

Final Exam (Chinmay Samak)

Solution 1:

$$1.5 \quad \delta_{sw} = G \cdot \delta \Rightarrow \delta = \delta_{sw} / G \quad (\text{rad})$$

$$\delta = \delta_A + \gamma u_a \stackrel{\circ}{=} \text{(neutral steer)} = \delta_A \Rightarrow \tan \delta = \frac{l}{R}$$

$$\Rightarrow R = \left| \frac{l}{\tan \delta} \right| = \left| \frac{l}{\tan(\delta_{sw}/G)} \right| \text{ (m)}$$

$$1.3 \quad \text{Stabilizing moment: } M = N_p \cdot \beta \stackrel{\circ}{=} \text{(Nm)} + N_r \cdot r \stackrel{\circ}{=} -12^\circ/\text{s} \stackrel{!}{=} 3200 \text{ lb.ft}$$

$$\Rightarrow \text{Yaw damping derivative: } N_r = M/r \quad (\text{Nm/s/rad})$$

$$1.4 \quad \Delta \beta \stackrel{\circ}{=} a c_f - b c_r \stackrel{!}{=} 0 \quad (\text{neutral steer})$$

$$N_r = \frac{a^2}{v} c_f + \frac{b^2}{v} c_r \quad (\text{soln 1.3})$$

$$\Rightarrow \begin{bmatrix} a & -b \\ a^2/v & b^2/v \end{bmatrix} \begin{bmatrix} c_f \\ c_r \end{bmatrix} = \begin{bmatrix} 0 \\ N_r \end{bmatrix} \quad \leftarrow \text{solve system of eq's}$$

$$1.2 \quad \text{Control moment derivative: } N\delta = -a c_f \quad (\text{Nm/rad})$$

$$1.1 \quad \text{Static stability derivative: } N\beta = a c_f - b c_r \stackrel{!}{=} 0 \quad (\text{neutral steer})$$

$$(\text{Nm/rad})$$

Table of Contents

Clear Workspace	1
Given Data	1
1.5	1
1.3	1
1.4	1
1.2	2
1.1	2

Clear Workspace

```
close all;
clear;
clc;
```

Given Data

```
l = 10/3.281; % Wheelbase (m)
a = 0.6*l; % Length of CG from front axle (m) for 40% front load
b = 0.4*l; % Length of CG from rear axle (m) for 40% front load
v = 66/3.281; % Velocity (m/s)
d = deg2rad(-40); % Steering wheel angle (rad)
G = 16; % Steering ratio
M = 3200*1.3558; % Stabilizing moment (Nm)
r = deg2rad(-12); % Yaw rate (rad/s)
```

1.5

```
R = abs(l/tan(d/G));
fprintf('Ans 1.5: The path radius (R) is %f m\n\n',R)
```

Ans 1.5: The path radius (R) is 69.807271 m

1.3

```
Nr = M/r;
fprintf('Ans 1.3: The yaw damping derivative (Nr) is %f Nms/rad\n\n',Nr)
```

Ans 1.3: The yaw damping derivative (Nr) is -20715.098097 Nms/rad

1.4

```
C = [a,-b;(a^2/v),(b^2/v)]\[0;Nr];
Cf = C(1);
Cr = C(2);
```

```
fprintf('Ans 1.4: The front (Cf) and rear (Cr) cornering stiffnesses are %f  
and %f N/rad\n\n',Cf,Cr)
```

*Ans 1.4: The front (Cf) and rear (Cr) cornering stiffnesses are -74762.860542
and -112144.290813 N/rad*

1.2

```
Nd = -a*Cf;  
fprintf('Ans 1.2: The control moment derivative (Nd) is %f Nm/rad\n\n',Nd)
```

Ans 1.2: The control moment derivative (Nd) is 136719.647440 Nm/rad

1.1

```
NB = a*Cf - b*Cr;  
fprintf('Ans 1.1: The static stability derivative (NB) is %f Nm/rad\n\n',NB)
```

Ans 1.1: The static stability derivative (NB) is 0.000000 Nm/rad

Published with MATLAB® R2022a

Solution 2:

$$V = 35 \text{ mph}$$

$$= 35/2.237 = 15.646 \text{ m/s}$$

$$a = -10 \text{ ft/sec}^2$$

$$= -10/3.281 = -3.048 \text{ m/s}^2$$

$$\theta = 30^\circ$$

$$= 30 \times (\pi/180) = 0.5236 \text{ rad}$$

$$\dot{\theta} = 10^\circ/\text{s}$$

$$= 10 \times (\pi/180) = 0.1745 \text{ rad/s}$$

$$b = 5 \text{ ft}$$

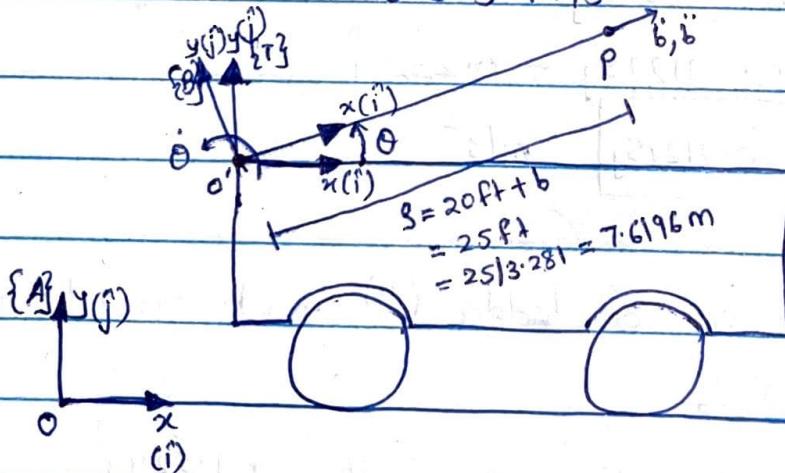
$$= 5/3.281 = 1.524 \text{ m}$$

$$\dot{b} = 2 \text{ ft/s}$$

$$= 2/3.281 = 0.6096 \text{ m/s}$$

$$\ddot{b} = -1 \text{ ft/s}^2$$

$$= -1/3.281 = -0.305 \text{ m/s}^2$$



By transport theorem

$$(a^P)_A = (a^o)_A + (a^r)_B + \alpha_{BA} \times s + 2\omega_{BA} \times (v^r)_B + \omega_{BA} \times (\omega_{BA} \times s)$$

Legend:
inertial acc (w.r.t ground)
relative acc (w.r.t truck)
body acc
tangential acc
Coriolis acc
centrifugal acc.

With respect to the established notations,

$$(\alpha^P)_B = \ddot{b} = -0.305 \hat{i} \text{ m/s}^2$$

$$\omega_{B/A} = 0 \hat{k} \text{ rad/s}$$

$$s = 7.6196 \hat{i} \text{ m}$$

$$\omega_{B/A} = \dot{\theta} = 0.1745 \hat{k} \text{ rad/s}$$

$$(v^P)_B = \dot{b} = 0.6096 \hat{i} \text{ m/s}$$

$$\begin{aligned} (\alpha^o)_A &= a \cos \theta \hat{i} - a \sin \theta \hat{j} = -3.048 \cos(30^\circ) \hat{i} + 3.048 \sin(30^\circ) \hat{j} \\ &= -2.6396 \hat{i} + 1.524 \hat{j} \text{ m/s}^2 \end{aligned}$$



