

Controller Design based on State Space Model

Controllability

- A system is said to be *controllable* at time t_0 if it is possible by means of an ***unconstrained control vector*** to transfer the system from any initial state x_0 to any other state *in a finite interval of time*
- Controllability depends upon the system matrix A and the control influence matrix B

Condition for Controllability: (single input case)

System: $\dot{X} = AX + Bu$

Solution:
$$X(t) = e^{At} X(0) + \int_0^t e^{A(t-\tau)} Bu(\tau) d\tau$$

Assuming $X(t_1) = 0$,

$$0 = e^{At_1} X(0) + \int_0^{t_1} e^{A(t_1-\tau)} Bu(\tau) d\tau$$

$$X(0) = - \int_0^{t_1} e^{-A\tau} Bu(\tau) d\tau$$

Condition for Controllability: (single input case)

$$e^{-A\tau} = \sum_{k=0}^{n-1} \alpha_k(\tau) A^k \quad (\text{Sylvester's formula})$$

$$\begin{aligned} X(0) &= -\int_0^{t_1} e^{-A\tau} B u(\tau) d\tau = -\sum_{k=0}^{n-1} A^k B \int_0^{t_1} \alpha_k(\tau) u(\tau) d\tau \\ &= -\sum_{k=0}^{n-1} A^k B \beta_k \quad \text{where} \quad \beta_k \triangleq \int_0^{t_1} \alpha_k(\tau) u(\tau) d\tau \\ &= -\begin{bmatrix} B & AB & \cdots & A^{n-1}B \end{bmatrix} \begin{bmatrix} \beta_0 & \beta_1 & \cdots & \beta_{n-1} \end{bmatrix}^T \end{aligned}$$

This system should have a non-trivial solution for $\begin{bmatrix} \beta_0 & \beta_1 & \cdots & \beta_{n-1} \end{bmatrix}^T$

Condition for Controllability (multiple input case)

The result obtained for single input case can be easily extended for m-dimensional control input U.

System: $\dot{X} = AX + BU$

$$\begin{aligned} X(0) &= - \int_0^{t_1} e^{-A\tau} BU(\tau) d\tau \\ &= [B \ AB \ \dots \ A^{n-1}B][\beta_0 \ \beta_1 \ \dots \ \beta_{n-1}]^T \end{aligned}$$

Condition for Controllability (multiple input case)

$$X(0) = [B \ AB \ \dots \ A^{n-1}B][\beta_0 \ \beta_1 \ \dots \ \beta_{n-1}]^T$$

For completely state controllability, this equation must be satisfied. This requires the rank of

$$C_B = [B \ AB \ \dots \ A^{n-1}B]$$

Should be n .

Controllability

Result: If the rank of $C_B \triangleq \begin{bmatrix} B & AB & \cdots & A^{n-1}B \end{bmatrix}$ is n ,
then the system is controllable.

Example:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 2 \\ 1 \end{bmatrix} u$$

$$C_B = \left[\begin{bmatrix} 2 \\ 1 \end{bmatrix} \quad \begin{bmatrix} -1 & 0 \\ 0 & -2 \end{bmatrix} \begin{bmatrix} 2 \\ 1 \end{bmatrix} \right] = \begin{bmatrix} 2 & -2 \\ 1 & -2 \end{bmatrix}$$

$\text{rank}(C_B) = 2 \quad \therefore$ The system is controllable.

Output Controllability

Result: $\dot{X} = AX + BU$
 $Y = CX + DU$

$$X \in \mathbb{R}^n, \quad U \in \mathbb{R}^m, \quad Y \in \mathbb{R}^p$$

If the rank of $C_B \triangleq \begin{bmatrix} CB & CAB & \cdots & CA^{n-1}B & D \end{bmatrix}$ is p ,
then the system is output controllable.

Note: The presence of DU term in the output equation
always helps to establish output controllability.

