
| 114 | 35.170 | -3124.0 | 2562.8 | 87.904 | 33.04 | 46.70 | 1402.0 |
| 116 | 35.049 | -3050.3 | 2656.0 | 88.715 | 32.87 | 46.54 | 1397.4 |
| 118 | 34.929 | -2977.0 | 2748.9 2841.6 | 89.510 | 32.70 | 46.39 | 1392.8 |
| 120 | 34.810 | -2904.0 | | 90.288 | 32.54 | 46.24 | 1388.3 |
| 122 | 34.691 | -2831.3 | 2933.9 | 91.051 | 32.38 | 46.09 | 1383.8 |
| 124 126 | 34.574 34.457 | -2758.8 -2686.6 | 3025.9 | 91.799 92.533 | 32.22 32.07 | 45.94 45.79 | 1379.4 1375.0 |
| 128 | 34.342 | -2686.6 -2614.8 | 3117.6 3209.1 | 93.253 | 31.92 | | 1373.0 |
| | | | | | | 45.65 | 13/0./ |
| 130 132 | 34.227 34.113 | -2543.2 | 3300.2 | 93.960 94.653 | 31.77 | 45.50 | 1366.4 |
| | | -2471.8 | 3391.1 | | 31.63 | 45.36 | |
| 134 | 33.999 | -2400.8 | 3481.7 | 95.334 | 31.49 | 45.22 | 1358.0 |
| 136 | 33.887 | -2330.0 | 3572.0 | 96.003 | 31.35 | 45.08 | 1353.9 |
| 138 | 33.775 | -2259.5 | 3662.0 | 96.661 | 31.22 | 44.95 | 1349.8 |
| 140 | 33.664 | -2189.2 | 3751.8 | 97.306 | 31.09 | 44.81 | 1345.8 |
| 142 | 33.554 | -2119.2 | 3841.3 | 97.941 | 30.96 | 44.68 | 1341.8 |
| 144 | 33.445 | -2049.5 | 3930.5 | 98.565 | 30.84 | 44.55 | 1337.8 |
| 146 | 33.336 | - 1980.0 | 4019.5 | 99.179 | 30.71 | 44.42 | 1333.9 |
| 148 | 33.228 | -1910.8 | 4108.2 | 99.782 | 30.59 | 44.30 | 1330.0 |
| 150 | 33.121 | -1841.8 | 4196.7 | 100.38 | 30.47 | 44.17 | 1326.2 |
| 155 | 32.856 | -1670.3 | 4416.7 | 101.82 | 30.19 | 43.86 | 1316.8 |
| 160 | 32.596 | -1500.4 | 4635.3 | 103.21 | 29.92 | 43.57 | 1307.7 |
| 165 | 32.340 | -1331.8 | 4852.4 | 104.54 | 29.66 | 43.28 | 1298.8 |
| 170 | 32.088 | -1164.7 | 5068.1 | 105.83 | 29.42 | 43.00 | 1290.2 |
| 175 | 31.841 | -998.86 | 5282.4 | 107.07 | 29.18 | 42.73 | 1281.9 |
| 180 185 | 31.597 | -834.33 | 5495.4 | 108.27 | 28.96 | 42.47 | 1273.7 |
| | 31.357 | -671.06 | 5707.1 | 109.43 | 28.74 | 42.21 | 1265.8 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|--------------|------------------------|----------------------|----------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 190 | 31.121 | -509.01 | 5917.5 | 110.56 | 28.54 | 41.97 | 1258.1 |
| 195 | 30.888 | -348.14 | 6126.8 | 111.64 | 28.34 | 41.73 | 1250.7 |
| 200 | 30.660 | -188.42 | 6334.8 | 112.70 | 28.16 | 41.50 | 1243.4 |
| 210 | 30.212 | 127.71 | 6747.6 | 114.71 | 27.81 | 41.05 | 1229.6 |
| 220 | 29.778 | 439.64 | 7156.0 | 116.61 | 27.48 | 40.64 | 1216.5 |
| 230 | 29.357 | 747.62 | 7560.4 | 118.41 | 27.19 | 40.25 | 1204.2 |
| 240 | 28.947 | 1051.9 | 7961.0 | 120.11 | 26.91 | 39.87 | 1192.5 |
