

High-Fidelity Computational Data of Transitional Boundary Layers for a Data-Driven Approach

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July 29th 2022

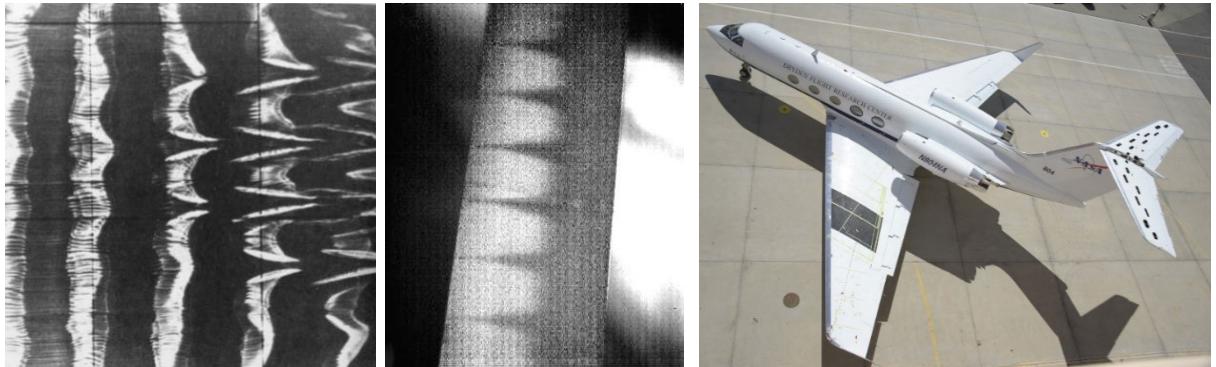
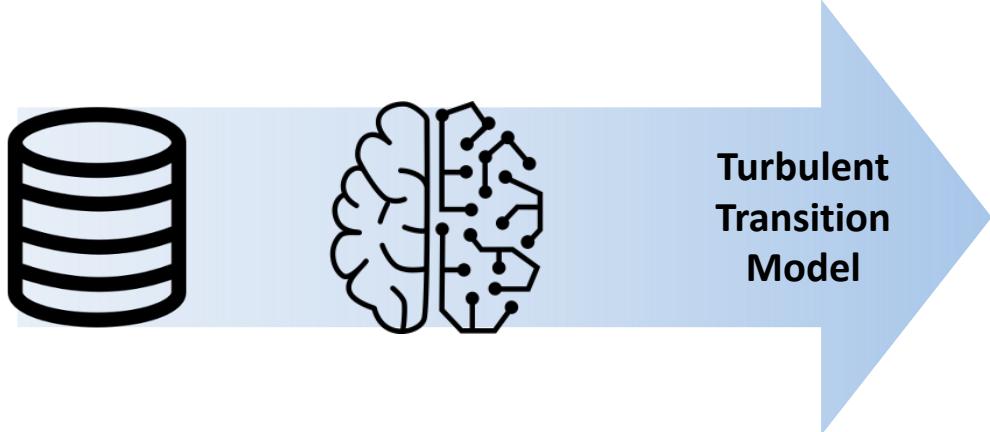
2022 Symposium on Turbulence Modeling



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Motivation on Sharing Transition Data

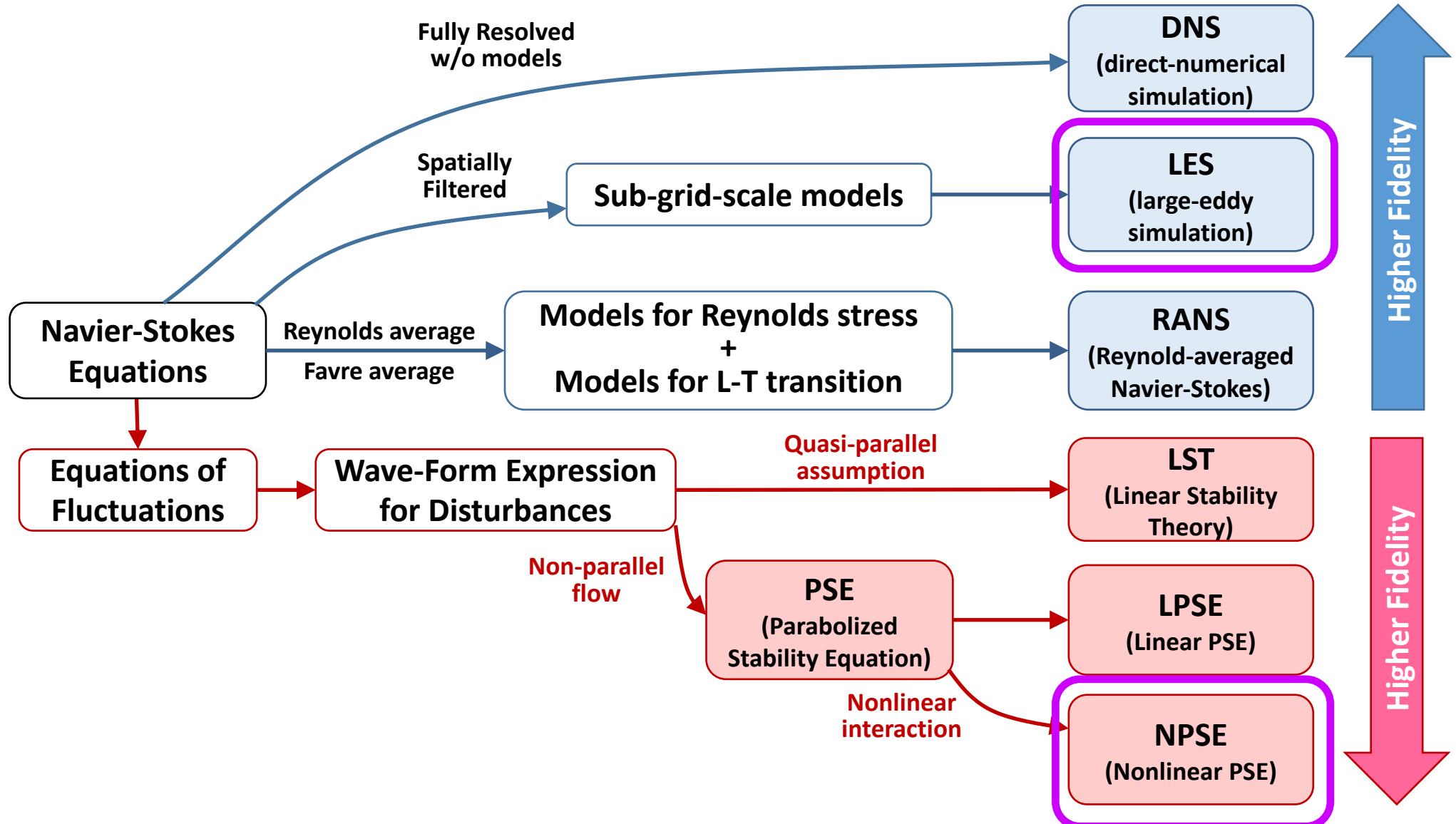
■ A new era of transition modeling with data-driven approaches



