

# MECH 260 - Unofficial Formula Sheet

## Average Stress and Strain

### Normal Stress and Strain

Stress  $\sigma = \frac{F}{A_{\perp}}$

Strain  $\varepsilon = \frac{\delta}{l_0}$

Young's Modulus  $E = \frac{\sigma}{\varepsilon}$

Deflection  $\delta = \frac{Fl}{AE}$

Poisson's Ratio  $\varepsilon_{x,y} = \frac{-\nu}{E} \sigma_z$

Normal Strain  $\varepsilon_{x|y|z} = \frac{1}{E} (\sigma_{\parallel} - \nu(\sigma_{\perp 1} + \sigma_{\perp 2}))$

Normal Stress  $\sigma_{x|y|z} = \left( \frac{E}{(1+\nu)(1-2\nu)} \right) ((1-\nu)\varepsilon_{\parallel} + \nu(\varepsilon_{\perp 1} + \varepsilon_{\perp 2}))$

Matrix Form 
$$\begin{bmatrix} \varepsilon_x \\ \varepsilon_y \\ \varepsilon_z \end{bmatrix} = \frac{1}{E} \begin{bmatrix} 1 & -\nu & -\nu \\ -\nu & 1 & -\nu \\ -\nu & -\nu & 1 \end{bmatrix} \begin{bmatrix} \sigma_x \\ \sigma_y \\ \sigma_z \end{bmatrix}$$

### Volumetric Stress and Strain

Volumetric Strain  $\varepsilon_V = \frac{\Delta V}{V_0} \approx \varepsilon_x + \varepsilon_y + \varepsilon_z = \left( \frac{1-2\nu}{E} \right) (\sigma_x + \sigma_y + \sigma_z)$

Bulk Modulus  $K = \frac{E}{3(1-2\nu)}$

### Shear Stress and Strain

Shear Stress  $\tau = \frac{V}{A_{\parallel}}$

Shear Strain  $\gamma = \frac{\delta}{l}$

Shear Modulus  $G = \frac{\tau}{\gamma}$

Matrix Form 
$$\begin{bmatrix} \tau_{xy} \\ \tau_{xz} \\ \tau_{yz} \end{bmatrix} = \begin{bmatrix} G & 0 & 0 \\ 0 & G & 0 \\ 0 & 0 & G \end{bmatrix} \begin{bmatrix} \gamma_{xy} \\ \gamma_{xz} \\ \gamma_{yz} \end{bmatrix}$$

### Thermal Stress and Strain

Thermal Strain  $\varepsilon = \alpha_L \Delta T$

$\varepsilon_V = \alpha_V \Delta T \approx 3\alpha_L \Delta T$

Total Strain  $\varepsilon_{x|y|z} = \varepsilon_{\text{normal}} + \alpha_L \Delta T$

Total Stress  $\sigma_{x|y|z} = \sigma_{\text{normal}} - \left( \frac{E}{1-2\nu} \right) \alpha_L \Delta T$

Volumetric  $\varepsilon_V = \left( \frac{1-2\nu}{E} \right) (\sigma_x + \sigma_y + \sigma_z) + 3\alpha_L \Delta T$

$\sigma_x + \sigma_y + \sigma_z = \left( \frac{E}{1-2\nu} \right) (\varepsilon_V - 3\alpha_L \Delta T)$

