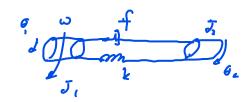
Design Example

This example is adapted from [1] and covered by the demo sateldem. The system is a satellite consisting of two rigid bodies (main body and instrumentation module) joined by a flexible link (the "boom"). The boom is modeled as a spring with torque constant k and viscous damping f and finite-element analysis gives the following uncertainty ranges for k and f:

The dynamical equations are



where θ_1 and θ_2 are the yaw angles for the main body and the sensor module, T is the control torque, and w is a torque disturbance on the main body.

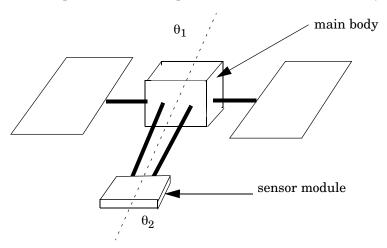


Figure 4-1: Satellite

The control purpose is to minimize the influence of the disturbance w on the angular position θ_2 . This goal is expressed through the following objectives:

1.

• Obtain a good trade-off between the RMS gain from w to θ_2 and the H_2 norm of the transfer function from w to



(LQG cost of control)

• Place the closed-loop poles in the region shown in Figure 4-3 to guarantee some minimum decay rate and closed-loop damping

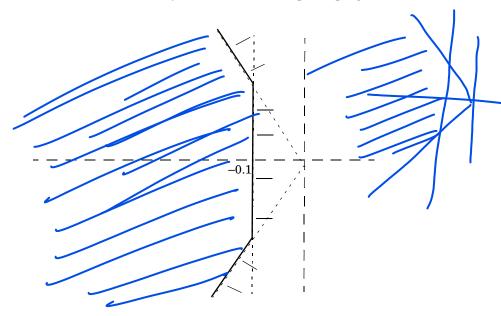


Figure 4-2: Pole placement region

 Achieve these objectives for all possible values of the varying parameters k and f. Since these parameters enter the plant state matrix in an affine manner, we can model the parameter uncertainty by a polytopic system with four vertices corresponding to the four combinations of extremal parameter values (see "From Affine to Polytopic Models" on page 2-20).

$$x_{p} = \begin{bmatrix} G_{1} \\ G_{2} \\ \vdots \\ G_{k} \end{bmatrix}$$

To solve this design problem with the LMI Control Toolbox, first specify the plant as a parameter-dependent system with affine dependence on k and f. A state-space description is readily derived from the dynamical equations as:

Next, specify the LMI region for pole placement as the intersection of the half-plane x < -0.1 and of the sector centered at the origin and with inner angle $3\pi/4$. This is done interactively with the function <code>lmireg</code>:

% parameter-dependent plant P= psys(pv,[ltisys(a0,b,c,d,e0) , ...

ltisys(ak,0*b,0*c,0*d,0) , ... ltisys(af,0*b,0*c,0*d,0)]) region = lmireg

To assess the trade-off between the H_{∞} and H_2 performances, first compute the optimal quadratic H_{∞} performance subject to the pole placement constraint by

gopt = msfsyn(P,[1 1],[0 0 1 0],region)This yields gopt ≈ 0 . For a prescribed H_{∞} performance g > 0, the best H_2 performance h2opt is computed by

[gopt,h2opt,K,Pcl] = msfsyn(P,[1 1],[g 0 0 1],region)Here obj = $[g \ 0 \ 0 \ 1]$ asks to optimize the H_2 performance subject to $||T_{\infty}||_{\infty}$ < g and the pole placement constraint. Repeating this operation for the values $g \in \{0.01, 0.1, 0.2, 0.5\}$ yields the Pareto-like trade-off curve shown in Figure 4-4.

By inspection of this curve, the state-feedback gain K obtained for g = 0.1 yields the best compromise between the H_{∞} and H_2 objectives. For this choice of K, Figure 4-5 superimposes the impulse responses from w to θ_2 for the four combinations of extremal values of k and f.

Finally, the closed-loop poles for these four extremal combinations are displayed in Figure 4-6. Note that they are robustly placed in the prescribed LMI region.

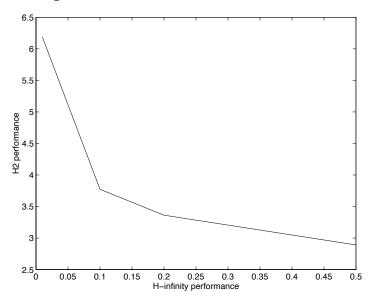


Figure 4-3: Trade-off between the H_{∞} and H_2 performances

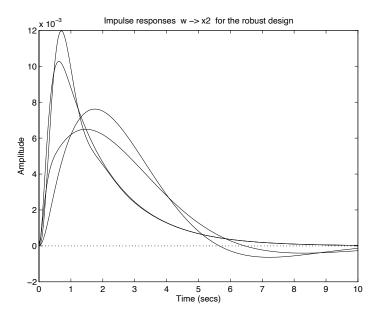


Figure 4-4: Impulse responses for the extremal values of \boldsymbol{k} and \boldsymbol{f}

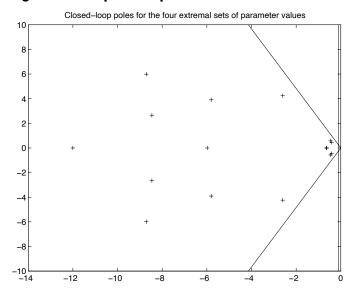


Figure 4-5: Corresponding closed-loop poles

References

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