

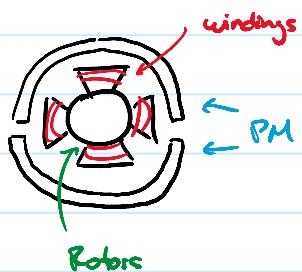
Lecture 3

September 14, 2021 3:10 PM

Non-Linear State Space Control

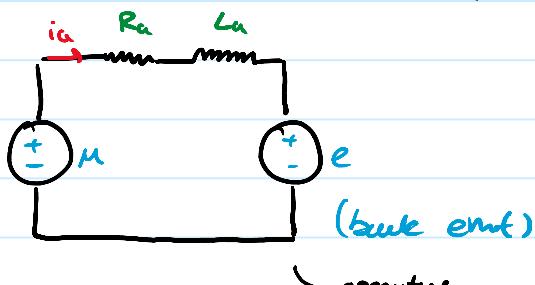
e.g. Permanent Magnet DC (PMDC)

position and speed control



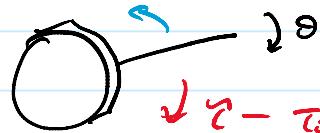
As the rotor rotates, windings move in a magnetic field (produced by PM's)

- ① Faraday's Law: Back EMF voltage induced in the windings
- ② Lorentz's Law: Current in the windings produces a force (torque) due to B field from PM's



Electrical

Friction Torque



Rotor

$\tau_f = -\gamma \dot{\theta}$ - Torque produced by windings

$$e = k_e \dot{\theta}$$

back emf constant

$$\tau = k_m i_a$$

motor torque constant

Armature:

$$La \frac{di_a}{dt} + Ra i_a + e = M$$

$k_e \dot{\theta}$

Coupled

$$I \ddot{\theta} = -\tau_{friction} + \tau$$

$$k_m i_a$$

assume $\tau_{friction} = b \dot{\theta}$ (viscous friction)

Rotor

Choose some states:

$$x_1 = \theta, x_2 = \dot{\theta}, x_3 = i_a$$

$$\dot{x}_1 = \dot{\theta} = x_2$$

$$\dot{x}_2 = \ddot{\theta} = -\frac{1}{J}(-b\dot{\theta} + k_t i_a) = -\frac{b}{J}x_2 + \frac{k_t}{J}x_3$$

$$\dot{x}_3 = \frac{dia}{dt} = -\frac{R_a}{L_a}i_a - \frac{k_e}{L_a}\dot{\theta} + \frac{1}{L_a}u = -\frac{R_a}{L_a}x_3 - \frac{k_e}{L_a}x_2 + \frac{1}{L_a}u$$

$$y = \theta = x_1 \quad (\text{position control})$$

$$y = \dot{\theta} = x_2 \quad (\text{speed control})$$

$$x = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} \quad \dot{x} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & -\frac{b}{J} & \frac{k_t}{J} \\ 0 & -\frac{k_e}{L_a} & -\frac{R_a}{L_a} \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{L_a} \end{bmatrix} u$$

$$y = [1 \ 0 \ 0]x \quad \text{or} \quad y = [0 \ 1 \ 0]x$$

LTI in state space model!

Summary:

Ex 1

$$\ddot{v} = -\frac{B}{M}v + \frac{1}{M}u \quad (\text{neglecting disturbance})$$

$$y = v$$

LTI in both I/O and state space

Ex 2

$$LC \frac{d^2y}{dt^2} + RL \frac{dy}{dt} + y = u$$

LTI I/O Model

$$\dot{x} = \begin{bmatrix} 0 & \frac{1}{C} \\ -\frac{1}{L} & -\frac{R}{L} \end{bmatrix}x + \begin{bmatrix} 0 \\ \frac{1}{L} \end{bmatrix}u$$

LTI State space model

$$y = [1 \ 0]x$$

Ex3

$$\dot{x}_1 = x_2$$

(neglecting disturbances)

$$\dot{x}_2 = -\frac{mgl}{I} \sin(x_1) + \frac{1}{I} u$$

$$y = x$$

Nonlinear State Space Model

Control Systems Models

↳ Non-Linear Time Invariant State Space Models

$$x = \begin{bmatrix} x_1 \\ \vdots \\ x_n \end{bmatrix} \quad x_i = f_i(x_1, \dots, x_n, u_1, \dots, u_m) \quad i=1, \dots, n$$

$$y_j = h_i(x_1, \dots, x_n, u_1, \dots, u_m) \quad j=1, \dots, p$$

$$\begin{aligned} \dot{x} &= f(x, u) \\ y &= h(x, u) \end{aligned} \quad x = \begin{bmatrix} x_1 \\ \vdots \\ x_n \end{bmatrix} \quad u = \begin{bmatrix} u_1 \\ \vdots \\ u_m \end{bmatrix} \quad y = \begin{bmatrix} y_1 \\ \vdots \\ y_p \end{bmatrix}$$

$$f = \begin{bmatrix} f_1(\cdot) \\ \vdots \\ f_n(\cdot) \end{bmatrix} \quad h = \begin{bmatrix} h_1(\cdot) \\ \vdots \\ h_p(\cdot) \end{bmatrix}$$

↳ LTI state space models

$$\dot{x}_i = a_{ii}x_i + \dots + a_{in}x_n + b_{i1}u_1 + \dots + b_{im}u_m \quad i=1, \dots, n$$

$$y_j = c_{j1}x_1 + \dots + c_{jn}x_n + d_{j1}u_1 + \dots + d_{jm}u_m$$

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

↳ LTI Input/Output (I/O) Models

1 input (u) — 1 output (y)

$$\frac{dy}{dt^n} + a_{n-1} \frac{d^{n-1}y}{dt^{n-1}} + \dots + a_1 \frac{dy}{dt} + a_0 y = b_m \frac{d^m u}{dt^m} + \dots + b_0 u$$