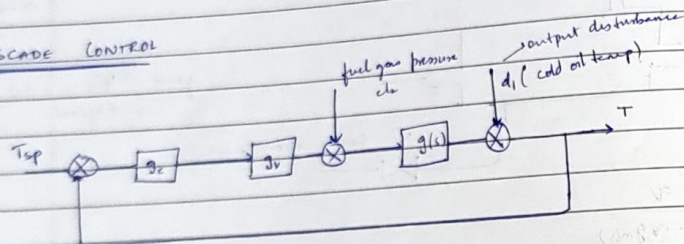



Traditional advanced control strategies

- ① High level application of PID
  - ⊙ Time delay compensator
  - ⊙ Cascade control
  - ⊙ split range control — manipulate multiple variables using 1 controller
- ② Feedforward and ratio control

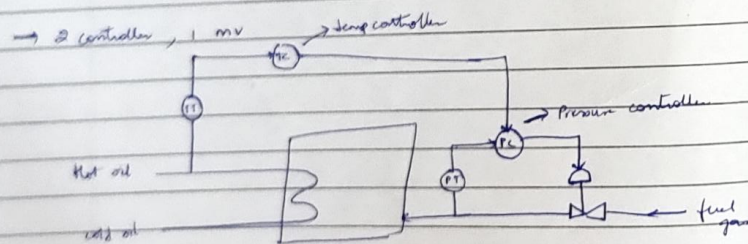
## CASCADE CONTROL



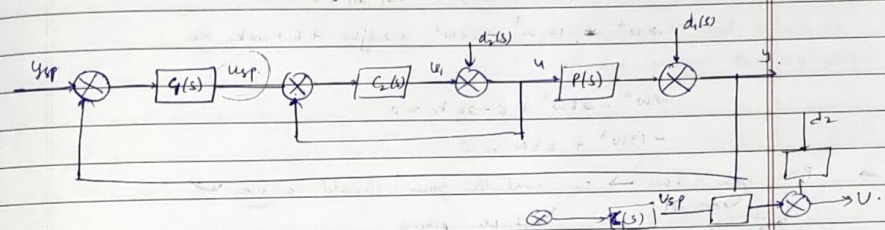
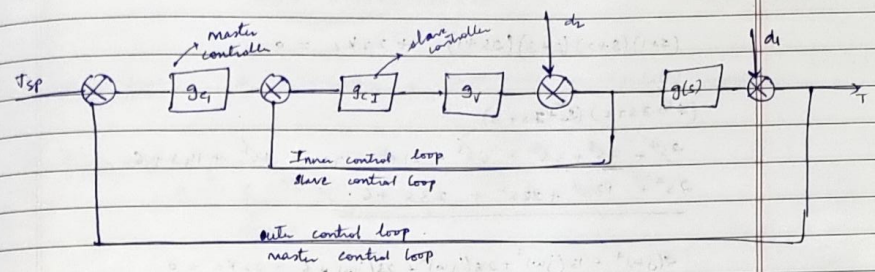

  
 furnace  
 (feedback control)  
 (normal temp)

→ regulatory control strategy cannot take care of input disturbances.  
(fuel gas pressure)

↙ This can be ~~the~~ ~~self~~ controlled  
using another controller at the input



cooling temp controller using pressure controller  
(TSPC cascade controller)



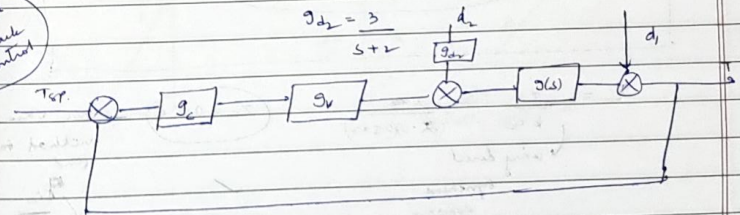
$$u_y(s) = \frac{c_z(s) V(s)}{1 + c_z(s) V(s)} u_{sp}(s) + \frac{1}{1 + c_z(s) V(s)} d_z(s) \checkmark$$

$$y(s) = \frac{y_{sp}(s) \cdot G_1(s) \cdot G_2(s) \cdot V(s)}{1 + G_2 V} + \frac{d_2 \cdot \frac{1}{1 + G_2 V}}{1 + \frac{G_1 G_2 V P}{1 + G_2 V}} + \frac{d_1}{1 + G_2 V P}$$

$$c) \quad y(s) = \frac{10}{(s+1)(s+2)(s+3)} u(s)$$

$$g(s) = \frac{10}{(s+1)(s+2)(s+3)}$$

$$g_v = \frac{2.2}{2s+1}$$



$$1 + k_{eq} g_v g = 0$$

$$1 + k_c \left( \frac{2.2}{s+1} \right) \left( \frac{10}{(s+1)(s+2)(s+3)} \right) = 0 \quad \underline{\underline{K_{crit} = ?}}$$