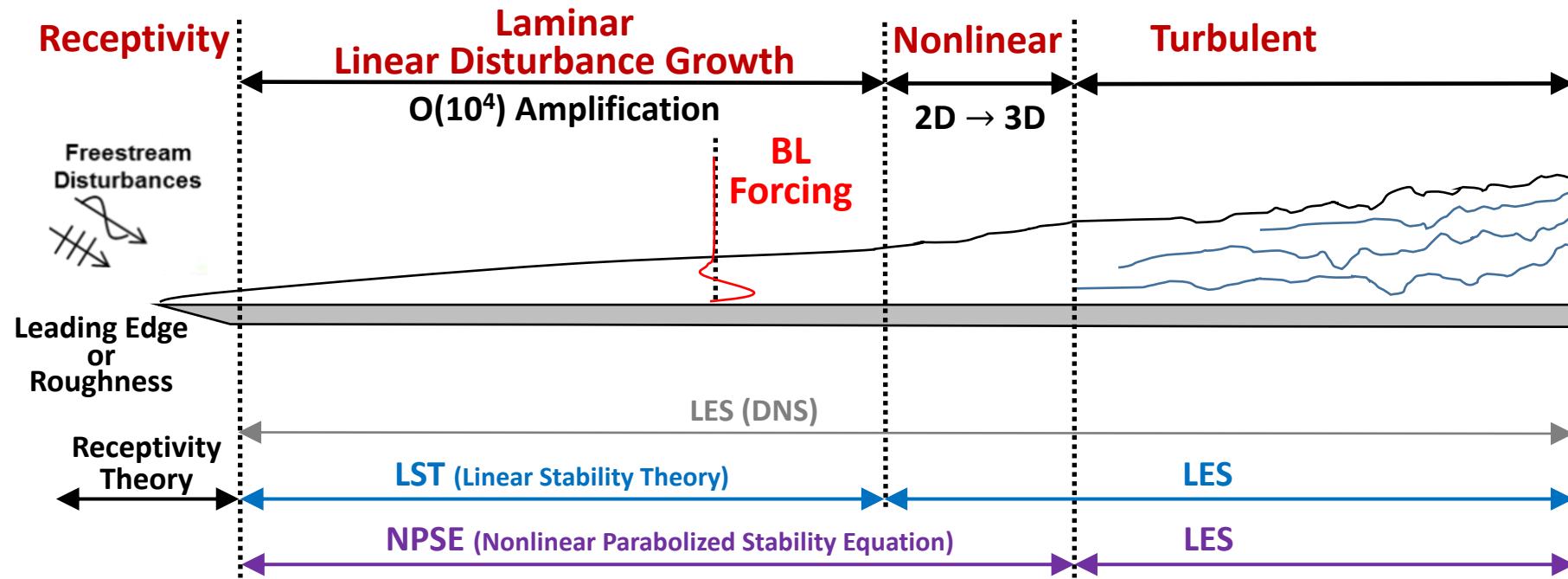
■ What kind of data do we need?

- Skin friction
- Instabilities (fluctuation in the pre-transition region)
 - Mode shape
 - Amplitude
- Velocity correlation (Reynolds stress)
 - May help to develop a RANS-based transition model
- Anything else?

High-Fidelity and Cost-Effective Computational Method : LES + PSE



High-Fidelity and Cost-Effective Computational Method : LES + PSE



- LES before L-T Transition is practically DNS
- A few modes of instability trigger L-T Transition
- Efficient method: stability theory -> forcing terms at LES Inlet
- **LES + PSE method is motivated by Philippe Spalart's early work**

[Bertolotti, F. P., Herbert, T., & Spalart, P. R. (1992). Linear and nonlinear stability of the Blasius boundary layer. *Journal of fluid mechanics*, 242, 441-474]

Two Canonical Transitional Boundary Layers

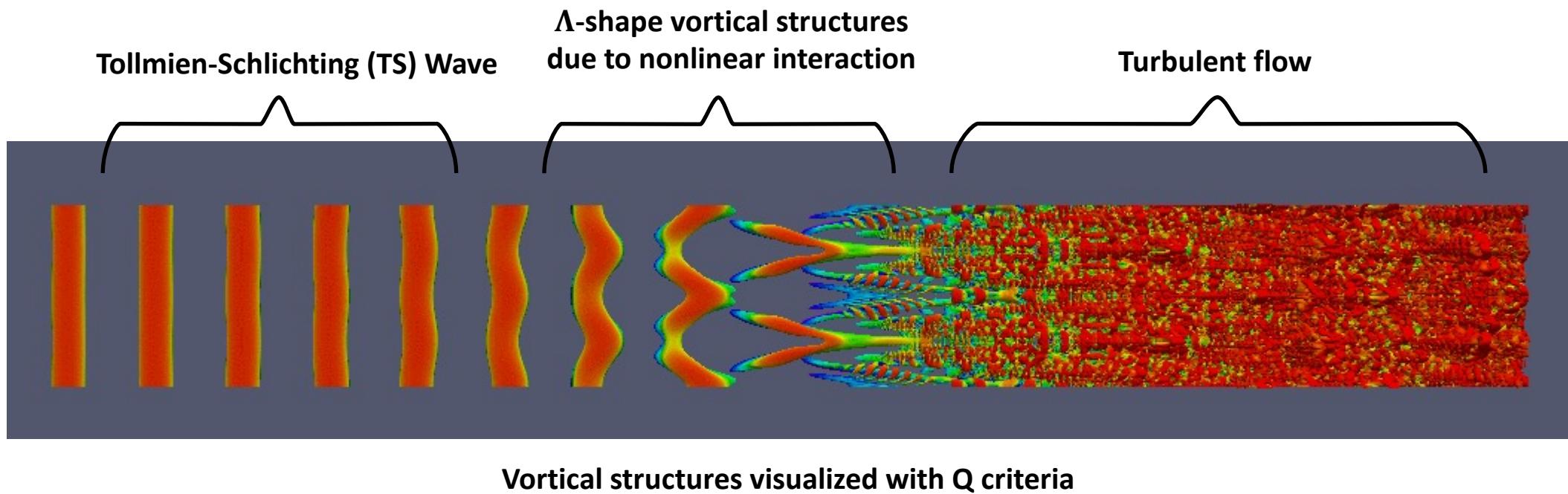
■ Case 1 : Incompressible BL

- ZPGBL on a flat plate
- Subharmonic-mode breakdown (H-type transition)
- Major instabilities
 - Fundamental planar wave (Tollmien-Schlichting wave)
 - Subharmonic oblique wave
- Solver: openFOAM
 - SGS model: WALE

