

Question 2

$$M(\dot{q})\ddot{q} + C(q, \dot{q})\dot{q} + D\dot{q} + g(q) = u$$

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$$e = q - q_r \quad \dot{e} = \dot{q} \quad \ddot{e} = \ddot{q}$$

q_r : constant reference trajectory

$$V_e = \frac{1}{2} e^T e + \frac{1}{2} \dot{e}^T \dot{e}$$

$$\begin{aligned} \dot{V}_e &= \dot{e}^T e + \dot{e}^T \dot{e} \\ &= \dot{e}^T e + \dot{e}^T \ddot{e} \\ &= \dot{e}^T [e + \ddot{e}] \end{aligned} \quad \therefore \ddot{e}^T \dot{e} = \dot{e}^T \ddot{e} \quad (\text{scalar})$$

$$e_1 = e \quad e_2 = \dot{e}$$

$$\dot{e}_1 = e_2$$

$$\dot{e}_2 = +[H(q)^{-1} \left[-C\dot{q} - D\ddot{q} - g + u \right]]$$

$$\dot{V}_e = \dot{q}^T \left[q - q_r + M^{-1} \underbrace{\left[u - C\dot{q} - D\ddot{q} - g \right]}_{-q} \right]$$

if

$$K_0 = u = M \left[-\dot{q} - (q - q_r) \right] + C\dot{q} + D\ddot{q} + g$$

$$\dot{V}_e = -\dot{q}^T \dot{q} < 0$$

now for the system with the integrator:

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$$\ddot{\hat{q}} = H(q)^{-1} \left[-C\dot{q} - D\ddot{q} - g + \xi \right]$$

$$\xi = \tau$$

$$\text{Let } V_1 = V_e + \frac{1}{2} [\xi - K_0]^T [\xi - K_0]$$

$$\dot{V}_1 = \dot{V}_e + [\xi - K_0]^T [\dot{\xi} - \dot{K}]$$

$$\dot{V}_1 = \dot{q}^T \left[(q - q_r) + H^{-1} [\xi - C\dot{q} - D\ddot{q} - g] \right] + \dot{q}^T \dot{q} - \dot{q}^T \dot{q}$$

$$+ [\xi - \cancel{H} \left[-\dot{q} - (q - q_r) \right] + C\dot{q} + D\ddot{q} + g] [\tau - \cancel{K}]$$

$$\ddot{V}_1 = -\ddot{q}^T \dot{q} + \dot{q}^T [(\ddot{q} - \ddot{q}_r) + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g) + \ddot{q}] \\ + [\underbrace{e_g - H(-\dot{q} - (q - q_r))}_{\text{忽略}} - \underbrace{(C\ddot{q} + D\dot{q} + g)}_{\text{忽略}}]^T [T - \dot{k}]$$

~~$$\ddot{V}_1 = -\ddot{q}^T \dot{q} + \dot{q}^T [(\ddot{q} - \ddot{q}_r) + H^{-1}[H\ddot{q}]] + \ddot{q} \\ + [H[\dot{q} + (q - q_r)] + H\ddot{q}]$$~~

$$\ddot{V} = -\ddot{q}^T \dot{q} + \dot{q}^T [(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)] \\ + [H[(q - q_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)]]^T [T - \dot{k}]$$

$$\ddot{V} = -\ddot{q}^T \dot{q} + [(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)] \underbrace{[\dot{q} + H^T T - H^T \dot{k}]}$$

$$\dot{q} + H^T T - H^T \dot{k} = -[(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)]$$

$$T = H^{-1} [H^T \dot{k} - \dot{q} - [(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)]]$$

$$T = \dot{k} + (H^T)^{-1} \left[-(\ddot{q} - \ddot{q}_r) - 2\dot{q} - H^{-1}(e_g - C\ddot{q} - D\dot{q} - g) \right]$$

Input from backstepping

$$\ddot{V} < 0$$

$$\ddot{V} = -\ddot{q}^T \dot{q} - [(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)]^T \\ [(\ddot{q} - \ddot{q}_r) + \dot{q} + H^{-1}(e_g - C\ddot{q} - D\dot{q} - g)]$$

Since g can be neglected.

$$K = M[-\ddot{q} - (q - q_r)] + C\dot{q} + D\ddot{q} + g \text{? neglected}$$

$$\ddot{K} = M[-\ddot{q} - (q - q_r)] + C\ddot{q} + D\ddot{\dot{q}} \\ + M[-\ddot{\dot{q}} - \ddot{q}] + \ddot{C}\dot{q}$$

Question 1: Robotic Manioulator using passivity based control

```
import numpy as np
from math import sin, cos, tanh
from numpy.linalg import inv
import matplotlib.pyplot as plt
# import scipy.integrate as integrate
import matplotlib.animation as animation

class RoboticManipulator():
    def __init__(self,
                 q1_ref_0 = 0 , 
                 q2_ref_0 = 0 , 
                 q1_0      = 0 , 
                 q2_0      = 0 , 
                 q1dot_0   = 0 , 
                 q2dot_0   = 0 , 
                 q1dot2_0  = 0 , 
                 q2dot2_0  = 0 ):
        """ CONSTANTS
        self.p1 = 3.31      #kg.m^2
        self.p2 = 0.116
        self.p3 = 0.16

        self.L1 = 0.8       #m
        self.L2 = 0.5
        self.01 = -0.666   #m
        self.02 = 0.333

        self.dt = 0.05      #s

        """ TUNING CONSTANTS
        self.k = 1           # control gain
        self.Kp = np.array([[1,0],
                           [0,1]])
        self.D = np.array([[1,0],
                          [0,1]])

    """ VARIABLES
    self.q1      = q1_0
    self.q2      = q2_0
    self.q1dot   = q1dot_0
    self.q2dot   = q2dot_0
    self.q1dot2 = q1dot2_0
    self.q2dot2 = q2dot2_0

    self.x1 = self.L1*cos(self.q1) + self.L2*cos(self.q1 + self.q2) + self.01
    self.x2 = self.L1*sin(self.q1) + self.L2*sin(self.q1 + self.q2) + self.02

    self.x21 = self.L1*cos(self.q1) + self.01
    self.x22 = self.L1*sin(self.q1) + self.02

    """ REFERNCE INPUT
    self.q1_ref = q1_ref_0
    self.q2_ref = q2_ref_0
```