| 250 | 28.549 | 1352.6 | 8358.0 | 121.73 | 26.66 | 39.52 | 1181.6 |
| 260 | 28.163 | 1650.0 | 8751.5 | 123.28 | 26.43 | 39.19 | 1171.3 |
| 270 | 27.787 | 1944.3 | 9141.9 | 124.75 | 26.21 | 38.88 | 1161.6 |
| 280 | 27.421 | 2235.6 | 9529.3 | 126.16 | 26.01 | 38.59 | 1152.5 |
| 290 | 27.065 | 2524.1 | 9913.7 | 127.51 | 25.83 | 38.31 | 1143.9 |
| 300 | 26.718 | 2810.0 | 10 295.0 | 128.80 | 25.66 | 38.05 | 1135.8 |
| 310 | 26.381 | 3093.4 | 10 675.0 | 130.05 | 25.50 | 37.80 | 1128.2 |
| 320 | 26.052 | 3374.5 | 11 052.0 | 131.24 | 25.36 | 37.56 | 1121.0 |
| 330 | 25.731 | 3653.4 | 11 426.0 | 132.40 | 25.23 | 37.34 | 1114.3 |
| 340 | 25.419 | 3930.3 | 11 798.0 | 133.51 | 25.10 | 37.13 | 1107.9 |
| 350 | 25.114 | 4205.2 | 12 169.0 | 134.58 | 24.99 | 36.94 | 1101.9 |
| 360 | 24.817 | 4478.3 | 12 537.0 | 135.62 | 24.89 | 36.75 | 1096.3 |
| 370 | 24.527 | 4749.7 | 12 904.0 | 136.62 | 24.80 | 36.58 | 1091.0 |
| 380 | 24.245 | 5019.5 | 13 269.0 | 137.60 | 24.71 | 36.41 | 1086.1 |
| 390 | 23.968 | 5287.8 | 13 632.0 | 138.54 | 24.63 | 36.26 | 1081.4 |
| 400 | 23.699 | 5554.8 | 13 994.0 | 139.46 | 24.57 | 36.12 | 1077.0 |
| 450 | 22.441 | 6872.0 | 15 784.0 | 143.67 | 24.32 | 35.53 | 1058.7 |
| 500 | 21.316 | 8167.3 | 17 550.0 | 147.39 | 24.22 | 35.12 | 1045.4 |
| 550 | 20.305 | 9449.0 | 19 299.0 | 150.73 | 24.23 | 34.86 | 1035.9 |
| 600 | 19.391 | 10 723.0 | 21 037.0 | 153.75 | 24.32 | 34.70 | 1029.4 |
| 650 | 18.562 | 11 995.0 | 22 770.0 | 156.53 | 24.46 | 34.64 | 1025.3 |
| 700 | 17.804 | 13 269.0 | 24 502.0 | 159.10 | 24.64 | 34.63 | 1023.1 |
| 750 | 17.111 | 14 545.0 | 26 234.0 | 161.49 | 24.84 | 34.67 | 1022.5 |
| 800 | 16.473 | 15 828.0 | 27 969.0 | 163.72 | 25.06 | 34.74 | 1023.1 |
| 900 | 15.338 | 18 412.0 | 31 452.0 | 167.83 | 25.50 | 34.93 | 1027.4 |
| 1000 | 14.359 | 21 027.0 | 34 955.0 | 171.52 | 25.92 | 35.15 | 1034.5 |
| 1100 | 13.505 | 23 671.0 | 38 481.0 | 174.88 | 26.31 | 35.37 | 1043.7 |
| 1200 | 12.752 | 26 345.0 | 42 029.0 | 177.97 | 26.66 | 35.59 | 1054.4 |
| 1300 | 12.083 | 29 045.0 31 770.0 | 45 598.0 | 180.82 | 26.97 | 35.79 | 1066.0 |
| 1400 | 11.484 10.944 | | 49 186.0 | 183.48 | 27.24 | 35.97 | 1078.4 |
| 1500 1600 | 10.455 | 34 517.0 37 284.0 | 52 791.0 56 412.0 | 185.97 | 27.49 27.71 | 36.13 | 1091.3 1104.5 |
| | | | | 188.30 | | 36.29 | |
| 1700 | 10.010 | 40 069.0 | 60 048.0 | 190.51 | 27.91 | 36.43 | 1118.0 |
| 1800 1900 | 9.6027 | 42 870.0 | 63 698.0 | 192.60 | 28.09 | 36.56 | 1131.6 |