Observability

- A system is said to be *observable* at time t_0 if, with the system in state $X(t_0)$, it is possible to determine this state from the observation of the output over a finite interval of time
- Observability depends upon the system matrix A and the output matrix C

Condition for Observability

System: $\dot{X} = AX$
 $Y = CX$

Output: $Y(t) = Ce^{At}X(0)$

Sylvester Formula: $e^{At} = \sum_{k=0}^{n-1} \alpha_k(t) A^k$

So, $Y(t) = \sum_{k=0}^{n-1} \alpha_k(t) CA^k X(0)$

Condition for Observability

Expanding,

$$Y(t) = \alpha_0(t)CX(0) + \alpha_1(t)CA X(0) + \cdots + \alpha_{n-1}(t)CA^{n-1}X(0)$$

If the system is completely observable, then given $Y(t)$ over a time interval $0 \leq t \leq t_1$, $X(0)$ should be determined from above equation. This requires the following $[np \times n]$ matrix,

$$O_B = \begin{bmatrix} C \\ CA \\ \vdots \\ CA^{n-1} \end{bmatrix} = [C^T \quad A^T C^T \quad \dots \quad (A^T)^{n-1} C^T]$$

must have rank n

Observability

Result: If the rank of $O_B \triangleq \begin{bmatrix} C^T & A^T C^T & \cdots & (A^T)^{n-1} C^T \end{bmatrix}$ is n ,
then the system is observable.

Example:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 0 & -2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 2 \\ 1 \end{bmatrix} u \quad y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$O_B = \begin{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} & \begin{bmatrix} -1 & 0 \\ 0 & -2 \end{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} \end{bmatrix} = \begin{bmatrix} 1 & -1 \\ 0 & 0 \end{bmatrix}$$

$\text{rank}(O_B) = 1 \neq 2 \quad \therefore$ The system is NOT observable.

Discuss State controllability of $\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -3 & 1 \\ -2 & 1.5 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 4 \end{bmatrix} u$

Discuss State controllability of $\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -3 & 1 \\ -2 & 1.5 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 4 \end{bmatrix} u$

So, $A = \begin{bmatrix} -3 & 1 \\ -2 & 1.5 \end{bmatrix}$; $B = \begin{bmatrix} 1 \\ 4 \end{bmatrix}$;

$M = M = [B \quad AB] = \begin{bmatrix} 1 & 1 \\ 4 & 4 \end{bmatrix}$. So $\text{rank}(M) = 1$

The system is not controllable. Why?

Discuss State controllability of $\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -3 & 1 \\ -2 & 1.5 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 4 \end{bmatrix} u$

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The system is not controllable. Why?

$$\frac{X_1(s)}{U(s)} = \frac{s + 2.5}{(s + 2.5)(s - 1)}$$

So, Pole-Zero cancellation occurs.

Discuss controllability & observability of

$$\begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -0.4 & -1.3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \quad y = [0.8 \quad 1] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Discuss controllability & observability of

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -0.4 & -1.3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \quad y = [0.8 \quad 1] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

So, $A = \begin{bmatrix} 0 & 1 \\ -0.4 & -1.3 \end{bmatrix}$; $B = \begin{bmatrix} 0 \\ 1 \end{bmatrix}$;

$M = [B \quad AB] = \begin{bmatrix} 0 & 1 \\ 1 & -1.3 \end{bmatrix}$. So $\text{rank}(M) = 2$ The system is controllable.

$N = [C^T \quad A^T C^T] = \begin{bmatrix} 0.8 & -0.4 \\ 1 & -0.5 \end{bmatrix}$. So $\text{rank}(N) = 1$ The system is not observable

If we write the system in observable canonical form then,

The system is observable but not controllable. Why?

