

Tightly-Coupled Visual-Inertial Localization and 3-D Rigid-Body Target Tracking

Kevin Eickenhoff, Yulin Yang, Patrick Geneva, and Guoquan Huang

Speaker: Erqun Dong

3-D Tracking 和VIO的联合估计

Tightly-Coupled Visual-Inertial Localization
and 3D Rigid-Body Target Tracking

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主要思想

- 将3D Tracking加到MSCKF的状态中，做联合估计
- 如何表示Target的位姿？
- 如何表示Target的移动模型？

MSCKF简单回顾

- MSCKF是一个error-state的EKF

误差状态的EKF模型

- 一个完整的采用误差状态EKF的估计系统

一套单独的系统状态递推方程（并非滤波器预测方程），用来预测系统状态：

$$\hat{\mathbf{X}}_{k+1|k} = \Phi_k^{k+1} (\hat{\mathbf{X}}_{k|k})$$

误差状态的EKF方程，当有效测量到来时才进行计算：

$$\mathbf{P}_{k+1|k} = \mathbf{f}_k^{k+1} \mathbf{P}_{k|k} \mathbf{f}_k^{k+1T} + \mathbf{Q}_k$$

$$\mathbf{K}_{k+1} = \mathbf{P}_{k+1|k} \mathbf{h}_{k+1}^T (\mathbf{h}_{k+1} \mathbf{P}_{k+1|k} \mathbf{h}_{k+1}^T + \mathbf{R}_{k+1})^{-1}$$

$$\delta \hat{\mathbf{X}}_{k+1|k+1} = \mathbf{K}_{k+1} \mathbf{r}_{k+1}$$

$$= \mathbf{K}_{k+1} (\mathbf{Z}_{k+1} - \mathbf{H}_{k+1} (\hat{\mathbf{X}}_{k+1|k}))$$

$$\mathbf{P}_{k+1|k+1} = (\mathbf{I} - \mathbf{K}_{k+1} \mathbf{h}_{k+1}) \mathbf{P}_{k+1|k}$$

其中有： $\mathbf{r}_{k+1} = \mathbf{Z}_{k+1} - \mathbf{H}_{k+1} (\hat{\mathbf{X}}_{k+1|k})$

注意小 \mathbf{f} 和小 \mathbf{h} 所代表的意义：

$$\delta \mathbf{X}_{k+1} = \mathbf{f}_k^{k+1} \delta \mathbf{X}_k + \mathbf{w}$$

$$\mathbf{r} = \mathbf{h} \delta \mathbf{X} + \mathbf{v}$$

最终的系统输出： $\hat{\mathbf{X}}_{k+1|k+1} = \hat{\mathbf{X}}_{k+1|k} + \delta \hat{\mathbf{X}}_{k+1|k+1}$

MSCKF简单回顾

- 某一时刻的状态

$$\mathbf{x}_I = \left[{}^I_G \bar{\mathbf{q}}^\top \quad \mathbf{b}_\omega^\top \quad {}^G \mathbf{v}_I^\top \quad \mathbf{b}_a^\top \quad {}^G \mathbf{p}_I^\top \right]^\top$$

- Error-state

$$\delta \mathbf{x}_I = \left[{}^I \delta \boldsymbol{\theta}_G^\top \quad \delta \mathbf{b}_\omega^\top \quad {}^G \delta \mathbf{v}_I^\top \quad \delta \mathbf{b}_a^\top \quad {}^G \delta \mathbf{p}_I^\top \right]^\top$$

MSCKF简单回顾

- 测量值

$$\mathbf{a}_m = \mathbf{a} + {}^I_G \mathbf{R}^G \mathbf{g} + \mathbf{b}_a + \mathbf{n}_a, \quad \boldsymbol{\omega}_m = \boldsymbol{\omega} + \mathbf{b}_\omega + \mathbf{n}_\omega$$

- 状态方程。由下面这个方程组就能推出error-state的状态递推方程

$${}^I_G \dot{\bar{\mathbf{q}}} = \frac{1}{2} \boldsymbol{\Omega}(\boldsymbol{\omega}) {}^I_G \bar{\mathbf{q}}, \quad {}^G \dot{\mathbf{v}} = {}^I_G \mathbf{R}^\top \mathbf{a}, \quad {}^G \dot{\mathbf{p}} = {}^G \mathbf{v}$$

$$\dot{\mathbf{b}}_\omega = \mathbf{n}_{b\omega}, \quad \dot{\mathbf{b}}_a = \mathbf{n}_{ba}$$