$$s = j\omega$$

$$(2s+1)(s+1)(s+2)(s+3) + K_c (22) = 0$$



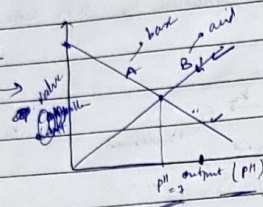




## Equation

→ PI neutralization reactor

PI, Controller output

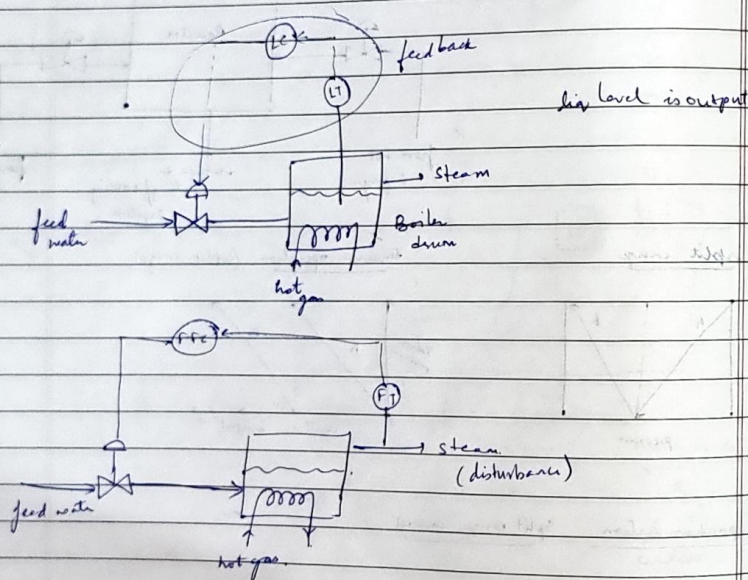


## Feedforward AND RATIO CONTROL

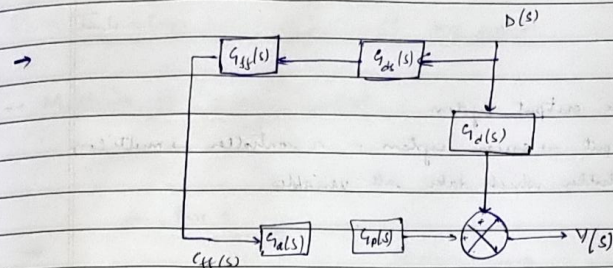
Feedback / Feedforward

a. Feed back control

Feedforward controller

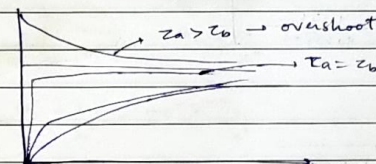


Only FFC strategy is never applied in an industry because it cannot solve servo problems.



$$Y(s) = D(s) \frac{G_1(s)}{1 + G_1(s)G_2(s)} + \frac{G_d(s)}{1 + G_1(s)G_2(s)} D(s) = 0$$

$$\frac{Y(s)}{U(s)} = \frac{G_2(s)}{Z_1 s + 1}$$



→ Combined ff & fb control

If we use a pure FFC strategy, there is no stability issue. If we have combination of FF & FB → stability will come into picture. Adding FFC will not hamper the stability of the small control loop.

## Ratio control

$$R = \frac{u}{d} \rightarrow mv$$

$d \rightarrow$  disturbance



## Multivariable control

- multiple input, multiple output system
- ①  $n$  input,  $n$  output → square system →  $n$  controller is multi-loop
- (or) single controller which take all variables

## Double tank system

$K_m$  = measurement gain

- a → find poles & zeros for a multivariable system

$$G(s) = \begin{bmatrix} \frac{4.153}{62.36s+1} & \frac{3.036}{(62.36s+1)(22.76s+1)} \\ \frac{1.75}{(90.63s+1)(20.07s+1)} & \frac{4.534}{90.63s+1} \end{bmatrix}$$

$$G(s) = \begin{bmatrix} a & b \\ c & d \end{bmatrix} \cdot \frac{1}{(62.36s+1)(22.76s+1)(90.63s+1)(20.07s+1)}$$

will give poles

→ poles of  $G^{-1}(s)$  are zeros of  $G(s)$

$$G^{-1}(s) = \frac{\text{Adj}(G(s))}{\det(G(s))}$$

$$\det(G(s))$$

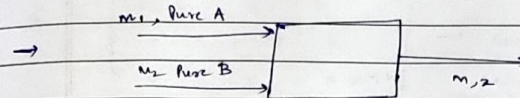
$$[\det(G(s)) = 0 \text{ will give poles of } G^{-1}(s) \text{ and zeros of } G(s)]$$

→ non-minimum phase system → designing the control strategy is difficult

Right hand pole is zero, time delay.