2(a) Acceleration of the end of ladder (P) w.r.t. truck is

$$(\alpha^P)_T = (\alpha^o)_A + -(\alpha^o)_A$$

$$= (\alpha^P)_B + \omega_{B/A} \times v^P + 2\omega_{B/A} \times (v^P)_B + \omega_{B/A} \times (\omega_{B/A} \times s)$$

$$= -0.305 \hat{i} + 2[0.1745 \hat{k} \times 0.6096 \hat{i}] + 0.1745 \hat{k} \times [0.1745 \hat{k} \times 7.6196 \hat{i}]$$

$$= -0.305 \hat{i} + 0.21275 \hat{j} + [0.1745 \hat{k} \times 1.3296 \hat{j}]$$

$$= -0.305 \hat{i} + 0.21275 \hat{j} - 0.232 \hat{i}$$

$$\Rightarrow (\alpha^P)_T = -0.537 \hat{i} + 0.21275 \hat{j} \text{ m/s}^2$$

2(b) Acceleration of the end of ladder (P) w.r.t. ground is

$$(\alpha^P)_A = (\alpha^P)_T + (\alpha^o)_A$$

$$= -0.537 \hat{i} + 0.21275 \hat{j} - 2.6396 \hat{i} + 1.524 \hat{j}$$

$$\Rightarrow (\alpha^P)_A = -3.1766 \hat{i} + 1.73675 \hat{j} \text{ m/s}^2$$

Solution 3A:

$$3A \cdot A) \delta_A = \tan^{-1} \left(\frac{1}{R} \right) \quad (\text{rad})$$

$$3A \cdot B) a_y = \frac{V^2}{R} \quad (\text{m/s}^2)$$

$$3A \cdot C) \dot{\psi} = \gamma = \frac{V}{R} \quad (\text{rad/s})$$

$$3A \cdot D) F_y = m a_y \quad (\text{N})$$

$$3A \cdot E) \sum F_y : F_{y_f} + F_{y_r} - F_s - F_f = 0 \quad (\text{N})$$

$$\sum M_2 : F_{y_r} \cdot b - F_{y_f} \cdot a + F_s \cdot (a+d) = 0 \quad (\text{Nm})$$

$$\Rightarrow \begin{bmatrix} 1 & 1 & -1 \\ -a & b & (a+d) \end{bmatrix} \begin{bmatrix} F_{y_f} \\ F_{y_r} \\ F_s \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$$

$$\Rightarrow \begin{bmatrix} 1 & 1 \\ -a & b \end{bmatrix} \begin{bmatrix} F_{y_f} \\ F_{y_r} \end{bmatrix} = \begin{bmatrix} F_s + F_y \\ -F_s(a+d) \end{bmatrix} \quad \leftarrow \text{solve system of equations}$$

$$3A \cdot F) C_f = 2 (0.2196 (0.5 \times 0.4 \times w) - 7.35e-6 (0.5 \times 0.4 \times w)^2) \times 4.448 \times \frac{180}{\pi}$$

$$3A \cdot G) \quad \begin{array}{c} \text{front load} \\ \text{distrib} \end{array} \quad \begin{array}{c} \uparrow \\ 1b \end{array} \quad \begin{array}{c} \text{front load} \\ \text{distrib} \end{array} \quad \begin{array}{c} \uparrow \\ 1b \end{array} \quad 1b \rightarrow N \quad \text{deg}^{-1} \rightarrow \text{rad}^+$$

$$C_r = 2 (0.2196 (0.5 \times 0.6 \times w) - 7.35e-6 (0.5 \times 0.6 \times w)^2) \times 4.448 \times \frac{180}{\pi}$$

(N/rad)

$$\alpha_f = F_{y_f} / C_f \quad (\text{rad})$$

$$\alpha_r = F_{y_r} / C_r \quad (\text{rad})$$

$$3A \cdot H) M_2 = a F_{y_f} - b F_{y_r} \quad (\text{Nm})$$

$$3A \cdot I) \beta = \frac{b}{R} - \alpha_r \quad (\text{rad})$$

$$3A \cdot J) \delta = \delta_A + \alpha_f - \alpha_r \quad (\text{rad})$$

Table of Contents

Clear Workspace	1
Given Data	1
3A.A	1
3A.B	1
3A.C	2
3A.D	2
3A.E	2
3A.F	2
3A.G	2
3A.H	2
3A.I	3
3A.J	3

Clear Workspace

```
close all;
clear;
clc;
```

Given Data

```
w = 4e4*4.448; % Weight (N)
m = w/9.81; % Mass (kg)
l = 15/3.281; % Wheelbase (m)
a = 0.6*l; % Length of CG from front axle (m) for 40% front load
b = 0.4*l; % Length of CG from rear axle (m) for 40% front load
d = 4/3.281; % Length of front axle from snowplow (m)
v = 50/2.237; % Velocity (m/s)
R = 500/3.281; % Turning radius
Fs = 1000*4.448; % Snow force (N)
```

3A.A

```
dA = atan2(l,R);
fprintf('Ans 3A.A: The Ackerman steering angle (dA) is %f rad\n\n',dA)
```

Ans 3A.A: The Ackerman steering angle (dA) is 0.029991 rad

3A.B

```
ay = v^2/R;
fprintf('Ans 3A.B: The lateral acceleration (ay) is %f m/s^2\n\n',ay)
```

Ans 3A.B: The lateral acceleration (ay) is 3.278267 m/s^2

3A.C

```
r = v/R;
fprintf('Ans 3A.C: The yaw rate (r) is %f rad/s\n\n',r)
```

Ans 3A.C: The yaw rate (r) is 0.146670 rad/s

3A.D

```
Fy = m*ay;
Fyfr = [1,1;-a,b]\[(Fs+Fy);-Fs*(a+d)];
Fyf = Fyfr(1);
fprintf('Ans 3A.D: The front lateral force (Fyf) is %f N\n\n',Fyf)
```

Ans 3A.D: The front lateral force (Fyf) is 29416.771056 N

3A.E

```
Fyr = Fyfr(2);
fprintf('Ans 3A.E: The rear lateral force (Fyr) is %f N\n\n',Fyr)
```

Ans 3A.E: The rear lateral force (Fyr) is 34487.823251 N

3A.F

```
Cf = 2*(0.2196*(0.5*0.4*(w/4.448)) -
(7.35e-6*(0.5*0.4*(w/4.448))^2))*4.448*(180/pi);
af = Fyf/Cf;
fprintf('Ans 3A.F: The front slip angle (af) is %f rad\n\n',af)
```

Ans 3A.F: The front slip angle (af) is 0.044864 rad

3A.G

```
Cr = 2*(0.2196*(0.5*0.6*(w/4.448)) -
(7.35e-6*(0.5*0.6*(w/4.448))^2))*4.448*(180/pi);
ar = Fyr/Cr;
fprintf('Ans 3A.G: The rear slip angle (ar) is %f rad\n\n',ar)
```

Ans 3A.G: The rear slip angle (ar) is 0.042911 rad

3A.H

```
Mz = a*Fyf - b*Fyr;
fprintf('Ans 3A.H: The total yaw moment (Mz) is %f Nm\n\n',Mz)
```

Ans 3A.H: The total yaw moment (M_z) is 17623.895154 Nm

3A.I

```
B = b/R - ar;  
fprintf('Ans 3A.I: The vehicle sideslip angle (B) is %f rad\n\n',B)
```

Ans 3A.I: The vehicle sideslip angle (B) is -0.030911 rad

3A.J

```
d = dA + af - ar;  
fprintf('Ans 3A.J: The steering angle (d) is %f rad\n\n',d)
```

