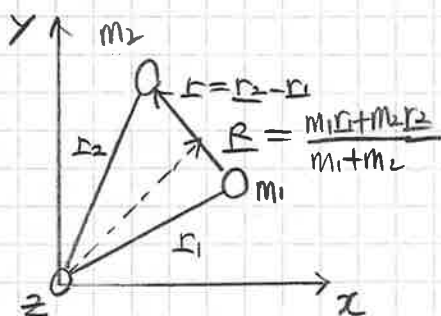


4. Central-force problem

① Central force problem and its constants of motion



$$(A) \quad L = \frac{1}{2}m_1\dot{r}_1^2 + \frac{1}{2}m_2\dot{r}_2^2 - V(|r_2 - r_1|)$$

$$\downarrow \left(R \equiv \frac{m_1 r_1 + m_2 r_2}{m_1 + m_2}, \quad r = r_2 - r_1 \right)$$

$$\boxed{L = \frac{1}{2}(m_1 + m_2)\dot{R}^2 + \frac{1}{2}\mu\dot{r}^2 - V(r)} \quad \left(\mu = \frac{m_1 m_2}{m_1 + m_2} \right)$$

(B) constants of motion

$$(i) \quad \frac{dL}{dR} = 0 \rightarrow \frac{dL}{dR} = (m_1 + m_2)\dot{R} = \text{const} \Rightarrow \boxed{L = \frac{1}{2}\mu\dot{r}^2 - V(r)} \quad (\text{one-body problem})$$

$$(ii) \quad \text{spherical symmetry: } L = \frac{1}{2}\mu(\dot{r}^2 + r^2\dot{\theta}^2) - V(r) \rightarrow \frac{dL}{d\theta} = 0 \rightarrow \frac{dL}{d\theta} = \mu r^2\dot{\theta} = l$$

$$\therefore \text{angular momentum conservation: } \boxed{l = \mu r^2\dot{\theta}}$$

$$(iii) \quad \text{time-symmetry: } h = \sum p_i \dot{q}_i - L = \frac{dL}{d\dot{\theta}}\dot{\theta} + \frac{dL}{d\dot{r}}\dot{r} - L = \frac{1}{2}\mu(\dot{r}^2 + r^2\dot{\theta}^2) + V(r) \\ = \frac{1}{2}\mu\dot{r}^2 + V(r) + \frac{l^2}{2\mu r^2} = E.$$

$$\therefore \text{energy conservation: } \boxed{E = \frac{1}{2}\mu\dot{r}^2 + V(r) + \frac{l^2}{2\mu r^2}} \quad (V_{\text{eff}} = V(r) + \frac{l^2}{2\mu r^2})$$

② Formal solution for the orbit geometry (dependence of r on θ)

$$l = \mu r^2\dot{\theta} \rightarrow dt = \frac{\mu r^2}{l} d\theta = \frac{\mu}{lu^2} d\theta$$

$$E = \frac{1}{2}\mu\dot{r}^2 + V(r) + \frac{l^2}{2\mu r^2} = \frac{l^2}{2\mu} \left(\frac{du}{d\theta} \right)^2 + V(u^{-1}) + \frac{l^2}{2\mu r^2}$$

$$\left(\frac{dr}{dt} = \left(-\frac{1}{u^2} \right) \frac{du}{dt} = \left(-\frac{1}{u^2} \right) \frac{lu^2 du}{\mu d\theta} = -\frac{l}{\mu} \frac{du}{d\theta} \right)$$

$$\boxed{\left(\frac{du}{d\theta} \right)^2 = \frac{2\mu}{l^2} [E - V(u^{-1})] - u^2}$$

$$\theta = \theta^* : \frac{du}{d\theta^*} = 0 : \text{apsis} \quad \begin{cases} \text{periapsis (근점)} \\ \text{apoapsis (원점)} \end{cases}$$

$$(r=r_{\min}, \theta=0) \rightarrow (r=r_{\max}, \theta=\theta'_n)$$

$$\frac{du}{d\theta} = - \left\{ \frac{2\mu}{l^2} [E - V(u^{-1})] - u^2 \right\}^{1/2}$$

$$\Rightarrow \theta = - \int_{r_{\min}^{-1}}^{r^{-1}} \frac{du}{\left\{ \frac{2\mu}{l^2} [E - V(u^{-1})] - u^2 \right\}^{1/2}}$$

