

ENERJİ EŞPARÇALANIMI İLE TİTREŞİM SÖNÜMLEMESİ ile ENERJİ GERİ KAZANIMI TEKNİKLERİ GELİŞTİRİLMESİ ÇALIŞMA RAPORU

INDEX :

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- 2) Analytic Calculation method of both Systems
- 3) Impulse Response of Lineer and Optimum Systems
- 4) Validation of Numeric Results via Comparision to Analytical Results
 - 4.1) External Excitation - Lineer Frequency Distribution
 - 4.2) External Excitation - Optimum Frequency Distribution
- 5) White Noise Input to Lineer and Optimum Frequency Dist Systems
- 6) Impulse Response of Optimum Frequency Dist Scaled to AFM frequencies

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- A.3 Comparision between Analytic Solution and Numeric Solution
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- A.9 Carbon x 32 Atom Lattice Simulation ~ A Work which is inspired from Dissipation in Solids Paper

Preparing:

- A.10 Analytic Solution Code - Numeric Solution Code - Convolution Code for Matlab
- A.11 Phase Portraits of Some NonLinear Systems - Transition to Chaos.

LITERATURE REVIEW

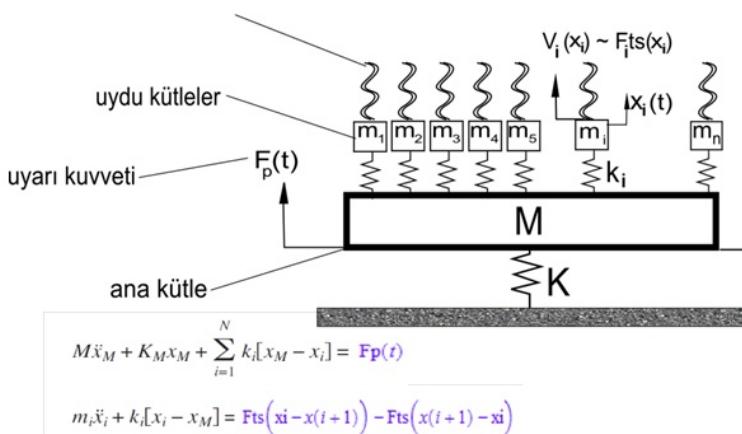
- L.1 AFM Cantilever Mechanics and Model
- L.2

I. Giriş :

Enerji eşparçalanımı metodu, titreşim hareketi yapan ana yapıya fiziksel olarak bağlı farklı özfrekanslara sahip uydu kütlelerin, kendi frekanslarında titreşerek yapılan titreşim enerjisini her birinde eşit olacak şekilde, kendi üzerilerine çekmeleriyle ana yapıdaki titreşimin enerji çekilerek sökümlendiği metottur.

Enerji geri kazanımındaki temel ilke, uydu kütleler tarafından çekilen enerjinin her bir uydu kütleye sökümlenerek, mekanik enerjiden elektrik enerjisine dönüştürülmesine dayanır.

uydu kütleye uygulanan kuvvet ya da varsa
non lineer potansiyel den türeyen (geri getirme) kuvvet



NonLinear Potansiyel Fonk :
 $V_i(x) = -E_{bond}(-2 \cdot \exp(-kd(x-x_e)) + \exp(-2kd(x-x_e)))$

Potansiyelden Türeyen Kuvvet Fonk :

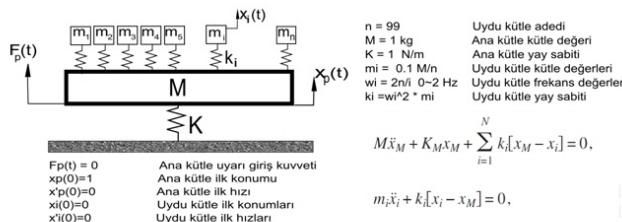
$$Fts(x) = E_{bond}((2*kd)/\exp(kd(x-x_e)) - (2*kd)/\exp(2*kd(x-x_e)))$$

avagadro=6.022142*10^23
Ebond=300/avagadro
kd=2.55*10^10
ze = 2.42*10^-10
avagadro sayısı
atomik ağırlık başına enerji (kj)
azalma faktörü
bag mesafesi (m) 2.42 A

$$V_i(x) = -E_{bond} \left(\frac{1}{2 kd(x-x_e)} - \frac{2}{e^{kd(x-x_e)}} \right)$$

$$Fts(x) = E_{bond} \left(\frac{2 kd}{e^{kd(x-x_e)}} - \frac{2 kd}{2 e^{kd(x-x_e)}} \right)$$

Lineer Frekans Dağılımı :



n = 99
M = 1 kg
K = 1 N/m
mi = 0.1 M/n
wi = 2*pi/0-2 Hz
ki = wi^2 * mi

Uydu kütleye adedi

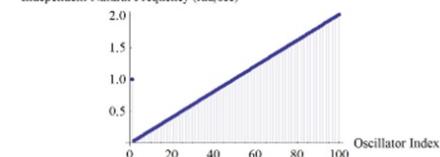
Ana kütleye adedi

Ana kütleye yay sabiti

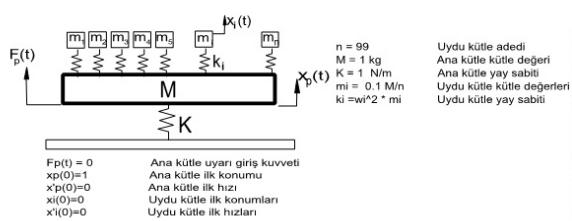
Uydu kütleye değerleri

Uydu kütleye yay sabiti

Independent Natural Frequency Distribution
Independent Natural Frequency (rad/sec)



Optimum Frekans Dağılımı :



n = 99
M = 1 kg
K = 1 N/m
mi = 0.1 M/n
ki = wi^2 * mi

Uydu kütleye adedi

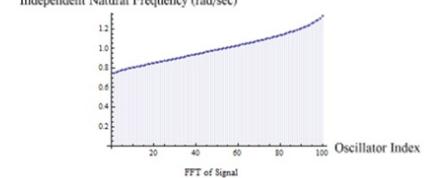
Ana kütleye adedi

Ana kütleye yay sabiti

Uydu kütleye değerleri

Uydu kütleye yay sabiti

Independent Natural Frequency Distribution
Independent Natural Frequency (rad/sec)



2. Analitik Hesap Yöntemi

MDOF HESAP AKIŞI

$\Rightarrow m$ - mass array
 $\Rightarrow k$ - stiffness array
 $\Rightarrow M$ - mass matrix
 $\Rightarrow K$ - stiffness matrix

$x(K-\lambda M)=0$

$\dot{q} = U\dot{u}$

$E_{top} = \frac{1}{2}\dot{q}^T M \dot{q} + \frac{1}{2}\dot{q}^T K q$

$U_m = \frac{U}{\sqrt{U^T M U}}$ \Rightarrow mass normalization

$E_{kinetic} = \frac{1}{2}m_2\dot{q}_2^2 + \frac{1}{2}(k_1 - k_2)\dot{q}_2^2 + \dots$

$L = T - V = \frac{1}{2}(\dot{q}^T M \dot{q} + \dot{q}^T K q)$

$\frac{d(L)}{dt} = \frac{dL}{dt} - \frac{\partial L}{\partial t} = M\ddot{q} + Kq = 0$

$U_m^T K U_m = [\omega_1^2 \dots]$

$U_m^T M U_m = I$

$$\dot{q} = \dot{q}_H + \dot{q}_P$$

$$q_H = \sum_{i=1}^n [U_m^T M \dot{q}_H(i) \cos \omega_i t + \frac{1}{\omega_i} U_m^T M \dot{q}_H(i) \sin \omega_i t] U_m \Rightarrow \text{Homogen Çözüm}$$

$$q_P = \sum_{i=1}^n \left[\frac{U_m^T}{\omega_i} \int_0^t q(i-t) \sin \omega_i t dt \right] U_m \Rightarrow \text{Partiküler Çözüm}$$

$$E_{MODAL} = \dot{q}^2 + [E_{\omega_i^2}] \dot{q}$$

$$\text{Excitation force } \underline{Q}(t) = Q_0 \cos \omega t$$

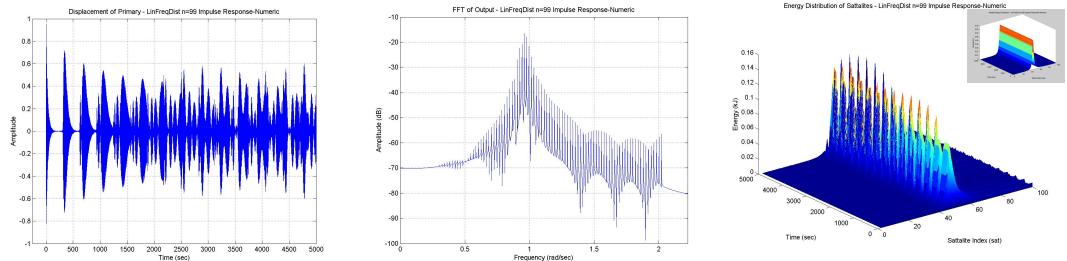
$$E_{error} = E_{tot} - E_{mod} - E_{out}$$

$$E_{out} = \frac{1}{2} (\dot{q}_H)^T M \dot{q}_H + q_0^T K q_0$$

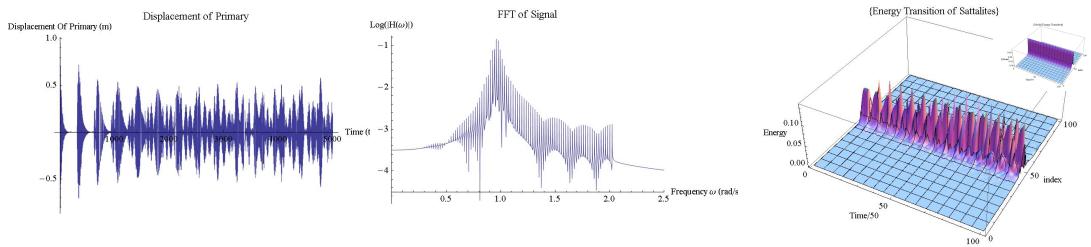
$$E_0 = \int q(i) dt \quad \text{constant için } E_{\omega_1^2} = 0 \text{ olursa}$$

