

Chapter 0 Formulae

0.1 Solid Sections

0.1.1 Section properties

$$A = \int_A dA = \sum_i (A_i)$$

$$Q_{XX} = \int_A Y dA = \sum_i (A_i \bar{Y}_i)$$

$$Q_{YY} = \int_A X dA = \sum_i (A_i \bar{X}_i)$$

$$\bar{X} = \frac{Q_{YY}}{A}$$

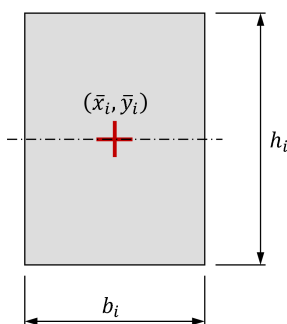
$$\bar{Y} = \frac{Q_{XX}}{A}$$

$$x = X - \bar{X}$$

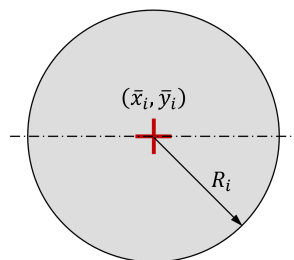
$$y = Y - \bar{Y}$$

$$\bar{x}_i = \bar{X}_i - \bar{X}$$

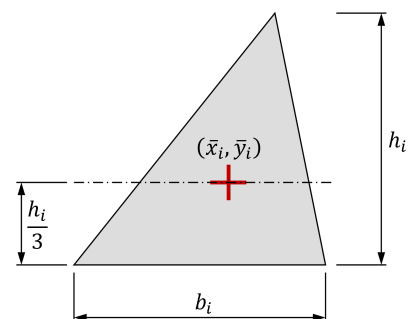
$$\bar{y}_i = \bar{Y}_i - \bar{Y}$$



$$I_{\bar{x}_i \bar{x}_i} = \frac{b_i h_i^3}{12}$$



$$I_{\bar{x}_i \bar{x}_i} = \frac{\pi R_i^4}{4}$$



$$I_{\bar{x}_i \bar{x}_i} = \frac{b_i h_i^3}{36}$$

$$I_{xx} = \int_A y^2 \, dA = \sum_i (I_{\bar{x}_i \bar{x}_i} + A_i \bar{y}_i^2)$$

$$I_{yy} = \int_A x^2 \, dA = \sum_i (I_{\bar{y}_i \bar{y}_i} + A_i \bar{x}_i^2)$$

$$I_{xy} = \int_A x y \, dA = \sum_i (I_{\bar{x}_i \bar{y}_i} + A_i \bar{x}_i \bar{y}_i)$$

0.1.2 Transformation of axes

For all equations in this section: $m = \cos \theta$ and $n = \sin \theta$.

$$\begin{Bmatrix} x' \\ y' \end{Bmatrix} = \begin{bmatrix} m & n \\ -n & m \end{bmatrix} \begin{Bmatrix} x \\ y \end{Bmatrix}$$

$$x' = x \cos \theta + y \sin \theta$$

$$y' = y \cos \theta - x \sin \theta$$

$$\begin{Bmatrix} I_{x'x'} \\ I_{y'y'} \\ I_{x'y'} \end{Bmatrix} = \begin{bmatrix} m^2 & n^2 & -2 m n \\ n^2 & m^2 & 2 m n \\ m n & -m n & m^2 - n^2 \end{bmatrix} \begin{Bmatrix} I_{xx} \\ I_{yy} \\ I_{xy} \end{Bmatrix}$$

$$I_{x'x'} = \frac{I_{xx} + I_{yy}}{2} + \left(\frac{I_{xx} - I_{yy}}{2} \right) \cos 2\theta - (I_{xy}) \sin 2\theta$$

$$I_{y'y'} = \frac{I_{xx} + I_{yy}}{2} - \left(\frac{I_{xx} - I_{yy}}{2} \right) \cos 2\theta + (I_{xy}) \sin 2\theta$$

$$I_{x'y'} = \left(\frac{I_{xx} - I_{yy}}{2} \right) \sin 2\theta + (I_{xy}) \cos 2\theta$$

$$\theta_p = -\beta_p = \frac{1}{2} \arctan \left(\frac{2 I_{xy}}{I_{yy} - I_{xx}} \right)$$

$$\beta_p = -\theta_p = \frac{1}{2} \arctan \left(\frac{2 I_{xy}}{I_{xx} - I_{yy}} \right)$$