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MSCKF简单回顾

- 滑窗内的所有状态

$$\mathbf{x}_k = [\mathbf{x}_I \quad \mathbf{x}_{cl}]^\top$$

$$\mathbf{x}_{cl} = \left[\begin{array}{cc|ccc} I_{k-1} & \bar{q}^\top & {}^G\mathbf{p}_{I_{k-1}}^\top & & & \\ G & & & \cdots & & \\ & & & & I_{k-m} & \bar{q}^\top \\ & & & & G & {}^G\mathbf{p}_{I_{k-m}}^\top \end{array} \right]^\top$$

- 视觉观测类似VINS，就是滑窗内Track到的特征点，每个点也是用逆深度来参数化。用滑窗里的观测来对逆深度做三角化。

$$\mathbf{z}_i = \mathbf{\Pi}({}^{C_i}\mathbf{p}_{fs}) + \mathbf{n}_{fi}, \quad \mathbf{\Pi}([x \ y \ z]^\top) = \begin{bmatrix} \frac{x}{z} & \frac{y}{z} \end{bmatrix}^\top$$

$${}^{C_i}\mathbf{p}_{fs} = {}^C_I \mathbf{R}_G^{I_i} \mathbf{R} ({}^G\mathbf{p}_{fs} - {}^G\mathbf{p}_{I_i}) + {}^C\mathbf{p}_I$$

Feature static

$${}^G\mathbf{p}_{fs} = {}^G_{C_a} \mathbf{R} \left(\frac{1}{\rho} \begin{bmatrix} \alpha \\ \beta \\ 1 \end{bmatrix} \right) + {}^G\mathbf{p}_{C_a}$$

anchor

MSCKF简单回顾

- 上式做一下等价变换（为了保证很远点的数值稳定性，乘以逆深度，里面变成归一化齐次坐标）。系统的观测方程就是

$$\mathbf{z}_i = \mathbf{\Pi}(\rho^{C_i} \mathbf{p}_{fs}) + \mathbf{n}_{fi}$$

- 对上面的方程线性化，就得到观测方程的error-state形式：

$$\delta \mathbf{z} = \mathbf{H}_x \delta \mathbf{x} + \mathbf{H}_f \delta \mathbf{m}_f + \mathbf{n}_f$$

- 但是MSCKF要解决EKF中存储特征点很占空间的问题。Trick：使用 H_f 的左零空间矩阵 Q_2^T 左乘上式，把和feature相关的这一项消掉

$$\delta \mathbf{z}' = \mathbf{H}'_x \delta \mathbf{x} + \mathbf{n}'_f$$

- MSCKF中把这个左零空间变换后的方程作为观测的error-state

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Target State Estimation

- 在这篇文章之前，Target State往往用单点来描述。但这样回丢失很多几何信息，也会因为遮挡等情况而出问题。
- 这篇文章用一个Representative Point, ${}^G\mathbf{p}_T$ 和一个orientation ${}^T_G\bar{q}$, 来表述Target的状态。
- 此外，还跟踪一些Non-Representative Point。Non-Representative Point像static feature一样，是在整个滑窗内Tracking到的所有点。区别是Non-Representative Point是表述在Target系里的。

Target State Estimation

- Representative Point的观测方程和static feature基本一致，只不过这里 ${}^G\mathbf{p}_T$ 是运动的，每个时刻都要优化一个新值：

$$\mathbf{z}_i = \mathbf{\Pi}({}^{C_i}\mathbf{p}_T) + \mathbf{n}_{fi}$$

$${}^{C_i}\mathbf{p}_T = {}^C_I \mathbf{R}_G^{I_i} \mathbf{R} ({}^G\mathbf{p}_T - {}^G\mathbf{p}_{I_i}) + {}^C\mathbf{p}_I$$

- Non-Representative Point的观测方程。由于每个Non-Representative Point都是在Target的坐标系里描述的，所以要多转换一次坐标系：

$$\mathbf{z}_i = \mathbf{\Pi}({}^{C_i}\mathbf{p}_{ft}) + \mathbf{n}_{fi}$$

$${}^{C_i}\mathbf{p}_{ft} = {}^C_I \mathbf{R}_G^{I_i} \mathbf{R} \left({}^G\mathbf{p}_{T_i} + {}^T_i \mathbf{R}^{\top T} \mathbf{p}_{ft} - {}^G\mathbf{p}_{I_i} \right) + {}^C\mathbf{p}_I$$

Target State Estimation

- 关于Non-Representative Point的观测就可以和static feature合并在一起。然后也用左零空间消掉。这样EKF中就不用存储Non-Representative Point的信息了。