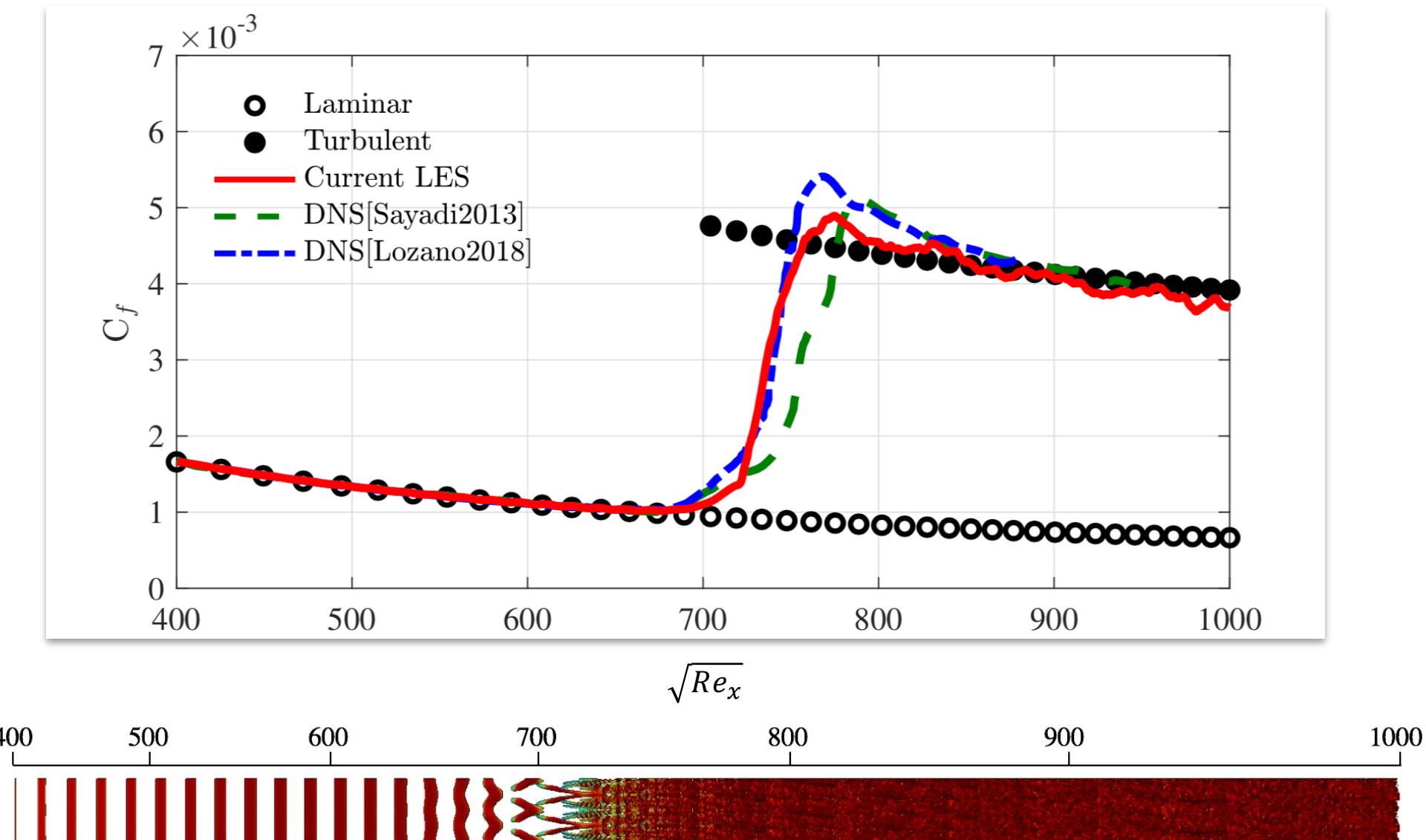
■ Case 2 : Compressible BL

- Mach = 3
- ZPGBL on a flat plate
- Oblique-mode breakdown
 - Oblique mode by itself is unstable for Mach range 2-4
- Solver: rhoEnergyFOAM [Modesti and Pirozzoli, *Comp. & Fluids*, 2017]
 - SGS model: WALE

