



University of
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Fatigue

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Fatigue

Part-3: Fatigue prevention

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Fatigue life

$$N = N_{\text{stage-1}} + N_{\text{stage-2}} + N_{\text{stage-3}}$$

$$N = N_{\text{initiation}} + N_{\text{slow propagation}} + \cancel{N_{\text{fast propagation}}}$$

$$N = N_i + N_{SP}$$

(since fast propagation time is negligible)

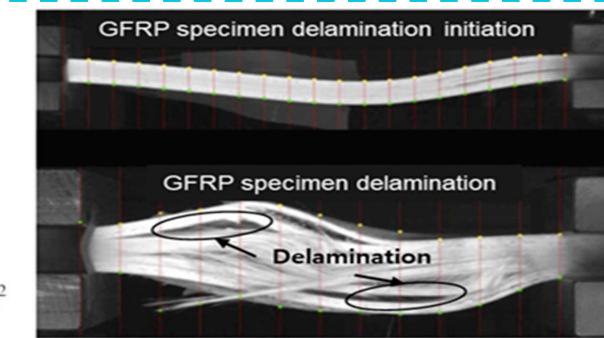
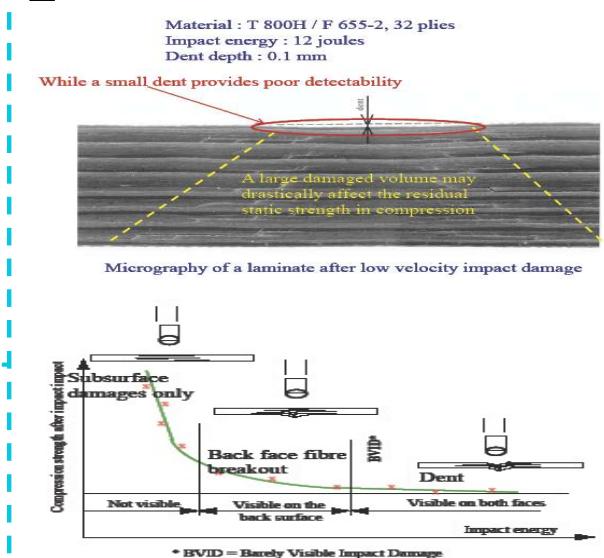
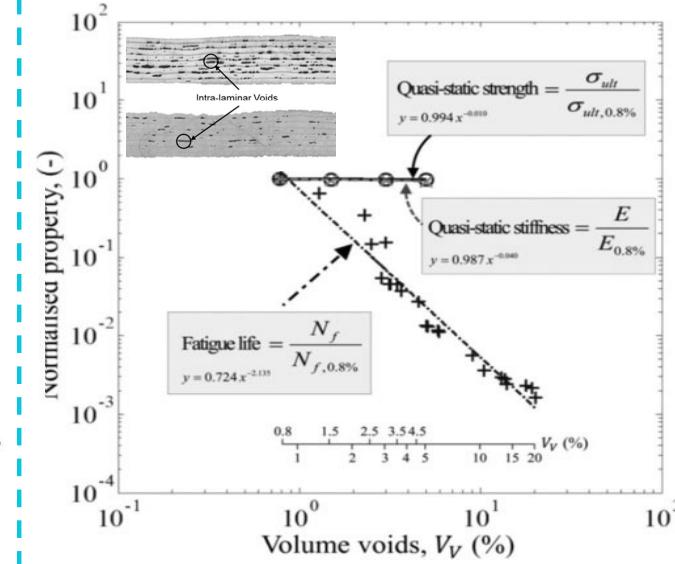
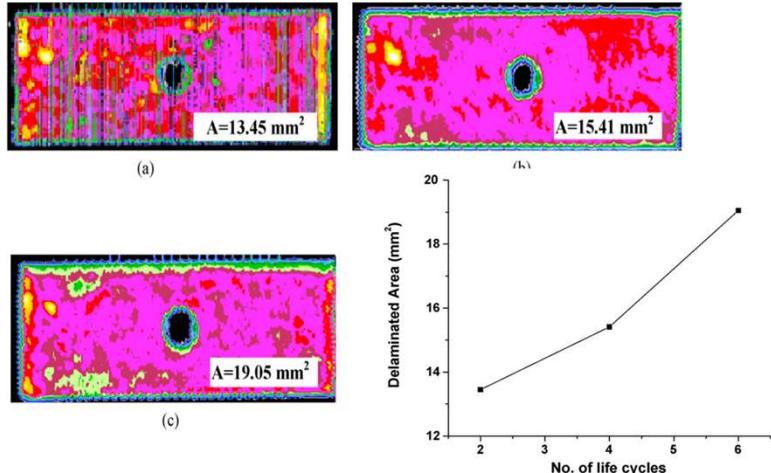
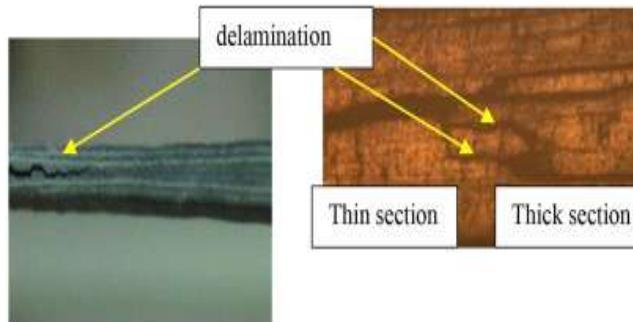
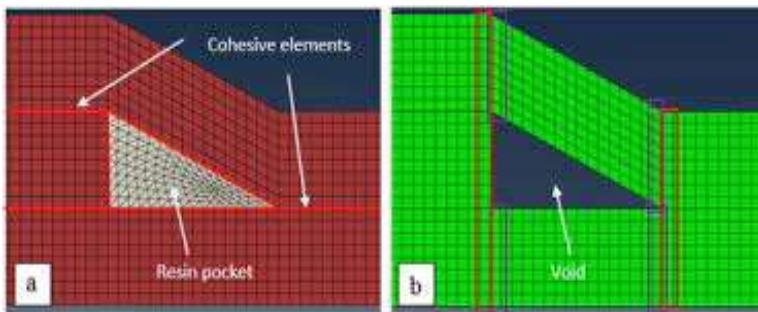
Prevent crack initiation / increase Ni

