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Introduction

This project involves evaluating a proposed temporary lifting apparatus for transporting goods to upper levels of a multi-story building. The design utilizes a pulley system constructed from repurposed steel automobile wheels and climbing rope, with a manual crank mechanism for lifting payloads. The objective is to develop a mathematical model of the system's dynamic behavior and assess its vibration characteristics using engineering principles and computational tools. Through this project, students will demonstrate their ability to apply scientific and engineering knowledge to solve complex engineering problems, in alignment with ECSA graduate attribute 2.

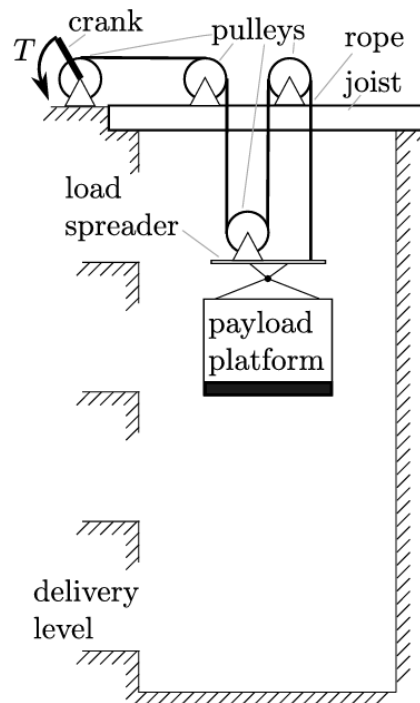
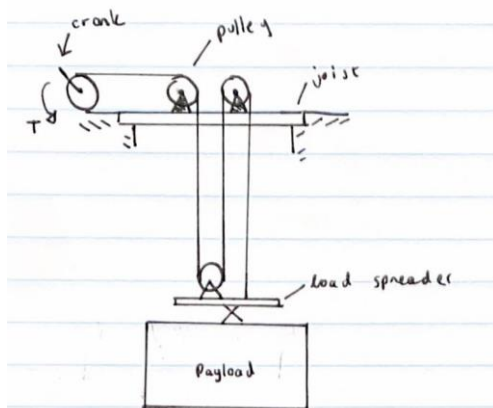


Figure 1: Make-shift lifting system.

Overview and Values

Input Values	Common Values
$m_{\text{wheel}} = 18 \text{ kg}$	$m_{\text{hub}} = 25 \text{ kg}$
$I_{\text{wheel}} = 0.6 \text{ kgm}^2$	$I_{\text{hub}} = 0.1 \text{ kgm}^2$
Rope Modulus = 80 kN	SA pine = 50 mm wide & 12 mm high
Rope mass / unit length 90 kg/m	$E = 10 \text{ GPa}$
Payload mass = 300 kg	Rope safe load = 20 kN

The above are the respective values and information given for the lifting mechanism.



There is a total of 4 pulleys, 1 payload, 1 load spreader and 1 rope. The torque is applied through a mechanism consisting of an arm and handle that will be investigated later. This mechanism is the crank mechanism.

The next step is to establish the number of degrees of freedom.

Assumptions & Simplifications

To determine the degrees of freedom to determine accurate equations of motion each part of the mechanism will be investigated while making logical assumptions.

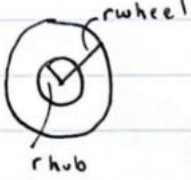
Load-spreader & Payload

It will be assumed that the load spreader and payload will not deform throughout operation and are connected by a light inextensible rope.

Therefore, the load-spreader, payload and rope connecting them will be neglected as degrees of freedom.

Pulleys

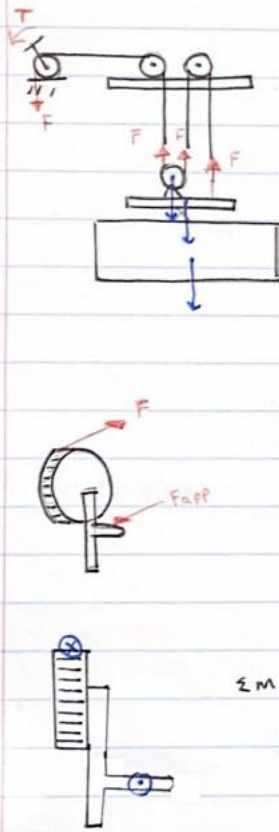
The pulleys are repurposed automotive wheels with hub inside. The hub axle assembly will be modelled as a solid cylinder and the wheel as a hollow cylinder. From the given values and model assumption the radius of the hub and wheel can be calculated.


$$J_{hub} = \frac{1}{2} m r^2$$
$$0.1 = \frac{1}{2} (25) (r_{hub}^2) \quad \therefore r_{hub} = 0.089m$$
$$J_{wheel} = \frac{1}{2} m (r_h^2 + r_w^2)$$
$$0.6 = \frac{1}{2} (18) (0.089^2 + r_w^2)$$
$$\therefore r_w = 0.242m$$

The pulleys are assumed to remain in a fixed position and are frictionless throughout operation.