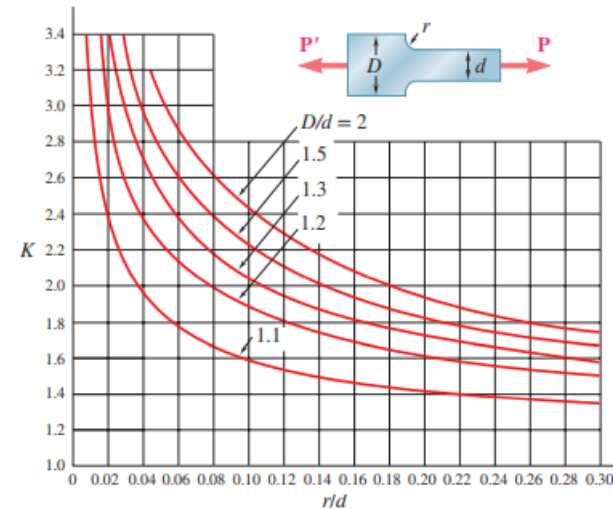
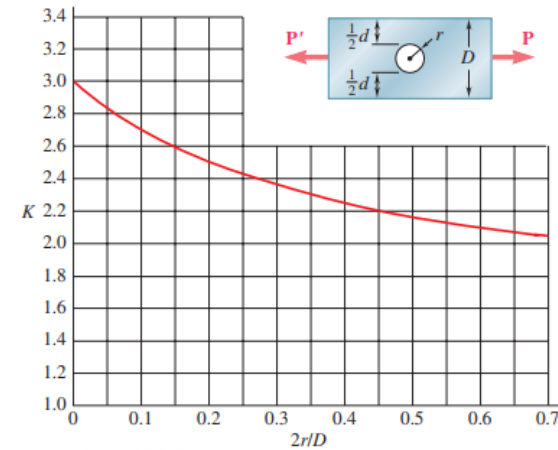
## Nonuniform Stress and Strain

### Nonuniform Stress

Safety Factor  $f_s = \begin{cases} \frac{\sigma_Y}{|\sigma_{\text{nom}}|} & \text{(ductile)} \\ \frac{\sigma_{UT}}{|\sigma_{\text{max}}|} & \text{(brittle, tension)} \\ \frac{|\sigma_{UC}|}{|\sigma_{\text{max}}|} & \text{(brittle, compression)} \end{cases}$

Nominal Stress  $\sigma_{\text{nominal}} = \frac{F_{\text{avg}}}{A_{\text{min}}}$

K Value  $K = \frac{\sigma_{\text{max}}}{\sigma_{\text{nominal}}}$



## Torsion and Bending

### Torsion

Shear Strain  $\gamma(r) = \frac{r}{l} \phi$

Shear Stress  $\frac{\tau(r)}{r} = \frac{T}{J} = \frac{G}{l} \phi$

Polar Moment of Area  $J = \begin{cases} \frac{\pi}{2} r_{\text{surf}}^4 & (\text{Solid Circular Shaft}) \\ \frac{\pi}{2} (r_{\text{surf}}^4 - r_{\text{in}}^4) & (\text{Hollow Circular Shaft}) \\ 2\pi t r_{\text{surf}}^3 & (\text{Hollow Circular, } t \ll r_{\text{surf}}) \end{cases}$

$J_{\text{rectangle}} = \frac{bh(b^2 + h^2)}{12}$

### Bending

Bending  $\frac{\sigma_x}{-y} = \frac{M}{I_z} = \frac{E}{\rho}$  (where  $\rho$  is radius of curvature)

Center of Area  $y_C^* = \frac{1}{A_{\text{total}}} \sum y_i^* A_i$

Parallel Axis Theorem  $(I_z)_O = (I_z)_C + A \Delta y^2$

Rectangular Moment of Area  $(I_z)_{\text{rectangle}} = \frac{bh^3}{12}$

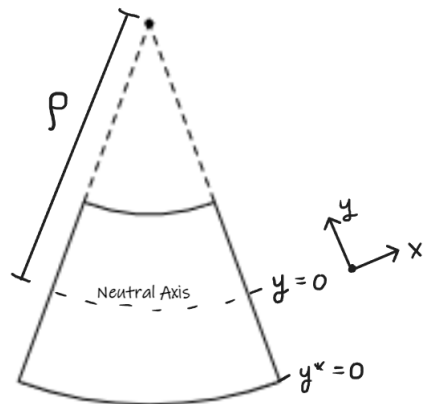
$(I_z)_{\text{circle}} = \frac{J}{2}$

Summation of  $I_z$   $I_z = \sum I_{z,i} + \sum A_i (y_{NA}^* - y_{NA,i}^*)^2$

Moment of Area  $J = I_z + I_y$

Composite Cross Sections  $b_{\text{transformed}} = b \left( \frac{E}{E_{\text{ref}}} \right)$