- Crack initiates when $S_{local} > YS$
- To reduce chance of crack initiation
 - Use material with higher YS
 - Stronger materials
 - Strengthening mechanisms
 - Lower S_{local} by reducing stress concentrations
 - Good surface finish
 - Proper component design (no abrupt changes in cross section, no sharp corners or edges etc.)
 - No microscopic defects like porosity, cavity, non-metallic inclusions (use advanced manufacturing like vacuum melting, ESR, preprints etc.)

Fatigue notch sensitivity

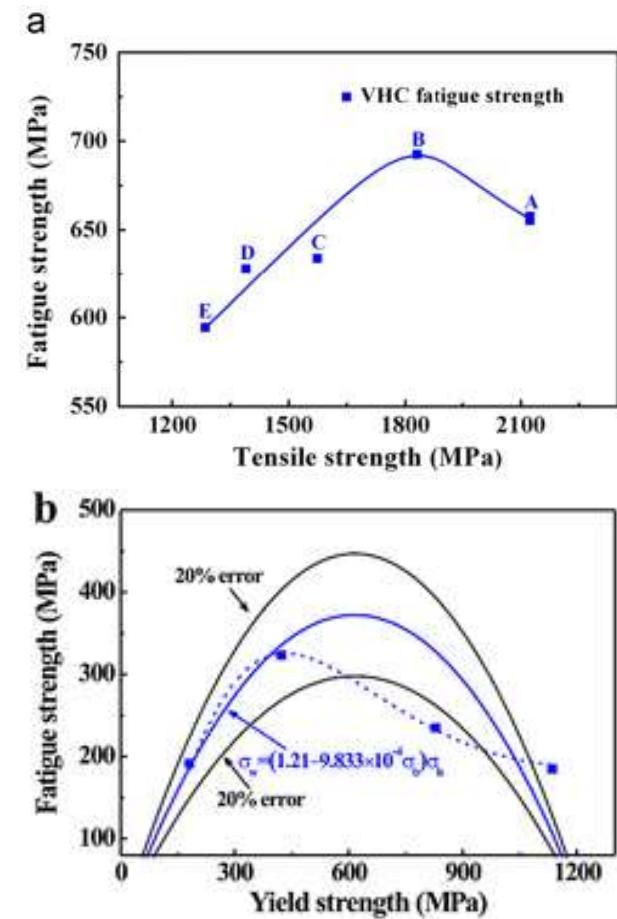
- Notch fatigue strength reduction factor (K_f) = fatigue strength without notch / fatigue strength with notch
 - ≠ Material property (depends on size, shape, stress etc.)
- Fatigue notch sensitivity parameter (q) = $(K_f - 1) / (K_t - 1)$
 - K_t = Monotonic notch concentration factor
 - Highest notch sensitivity ($q = 1$) when $K_f = K_t$
 - Least notch sensitivity ($q = 0$) when $K_f = 1$
 - Only indicative