Rope Configuration

The rope being used to pull up the load has a can withstand a maximum load of 20kN, the safety of the rope will be investigated.



$\Sigma \text{load} = m_{\text{pulley}} + m_{\text{load}} + m_{\text{payload}}$
 $= (25 + 18) + (110) + 300 = 453 \text{ kg}$
 $453 \times 9.81 = 4443.93 \text{ N}$

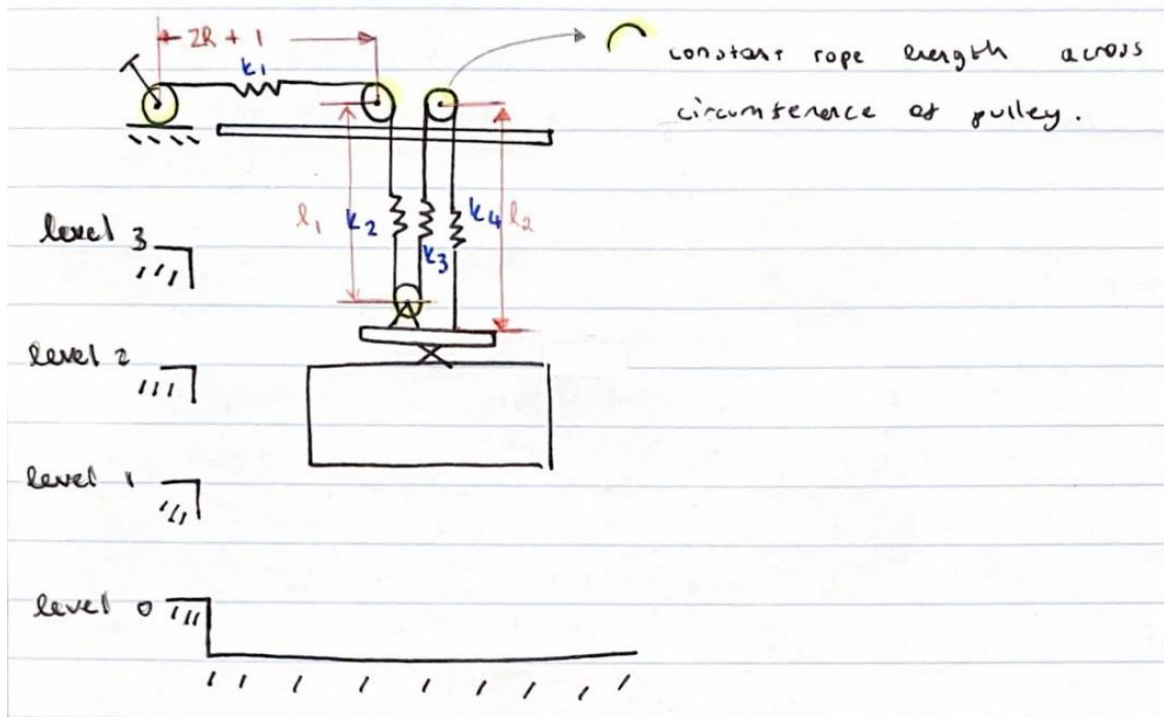
$\Sigma F_y = 0 \therefore 3F = 4443.93$
 $F = 1481.31 \text{ N} = 1.48 \text{ kN}$

$F_{\text{rope max}} = 20 \text{ kN}$
 $F \ll F_{\text{rope max}} \therefore \text{the rope can take the load.}$

$\Sigma M = 0 \therefore (242)(1481.31) = (600)(F_{\text{app}})$
 $\therefore F_{\text{app}} = 597.46 \text{ N}$
 $\frac{597}{9.81} = 60.9 \text{ kg}$

$\therefore \text{A feasible amount of weight to be pulled by a human.}$

Therefore, the rope can safely carry the load and will not break through operation.



The distance between the bottom of the joist to level 0 is 16m, each level is 4m apart.

The rope in contact with the circumference of the pulleys is constant and does not change.

The distance between the 2 pulleys is assumed to be $2R + 1\text{m} = 1.484\text{m}$, this length is constant at all levels therefore K_1 is constant at all 4 levels.

$$l_1 = r_{\text{pulley}} + l_{\text{rope}} + h_{\text{joist}}$$

$$l_2 = 2r_{\text{pulley}} + l_{\text{rope}} + h_{\text{joist}}$$

Level	L1	L2
0	15.394	15.636
1	11.394	11.636
2	7.394	7.636
3	3.394	3.636

$$k_x = AE/l_y$$

$$\text{Rope Modulus} = AE = 80\text{kN}$$

Level	K1	K2	K3	K4
0	53.91	5.20	5.20	5.11
1	53.91	7.02	7.02	6.87
2	53.91	10.82	10.82	10.48
3	53.91	23.57	23.57	22.00

At level 3 the largest difference between k2/3 and k4. There is a 6.66% difference between them when looking at the most extreme case, this is smaller than the difference with k1 by more than a factor of 10.