Target State Estimation

- MSCKF中还要求给出feature的初始估计。Non-Representative Point 满足这个方程。其中d是深度，r是feature的归一化齐次坐标：

$${}^T\mathbf{p}_{ft} = {}^{T_i}_G\mathbf{R} \left({}^G\mathbf{p}_{C_i} - {}^G\mathbf{p}_{T_i} \right) + d_i {}^{T_i}_G\mathbf{R} {}^G_{C_i}\mathbf{R} {}^{C_i}\mathbf{r}_i$$

- 上面的方程左乘 ${}^{T_i}_G\mathbf{R} {}^G_{C_i}\mathbf{R} {}^{C_i}\mathbf{r}_i$ 对应的反对称矩阵，刚好就消掉了这一项：

$$\left[{}^{T_i}_G\mathbf{R} {}^G_{C_i}\mathbf{R} {}^{C_i}\mathbf{r}_i \right] {}^T\mathbf{p}_{ft} = \left[{}^{T_i}_G\mathbf{R} {}^G_{C_i}\mathbf{R} {}^{C_i}\mathbf{r}_i \right] {}^{T_i}_G\mathbf{R} \left({}^G\mathbf{p}_{C_i} - {}^G\mathbf{p}_{T_i} \right)$$

- 然后就可以解出 ${}^T\mathbf{p}_{ft}$ 的初值。之后可以对 ${}^T\mathbf{p}_{ft}$ 做一个local BA。

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Target Motion Model

- Model 1. Constant Global Linear Velocity. 这个模型假设Target在Global系下的速度和角速度在短时间内是恒定的。这种一般发生在速度和角速度是解耦的情况，比如UAV。
- 状态、状态方程、Error-state的状态方程分别如下：

$$\mathbf{x}_T^{(1)} = [{}^T_G \bar{\mathbf{q}}^\top \quad {}^T \boldsymbol{\omega}^\top \quad {}^G \mathbf{p}_T^\top \quad {}^G \mathbf{v}_T^\top]^\top$$

$${}^T_G \dot{\bar{\mathbf{q}}} = \frac{1}{2} \boldsymbol{\Omega}({}^T \boldsymbol{\omega}) {}^T_G \bar{\mathbf{q}}, \quad {}^G \dot{\mathbf{p}}_T = {}^G \mathbf{v}_T, \quad {}^G \dot{\mathbf{v}}_T = \mathbf{n}_{tv}, \quad {}^T \dot{\boldsymbol{\omega}} = \mathbf{n}_{t\omega}$$

$$\begin{bmatrix} {}^T \delta \dot{\boldsymbol{\theta}}_G \\ {}^T \delta \dot{\boldsymbol{\omega}} \\ {}^G \delta \dot{\mathbf{p}}_T \\ {}^G \delta \dot{\mathbf{v}}_T \end{bmatrix} = \begin{bmatrix} -[{}^T \hat{\boldsymbol{\omega}}] & \mathbf{I}_3 & \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{I}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 \end{bmatrix} \begin{bmatrix} {}^T \delta \boldsymbol{\theta}_G \\ {}^T \delta \boldsymbol{\omega} \\ {}^G \delta \mathbf{p}_T \\ {}^G \delta \mathbf{v}_T \end{bmatrix} + \begin{bmatrix} \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{0}_3 & \mathbf{I}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{I}_3 & \mathbf{0}_3 \end{bmatrix} \begin{bmatrix} \mathbf{n}_{tv} \\ \mathbf{n}_{t\omega} \end{bmatrix}$$

Target Motion Model

- Model 2. Constant Local Linear Velocity. 这个模型假设Target的速度在Target系下是短时间恒定的。但是Target的角速度还是在Global系下是短时间恒定的。比如地面的车辆、固定翼飞行器。
- 状态、状态方程、Error-state的状态方程分别如下：

$$\mathbf{x}_T^{(2)} = \begin{bmatrix} {}^T_G \bar{\mathbf{q}}^\top & {}^T \boldsymbol{\omega}^\top & {}^G \mathbf{p}_T^\top & {}^T \mathbf{v}_T^\top \end{bmatrix}^\top$$

$${}^T_G \dot{\bar{\mathbf{q}}} = \frac{1}{2} \boldsymbol{\Omega}({}^T \boldsymbol{\omega}) {}^T_G \bar{\mathbf{q}}, \quad {}^G \dot{\mathbf{p}}_T = {}^G_T \mathbf{R}^T \mathbf{v}_T, \quad {}^T \dot{\mathbf{v}}_T = \mathbf{n}_{tv}, \quad {}^T \dot{\boldsymbol{\omega}} = \mathbf{n}_{tw}$$

