

Questions

1. What happens to the drag of a jet aeroplane if, during the initial climb after take-off, a constant IAS and constant configuration is maintained? (Assume a constant mass.)
 - a. The drag decreases
 - b. The drag increases initially and decreases thereafter
 - c. The drag remains almost constant
 - d. The drag increases considerably

2. The speed for best rate of climb is called:
 - a. V_O
 - b. V_Y
 - c. V_X
 - d. V_2

3. An increase in atmospheric pressure has, among other things, the following consequences on take-off performance:
 - a. a reduced take-off distance and degraded initial climb performance
 - b. a reduced take-off distance and improved initial climb performance
 - c. an increased take-off distance and degraded initial climb performance
 - d. an increased take-off distance and improved initial climb performance

4. A higher outside air temperature:
 - a. does not have any noticeable effect on climb performance
 - b. reduces the angle of climb but increases the rate of climb
 - c. reduces the angle and the rate of climb
 - d. increases the angle of climb but decreases the rate of climb

5. In unaccelerated climb:
 - a. thrust equals drag plus the uphill component of the gross weight in the flight path direction
 - b. thrust equals drag plus the downhill component of the gross weight in the flight path direction
 - c. lift is greater than the gross weight
 - d. lift equals weight plus the vertical component of the drag

6. A jet aeroplane is climbing at a constant IAS with maximum climb thrust. How will the climb angle / the pitch angle change?
 - a. Remain constant / decrease
 - b. Remain constant / become larger
 - c. Reduce / decrease
 - d. Reduce / remain constant

7. Take-off performance data, for the ambient conditions, show the following limitations with flap 10° selected:

Runway or field limit mass: 5270 kg
Obstacle limit mass: 4630 kg

If the estimated take-off mass is 5000 kg, it would be prudent to consider a take-off with flaps at:

- a. 20°, both limitations are increased
 - b. 5°, the obstacle limit mass is increased but the runway limit mass decreases
 - c. 5°, both limitations are increased
 - d. 20°, the obstacle limit mass is increased but the runway limit mass decreases
8. A four jet engine aeroplane whose mass is 150 000 kg is established on climb with engines operating. The lift over drag ratio is 14:1. Each engine has a thrust of 75 000 Newtons.
The gradient of climb is: (given: $g = 10 \text{ m/s}^2$)
- a. 12.86%
 - b. 27%
 - c. 7.86%
 - d. 92%
9. How does the best angle of climb and best rate of climb vary with increasing altitude?
- a. Both decrease
 - b. Both increase
 - c. Best angle of climb increases while best rate of climb decreases
 - d. Best angle of climb decreases while best rate of climb increases
10. Following a take-off determined by the 50 ft (15 m) screen height, a light twin climbs on a 10% ground gradient. It will clear a 900 m high obstacle situated at 10 000 m from the 50 ft clearing point with an obstacle clearance of:
- a. 85 m
 - b. It will not clear the obstacle
 - c. 115 m
 - d. 100 m
11. The rate of climb:
- a. is approximately the climb gradient multiplied by the true airspeed divided by 100
 - b. is the downhill component of the true airspeed
 - c. is angle of climb multiplied by the true airspeed
 - d. is the horizontal component of the true airspeed