Discuss controllability & observability of

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -0.4 & -1.3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u \quad y = [0.8 \quad 1] \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$\text{So, } A = \begin{bmatrix} 0 & 1 \\ -0.4 & -1.3 \end{bmatrix}; \quad B = \begin{bmatrix} 0 \\ 1 \end{bmatrix};$$

$$M = [B \quad AB] = \begin{bmatrix} 0 & 1 \\ 1 & -1.3 \end{bmatrix}. \quad \text{So rank}(M) = 2 \quad \text{The system is controllable.}$$

$$N = [C^T \quad A^T C^T] = \begin{bmatrix} 0.8 & -0.4 \\ 1 & -0.5 \end{bmatrix}. \quad \text{So rank}(N) = 1 \quad \text{The system is not observable}$$

If we write the system in observable canonical form then,

The system is observable but not controllable. Why?

$$\frac{X_1(s)}{U(s)} = \frac{s + 0.8}{(s + 0.8)(s + 0.5)}$$

So, Pole-Zero cancellation occurs.

Controllability and Observability in Transfer Function Domain

- The system is both controllable and observable if there is no Pole-Zero cancellation.
- **Note:** The cancelled pole-zero pair suppresses part of the information about the system

Principle of Duality

$$\begin{aligned} \text{System } \mathbf{S}_1: \quad \dot{X} &= AX + BU & C_B &= [B \quad AB \quad A^2B \quad \dots \quad A^{n-1}B] \\ Y_1 &= CX & O_B &= [C^T \quad A^T C^T \quad A^{T^2} C^T \quad \dots \quad A^{T^{n-1}} C^T] \end{aligned}$$

$$\begin{aligned} \text{System } \mathbf{S}_2: \quad \dot{Z} &= A^T Z + C^T V & C_B &= [C^T \quad A^T C^T \quad A^{T^2} C^T \quad \dots \quad A^{T^{n-1}} C^T] \\ Y_2 &= B^T Z & O_B &= [B \quad AB \quad A^2B \quad \dots \quad A^{n-1}B] \end{aligned}$$

The principle of duality states that the system \mathbf{S}_1 is controllable if and only if system \mathbf{S}_2 is observable; and vice-versa!

Hence, the problem of observer design for a system is actually a problem of control design for its dual system.

Stabilizability and Detectability

- Stabilizable system: Uncontrollable system in which uncontrollable part is stable
- Detectable system: Unobservable system in which the unobservable subsystem is stable

State Space Controller Design

System :

$$\begin{aligned}\dot{\mathbf{X}} &= \mathbf{A}\mathbf{X} + \mathbf{B}\mathbf{U} & \mathbf{X} \in \mathbb{R}^n, \mathbf{U} \in \mathbb{R}^m \\ \mathbf{Y} &= \mathbf{C}\mathbf{X} + \mathbf{D}\mathbf{U} & \mathbf{Y} \in \mathbb{R}^p\end{aligned}$$

Controllability:

State: Rank of $\mathcal{C}_B = [B \ AB \ \dots \ A^{n-1}B]$ is n

Output: Rank of $\mathcal{C}_B = [CB \ CAB \ \dots \ CA^{n-1}B \ D]$ is p

Observability:

Rank of $\mathcal{O}_B = [C^T \ A^T C^T \ \dots \ A^{T(n-1)} C^T]$ is n

State Feedback Controller

System : $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$

$$y = \mathbf{C}\mathbf{x} + Du$$

where \mathbf{x} = state vector (n -vector)

y = output signal (scalar)

u = control signal (scalar)