Ans 3A.J: The steering angle (d) is 0.031944 rad

Published with MATLAB® R2022a

Solution 3B:

$$3B.A \quad \delta_A = \tan^{-1} \left(\frac{1}{R} \right) \quad (\text{rad})$$

$$3B.B \quad a_y = v^2/R \quad (\text{m/s}^2)$$

$$3B.C \quad \dot{\psi} = r = v/R \quad (\text{rad/s})$$

$$3B.D \quad F_y = m_{ay} \quad (\text{N})$$

$$3B.E \quad \sum F_y: f_{y_f} + f_{y_r} - f_s - f_y = 0 \quad (\text{N})$$

$$\sum M_z: F_{y_f} \cdot b - F_{y_f} \cdot a + F_s (a+d) = 0 \quad (\text{Nm})$$

$$\begin{bmatrix} 1 & 1 \\ -a & b \end{bmatrix} \begin{bmatrix} F_{y_f} \\ f_{y_r} \end{bmatrix} = \begin{bmatrix} f_s + F_y \\ -F_s(a+d) \end{bmatrix} \quad \leftarrow \text{solve system of equations}$$

$$3B.F \quad S_f = 0.4 w/2 \quad (\text{N}) \quad | \quad S_r = 0.6 w/2 \quad (\text{N})$$

$$3B.G \quad F_{2fx} = S_f - \frac{m_{ay} h}{t} 0.45 \quad (\text{N}) \quad | \quad F_{2rx} = S_r - \frac{m_{ay} h}{t} 0.55 \quad (\text{N})$$

$$F_{2fr} = S_f + \frac{m_{ay} h}{t} 0.45 \quad (\text{N}) \quad | \quad F_{2rr} = S_r + \frac{m_{ay} h}{t} 0.55 \quad (\text{N})$$

$$C_{fl} = \left[0.2196 \cdot \left(\frac{F_{2fx}}{4.448} \right) - 7.35e-6 \left(\frac{F_{2rx}}{4.448} \right)^2 \right] 4.448 \times \frac{180}{\pi} \quad (\text{N/rad})$$

$$C_{fr} = \left[0.2196 \left(\frac{F_{2fr}}{4.448} \right) - 7.35e-6 \left(\frac{F_{2rr}}{4.448} \right)^2 \right] 4.448 \times \frac{180}{\pi} \quad (\text{N/rad})$$

$$C_{rl} = \left[0.2196 \left(\frac{F_{2rl}}{4.448} \right) - 7.35e-6 \left(\frac{F_{2rr}}{4.448} \right)^2 \right] 4.448 \times \frac{180}{\pi} \quad (\text{N/rad})$$

$$C_{rr} = \left[0.2196 \left(\frac{F_{2rr}}{4.448} \right) - 7.35e-6 \left(\frac{F_{2rr}}{4.448} \right)^2 \right] 4.448 \times \frac{180}{\pi} \quad (\text{N/rad})$$

$$c_f = c_{fx} + c_{fr} \quad (\text{N/rad})$$

$$c_r = c_{rx} + c_{rr} \quad (\text{N/rad})$$

$$\alpha_f = F_{yf} / c_f \quad (\text{rad})$$

$$\alpha_r = F_{yr} / c_r \quad (\text{rad})$$

$$3B\cdot H \quad M_2 = a F_{yf} - b F_{yr} \quad (\text{Nm})$$

$$3B\cdot I \quad \beta = \frac{b}{R} - \alpha_r \quad (\text{rad})$$

$$3B\cdot J \quad \delta = \delta_A + \alpha_f - \alpha_r \quad (\text{rad})$$

Table of Contents

Clear Workspace	1
Given Data	1
3B.A	1
3B.B	1
3B.C	2
3B.D	2
3B.E	2
3B.F	2
3B.G	2
3B.H	3
3B.I	3
3B.J	3

Clear Workspace

```
close all;
clear;
clc;
```

Given Data

```
w = 4e4*4.448; % Weight (N)
m = w/9.81; % Mass (kg)
l = 15/3.281; % Wheelbase (m)
t = 9/3.281; % Trackwidth (m)
h = 4.7/3.281; % Height of CG (m)
a = 0.6*l; % Length of CG from front axle (m) for 40% front load
b = 0.4*l; % Length of CG from rear axle (m) for 40% front load
d = 4/3.281; % Length of front axle from snowplow (m)
v = 50/2.237; % Velocity (m/s)
R = 500/3.281; % Turning radius
Fs = 1000*4.448; % Snow force (N)
```

3B.A

```
dA = atan2(l,R);
fprintf('Ans 3B.A: The Ackerman steering angle (dA) is %f rad\n\n',dA)
Ans 3B.A: The Ackerman steering angle (dA) is 0.029991 rad
```

3B.B

```
ay = v^2/R;
fprintf('Ans 3B.B: The lateral acceleration (ay) is %f m/s^2\n\n',ay)
Ans 3B.B: The lateral acceleration (ay) is 3.278267 m/s^2
```

3B.C

```
r = v/R;
fprintf('Ans 3B.C: The yaw rate (r) is %f rad/s\n\n',r)
```

Ans 3B.C: The yaw rate (r) is 0.146670 rad/s

3B.D

```
Fy = m*ay;
Fyfr = [1,1;-a,b]\[(Fs+Fy);-Fs*(a+d)];
Fyf = Fyfr(1);
fprintf('Ans 3B.D: The front lateral force (Fyf) is %f N\n\n',Fyf)
```

Ans 3B.D: The front lateral force (Fyf) is 29416.771056 N

3B.E

```
Fyr = Fyfr(2);
fprintf('Ans 3B.E: The rear lateral force (Fyr) is %f N\n\n',Fyr)
```

Ans 3B.E: The rear lateral force (Fyr) is 34487.823251 N

3B.F

```
Sf = (w*0.4)/2; % 40% front (static) load distribution
Fzfl = Sf - ((m*ay*h)/t)*0.45;
Fzfr = Sf + ((m*ay*h)/t)*0.45;
Cfl = (0.2196*((Fzfl/4.448)) - (7.35e-6*((Fzfl/4.448))^2))*4.448*(180/pi);
Cfr = (0.2196*((Fzfr/4.448)) - (7.35e-6*((Fzfr/4.448))^2))*4.448*(180/pi);
Cf = Cfl + Cfr;
af = Fyf/Cf;
fprintf('Ans 3B.F: The front slip angle (af) is %f rad\n\n',af)
```

Ans 3B.F: The front slip angle (af) is 0.047545 rad

3B.G

```
Sr = (w*0.6)/2; % 60% rear (static) load distribution
Fzrl = Sr - ((m*ay*h)/t)*0.55;
Fzrr = Sr + ((m*ay*h)/t)*0.55;
Crl = (0.2196*((Fzrl/4.448)) - (7.35e-6*((Fzrl/4.448))^2))*4.448*(180/pi);
Crr = (0.2196*((Fzrr/4.448)) - (7.35e-6*((Fzrr/4.448))^2))*4.448*(180/pi);
Cr = Crl + Crr;
ar = Fyr/Cr;
```

```
fprintf('Ans 3B.G: The rear slip angle (ar) is %f rad\n\n',ar)
```

```
Ans 3B.G: The rear slip angle (ar) is 0.046077 rad
```

3B.H

```
Mz = a*Fyf - b*Fyr;
```

```
fprintf('Ans 3B.H: The total yaw moment (Mz) is %f Nm\n\n',Mz)
```

```
Ans 3B.H: The total yaw moment (Mz) is 17623.895154 Nm
```

3B.I

```
B = b/R - ar;
```

```
fprintf('Ans 3B.I: The vehicle sideslip angle (B) is %f rad\n\n',B)
```

```
Ans 3B.I: The vehicle sideslip angle (B) is -0.034077 rad
```

3B.J

```
d = dA + af - ar;
```

```
fprintf('Ans 3B.J: The steering angle (d) is %f rad\n\n',d)
```

```
Ans 3B.J: The steering angle (d) is 0.031459 rad
```

Published with MATLAB® R2022a

Comparing Solution 3A and 3B

Parameter	Unit	Value (3A)	Value (3B)
δ_A	rad	0.029991	= 0.029991
a_y	m/s^2	3.278267	= 3.278267
$\dot{\psi}$	rad/s	0.146670	= 0.146670
F_{yf}	N	29416.771056	= 29416.771056
F_{yr}	N	34487.823251	= 34487.823251
α_F	rad	0.044864	< 0.047545
α_r	rad	0.042911	< 0.046077
M_2	Nm	17623.895154	= 17623.895154
β	rad	-0.030911	> -0.034077
δ	rad	0.031944	> 0.031459

From the above table it can be clearly seen that without (3A) or with (3B) load transfer taken into consideration, the parameters that change are α_f , α_r , β and δ . α_f and α_r increase as the vehicle has to now react with larger slip angles (in response to the load transfer). β is decreasing in -ve direction (cw) i.e. increasing in magnitude due to load transfer. Finally, δ decreases indicating that the understeering behavior of the vehicle is reduced.