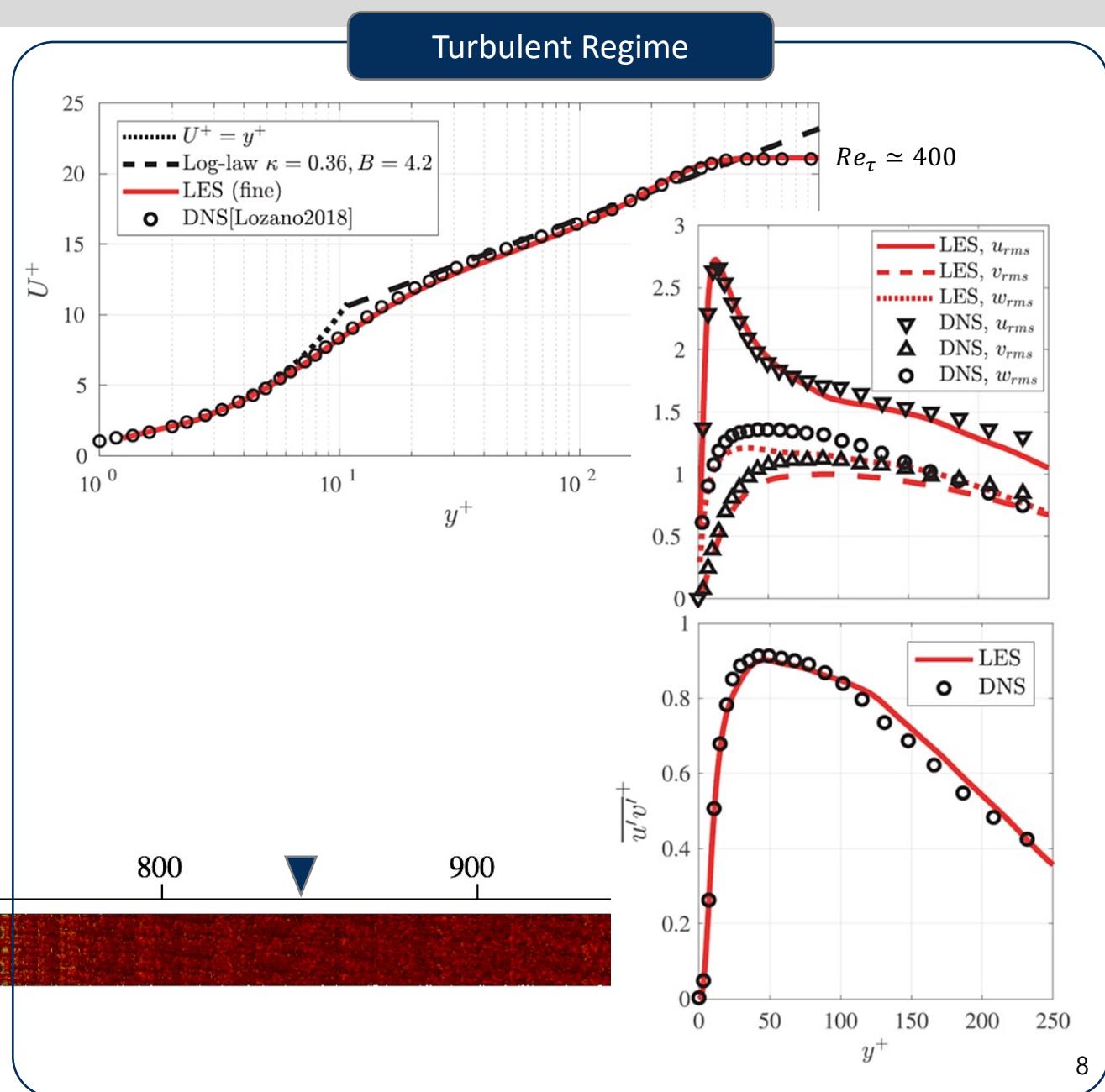
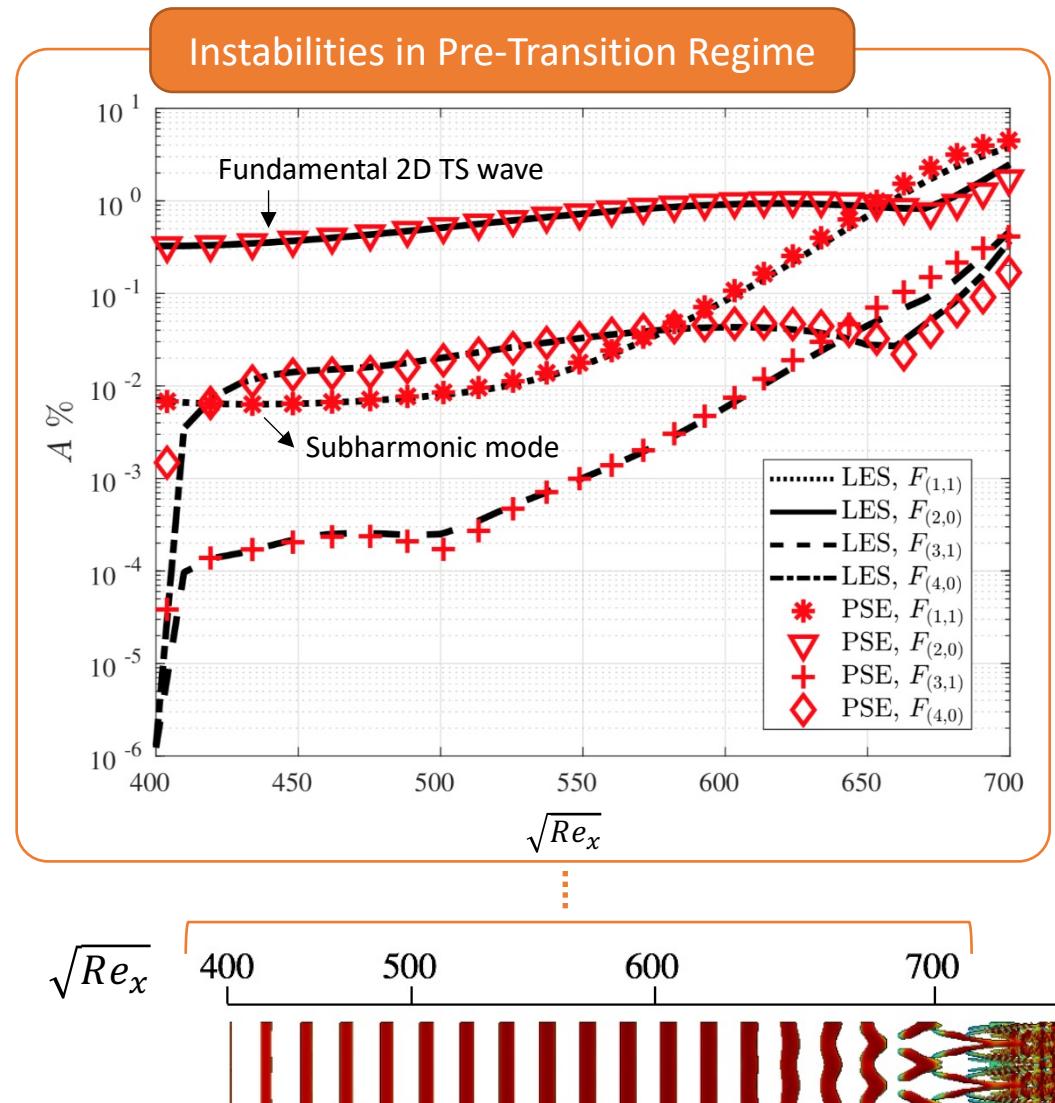
Overview on Case 1 (Incompressible BL Transition, H-type)



