

Documentation of RVortexInt

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1 Introduction

The function RVortexInt evaluates the dimensionless induced velocity of a ring vortex by integration of the Biot-Savart law:

$$\mathbf{V}(\mathbf{P}) = -\frac{\Gamma}{4\pi} \int_0^{2\pi} \frac{\mathbf{R} \otimes d\mathbf{l}}{R^3} \quad (1.1)$$

Where:

$$\mathbf{R} = (x - \bar{x})\mathbf{i} + (y - \bar{y})\mathbf{j} + z\mathbf{k} \quad (1.2)$$

Since the ring vortex equations are:

$$\bar{x} = \bar{r}\cos(\vartheta), \quad \bar{y} = \bar{r}\sin(\vartheta), \quad \bar{z} = 0 \quad (1.3)$$

The vector product is developed in this way:

$$d\mathbf{l} = \frac{d\mathbf{s}}{d\vartheta} d\vartheta = (-\bar{r}\sin\vartheta\mathbf{i} + \bar{r}\cos\vartheta\mathbf{j})d\vartheta \quad (1.4)$$

$$\mathbf{R} \otimes d\mathbf{l} = -z\bar{r}\cos\vartheta\mathbf{i} + z\bar{r}\sin\vartheta\mathbf{j} + [(x - \bar{x})\bar{r}\cos\vartheta + (y - \bar{y})\bar{r}\sin\vartheta]\mathbf{k} \quad (1.5)$$

And:

$$R^3 = [(x - \bar{x})^2 + (y - \bar{y})^2 + z^2]^{\frac{3}{2}} \quad (1.6)$$

The three induced velocities are:

$$V_x(x, y, z) = \frac{\Gamma}{4\pi} \bar{r}z \int_0^{2\pi} \frac{\cos\vartheta d\vartheta}{R^3} \quad (1.7)$$

$$V_y(x, y, z) = -\frac{\Gamma}{4\pi} \bar{r}z \int_0^{2\pi} \frac{\sin\vartheta d\vartheta}{R^3} \quad (1.8)$$

$$V_z(x, y, z) = -\frac{\Gamma}{4\pi} \bar{r} \int_0^{2\pi} \frac{[(x - \bar{x})\bar{r}\cos\vartheta + (y - \bar{y})\bar{r}\sin\vartheta] d\vartheta}{R^3} \quad (1.9)$$

The velocity component of interests is V_x calculated imposing $y = 0$ because it provide exactly the radial component:

$$V_x(x, 0, z) = \frac{\Gamma}{4\pi} \bar{r}z \int_0^{2\pi} \frac{\cos\vartheta d\vartheta}{[(x - \bar{r}\cos\vartheta)^2 + (\bar{r}\sin\vartheta)^2 + z^2]^{\frac{3}{2}}} \quad (1.10)$$

This model is based on the McCornick's Theory: The point at 3/4 of chord of the shroud is classified as a control point and of course it is the point where

the velocity of interests is evaluated. Instead the Ring Vortex is placed at 1/4 of the shroud.

$$f\left(\frac{c}{D_{1/4}}, \frac{D_{3/4}}{D_{1/4}}\right) = \frac{V_x \pi D_{1/4}}{\Gamma} = \frac{1}{4} \bar{r} z D_{1/4} \int_0^{2\pi} \frac{\cos \vartheta d\vartheta}{[(x - \bar{r} \cos \vartheta)^2 + (\bar{r} \sin \vartheta)^2 + z^2]^{\frac{3}{2}}} \quad (1.11)$$

The results that are expected are represented on the following picture for different ratio of $\frac{D_{3/4}}{D_{1/4}}$ and $\frac{C}{D_{1/4}}$. [1],[2]