$\mathbf{A} = n \times n$ constant matrix

$\mathbf{B} = n \times 1$ constant matrix

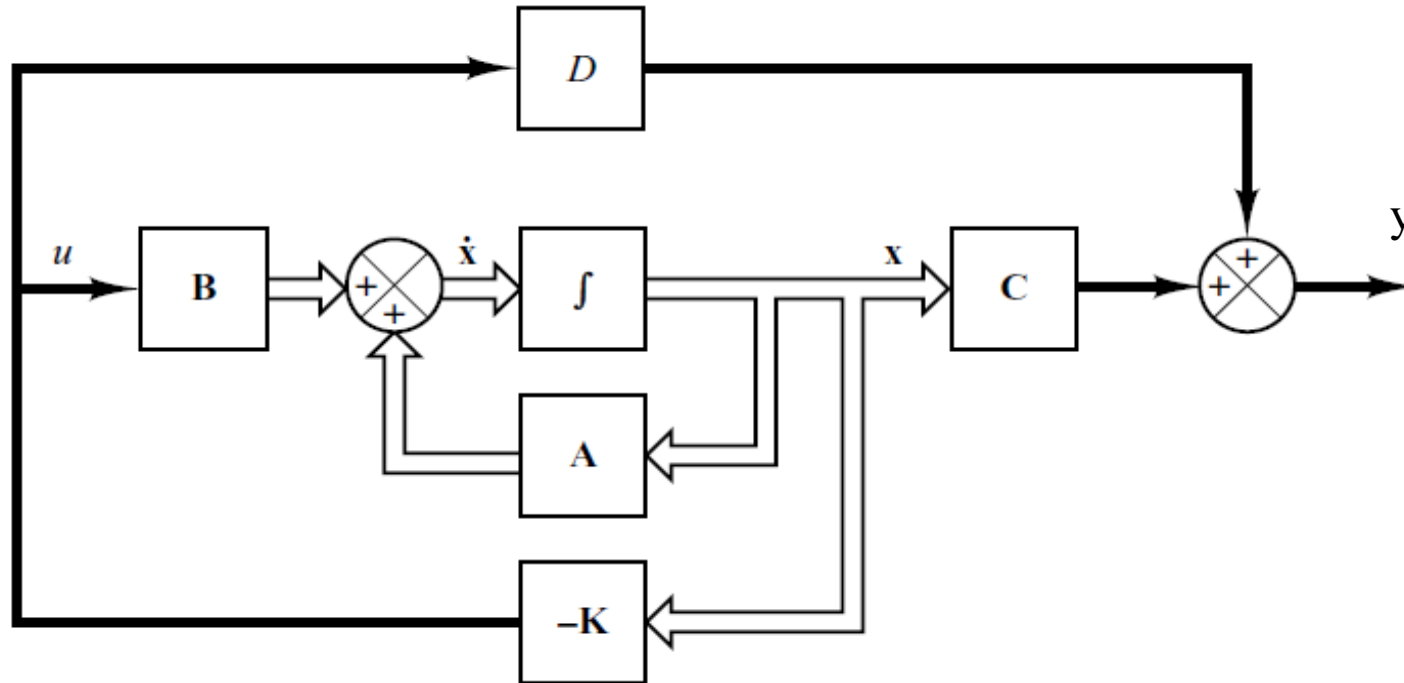
$\mathbf{C} = 1 \times n$ constant matrix

D = constant (scalar)

Control signal : $u = -\mathbf{K}\mathbf{x}$ where, $\mathbf{K} = 1 \times n$ matrix

This means u is determined from instantaneous state.

Block Diagram



- This scheme is called state feedback and \mathbf{K} matrix is called state feedback gain matrix.
- Non-zero output will be returned to zero reference input because of the state feedback scheme

Closed loop system

- System State equation: $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u$
- State feedback Control: $u = -\mathbf{K}\mathbf{x}$
- Closed loop system : $\dot{\mathbf{x}}(t) = (\mathbf{A} - \mathbf{B}\mathbf{K})\mathbf{x}(t)$
- Solution is : $\mathbf{x}(t) = e^{(\mathbf{A} - \mathbf{B}\mathbf{K})t}\mathbf{x}(0)$

Where $\mathbf{x}(0)$ is the initial state

- eigenvalues of $\mathbf{A} - \mathbf{B}\mathbf{K}$ are called regulator poles
- Placement of regulator poles in left half of s-plane ensure $\mathbf{x}(t)$ approaches $\mathbf{0}$ as t tending to ∞
- Problem of placing regulator poles at desired location is called Pole Placement Problem

Necessary & sufficient condition

Arbitrary pole placement of a given system is possible if and only if the system is completely state controllable

Proof:

Suppose the system is not fully state controllable i.e,

$$\text{rank } [\mathbf{P}] = \text{rank } [\mathbf{B} \mid \mathbf{AB} \mid \dots \mid \mathbf{A}^{n-1}\mathbf{B}] = q < n$$

Let us *write*

$$\mathbf{P} = [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \dots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \mathbf{v}_{q+2} \mid \dots \mid \mathbf{v}_n] \text{ of rank } n.$$

Where \mathbf{f}_i ($i = 1, q$) are q linearly independent column vectors and \mathbf{v}_j ($j = q+1, n$) are $n-q$ additional chosen vectors to make $\text{rank } [\mathbf{P}] = n$.