Solution 4 A:

$$4A\cdot A \quad LF + RF + LR + RR - w = 0$$

$$LF + RF - w(w_{Front}) = 0$$

$$RF + RR - w(w_{Right}) = 0$$

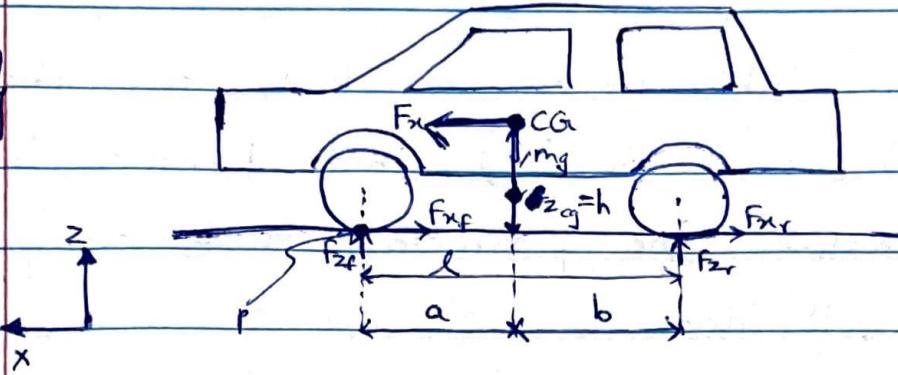
$$(LF + RR) - (RF + LR) - D = 0$$

4 eq's in 4 unknowns

Solved using MATLAB symbolic
toolbox

4A.B)

4A.C)



$$\text{Let } z_{cg} = h$$

Assume CG @ $l/2 \Rightarrow$ when $a_x = 0$, $|F_{zr}| = |F_{zf}| = |\frac{mg}{2}|$

Taking moment about point P (front tire contact): (↷)

$$0 = -mg\frac{l}{2} + |F_{zr}|l + m a_x h$$

$$\Rightarrow F_{zr} = \frac{mg}{2} - \frac{m a_x h}{l} \quad (\text{Rear})$$

$$\Rightarrow F_{zf} = \frac{mg}{2} + \frac{m a_x h}{l} \quad (\text{Front})$$

\uparrow Dynamic Wheel Load
 \uparrow Static Wheel Load
 \uparrow Load Transfer Due to a_x

$$\Delta W_n = \frac{m a_x h}{l} = \frac{W A_x h}{l}$$

$$\left\{ \begin{array}{l} W = mg \\ A_x = a/g \end{array} \right.$$

4A-D For all load to be transferred to front from rear in a straight line (i.e. no lateral load transfer),

$$F_{Zr} = 0$$

$$\Rightarrow \frac{mg}{2} - \frac{m a_n h}{l} = 0$$

$$\Rightarrow \frac{mg}{2} = \frac{m a_n h}{l}$$

$$\Rightarrow a_n = \frac{l g}{2h}$$

Generalizing above eqⁿ specific to $a = \frac{l}{2}$ to a general "a" distance we get:

$$a_n = \frac{g a}{h}$$

(m/s^2)

Table of Contents

Clear Workspace	1
4A.A	1
4A.B	1
4A.C	2
4A.D	2

Clear Workspace

```
close all;
clear;
clc;
```

4A.A

```
W = 143*14.594*9.81; % Total vehicle weight
WFront = 0.54; % Front weight ratio
WRear = 1-WFront; % Rear weight ratio
WLeft = 0.52; % Left weight ratio
WRight = 1-WLeft; % Right weight ratio
D = 500; % Diagonal weight (N)
syms LF RF LR RR
assume(LF>=0 & RF>=0 & LR>=0 & RR>=0); % Assume all static loads are positive
equation1 = LF+RF+LR+RR-W; % W = LF+RF+LR+RR
equation2 = LF+RF-(W*WFront); % WFront = ((LF+RF)/W)
equation3 = RF+RR-(W*WRight); % WRight = ((RF+RR)/W)
equation4 = (LF+RR)-(RF+LR)-D; % D = (LF+RR)-(RF+LR)
solution = solve(equation1, equation2, equation3, equation4); % Solve 4 equations
simultaneously
LF = double(solution.LF); % Left front wheel load (N)
RF = double(solution.RF); % Right front wheel load (N)
LR = double(solution.LR); % Left rear wheel load (N)
RR = double(solution.RR); % Right rear wheel load (N)
fprintf('Ans 4A.A: The static wheel loads for given vehicle parameters are LF
= %.2f N, RF = %.2f N, LR = %.2f N, RR = %.2f N\n\n', LF, RF, LR, RR)
```

Ans 4A.A: The static wheel loads for given vehicle parameters are LF = 5857.41 N, RF = 5197.95 N, LR = 4788.50 N, RR = 4629.04 N

4A.B

$$dWx = (m*ax*h)/l = (W*Ax*h)/l$$

```
fprintf('Ans 4A.B: Expression "dWx = (m*ax*h)/l = (W*Ax*h)/l" gives
longitudinal load transfer while acceleration/braking\n\n')
```

Ans 4A.B: Expression "dWx = (m*ax*h)/l = (W*Ax*h)/l" gives longitudinal load transfer while acceleration/braking

4A.C

```
m = 143*14.594; % Vehicle mass (kg)
ax = 0.5*9.81; % Longitudinal (braking) acceleration (m/s^2)
h = 2.3/3.281; % Height of CG (m)
l = 9.45/3.281; % Wheelbase (m)
dWx = (m*ax*h)/l; % Longitudinal load transfer
fprintf('Ans 4A.C: The load transferred from the rear tires to the front tires
when the vehicle is braking at 0.5g is %f N\n\n',dWx)
```

Ans 4A.C: The load transferred from the rear tires to the front tires when the vehicle is braking at 0.5g is 2491.411235 N