3. Impulse Response Simulation Results

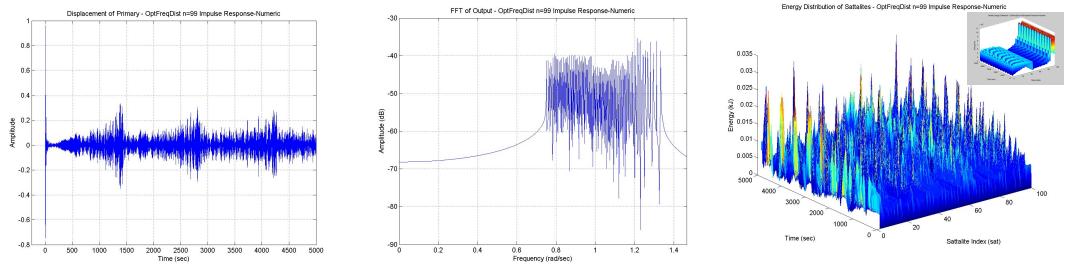
Lineer F.D.
Impulse Response
Numeric Solution



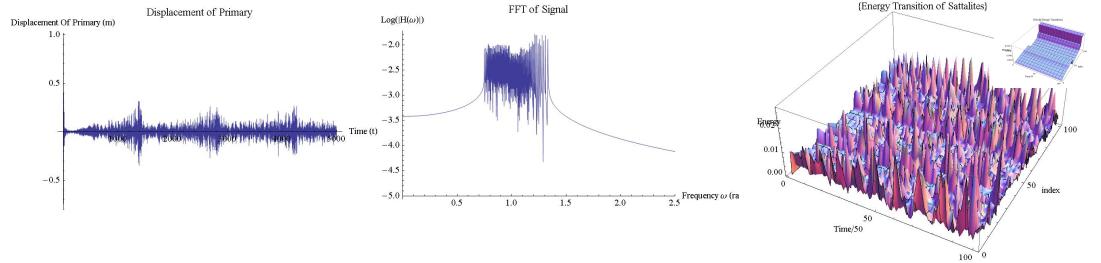
Lineer F.D.
Impulse Response
Analytic Solution



Optimum F.D.
Impulse Response
Numeric Solution



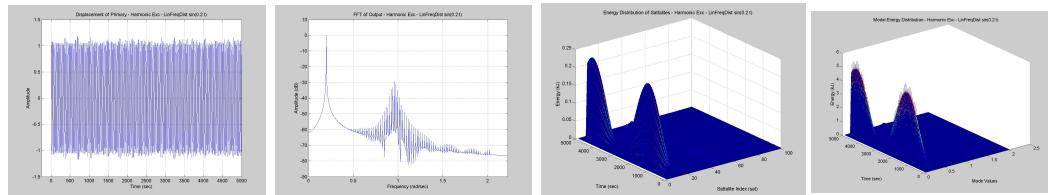
Optimum F.D.
Impulse Response
Analytic Solution



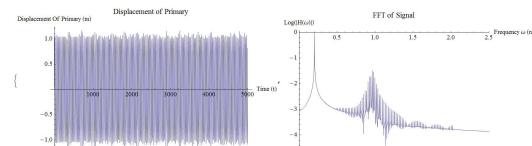
4.1 Harmonic Response of Linear Freq. Dist.