$$\begin{bmatrix} {}^T \delta \dot{\boldsymbol{\theta}}_G \\ {}^T \delta \dot{\boldsymbol{\omega}} \\ {}^G \delta \dot{\mathbf{p}}_T \\ {}^T \delta \dot{\mathbf{v}}_T \end{bmatrix} = \begin{bmatrix} -[{}^T \hat{\boldsymbol{\omega}}] & \mathbf{I}_3 & \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 \\ -{}^G_T \hat{\mathbf{R}} [{}^T \hat{\mathbf{v}}_T] & \mathbf{0}_3 & \mathbf{0}_3 & {}^G_T \hat{\mathbf{R}} \\ \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 & \mathbf{0}_3 \end{bmatrix} \begin{bmatrix} {}^T \delta \boldsymbol{\theta}_G \\ {}^T \delta \boldsymbol{\omega} \\ {}^G \delta \mathbf{p}_T \\ {}^T \delta \mathbf{v}_T \end{bmatrix} + \begin{bmatrix} \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{0}_3 & \mathbf{I}_3 \\ \mathbf{0}_3 & \mathbf{0}_3 \\ \mathbf{I}_3 & \mathbf{0}_3 \end{bmatrix} \begin{bmatrix} \mathbf{n}_{tv} \\ \mathbf{n}_{tw} \end{bmatrix}$$

Target Motion Model

- Model 3. Local Planar Velocity. 这个模型假设速度都发生在一个平面上，这样的话，一方面，速度在Target系里是恒定的；另一方面，角速度只有yaw这个分量。Model 3其实就是Model 2限制角速度只能在yaw上变化。这个模型适用于地面车辆。

$$\mathbf{x}_T^{(3)} = \begin{bmatrix} {}^T_G \bar{\mathbf{q}}^\top & \omega_z & {}^G \mathbf{p}_T^\top & v_x & v_y \end{bmatrix}^\top$$

$${}^T_G \dot{\bar{\mathbf{q}}} = \frac{1}{2} \boldsymbol{\Omega} \left(\begin{bmatrix} n_{\omega x} \\ n_{\omega y} \\ \omega_z \end{bmatrix} \right) {}^T_G \bar{\mathbf{q}}, \quad {}^G \dot{\mathbf{p}}_T = {}^G_T \mathbf{R} \begin{bmatrix} v_x \\ v_y \\ n_{vz} \end{bmatrix}$$

where

$$\mathbf{L} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}, \mathbf{K} = \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 1 \end{bmatrix}, \mathbf{J} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix},$$

and \mathbf{e}_i is the 3×1 unit vector in the i -th axis direction.

$$\dot{v}_x = n_{vx}, \quad \dot{v}_y = n_{vy}, \quad \dot{\omega}_z = n_{\omega z}$$

$$\begin{bmatrix} {}^T \delta \boldsymbol{\theta}_G \\ \delta \omega_z \\ {}^G \delta \mathbf{p}_T \\ \delta v_x \\ \delta v_y \end{bmatrix} = \begin{bmatrix} -[{}^T \hat{\boldsymbol{\omega}}] & \mathbf{e}_3 & \mathbf{0}_3 & \mathbf{0}_{3 \times 2} \\ \mathbf{0}_3 & \mathbf{0}_{3 \times 1} & \mathbf{0}_3 & \mathbf{0}_{3 \times 2} \\ -{}^G_T \hat{\mathbf{R}} [{}^T \hat{\mathbf{v}}_T] & \mathbf{0}_{3 \times 2} & \mathbf{0}_3 & {}^G_T \hat{\mathbf{R}} [\mathbf{e}_1 \quad \mathbf{e}_2] \\ \mathbf{0}_{2 \times 3} & \mathbf{0}_{2 \times 1} & \mathbf{0}_{2 \times 3} & \mathbf{0}_{2 \times 2} \end{bmatrix} \begin{bmatrix} {}^T \delta \boldsymbol{\theta}_G \\ \delta \omega_z \\ {}^G \delta \mathbf{p}_T \\ \delta v_x \\ \delta v_y \end{bmatrix} + \begin{bmatrix} \mathbf{0}_3 & \mathbf{L} \\ \mathbf{0}_{1 \times 3} & \mathbf{e}_3^\top \\ {}^G_T \hat{\mathbf{R}} \mathbf{K} & \mathbf{0}_3 \\ \mathbf{J} & \mathbf{0}_{2 \times 3} \end{bmatrix} \begin{bmatrix} n_{vx} \\ n_{vy} \\ n_{vz} \\ n_{\omega x} \\ n_{\omega y} \\ n_{\omega z} \end{bmatrix}$$

能观性分析

- 这块我还是零基础，没有看懂。

附录——Target Motion Model的error-state推导

• Model 1.