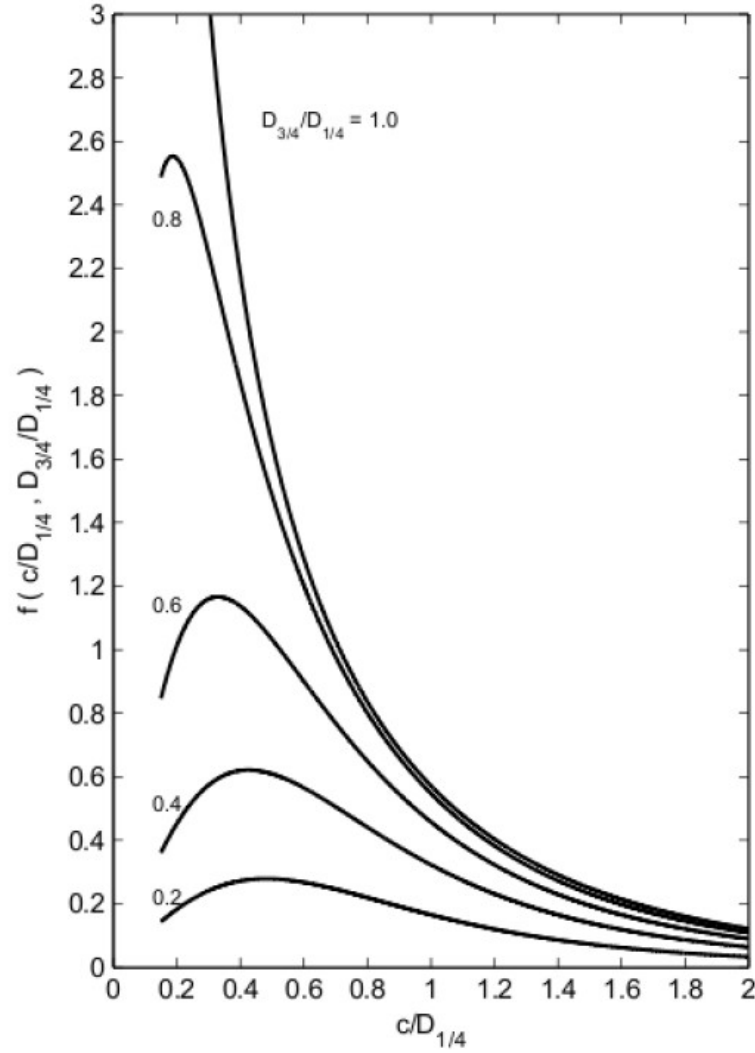


Figure 1.1: Velocity induced by vortex ring

2 Algorithm description

The user has to give as inputs in the GUI three different values: the shroud length and the diameter at the beginning and at the end of the shroud too.

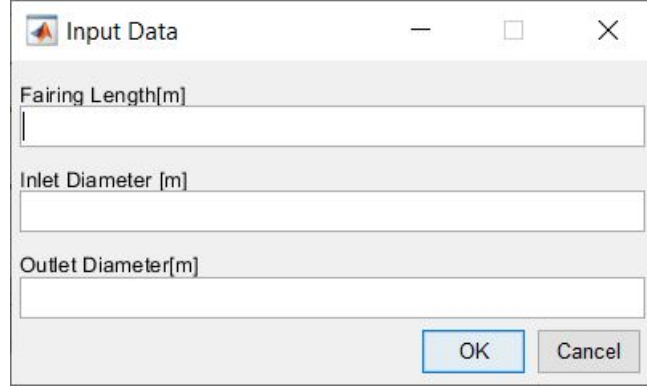


Figure 2.1: Code's GUI

The first part of the code rebuilds the shroud and it evaluates the diameters at $1/4$ and at $3/4$ of the shroud. In this way, it could correctly place the radius of the ring vortex at $1/4$ and the velocity control point at $3/4$ of the shroud length.

```
1 t=linspace(0,1,1000); % Recreating the shroud
2   for i=1:length(t)
3       xs(i)=0+t(i)*(ch);
4       ys(i)=Di/2+t(i)*(De/2-Di/2);
5   end
6
7 D14=2*ys(250); % Diameter at 1/4 of the shroud
8 D34=2*ys(750); % Diameter at 3/4 of the shroud
9 r=D14/2; % Ring Vortex radius
10 x=D34/2; % Speed Control Point
11 z=ch*(3/4); % Three quarters of the shroud
12 y=0;
```