③ Kepler problem

(A) orbit geometry. ($V(r) = -\frac{k}{r} = -ku$)

$$\theta = - \int_{r_{\min}}^{r^{-1}} \frac{dn}{\sqrt{\frac{2\mu E}{l^2} + \frac{2\mu k}{l^2} u - u^2}} = \arccos \frac{2u - \frac{2\mu k}{l^2}}{\sqrt{\frac{4\mu^2 k^2}{l^4} + \frac{8\mu E}{l^2}}} \Bigg|_{u=r_{\min}^{-1}}^{u=r^{-1}}$$

$$\int \frac{dx}{\sqrt{\alpha + \beta x - x^2}} = -\arccos \frac{2x - \beta}{\sqrt{\beta^2 + 4\alpha}} + \text{const}$$

$$\theta = \arccos \frac{\frac{l^2}{\mu k r} - 1}{\sqrt{1 + \frac{2El^2}{\mu k^2}}}$$

($\because \theta' = 0$ at $r = r_{\min}$) \leftarrow 우리가 이렇게 정의함.
 $\hookrightarrow \cos \theta' = \cos 0 = 1$

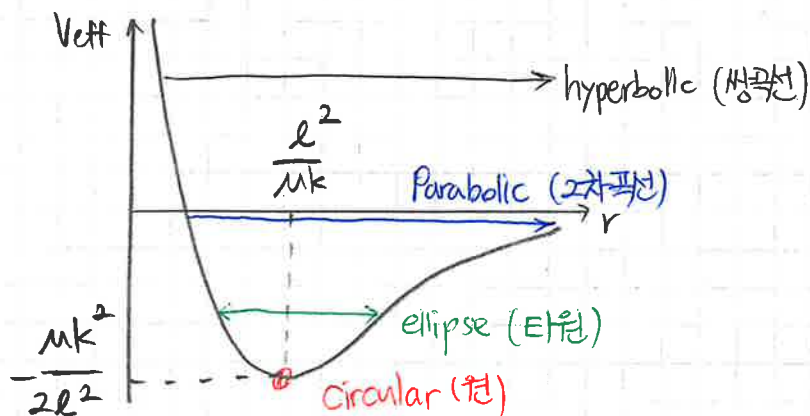
Define $e = \sqrt{1 + \frac{2El^2}{\mu k^2}}$

eccentricity (이심률)

$$e \cos \theta + 1 = \frac{l^2}{\mu k r}$$

$$r = \frac{l^2}{\mu k} (1 + e \cos \theta)$$

• 4 categories ($V_{\text{eff}} = -\frac{k}{r} + \frac{l^2}{2\mu r^2}$, $V_{\text{eff}}' = +\frac{k}{r^2} - \frac{l^2}{\mu r^3}$)



(i) Hyperbolic (쌍곡선) ($E > 0, e > 1$)

$$r = \frac{l^2}{\mu k} \frac{1}{(1 + e \cos \theta)} = \frac{\mu k^2}{2E l^2} \cdot \frac{l^2 (e^2 - 1)}{\mu k (1 + e \cos \theta)} = \frac{k}{2E} \frac{e^2 - 1}{1 + e \cos \theta}$$

$$(\text{semi-major axis: } a \equiv \frac{k}{2E}) = \frac{a(e^2 - 1)}{1 + e \cos \theta} > r_{\min} = a(e - 1)$$

$$\therefore r = \frac{a(e^2 - 1)}{1 + e \cos \theta} > r_{\min} = a(e - 1)$$

$$\text{where } |\cos \theta| < e^{-1} \leftrightarrow |\theta| < \cos^{-1} e^{-1}$$

