

**Project name:-** FCND-Controls-CPP

**Submitted by:-** Bishwajit Kumar

**Objective:** - To build a controller in C++ along with implementing and tuning these controllers in multiple steps for 5 scenarios(for 5 different controllers).

- 1) Body rate control
- 2) Roll pitch control
- 3) Altitude control
- 4) Lateral control
- 5) Yaw control

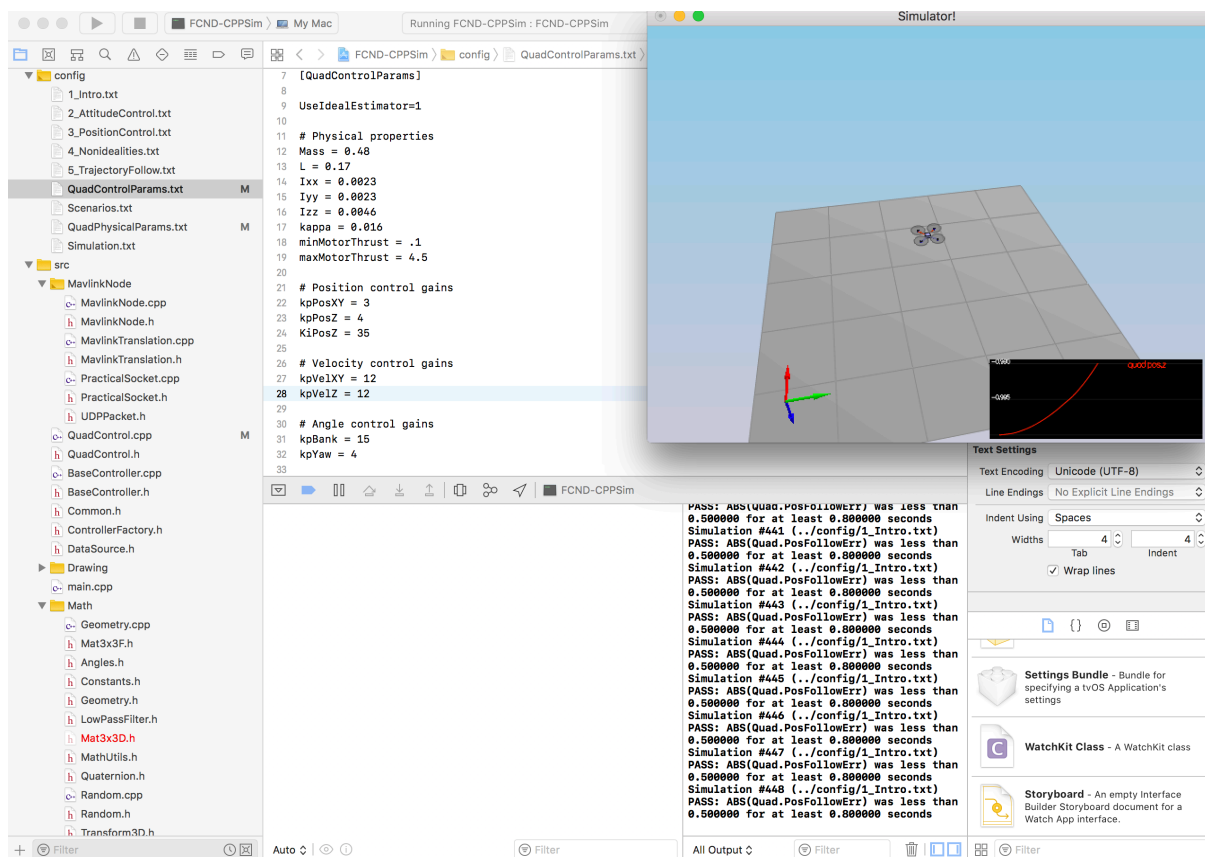
Mainly modification in 2 file was required to complete the project:-

- 1) /config/QuadControlParams.txt: Contains all control gains and other desired tuning parameters.
- 2) /src/QuadControl.cpp: Contains all of the code for different controllers.

## Implementation:-

### Scenario 1: - Inro\_1

Simulator provided for intro scenario required changes in mass value of drone to make it stable. Changed mass to 0.48g to make it stable in QuadControlParams.txt file.