$\text{Sin}(0.2 t)$

Numeric Solution

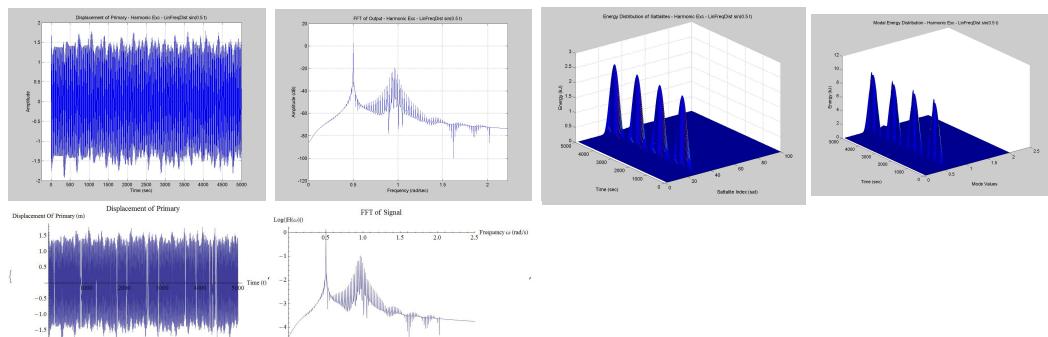


Analytic Solution

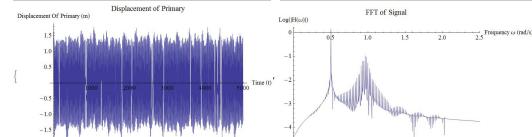


$\text{Sin}(0.5 t)$

Numeric Solution

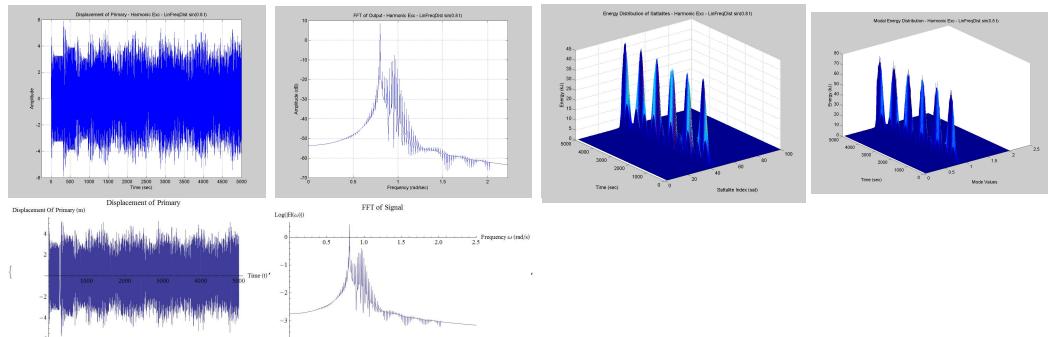


Analytic Solution

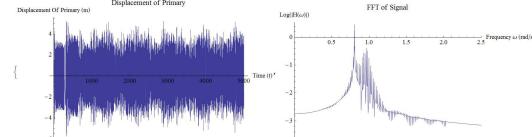


$\text{Sin}(0.8 t)$

Numeric Solution

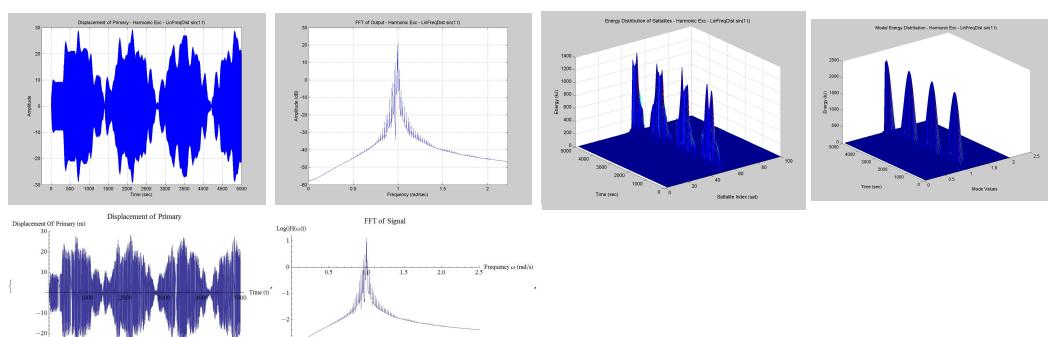


Analytic Solution

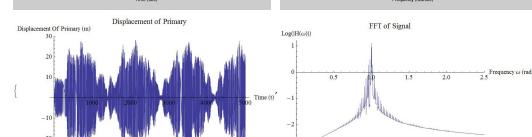


$\text{Sin}(1 t)$

Numeric Solution

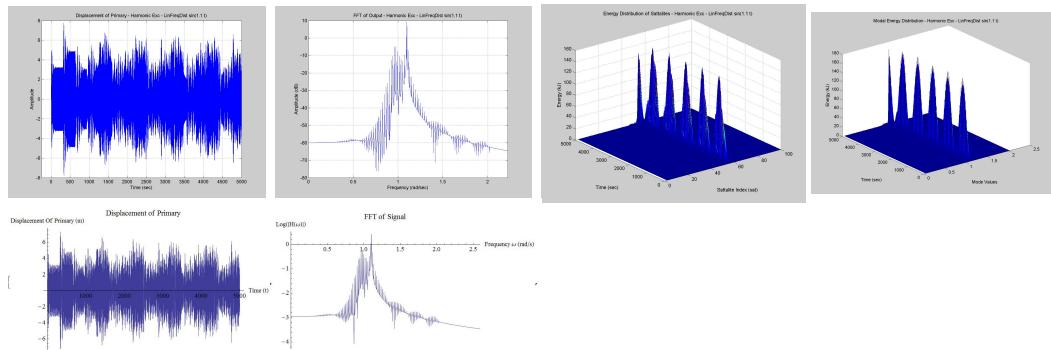


Analytic Solution



$\text{Sin}(1.1t)$

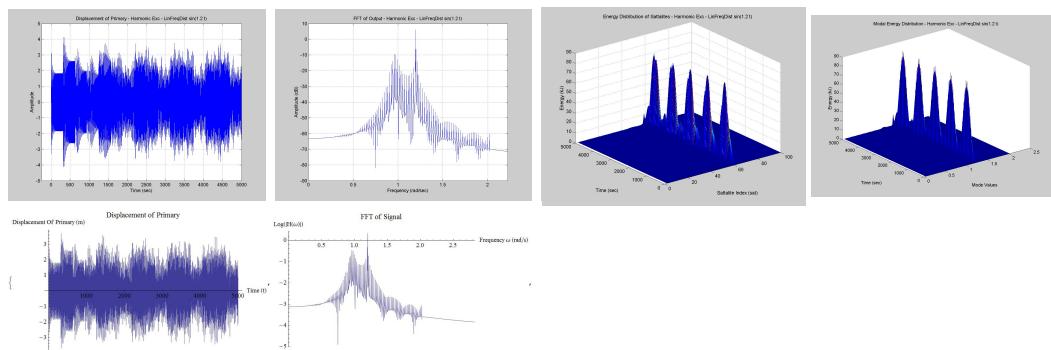
Numeric Solution



Analytic Solution

$\text{Sin}(1.2t)$

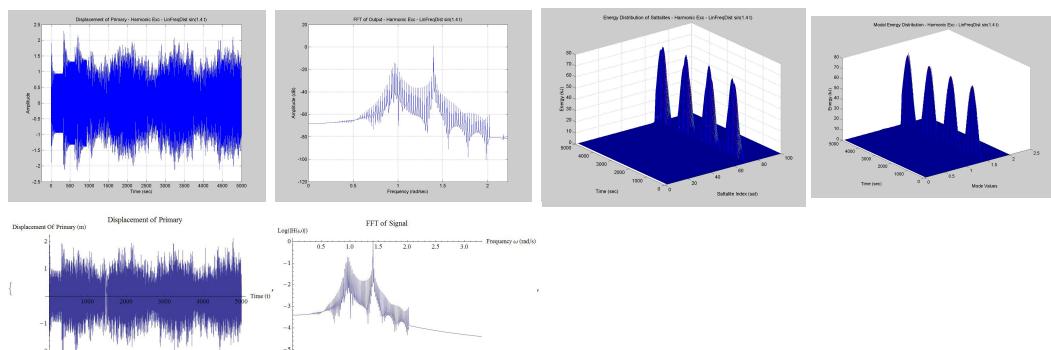
Numeric Solution



Analytic Solution

$\text{Sin}(1.4t)$

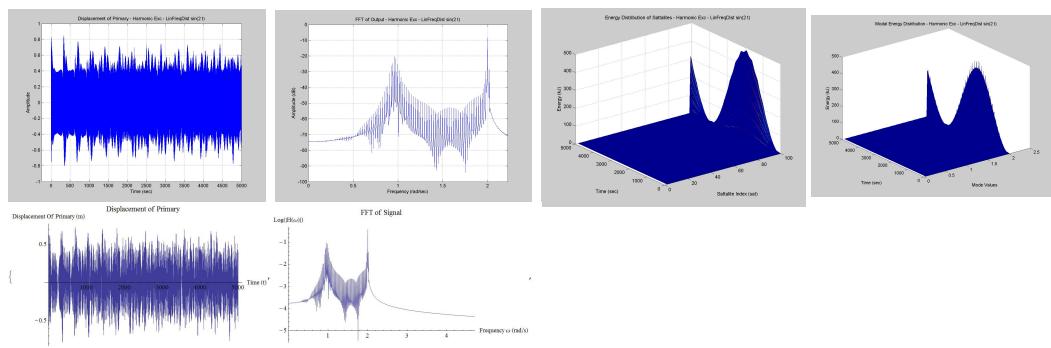
Numeric Solution



Analytic Solution

$\text{Sin}(2t)$

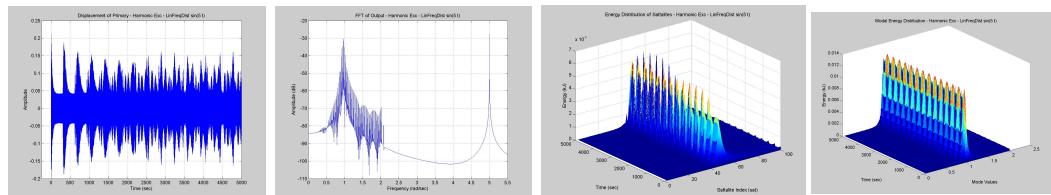
Numeric Solution



Analytic Solution

$\text{Sin}(5t)$

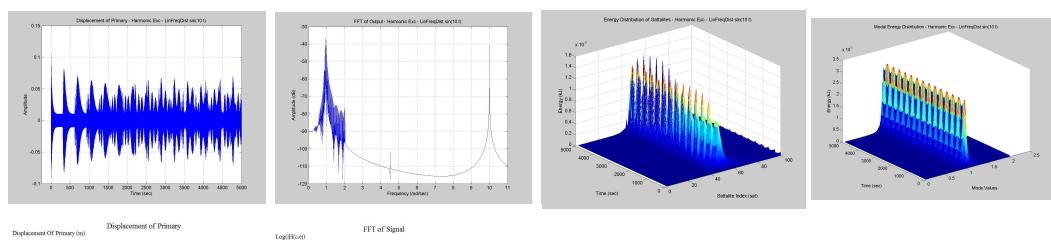
Numeric Solution



Analytic Solution

$\text{Sin}(10t)$

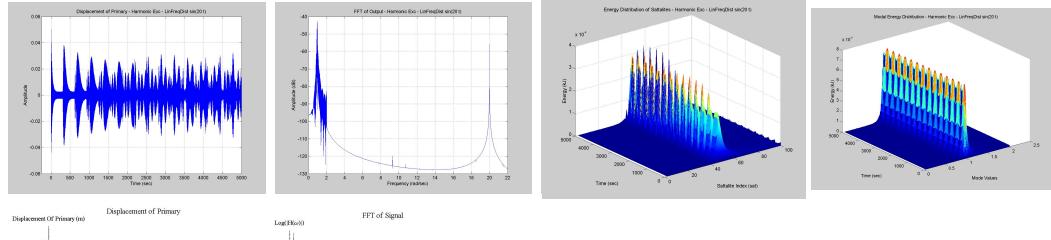
Numeric Solution



Analytic Solution

$\text{Sin}(20t)$

Numeric Solution

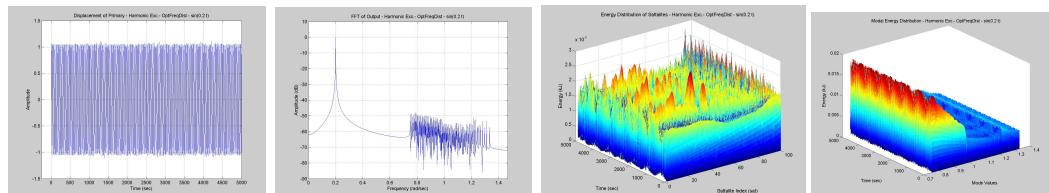


Analytic Solution

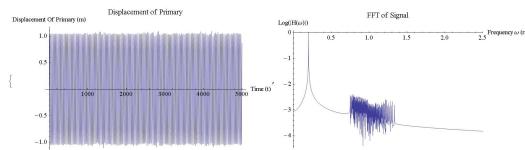
4.2 Harmonic Response of Opt. Freq. Dist.