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self.x1_ref = self.L1*cos(self.q1_ref) + self.L2*cos(self.q1_ref + self.q2_ref) + self.01
self.x2_ref = self.L1*sin(self.q1_ref) + self.L2*sin(self.q1_ref + self.q2_ref) + self.02

self.x21_ref = self.L1*cos(self.q1_ref) + self.01
self.x22_ref = self.L1*sin(self.q2_ref) + self.02

self.update_state_matrices()

def update_state_matrices(self):
    self.M = np.array([ [ self.p1 + 2*self.p3*cos(self.q2) ,   self.p2 + self.p3*cos(self.q2) ] ,
                        [ self.p2 + self.p3*cos(self.q2) ,   self.p2 ] ] )

    self.C = np.array([ [ -self.q2dot ,  self.q1dot + self.q2dot ] ,
                        [ self.q1dot ,  0 ] ] ) *self.p3*sin(self.q2)

    self.g = np.array([ [ 0 ] ,
                        [ 0 ] ] )

    self.Omega = np.array([ [ self.x1 ] ,
                           [ self.x2 ] ] )

    self.omega2 = np.array([ [ self.x21 ] ,
                           [ self.x22 ] ] )

    self.q_ref = np.array([ [ self.q1_ref ] ,
                           [ self.q2_ref ] ] )

    self.x_ref = np.array([[self.x1_ref],
                          [self.x2_ref]])

    self.q = np.array([ [ self.q1 ] ,
                       [ self.q2 ] ] )

    self.qdot = np.array([ [ self.q1dot ] ,
                          [ self.q2dot ] ] )

    self.qdot2 = np.array([ [ self.q1dot2 ] ,
                           [ self.q2dot2 ] ] )

    self.E = np.copy(self.q - self.q_ref)

    self.Edot = np.copy(self.qdot)

    self.Edot2 = np.copy(self.qdot2)

    self.v_input = -self.k* np.array([ [ tanh(self.Edot[0][0]) ] ,
                                       [ tanh(self.Edot[1][0]) ] ] )

    self.U = self.g -np.dot(self.Kp,self.E) + self.v_input

def advance_and_update_states(self):

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        self.qdot2 = np.dot(inv(self.M), -np.dot(self.Kp, self.E) - np.dot(self.D, self.Edot)
- np.dot(self.C, self.Edot) + self.v_input)
        self.qdot = self.qdot + self.qdot2 * self.dt
        self.q = self.q + self.qdot * self.dt

        self.q1,      self.q2      = self.q[0][0]      , self.q[1][0]
        self.q1dot,   self.q2dot = self.qdot[0][0] , self.q[1][0]
        self.q1dot2, self.q2     = self.qdot2[0][0], self.q[1][0]

        self.x1 = self.L1*cos(self.q1) + self.L2*cos(self.q1 + self.q2) + self.01
        self.x2 = self.L1*sin(self.q1) + self.L2*sin(self.q1 + self.q2) + self.02

        self.x21 = self.L1*cos(self.q1) + self.01
        self.x22 = self.L1*sin(self.q1) + self.02

        self.update_state_matrices()

def simulate(self, t_f):
    self.t = np.linspace(0,t_f,int(t_f/self.dt))
    self.u_array      = []
    self.u_1_array    = []
    self.u_2_array    = []

    self.v_array      = []
    self.v_1_array    = []
    self.v_2_array    = []

    self.q_array      = []
    self.q_r_array    = []
    self.q_1_array    = []
    self.q_1_r_array = []
    self.q_2_array    = []
    self.q_2_r_array = []

    self.x_array      = []
    self.x_r_array    = []
    self.x_1_array    = []
    self.x_2_array    = []
    self.x_1_r_array = []
    self.x_2_r_array = []

    self.x02_1_array = []
    self.x02_2_array = []
    self.x02_1_r_array= []
    self.x02_2_r_array= []

    for _ in range(len(self.t)):
        self.u_array.append(np.linalg.norm(self.U))
        self.u_1_array.append(self.U[0])
        self.u_2_array.append(self.U[1])

        self.v_array.append(np.linalg.norm(self.v_input))
        self.v_1_array.append(self.v_input[0])
        self.v_2_array.append(self.v_input[1])

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        self.q_array.append(np.linalg.norm(self.q))
        self.q_r_array.append(np.linalg.norm(self.q_ref))
        self.q_1_array.append(self.q[0])
        self.q_2_array.append(self.q[1])
        self.q_1_r_array.append(self.q1_ref)
        self.q_2_r_array.append(self.q2_ref)

        self.x_r_array.append(self.x_ref)
        self.x_1_r_array.append(self.x1_ref)
        self.x_2_r_array.append(self.x2_ref)
        self.x_array.append(self.Omega)
        self.x_1_array.append(self.x1)
        self.x_2_array.append(self.x2)

        self.x02_1_array.append(self.x21)
        self.x02_2_array.append(self.x22)
        self.x02_1_r_array.append(self.x21_ref)
        self.x02_2_r_array.append(self.x22_ref)

    self.advance_and_update_states()

def plot(self):
    fig, ax = plt.subplots(1, 4)