(ii) Parabolic (2차곡선) ($E = 0, e = 1$)

$$r = \frac{l^2}{\mu k} \frac{1}{(1 + e \cos \theta)} = \frac{l^2}{\mu k} \frac{1}{1 + \cos \theta} > r_{\min} = \frac{l^2}{2\mu k}$$

$$r_{\min} = \frac{l^2}{2\mu k} \rightarrow r = \frac{2r_{\min}}{1 + \cos \theta}$$

(iii) Ellipse (타원) ($-\frac{\mu k^2}{2l^2} < E < 0, 0 < e < 1$)

$$r = \frac{l^2}{\mu k} \frac{1}{1 + e \cos \theta} = \frac{k}{2E} \frac{e^2 - 1}{1 + e \cos \theta} = -\frac{k}{2E} \frac{1 - e^2}{1 + e \cos \theta}$$

$$(\text{semi-major axis: } a \equiv -\frac{k}{2E}) = \frac{a(1 - e^2)}{1 + e \cos \theta} \rightarrow r = \frac{a(1 - e^2)}{1 + e \cos \theta}$$

$$r = \frac{a(1 - e^2)}{1 + e \cos \theta}, \quad r_{\min} = a(1 - e) < r < r_{\max} = a(1 + e)$$

(iv) circular (원) ($E = -\frac{\mu k^2}{2l^2}, e = 0$)

$$\therefore r = \frac{l^2}{\mu k}$$

(B) Kepler's Law of planetary motion

- { ① 행성은 타원운동
 ② 행성과 태양이 휩쓰는 단위시간 당 면적은 항상 같다.
 ③ 행성주기 (T) 와 단반경(a)는 $T^2 \propto a^3$ 관계가 있다. }

proof) ②

$\Delta A \approx \frac{1}{2} r^2 \Delta \theta$
 $\frac{dA}{dt} = \frac{1}{2} r^2 \dot{\theta} = \frac{l}{2\mu} = \text{const}$

② $A = \pi a^2 \sqrt{1-e^2} = \pi a^2 \sqrt{-\frac{2El^2}{\mu k^2}}$
 $(a = -\frac{\mu}{2E})$
 $= \pi a^2 \sqrt{\frac{l^2}{a\mu k}} = \frac{\pi a^{3/2} l}{\sqrt{\mu k}}$
 $(e = \sqrt{1 + \frac{2El^2}{\mu k^2}})$

$T = \frac{A}{dA/dt} = \frac{\pi a^{3/2} l}{\sqrt{\mu k}} \cdot \frac{2\mu}{l} = 2\pi a^{3/2} \sqrt{\frac{\mu}{k}} = \frac{2\pi a^{3/2}}{\sqrt{G(M_1+M_2)}} \approx \frac{2\pi a^{3/2}}{\sqrt{GM_0}}$

(c) Laplace-Runge-Lenz vector

$$\frac{1}{2} \frac{d}{dt} (r \cdot r) = r \dot{r}$$

(1) conservation law

$\dot{\mathbf{p}} = -\frac{k}{r^3} \mathbf{r} \rightarrow \dot{\mathbf{p}} \times \mathbf{L} = -\frac{\mu k}{r^3} \mathbf{r} \times (\mathbf{r} \times \dot{\mathbf{r}}) = -\frac{\mu k}{r^3} [\mathbf{r} (\mathbf{r} \cdot \dot{\mathbf{r}}) - r^2 \dot{\mathbf{r}}]$
 $= \mu k \left(-\frac{\dot{r}}{r^2} \mathbf{r} + \frac{1}{r} \dot{\mathbf{r}} \right) = \frac{d}{dt} (\mu k \hat{\mathbf{r}})$