## Interaction analysis for multi-loop control

→ Multivariable process control



$$M = M_1 + M_2$$

$$x = \frac{M_1}{M_1 + M_2}$$

$$\frac{\partial M}{\partial M_1} \Big|_{\text{both loops open}} = 1$$

$$\frac{\partial M}{\partial M_1} \Big|_{\text{other loop is closed}} = 1$$

2 loops  
control  
x by manipulating  
 $M_1$

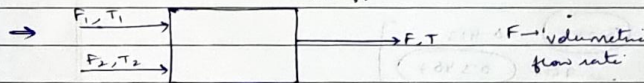
$$\rightarrow x^* = \frac{M_1}{M_1 + M_2} \quad M_2 = \frac{M_1 - M_1}{x^*}$$

$$M = M_1 + M_2 - M_1 = \frac{M_1}{x^*}$$

$$\rightarrow \frac{\partial M}{\partial M_1} \Big|_{\text{other loop closed}} = \frac{1}{x^*}$$

$$\rightarrow \lambda_H = x^*$$

$R_{HA}??$



## Analysis of RGA

$$\lambda = \frac{\Delta y}{\Delta y + \Delta y_r} = \frac{k_d}{k_d + k_r}$$

$$\lambda = 1, \rightarrow k_r = 0$$

→  $\lambda < 1$ , means  $k_r$  is finite and in the same direction to  $k_d$

→  $\lambda > 1 \rightarrow |k_d| > |k_r|$   $k_r$  is acting in opp direction to  $k_d$



$\lambda_{ii} = 1 \rightarrow v_d$   
 $k_d + k_v$   
 no interaction from other loops.

$\lambda_{ii} < 1 \rightarrow k_d T, k_v T$   
 $\lambda_{ii} = 0 \rightarrow$  assumed loop pairing is incorrect.  
 $\lambda_{ii} > 1 \rightarrow k_d T, k_v T$   
 $\lambda_{ii} < 0$

$\Rightarrow RGA$ 

	$u_1$	$u_2$	$u_3$	$u_4$
$y_1$	0.93	0.15	0.08	-0.164
$y_2$	-0.011	-0.429	0.286	1.154
$y_3$	-0.135	3.314	-0.27	-1.11
$y_4$	0.215	-2.03	0.7	1.917

  
 $\rightarrow$  avoid -ve values  
 $\rightarrow$  choose values close to 1  
 loop pairing should ~~minimize~~ min interaction

### Singular value analysis

relation between  $CN$  & linear independence  
 $\rightarrow CN \uparrow$ , independence  $\uparrow$

if  $k$  is all-conditioned  $\rightarrow$  Pairing is ill-conditioned

$\Rightarrow RGA =$ 

-2.42	3.024	0.413
1.2211	-0.7617	0.5467
2.2165	-1.2622	0.0458

$y_1 \rightarrow u_2$   
 $y_2 \rightarrow u_3$   
 $y_3 \rightarrow u_1$   
 better pairing

$y_1 \rightarrow u_2$   
 $y_2 \rightarrow u_1$  (X)  
 $y_3 \rightarrow u_3$  is very low

$y_1 \rightarrow u_2$   
 $y_2 \rightarrow u_1$   
 $y_3 \rightarrow u_3 \rightarrow$  not possible

(same variables)

$$W = \begin{bmatrix} 0.5714 & 0.376 & 0.2272 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

$\downarrow$  find mag

$$S = \begin{bmatrix} 1.618 & 0 & 0 \\ 0 & 1.143 & 0 \\ 0 & 0 & 0.0097 \end{bmatrix} \rightarrow CN \text{ is very high}$$

$$CN = \frac{1.618}{0.0097}$$

$CN$  is very high  
 unstable loop

Whatever loop pairing you consider, loop will be unstable.

$\rightarrow$  3 control loops  $\rightarrow$  system is unstable  
 design 2 control loops

$\rightarrow$  find the possible combinations

$\rightarrow$  Find the possible combinations to get the 2 control loops  
 $\rightarrow$  choose the loops where  $CN \approx 1$



# Controller design based on state space model

→ state space model

(1) state variable →

(2) Non linear SSM

control affine form  
 $\dot{x} = f(x) + g(x)u$  → state eqn  
 $y = h(x, u)$  → output eqn  
 Input is represented explicitly.

$$x = [x_1 \ x_2 \ x_3 \ \dots \ x_n]^T \quad \text{for system}$$

$$u = [u_1 \ u_2 \ \dots \ u_m]^T \quad \text{overall dynamics depends on } n$$

$$y = [y_1 \ y_2 \ \dots \ y_p]^T$$

(3) linear SSM

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

$x$  → state variable vector  
 $u$  → input vector  
 $y$  → output vector  
 as deviation form.