    # ax[0].plot(self.t, self.u_array, '-c')
    ax[0].plot(self.t, self.u_1_array, '-g')
    ax[0].plot(self.t, self.u_2_array, '-r')

    ax[0].legend(["u1", "u2"])
    ax[0].set_ylabel("u")
    ax[0].set_xlabel("t")
    ax[0].grid()
    ax[0].set_title("Control input u v/s iteration t")

    # ax[1].plot(self.t, self.q_array, '-b')
    ax[1].plot(self.t, self.q_1_array, '-g')
    ax[1].plot(self.t, self.q_2_array, '-r')
    ax[1].plot(self.t, self.q_1_r_array, ':b')
    ax[1].plot(self.t, self.q_2_r_array, ':k')

    ax[1].legend(["q1", "q2", "q1_ref", "q2_ref"])
    ax[1].set_ylabel("q")
    ax[1].set_xlabel("t")
    ax[1].grid()
    ax[1].set_title("Angle q vs iteration t")

    # ax[2].plot(self.t, self.x_array, '-b')
    ax[2].plot(self.t, self.x_1_array, '-g')
    ax[2].plot(self.t, self.x_2_array, '-r')
    ax[2].plot(self.t, self.x_1_r_array, ':b')
    ax[2].plot(self.t, self.x_2_r_array, ':k')

    ax[2].legend(["x1", "x2", "x1_ref", "x2_ref"])
    ax[2].set_ylabel("x")

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ax[2].set_xlabel("t")
ax[2].grid()
ax[2].set_title("Cartesian coordinate x vs iteration t")

# ax[3].plot(self.t, self.v_array, '-b')
ax[3].plot(self.t, self.v_1_array, '-g')
ax[3].plot(self.t, self.v_2_array, '-r')

ax[3].legend(["v1", "v2"])
ax[3].set_ylabel("v")
ax[3].set_xlabel("t")
ax[3].grid()
ax[3].set_title("Passivity control input of v vs iteration t")

# plt.show()

fig, ax = plt.subplots(1, 2)

ax[0].plot(self.x_1_array, self.x_2_array, '-g')
ax[0].plot(self.x_1_r_array, self.x_2_r_array, 'X-b')

ax[0].legend([(x1, x2), "(x1_ref, x2_ref)"])
ax[0].set_ylabel("x2")
ax[0].set_xlabel("x1")
ax[0].grid()
ax[0].set_title("Cartesian coordinate x2 vs x1")

ax[1].plot(self.q_1_array, self.q_2_array, '-g')
ax[1].plot(self.q_1_r_array, self.q_2_r_array, 'X-b')
ax[1].legend([(q1, q2), "(q1_ref, q2_ref)"])
ax[1].set_ylabel("q2")
ax[1].set_xlabel("q1")
ax[1].grid()
ax[1].set_title("Angle q2 vs q1")

# plt.show()

def animate_this(self):
    fig = plt.figure()
    ax = fig.add_subplot(111, autoscale_on=False, xlim=(-2, 2), ylim=(-2, 2))
    ax.set_aspect('equal')
    ax.grid()

    line, = ax.plot([], [], 'o-', lw=2)
    time_template = 'time = %.1fs'
    time_text = ax.text(0.05, 0.9, '', transform=ax.transAxes)
    def init():
        line.set_data([], [])
        time_text.set_text('')
        return line, time_text
    def animate(i):
        thisx = [self.o1, self.x02_1_array[i], self.x_1_array[i]]
        thisy = [self.o2, self.x02_2_array[i], self.x_2_array[i]]

        line.set_data(thisx, thisy)
        time_text.set_text(time_template % i)
        return line, time_text

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        time_text.set_text(time_template % (i*self.dt))
        return line, time_text

    ani = animation.FuncAnimation(fig, animate, np.arange(1, len(self.x_1_array)),
                                 interval=25, blit=True, init_func=init)

    plt.show()

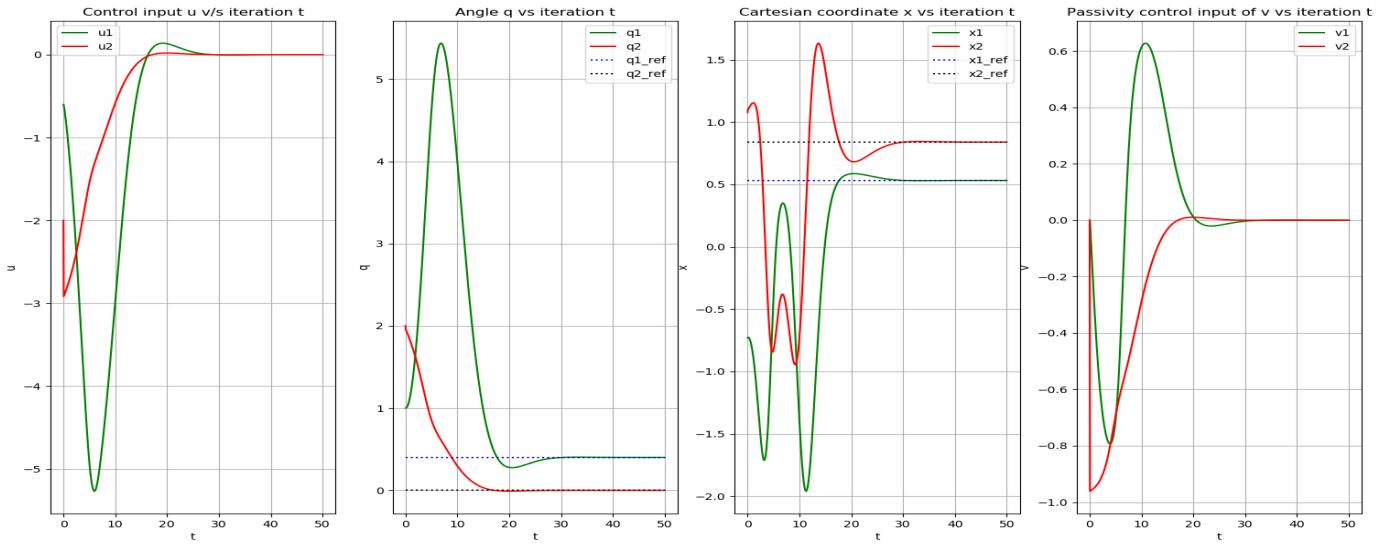
if __name__ == '__main__':
    q_ref = np.array([[3],
                     [0 ]])
    q_0    = np.array([[3],
                     [1]])
    simulation_time = 50

