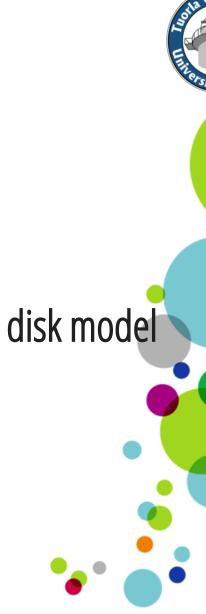




Ilia Kosenkov, 2018-01-19 @ Tuorla







# Starting from those scary equations....

1. 
$$ho=\Sigma/H$$

2. 
$$H = c_{
m s} R^{3/2}/(GM)^{1/2}$$

3. 
$$c_{
m s}^2=P/
ho$$

4. 
$$P = \frac{
ho k T_{
m c}}{\mu m_{
m p}} + \frac{4\sigma}{3c} T_{
m c}^4$$

5. 
$$\frac{4\sigma T_{
m c}^4}{3 au}=rac{3GM\dot{M}}{8\pi R^3}iggl[1-\left(rac{R_*}{R}
ight)^{1/2}iggr]$$

6. 
$$au = \Sigma \kappa_{
m R}(
ho, T_{
m c}) = au(\Sigma, 
ho, T_{
m c})$$

7. 
$$u\Sigma = \frac{\dot{M}}{3\pi} \left[ 1 - \left( \frac{R_*}{R} \right)^{1/2} \right]$$

8. 
$$u = \nu(\rho, T_{\rm c}, \Sigma, \alpha, \dots)$$





#### ... lets think of the emitted disk radiation.

Providing disc is optically thick in z-direction, each element of the disc face radiates as (approximately) a blackbody of some temperature T(R).

This temperature profile can be obtained from dissipation rate D(R)

$$\sigma T^4(R) = D(R)$$
 , which gives

$$T(R) = \left\{rac{3GM\dot{M}}{8\pi R^3\sigma}iggl[1-\left(rac{R_*}{R}
ight)^{1/2}iggr]
ight\}^{1/4}$$





#### For $R>>R_{st}$

$$T=T_st(R/R_st)^{-3/4}$$
 , where

$$T_* = \left(rac{3GM\dot{M}}{8\pi R_*^3\sigma}^{1/4}
ight) = 4.1 imes 10^4 \dot{M}_{16}^{1/4} m_1^{1/4} R_9^{-3/4} \; {
m K}$$





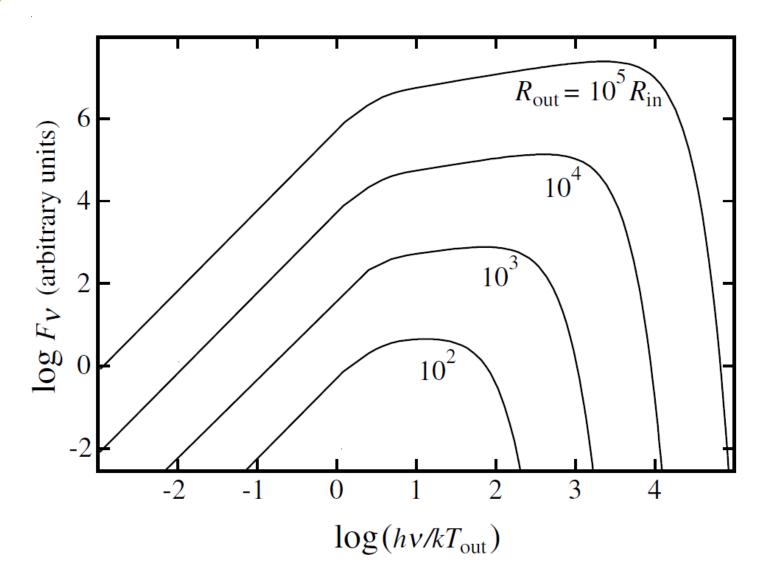
#### For an observer at distance D

$$F_
u = rac{4\pi h \cos i
u^3}{c^2D^2} \int\limits_{R_*}^{R_{
m out}} rac{R \mathrm{d}R}{\exp[h
u/kT(R)]-1}$$

- ullet For low frequencies  $u << kT(R_{
  m out})/h \implies F_
  u \propto 
  u^2$
- ullet For high frequencies  $u>>kT_*/h \implies F_
  u\propto 
  u^3 \exp[-h
  u/kT_*]$
- ullet For intermediate values  $F_
  u\propto
  u^{1/3}\int\limits_0^\inftyrac{x^{5/3}}{e^x-1}\mathrm{d}x,\;\;x=(h
  u/kT_*)(R/R_*)^{3/4}$