| 2000 | 9.2283 8.8829 | 45 688.0 | 67 360.0 71 034.0 | 194.58 196.46 | 28.26 28.41 | 36.68 36.80 | 1145.2 |
| 2000 | 0.0029 | 48 519.0 | 500.0 MPa i | | 20.41 | 30.80 | 1158.9 |
| 123.66 | 39.920 | -2818.8 | 9706.4 | 81.122 | 36.30 | 47.19 | 1860.1 |
| 123.00 | | -2818.8 -2807.3 | 9700.4 | 81.252 | | 47.17 | 1859.6 |
| 124 | 39.905 39.820 | - 2807.3 - 2739.8 | 9722.3 9816.7 | 81.252 82.006 | 36.27 36.12 | 47.17 47.04 | 1859.6 |
| 128 | 39.736 | -2739.8 -2672.5 | 9910.6 | 82.745 | 35.97 | 46.90 | 1853.9 |
| 130 | 39.652 | -2672.5 -2605.5 | 10 004.0 | 83.472 | 35.82 | 46.77 | 1851.0 |
| 132 | 39.569 | - 2538.6 | 10 004.0 | 84.185 | 35.67 | 46.77 | 1848.2 |
| 134 | 39.486 | - 2338.6 - 2471.9 | 10 191.0 | 84.885 | 35.53 | 46.51 | 1845.4 |
| 136 | 39.404 | -2471.9 -2405.3 | 10 191.0 | 85.573 | 35.39 35.39 | 46.38 | 1842.7 |
| 138 | 39.322 | -2403.3 -2339.0 | 10 284.0 | 86.249 | 35.25 | 46.25 | 1839.9 |
| 140 | 39.241 | - 2339.0 - 2272.9 | 10 370.0 | 86.914 | 35.12 | 46.13 | 1837.2 |
| 140 | 39.161 | -2272.9 -2206.9 | 10 561.0 | 87.567 | 34.98 | 46.00 | 1834.5 |
| 144 | 39.081 | -2200.9 -2141.1 | 10 653.0 | 88.210 | 34.85 | 45.88 | 1831.8 |
| 144 | 39.002 | -2141.1 -2075.5 | 10 633.0 | 88.842 | 34.83 | 45.75 | 1829.1 |
| 148 | 38.923 | -2075.5 -2010.1 | 10 836.0 | 89.464 | 34.72 | 45.63 | 1826.5 |
| 150 | 38.923 38.845 | - 2010.1 - 1944.8 | 10 836.0 | 90.075 | 34.39 34.47 | 45.51 | 1820.5 |
| 155 | | | | 91.563 | | | |
| | 38.651 | - 1782.4 - 1621.1 | 11 154.0 | 91.563 92.994 | 34.17 33.87 | 45.21 44.92 | 1817.5 1811.2 |
| 160 | 38.461 38.274 | - 1621.1 - 1460.8 | 11 379.0 11 603.0 | 92.994 94.372 | 33.60 | 44.92 44.64 | 1811.2 |
| 165 | | | | | | | |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|--------------|------------------------|----------------------|----------------------|------------------|----------------|----------------|-------------------|
| (K) | (mol/dm ³) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 175 | 37.908 | -1143.2 | 12 047.0 | 96.982 | 33.07 | 44.10 | 1793.1 |
| 180 | 37.729 | -985.89 | 12 267.0 | 98.221 | 32.82 | 43.84 | 1787.4 |
| 185 | 37.553 | -829.45 | 12 485.0 | 99.419 | 32.58 | 43.58 | 1781.7 |
| 190 | 37.380 | -673.92 | 12 702.0 | 100.58 | 32.35 | 43.33 | 1776.2 |
| 195 | 37.209 | -519.26 | 12 918.0 | 101.70 | 32.13 | 43.09 | 1770.8 |
| 200 | 37.041 | -365.46 | 13 133.0 | 102.79 | 31.91 | 42.86 | 1765.5 |
| 210 | 36.711 | -60.311 | 13 560.0 | 104.87 | 31.51 | 42.40 | 1755.2 |