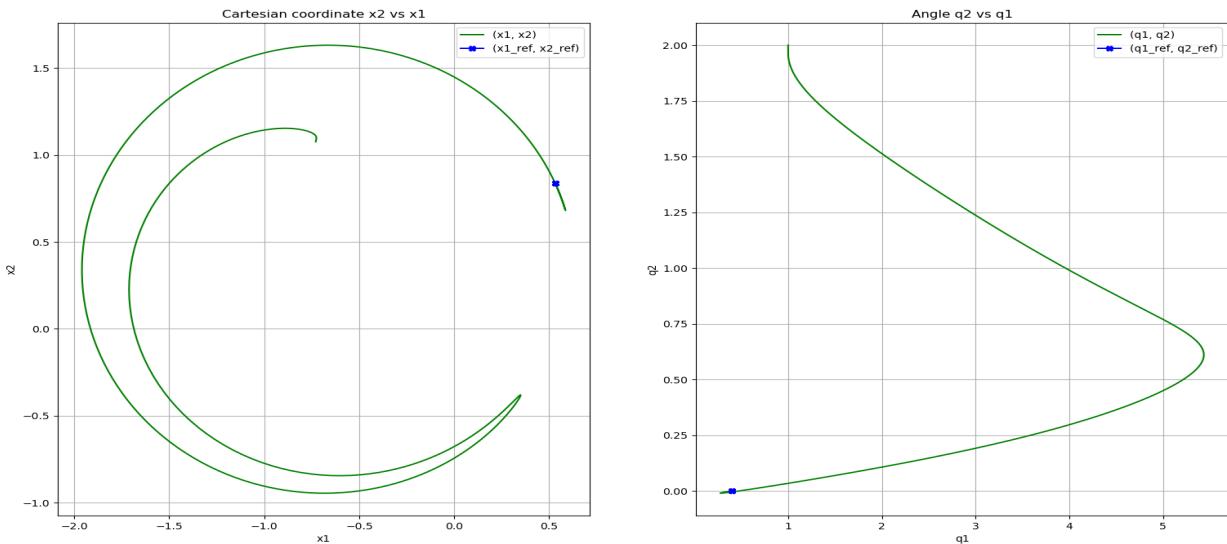
    tlm1 = RoboticManipulator(q_ref[0][0],
                               q_ref[1][0],
                               q_0[0][0] ,
                               q_0[1][0] )
    tlm1.simulate(simulation_time)

    tlm1.plot()
    tlm1.animate_this()

```

PLOTS:





Question 2: Robotic Manipulator using Backstepping

```
import numpy as np
import matplotlib.pyplot as plt

class manipulator:
    def __init__(self, p_1, p_3, q_0, q_r):
        # system states
        self.q = q_0
        self.q_r = q_r
        self.e = self.q - self.q_r
        self.q_dot = 0.
        self.q_dot_dot = 0.
        self.xi = 0.
        self.xi_dot = 0

        # constant values
        self.p_1 = p_1
        self.p_3 = p_3
        self.dt = 0.1

    def update_mc():
        self.m = self.p_1 + (2 * self.p_3 * np.cos(self.q))
        self.c = -self.p_3 * np.sin(self.q) * self.q_dot
        self.m_dot = -2 * self.p_3 * np.sin(self.q) * self.q_dot
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        self.c_dot = -(self.p_3 * np.cos(self.q) * (self.q_dot**2)) - (self.p_3 *
np.sin(self.q) * self.q_dot_dot)

def advance_state(self, tau):
    dt = self.dt

    # dynamics
    self.q_dot_dot = (self.xi - (self.c * self.q_dot)) / self.m
    self.xi_dot = tau

    # update states
    self.q_dot += self.q_dot_dot * dt
    self.q += self.q_dot * dt
    self.xi += self.xi_dot * dt
    self.update_mc()

def compute_control_law(self): # taking the control lyapunov function to be
V(q,q_dot) = e^2/2 + q_dot^2/2
    # k_0 = Cq_dot + mq_r - mq - mq_dot
    k_0_dot = (self.c_dot * self.q_dot + self.c * self.q_dot_dot) + self.m_dot *
self.q_r - (self.m_dot*self.q + self.m*self.q_dot) - (self.m_dot*self.q_dot +
self.m*self.q_dot_dot)
    self.tau = k_0_dot - (2*self.q_dot/self.m) - (self.xi/(self.m**2)) +
(self.c*self.q_dot/(self.m**2)) + ((self.q_r - self.q)/self.m)

def simulate(self, t_f) :
    self.t = np.linspace(0,t_f,int(t_f/self.dt))
    self.compute_control_law()
    self.tau_array = []
    self.q_array = []
    self.q_r_array = []
    for _ in range (len(self.t)):
        self.tau_array.append(self.tau)
        self.q_array.append(self.q)
        self.q_r_array.append(self.q_r)
        self.advance_state(self.tau)
        self.compute_control_law()

def plot(self):

    fig, ax = plt.subplots(1, 2)

    ax[0].plot(self.t, self.tau_array, '-g')