Necessary & sufficient condition

- By using \mathbf{P} as transformation matrix for \mathbf{A} and \mathbf{B} ,

- $\mathbf{A} \mathbf{P} = \mathbf{P} \hat{\mathbf{A}}$ or $[\mathbf{A}\mathbf{f}_1 \mid \mathbf{A}\mathbf{f}_2 \mid \cdots \mid \mathbf{A}\mathbf{f}_q \mid \mathbf{A}\mathbf{v}_{q+1} \mid \cdots \mid \mathbf{A}\mathbf{v}_n]$
 $= [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \cdots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \cdots \mid \mathbf{v}_n] \hat{\mathbf{A}}$

- Using Cayley-Hamilton's theorem to express $\mathbf{A}\mathbf{f}_1, \mathbf{A}\mathbf{f}_2, \dots, \mathbf{A}\mathbf{f}_q$

$$\mathbf{A}\mathbf{f}_1 = a_{11}\mathbf{f}_1 + a_{21}\mathbf{f}_2 + \cdots + a_{q1}\mathbf{f}_q$$

$$\mathbf{A}\mathbf{f}_2 = a_{12}\mathbf{f}_1 + a_{22}\mathbf{f}_2 + \cdots + a_{q2}\mathbf{f}_q$$

.

.

.

$$\mathbf{A}\mathbf{f}_q = a_{1q}\mathbf{f}_1 + a_{2q}\mathbf{f}_2 + \cdots + a_{qq}\mathbf{f}_q$$

Necessary & sufficient condition

- Hence,

$$\begin{aligned}
 & [\mathbf{A}\mathbf{f}_1 \mid \mathbf{A}\mathbf{f}_2 \mid \cdots \mid \mathbf{A}\mathbf{f}_q \mid \mathbf{A}\mathbf{v}_{q+1} \mid \cdots \mid \mathbf{A}\mathbf{v}_n] \\
 &= [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \cdots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \cdots \mid \mathbf{v}_n] \begin{bmatrix} a_{11} & \cdots & a_{1q} & a_{1q+1} & \cdots & a_{1n} \\ a_{21} & \cdots & a_{2q} & a_{2q+1} & \cdots & a_{2n} \\ \cdot & & \cdot & \cdot & & \cdot \\ \cdot & & \cdot & \cdot & & \cdot \\ \cdot & & \cdot & \cdot & & \cdot \\ a_{q1} & \cdots & a_{qq} & a_{qq+1} & \cdots & a_{qn} \\ \hline 0 & \cdots & 0 & a_{q+1q+1} & \cdots & a_{q+1n} \\ \cdot & & \cdot & \cdot & & \cdot \\ \cdot & & \cdot & \cdot & & \cdot \\ \cdot & & \cdot & \cdot & & \cdot \\ 0 & \cdots & 0 & a_{nq+1} & \cdots & a_{nn} \end{bmatrix}
 \end{aligned}$$

Necessary & sufficient condition

- Hence,

$$\begin{aligned} & [\mathbf{A}\mathbf{f}_1 \mid \mathbf{A}\mathbf{f}_2 \mid \cdots \mid \mathbf{A}\mathbf{f}_q \mid \mathbf{A}\mathbf{v}_{q+1} \mid \cdots \mid \mathbf{A}\mathbf{v}_n] \\ &= [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \cdots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \cdots \mid \mathbf{v}_n] \left[\begin{array}{c|c} \mathbf{A}_{11} & \mathbf{A}_{12} \\ \hline \mathbf{0} & \mathbf{A}_{22} \end{array} \right] \end{aligned}$$

