

Mechanics of Materials Formula and Data Sheet

Some Engineering Material Properties:

Property	Steel	Aluminum	Brass
Young's Modulus, E	210 GPa (30 psi x 10 ⁶)	70 GPa (10 psi x 10 ⁶)	105 GPa (15 psi x 10 ⁶)
Shear Modulus, G	81 GPa (11.6 psi x 10 ⁶)	26 GPa (3.7 psi x 10 ⁶)	39 GPa (5.6 psi x 10 ⁶)
Poisson's Ratio, ν	0.30	0.33	0.35
Thermal Expansion, α	11x10 ⁻⁶ /°C (6x10 ⁻⁶ °/F)	22x10 ⁻⁶ /°C (12x10 ⁻⁶ °/F)	20x10 ⁻⁶ /°C (11x10 ⁻⁶ °/F)
Density, ρ	7850 kg/m ³ (0.28 lb/in ³)	2720 kg/m ³ (0.10 lb/in ³)	8410 kg/m ³ (0.30 lb/in ³)
Yield stress, σ_y	250 MPa (36 x 10 ³ psi)	230 MPa (33 x 10 ³ psi)	210 MPa (30 x 10 ³ psi)

Moments of Area for Common Shapes:

Shape	Cross-section A	I _x	I _y	J = I _x + I _y
solid circle	π c ²	π c ⁴ / 4	π c ⁴ / 4	π c ⁴ / 2
thin-wall circle	2 π t c	π t c ³	π t c ³	2 π t c ³
solid rectangle	b h	b h ³ / 12	h b ³ / 12	b h (b ² + h ²) / 12

Elastic relationships:

$$\epsilon_x = \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} - \nu \frac{\sigma_z}{E} + \alpha \Delta T \quad \gamma = \frac{\tau}{G} \quad G = \frac{E}{2(1+\nu)} \quad K = \frac{E}{3(1-2\nu)}$$

Failure Criteria: (for principal stresses $\sigma_1 \geq \sigma_2 \geq \sigma_3$)

Tresca: $\sigma_1 - \sigma_3 = \sigma_y$ **Mises:** $(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 = 2\sigma_y^2$ **Mohr:** $\sigma_1/\sigma_{UT} - \sigma_3/\sigma_{UC} = 1$

Thin-Walled Pressure Vessels:

Cylinder: $\sigma_\theta = \frac{pR}{t}$ $\sigma_a = \frac{pR}{2t}$ $\sigma_r \approx 0$ Sphere: $\sigma_\theta = \sigma_a = \frac{pR}{2t}$ $\sigma_r \approx 0$

Shaft Torsion:

$$\frac{\tau}{r} = \frac{T}{J} = \frac{G\phi}{L}$$

Beam Bending:

$$\frac{\sigma}{-Y} = \frac{M}{I} = \frac{E}{R}$$

Parallel Axis Theorem:

$$I(d) = I_0 + A d^2$$

Power Transmission:

$$P = \frac{2\pi f J \tau_{\max}}{c} \text{ watts (metric units)} \quad P = \frac{15.87 \times 10^{-6} \text{ rpm } J \tau_{\max}}{c} \text{ h.p. (lb.in units)}$$

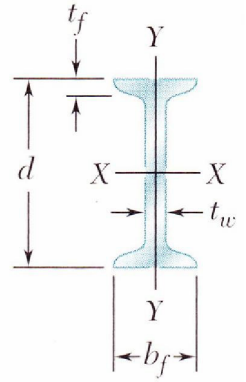
Beam Formulas:

$$V = \frac{dM}{dx} \quad w = \frac{dV}{dx} \quad \frac{d^2y}{dx^2} = \frac{M}{EI} \quad q = \frac{VA\bar{y}}{I} = \frac{VQ}{I} \quad \text{where } Q = \int y dA$$