## The Standard Model. Structure of steady lpha- disk

To solve that scary system we need to rethink equations 6 and 8

$$ullet \ au = \Sigma \kappa_{
m R}(
ho, T_{
m c}) = au(\Sigma, 
ho, T_{
m c})$$

• 
$$\nu = \nu(\rho, T_{\rm c}, \Sigma, \alpha, \dots)$$

Let's introduce lpha- prescription  $u=lpha c_{
m s} H$ 

and assume Rosseland mean opacity is well approximated by Kramers' law

$$\kappa_{
m R} = 5 imes 10^{24} 
ho T_{
m c}^{-7/2} {
m ~cm^2 ~g^{-1}}$$





#### Now equations are algebraic (almost)

1. 
$$\Sigma = 5.2 \alpha^{-4/5} \dot{M}_{16}^{7/10} m_1^{1/4} R_{10}^{-3/4} f^{14/5} \text{ g cm}^{-2}$$
  
2.  $H = 1.7 \times 10^8 \alpha^{-1/10} \dot{M}_{16}^{3/20} m_1^{-3/8} R_{10}^{9/8} f^{3/5} \text{ cm}$   
3.  $\rho = 3.1 \times 10^{-8} \alpha^{-7/10} \dot{M}_{16}^{11/20} m_1^{5/8} R_{10}^{-15/8} f^{11/5} \text{ g cm}^{-3}$   
4.  $T_{\rm c} = 1.4 \times 10^4 \alpha^{-1/5} \dot{M}_{16}^{3/10} m_1^{1/4} R_{10}^{-3/4} f^{6/5} \text{ K}$   
5.  $\tau = 190 \alpha^{-4/5} \dot{M}_{16}^{1/5} f^{4/5}$ ,  $\left\{ f = \sqrt{1 - (R_*/R)^{1/2}} \right\}$   
6.  $\nu = 1.8 \times 10^{14} \alpha^{4/5} \dot{M}_{16}^{3/10} m_1^{-1/4} R_{10}^{3/4} f^{6/5} \text{ cm}^{2-1}$   
7.  $v_R = 2.7 \times 10^4 \alpha^{4/5} \dot{M}_{16}^{3/10} m_1^{-1/4} R_{10}^{-1/4} f^{-14/6} \text{ cm s}^{-s}$ 





## Looking closely

1. 
$$H/R=1.7 imes 10^{-2}lpha^{-1/30} \dot{M}_{16}^{3/20} m_1^{-3/8} R_{10}^{1/8} f^{3/5}$$

2. 
$$M_{
m disc} = 2\pi \int\limits_{R_*}^{R_{
m out}} \Sigma R {
m d}R \lesssim (10^{-10} M_{\odot}) lpha^{-4/5} \dot{M}_{16}^{7/10}$$

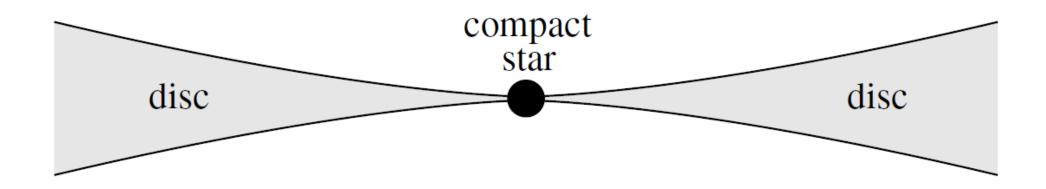
3. 
$$ho << M/R^3$$

4. 
$$\kappa_{
m R}(Krm) = au/\Sigma = 36 \dot{M}_{16}^{-1/2} m_1^{1/4} R_{10}^{3/4} f^{-2} {
m cm}^2 {
m g}$$

which dominates for  $R\gtrsim 2.5 imes 10^7 \dot{M}_{16}^{2/3} m_1^{1/3} f^{8/3} {
m ~cm}$ 







$$rac{P_{
m r}}{P_{
m g}} = 2.8 imes 10^{-3} lpha^{1/10} \dot{M}_{16}^{7/10} R_{10}^{-3/8} f^{7/5}$$

Radiation pressure exceedes gas pressure at

$$R \lesssim 24 lpha^{2/21} \dot{M}_{16}^{16/21} m_1^{-3/21} f^{4/21} \; {
m km}$$





## For radiation-pressure-domianted region

$$au = \Sigma \kappa_{
m R} = 
ho H \sigma_{
m T}/m_{
m p}$$

$$c_{
m s}^2 = rac{3GM\dot{M}\sigma_{
m T}H}{8\pi R^3 m_{
m p}c}iggl[1-\left(rac{R_*}{R}
ight)^{1/2}iggr]$$

and 
$$H \cong rac{2\sigma_{
m T}\dot{M}}{8\pi m_{
m p}c}iggl[1-\left(rac{R_*}{R}
ight)^{1/2}iggr]$$

$$\dot{M}_{
m crit} = rac{L_{
m Edd}R_*}{2\eta M} =$$

$$1.5 imes10^{18}\left(rac{R_*}{3~\mathrm{km}}
ight)\left(rac{\eta}{0.1}
ight)^{-1}~\mathrm{g~s^{-1}}$$

$$\dot{M}_{
m crit} = 2\pi R_* m_{
m p} c/(\eta \sigma_{
m T})$$

$$H\congrac{3R_*}{4\eta}rac{\dot{M}}{\dot{M}_{
m crit}}iggl[1-\left(rac{R_*}{R}
ight)^{1/2}iggr]$$