12. Assuming that the required lift exists, which forces determine an aeroplane's angle of climb?
- Thrust and drag only
 - Weight and thrust only
 - Weight, drag and thrust
 - Weight and drag only
13. Which of the following provides maximum obstacle clearance during climb?
- $1.2V_s$
 - The speed for maximum rate of climb
 - The speed at which the flaps may be selected one position further UP
 - The speed for maximum climb angle V_x
14. Which speed provides maximum obstacle clearance during climb?
- The speed which gives maximum excess thrust
 - $V_2 + 10 \text{ kt}$
 - The speed for maximum rate of climb
 - V_2
15. Which of the equations below expresses approximately the unaccelerated percentage climb gradient for small climb angles?
- Climb Gradient = $(\text{Thrust} - \text{Drag}) / \text{Weight} \times 100$
 - Climb Gradient = $(\text{Thrust} + \text{Drag}) / \text{Lift} \times 100$
 - Climb Gradient = $(\text{Thrust} - \text{Mass}) / \text{Lift} \times 100$
 - Climb Gradient = $\text{Lift} / \text{Weight} \times 100$
16. The absolute ceiling:
- is the altitude at which the best climb gradient attainable is 5%
 - is the altitude at which the aeroplane reaches a maximum rate of climb of 100 ft/min
 - is the altitude at which the rate of climb is theoretically zero
 - can be reached only with minimum steady flight speed
17. The climb gradient of an aircraft after take-off is 6% in standard atmosphere, no wind, at 0 ft pressure altitude.

Using the following corrections:

- $\pm 0.2\% / 1000 \text{ ft field elevation}$
- $\pm 0.1\% / ^\circ\text{C from standard temperature}$
- 1% with wing anti-ice
- 0.5% with engine anti-ice

The climb gradient after take-off from an airport situated at 1000 ft, 17°C ; QNH 1013.25 hPa, with wing and engine anti-ice operating for a functional check is:

- 3.9%
- 4.3%
- 4.7%
- 4.9%

18. As long as an aeroplane is in a positive climb:
- V_x is always below V_y
 - V_x is sometimes below and sometimes above V_y depending on altitude
 - V_x is always above V_y
 - V_y is always above V_{MO}
19. A constant headwind component:
- increases the angle of flight path during climb
 - increases the best rate of climb
 - decreases the angle of climb
 - increases the maximum endurance
20. A higher gross mass at the same altitude will cause:
- V_y and V_x to decrease
 - V_x to increase and V_y to decrease
 - V_y and V_x to remain constant since they are not affected by a higher gross mass
 - V_y and V_x to increase
21. With a true airspeed of 194 kt and a vertical speed of 1000 ft/min, the climb gradient is approximately:
- 3°
 - 3%
 - 5°
 - 8%
22. With take-off flaps set, V_x and V_y will be:
- lower than that for clean configuration
 - higher than that for clean configuration
 - the same as that for clean configuration
 - changed so that V_x increases and V_y decreases compared to clean configuration
23. The maximum rate of climb that can be maintained at the absolute ceiling is:
- 0 ft/min
 - 125 ft/min
 - 500 ft/min
 - 100 ft/min
24. A headwind will:
- increase the rate of climb
 - shorten the time of climb
 - increase the climb flight path angle
 - increase the angle of climb

25. V_x is:
- the speed for best rate of climb
 - the speed for best specific range
 - the speed for best angle of flight path
 - the speed for best angle of climb
26. The best rate of climb at a constant gross mass:
- decreases with increasing altitude since the thrust available decreases due to the lower air density
 - increases with increasing altitude since the drag decreases due to the lower air density
 - increases with increasing altitude due to the higher true airspeed
 - is independent of altitude
27. With a jet aeroplane, the maximum climb angle can be flown at approximately:
- $1.2V_s$
 - $1.1V_s$
 - the highest L/D ratio
 - the highest L/D² ratio
28. During a climb with all engines operating, the altitude where the rate of climb reduces to 100 ft/min is called:
- thrust ceiling
 - maximum transfer ceiling
 - service ceiling
 - absolute ceiling
29. With all other factors remaining constant, how does increasing altitude affect V_x and V_y as a TAS:
- V_x will decrease and V_y will increase
 - Both will increase
 - Both will remain the same
 - Both will decrease
30. Any acceleration in climb, with a constant power setting:
- improves the climb gradient if the airspeed is below V_x
 - improves the rate of climb if the airspeed is below V_y
 - decreases rate of climb and increases angle of climb
 - decreases the rate of climb and the angle of climb
31. For an aircraft maintaining 100 kt true airspeed and a climb gradient of 3.3% with no wind, what would be the approximate rate of climb?
- 3.30 m/s
 - 33.0 m/s
 - 330 ft/min
 - 3300 ft/min