| 220 | 36.391 | 241.69 | 13 981.0 | 106.83 | 31.13 | 41.98 | 1745.3 |
| 230 | 36.079 | 540.72 | 14 399.0 | 108.69 | 30.78 | 41.57 | 1735.8 |
| 240 | 35.776 | 836.94 | 14 813.0 | 110.45 | 30.45 | 41.19 | 1726.7 |
| 250 | 35.480 | 1130.5 | 15 223.0 | 112.12 | 30.14 | 40.82 | 1718.0 |
| 260 | 35.192 | 1421.5 | 15 629.0 | 113.72 | 29.85 | 40.47 | 1709.5 |
| 270 | 34.911 | 1710.2 | 16 032.0 | 115.24 | 29.58 | 40.15 | 1701.4 |
| 280 | 34.636 | 1996.6 | 16 432.0 | 116.69 | 29.33 | 39.84 | 1693.6 |
| 290 | 34.368 | 2280.8 | 16 829.0 | 118.09 | 29.10 | 39.54 | 1686.1 |
| 300 | 34.106 | 2563.1 | 17 223.0 | 119.42 | 28.88 | 39.27 | 1678.8 |
| 310 | 33.850 | 2843.4 | 17 615.0 | 120.70 | 28.67 | 39.00 | 1671.8 |
| 320 | 33.599 | 3122.0 | 18 003.0 | 121.94 | 28.47 | 38.75 | 1665.0 |
| 330 | 33.354 | 3398.9 | 18 390.0 | 123.13 | 28.29 | 38.52 | 1658.5 |
| 340 | 33.114 | 3674.2 | 18 774.0 | 124.27 | 28.12 | 38.30 | 1652.2 |
| 350 | 32.878 | 3948.0 | 19 156.0 | 125.38 | 27.96 | 38.09 | 1646.0 |
| 360 | 32.648 | 4220.5 | 19 536.0 | 126.45 | 27.82 | 37.89 | 1640.1 |
| 370 | 32.421 | 4491.6 | 19 913.0 | 127.49 | 27.68 | 37.70 | 1634.4 |
| 380 | 32.200 | 4761.5 | 20 290.0 | 128.49 | 27.55 | 37.52 | 1628.8 |
| 390 | 31.982 | 5030.3 | 20 664.0 | 129.46 | 27.43 | 37.36 | 1623.4 |
| 400 | 31.769 | 5298.0 | 21 037.0 | 130.41 | 27.32 | 37.20 | 1618.2 |
| 450 | 30.756 | 6623.2 | 22 880.0 | 134.75 | 26.89 | 36.56 | 1594.3 |
| 500 | 29.826 | 7932.1 | 24 696.0 | 138.58 | 26.61 | 36.11 | 1573.6 |
| 550 | 28.965 | 9231.2 | 26 494.0 | 142.00 | 26.47 | 35.81 | 1555.7 |
| 600 | 28.164 | 10 526.0 | 28 279.0 | 145.11 | 26.41 | 35.63 | 1540.0 |
| 650 | 27.416 | 11 820.0 | 30 058.0 | 147.96 | 26.43 | 35.53 | 1526.5 |
| 700 | 26.714 | 13 116.0 | 31 833.0 | 150.59 | 26.50 | 35.49 | 1514.8 |
| 750 | 26.054 | 14 417.0 | 33 608.0 | 153.04 | 26.60 | 35.51 | 1504.7 |
| 800 | 25.431 | 15 724.0 | 35 384.0 | 155.33 | 26.72 | 35.55 | 1496.1 |
| 900 | 24.283 | 18 356.0 | 38 947.0 | 159.53 | 26.99 | 35.70 | 1482.8 |
| 1000 | 23.247 | 21 018.0 | 42 526.0 | 163.30 | 27.27 | 35.89 | 1473.8 |
| 1100 | 22.305 | 23 707.0 | 46 124.0 | 166.73 | 27.54 | 36.08 | 1468.3 |
| 1200 1300 | 21.443 | 26 424.0 29 164.0 | 49 741.0 53 375.0 | 169.87 | 27.79 | 36.26 | 1465.5 1465.0 |
| | 20.651 | | | 172.78 | 28.01 | 36.43 | |
| 1400 | 19.921 | 31 926.0 | 57 026.0 | 175.49 | 28.21 | 36.58 | 1466.3 |
| 1500 1600 | 19.243 | 34 708.0 | 60 691.0 | 178.02 | 28.39 | 36.72 | 1469.2 |