## Scenario 2:- Body rate and roll/pitch control

- Implement calculating the motor commands given commanded thrust and moments in C++  
Implemented **GenerateMotorCommands** method in QuadControl.cpp as below: -

```
float l = L / sqrtf(2.f);  
float f1 = momentCmd.x / l;  
float f2 = momentCmd.y / l;  
float f3 = - momentCmd.z / kappa;  
float f4 = collThrustCmd;  
cmd.desiredThrustsN[0] = (f1 + f2 + f3 + f4)/4.f;  
cmd.desiredThrustsN[1] = (-f1 + f2 - f3 + f4)/4.f;  
cmd.desiredThrustsN[2] = (f1 - f2 - f3 + f4)/4.f;  
cmd.desiredThrustsN[3] = (-f1 - f2 + f3 + f4)/4.f;
```

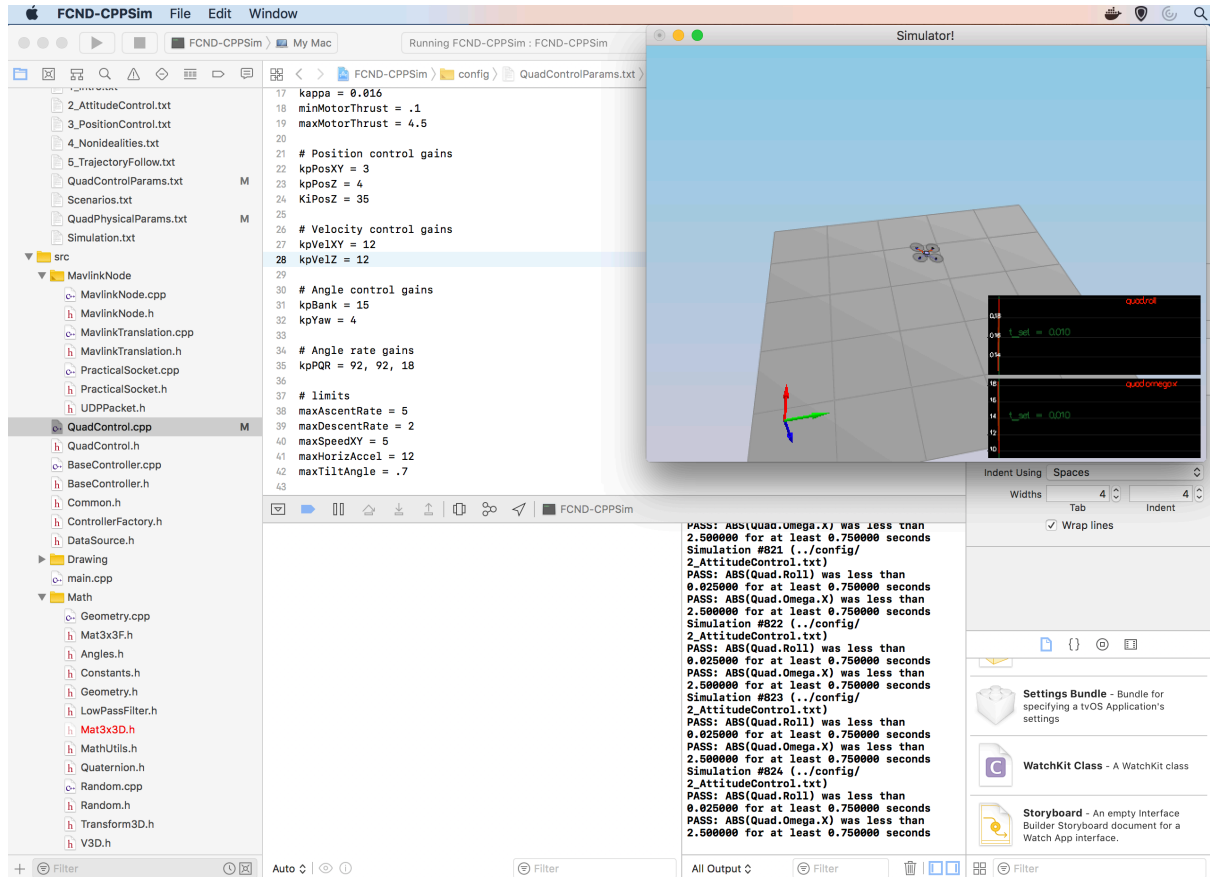
- Implement body rate control in C++ :-  
This is a single order P controller and controls roll,pitch and yaw in 3 axes.  
Modified **BodyRateControl** method to return desired moments for all 3 axes(X,Y,Z).

```
V3F MI;  
MI.x = lxx;  
MI.y = lyy;  
MI.z = lzz;  
momentCmd = MI * kpPQR * ( pqrCmd - pqr );
```