The code performs the Biot-Savart's integrals on the three directions using anonymous functions based on the value of ϑ . The MatLab function *Integral* is used for the calculations.

```
1 % Anonymous Function
2 R1=@(t) cos(t)/(((x-r.*cos(t)).^2 +(y-r.*sin(t)).^2+z(i).^2)).^(3/2);
3 R2=@(t) sin(t)/(((x-r.*cos(t)).^2 +(y-r.*sin(t)).^2+z(i).^2)).^(3/2);
4 R3=@(t)((x-r.*cos(t)).*cos(t)+(y-r.*sin(t)).*sin(t))/((x-r.*cos(t)).^2+(y-r.*sin(t)).^2+z(i).^2)).^(3/2);
5 % Integrals
6 Ix(i)=integral(R1,0,2*pi);
7 Iy(i)=integral(R2,0,2*pi);
8 Iz(i)=integral(R3,0,2*pi);
```

The code evaluates the dimensionless components of the velocity induced by the ring vortex based on the Biot-Savart's Law at 3/4 of the shroud.

```
1 %Dimensionless;
2 fx(i)=(r*z(i)*D14/4)*Ix(i);
3 fy(i)=-(r*z(i)*D14/4)*Iy(i);
4 fz(i)=-(r*D14/4)*Iz(i);
```

The code's output is the value of f_x . It has been developed another version of the code that gives as output the curves of different Ratio of $\frac{D_3}{D_{1/4}}$ and $\frac{C}{D_{1/4}}$.

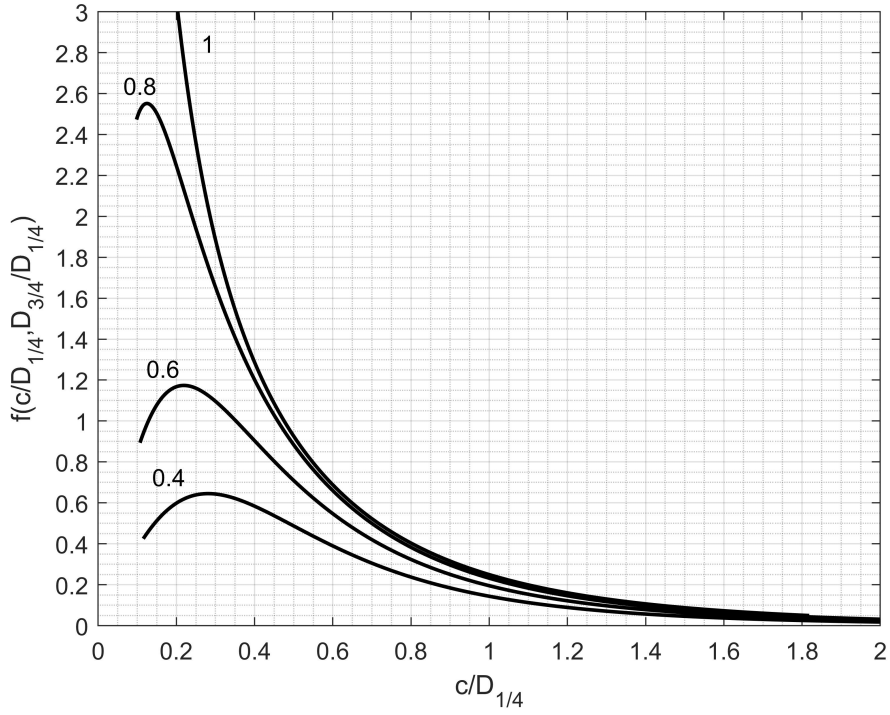


Figure 2.2: Induced speed for different ratios- MatLab

3 Warnings indicators

There is a warning when the user runs the code. According to the input data there are three different warnings pop-ups that advice the user of which shroud is being analyzed.