| 1700 | 18.614 18.027 | 37 508.0 40 324.0 | 64 370.0 68 061.0 | 180.39 182.63 | 28.55 28.70 | 36.85 | 1473.2 1478.2 |
| 1800 | | | | 184.74 | 28.83 | 36.97 37.08 | 1478.2 |
| 1900 | 17.478 16.963 | 43 155.0 | 71 763.0 | 186.75 | | | 1490.5 |
| 2000 | 16.480 | 46 000.0 48 857.0 | 75 476.0 79 198.0 | 188.66 | 28.95 29.07 | 37.18 37.27 | 1490.3 |
| 2000 | 10.460 | 46 63 7.0 | 1000.0 MPa i | | 29.07 | 31.21 | 1497.0 |
| 167.86 | 12 115 | _726.70 | | | 37.29 | 16.71 | 2212.2 |
| 170 | 43.415 43.352 | -736.70 | 22 297.0 22 397.0 | 85.397 | 37.17 | 46.74 | 2313.2 2311.1 |
| 170 | | -670.08 | | 85.987 87.335 | 36.90 | 46.63 | 2311.1 |
| | 43.208 | -514.75 | 22 629.0 | | | 46.38 | |
| 180 185 | 43.065 42.925 | -360.10 -206.12 | 22 860.0 23 091.0 | 88.639 89.899 | 36.65 36.40 | 46.14 45.90 | 2301.7 2297.0 |
| 190 | | - 206.12 - 52.792 | | 91.120 | 36.40 36.15 | | 2297.0 |
| 190 | 42.786 42.649 | - 52.792 99.887 | 23 319.0 23 547.0 | 92.303 | 35.92 | 45.66 45.43 | 2292.5 |
| 200 | 42.514 | 251.93 | 23 774.0 | 92.303 | 35.92 35.69 | 45.43 45.20 | 2288.0 |
| 210 | 42.249 | 554.16 | | 95.451 95.645 | 35.09 35.25 | 45.20 44.76 | 2283.0 |
| 210 | | | 24 224.0 | 95.645 97.718 | | | |
| 230 | 41.990 41.738 | 853.98 | 24 669.0 | 97.718 99.679 | 34.84 34.45 | 44.33 43.92 | 2266.7 2258.6 |
| 240 | | 1151.5 | 25 110.0 25 548 0 | 101.54 | 34.45 | | 2250.8 |
| | 41.493 | 1446.8 | 25 548.0 | | | 43.53 | |
| 250 | 41.252 | 1740.0 | 25 981.0 | 103.31 | 33.74 | 43.16 | 2243.3 |
| 260 | 41.018 | 2031.1 | 26 411.0 | 105.00 106.60 | 33.41 33.10 | 42.80 42.45 | 2235.9 2228.8 |
| 270 | 40.789 | 2320.3 | 26 837.0 | | | | |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature | Density | Internal energy | Enthalpy | Entropy | c_v | c_p | Speed of sound |
|-------------|---------------------|----------------------|------------------------|------------------|----------------|----------------|------------------|
| (K) | (mol/dm^3) | (J/mol) | (J/mol) | J/(mol-K) | J/(mol-K) | J/(mol-K) | (m/s) |
| 290 | 40.345 | 2893.3 | 27 679.0 | 109.61 | 32.53 | 41.81 | 2215.1 |
| 300 | 40.130 | 3177.3 | 28 096.0 | 111.03 | 32.27 | 41.51 | 2208.5 |
| 310 | 39.920 | 3459.6 | 28 510.0 | 112.38 | 32.02 | 41.22 | 2202.1 |
| 320 | 39.714 | 3740.5 | 28 921.0 | 113.69 | 31.79 | 40.95 | 2195.9 |
| 330 | 39.512 | 4019.9 | 29 329.0 | 114.94 | 31.57 | 40.69 | 2189.8 |
| 340 | 39.314 | 4297.9 | 29 734.0 | 116.15 | 31.36 | 40.44 | 2183.9 |
| 350 | 39.119 | 4574.7 | 30 138.0 | 117.32 | 31.17 | 40.21 | 2178.1 |