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# ax[0].plot(self.t, self.tau_array, '-r')

ax[0].legend(["tau"])
ax[0].set_ylabel("tau")
ax[0].set_xlabel("t")
ax[0].grid()
ax[0].set_title("Control input tau v/s iteration t")

ax[1].plot(self.t, self.q_array, '-g')
ax[1].plot(self.t, self.q_r_array, ':b')
ax[1].legend(["q", "q_ref"])
ax[1].set_ylabel("q")
ax[1].set_xlabel("t")
ax[1].grid()
ax[1].set_title("Control input q v/s iteration t")

plt.show()

if __name__ == '__main__':
    p_1 = 3.31
    p_3 = 0.16

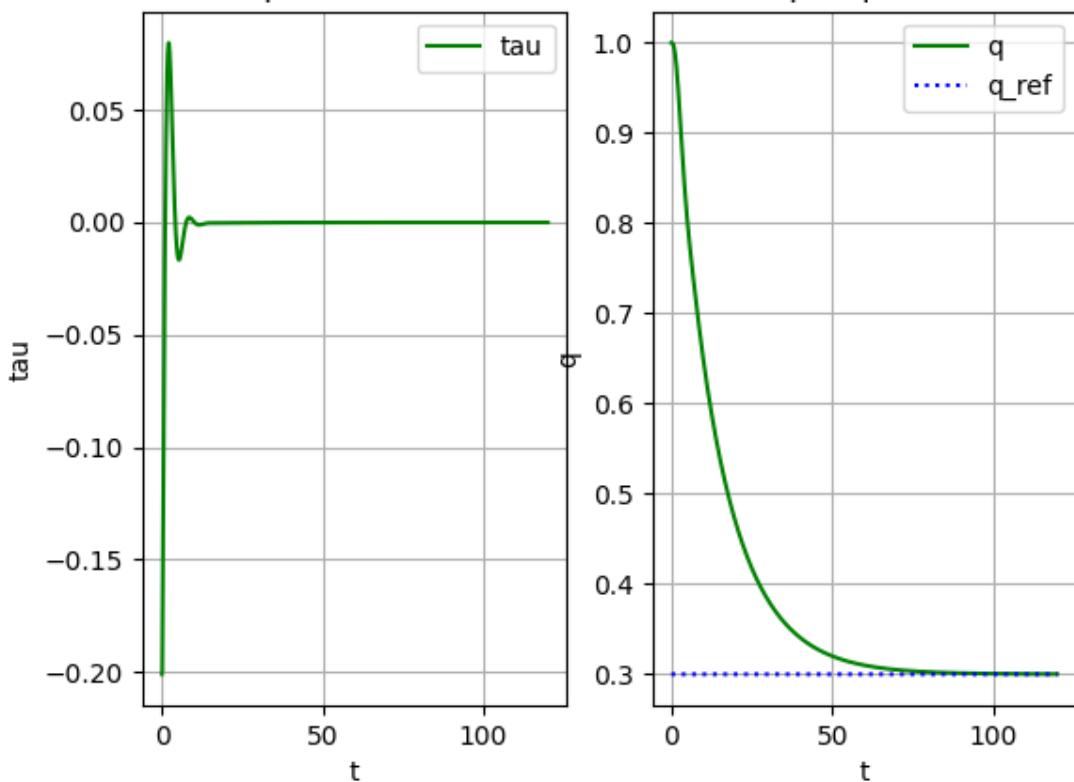
    q_r = 0.3
    q_0 = 1.

    m = manipulator(p_1, p_3, q_0, q_r)
    m.simulate(120)
    m.plot()

```

PLOTS:

Control input tau v/s iteration t Control input q v/s iteration t



Question 3: Rigid Body controller using passivity

```
import numpy as np
import matplotlib.pyplot as plt
from math import sqrt, tanh, tan

class rigid_body:
    def __init__(self, J, rho_0, omega_0):
        self.J = J
        self.rho = rho_0
        print(self.rho)
        self.rho_cross = self.get_curl(self.rho)
        self.omega = omega_0
        self.I = np.identity(3)
        self.dt = 0.05
        self.k_1 = 1 # storage function gain
        self.k_2 = 12 # control law gain
        self.compute_dW_drho()

    def get_curl(self, vec):
        curl = np.array([[0 , -vec[2], vec[1] ],
                        [vec[2] , 0 , -vec[0]]])
```

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[-vec[1], vec[0] , 0      ])
return curl

# now compute the jacobian of the storage function to compute the control law
def compute_dW_drho(self):
    self.dW_drho = (2 * self.k_1 / (1 + np.dot(self.rho.T, self.rho))) * self.rho

def advance_state(self, u) :
    dt = self.dt

    # dynamics
    self.rho_dot = np.dot((self.I + self.rho_cross + np.dot(self.rho,
self.rho.T)), self.omega)
    self.omega_dot = np.dot(np.linalg.inv(self.J),(-(np.cross(self.omega ,
np.dot(self.J, self.omega))) + u))
    self.y = self.omega
    self.compute_dW_drho()

    # update states
    self.omega += self.omega_dot * dt
    self.rho += self.rho_dot * dt
    self.rho_cross = self.get_curl(self.rho)

def compute_control_law(self):
    self.nu = -self.k_2 * np.tanh(self.omega)
    self.u = self.nu - np.dot(self.dW_drho,(self.I + self.rho_cross +
np.dot(self.rho, self.rho.T))).T

def simulate(self, t_f):
    self.t = np.linspace(0,t_f,int(t_f/self.dt))
    self.compute_control_law()
    self.u_1_array      = []
    self.u_2_array      = []
    self.u_3_array      = []
    self.rho_1_array    = []
    self.rho_2_array    = []
    self.rho_3_array    = []
    self.omega_1_array = []
    self.omega_2_array = []
    self.omega_3_array = []

    for _ in range (len(self.t)):
        self.rho_1_array.append(self.rho[0])
        self.rho_2_array.append(self.rho[1])
        self.rho_3_array.append(self.rho[2])