$\text{Sin}(0.2t)$

Numeric Solution

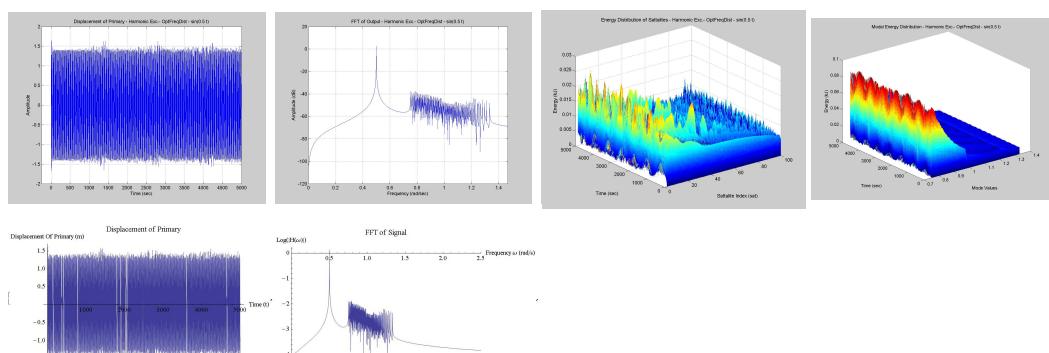


Analytic Solution

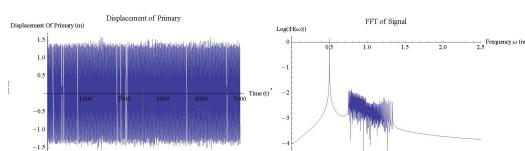


$\text{Sin}(0.5t)$

Numeric Solution

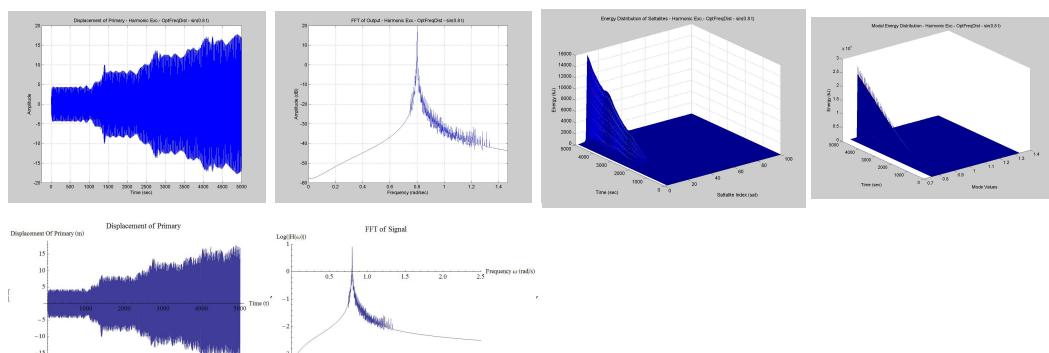


Analytic Solution

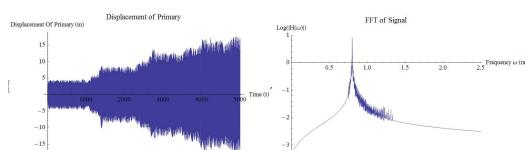


$\text{Sin}(0.8t)$

Numeric Solution

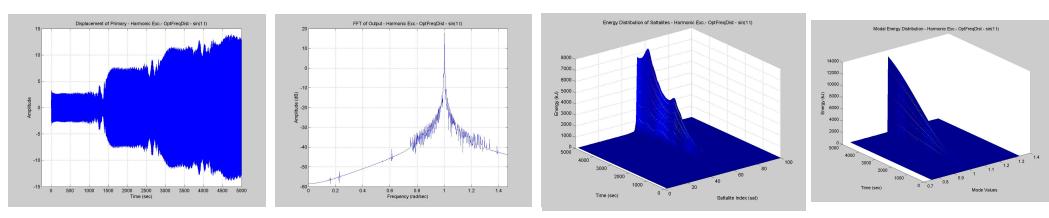


Analytic Solution

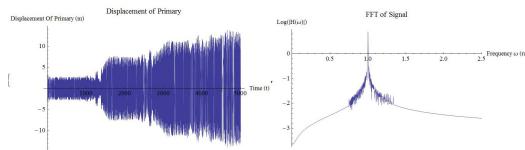


$\text{Sin}(1t)$

Numeric Solution

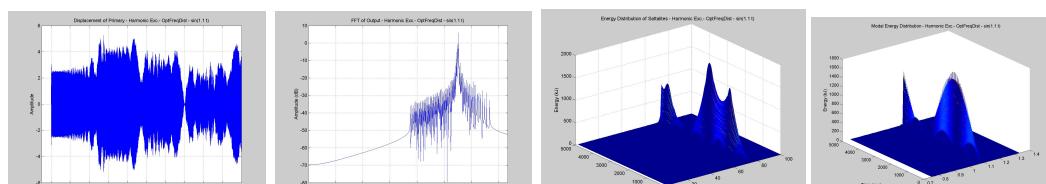


Analytic Solution

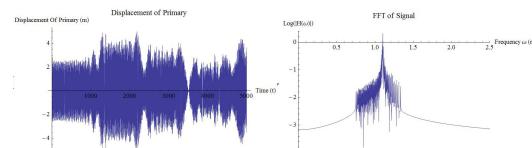


$\text{Sin}(1.1 t)$

Numeric Solution

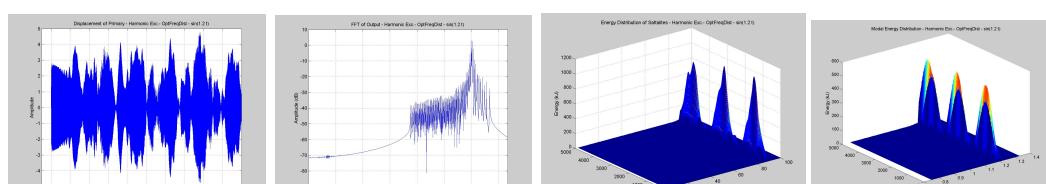


Analytic Solution

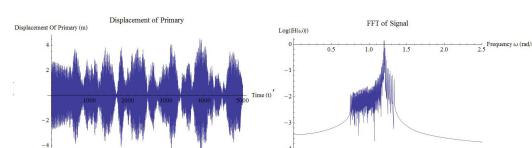


$\text{Sin}(1.2 t)$

Numeric Solution

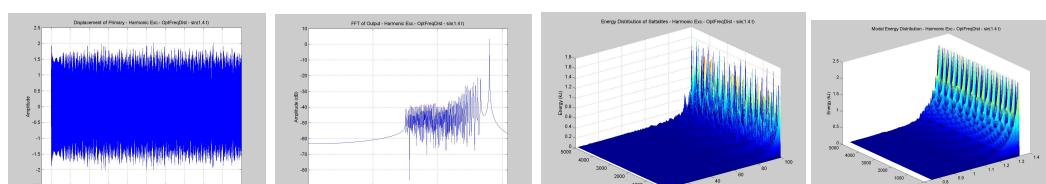


Analytic Solution

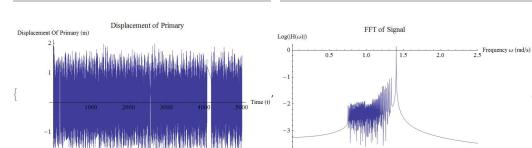


$\text{Sin}(1.4 t)$

Numeric Solution

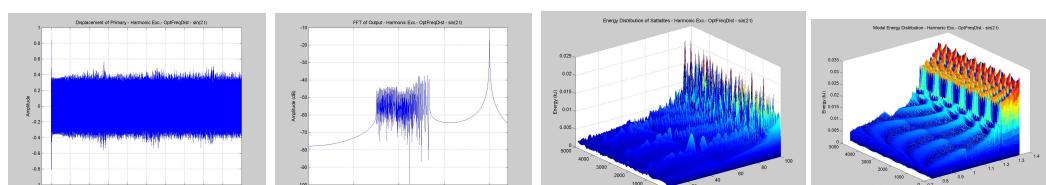


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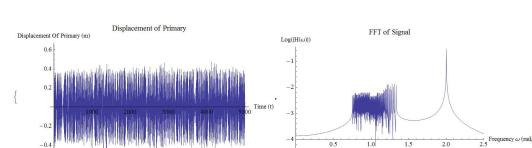


$\text{Sin}(2 t)$

Numeric Solution

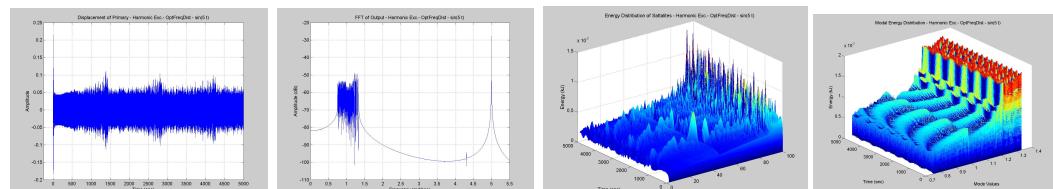


Analytic Solution

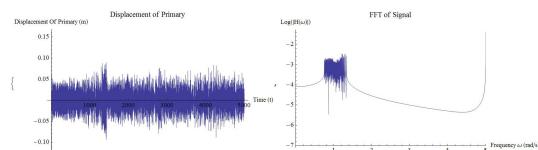


$\text{Sin}(5t)$

Numeric Solution

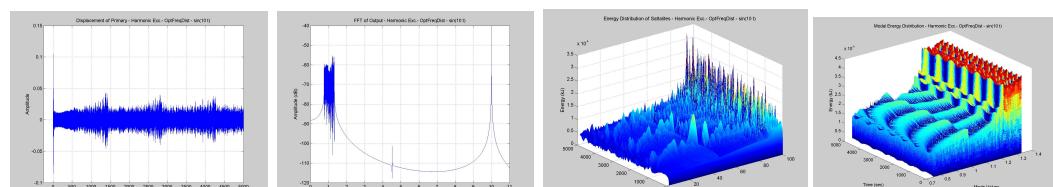


Analytic Solution

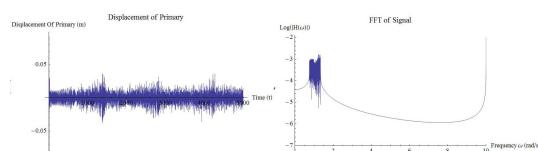


$\text{Sin}(10t)$

Numeric Solution

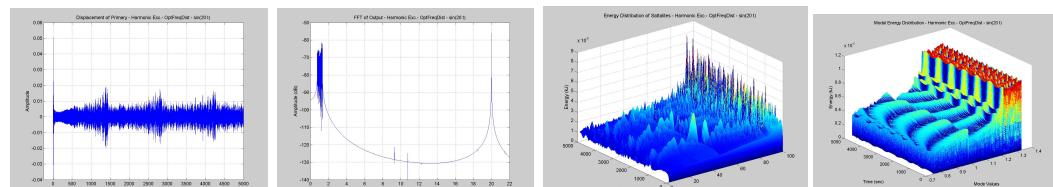


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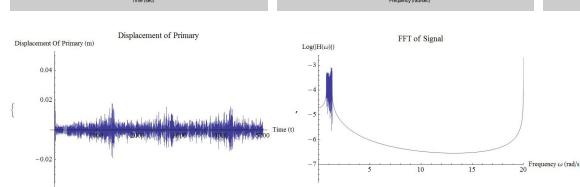