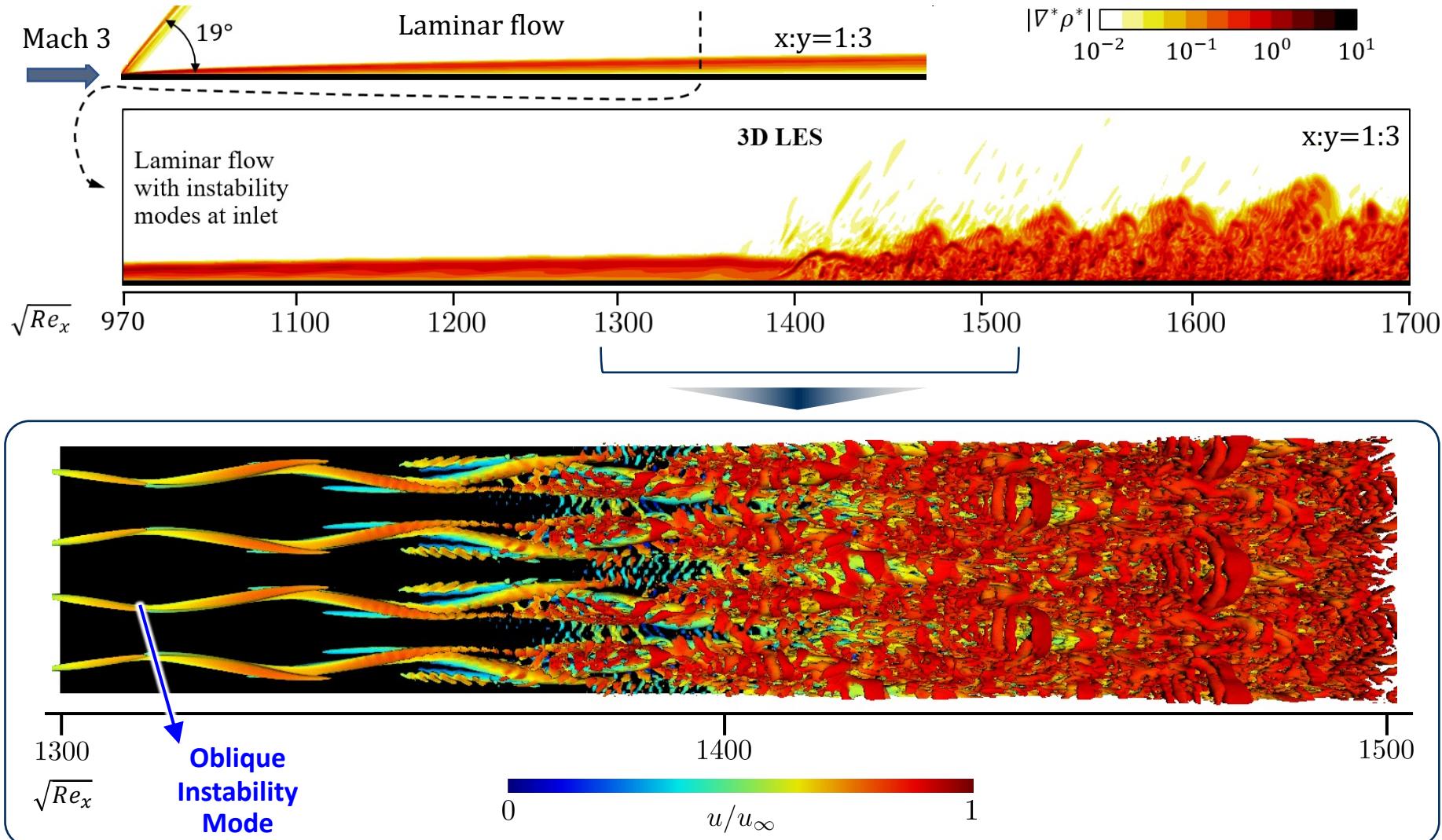
Selected Data from Case 1 (Incompressible BL Transition, H-type)