```

```

        self.omega_1_array.append(self.omega[0])
        self.omega_2_array.append(self.omega[1])
        self.omega_3_array.append(self.omega[2])
        self.u_1_array.append(self.u[0])
        self.u_2_array.append(self.u[1])
        self.u_3_array.append(self.u[2])
        self.advance_state(self.u)
        self.compute_control_law()

def plot(self):

    fig, ax = plt.subplots(1, 3)

    ax[0].plot(self.t, self.rho_1_array, '-b')
    ax[0].plot(self.t, self.rho_2_array, '-g')
    ax[0].plot(self.t, self.rho_3_array, '-r')

    ax[0].legend(["rho1", "rho2", "rho3"])
    ax[0].set_ylabel("v")
    ax[0].set_xlabel("t")
    ax[0].grid()
    ax[0].set_title("Components of Rho vs iteration t")

    ax[1].plot(self.t, self.omega_1_array, '-b')
    ax[1].plot(self.t, self.omega_2_array, '-g')
    ax[1].plot(self.t, self.omega_3_array, '-r')

    ax[1].legend(["omega1", "omega2", "omega3"])
    ax[1].set_ylabel("w")
    ax[1].set_xlabel("t")
    ax[1].grid()
    ax[1].set_title("Components of Omega vs iteration t")

    ax[2].plot(self.t, self.u_1_array, '-b')
    ax[2].plot(self.t, self.u_2_array, '-g')
    ax[2].plot(self.t, self.u_3_array, '-r')

    ax[2].legend(["u1", "u2", "u3"])
    ax[2].set_ylabel("w")
    ax[2].set_xlabel("t")
    ax[2].grid()
    ax[2].set_title("Components of input u vs iteration t")

plt.show()

```

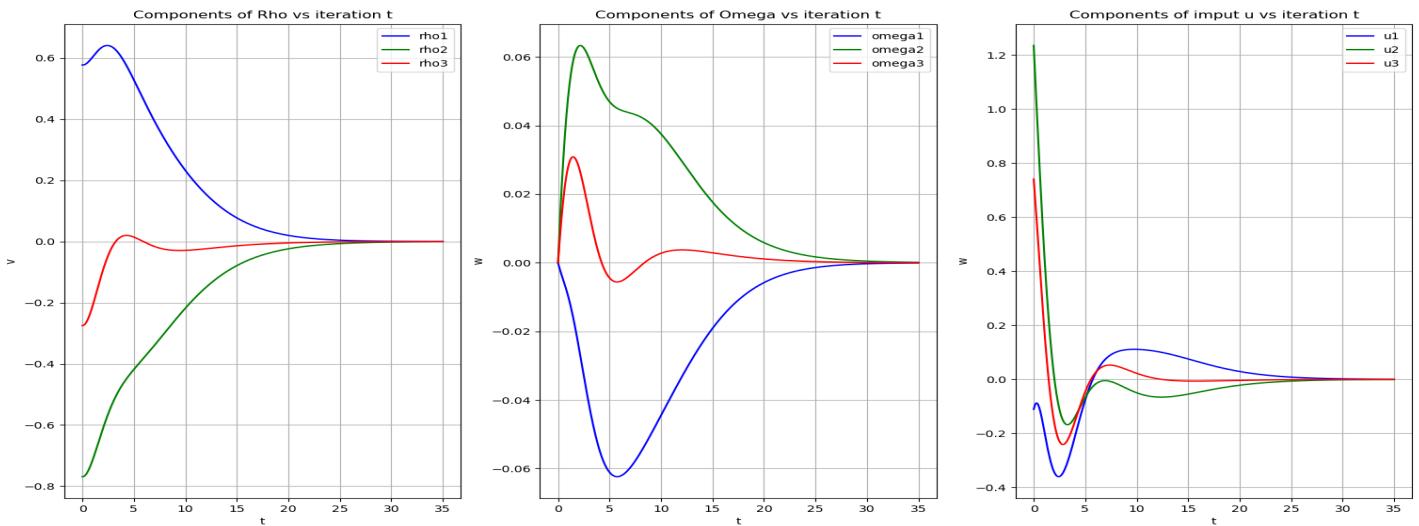
```

if __name__ == '__main__':
    alpha_1 = 0
    theta_initial = -35
    J = np.array([[20, 1.2, 0.9], [1.2, 17, 1.4], [0.9, 1.4, 15]])
    orient = np.array([1/sqrt(3), 1/sqrt(3), 1/sqrt(3)]).T
    k = np.array([1, 0, 0]).T
    rho_0 = orient*np.cos(theta_initial) + np.cross(k, orient)*np.sin(theta_initial) + k*(np.dot(k, orient))*(1-np.cos(theta_initial))
    omega_0 = np.array([0.0, 0.0, 0.0]).T

    rot_dyn = rigid_body(J, rho_0, omega_0)
    rot_dyn.simulate(35)
    rot_dyn.plot()

```

Plots:



Q1

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + Dq + g(q) = u$$

$q \in \mathbb{R}^n$: generalized coordinates

\dot{q} : generalized velocities

$M(q) > 0 \in \mathbb{R}^{n \times n}$: positive definite inertia matrix

$C(q, \dot{q}) \in \mathbb{R}^{n \times n}$: centrifugal + coriolis force

$D \in \mathbb{R}^{n \times n}$: viscous damping

$g(q) \in \mathbb{R}^n$: gravity

If it is known $i - 2C$ is skew symmetric

$D > 0$: is a positive definite

Objective : track constant reference q_r

so that

$e = q - q_r$, and dynamics of e

$$M(q)\ddot{e} + C(q, \dot{q})\dot{e} + D\dot{e} + g(q) = u$$

we want ~~$\ddot{e} = 0$~~ $e = 0$ as GAS

consider control

$$u = g(q) - K_p e + \nu \quad \text{for feedback passivation}$$

$$K_p = K_p^T > 0$$

$$K_p \in \mathbb{R}^{n \times n}$$

Ans: dynamics of the above feedback :

$$e_1 = e$$

$$\dot{e}_1 = \dot{e} = e_2$$

$$e_2 = \ddot{e}$$

$$\dot{e}_2 = \ddot{e}$$

$$M(q) \ddot{e}_2 + C(q, \dot{q}) e_2 + D e_2 + g(q) = u = g(q) - K_p e_1 + v$$

$$u = g(q) - K_p e_1 + v$$

Dynamics — State space equation.

$$\dot{e}_1 = e_2$$

$$M(q) \ddot{e}_2 = -[K_p e_1 + D e_2 + C(q, \dot{q}) e_2] + v$$

consider the storage function:

$$V(s) = \frac{1}{2} \dot{e}^T M(q) \dot{e} + \frac{1}{2} e^T K_p e$$

$$\dot{V}(s) = \frac{1}{2} \dot{e}^T M(q) \ddot{e} \times 2 + \frac{1}{2} \dot{e}^T M(q) \dot{e} + \frac{1}{2} e^T K_p \dot{e} \times 2$$

$$= \dot{e}^T [M(q) \ddot{e} + \frac{1}{2} \dot{e}^T M(q) \dot{e}] + e^T K_p \dot{e}$$

$$= \dot{e}^T [v - K_p e - D \dot{e} - C \dot{e} + \frac{1}{2} \dot{e}^T M(q) \dot{e}] + e^T K_p \dot{e}$$

we know $(M - 2C)$ is skew symmetric
and $\dot{e}^T (M - 2C) \dot{e}$ for skew-symmetric
matrix is zero.

$$= \dot{e}^T [v - K_p e - D \dot{e}] + e^T K_p \dot{e}$$

$$= -\dot{e}^T D \dot{e} + \dot{e}^T v - \dot{e}^T K_p e + e^T K_p \dot{e}$$

$$(\dot{e}^T K_p e)^T = (\dot{e}^T K_p e) \quad \text{(scalar)}$$

$$e^T K_p \dot{e} = \dot{e}^T K_p e$$

$$\dot{V}(s) = -\dot{e}^T D \dot{e} + \dot{e}^T v$$

$$\dots K_p = K_p^T > 0 \quad \text{(given)}$$

$$\dot{V}(x) = -\dot{e}^T D \dot{e} + \dot{e}^T v$$

$$\dot{V}(x) + \dot{e}^T D \dot{e} = \dot{e}^T v$$

$$\therefore c = q - q_r \rightarrow \dot{e} = \dot{q} - 0$$

Output: \dot{q}

$$\dot{V}(x) + \dot{q}^T (D \dot{q}) = \dot{e}^T v$$

we know $D > 0$ Positive definite

hence $\dot{q}^T (D \dot{q}) > 0 \neq q \neq 0$

hence,

$$\dot{V}(x) + y^T S(y) = \dot{e}^T v$$

strictly output passive

$$\dot{V}(x) \leq \dot{e}^T v$$

$\dot{e}^T v$: scalar

$$\dot{e}^T v = (\dot{e}^T v)^T = v^T \dot{e}$$

$$\dot{V}(x) \leq v^T \dot{e} \quad \text{... passivity proved}$$

$$\text{for } y = \dot{e} - e_2$$

$$\dot{V}(x) \leq v^T y$$

for output chosen

checking for zero-state observability for the chosen output

$$y = \dot{e} = 0 \rightarrow e_2 = 0 \quad [e_2 = \dot{e}]$$

$$\dot{e}_2 = 0 \quad (\text{derivative of } 0 \text{ is } 0)$$

$$H(q) \dot{e}_1 = [-k_p e_1 - C e_2 - D \dot{e}_2] + i \dot{q} \quad \text{when } v = 0$$

$$-k_p e_1 = 0 \rightarrow e_1 = 0 \quad \because k_p > 0$$

hence $y = 0 \rightarrow (e_1, e_2) = (0, 0)$ when $v = 0$

hence \rightarrow zero-state observable.

~~$e_1 = e_2$~~

~~$M(q) \dot{e}_2 = -[K_p e_1 + C(q, q) e_2 + D(q) e_2] + v$~~

~~$Q = \begin{pmatrix} 0 \\ 1 \end{pmatrix}^T$~~

$$\frac{\partial V}{\partial e} = \left[\begin{array}{c} e_1^T K_p \\ e_2^T M \\ e_2^T H + \frac{1}{2} e_2^T \dot{H} e_2 \end{array} \right]^T$$

where

~~$V = \frac{1}{2} e_2^T H e_2 + \frac{1}{2} e_1^T K_p e_1$~~

~~$\left[\left[\frac{\partial V_{(0)}}{\partial e} \right] Q(v) \right]^T = e_2^T M + \frac{1}{2} e_2^T \left(\frac{\partial H}{\partial e} \right) e_2$~~