$\text{Sin}(20t)$

Numeric Solution



Analytic Solution

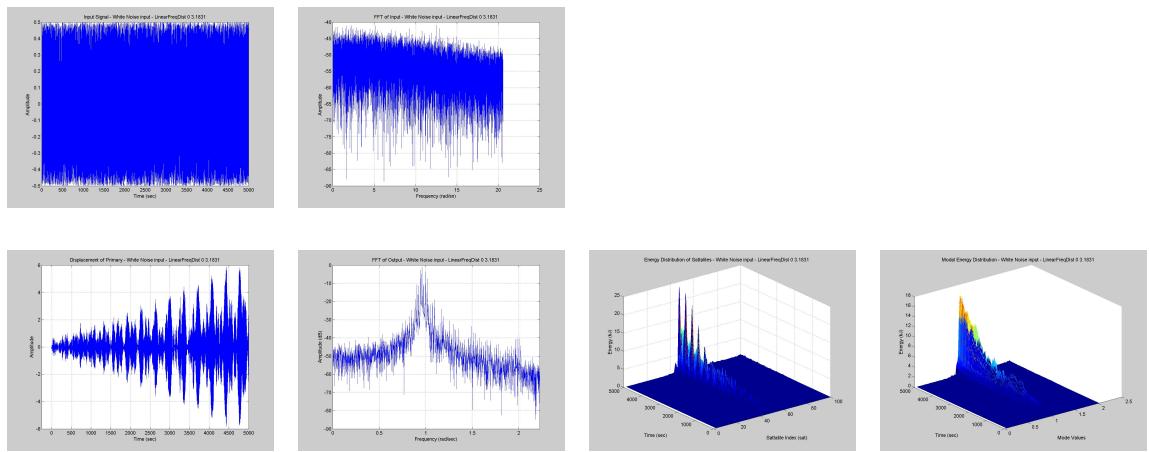


5. White Noise Input to Lin And Opt Freq Dist

Lineer Freq Dist

White Noise

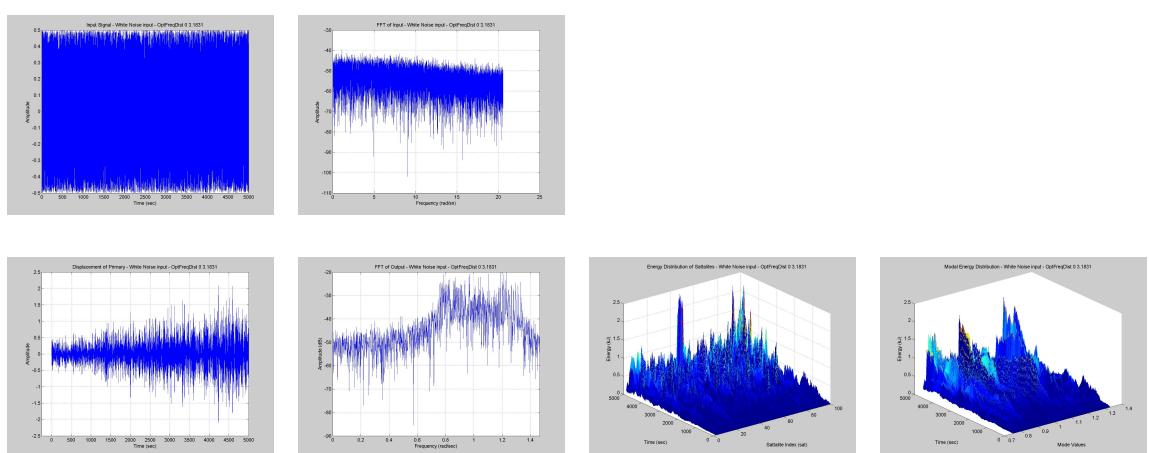
0-20 rad/sec



Optimum Freq Dist

White Noise

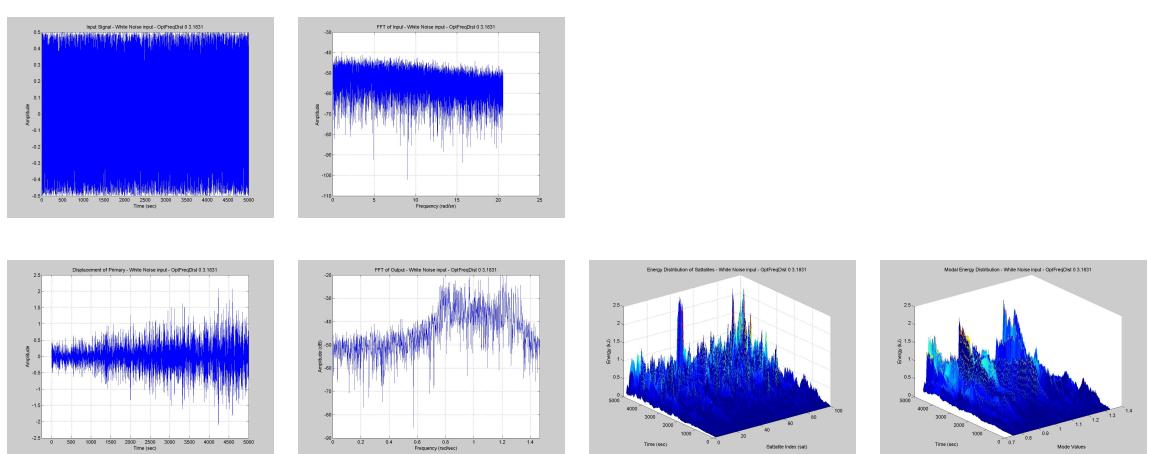
0-20 rad/sec



Optimum Freq Dist

White Noise

0-1.34 rad/sec

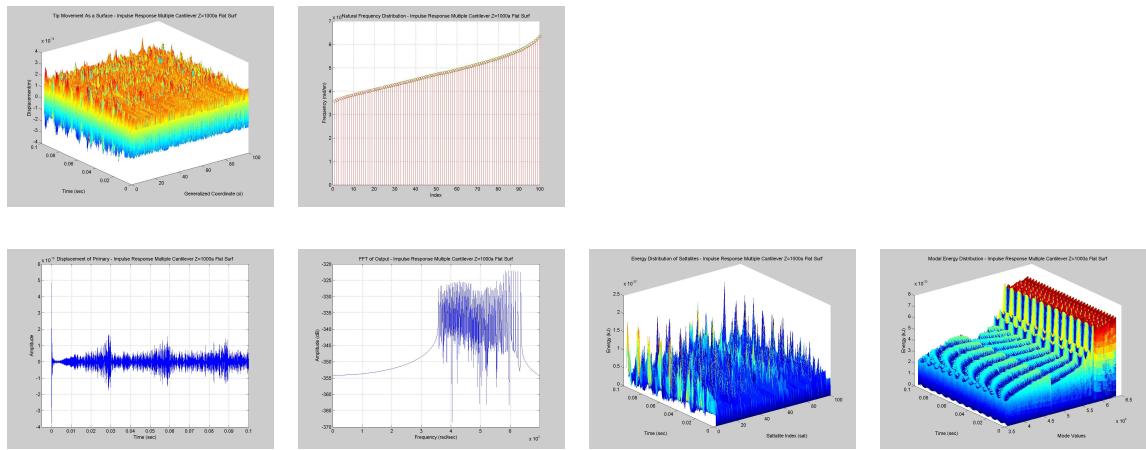


6. Impulse Response - Optimum Frequency Scaled to AFM Frequencies

Impulse Response

Scaled Opt

$Z=1000^*a$



Appendix

A0. Summary OF MDOF Systems

$M\ddot{q}(t) + C\dot{q}(t) + Kq(t) = Q(t) \rightarrow M = M^T \quad C = CT \quad K = K^T$ simetrik 22-24/06/19
Gelişme

General form of equations, derived from Lagrangian mechanics.

Influence coefficients

Stiffness inf. coeffs $F_i = \sum_{j=1}^n k_{ij} u_j$ Flexibility inf. coeffs $u_i = \sum_{j=1}^n a_{ij} F_j$

$$A = [a_{ij}] \quad K = [k_{ij}]$$

$$A \cdot K = I$$

↳ j. den birim deformaşon uygulandığında

i. yi etkili tutacak kuvvet

↳ j. den birim kuvvet uygulandığında, i. deli deplasman.

$$V = \frac{1}{2} u^T K u \rightarrow \text{potansiyel enerji} \text{ quadratik Formda}$$

$$T = \frac{1}{2} u^T M u \rightarrow \text{kinetik enerji} \text{ quadratik Formda}$$

$$\mathcal{F} \text{ or } D = \frac{1}{2} q^T C q \rightarrow \text{dissipasyon enerjisi}$$

$$\frac{\partial(\frac{1}{2} T)}{\partial t} - \frac{\partial T}{\partial q_2} + \frac{\partial V}{\partial q_2} + \frac{\partial D}{\partial q_2} = 0$$

Lagrangian Mechanics

— MODAL Uzayda —

$$q(t) = U\tilde{q}(t), \quad \ddot{q}(t) = U\ddot{\tilde{q}}(t)$$

Burda $U \rightarrow$ öznektörler oslinda bir (modal uzay) transformasyon matrisi olarak ~~gelişte~~ olsadır.

$$M\ddot{q}(t) + Kq(t) = Q(t) \rightarrow M' \ddot{\tilde{q}}(t) + K' \tilde{q}(t) = N(t)$$

$$M' = U^T M U = M'^T \rightarrow ?$$

$$K' = U^T K U = K'^T \rightarrow ?$$

mass normalisation?

$$N(t) = U^T Q(t)$$

$$\begin{cases} c_1 \cos(\omega_1 t + \phi_1) \\ c_2 \cos(\omega_2 t + \phi_2) \end{cases}$$

*Burdaki M' ve K' matrisleri modal uzaydaki stiffness ve mass matrisleridir.

Şimdi öyle bir U bulalım ki bu M' ve K' yi sırasıyla inertially ve elastically uncoupled yapalım. Böylece

Böyle bir U matrisi vardır adı modal matrisdir

ve doğal modaller içersi.

$$M'_\text{jj} \ddot{q}_j(t) + K'_{jj} q_j(t) = N_j(t) \text{ olsun.}$$

↳ independent equations are referred as modal equations.

↳ solution of these diff eqs called modal analysis

* Eger U , M' ve K' i diagonal yapan bir matris ise ikisine göre de ortogonaldır. Eger \tilde{M}' i bir I yapanın normalasyon yapmış olyorum.

ORTHOGONALITY OF MODAL VEC.

$$U_r^T K U_s = w_r^2 U_r^T M U_s \rightarrow K_{rs} = w_r^2 M_{rs} \leftarrow x = q = U\tilde{q} = [U_r, U_s] \tilde{q} \rightarrow Kx + M\ddot{x} = 0$$

$$U_r^T M U_s = w_s^2 U_r^T U_s \rightarrow M_{rs} = w_s^2 U_r^T U_s$$

↳ Zaten modal koordinatlara genis kitle ve yep matrislerini

decoupled hale getirmekti (modal coordinates cinsinden)

öyleyse modal vektörde oğrınca eline doğal faktörler n. modalın genelde farklılık bir durum

A.1 Summary Of Calculations MDOF System

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{x}_i} \right) - \frac{\partial L}{\partial x_i} = Q_i^m \rightarrow M\ddot{x} + Kx = Q \rightarrow x(K - \omega^2 M) = 0$$

$\omega = \omega^2 \rightarrow [E_{\omega_i^2}]$

$\begin{array}{l} \mathcal{L} = U^{-1} \cdot x \\ \downarrow \text{modal} \\ \mathcal{L} = U^T K U \\ \downarrow \text{modal stiffness} \\ \tilde{M} = U^T M U \\ \downarrow \text{modal Mass} \\ N(t) = U^T Q(t) \\ \downarrow \text{modal forces} \end{array}$

$$M\ddot{x} + Kx = Q$$

$$\mathcal{L}(-\omega U^T M U + U^T K U) = U^T Q$$

$$\mathcal{L}(-\omega \tilde{M} + \tilde{K}) = \tilde{N} \rightarrow \mathcal{L}(-[E_{\omega_i^2}] \tilde{M} + \tilde{K}) = \tilde{N}$$

$$\mathcal{L} \downarrow [E_{\omega_i^2}] \quad \tilde{N} \downarrow [E_{\omega_i^2}]$$