- Implement roll pitch control in C++:-

This is again a single order P controller and works by receiving commanded acceleration in x and y axes. It basically converts rotation matrix from world frame to body frame. It returns a V3F(pqrCmd below) containing desired pitch and roll rates. Modified **RollPitchControl** method to implement the same.

```
float R11 = R(0, 0);  
float R12 = R(0, 1);  
float R21 = R(1, 0);  
float R22 = R(1, 1);  
float R33 = R(2, 2);  
  
float c = collThrustCmd / mass;  
V3F b = V3F(R(0, 2), R(1, 2), 0.f);  
V3F b_c = V3F(accelCmd.x / -c, accelCmd.y / -c, 0.f);  
b_c.constrain(-maxTiltAngle, maxTiltAngle);  
  
V3F b_error = b_c - b;  
V3F b_c_dot = kpBank * b_error;  
  
pqrCmd.x = (R21 * b_c_dot.x - R11 * b_c_dot.y) / R33;  
pqrCmd.y = (R22 * b_c_dot.x - R12 * b_c_dot.y) / R33;  
pqrCmd.z = 0.f;
```



### Scenario 3:- Position/velocity and yaw angle control

- Implement lateral position control in C++:-

This control returns a desired acceleration in world frame. Implemented **LateralPositionControl** method to return a V3F with desired horizontal acceleration.

```
const V3F error = posCmd - pos;
velCmd += kpPosXY * error;
```

```
velCmd.x = CONSTRAIN(velCmd.x, -maxSpeedXY, maxSpeedXY);
velCmd.y = CONSTRAIN(velCmd.y, -maxSpeedXY, maxSpeedXY);
```

```
const V3F error_dot = velCmd - vel;
accelCmd += kpVelXY * error_dot;
```

```
accelCmd.x = CONSTRAIN(accelCmd.x, -maxAccelXY, maxAccelXY);
accelCmd.y = CONSTRAIN(accelCmd.y, -maxAccelXY, maxAccelXY);
```

```
accelCmd.z = 0.0F;
```

- Implement altitude control in C++:-

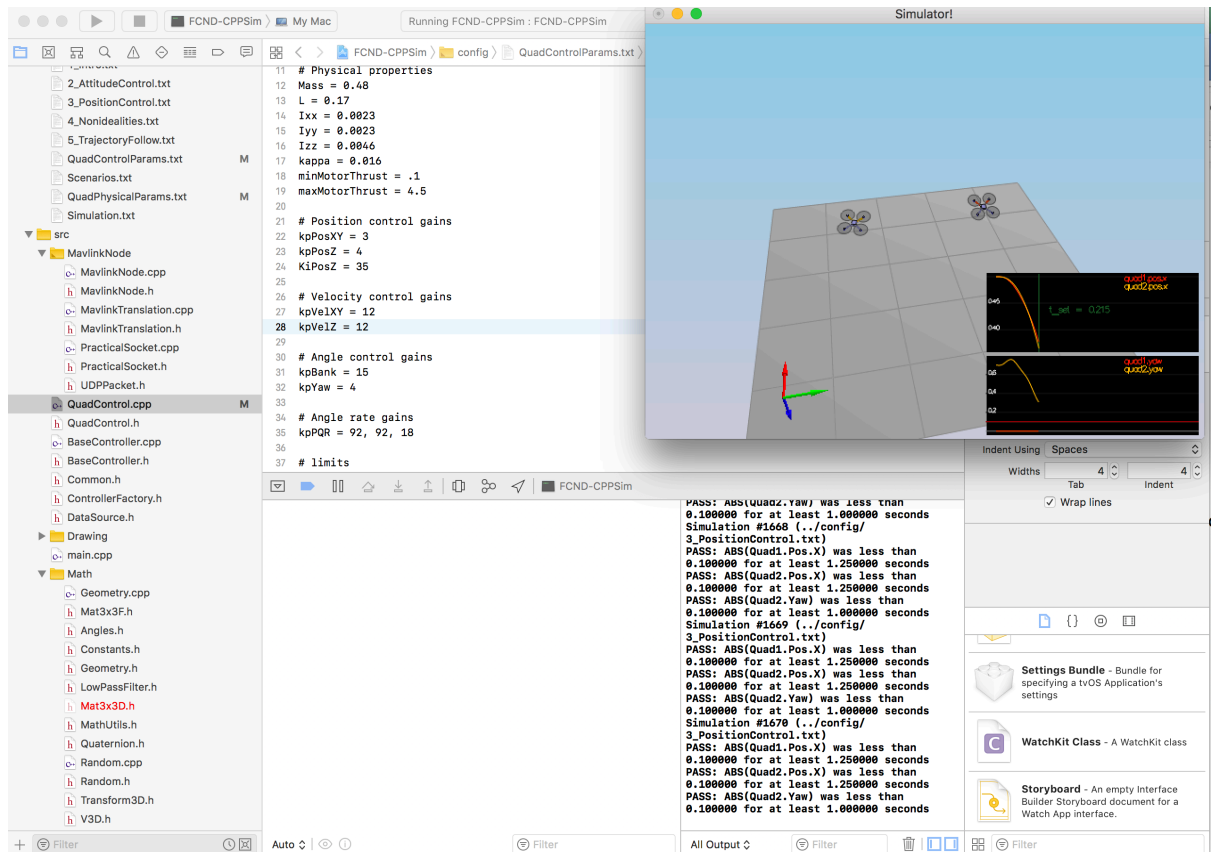
This control returns thrust to control the acceleration in z direction. Implemented **AltitudeControl** method which returns overall thrust (in N).

```
float z_target = posZCmd;  
float z_dot_target = velZCmd;  
float z_actual = posZ;  
float z_dot_actual = velZ;  
integratedAltitudeError = integratedAltitudeError + ((z_target - z_actual) * dt);  
float z_dot_dot_target = accelZCmd;  
  
float u_bar_1 = (kpPosZ * (z_target - z_actual)) + (kpVelZ * (z_dot_target - z_dot_actual)) +  
(KiPosZ * integratedAltitudeError) + z_dot_dot_target;  
float b = R(2,2);  
float c = (u_bar_1 - CONST_GRAVITY) / b;  
float c_const = CONSTRAIN(c, -maxAscentRate / dt, maxAscentRate / dt);  
thrust = - c_const * mass;
```