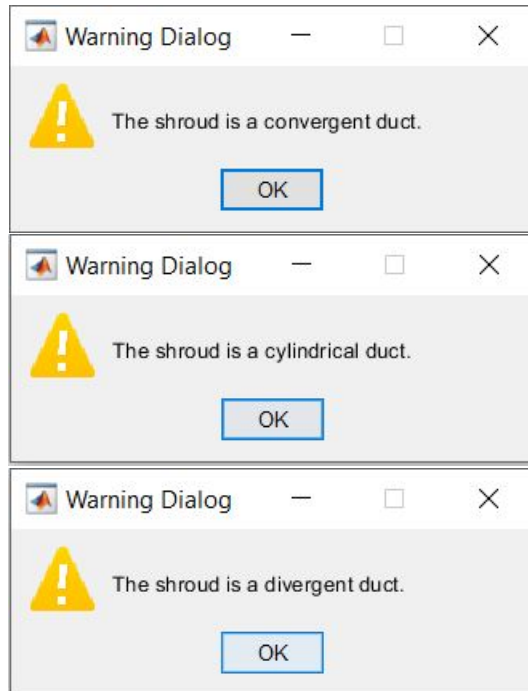


Figure 3.1: Code's warnings for shroud geometry.

4 Test Case

In order to validate the function it is done a comparison between the results obtained with the integration to the ones of reference on the [2].

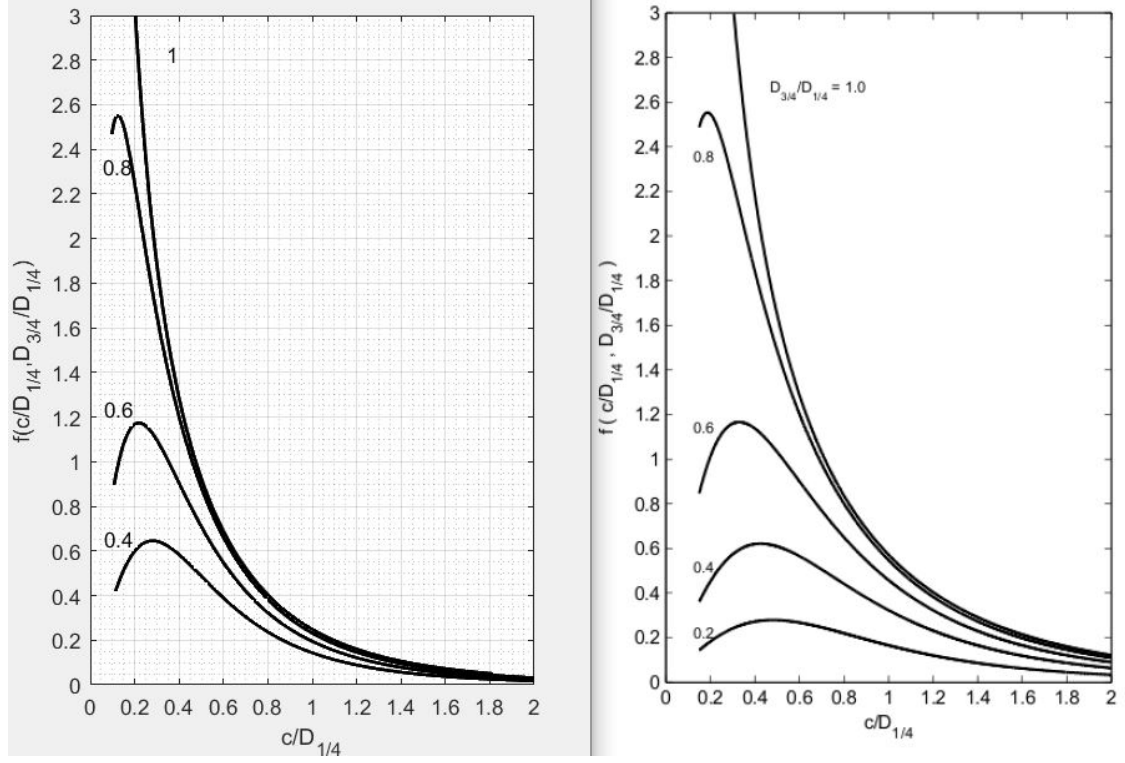


Figure 4.1: Test cases: Left: MatLab code output. Right: Reference data

There is a good overlap between the two graphics.