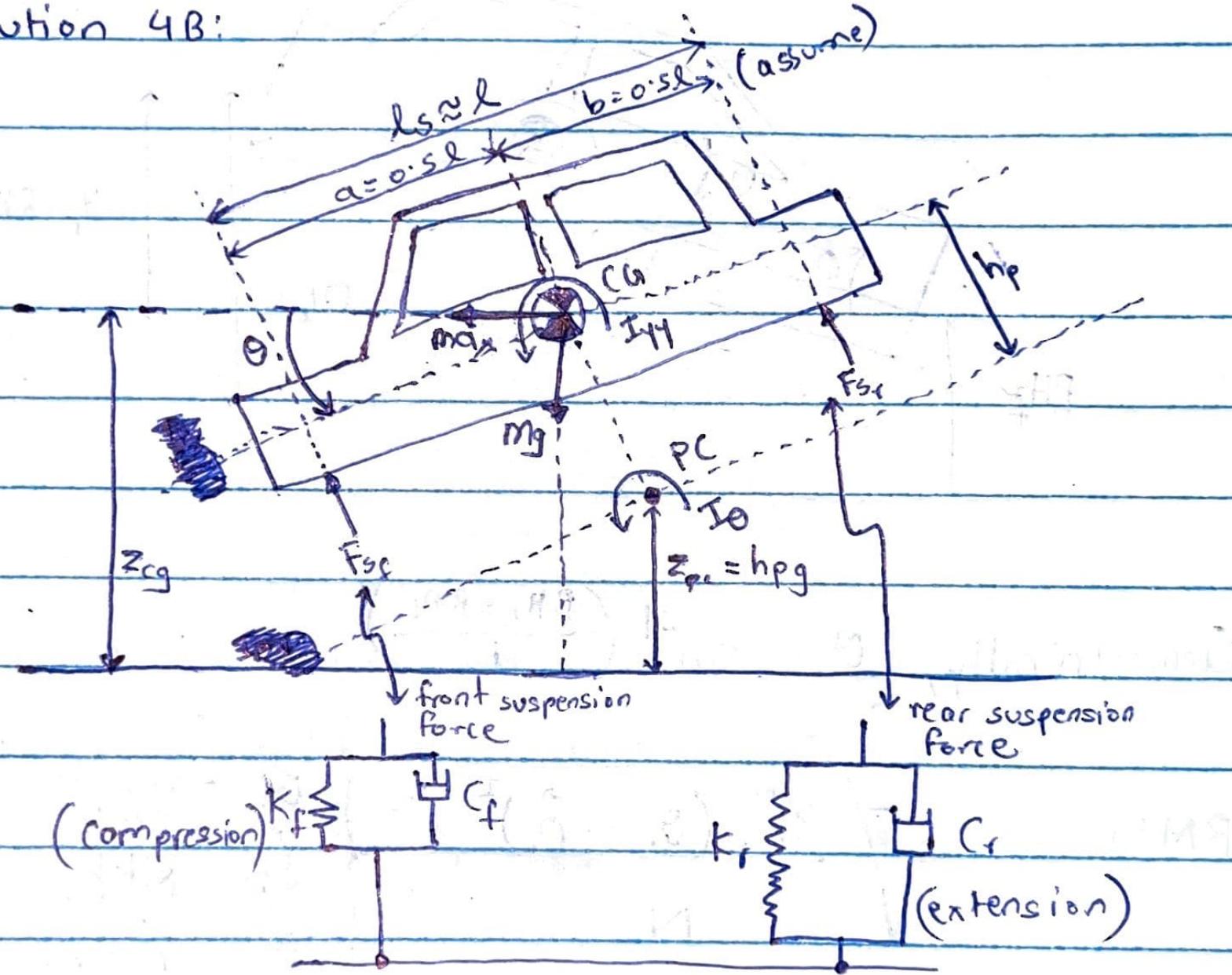
4A.D

```
l = 9.45/3.281; % Wheelbase (m)
g = 9.81; % Acceleration due to gravity (m/s^2)
a = (1-0.54)*l; % Distance of CG from front axle (m)
h = 2.3/3.281; % Height of CG (m)
ax = (g*a)/h;
fprintf('Ans 4A.D: At %f m/s^2 longitudinal acceleration all the load will be
transferred to front from rear in a straight-line motion\n\n',ax)
```

Ans 4A.D: At 18.540900 m/s^2 longitudinal acceleration all the load will be transferred to front from rear in a straight-line motion

Published with MATLAB® R2022a

Solution 4B:



Taking moment about PC (7)

$$(I_{yy} + m h_p^2) \ddot{\theta} = \sum M_y$$

$$= m \cdot a_x \cdot h_p \cdot \cos \theta + m g \cdot h_p \cdot \sin \theta - F_{sf} \frac{l_s}{2} + F_{sr} \frac{l_s}{2}$$

$$= m a_x h_p \cos \theta + m g h_p \sin \theta + \frac{l_s}{2} (F_{sr} - F_{sf}) \quad (1)$$

Deflection of sprung mass  (from undeflected stat)

$$z_{sf} = -\frac{l_s}{2} \sin \theta \quad (\text{front}) \quad - (2)$$

$$z_{sr} = +\frac{l_s}{2} \sin \theta \quad (\text{rear}) \quad - (3)$$

Velocity of sprung mass  (deflection derivative)

Differentiating ② & ③ we get

$$\dot{z}_{sf} = -\frac{l_s}{2} \cos \theta \dot{\theta} \quad (\text{front}) \quad - (4)$$

$$\dot{z}_{sr} = +\frac{l_s}{2} \cos \theta \dot{\theta} \quad (\text{rear}) \quad - (5)$$

Dynamic suspension forces (compression +)

From ②, ③, ④, ⑤ we get

$$F_{sp} = k_f \frac{l_s}{2} \sin\theta + c_f \frac{l_s}{2} \cos\theta \quad \text{--- ⑥}$$

$$F_{sr} = -k_r \frac{l_s}{2} \sin\theta - c_r \frac{l_s}{2} \cos\theta \quad \text{--- ⑦}$$

Spring
Damper

Substituting ⑥ & ⑦ in ① we get

$$(I_{yy} + m h_p^2) \ddot{\theta} = m a_n h_p \cos\theta + m g h_p \sin\theta -$$

$$\frac{1}{2} \left[k_r \frac{l_s}{2} \sin\theta + c_r \frac{l_s}{2} \cos\theta + k_f \frac{l_s}{2} \sin\theta + c_f \frac{l_s}{2} \cos\theta \right]$$

$$= m a_n h_p \cos\theta + m g h_p \sin\theta - \frac{k_f l_s^2 \sin\theta}{4} - \frac{k_r l_s^2 \sin\theta}{4} -$$

$$\frac{c_f l_s^2 \cos\theta}{4} - \frac{c_r l_s^2 \cos\theta}{4}$$

r = resultant

$$\text{Let } K_f = k_f l_s^2 / 4$$

$$K_r = k_r l_s^2 / 4$$

$$C_f = c_f l_s^2 / 4$$

$$C_r = c_r l_s^2 / 4$$

pitch axis

$$I_0 = I_{yy} + m h_p^2$$

~~Eqn ⑧~~

$$\Rightarrow I_0 \ddot{\theta} = m a_n h_p \cos\theta + m g h_p \sin\theta - k_f r \sin\theta - k_r r \sin\theta - C_f \cos\theta - C_r \cos\theta \quad \text{--- ⑧}$$

Applying small angle approximation on ⑧ we get

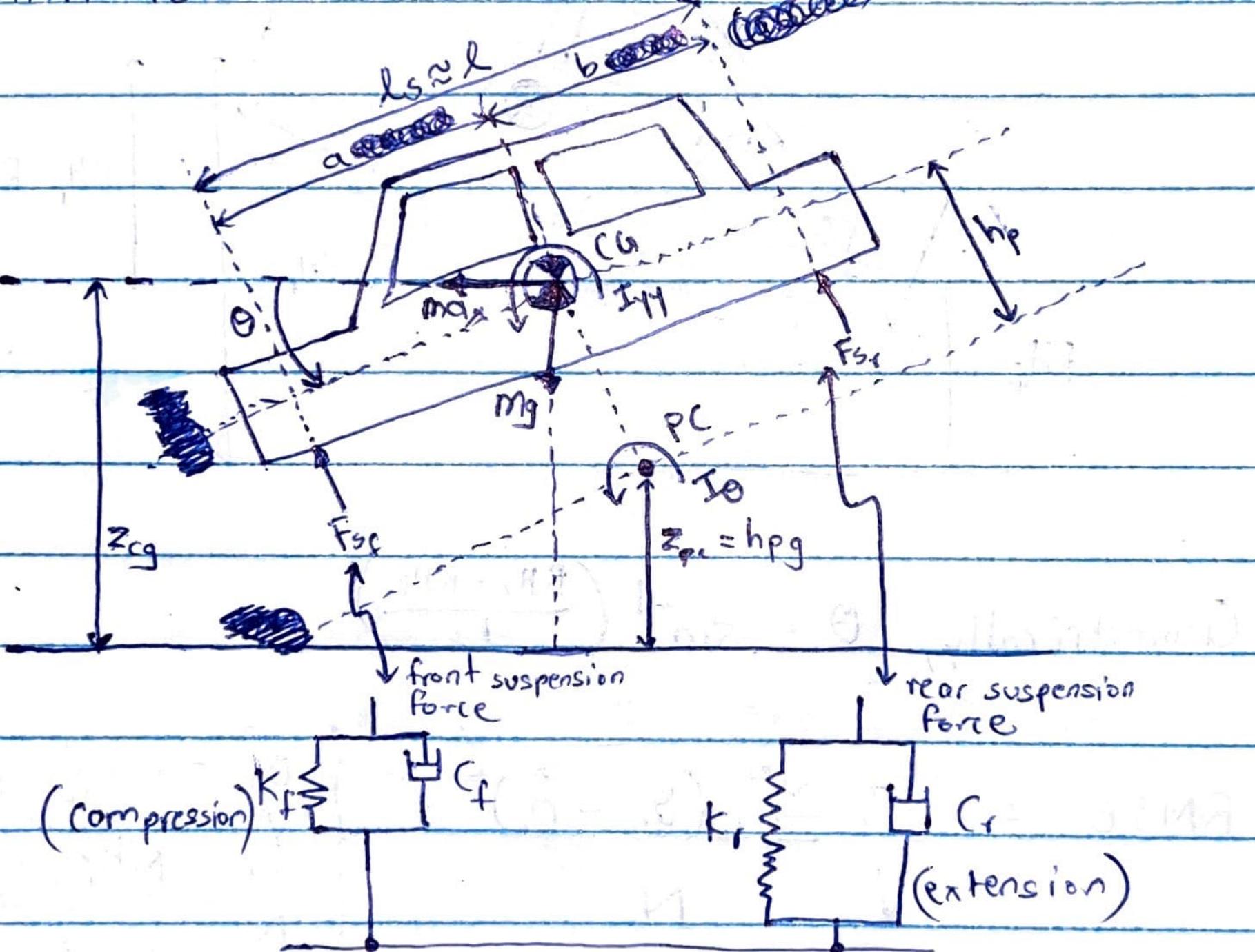
$$I_0 \ddot{\theta} = m a_n h_p \theta + m g h_p \theta - k_f r \theta - k_r r \theta - C_f \dot{\theta} - C_r \dot{\theta} \quad \text{--- ⑨}$$