$\frac{\sigma_x}{-y} \left( \frac{E_{\text{ref}}}{E} \right) = \frac{M}{I_z} = \frac{E_{\text{ref}}}{\rho}$



$$y = y^* - y_{NA}^*$$

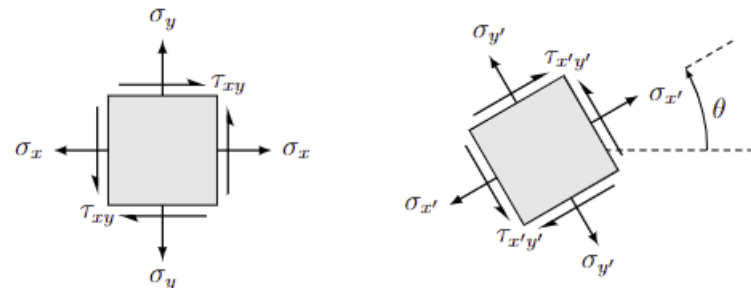
## Mohr's Circle

### Stress Transformation

$$\sigma_{x'} = \left( \frac{\sigma_x + \sigma_y}{2} \right) + \left( \frac{\sigma_x - \sigma_y}{2} \right) \cos 2\theta + \tau_{xy} \sin 2\theta$$

$$\sigma_{y'} = \left( \frac{\sigma_x + \sigma_y}{2} \right) - \left( \frac{\sigma_x - \sigma_y}{2} \right) \cos 2\theta - \tau_{xy} \sin 2\theta$$

$$\tau_{x'y'} = - \left( \frac{\sigma_x - \sigma_y}{2} \right) \sin 2\theta + \tau_{xy} \cos 2\theta$$



Note that  $\theta$  for the stress cube corresponds to  $2\theta$  in Mohr Circle space.

### Principal Planes and Max Shear Stress

Center Point  $C = \sigma_{avg} = \frac{\sigma_x + \sigma_y}{2}$

Radius  $R = \sqrt{\left( \frac{\sigma_x - \sigma_y}{2} \right)^2 + (\tau_{xy})^2}$

Principal Stresses  $\sigma_{p1,p2} = C \pm R$

Principal Angles  $2\theta_{p1} = \tan^{-1} \left( \frac{2\tau_{xy}}{\sigma_x - \sigma_y} \right)$

$$\theta_{p2} = \theta_{p1} + 90^\circ$$

Max Shear Stress  $\tau_{max} = R$

Critical Planes  $\theta_{cp1,cp2} = \theta_{p1,p2} + 45^\circ$

### Failure Criteria

Tresca  $\sigma_{\max} - \sigma_{\min} = 2\tau_{\max} = \frac{\sigma_Y}{f_s^{\text{Tresca}}}$

von Mises  $\frac{1}{2} ((\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_1 - \sigma_3)^2) = \left( \frac{\sigma_Y}{f_s^{\text{vM}}} \right)^2$

Mohr  $\frac{\sigma_{\max}}{\sigma_{UT}} - \frac{\sigma_{\min}}{|\sigma_{UC}|} = \frac{1}{f_s^{\text{Mohr}}} \text{ (for brittle materials)}$

### Thin-Walled Pressure Vessels

Sphere  $\sigma_\theta^{\text{sph}} = \frac{Pr}{2t}$

Cylinder  $\sigma_{\text{ax}} = \frac{Pr}{2t}$   $\sigma_\theta^{\text{cyl}} = \frac{Pr}{t}$