- Implement Yaw control in C++:-

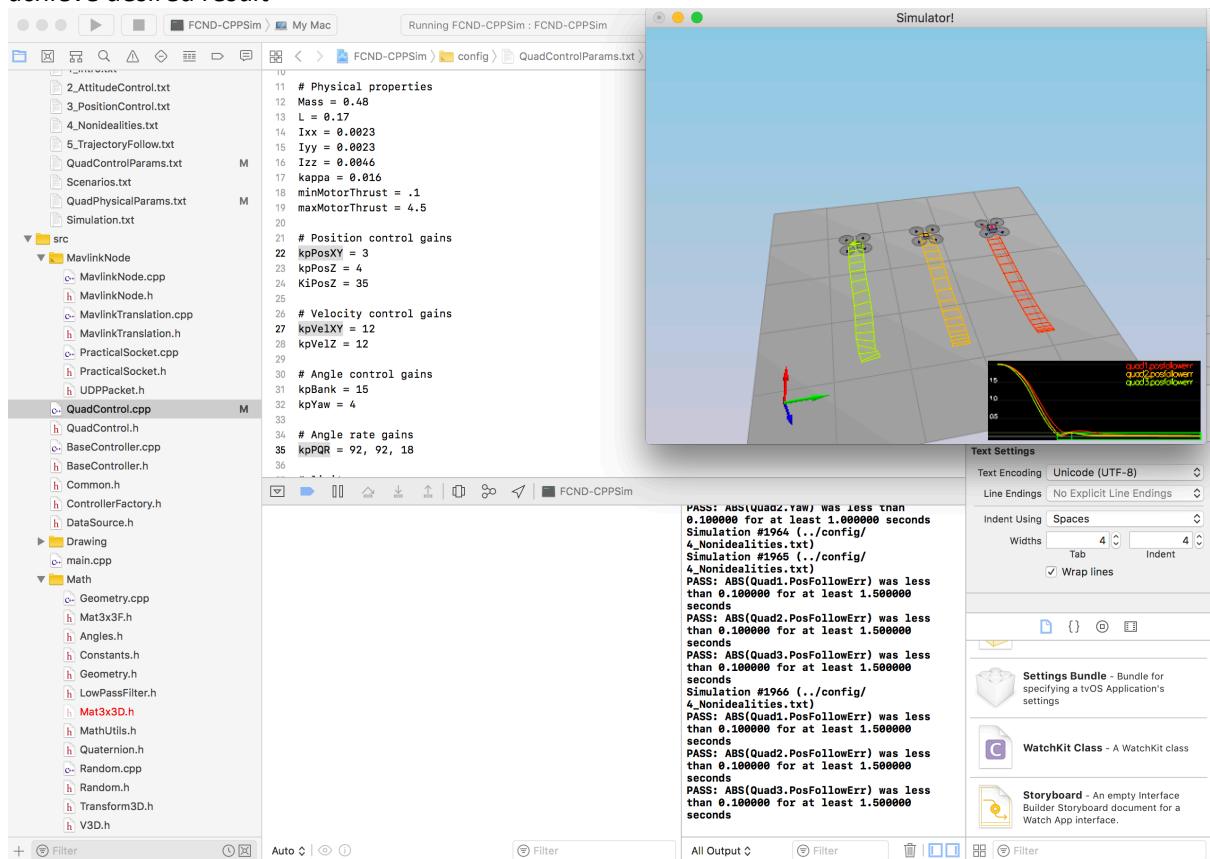
This control returns desired yaw rate. Implemented **AltitudeControl** method which returns yaw rate (in rad/s)

```
float yaw_error = yawCmd - yaw;  
yaw_error = fmodf(yaw_error, 2.*F_PI);  
yawRateCmd = kpYaw * yaw_error;
```