At steady state, $\dot{\theta} = 0$ & $\ddot{\theta} = 0$

$$\Rightarrow m_a h_p + mg h_p \theta - k_{fr} \theta - k_{rr} \theta = 0$$

$$\Rightarrow \theta (k_{fr} + k_{rr} - mg h_p) = m_a h_p$$

$$\Rightarrow \boxed{\theta = \frac{m_a h_p}{(k_{fr} + k_{rr} - mg h_p)}} \quad (\text{rad})$$



Assuming a & b are not equal to 0.5 ls

Taking moment about PC (F)

$$(T_y + m h_p^2) \ddot{\theta} = \sum M_y$$

$$= m a h_p \cos \theta + m g h_p \sin \theta - a F_{sf} + b F_{sr} \quad \text{--- (1)}$$

Deflection of sprung mass (from undeflected state)

$$z_{sf} = -a \sin \theta \quad (\text{Front}) \quad \text{--- (2)}$$

$$z_{sr} = b \sin \theta \quad (\text{Rear}) \quad \text{--- (3)}$$

Velocity of sprung mass (deflection derivative)

Differentiating ② & ③ wrt time we get

$$\dot{z}_{sf} = -a \cos \theta \dot{\theta} \quad (\text{Front}) \quad \text{--- (4)}$$

$$\dot{z}_{sr} = b \cos \theta \dot{\theta} \quad (\text{Rear}) \quad \text{--- (5)}$$

Dynamic suspension forces (compression +)

From ④, ⑤, ④, ⑤ we have

$$F_{sf} = \underbrace{k_f a \sin \theta}_{\text{spring}} + \underbrace{c_f a \cos \theta \dot{\theta}}_{\text{damper}} \quad \text{--- (6)}$$

$$F_{sr} = -k_r b \sin \theta - c_r b \cos \theta \dot{\theta} \quad \text{--- (7)}$$

Substituting ⑥ + ⑦ in ① we get

$$\underbrace{(I_{yy} + mhp^2)}_{I_0} \ddot{\theta} = m a_n h_p \cos \theta + mg h_p \sin \theta - k_f a^2 \sin \theta - c_f a^2 \cos \theta \dot{\theta} - k_r b^2 \sin \theta - c_r b^2 \cos \theta \dot{\theta} \quad - ⑧$$

Applying small angle approximation on ⑧ we get

$$I_0 \ddot{\theta} = m a_n h_p \theta + mg h_p \theta - k_f a^2 \theta - c_f a^2 \dot{\theta} - k_r b^2 \theta - c_r b^2 \dot{\theta} \quad - ⑨$$

At steady state, $\ddot{\theta} = 0$ & $\dot{\theta} = 0$

$$\Rightarrow m a_n h_p + mg h_p \theta - k_f a^2 \theta - k_r b^2 \theta = 0$$

$$\Rightarrow \theta (k_f a^2 + k_r b^2 - m g h_p) = m a_n h_p$$

$$\Rightarrow \boxed{\theta = \frac{m a_n h_p}{(k_f a^2 + k_r b^2 - m g h_p)} \text{ (rad)}}$$

Solution 4C:

4C.A Assumed vehicle parameters from 4B

$$h_p = h - h_{pg} \quad @ \text{ zero roll}$$

W = trackwidth, L = wheelbase

$$I_{yy} = 0.4 \frac{m}{12} (L^2 + w^2)$$

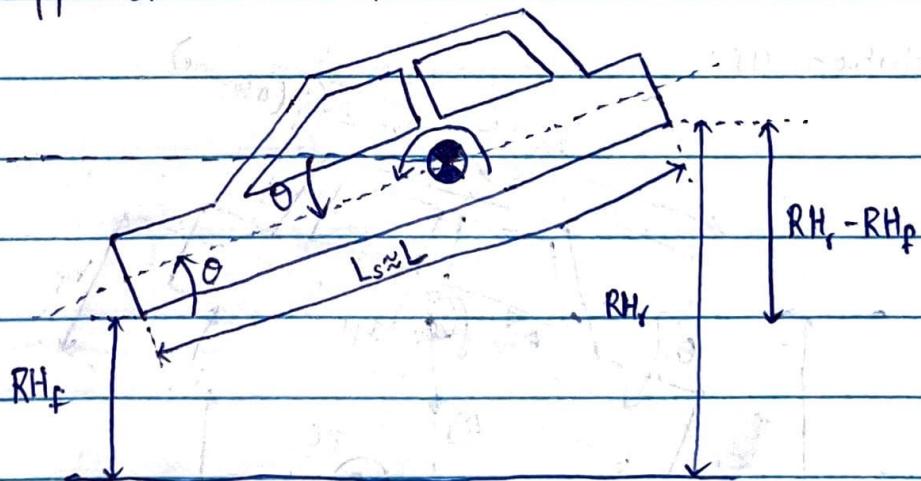
$$I_0 = I_{yy} + mh_p^2$$

$$\ddot{\theta} = \frac{1}{I_0} [m a_x h_p + mg h_p \dot{\theta} - K_r \theta - K_{rr} \dot{\theta} - C_{fr} \dot{\theta} - C_{tr} \dot{\theta}]$$

$$\theta_{t+1} = \theta_t + \ddot{\theta}_t \Delta t \quad \text{Forward Euler method.}$$

$$\theta_{t+1} = \theta_t + \dot{\theta}_t \Delta t$$

4C.B Similar approach as 4C.A



$$\text{Geometrically, } \theta = \sin^{-1} \left(\frac{RH_r - RH_f}{L} \right)$$

$$RMSE = \sqrt{\frac{\sum_{i=1}^N (\theta_i - \hat{\theta}_i)^2}{N}}$$

θ_i = recorded/measured pitch angle
 $\hat{\theta}_i$ = pitch angle updated using model developed in 4B
 N = number of data points