Aynı şekilde Emodal

$$E_{\text{top}} \in D[\mathcal{L}] \rightarrow E_{\text{modal}} \in D[\mathcal{L}] \quad \mathcal{L} = \frac{1}{2} \mathcal{L}^2 (U^T M U [E_{\omega_i^2}] + U^T K U) \rightarrow \frac{1}{2} \mathcal{L}^2 (E_{\omega_i^2} \tilde{M} + \tilde{K})$$

$$E_{\text{modal}} = \frac{1}{2} (\tilde{M} + E_{\omega_i^2} \tilde{K}^2)$$

burdan ağırlık (diagonalize edilmiş)
sistemde

$$E_{i,\text{modal}} = \frac{1}{2} (\tilde{K}_i + \omega_i^2 \tilde{M}_i) \quad i. \text{ modalları enerji}$$

$$E_{\text{modal}_i} = \frac{1}{2} C_i^2 \omega_i^2$$

ANALİTİK GÖRÜM:

$$q_r(t) = C_r \cos(\omega_r t - \phi_r) = \underbrace{C_r}_{\text{eğer mass normalize}} \underbrace{\cos(\omega_r t)}_{\text{etrafında dönerse}} + \underbrace{\frac{C_r}{\omega_r}}_{\text{sin } \omega_r t} \text{ Modal initial Conditions}$$

$$(U^T M U)$$

etrafında dönerse.

$$C_r = \sqrt{C_r^2 + \dot{C}_r^2}$$

$$\phi_r = \arctan\left(\frac{-\dot{C}_r}{C_r}\right)$$

$$\text{böyledice } q(t) = \sum_{r=1}^n (U_r^T M q(0) \cos \omega_r t + \frac{1}{\omega_r} U_r^T M \dot{q}(0) \sin \omega_r t) U_r$$

A.2 Example Problem (Validation of Operator and Function use in Mathematica)

"CONTENT"

"Mathematica Built-in Function Test using a problem which has a known solution
- Meireovitch // Fundamentals of Vibration // Two Degree Of Freedom Systems // Example 5.2 // Page 221"

"Question"

Example 5.2. Consider the simplified model of an automobile shown in Fig. 5.5, let the parameters have the values $m = 1,300 \text{ kg}$, $I_c = 2,000 \text{ kg m}^2$, $k_1 = 36,000 \text{ kg/m}$, $k_2 = 40,000 \text{ kg/m}$, $a = 1.3 \text{ m}$ and $b = 1.7 \text{ m}$, calculate the natural modes of the system and write an expression for the response.

"Equations Of Motion in Matrix Form"

$$\begin{bmatrix} 1,500 & 0 \\ 0 & 2,000 \end{bmatrix} \begin{bmatrix} \ddot{x} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 76,000 & 21,200 \\ 21,200 & 176,440 \end{bmatrix} \begin{bmatrix} x \\ \theta \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix} \quad (a)$$

But, free vibration is harmonic, so that by analogy with Eqs. (5.23) and (5.31) we can write

$$x(t) = X \cos(\omega t - \phi), \quad \theta(t) = \Theta \cos(\omega t - \phi) \quad (b)$$

"Results"

$$x(t) = C_1 \cos(6.857811t - \phi_1) \begin{bmatrix} 1 \\ -0.257341 \end{bmatrix} + C_2 \cos(9.584211t - \phi_2) \begin{bmatrix} 1 \\ 2.914417 \end{bmatrix} \quad (i)$$

where the amplitudes C_1 , C_2 and the phase angles ϕ_1 , ϕ_2 must be determined from the initial displacements $x(0)$, $\dot{x}(0)$ and the initial velocities $\theta(0)$, $\dot{\theta}(0)$, as will be shown in Sec. 5.4.

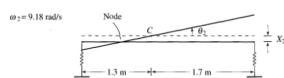
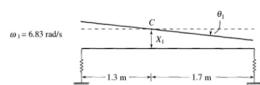


FIGURE 5.8
Natural modes for the automobile model of Fig. 5.5

"Mass Matrix"

$$\begin{pmatrix} 1500. & 0. \\ 0. & 2000. \end{pmatrix}$$

"Stiffness Matrix"

$$\begin{pmatrix} 76000. & 21200. \\ 21200. & 176440. \end{pmatrix}$$

"Eigen Frequencies"

$$\{6.85781, 9.58421\}$$

"Eigen Vectors"

$$\begin{pmatrix} -0.968447 & 0.324548 \\ 0.249221 & 0.945869 \end{pmatrix}$$

"Checking Orthogonality"

$$3.63798 \times 10^{-12}$$

"Mass Normalization Formula"

$$Us2 = \frac{Us}{\sqrt{Us \cdot M \cdot Us}}$$

"Mass Normalized Eigen Vectors"

$$\begin{pmatrix} -0.0247503 & 0.0073546 \\ 0.00636927 & 0.0214344 \end{pmatrix}$$

"Unit Normalized Eigen Vectors"

$$\begin{pmatrix} 1. & 1. \\ -0.257341 & 2.91442 \end{pmatrix}$$

A3. Comparision between Analytic Solution and Numeric Solution

```

M = {{1, 0, 0}, {0, 0.04, 0}, {0, 0, 0.06}}
{{1, 0, 0}, {0, 0.04, 0}, {0, 0, 0.06}}

K = {{1.26, -0.02, -0.24}, {-0.02, 0.02, 0}, {-0.24, 0, 0.24}}
{{1.26, -0.02, -0.24}, {-0.02, 0.02, 0}, {-0.24, 0, 0.24}}

```

Impulse Response 3DOF System

```

"ANALITIK COZUM"
q0 = {0, 0, 0};
dq0 = {1, 0, 0};
q = Sum[Um[[All, i]].M.q0 * Cos[w[i][i]*t] + 1/w[i][i]*Um[[All, i]].M.dq0 * Sin[w[i][i]*t] *
Um[[All, i]], {i, 1, 3}]

ANALITIK COZUM
(-0.193239 (0. Cos[0.693019 t] - 0.278837 Sin[0.693019 t]) +
0.932324 (0. Cos[0.982431 t] + 0.948997 Sin[0.982431 t]) -
0.305665 (0. Cos[2.07715 t] - 0.147156 Sin[2.07715 t]),
-0.98938 (0. Cos[0.693019 t] - 0.278837 Sin[0.693019 t]) -
1.00213 (0. Cos[0.982431 t] + 0.948997 Sin[0.982431 t]) +
0.0400656 (0. Cos[2.07715 t] - 0.147156 Sin[2.07715 t]),
-0.219607 (0. Cos[0.693019 t] - 0.278837 Sin[0.693019 t]) +
1.22883 (0. Cos[0.982431 t] + 0.948997 Sin[0.982431 t]) +
3.88695 (0. Cos[2.07715 t] - 0.147156 Sin[2.07715 t]))
```

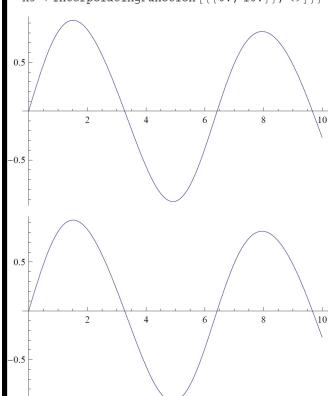
SolutionBySolver

```

TIME = 10;
X = {x1[t], x2[t], x3[t]};
D2X = D[X, {t, 2}];
Eqs = {0, 0, 0};

Table[Eqs[[i]] = (M.D2X + K.X)[[i]] == 0, {i, 1, 3}];
Eqs // MatrixForm
vars = {x1, x2, x3}
IC = {x1[0] == 0, x2[0] == 0, x3[0] == 0, x1'[0] == 1, x2'[0] == 0, x3'[0] == 0}
dsol = NDSolve[Flatten[{Eqs, IC}], vars, {t, 0, TIME}, AccuracyGoal -> 16,
PrecisionGoal -> 16, Method -> "ExplicitRungeKutta", MaxSteps -> \[Infinity]];
Plot[Evaluate[x1[t]], {t, 0, TIME}]
Plot[q[[1]], {t, 0, TIME}]
Plot[q[[1]] - Evaluate[x1[t]/.dsol], {t, 0, TIME}, PlotRange -> All]
SolutionBySolver
{x1, x2, x3}

{x1[0] == 0, x2[0] == 0, x3[0] == 0, x1'[0] == 1, x2'[0] == 0, x3'[0] == 0}
{({x1 -> InterpolatingFunction[{{0., 10.}}, <>]}, x2 -> InterpolatingFunction[{{0., 10.}}, <>],
x3 -> InterpolatingFunction[{{0., 10.}}, <>])}
```



Analytic Solution of Harmonic Excitation 3DOF ***

```

0.0538821 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.5 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t]) + 0.884772 (1.3737 Cos[0.5 t] - 1.3737 Cos[0.982431 t] + 0. i Sin[0.5 t] + 0. i Sin[0.982431 t]) + 0.0449804 (0.51104 Cos[2.07715 t] + 0. i Sin[0.5 t] + 0. i Sin[2.07715 t])
1.36585 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.693019 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t]) - 0.95102 (1.3737 Cos[0.5 t] - 1.3737 Cos[0.982431 t] + 0. i Sin[0.5 t] + 0. i Sin[0.982431 t]) - 0.0589589 (0.51104 Cos[0.5 t] - 0.51104 Cos[2.07715 t] + 0. i Sin[0.5 t] + 0. i Sin[2.07715 t])
0.0612345 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.693019 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t]) + 1.1616 (1.3737 Cos[0.5 t] - 1.3737 Cos[0.982431 t] + 0. i Sin[0.5 t] + 0. i Sin[0.982431 t]) - 0.571988 (0.51104 Cos[0.5 t] - 0.51104 Cos[2.07715 t] + 0. i Sin[0.5 t] + 0. i Sin[2.07715 t])