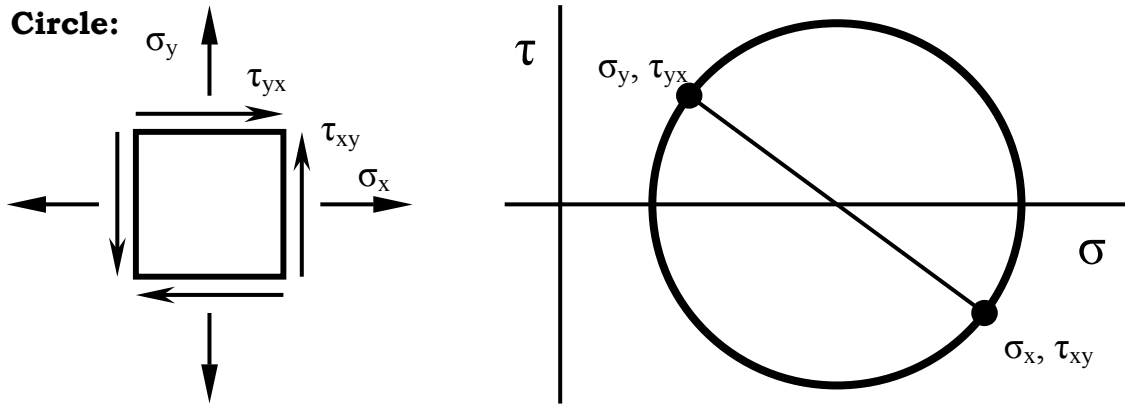
$$< x - a > = (x - a) \text{ if } x - a > 0, \quad = 0 \text{ if } x - a \leq 0$$

Some Typical I-Beams:

Designation†	Area A, mm ²	Depth d, mm	Flange		Web Thick- ness t _w , mm	Axis X-X			Axis Y-Y		
			Width b _f , mm	Thick- ness t _f , mm		I _x 10 ⁶ mm ⁴	S _x 10 ³ mm ³	r _x mm	I _y 10 ⁶ mm ⁴	S _y 10 ³ mm ³	r _y mm
S150 × 25.7	3260	152	90.7	9.12	11.8	10.9	143	57.9	0.953	21.0	17.1
18.6	2360	152	84.6	9.12	5.89	9.16	120	62.2	0.749	17.7	17.8
S130 × 15	1890	127	76.2	8.28	5.44	5.12	80.3	52.1	0.495	13.0	16.2
S100 × 14.1	1800	102	71.1	7.44	8.28	2.81	55.4	39.6	0.369	10.4	14.3
11.5	1460	102	67.6	7.44	4.90	2.52	49.7	41.7	0.311	9.21	14.6
S75 × 11.2	1420	76.2	63.8	6.60	8.86	1.21	31.8	29.2	0.241	7.55	13.0
8.5	1070	76.2	59.2	6.60	4.32	1.04	27.4	31.2	0.186	6.28	13.2



Mohr's Circle:



Use sign convention “***in the kitchen, the clock is above and the counter is below***” for the shear stresses. Then, rotations in the Mohr's circle have the same direction and double the rotation angle of the physical stresses. For Mohr's circle of strain, use ϵ and $\gamma/2$ in place of σ and τ .

Column Buckling:

$$P_{CR} = \frac{\pi^2 EI}{L_e^2} = \frac{\pi^2 EA}{(L_e/r)^2} \quad \text{where: } \begin{array}{ll} L_e = L \text{ (pinned-pinned)} & L_e = 2L \text{ (free-fixed)} \\ L_e = 0.7L \text{ (pinned-fixed)} & L_e = 0.5L \text{ (sliding-fixed)} \end{array}$$

$$\text{Eccentric load: } y_{\max} = e \left(\sec \left(\frac{\pi}{2} \sqrt{\frac{P}{P_{CR}}} \right) - 1 \right) \quad \text{Initial curvature: } y_{\max} = \frac{aP}{P_{CR} - P}$$

$$I = A r^2 \quad \text{For “short” steel columns (Johnson): } \frac{L_e}{r} < \sqrt{\frac{2\pi^2 E}{\sigma_Y}} \quad \sigma_{CR} = \sigma_Y - \left(\frac{\sigma_Y^2}{4\pi^2 E} \right) \left(\frac{L_e}{r} \right)^2$$

Strain Energy:

$$\text{rod: } U = \int \frac{P^2}{2EA} dx \quad \text{beam: } U = \int \frac{M^2}{2EI} dx \quad \text{shaft: } U = \int \frac{T^2}{2GJ} dx$$

Castigliano's Theorem:

$$\delta_j = \frac{\partial U}{\partial P_j} \quad \theta_j = \frac{\partial U}{\partial M_j} \quad \text{rod: } \delta_j = \int \frac{P}{EA} \frac{\partial P}{\partial P_j} dx \quad \text{beam: } \delta_j = \int \frac{M}{EI} \frac{\partial M}{\partial P_j} dx$$