Finding principal values from arbitrary orientation values:

$$I_{11} = \frac{I_{xx} + I_{yy}}{2} + \left(\frac{I_{xx} - I_{yy}}{2} \right) \cos 2\theta_p - (I_{xy}) \sin 2\theta_p$$

$$I_{22} = \frac{I_{xx} + I_{yy}}{2} - \left(\frac{I_{xx} - I_{yy}}{2} \right) \cos 2\theta_p + (I_{xy}) \sin 2\theta_p$$

$$I_{12} = 0$$

Finding arbitrary orientation values from principal values:

$$I_{xx} = \frac{I_{11} + I_{22}}{2} + \left(\frac{I_{11} - I_{22}}{2} \right) \cos 2\beta_p$$

$$I_{yy} = \frac{I_{11} + I_{22}}{2} - \left(\frac{I_{11} - I_{22}}{2} \right) \cos 2\beta_p$$

$$I_{xy} = \left(\frac{I_{11} - I_{22}}{2} \right) \sin 2\beta_p$$

0.1.3 Mohr's circle

$$C = \frac{I_{xx} + I_{yy}}{2}$$

$$R_{\text{Mohr}} = \sqrt{\left(\frac{I_{xx} - I_{yy}}{2} \right)^2 + I_{xy}^2}$$

$$I_{11} = C + R_{\text{Mohr}}$$

$$I_{22} = C - R_{\text{Mohr}}$$

$$\theta_p = \frac{1}{2} \arctan \left(\frac{I_{xy}}{C - I_{xx}} \right)$$

0.1.4 Direct stresses

$$\sigma_z = - \left(\frac{M_y I_{xx} + M_x I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right) x + \left(\frac{M_x I_{yy} + M_y I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right) y$$

$$\alpha = \arctan \left(\frac{M_y I_{xx} + M_x I_{xy}}{M_x I_{yy} + M_y I_{xy}} \right)$$

0.1.5 Shear stresses

$$\tau_{(y_1)} = \frac{S_Y}{I_{xx} b_{(y_1)}} \int_{y_1}^h y \, dA$$

0.2 Thin-walled Sections

$$A = \int_0^b t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i t_{(s)} \, ds$$

$$Q_{XX} = \int_0^b Y_{(s)} t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i Y_{(s)} {}^i t_{(s)} \, ds$$

$$Q_{YY} = \int_0^b X_{(s)} t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i X_{(s)} {}^i t_{(s)} \, ds$$

$$\bar{X} = \frac{Q_{YY}}{A}$$

$$\bar{Y} = \frac{Q_{XX}}{A}$$

$$x_{(s)} = X_{(s)} - \bar{X}$$

$$y_{(s)} = Y_{(s)} - \bar{Y}$$

$${}^i x_{(s)} = {}^i X_{(s)} - \bar{X}$$

$${}^i y_{(s)} = {}^i Y_{(s)} - \bar{Y}$$

$$I_{xx} = \int_0^b y_{(s)}^2 t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i y_{(s)}^2 {}^i t_{(s)} \, ds$$

$$I_{yy} = \int_0^b x_{(s)}^2 t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i x_{(s)}^2 {}^i t_{(s)} \, ds$$

$$I_{xy} = \int_0^b x_{(s)} y_{(s)} t_{(s)} \, ds = \sum_i \int_0^{b_i} {}^i x_{(s)} {}^i y_{(s)} {}^i t_{(s)} \, ds$$

0.3 Open Thin-walled Sections

0.3.1 Shear flow

$$q_{(s)} = \tau_{zs(s)} t_{(s)}$$

$$C_x = \left(\frac{S_x I_{xx} - S_y I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right) \quad C_y = \left(\frac{S_y I_{yy} - S_x I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right)$$

$$q_{(s)} = q_0 - C_x \int_0^s x_{(s)} t_{(s)} ds - C_y \int_0^s y_{(s)} t_{(s)} ds$$

$$^i q_{(s)} = q_{[i-1]} - C_x \int_0^s ^i x_{(s)} ^i t_{(s)} ds - C_y \int_0^s ^i y_{(s)} ^i t_{(s)} ds$$

0.3.2 Shear centre

$$S_Y e_X - S_X e_Y = \int_0^b q_{(s)} r_{(s)} ds = \sum_i \int_0^{b_i} ^i q_{(s)} ^i r_{(s)} ds$$