| 360 | 38.928 | 4850.3 | 30 539.0 | 118.45 | 30.98 | 39.98 | 2172.4 |
| 370 | 38.741 | 5124.8 | 30 937.0 | 119.55 | 30.81 | 39.77 | 2166.9 |
| 380 390 | 38.557 38.376 | 5398.2 5670.7 | 31 334.0 31 729.0 | 120.60 121.63 | 30.65 30.49 | 39.57 39.38 | 2161.5 |
| 400 | 38.198 | 5942.2 | 31 729.0 | 121.63 | 30.49 | 39.38 39.19 | 2156.2 2151.1 |
| 450 | 37.351 | 7288.1 | 34 061.0 | 127.19 | 29.75 | 38.42 | 2126.8 |
| 500 | 36.567 | 8619.7 | 35 967.0 | 131.21 | 29.33 | 37.84 | 2120.8 |
| 550 | 35.836 | 9942.8 | 37 848.0 | 134.80 | 29.06 | 37.42 | 2084.6 |
| 600 | 35.150 | 11 262.0 | 39 711.0 | 138.04 | 28.88 | 37.13 | 2066.2 |
| 650 | 34.505 | 12 581.0 | 41 562.0 | 141.00 | 28.79 | 36.93 | 2049.3 |
| 700 | 33.895 | 13 902.0 | 43 405.0 | 143.73 | 28.75 | 36.80 | 2033.9 |
| 750 | 33.316 | 15 228.0 | 45 243.0 | 146.27 | 28.76 | 36.73 | 2019.9 |
| 800 | 32.765 | 16 559.0 | 47 079.0 | 148.64 | 28.80 | 36.69 | 2007.0 |
| 900 | 31.736 | 19 238.0 | 50 748.0 | 152.96 | 28.92 | 36.70 | 1984.7 |
| 1000 | 30.791 | 21 944.0 | 54 421.0 | 156.83 | 29.07 | 36.77 | 1966.3 |
| 1100 | 29.916 | 24 676.0 | 58 103.0 | 160.34 | 29.22 | 36.86 | 1951.2 |
| 1200 | 29.101 | 27 431.0 | 61 794.0 | 163.55 | 29.36 | 36.96 | 1939.0 |
| 1300 | 28.338 | 30 207.0 | 65 495.0 | 166.51 | 29.49 | 37.06 | 1929.2 |
| 1400 | 27.622 | 33 003.0 | 69 206.0 | 169.26 | 29.60 | 37.16 | 1921.4 |
| 1500 | 26.946 | 35 816.0 | 72 927.0 | 171.83 | 29.70 | 37.25 | 1915.5 |
| 1600 | 26.307 | 38 644.0 | 76 656.0 | 174.24 | 29.79 | 37.34 | 1911.0 |
| 1700 | 25.701 | 41 486.0 | 80 395.0 | 176.50 | 29.88 | 37.42 | 1908.0 |
| 1800 | 25.125 | 44 341.0 | 84 141.0 | 178.65 | 29.95 | 37.51 | 1906.1 |
| 1900 | 24.577 | 47 208.0 | 87 896.0 | 180.68 | 30.02 | 37.58 | 1905.2 |
| 2000 | 24.054 | 50 085.0 | 91 658.0 | 182.60 | 30.08 | 37.66 | 1905.3 |
| 236.19 | 47.967 | 3353.0 | 2000.0 MPa 45 048.0 | 90.033 | 38.84 | 46.92 | 2923.5 |
| 240 | 47.893 | 3466.7 | 45 227.0 | 90.783 | 38.69 | 46.79 | 2920.7 |
| 250 | 47.700 | 3764.2 | 45 693.0 | 92.685 | 38.33 | 46.43 | 2913.6 |
| 260 | 47.512 | 4060.2 | 46 155.0 | 94.499 | 37.98 | 46.08 | 2906.7 |
| 270 | 47.327 | 4354.8 | 46 614.0 | 96.232 | 37.64 | 45.74 | 2899.8 |
| 280 | 47.145 | 4648.0 | 47 070.0 | 97.890 | 37.32 | 45.41 | 2893.2 |
| 290 | 46.968 | 4939.9 | 47 522.0 | 99.477 | 37.01 | 45.09 | 2886.7 |
| 300 | 46.793 | 5230.5 | 47 972.0 | 101.00 | 36.72 | 44.78 | 2880.3 |
| 310 | 46.622 | 5519.8 | 48 418.0 | 102.46 | 36.44 | 44.49 | 2874.1 |
