

Appendix 2: Definitions of Aerodynamic Stability and Control Derivatives

A2.1 Notes

The derivatives given in [Tables A2.5 through A2.8](#) are referred to **generalised body axes**; $U_e = V_0 \cos \theta_e$ and $W_e = V_0 \sin \theta_e$. In the particular case when the derivatives are referred to wind axes, $\theta_e = 0$ and the following simplifications can be made: $U_e = V_0$, $W_e = 0$, $\sin \theta_e = 0$ and $\cos \theta_e = 1$. The equivalent algebraic expressions in [Tables A2.5 to A2.8](#) were derived with the aid of the computer program Mathcad, which includes a facility for symbolic calculation. In [Tables A2.5, A2.6, A2.7, and A2.8](#) normalised mass and inertias are used. They are defined as follows:

$$m' = \frac{m}{\frac{1}{2}\rho V_0 S}$$

$$I'_x = \frac{I_x}{\frac{1}{2}\rho V_0 S b}$$

$$I'_y = \frac{I_y}{\frac{1}{2}\rho V_0 S \bar{c}}$$

$$I'_z = \frac{I_z}{\frac{1}{2}\rho V_0 S b}$$

$$I'_{xz} = \frac{I_{xz}}{\frac{1}{2}\rho V_0 S b}$$

Table A2.1 Longitudinal Aerodynamic Stability Derivatives		
Dimensionless	Multiplier	Dimensional
X_u	$\frac{1}{2}\rho V_0 S$	$\overset{\circ}{X}_u$
X_w	$\frac{1}{2}\rho V_0 S$	$\overset{\circ}{X}_w$
$X_{\dot{w}}$	$\frac{1}{2}\rho S \bar{c}$	$\overset{\circ}{X}_{\dot{w}}$
X_q	$\frac{1}{2}\rho V_0 S \bar{c}$	$\overset{\circ}{X}_q$
Z_u	$\frac{1}{2}\rho V_0 S$	$\overset{\circ}{Z}_u$
Z_w	$\frac{1}{2}\rho V_0 S$	$\overset{\circ}{Z}_w$
$Z_{\dot{w}}$	$\frac{1}{2}\rho S \bar{c}$	$\overset{\circ}{Z}_{\dot{w}}$
Z_q	$\frac{1}{2}\rho V_0 S \bar{c}$	$\overset{\circ}{Z}_q$
M_u	$\frac{1}{2}\rho V_0 S \bar{c}$	$\overset{\circ}{M}_u$
M_w	$\frac{1}{2}\rho V_0 S \bar{c}$	$\overset{\circ}{M}_w$
$M_{\dot{w}}$	$\frac{1}{2}\rho S \bar{c}^2$	$\overset{\circ}{M}_{\dot{w}}$
M_q	$\frac{1}{2}\rho V_0 S \bar{c}^2$	$\overset{\circ}{M}_q$

Table A2.2 Longitudinal Control Derivatives		
Dimensionless	Multiplier	Dimensional
X_η	$\frac{1}{2}\rho V_0^2 S$	$\overset{\circ}{X}_\eta$
Z_η	$\frac{1}{2}\rho V_0^2 S$	$\overset{\circ}{Z}_\eta$
M_η	$\frac{1}{2}\rho V_0^2 S \bar{c}$	$\overset{\circ}{M}_\eta$
X_τ	1	$\overset{\circ}{X}_\tau$
Z_τ	1	$\overset{\circ}{Z}_\tau$
M_τ	\bar{c}	$\overset{\circ}{M}_\tau$

Table A2.3 Lateral-Directional Aerodynamic Stability Derivatives

Dimensionless	Multiplier	Dimensional
Y_v	$\frac{1}{2}\rho V_0 S$	\dot{Y}_v
Y_p	$\frac{1}{2}\rho V_0 S b$	\dot{Y}_p
Y_r	$\frac{1}{2}\rho V_0 S b$	\dot{Y}_r
L_v	$\frac{1}{2}\rho V_0 S b$	\dot{L}_v
L_p	$\frac{1}{2}\rho V_0 S b^2$	\dot{L}_p
L_r	$\frac{1}{2}\rho V_0 S b^2$	\dot{L}_r
N_v	$\frac{1}{2}\rho V_0 S b$	\dot{N}_v
N_p	$\frac{1}{2}\rho V_0 S b^2$	\dot{N}_p
N_r	$\frac{1}{2}\rho V_0 S b^2$	\dot{N}_r

Table A2.4 Lateral-Directional Aerodynamic Control Derivatives

Dimensionless	Multiplier	Dimensional
Y_ξ	$\frac{1}{2}\rho V_0^2 S$	\dot{Y}_ξ
L_ξ	$\frac{1}{2}\rho V_0^2 S b$	\dot{L}_ξ
N_ξ	$\frac{1}{2}\rho V_0^2 S b$	\dot{N}_ξ
Y_ζ	$\frac{1}{2}\rho V_0^2 S$	\dot{Y}_ζ
L_ζ	$\frac{1}{2}\rho V_0^2 S b$	\dot{L}_ζ
N_ζ	$\frac{1}{2}\rho V_0^2 S b$	\dot{N}_ζ