$\frac{dA}{dt} = \frac{d}{dt} (\mathbf{p} \times \mathbf{L} - \mu k \hat{\mathbf{r}}) = 0 \rightarrow \boxed{A = \mathbf{p} \times \mathbf{L} - \mu k \hat{\mathbf{r}}}$ is conserved
 (Laplace-Runge-Lenz vector)

(2) Derivation of orbit equation

$$\underline{L} = L \hat{z} \rightarrow \underline{L} \cdot \underline{A} = 0 \rightarrow \underline{A} \text{ is in } x-y \text{ plane.}$$

$$\underline{r} \cdot \underline{A} = \underline{r} \cdot (\underline{p} \times \underline{L}) - \mu k r = \underline{L} \cdot (\underline{r} \times \underline{p}) - \mu k r = L^2 - \mu k r$$

$$= A r \cos \theta$$

$$\therefore (A \cos \theta + \mu k) r = L^2 \quad \therefore r = \frac{L^2}{\mu k + A \cos \theta} \geq r_{\min} = \frac{L^2}{\mu k + A}$$

$$\text{Since } E = -\frac{k}{r_{\min}} + \frac{L^2}{2\mu r_{\min}^2} = \frac{A^2 - \mu^2 k^2}{2\mu L^2} \rightarrow A = \mu k \sqrt{1 + \frac{2EL^2}{\mu k^2}} = \mu k e.$$

$$\therefore r = \frac{L^2}{\mu k} \frac{1}{1 + e \cos \theta}$$

⊕ Conditions for closed orbits

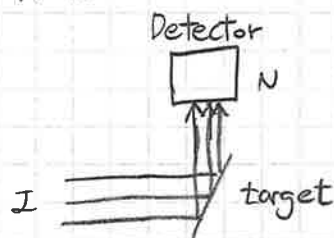
모든 bound orbit 이 닫힌 orbit 이 되기 위해서는, 아래 즉 종류의 퍼텐셜만 가능하다.

$$(1) \underline{V(r) = -k/r}, \quad (2) \underline{V(r) = \frac{1}{2}kr^2}$$

⑤ Scattering in a central-force field

(A) Scattering cross-section

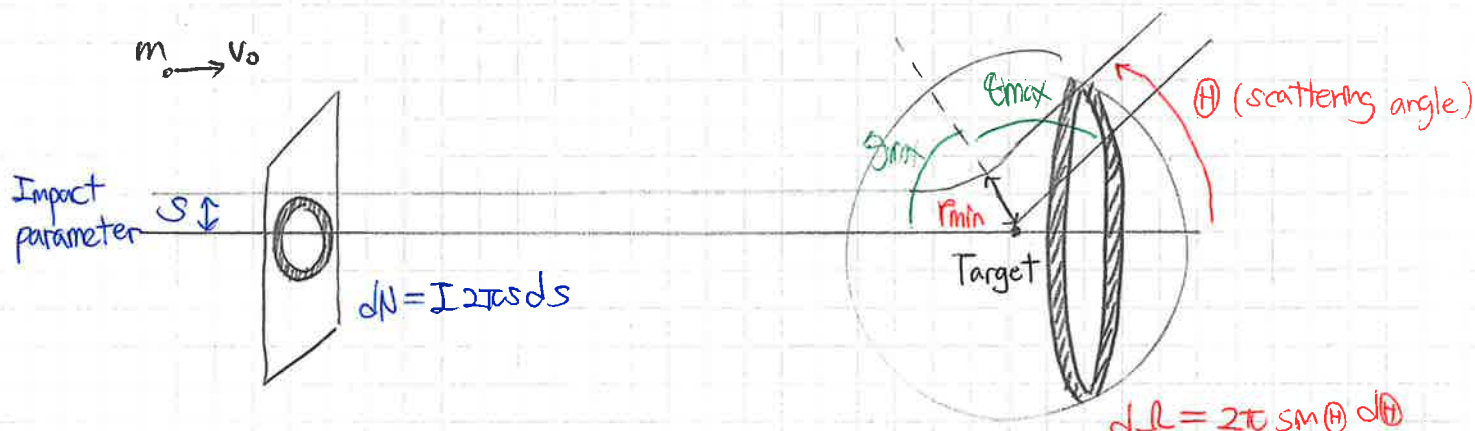
$$\sigma \equiv N/I \quad (\text{area})$$