$A, B$  → matrices

→ (i) develop dynamic model

$$\dot{x} = f(x, u) \rightarrow (1)$$

(ii) define control signal

$$y = h(x, u) \rightarrow (2)$$

(iii) control-affine form

$$\dot{x} = f(x) + g(x)u$$

$$y = h(x, u)$$

(iv) linear SSM → linearize around  $x_s, u_s$

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

Dynamic model :  $\dot{x} = f(x, u) \rightarrow (1)$

$$y = h(x, u) \rightarrow (2)$$

Use Taylor series approx around  $(x_s, u_s)$

$$\dot{x}_1 = f_1(x_1^s, x_2^s, \dots, x_n^s, u_1^s, u_2^s, \dots, u_m^s) +$$

$$\frac{\partial f_1}{\partial x_1} (x_1 - x_1^s) + \frac{\partial f_1}{\partial x_2} (x_2 - x_2^s) + \dots + \frac{\partial f_1}{\partial x_n} (x_n - x_n^s)$$

$$+ \dots$$

$\dot{x}$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_n \end{bmatrix} = \underbrace{\begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \dots & \frac{\partial f_1}{\partial x_n} \\ \vdots & \ddots & \vdots \\ \frac{\partial f_n}{\partial x_1} & \dots & \frac{\partial f_n}{\partial x_n} \end{bmatrix}}_A \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + \underbrace{\begin{bmatrix} \frac{\partial f_1}{\partial u_1} & \dots & \frac{\partial f_1}{\partial u_m} \\ \vdots & \ddots & \vdots \\ \frac{\partial f_n}{\partial u_1} & \dots & \frac{\partial f_n}{\partial u_m} \end{bmatrix}}_B \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_m \end{bmatrix}$$

→ Transfer function model → is only for linear equation

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

take Laplace

$$sX(s) = AX(s) + BU(s)$$

$$Y(s) = CX(s) + DU(s)$$

$$(sI - A)X(s) = BU(s); \text{ i.e. } X(s) = (sI - A)^{-1}BU(s)$$

$$Y(s) = [C(sI - A)^{-1}B + D]U(s)$$

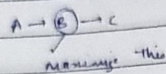
$$U = \begin{bmatrix} M \\ L \end{bmatrix}$$

$$M \rightarrow u, L \rightarrow v$$

$$Y(s) = G_p(s)M(s) + G_L(s)L(s)$$



Example: Van De Vusse reactor



Realization from TF

→ If I have TF model, can I get the SSM?

Controllable canonical form

non-rational → time delay associated with TF

$$G(s) = \frac{V(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + \dots + b_{n-1} s + b_n}{s^n + a_1 s^{n-1} + \dots + a_{n-1} s + a_n}$$

Consider rational TF if time delay represent separate mat part

→  $\frac{V(s)}{U(s)} = \frac{b_0 s^n + b_1 s^{n-1} + \dots + b_{n-1} s + b_n}{s^n + a_1 s^{n-1} + \dots + a_{n-1} s + a_n}$

Jordan canonical form

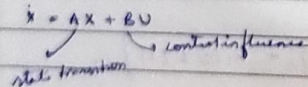
Observer canonical form

→ matrix = transpose of matrix in controllable canonical form

Designing the control system

① Controllability

At time  $t_0$ , if it is possible by means of an unconfined control vector to transfer the system from any initial state  $x_0$  to any other state in a finite interval of time (whatever input  $u$  gives, ~~and~~ system can be controlled)



system:  $\dot{X} = AX + BU$

Linear ODE

$$X(t) = e^{At} X(0) + \int_0^t e^{A(t-\tau)} B U(\tau) d\tau$$

at  $t=0$   $X(t_0) = 0$   
( $X$  is deviation variable)

$$X(0) = - \int_0^t e^{A(t-\tau)} B U(\tau) d\tau$$

$$\Rightarrow [B \quad AB \quad \dots \quad A^{n-1} B]$$

rank should be  $n$  to solve  
→ no. of states of the system

→  $P$  = no. of outputs

Observability

whether the system is observable or not

n states

$$\dot{X} = AX + BU$$

$$Y = CX + DU$$

$y = f(x_1, x_2) \rightarrow$  we will measure  $x_1, x_2$  to get the output.  
→  $n-2$  state info is not available.

Principle of duality

$$S_1: \dot{X} = AX + BU$$

$$Y_1 = CX$$

$$S_2: \dot{Z} = A^T Z + C^T V$$

$$Y_2 = B^T Z$$



## State space controller design

$$\dot{X} = AX + BU$$

$$Y = CX + DV$$

$Y = CX + DV$

check whether the states, outputs are controllable or not

→ control signal :  $U = -kx$