Few unique stress raisers in composites



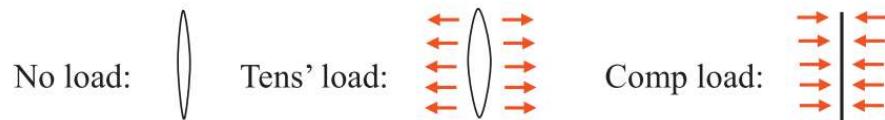
Delay slow propagation of crack / increase N_{SP}

- $da/dN = (\text{crack advancement} - \text{crack closure})$
 - To reduce crack advancement: Higher YS (lower plastic deformation)
 - To increase crack closure: Higher plastic deformation (lower YS)
 - Contradictory requirement needing optimization
 - Surface / case hardening
 - Residual compressive stresses



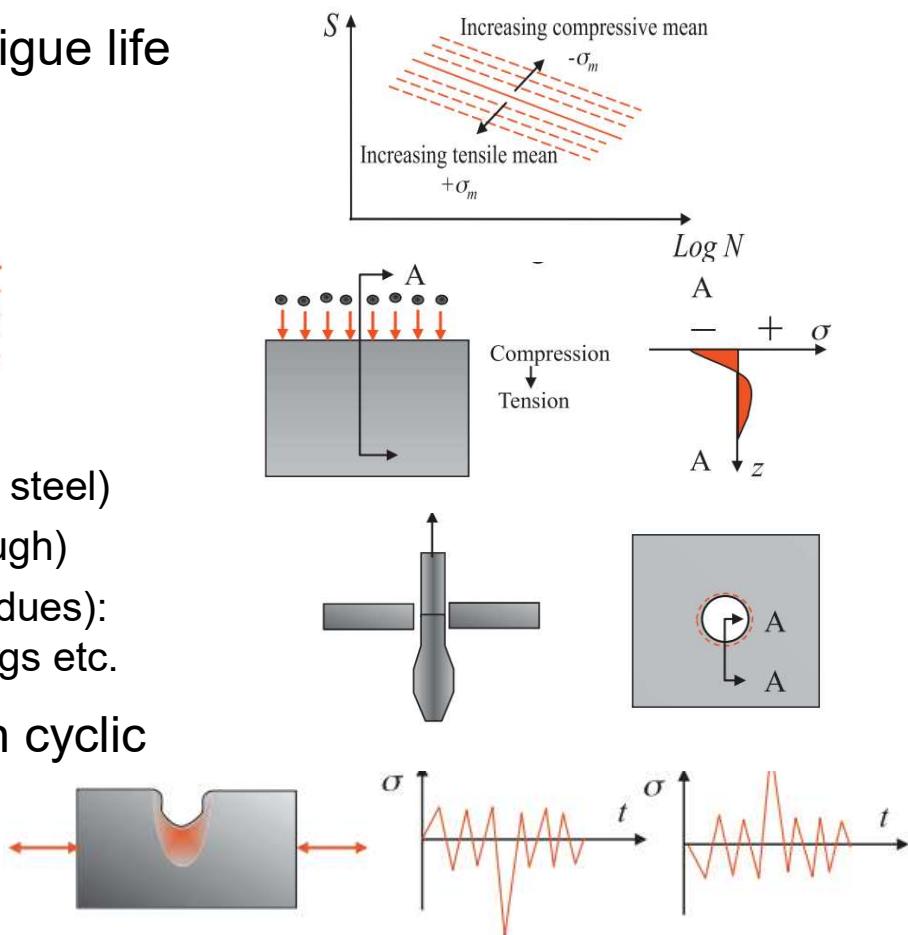
Residual compressive stresses

- Recall: Compressive mean stress improves fatigue life
 - Tension: opens up crack
 - Compression: Closes crack



- Residual stress
 - Shot peening / sand blasting (10x fatigue life for mild steel)
 - Cold worked holes (with oversized mandrel pull through)
 - Overloads (tensile overload leaves compressive residues):
e.g. freak gust, emergency maneuvers, heavy landings etc.
- Need to be careful about relaxation effects with cyclic loading / heating

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End of Part-3

Fatigue

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Thank you

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