Table A2.5 Concise Longitudinal Aerodynamic Stability Derivatives

Equivalent Expressions		
Concise Derivative	In Terms of Dimensional Derivatives	In Terms of Dimensionless Derivatives
x_u	$\frac{\dot{X}_u}{m} + \frac{\dot{X}_w \dot{Z}_u}{m(m - \dot{Z}_w)}$	$\frac{X_u}{m'} + \frac{\frac{\bar{c}}{V_0} \dot{X}_w Z_u}{m'(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
z_u	$\frac{\dot{Z}_u}{m - \dot{Z}_w}$	$\frac{Z_u}{m' - \frac{\bar{c}}{V_0} \dot{Z}_w}$
m_u	$\frac{\dot{M}_u}{I_y} + \frac{\dot{Z}_u \dot{M}_w}{I_y(m - \dot{Z}_w)}$	$\frac{M_u}{I'_y} + \frac{\frac{\bar{c}}{V_0} \dot{M}_w Z_u}{I'_y(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
x_w	$\frac{\dot{X}_w}{m} + \frac{\dot{X}_w \dot{Z}_w}{m(m - \dot{Z}_w)}$	$\frac{X_w}{m'} + \frac{\frac{\bar{c}}{V_0} \dot{X}_w Z_w}{m'(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
z_w	$\frac{\dot{Z}_w}{m - \dot{Z}_w}$	$\frac{Z_w}{m' - \frac{\bar{c}}{V_0} \dot{Z}_w}$
m_w	$\frac{\dot{M}_w}{I_y} + \frac{\dot{Z}_w \dot{M}_w}{I_y(m - \dot{Z}_w)}$	$\frac{M_w}{I'_y} + \frac{\frac{\bar{c}}{V_0} \dot{M}_w Z_w}{I'_y(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
x_q	$\frac{\dot{X}_q - mW_e}{m} + \frac{(Z_q + mU_e) \dot{X}_w}{m(m - \dot{Z}_w)}$	$\frac{\bar{c}X_q - m'W_e}{m'} + \frac{(\bar{c}Z_q + m'U_e) \frac{\bar{c}}{V_0} \dot{X}_w}{m'(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
z_q	$\frac{\dot{Z}_q + mU_e}{m - \dot{Z}_w}$	$\frac{\bar{c}Z_q + m'U_e}{m' - \frac{\bar{c}}{V_0} \dot{Z}_w}$
m_q	$\frac{\dot{M}_q}{I_y} + \frac{(Z_q + mU_e) \dot{M}_w}{I_y(m - \dot{Z}_w)}$	$\frac{\bar{c}M_q}{I'_y} + \frac{(\bar{c}Z_q + m'U_e) \frac{\bar{c}}{V_0} \dot{M}_w}{I'_y(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$
x_θ	$-g \cos \theta_e - \frac{\dot{X}_w g \sin \theta_e}{m - \dot{Z}_w}$	$-g \cos \theta_e - \frac{\frac{\bar{c}}{V_0} \dot{X}_w g \sin \theta_e}{m' - \frac{\bar{c}}{V_0} \dot{Z}_w}$
z_θ	$-\frac{mg \sin \theta_e}{m - \dot{Z}_w}$	$-\frac{m'g \sin \theta_e}{m' - \frac{\bar{c}}{V_0} \dot{Z}_w}$
m_θ	$-\frac{\dot{M}_w mg \sin \theta_e}{I_y(m - \dot{Z}_w)}$	$-\frac{\frac{\bar{c}}{V_0} \dot{M}_w m'g \sin \theta_e}{I'_y(m' - \frac{\bar{c}}{V_0} \dot{Z}_w)}$

Table A2.6 Concise Longitudinal Control Derivatives

Equivalent Expressions		
Concise Derivative	In Terms of Dimensional Derivatives	In Terms of Dimensionless Derivatives
x_η	$\frac{\dot{X}_\eta}{m} + \frac{\dot{X}_{\dot{w}} \dot{Z}_\eta}{m(m - \dot{Z}_{\dot{w}})}$	$\frac{V_0 X_\eta}{m'} + \frac{\frac{\bar{c}}{V_0} X_{\dot{w}} Z_\eta}{m'(m' - \frac{\bar{c}}{V_0} Z_{\dot{w}})}$
z_η	$\frac{\dot{Z}_\eta}{m - \dot{Z}_{\dot{w}}}$	$\frac{V_0 Z_\eta}{m' - \frac{\bar{c}}{V_0} Z_{\dot{w}}}$
m_η	$\frac{\dot{M}_\eta}{I_y} + \frac{\dot{M}_{\dot{w}} \dot{Z}_\eta}{I_y(m - \dot{Z}_{\dot{w}})}$	$\frac{V_0 M_\eta}{I'_y} + \frac{\bar{c} M_{\dot{w}} Z_\eta}{I'_y(m' - \frac{\bar{c}}{V_0} Z_{\dot{w}})}$
x_τ	$\frac{\dot{X}_\tau}{m} + \frac{\dot{X}_{\dot{w}} \dot{Z}_\tau}{m(m - \dot{Z}_{\dot{w}})}$	$\frac{V_0 X_\tau}{m'} + \frac{\frac{\bar{c}}{V_0} X_{\dot{w}} Z_\tau}{m'(m' - \frac{\bar{c}}{V_0} Z_{\dot{w}})}$
z_τ	$\frac{\dot{Z}_\tau}{m - \dot{Z}_{\dot{w}}}$	$\frac{V_0 Z_\tau}{m' - \frac{\bar{c}}{V_0} Z_{\dot{w}}}$
m_τ	$\frac{\dot{M}_\tau}{I_y} + \frac{\dot{M}_{\dot{w}} \dot{Z}_\tau}{I_y(m - \dot{Z}_{\dot{w}})}$	$\frac{V_0 M_\tau}{I'_y} + \frac{\bar{c} M_{\dot{w}} Z_\tau}{I'_y(m' - \frac{\bar{c}}{V_0} Z_{\dot{w}})}$