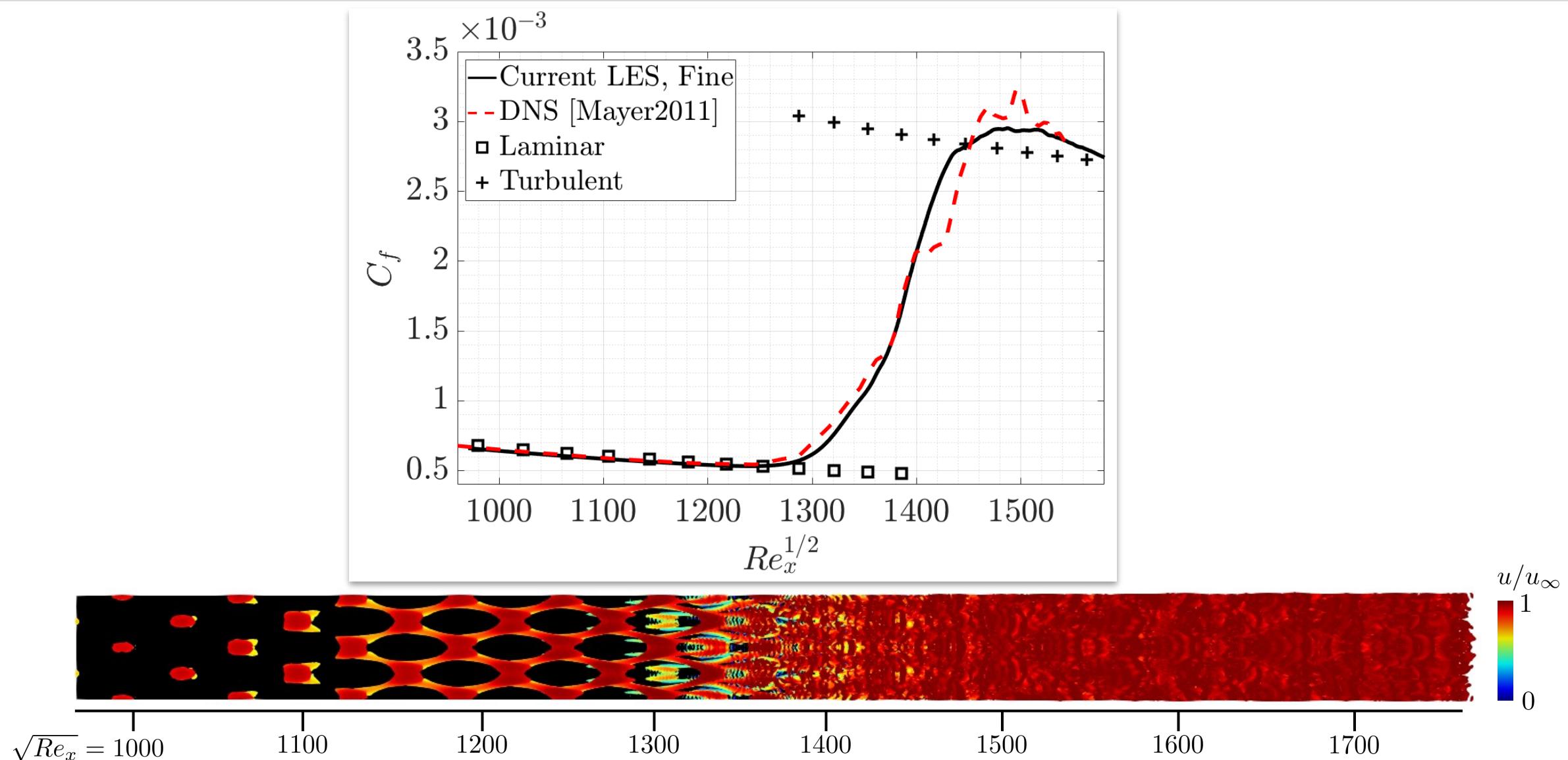
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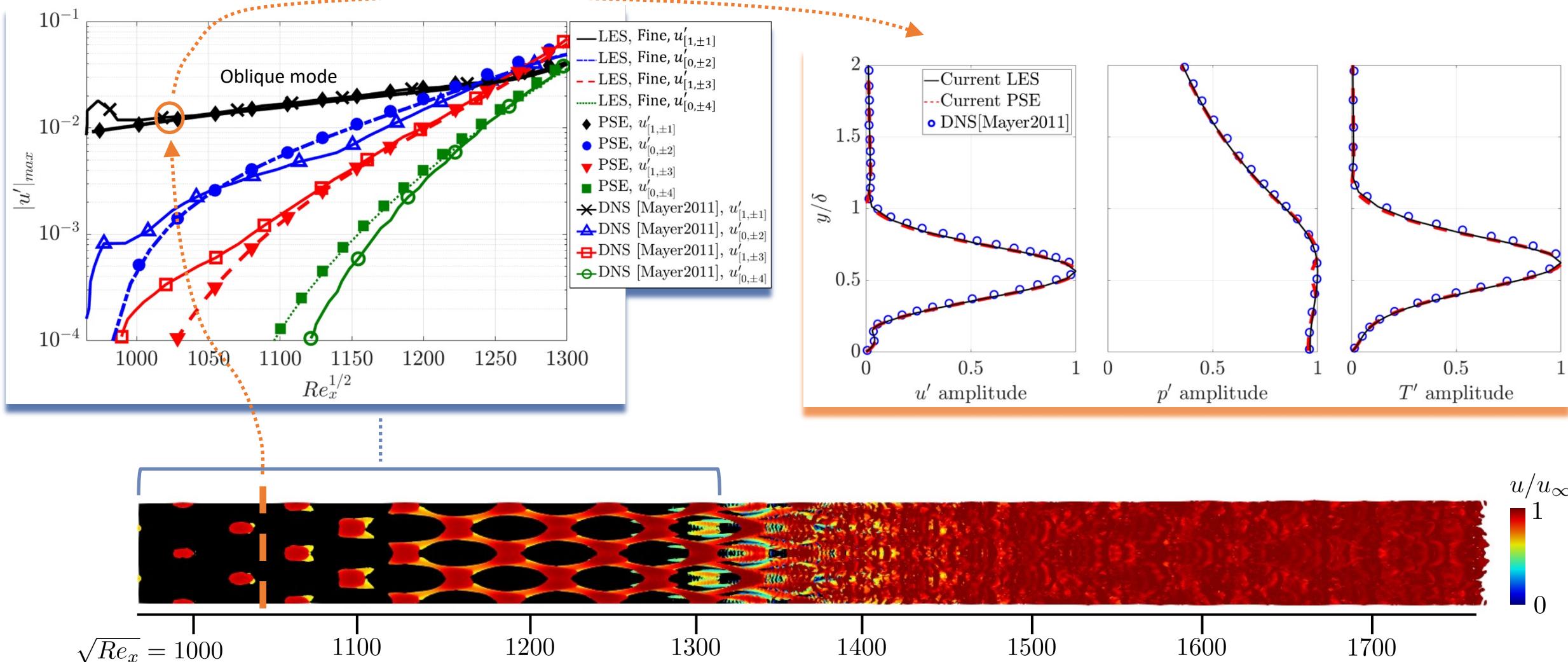
Overview on Case 2 (Supersonic BL Transition at Mach 3)