0.4 Closed Thin-walled Sections

0.4.1 Pure torsion

$$T = \bar{q} \oint r_{(s)} ds \quad \frac{d\theta}{dz} = \frac{T}{4 \bar{A}^2} \oint \frac{1}{G t_{(s)}} ds$$

$$T = \bar{q} (2 \bar{A}) \quad \frac{d\theta}{dz} = \frac{1}{2 \bar{A}} \oint \frac{\bar{q}}{G t_{(s)}} ds$$

0.4.2 Torsional constant

$$T = \frac{d\theta}{dz} G J$$

$$J = \frac{4 \bar{A}^2}{\oint \frac{ds}{t}}$$

0.4.3 Open shear flow

$$C_x = \left(\frac{S_x I_{xx} - S_y I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right) \quad C_y = \left(\frac{S_y I_{yy} - S_x I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right)$$

$$q_{(s)}^{\text{open}} = q_0 - C_x \int_0^s x_{(s)} t_{(s)} ds - C_y \int_0^s y_{(s)} t_{(s)} ds$$

$$^i q_{(s)}^{\text{open}} = q_{[i-1]} - C_x \int_0^s ^i x_{(s)} ^i t_{(s)} ds - C_y \int_0^s ^i y_{(s)} ^i t_{(s)} ds$$

0.4.4 Closed-cell constant

$$\bar{q} = - \frac{\oint q_{(s)}^{\text{open}} ds}{\oint ds}$$

0.4.5 Closed shear flow

$$q_{(s)}^{\text{closed}} = q_{(s)}^{\text{open}} + \bar{q}$$

0.4.6 Angle of twist

$$\frac{d\theta}{dz} = \frac{1}{2A} \oint \frac{q_{(s)}^{\text{closed}}}{G t_{(s)}} ds$$

0.4.7 Shear centre

$$S_Y e_X - S_X e_Y = \oint q_{(s)}^{\text{closed}} r_{(s)} ds = \sum_i \int_0^{b_i} ^i q_{(s)}^{\text{closed}} ^i r_{(s)} ds$$

0.5 Idealised Sections

0.5.1 Effective boom areas

Shear load S_X :

$$B_k = J_k + \sum_{\text{connected booms } l} \frac{b_{kl} t_{kl}}{6} \left(2 + \frac{x_l}{x_k} \right)$$

Shear load S_Y :

$$B_k = J_k + \sum_{\text{connected booms } l} \frac{b_{kl} t_{kl}}{6} \left(2 + \frac{y_l}{y_k} \right)$$

where:

- J_k is the cross-sectional area of any 'joint' (or local reinforcement) at boom k ;
- l indexes all neighbouring booms connected to boom k ;
- b_{kl} is the arc-length of the skin connecting k and l ;
- t_{kl} is the thickness of the skin connecting k and l .

0.5.2 Idealised shear flow

$$C_x = \left(\frac{S_x I_{xx} - S_y I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right) \quad C_y = \left(\frac{S_y I_{yy} - S_x I_{xy}}{I_{xx} I_{yy} - I_{xy}^2} \right)$$

$$q_i^{\text{open}} = - C_x \sum_k x_k B_k - C_y \sum_k y_k B_k$$

$$q_i^{\text{closed}} = q_i^{\text{open}} + \bar{q}$$

0.5.3 Multi-cell sections

Rate of twist of each cell $j = 1 \dots n$:

$$\sum_{i \in j} \left[(q_i^{\text{open}} + {}^* \bar{q}_i) \frac{b_i}{t_i} \right] = (2 \bar{A}_j G) \frac{d\theta}{dz}$$

where:

$${}^* \bar{q}_i = \begin{cases} \bar{q}_j & \text{if wall } i \text{ is exclusive to cell } j \\ \bar{q}_j - \bar{q}_{[j \pm 1]} & \text{if wall } i \text{ is shared with cell } [j \pm 1] \end{cases}$$

Balance of torques about the origin of (X, Y) :

$$T + S_Y e_X - S_X e_Y = \sum_{\text{all walls } i} (q_i^{\text{open}} b_i r_i) + \sum_{\text{all cells } j} (2 \bar{A}_j \bar{q}_j)$$

where r_i is the orthogonal distance function $r_{(s)}$ for wall i with respect to the origin of (X, Y) .

If $r_{(s)}$ is not constant within wall i then the 'subtended area' \hat{A}_i can be used instead:

$$T + S_Y e_X - S_X e_Y = \sum_{\text{all walls } i} (2 \hat{A}_i q_i^{\text{open}}) + \sum_{\text{all cells } j} (2 \bar{A}_j \bar{q}_j)$$