```

Harmonic Excitation 3DOF System

```

"FOR HARMONIC EXCITATIONS"
Q0 = {1, 0, 0};
\alpha = 0.5;
Q[t_] = Q0 * Cos[\alpha * t];
% // MatrixForm
NF = Table[Um[[All, i]].Q0 * Cos[\alpha * t], {i, 1, 3}];
% // MatrixForm

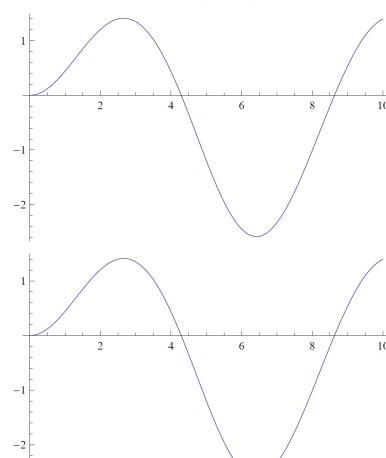
qharm = Sum[Um[[All, i]] / wi[[i]] *
Integrate[Q[t - \tau] * Sin[wi[[i]] * \tau], {\tau, 0, t}].Um[[All, i]], {i, 1, 3}];
% // MatrixForm
Plot[qharm[[1]], {t, 0, TIME}]
```

```

Table[Eqs[[i]] = (M.D2X + K.X)[[i]] == Q0[[i]] * Cos[\alpha * t], {i, 1, 3}];
IC = {x1[0] == 0, x2[0] == 0, x1'[0] == 0, x2'[0] == 0, x3'[0] == 0};
dsol2 = NDSolve[Flatten[{Eqs, IC}], vars, {t, 0, TIME}, AccuracyGoal -> 16,
PrecisionGoal -> 16, Method -> "ExplicitRungeKutta", MaxSteps -> \[Infinity]];
Plot[Evaluate[x1[t]] /. dsol2, {t, 0, TIME}]
Plot[qharm[[1]] - Evaluate[x1[t]] /. dsol2, {t, 0, TIME}, PlotRange -> All]
FOR HARMONIC EXCITATIONS
```

```

Cos[0.5 t]
0
0
(-0.193239 Cos[0.5 t])
0.932324 Cos[0.5 t]
-0.305665 Cos[0.5 t]
0.0538821 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.693019 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t])
1.36585 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.693019 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t])
0.0612345 (3.00952 Cos[0.5 t] - 3.00952 Cos[0.693019 t] + 0. i Sin[0.5 t] + 0. i Sin[0.693019 t])
*****
```



A.4 Calculation of Modal Coordinates and Energies

"Analytic Solution of Generalized and Modal Coordinates"

$$\eta h = \text{Table}\left[\frac{\sin(t \Omega[[i]]) \{Umt[[All, i]]\}.Mt.q0t}{\Omega[[i]]} + \cos(t \Omega[[i]]) \{Umt[[All, i]]\}.Mt.q0t, \{i, 1, nt\}\right];$$

$$q_h = \sum_{i=1}^{nt} \eta h[i].\{U_{mt}[All, i]\};$$

$$Nf = \text{Table}[\{Umt[\text{All}, i]\}.Qt, \{i, 1, nt\}]$$

$$\eta p = \text{Table}\left[\frac{\int_0^t \sin(\tau \Omega t[i]) (Nf[i] / . t \rightarrow t - \tau) d\tau}{\Omega t[i]}, \{i, 1, nt\}\right];$$

$$qp = \sum_{i=1}^{nt} np[i].\{Umt[All, i]\};$$

"Analytic Solution of Energies"

```
System(EperSat) = Table[0.5 (kt[[i]] (qt[[i]] - qt[[1]])2 + mt[[i]] qdot[[i]]2), {i, 2, nt}];
```

$$\text{System(ETot)} = 0.5 (\text{qt.Kt.qt} + \text{qdot.Mt.qdot});$$

$$E_{Force} = \int Q_t \cdot q_{dot} dt;$$

$$\mathbf{EF0} = \mathbf{EForcet} / . t \rightarrow 0;$$

$$\text{System(EForce)} = \text{EForcet} - \text{EF0};$$

System(EModal) =

$$\text{System(EImp)} = 0.5 (\mathbf{dq0t.Mt.dq0t})$$

Umt - Mass Normalized Eigen Vectors

etah - homogenous solution eta

etap - particular solution eta

N_f - modal forces transposed from generalized forces

qp - particular solution

qh - homogenous solution

A.5 Energy Check for Impulse Response (Time independent modal energy distribution).

```

In[1]: = evns = MatrixForm[Inverse[M].K];
U = Transpose[{Revers[eigs[[1]]]];
w2 = Revers[eigs[[2]]];
w3 = Revers[eigs[[3]]];
% // MatrixForm
U // MatrixForm
Out[1]:=
Table[
  U[[All, i]] = U[[All, i]] / Sqrt[U[[All, i]].M.U[[All, i]]]
,{i, 1, 3}];
% // MatrixForm

Out[1]//MatrixForm<=

$$\begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix} \rightarrow \boxed{\text{Uw}_1^2}$$


Out[15]//MatrixForm<=

$$\begin{pmatrix} -0.33849794 & -0.568683 & -0.7989355 \\ -0.998225 & 0.544494 & 0.0102794 \\ -0.04437528 & 0.686986 & 0.99887 \end{pmatrix} \rightarrow \boxed{\text{U (eベクトル)}}$$


Out[3]//MatrixForm<=

$$\begin{pmatrix} -0.104239 & 0.932324 & -0.305665 \\ -0.489075 & 0.052361 & 0.865651 \\ -0.218607 & 1.228883 & 3.88695 \end{pmatrix} \rightarrow \text{K' kitle normalize eベクトルler } \boxed{U_m}$$


Re[mat]=Round[Transpose[U].M.U, 0.000000000001];
% // MatrixForm
K = Round[Transpose[U].K.U, 0.0000000000001];
% // MatrixForm

Out[3]//MatrixForm<=

$$\begin{pmatrix} -0.0145288 & 0 & 0 \\ 0 & 0.29555 & 0 \\ 0 & 0 & 0.0657746 \end{pmatrix} \rightarrow \boxed{U^T = Nm = U^T M U} \quad \rightarrow \text{Nodal Transformasyona ugrayan kitle ve varyans matrisi}$$


Out[22]//MatrixForm<=

$$\begin{pmatrix} 0.0199452 & 0 & 0 \\ 0 & 0.285256 & 0 \\ 0 & 0 & 0.283788 \end{pmatrix} \rightarrow \boxed{K' = Km = U^T K U} \quad \rightarrow \text{Diagonalize edildi (Uncoupling)}$$


H[1] = (Km // MatrixForm, w2 // MatrixForm, Mm // MatrixForm)

Out[23]//MatrixForm<=

$$\begin{pmatrix} 0.0199452 & 0 & 0 \\ 0 & 0.285256 & 0 \\ 0 & 0 & 0.283788 \end{pmatrix} \rightarrow \boxed{Km}$$


Out[24]//MatrixForm<=

$$\begin{pmatrix} 0.488075 & 0 & 0 \\ 0 & 0.965317 & 0 \\ 0 & 0 & 0.134455 \end{pmatrix}, \begin{pmatrix} 0 & 0.0415288 & 0 \\ 0 & 0 & 0.29555 \\ 0 & 0 & 0 \end{pmatrix}, \begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix} \rightarrow \boxed{Nm}$$


H[2] = Round[Km.Mm, 0.000000000001] == 0
Round[Km.Mm, 0.0000000000001] == 0

Out[27]//MatrixForm<=

$$\begin{pmatrix} 2.76515 \times 10^{-15} & 0 & 0 \\ 0 & 3.33067 \times 10^{-15} & 0 \\ 0 & 0 & 2.10942 \times 10^{-15} \end{pmatrix} \rightarrow \boxed{Km - A Mm \approx 0}$$


Output: True

```

* ÖNEK PROBLEM ÜZERİNDEN DECOUPLING
MODAL TRANSFORMASYON-MODAL ENERJİ KAVRAMI
nin sonucu birbirine galişmesi.

$$\begin{aligned} \text{Yani: } E_{\text{top}} &= E_{\text{model}} \\ \Rightarrow \frac{1}{2} (x^T C x + x^T M x) &= \frac{1}{2} (w_c^T j^2 + j^2) \\ = \frac{1}{2} (w_1^2 C_1^2 + w_2^2 C_2^2 + w_3^2 C_3^2) & \quad C \text{ler boglong} \end{aligned}$$

***) Feki birde Primary'nin ayrı sotthalıkların ayrı model Enerjisi gözlemleniyor Neden? (Görünüş bı Neden, Enerji TOPLANI)**