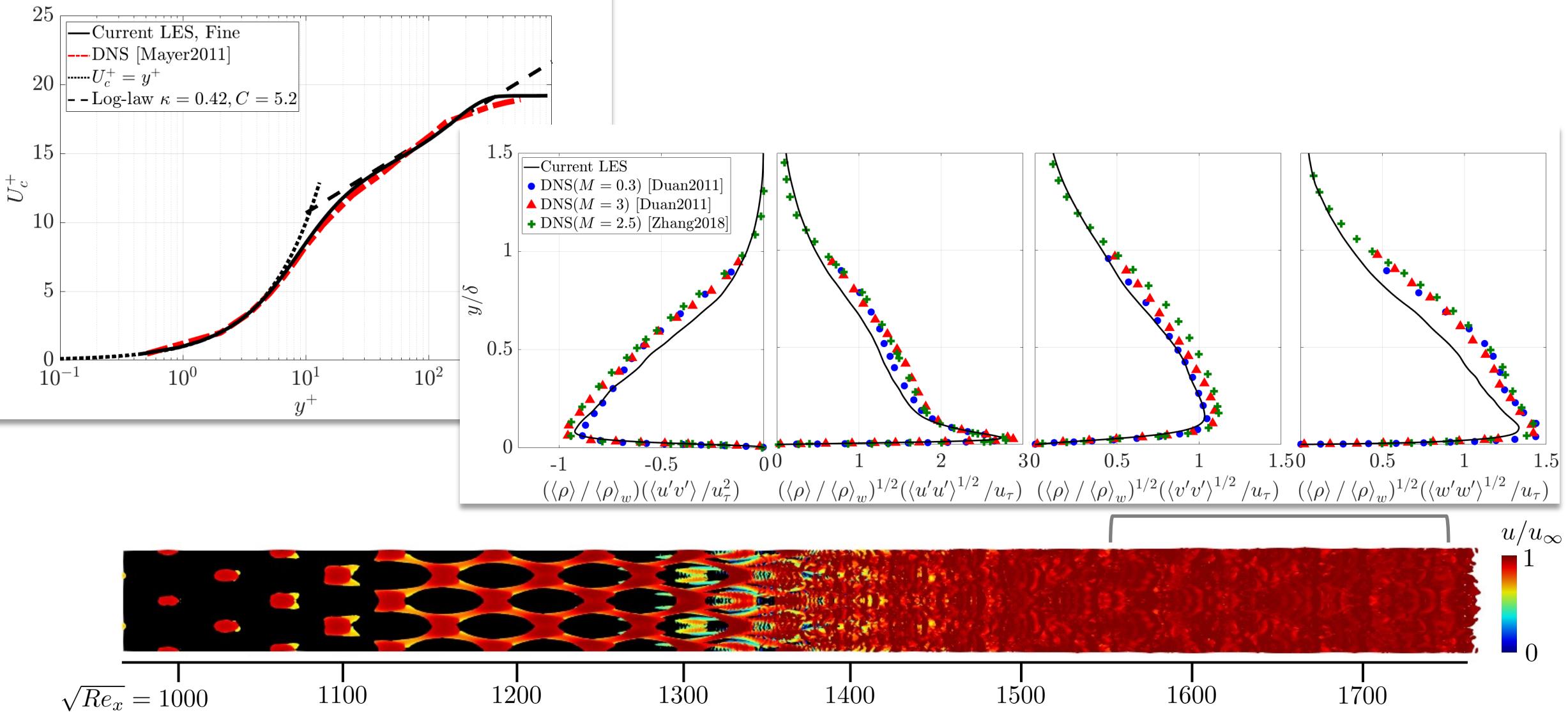
Selected Data from Case 2 (Supersonic BL Transition at Mach 3)



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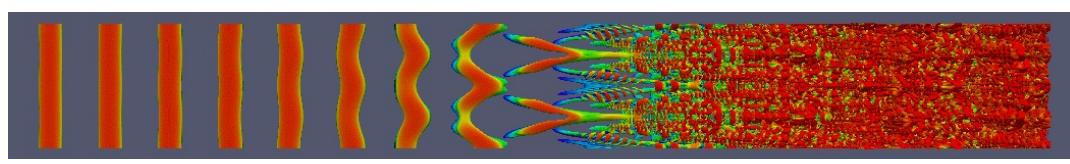
Selected Data from Case 2 (Supersonic BL Transition at Mach 3)



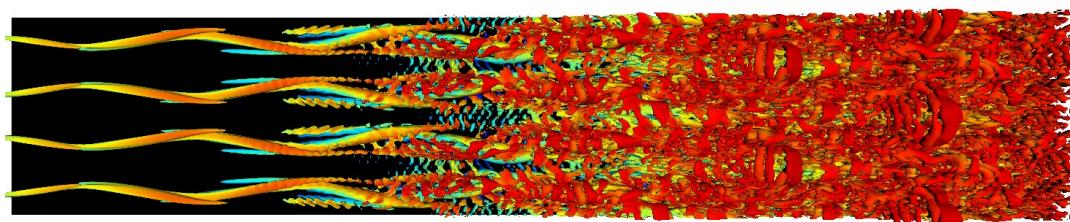
Summary – 1/2

■ Two canonical BL transition cases are available

- Incompressible BL: subharmonic breakdown (H-type)



- Supersonic BL at Mach 3: oblique-mode breakdown

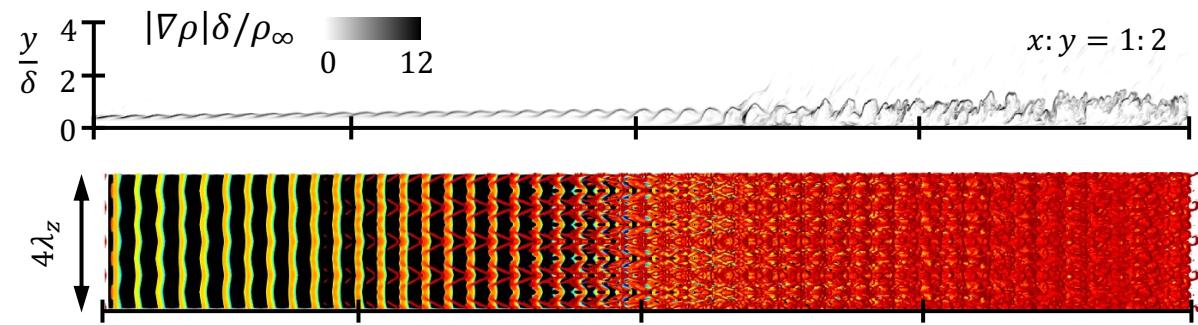


■ Vortical structures in pre-transition regions may provide insights for physics-based, data-driven transition modeling

■ Machine learning may pick up relevant flow features for transition models

■ More cases could be added

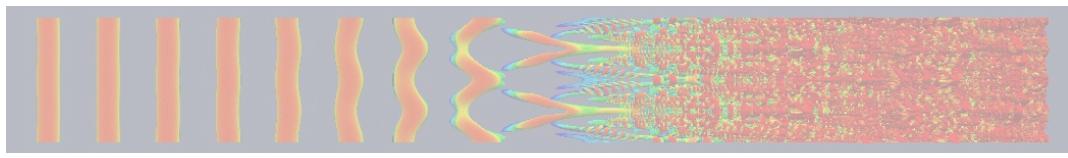
- Hypersonic BL transition at Mach 6: Mack 2nd mode and fundamental oblique mode