4C.A

```
close all;
clear;
clc;

% Data assumed from 4A
m = 143*14.594; % Vehicle mass (kg)
g = 9.81; % Acceleration due to gravity (m/s^2)
h = 2.3/3.281; % Height of CG (m)
L = 9.45/3.281; % Wheelbase (m)
W = 5.25/3.281; % Trackwidth (m)
a = (1-0.54)*L; % Distance of CG from front axle (m)
b = L-a; % Distance of CG from rear axle (m)

% Data given in 4C
ax = 2; % Longitudinal acceleration (m/s^2) (braking assumed positive in 4B)
hpg = 1.5/3.281; % Height of PC from ground (m)
Kf = 45e3; % Front suspension stiffness (N/m)
Kr = 52e3; % Rear suspension stiffness (N/m)
Cf = 0; % Front suspension damping (Ns/m)
Cr = 0; % Rear suspension damping (Ns/m)

% Intermediate variables
hp = h - hpg; % Height of PC from CG (m)
Iyy = 0.4*((m/12)*(L^2+W^2)); % Pitch MOI about CG (kg-m^2)
Itheta = Iyy + m*hp^2; % Pitch MOI about PC (kg-m^2)

% Simulation settings
ms = 1*m; % Sprung mass assumed to be 100% of total vehicle mass (kg)
theta_sim = [0]; % Initial pitch angle assumption (rad)
theta_dot = [0]; % Initial pitch velocity assumption (rad/s)
dt = 1e-6; % Simulation timestep (s)
i = 1; % Iterator count

% Simulate from i=0 to i=5 (@ 2 m/s^2 braking acceleration)
for t = 0+dt:dt:5
    theta_ddot(i) = 1/Itheta * (ms*ax*hp + ms*g*hp*theta_sim(i) -
    Kf*a^2*theta_sim(i) - Kr*b^2*theta_sim(i) - Cf*a^2*theta_dot(i) -
    Cr*b^2*theta_dot(i));
    theta_dot(i+1) = theta_dot(i) + theta_ddot(i)*dt;
    theta_sim(i+1) = theta_sim(i) + theta_dot(i)*dt;
    i = i+1;
end

% Plot theta vs t
figure()
plot((0:dt:5),theta_sim,'.', 'color', 'red')
xlabel('${t}$(sec)', 'interpreter', 'latex')
ylabel('${\theta}$(rad)', 'interpreter', 'latex')

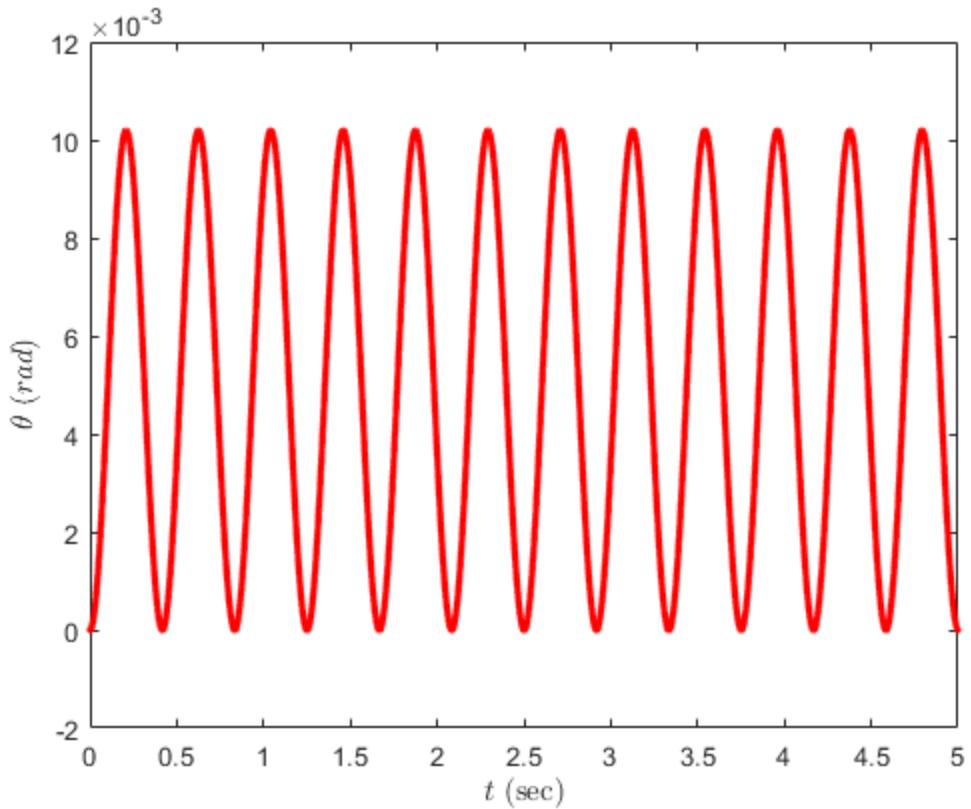
% Plot theta_dot vs t
```

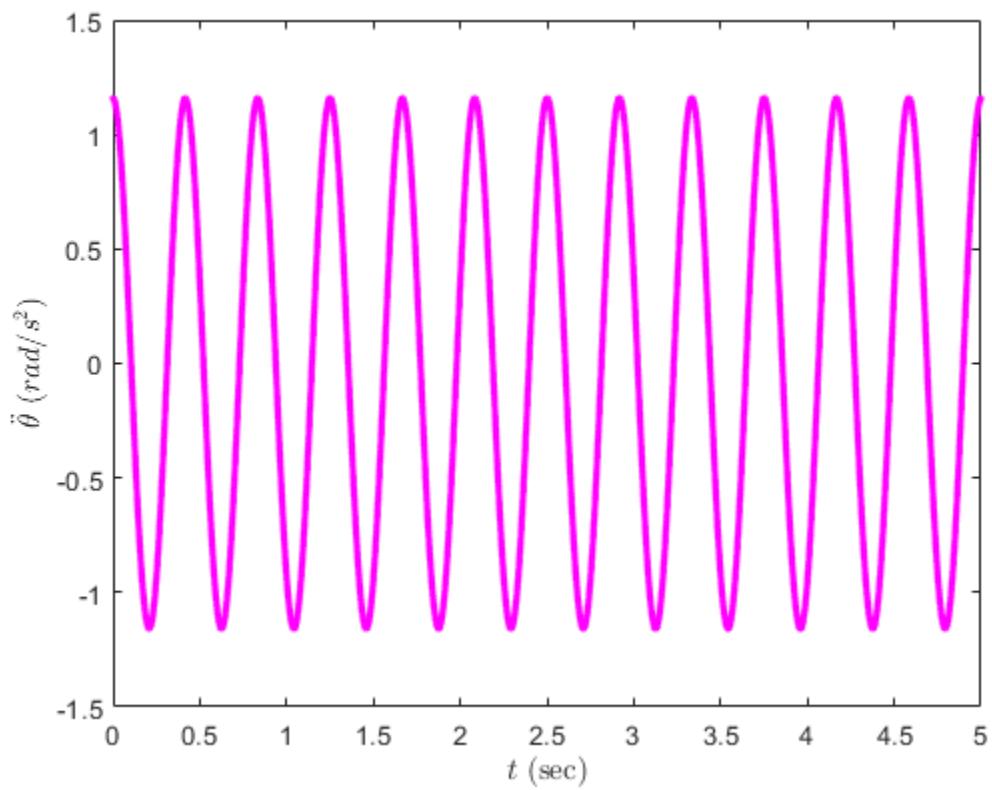
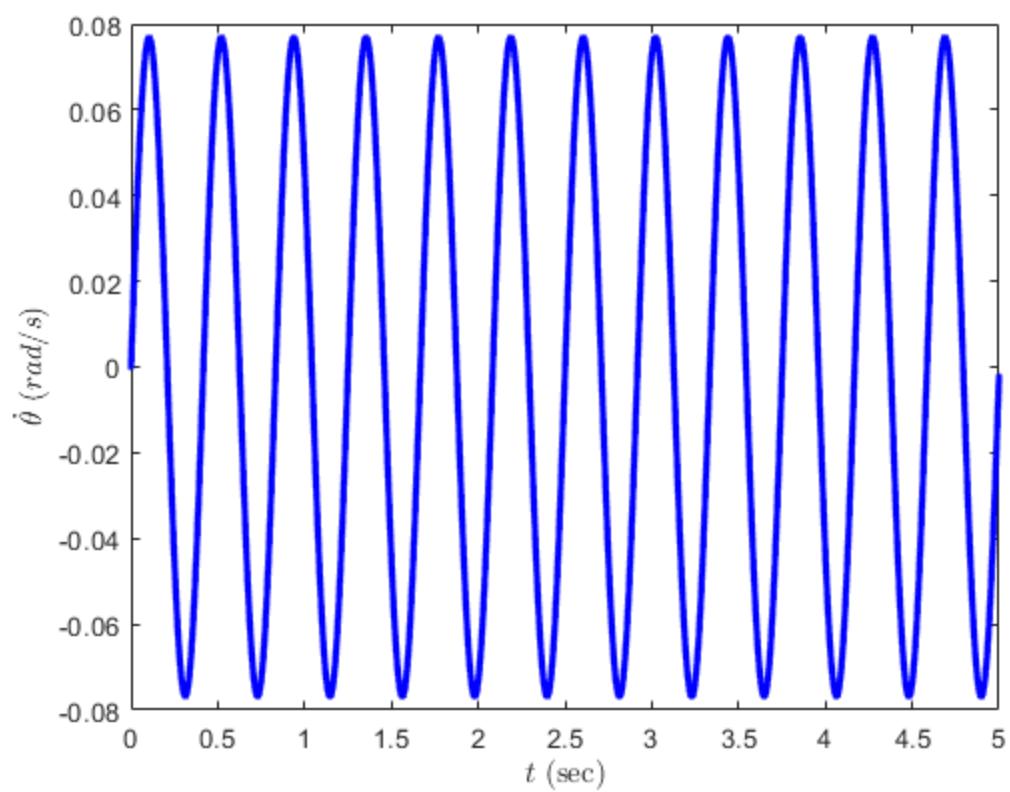
```

figure()
plot((0:dt:5),theta_dot,'.','color','blue')
xlabel('${t} \text{ (sec)}', 'interpreter', 'latex')
ylabel('${\dot{\theta}} \text{ ($rad/s$)}', 'interpreter', 'latex')

% Plot theta_ddot vs t
figure()
plot((dt:dt:5),theta_ddot,'.','color','magenta')
xlabel('${t} \text{ (sec)}', 'interpreter', 'latex')
ylabel('${\ddot{\theta}} \text{ ($rad/s^2$)}', 'interpreter', 'latex')