推公式时,不要急着把一个整体展开.

true: $\frac{1}{2}\omega_t \otimes q_t = \dot{q}_t$ error-state: $q_t = s q \otimes q. = \begin{bmatrix} 1 \\ s\theta \end{bmatrix} \otimes q$

nominal: $\frac{1}{2}\omega \otimes q = \dot{q}$ $\omega_t = \omega + \delta\omega$

则有: $\frac{1}{2}\omega_t \otimes q_t = \dot{q}_t = s\dot{q} \otimes q + s q \otimes \dot{q}$

$\frac{1}{2}\omega_t \otimes s q \otimes q = s\dot{q} \otimes q + s q \otimes \frac{1}{2}\omega \otimes q$

同时消去 q

$\Rightarrow 2s\dot{q} = \omega_t \otimes s q - s q \otimes \omega$

$= ([\omega_t]_R - [\omega]_L) s q$

$= \begin{bmatrix} 0 & -(\omega_t - \omega)^T \\ (\omega_t - \omega) - [\omega_t + \omega]_x \end{bmatrix} s q$

所以, $2 \begin{bmatrix} 0 \\ \frac{1}{2}s\dot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & -(\omega_t - \omega)^T \\ (\omega_t - \omega) - [\omega_t + \omega]_x \end{bmatrix} \begin{bmatrix} 1 \\ \frac{1}{2}s\theta \end{bmatrix}$

$\Rightarrow \dot{s}\theta = -\frac{1}{2}[\omega_t + \omega]_x s\theta + (\omega_t - \omega)$

$= -[\omega]_x s\theta + \delta\omega + O(\|s\theta\|^2)$

Model 2.

true: $\dot{P}_t = R_t V_t$ error-state: $P_t = P + \delta P$

nominal: $\dot{P}_t = R V$

用李代数推:

$R_t V_t = \dot{P}_t = (P + \delta P) = \dot{P} + \dot{\delta P} = R V + \dot{\delta P}$

$\Rightarrow \dot{\delta P} = R_t V_t - R V = R \cdot \text{Exp}([s\theta]_x) V_t - R V$

因为 R 表示 R_{GT} , 所以应该用右乘扰动

$= R(1 + [s\theta]_x + \frac{1}{2}[s\theta]_x^2 + \dots)(V + \delta V) - R V$

$= R[s\theta]_x V + R\delta V + O(\|s\theta\|^2)$

$= \frac{1}{2}([1 + [s\theta]_x + [s\theta]_x^2 + \dots]) R(V + \delta V) - R V$

$= \frac{1}{2}([s\theta]_x R V + R\delta V + O(\|s\theta\|^2))$

$= \frac{1}{2}([R V]_x s\theta + R\delta V)$

$= -R[V]_x s\theta + R\delta V$

用四元数推不出来...

$\dot{\delta P} = s q \otimes q \otimes (V + \delta V) - q \otimes V$

$= s q \otimes q \otimes V - q \otimes V + s q \otimes q \otimes \delta V$

$= \begin{bmatrix} 1 \\ \frac{1}{2}s\theta \end{bmatrix} \otimes q \otimes V - \begin{bmatrix} 1 \\ 0 \end{bmatrix} \otimes q \otimes V + s q \otimes q \otimes \delta V$

$= \underbrace{\begin{bmatrix} 0 \\ \frac{1}{2}s\theta \end{bmatrix} \otimes q \otimes V}_{\text{pure}} + s q \otimes q \otimes \delta V$

Model 3.

Model 3. 就是 Model 2 的变种.

把 Model 2 的 ω 特殊化为 $\begin{bmatrix} n_{wx} \\ n_{wy} \\ \omega_z \end{bmatrix}$

只需在 error-state 中, 注意一下

$\omega_t = \omega + \delta\omega$

其中, nominal 值 $\omega = \begin{bmatrix} 0 \\ 0 \\ \omega_z \end{bmatrix}$ 这两维就是高斯白噪声

$\delta\omega = \begin{bmatrix} n_{wx} \\ n_{wy} \\ \delta\omega_z \end{bmatrix}$