- Thus,

$$\mathbf{A}\mathbf{P} = \mathbf{P} \left[\begin{array}{c|c} \mathbf{A}_{11} & \mathbf{A}_{12} \\ \hline \mathbf{0} & \mathbf{A}_{22} \end{array} \right]$$

$$\mathbf{P}^{-1}\mathbf{A}\mathbf{P} = \hat{\mathbf{A}} = \left[\begin{array}{c|c} \mathbf{A}_{11} & \mathbf{A}_{12} \\ \hline \mathbf{0} & \mathbf{A}_{22} \end{array} \right]$$

Necessary & sufficient condition

- Similarly, $\mathbf{B} = \mathbf{P}\hat{\mathbf{B}}$

$$\mathbf{B} = [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \cdots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \cdots \mid \mathbf{v}_n] \hat{\mathbf{B}}$$

- So,

$$b_{11}\mathbf{f}_1 + b_{21}\mathbf{f}_2 + \cdots + b_{q1}\mathbf{f}_q = [\mathbf{f}_1 \mid \mathbf{f}_2 \mid \cdots \mid \mathbf{f}_q \mid \mathbf{v}_{q+1} \mid \cdots \mid \mathbf{v}_n] \begin{bmatrix} b_{11} \\ b_{21} \\ \cdot \\ \cdot \\ \cdot \\ b_{q1} \\ 0 \\ \cdot \\ \cdot \\ \cdot \\ 0 \end{bmatrix}$$

Where \mathbf{B} can be written in terms of independent q column vectors

$$\mathbf{B} = b_{11}\mathbf{f}_1 + b_{21}\mathbf{f}_2 + \cdots + b_{q1}\mathbf{f}_q$$

$$\hat{\mathbf{B}} = \begin{bmatrix} \mathbf{B}_{11} \\ \mathbf{0} \end{bmatrix}$$

Necessary & sufficient condition

- Now define, $\hat{\mathbf{K}} = \mathbf{K}\mathbf{P} = [\mathbf{k}_1 \quad \vdots \quad \mathbf{k}_2]$
- Then we have,

$$\begin{aligned} |s\mathbf{I} - \mathbf{A} + \mathbf{BK}| &= |\mathbf{P}^{-1}(s\mathbf{I} - \mathbf{A} + \mathbf{BK})\mathbf{P}| \\ &= |s\mathbf{I} - \mathbf{P}^{-1}\mathbf{A}\mathbf{P} + \mathbf{P}^{-1}\mathbf{B}\mathbf{K}\mathbf{P}| \\ &= |s\mathbf{I} - \hat{\mathbf{A}} + \hat{\mathbf{B}}\hat{\mathbf{K}}| \\ &= \left| s\mathbf{I} - \left[\begin{array}{c|c} \mathbf{A}_{11} & \mathbf{A}_{12} \\ \hline \mathbf{0} & \mathbf{A}_{22} \end{array} \right] + \left[\begin{array}{c} \mathbf{B}_{11} \\ \hline \mathbf{0} \end{array} \right] [\mathbf{k}_1 \quad \vdots \quad \mathbf{k}_2] \right| \\ &= \left| \begin{array}{cc} s\mathbf{I}_q - \mathbf{A}_{11} + \mathbf{B}_{11}\mathbf{k}_1 & -\mathbf{A}_{12} + \mathbf{B}_{11}\mathbf{k}_2 \\ \mathbf{0} & s\mathbf{I}_{n-q} - \mathbf{A}_{22} \end{array} \right| \\ &= |s\mathbf{I}_q - \mathbf{A}_{11} + \mathbf{B}_{11}\mathbf{k}_1| \cdot |s\mathbf{I}_{n-q} - \mathbf{A}_{22}| = 0 \end{aligned}$$

- Notice that the eigenvalues of \mathbf{A}_{22} do not depend on \mathbf{K} . Thus if the system is not completely state controllable, then there are eigenvalues of \mathbf{A} that cannot be arbitrarily placed.