5 Appendix: Code

```

1 function [fx] = RVortexInt(ch,Di,De)
2 % % \RVortexInt.m
3 % \The function evaluates the dimensionless radial velocity component
4 %   of the ring vortex by means of integration of Biot-Savart law
5 % \Olino Massimiliano, Marino Giuseppe
6 % \1.0
7 %
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24 %
25 % =====
26 % Name      : RVortexInt.m
27 % Author    : Olino Massimiliano, Marino Giuseppe
28 %           : University of Naples Federico II.
29 % Version   : 1.0
30 % Date      : 01/12/2020
31 % Modified  : 20/01/2021
32 % Description: The function evaluates the dimensionless radial velocity component
33 %               of the ring vortex by means of integration of Biot-Savart law
34 % Reference  : McCormick, B.W.,(1967), Aerodynamics of V/STOL Flight, Academic Press.
35 %           : Tognaccini, R., (2020), Lezioni di AERODINAMICA DELL'ALA ROTANTE.
36 % Input     : ch (chord), Di (inlet diameter), De (exit diameter) of the shroud
37 % Output    : fx (the dimensionless radial velocity component of the ring vortex )
38 % Note      :
39 % =====
40 %
41 %Gui for the input by users.
42 title='Input Data';
43 prompt={'Shroud chord [m]','Inlet diameter [m]','Exit diameter [m]'};
44 answer=inputdlg(prompt,title,[1 55;1 55;1 55]);
45 ch=str2double(answer{1});
46 Di=str2double(answer{2});
47 De=str2double(answer{3});
48
49 % Creation of shroud geometry
50 t=linspace(0,1,1000);
51 for i=1:length(t)
52     xs(i)=0+t(i)*(ch);
53     ys(i)=Di/2+t(i)*(De/2-Di/2);
54 end
55 D14=2*ys(250);           % Diameter at 1/4 of the shroud
56 D34=2*ys(750);           % Diameter at 3/4 of the shroud
57 c_D14=ch/D14 ;           % Ratio between chord and diameter at 1/4
58 RatioD=D34/D14;         % Ratio between diameter at 3/4 and 1/4
59
60 r=(D14/2);               % flow radius
61 x=D34/2;                 %
62 z=ch*(3/4);              % Speed control point coordinates
63 y=0;                     %
64
65 % Anonymous Function

```

```

66 R1=@(t) cos(t)./(((x-r.*cos(t)).^2 +((y-r.*sin(t)).^2+z^2)).^(3/2);
67 R2=@(t) sin(t)./(((x-r.*cos(t)).^2 +(y-r.*sin(t)).^2+z^2)).^(3/2);
68 R3=@(t) ((x-r.*cos(t)).*cos(t)+(y-r.*sin(t)).*sin(t))./(((x-r.*cos(t)).^2 +(y-r.*sin(t)).^2+z^2)).^(3/2);
69
70 % Integrals
71 Ix=integral(R1,0,2*pi);
72 Iy=integral(R2,0,2*pi);
73 Iz=integral(R3,0,2*pi);
74
75 % Adimesionalisation of the velocity components
76
77 fx=(r*z*D14/4)*Ix;           % Component of interests
78 fy=-(r*z*D14/4)*Iy;
79 fz=-(r*D14/4)*Iz;
80
81 % Vector sum
82 f=sqrt(fx^2+fy^2+fz^2);
83
84 % Results
85 figure(1);
86 plot(f,f)
87 axis ([0 1 0 1]);
88 text(0.25,1,'Velocity induced by vortex ring:');
89 text(0.25,0.90,['fx= ',num2str(fx)]);
90 axis off;
91
92 % Warnings
93 if Di>De
94     warndlg('The shroud is a convergent duct.');
```

Bibliography

- [1] McCORMICK, B.W.,(1967), *Aerodynamics of V/STOL Flight*, Academic Press.
- [2] TOGNACCINI, R., (2020), *Lezioni di AERODINAMICA DELL'ALA ROTANTE*.