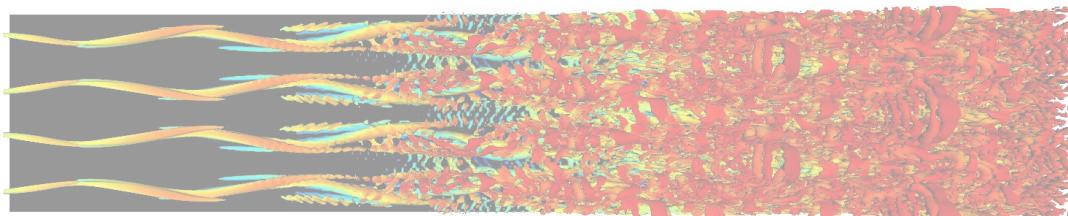
Summary – 2/2

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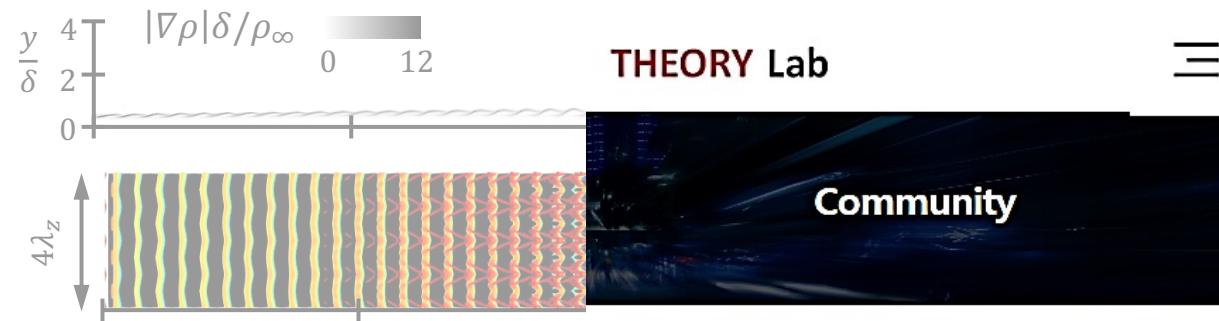


■ Data available

- Skin friction, mean flow info, anything else in papers
- Spatial and temporal development of instabilities
- Unsteady dynamics of major instability modes
- Any other upon request

■ More cases could be added

- Hypersonic BL transition at Mach 6: Mack 2nd mode and fundamental oblique mode



■ Where : theory.gist.ac.kr

- Feel free to contact me
(sjee@gist.ac.kr) for transition data

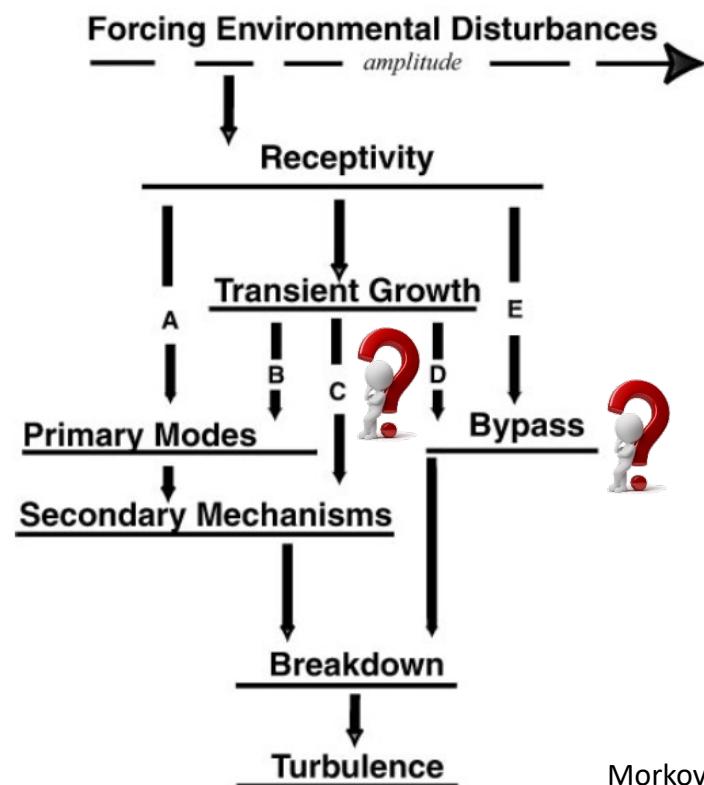
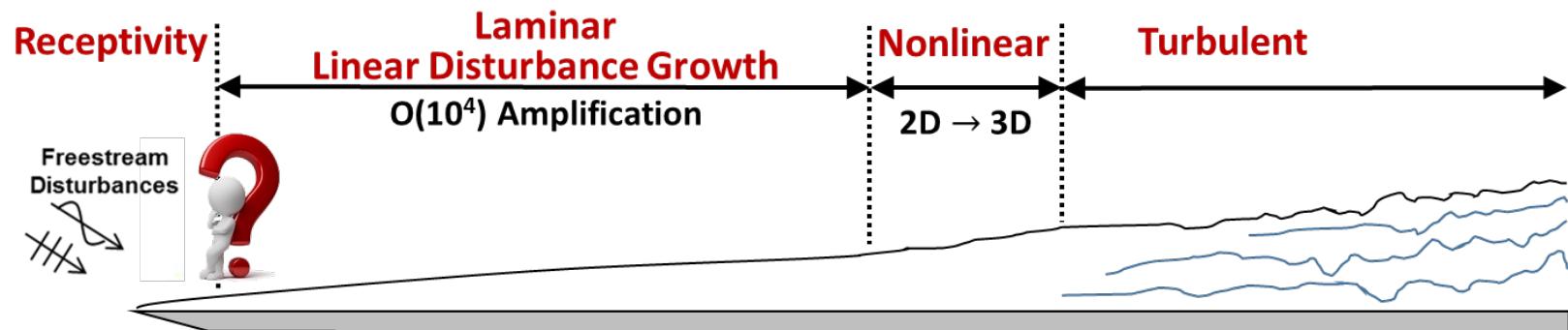
Useful Links

[HOME](#) > [Community](#) > [Useful Links](#)

- Boundary Layer Transition Data from THEORY LAB
(Click)
- NASA Turbulence Modeling Resource (Click)
- NASA Transition Modeling and CFD Vision 2030
(Click)

Final Remarks

- What else data do we need?
- Freestream disturbances?
- Other than natural transition – bypass and transient growth path?



Morkovin (1969)

Acknowledgements



National Research
Foundation of Korea



Selected Data from Case 1 (Incompressible BL Transition, H-type)

- Vortical structures in pre-transition region may provide a clue for transition location
- Here, different evolution of vortical structure comes from initial phase difference between two modes (2,0) and (1,1) at inlet
 $\sqrt{Re_x} = 400$
 - Resonance : subharmonic mode (1,1) grows exponentially almost from the beginning
 - Anti-resonance : subharmonic mode (1,1) initially damps then grow