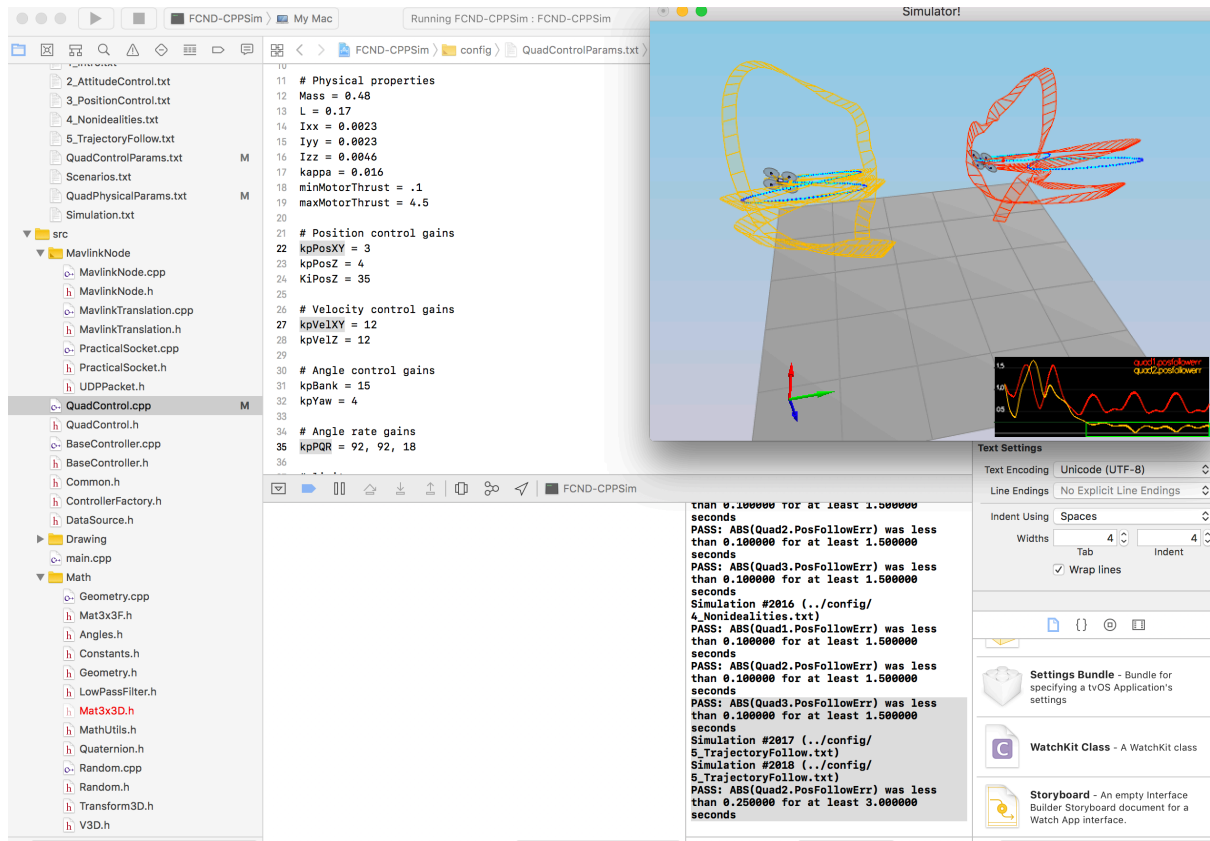
## Scenario 4:- Non idealities and robustness

Edited **AltitudeControl** method and tuned kpPosXY and kpVelXY parameters along with kpPQR to achieve desired result



**Scenario 5:-** Testing it all together to make an 8 shape.

Able to make 8 shape with pass results( Though seems some more fine tuning is required in the provided parameters as it is not working as provided in sample example)



Could not try extra challenges and will do the same later.

Please find Project3.mov video in the attached folder for all scenarios video.