A.6 Linearization Of Nonlinear Spring K(x)=k Log(1-x(t))

```

Yay Tipi:Sertlesen Yay
k Log[1 + x[t]]
Sistemin Toplam Kinetik Enerjisi:

$$\frac{1}{2} \sum_{i=2}^n m_i (x_i'(t))^2 + \frac{1}{2} M_R (x_R)'(t)^2$$

Sistemin Toplam Potansiyel Enerjisi:

$$\sum_{i=2}^n k_i (-x_i(t) + \log(1 + x_i(t) - x_R(t)) (1 + x_i(t) - x_R(t)) + x_R(t)) +$$


$$K_R (-x_R(t) + \log(1 + x_R(t)) (1 + x_R(t)))$$

Lagrange EİitiliEri:

$$-\sum_{i=2}^n k_i (-x_i(t) + \log(1 + x_i(t) - x_R(t)) (1 + x_i(t) - x_R(t)) + x_R(t)) +$$


$$\frac{1}{2} \sum_{i=2}^n m_i (x_i'(t))^2 - K_R (-x_R(t) + \log(1 + x_R(t)) (1 + x_R(t))) + \frac{1}{2} M_R (x_R)'(t)^2$$

Hareketin Diferansiyel Deklemleri Euler Lagrange Denklemlerinden Çıkarılması

$$\frac{\partial L}{\partial x_i(t)} - \frac{\partial}{\partial t} \frac{\partial L}{\partial x_i'(t)} = 0 \quad \frac{\partial L}{\partial x_R(t)} - \frac{\partial}{\partial t} \frac{\partial L}{\partial x_R'(t)} = 0$$


$$-\sum_{i=2}^n \log(1 + x_i(t) - x_R(t)) k_i - \frac{1}{2} \sum_{i=2}^n 2 m_i (x_i)''(t) = 0$$


$$-\log(1 + x_R(t)) K_R - \sum_{i=2}^n -\log(1 + x_i(t) - x_R(t)) k_i - M_R (x_R)''(t) = 0$$

Sistemin Linearize Edilmiş Hali:
Kinetik En:

$$\frac{1}{2} \sum_{i=2}^n m_i (x_i'(t))^2 + \frac{1}{2} M_R (x_R)'(t)^2$$

Pot. En:

$$\sum_{i=2}^n (0.0721318 k_i (x_i(t) - x_R(t)) + 0.333333 k_i (x_i(t) - x_R(t))^2) +$$


$$0.0721318 K_R x_R(t) + 0.333333 K_R x_R(t)^2$$

Lagrange Eq:

$$-\sum_{i=2}^n (0.0721318 k_i (x_i(t) - x_R(t)) + 0.333333 k_i (x_i(t) - x_R(t))^2) +$$


$$\frac{1}{2} \sum_{i=2}^n m_i (x_i'(t))^2 - 0.0721318 K_R x_R(t) - 0.333333 K_R x_R(t)^2 + \frac{1}{2} M_R (x_R)'(t)^2$$

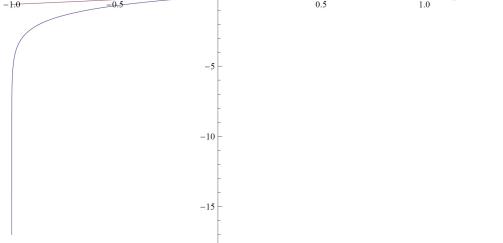

$$-\sum_{i=2}^n (0.0721318 k_i + 0.666667 k_i (x_i(t) - x_R(t))) - \frac{1}{2} \sum_{i=2}^n 2 m_i (x_i)''(t) = 0$$


$$-0.0721318 K_R -$$


$$\sum_{i=2}^n (-0.0721318 k_i - 0.666667 k_i (x_i(t) - x_R(t))) - 0.666667 K_R x_R(t) - M_R (x_R)''(t) = 0$$


```

[Spring Force vs Displacement F=, log(x(t)+1), $F_{\text{lin}}=$, 0.666667(x(t)-0.5)+0.405465], Force F[t]



A.7 Potential Functions Leonard Jones - Morse Potential

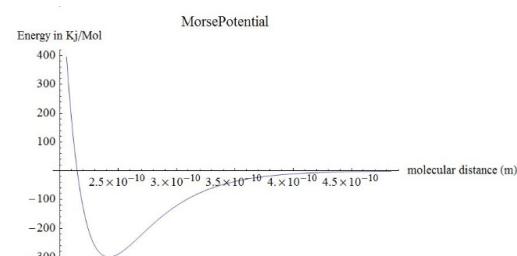
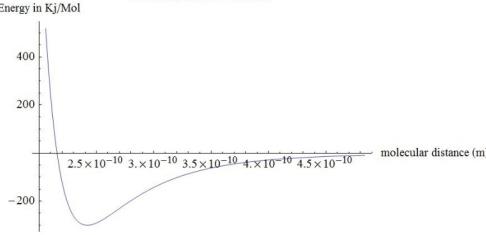
```

Avagadro=6.022142*10^23;
ze=2.42*10^-10; (*bond distance 2.42 A in m *)
Ebond=300; (*bond Energy 300 kJ/mol *)
κ=2.55*10^10; (*decay factor in 1/angstrom*)
L = -Ebond  $(2 \left(\frac{ze}{z}\right)^6 - \left(\frac{ze}{z}\right)^{12})$ 
V = -Ebond  $(2 \exp(-\kappa(z - ze)) - \exp(-2\kappa(z - ze)))$ 
Fm =  $\frac{\partial V}{\partial z}$ 

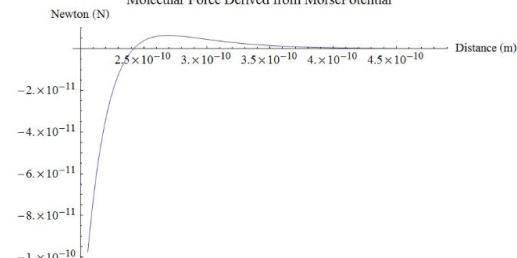
```

$$-4.98162 \times 10^{-22} \left(5.1 \times 10^{10} e^{-5.1 \times 10^{10} (-2.42 \times 10^{-10} - z)} - 5.1 \times 10^{10} e^{-2.55 \times 10^{10} (-2.42 \times 10^{-10} - z)}\right)$$

Leonard Jones Potential



Molecular Force Derived from MorsePotential



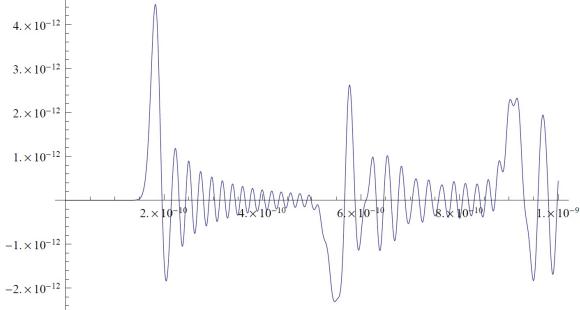
A.8 From Linear to Nonlinear - Linear Frequency Distribution with hardening spring K(x)=k (x(t) + eps x(t)^3)



A.9 Carbon x 32 Atom Lattice Simulation ~ A Work which is inspired from Dissipation in Solids Paper

Figures : First Atom - First Atom - Last Atom - All Atoms

Atomic Displacements



```

Avogadro = 6.022142 * 10^23;
ze = 2.42 * 10^-10; (*bond distance 2.42 Å in m *)
Ebond = 300; (*bond Energy 300 kJ/mol *)
κ = 2.55 * 10^-10; (*decay factor in 1/angstrom*)
m1 = 12 / Avogadro; (* weight of One Carbon Atom *)
IVel = 1; (* Initial Velocity of 1DOF model *)
IPos = ze; (*Initial Position of 1DOF atom *)
endtime = N[10^-8]; (*Simulation End Time*)

(*Formulas And Calculations*)
L = -Ebond * (2 + (ze / z)^6 - (ze / z)^12);
V = -Ebond * (2 + Exp[-κ*z] - Exp[-2 * κ * z]);
(*Vz=2/2; For test Purposes*)
Va = V / Avogadro;
Fm = D[V, z] / Avogadro;

n = 32; (*lattice atoms*)
IPos = ConstantArray[0, n];
IVel = ConstantArray[0, n]; (*Initial Position of 1DOF atom *)
IVel[[n]] = 1;

Z = Table[ToExpression[StringJoin["z", ToString[i], "[t]"]], {i, 1, n}];

CarbonAtomWeight = N[12 / Avogadro];
OxygenAtomWeight = N[16 / Avogadro];

Marray = Table[CarbonAtomWeight, {i, 1, n}];
M = DiagonalMatrix[Marray];

VPotTemp = Total[Table[Va /. z → Z[[i]] - Z[[i + 1]]], {i, 1, n - 1}];
VPot = (Va /. z → Z[[1]]) + VPotTemp;

TKin = Sum[0.5 * M[[i, i]] * D[Z[[i]], t]^2, {i, 1, n}];
Lagrange = TKin - VPot;

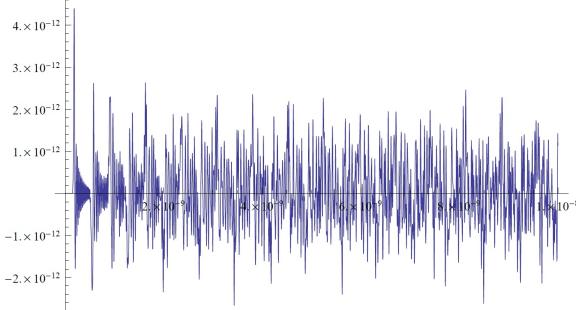
temp1 = Table[D[Lagrange, D[Z[[i]], t], t] - D[Lagrange, Z[[i]]] == 0, {i, 1, n}];
temp2 =
  Table[{(Z[[i]] /. t → 0) == IPos[[i]], (D[Z[[i]], t] /. t → 0) == IVel[[i]]}, {i, 1, n}];

Eqs = Flatten[{temp1, temp2}];

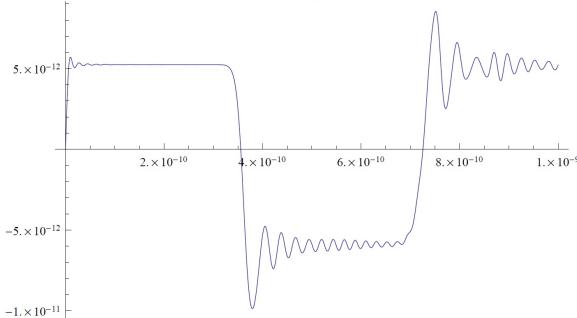
sol1MDOF =
  NDSolve[Eqs, Z, {t, 0, endtime}, MaxSteps → ∞, PrecisionGoal → 13, AccuracyGoal → 13];
Plot[Evaluate[Z[[1]] /. sol1MDOF], {t, 0, endtime}, PlotRange → All,
  PlotLabel → "Atomic Displacements", ImageSize → 420];
Plot[Evaluate[Z[[1]] /. sol1MDOF], {t, 0, endtime / 10}, PlotRange → All,
  PlotLabel → "Atomic Displacements", ImageSize → 420];
Plot[Evaluate[Z[[n]] /. sol1MDOF], {t, 0, endtime / 10}, PlotRange → All,
  PlotLabel → "Atomic Displacements", ImageSize → 420];
Plot[Evaluate[Z /. sol1MDOF], {t, 0, endtime / 10}, PlotRange → All,
  PlotLabel → "Atomic Displacements", ImageSize → 420]

```

Atomic Displacements



Atomic Displacements



Atomic Displacements