Table A2.7 Concise Lateral-Directional Aerodynamic Stability Derivatives

Equivalent Expressions		
Concise Derivative	In Terms of Dimensional Derivatives	In Terms of Dimensionless Derivatives
y_v	$\frac{\dot{Y}_v}{m}$	$\frac{Y_v}{m'}$
y_p	$\frac{(\dot{Y}_p + mW_e)}{m}$	$\frac{(bY_p + m'W_e)}{m'}$
y_r	$\frac{(\dot{Y}_r - mU_e)}{m}$	$\frac{(bY_r - m'U_e)}{m'}$
y_ϕ	$g \cos\theta_e$	$g \cos\theta_e$
y_ψ	$g \sin\theta_e$	$g \sin\theta_e$
l_v	$\frac{(I_z \dot{L}_v + I_{xz} \dot{N}_v)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_z L_v + I'_{xz} N_v)}{(I'_x I'_z - I_{xz}^2)}$
l_p	$\frac{(I_z \dot{L}_p + I_{xz} \dot{N}_p)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_z L_p + I'_{xz} N_p)b}{(I'_x I'_z - I_{xz}^2)}$
l_r	$\frac{(I_z \dot{L}_r + I_{xz} \dot{N}_r)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_z L_r + I'_{xz} N_r)b}{(I'_x I'_z - I_{xz}^2)}$
l_ϕ	0	0
l_ψ	0	0
n_v	$\frac{(I_x \dot{N}_v + I_{xz} \dot{L}_v)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_x N_v + I'_{xz} L_v)}{(I'_x I'_z - I_{xz}^2)}$
n_p	$\frac{(I_x \dot{N}_p + I_{xz} \dot{L}_p)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_x N_p + I'_{xz} L_p)b}{(I'_x I'_z - I_{xz}^2)}$
n_r	$\frac{(I_x \dot{N}_r + I_{xz} \dot{L}_r)}{(I_x I_z - I_{xz}^2)}$	$\frac{(I'_x N_r + I'_{xz} L_r)b}{(I'_x I'_z - I_{xz}^2)}$
n_ϕ	0	0
n_ψ	0	0

Table A2.8 Concise Lateral-Directional Control Derivatives		
Equivalent Expressions		
Concise Derivative	In Terms of Dimensional Derivatives	In Terms of Dimensionless Derivatives
y_ξ	$\frac{\overset{\circ}{Y}_\xi}{m}$	$\frac{V_0 Y_\xi}{m'}$
l_ξ	$\frac{\left(I_z \overset{\circ}{L}_\xi + I_{xz} \overset{\circ}{N}_\xi\right)}{(I_x I_z - I_{xz}^2)}$	$\frac{V_0(I'_z L_\xi + I'_{xz} N_\xi)}{(I'_x I'_z - I_{xz}^2)}$
n_ξ	$\frac{\left(I_x \overset{\circ}{N}_\xi + I_{xz} \overset{\circ}{L}_\xi\right)}{(I_x I_z - I_{xz}^2)}$	$\frac{V_0(I'_x N_\xi + I'_{xz} L_\xi)}{(I'_x I'_z - I_{xz}^2)}$
y_ζ	$\frac{\overset{\circ}{Y}_\zeta}{m}$	$\frac{V_0 Y_\zeta}{m'}$
l_ζ	$\frac{\left(I_z \overset{\circ}{L}_\zeta + I_{xz} \overset{\circ}{N}_\zeta\right)}{(I_x I_z - I_{xz}^2)}$	$\frac{V_0(I'_z L_\zeta + I'_{xz} N_\zeta)}{(I'_x I'_z - I_{xz}^2)}$
n_ζ	$\frac{\left(I_x \overset{\circ}{N}_\zeta + I_{xz} \overset{\circ}{L}_\zeta\right)}{(I_x I_z - I_{xz}^2)}$	$\frac{V_0(I'_x N_\zeta + I'_{xz} L_\zeta)}{(I'_x I'_z - I_{xz}^2)}$