~~$v = -k \tanh(y) \text{ or } -ky$~~

To make actuator input bounded

taking $v = -k \tanh(y)$ as v -input

Since $\dot{v}(a) = 0$ only at $y = a \rightarrow \dot{y} = 0$

and $\dot{v}(a) < 0$ otherwise

$\therefore v = -k \tanh(y)$ and $y = 0$ only at $\dot{y} = 0$

and $v(a) \leq v^T y$

hence locally asymptotically stable.

and

Since a has no bounds \rightarrow globally asymptotically stable (RUB)

Q3.

$$\dot{\vec{s}} = [I + \vec{s}^* + \vec{s}\vec{s}^T] \vec{w}$$

$$J\dot{\vec{w}} = -\vec{w} \times J\vec{w} + u$$

with $y = \vec{w}$

$u \in \mathbb{R}^3$: thrust
 $\vec{s} \in \mathbb{R}^3 \leftarrow$ Rodriguez parameters
 $\vec{w} \in \mathbb{R}^3$
 $J = J^T > 0 \leftarrow$

Inertia matrix

with $\vec{w} = 0 \rightarrow \dot{\vec{s}} = 0 \Rightarrow$ Stable

hence any $\vec{w}(s) > 0$ (RUB)

will ensure $\dot{w}(s) \leq 0$

considering $V(\vec{w}) = \frac{1}{2} \vec{w}^T J \vec{w} > 0$

$$\dot{V}(\vec{w}) = \vec{w}^T J \dot{\vec{w}} + \frac{1}{2} \vec{w}^T J \vec{w}$$

$$\begin{aligned} &= \vec{w}^T [-\vec{w} \times J\vec{w} + u] \\ &= \cancel{\vec{w}^T} \vec{w}^T [-\vec{w} \times J\vec{w} + u] \\ &= \vec{w}^T [J\vec{w} \times \vec{w} + u] \\ &= \vec{w}^T (J\vec{w} \times \vec{w}) + \vec{w}^T u \\ &= -\vec{w}^T (\vec{w} \times J\vec{w}) + \vec{w}^T u \end{aligned}$$

negative definite

hence

$$\dot{V} < \underbrace{\vec{w}^T u}_{= u^T \vec{w}}$$

$$\dot{V} < u^T \vec{w}$$

where $y = \vec{w}$

hence

Proved

$$\boxed{\dot{V} < u^T y}$$

Taking

Storage function

$$U(\vec{w}, s) = V(\vec{w}) + W(s)$$

$$\dot{U}(\vec{w}, s) = \dot{V}(\vec{w}) + \dot{W}(s)$$

$$\begin{aligned} &= -\vec{w}^T (\vec{w} \times J\vec{w}) + \underbrace{\frac{\partial W}{\partial s} s}_{{\text{-ve semi definite}}} + \vec{w}^T u \end{aligned}$$

$$\dot{U}(\vec{w}, s) < \frac{\partial W}{\partial s} [I + [s] + \vec{s}\vec{s}^T] \vec{w} + \vec{w}^T u$$

$$\dot{u}(w, s) \leq \frac{\partial w}{\partial s} [I + [s] + ss^T] w + u^T w$$

$$\because u^T w = w^T u \text{ (scalar)}$$

for $u = v - \left[\frac{\partial w}{\partial s} [I + [s] + ss^T] \right]^T$

with ~~w(s)~~ $w(s) = k \ln(1 + s^T s)$

find u ;

and prove

the system to be zero state observable in (v, y)

$$\frac{\partial w}{\partial s} = \frac{k \cdot 2(s^T I)}{(1 + s^T s)}$$

$$\frac{\partial w}{\partial s} = \frac{2k \cdot s^T}{1 + s^T s} \cancel{[] }$$

$$\dot{u}(w, s) \leq \frac{\partial w}{\partial s} [I + [s] + ss^T] w + v^T w - \left[\frac{\partial w}{\partial s} [I + [s] + ss^T] \right] w$$

$$\dot{u}(w, s) \leq \underbrace{v^T w}_{w=y} \rightarrow \dot{u}(w, s) \leq v^T y$$

~~to prove zero state observability~~ $\rightarrow (v, y)$

for $y = 0 \rightarrow w = 0$

$w = 0 \rightarrow \dot{w} = 0$

~~so~~ $w = 0 \rightarrow \dot{s} = 0$

$$\dot{J}\dot{w} = -\omega \times J\omega + v^T - \left[\frac{\partial w(s)}{\partial s} [I + [s] + ss^T] \right]^T$$

$$\text{hence } \frac{\partial W}{\partial S} [I + [S] + SS^T] = 0$$

$$\frac{2kS^T}{(1+S^T S)} [I + [S] + SS^T] = 0$$

$$k \neq 0$$

$$I + S^T S \neq 0 \quad \because \text{argument of } \ln(a) \quad a > 0$$

$$\underbrace{I + [S] + SS^T}_{=0} \quad \text{OR} \quad S^T = 0$$

$$\begin{bmatrix} 1 & -S_z & S_y \\ S_z & 1 & -S_x \\ -S_y & S_x & 1 \end{bmatrix} + \begin{bmatrix} S_x^2 & S_x S_y & S_x S_z \\ S_x S_y & S_y^2 & S_y S_z \\ S_x S_z & S_y S_z & S_z^2 \end{bmatrix}$$

$$1 + S_x^2 \neq 0$$

$$1 + S_y^2 \neq 0$$

$$1 + S_z^2 \neq 0$$

hence only $S^T = 0 \Rightarrow$ the only solution

$$S = 0 \quad \therefore S^T = 0$$

hence zero-state observability proved

$$\therefore u(w, \varphi) \leq v^T w \rightarrow \text{proves passivity}$$

$$\text{since } y = w$$

$$\text{choosing } \varphi = -k \tanh(y)$$

$$u = -k \tanh(y) - \left[\frac{\partial W}{\partial S} [I + [S] + SS^T] \right]^T$$