| 320 | 46.454 | 5808.0 | 48 862.0 | 103.87 | 36.17 | 44.20 | 2867.9 |
| 330 | 46.289 | 6094.9 | 49 302.0 | 105.23 | 35.92 | 43.92 | 2861.9 |
| 340 | 46.126 | 6380.8 | 49 740.0 | 106.54 | 35.68 | 43.66 | 2856.1 |
| 350 | 45.967 | 6665.7 | 50 175.0 | 107.80 | 35.44 | 43.40 | 2850.3 |
| 360 | 45.810 | 6949.6 | 50 608.0 | 109.02 | 35.22 | 43.15 | 2844.6 |
| 370 | 45.656 | 7232.5 | 51 038.0 | 110.20 | 35.02 | 42.92 | 2839.0 |
| 380 | 45.504 | 7514.5 | 51 466.0 | 111.34 | 34.82 | 42.69 | 2833.6 |
| 390 | 45.355 | 7795.7 | 51 892.0 | 112.44 | 34.63 | 42.48 | 2828.2 |
| 400 | 45.208 | 8076.2 | 52 316.0 | 113.52 | 34.45 | 42.27 | 2822.9 |
| 450 500 | 44.506 43.852 | 9468.3 | 54 406.0 56 455.0 | 118.44 | 33.68 | 41.36 | 2797.6 2774.1 |
| 550 | 43.852 | 10 848.0 12 220.0 | 58 473.0 | 122.76 126.61 | 33.11 32.68 | 40.65 40.09 | 2774.1 |
| 600 | 42.664 | 13 588.0 | 60 466.0 | 130.07 | 32.37 | 39.66 | 2732.1 |
| 650 | 42.120 | 14 956.0 | 62 440.0 | 133.23 | 32.37 | 39.32 | 2731.0 |
| 700 | 41.603 | 16 326.0 | 64 400.0 | 136.14 | 32.00 | 39.32 | 2694.4 |
| 750 | 41.111 | 17 700.0 | 66 349.0 | 138.83 | 31.90 | 38.88 | 2677.5 |
| 800 | 40.641 | 19 078.0 | 68 289.0 | 141.33 | 31.84 | 38.73 | 2661.8 |
| 900 | 39.760 | 21 849.0 | 72 151.0 | 145.88 | 31.78 | 38.53 | 2633.2 |
| 1000 | 38.945 | 24 643.0 | 75 998.0 | 149.94 | 31.77 | 38.41 | 2608.2 |
| 1100 | 38.184 | 27 458.0 | 79 835.0 | 153.59 | 31.79 | 38.34 | 2586.1 |
| 1200 | 37.471 | 30 292.0 | 83 666.0 | 156.93 | 31.80 | 38.29 | 2566.7 |

TABLE A2. Thermodynamic properties of air—Continued

| Temperature (K) | Density (mol/dm ³) | Internal energy (J/mol) | Enthalpy (J/mol) | Entropy J/(mol-K) | c_v J/(mol-K) | c_p J/(mol-K) | Speed of sound (m/s) |
|-----------------|--------------------------------|-------------------------------|------------------|----------------------|-----------------|-----------------|----------------------------|
| 1300 | 36.799 | 33 143.0 | 87 493.0 | 159.99 | 31.82 | 38.25 | 2549.5 |
| 1400 | 36.162 | 36 010.0 | 91 317.0 | 162.82 | 31.83 | 38.22 | 2534.2 |
| 1500 | 35.556 | 38 889.0 | 95 138.0 | 165.46 | 31.84 | 38.21 | 2520.6 |
| 1600 | 34.978 | 41 780.0 | 98 958.0 | 167.93 | 31.85 | 38.20 | 2508.5 |
| 1700 | 34.425 | 44 680.0 | 102 780.0 | 170.24 | 31.85 | 38.19 | 2497.8 |
| 1800 | 33.894 | 475 90.0 | 106 600.0 | 172.42 | 31.85 | 38.19 | 2488.2 |
| 1900 | 33.384 | 505 07.0 | 110 420.0 | 174.49 | 31.86 | 38.20 | 2479.6 |
| 2000 | 32.893 | 534 33.0 | 114 240.0 | 176.45 | 31.86 | 38.21 | 2472.1 |