```





4C.B

```
clear;
clc;

% Data assumed from 4A
m = 143*14.594; % Vehicle mass (kg)
g = 9.81; % Acceleration due to gravity (m/s^2)
h = 2.3/3.281; % Height of CG (m)
L = 9.45/3.281; % Wheelbase (m)
W = 5.25/3.281; % Trackwidth (m)
a = (1-0.54)*L; % Distance of CG from front axle (m)
b = L-a; % Distance of CG from rear axle (m)

% RECORDED DATA
load('FinalExamData.mat'); % Load given data
theta_rec = asin((Rh_lr-Rh_lf)./L);

% (SUB)OPTIMAL ESTIMATION OF SUSPENSION DAMPING COEFFICIENTS (ASSUMING THEY
% ARE EQUAL)
ax = Acc_x; % Longitudinal acceleration (m/s^2) (braking assumed positive in
% 4B)
hpg = 1.5/3.281; % Height of PC from ground (m)
Kf = 45e3; % Front suspension stiffness (N/m)
Kr = 52e3; % Rear suspension stiffness (N/m)
hp = h - hpg; % Height of PC from CG (m)
Iyy = 0.4*((m/12)*(L^2+W^2)); % Pitch MOI about CG (kg-m^2)
Itheta = Iyy + m*hp^2; % Pitch MOI about PC (kg-m^2)
ms = 1*m; % Sprung mass assumed to be 100% of total vehicle mass (kg)
theta_sim = [0]; % Initial pitch angle assumption (rad)
theta_dot = [0]; % Initial pitch velocity assumption (rad/s)
dt = time(2) - time(1); % Simulation timestep (s)
i = 1; % Iterator count
C_optimal = 0;
error_minimum = Inf;
for C = 0:10:1e6
    i = 1;
    for t = 0+dt:dt:time(end)
        theta_ddot(i) = 1/Itheta * (ms*ax(i)*hp + ms*g*hp*theta_sim(i)
        - Kf*a^2*theta_sim(i) - Kr*b^2*theta_sim(i) - C*a^2*theta_dot(i) -
        C*b^2*theta_dot(i));
        theta_dot(i+1) = theta_dot(i) + theta_ddot(i)*dt;
        theta_sim(i+1) = theta_sim(i) + theta_dot(i)*dt;
        i = i+1;
    end
    error = sqrt((sum((theta_rec - theta_sim).^2))/length(theta_rec)); % RMSE
    if error < error_minimum
        error_minimum = error;
        C_optimal = C;
    end
end
fprintf(['Ans 4C.B: The estimated suspension damping coefficients (assumung
they are equal)\n' ...
```

```

'considering RMSE between given data and simulation results are %f
Ns/m\n\n'],C_optimal)

% SIMULATION
ax = Acc_x; % Longitudinal acceleration (m/s^2) (braking assumed positive in
4B)
hpg = 1.5/3.281; % Height of PC from ground (m)
Kf = 45e3; % Front suspension stiffness (N/m)
Kr = 52e3; % Rear suspension stiffness (N/m)
Cf = C_optimal; % Front suspension damping (Ns/m)
Cr = C_optimal; % Rear suspension damping (Ns/m)
hp = h - hpg; % Height of PC from CG (m)
Iyy = 0.4*((m/12)*(L^2+W^2)); % Pitch MOI about CG (kg-m^2)
Itheta = Iyy + m*hp^2; % Pitch MOI about PC (kg-m^2)
ms = 1*m; % Sprung mass assumed to be 100% of total vehicle mass (kg)
theta_sim = [0]; % Initial pitch angle assumption (rad)
theta_dot = [0]; % Initial pitch velocity assumption (rad/s)
dt = time(2) - time(1); % Simulation timestep (s)
i = 1; % Iterator count
for t = 0:dt:time(end)
    theta_ddot(i) = 1/Itheta * (ms*ax(i)*hp + ms*g*hp*theta_sim(i) -
    Kf*a^2*theta_sim(i) - Kr*b^2*theta_sim(i) - Cf*a^2*theta_dot(i) -
    Cr*b^2*theta_dot(i));
    theta_dot(i+1) = theta_dot(i) + theta_ddot(i)*dt;
    theta_sim(i+1) = theta_sim(i) + theta_dot(i)*dt;
    i = i+1;
end

% Plot theta vs t
figure()
plot((0:dt:time(end)),theta_rec,'.', 'color', 'blue')
hold on
plot((0:dt:time(end)),theta_sim,'.', 'color', 'red')
xlabel('${t} \text{ (sec)}', 'interpreter', 'latex')
ylabel('${\theta} \text{ ($rad$)}', 'interpreter', 'latex')
legend('${\theta}_{\text{recorded}}$', '${\theta}_{\text{simulated}}$', 'interpreter', 'latex')
hold off

% Plot theta_dot vs t
figure()
plot((0:dt:time(end)),theta_dot,'.', 'color', 'blue')
xlabel('${t} \text{ (sec)}', 'interpreter', 'latex')
ylabel('${\dot{\theta}} \text{ ($rad/s$)}', 'interpreter', 'latex')

% Plot theta_ddot vs t
figure()
plot((dt:dt:time(end)),theta_ddot,'.', 'color', 'magenta')
xlabel('${t} \text{ (sec)}', 'interpreter', 'latex')
ylabel('${\ddot{\theta}} \text{ ($rad/s^2$)}', 'interpreter', 'latex')

```

*Ans 4C.B: The estimated suspension damping coefficients (assumung they are equal)
considering RMSE between given data and simulation results are 18470.000000
Ns/m*