32. During a climb to the cruising level, any headwind component:
- a. decreases the climb time
 - b. decreases the ground distance flown during that climb
 - c. increases the amount of fuel for the climb
 - d. increases the climb time
33. The pilot of a single-engine aircraft has established the climb performance. The carriage of an additional passenger will cause the climb performance to be:
- a. degraded
 - b. improved
 - c. unchanged
 - d. unchanged, if a short field take-off is adopted
34. A headwind component increasing with altitude, as compared to zero wind condition: (assuming IAS is constant.)
- a. improves angle and rate of climb
 - b. decreases angle and rate of climb
 - c. has no effect on rate of climb
 - d. does not have any effect on the angle of flight path during climb
35. Which of the following combinations adversely affects take-off and initial climb performance?
- a. High temperature and low relative humidity
 - b. Low temperature and low relative humidity
 - c. High temperature and high relative humidity
 - d. Low temperature and high relative humidity
36. A decrease in atmospheric pressure has, among other things, the following consequences on take-off performance:
- a. a reduced take-off distance and degraded initial climb performance
 - b. an increased take-off distance and degraded initial climb performance
 - c. a reduced take-off distance and improved initial climb performance
 - d. an increased take-off distance and improved initial climb performance
37. The angle of climb with flaps extended, compared to that with flaps retracted, will normally be:
- a. increased at moderate flap setting, decreased at large flap setting
 - b. smaller
 - c. larger
 - d. not changed
38. What is the effect of tailwind on the time to climb to a given altitude?
- a. The time to climb increases
 - b. The time to climb decreases
 - c. The effect on time to climb will depend on the aeroplane type
 - d. The time to climb does not change

39. Changing the take-off flap setting from flap 15° to flap 5° will normally result in:
- a. a longer take-off distance and a better climb
 - b. a shorter take-off distance and an equal climb
 - c. a better climb and an equal take-off distance
 - d. a shorter take-off distance and a better climb
40. What is the influence of the mass on maximum rate of climb (ROC) speed if all other parameters remain constant?
- a. The ROC is affected by the mass, but not the ROC speed
 - b. The ROC and the ROC speed are independent of the mass
 - c. The ROC speed increases with increasing mass
 - d. The ROC speed decreases with increasing mass
41. Following a take-off to the 50 ft (15 m) screen height, a light twin climbs on a gradient of 5%. It will clear a 160 m obstacle situated at 5000 m from the 50 ft point with an obstacle clearance margin of:
- a. it will not clear the obstacle
 - b. 105 m
 - c. 90 m
 - d. 75 m
42. The climb "gradient" is defined as the ratio of:
- a. true airspeed to rate of climb
 - b. rate of climb to true airspeed
 - c. the increase of altitude to horizontal air distance expressed as a percentage
 - d. the horizontal air distance over the increase of altitude expressed as a percentage

43. When flying an aircraft at:

- i. V_x without flap.
- ii. V_x with flap.
- iii. V_y without flap.
- iv. V_y with flap.

the aircraft should be achieving:

- a.
 - i. the best rate of climb
 - ii. the best rate of climb, but using a slightly faster speed than in (i)
 - iii. the best angle of climb
 - iv. the best angle of climb, but using a slightly faster speed than in (iii)
- b.
 - i. a good angle of climb
 - ii. the best angle of climb
 - iii. a good rate of climb
 - iv. the best rate of climb
- c.
 - i. the best angle of climb
 - ii. a slightly reduced angle of climb compared to (i) if using a slightly reduced speed than in (i)
 - iii. the best rate of climb
 - iv. a slightly reduced rate of climb compared to (iii) if using a slightly reduced speed than in (iii)
- d.
 - i. a good rate of climb
 - ii. the best rate of climb
 - iii. a good angle of climb
 - iv. the best angle of climb