I [unit area · time]

N [unit · time]

(b) Differential scattering cross-section in a central-force field.



$$dN = I d\sigma = I \frac{d\sigma}{d\Omega} d\Omega \Rightarrow \frac{d\sigma}{d\Omega} = \frac{dN}{I d\Omega} = \frac{2\pi s ds}{2\pi s \sin \Theta d\Theta} \quad \therefore \frac{d\sigma}{d\Omega} = \frac{s}{\sin \Theta} \left| \frac{ds}{d\Theta} \right|$$

(c) Rutherford Scattering

$$\Theta = - \int_{r_{\min}}^{r-1} \frac{du}{\sqrt{\frac{2uE}{l^2} - \frac{2uk}{l^2} - u^2}} = \arccos \left. \frac{2u + \frac{2uk}{l^2}}{\sqrt{\frac{4u^2 k^2}{l^4} + \frac{8uE}{l^2}}} \right|_{u=r_{\min}^{-1}}^{u=r-1}$$

$$\Theta = \arccos \frac{\frac{l^2}{mk r} + 1}{\sqrt{1 + \frac{2El^2}{mk^2}}} \quad \rightarrow \quad 1 + \frac{l^2}{mk r} = e \cos \Theta \quad (u=m) \quad \rightarrow \quad r = \frac{l^2}{mk} \frac{1}{e \cos \Theta - 1}$$

since $E > 0, e > 1$, $|\Theta| < \Theta_{\max} = \arccos e^{-1}$ ($\cos \Theta_{\max} = \frac{1}{e}$)

$$\Theta = \pi - 2\Theta_{\max}$$

$$i) \sin \frac{\theta}{2} = \sin \left(\frac{\pi}{2} - \theta_{\max} \right) = \cos \theta_{\max} = \frac{1}{e}$$

$$ii) \cot^2 \frac{\theta}{2} = \csc^2 \frac{\theta}{2} - 1 = e^2 - 1 = \frac{2El^2}{mk^2} = \frac{2Es^2 2mE}{mk^2} \rightarrow \cot \frac{\theta}{2} = \frac{2Es}{k}$$

$$l = smv_0 = s\sqrt{2mE}$$

$$iii) -\csc^2 \frac{\theta}{2} \cdot \frac{d\theta}{2} = \frac{2E}{k} ds \rightarrow \frac{ds}{d\theta} = -\frac{k}{4E} \csc^2 \frac{\theta}{2}$$

$$iv) \sin \theta = 2 \sin \frac{\theta}{2} \cos \frac{\theta}{2}$$

$$\therefore \frac{ds}{d\theta} = \frac{s}{\sin \theta} \left| \frac{ds}{d\theta} \right| = \left(\frac{k}{2E} \cot \frac{\theta}{2} \right) \cdot \left(\frac{1}{2 \sin \frac{\theta}{2} \cos \frac{\theta}{2}} \right) \cdot \left(\frac{k}{4E} \csc^2 \frac{\theta}{2} \right)$$

$$= \frac{k^2}{16E^2} \csc^4 \frac{\theta}{2}$$

$$\therefore \frac{ds}{d\theta} = \frac{s}{\sin \theta} \left| \frac{ds}{d\theta} \right| = \frac{k^2}{16E^2} \csc^4 \frac{\theta}{2}$$