Rope masses at lowest level

$$m_{horizontal} = (rope\ mass\ length) \left(2 r_{pulley} + 1 \right) = (90)(1.484) = 133.56\ g$$

$$m_{vertical\ 1} = 2(rope\ mass\ length) \left(r_{pulley} + 1 \right) = (90)(15.394) = 2.77\ kg$$

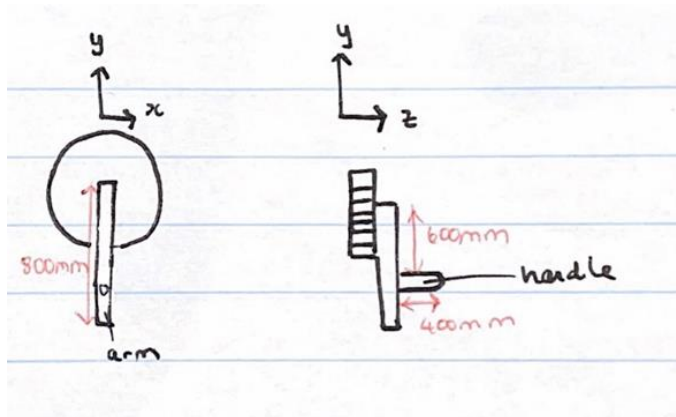
$$m_{vertical\ 2} = (rope\ mass\ length) \left(2r_{pulley} + 1 \right) = (90)(15.636) = 1.41\ kg$$

$$m_{vertical\ 1} + m_{vertical\ 2} = 4.18\ kg$$

The sum of the ropes mass at the maximum case is less than the payload and load spreader by more than a factor of 10 therefore the mas is not significant enough to change the load rating of the system. Therefore, it will be ignored in the analysis.

Crank Mechanism

The crank mechanism consists of an arm and handle. The handle is a solid circular steel shaft while the arm is a hollow steel shaft with significant thickness. The orientation of the handle is shown in the illustration below.



The arm is welded to the pulley, making it rigid.

The orientation of the handle has been established. Now the stiffness and maximum deflection of the arm and handle will be investigated if the arm and handle need to be considered as degrees of freedom.

Handle

$r = 10\text{mm}$
 $E_{\text{steel}} = 210\text{GPa}$
 $\rho_{\text{st}} = 7850\text{kg/m}^3$

$$m = v\rho = \frac{(\pi (10)^2)(400)(7850)}{1000^3} = 0.986\text{kg}$$

$$I_{\text{handle}} = \frac{\pi r^4}{4} = \frac{\pi (10)^4}{4} = 7.85 \times 10^3\text{mm}^4$$

Arm

40mm
 $t = 8\text{mm}$
 $E_{\text{steel}} = 210\text{GPa}$
 $\rho_{\text{steel}} = 7850\text{kg/m}^3$

$$m = v\rho = \frac{((40)^2 - (34)^2)(800)(7850)}{1000^3} = 2.79\text{kg}$$

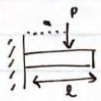
$$I = \frac{1}{12}bh^3$$

$$I_{\text{arm}} = I_{\text{outer}} - I_{\text{inner}}$$

$$= \frac{1}{12}[(40)^4 - (34)^4] = 101472\text{mm}^4$$

Handle

Handle



$$\delta_{max} = \frac{Pa^2}{6EI} (3l - a)$$

$$= \frac{(597)(0.2)^2}{6(210 \times 10^9)(7.85 \times 10^{-9})} (3 \times 0.4 - 0.2)$$

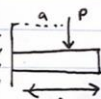
$$= 2.41 \text{ mm}$$

$$\therefore k_H = \frac{6EI}{a^2} \left(\frac{1}{3l - a} \right)$$

$$k_H = 412.13 \text{ kN/m}$$

Arm

Arm



$$\delta_{max} = \frac{Pa^2}{6EI} (3l - a)$$

$$= \frac{(597)(0.6)^2}{6(210 \times 10^9)(1.01 \times 10^{-7})} (3 \times 0.8 - 0.6)$$

$$= 3.03 \text{ mm}$$


$$\therefore k_A = \frac{6EI}{a^2} \left(\frac{1}{3l - a} \right)$$

$$k_A = 589.17 \text{ kN/m}$$

The arm and handle are stiff when compared to the stiffness of the rope by a factor of 10 therefore both the arm and the handle will not be considered as degrees of freedom. Despite the handle having a lower stiffness, if the handle bends the arm will not necessarily. This also solidifies the assumption to not include these as DOF. The minimum distance the ropes will extend to is 4m therefore the calculated deflections are less than a factor of 10 for both the arm and handle this also solidifies the assumption and decision to not include the arms and handles as degrees of freedom.

Joist

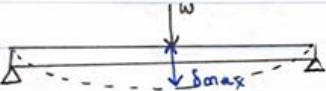
2 pulleys are mounted in a one wood joist, the integrity of the joist will be investigated to establish if it needs to be included as a degree of freedom in the lifting system.



$I = \frac{1}{12}bh^3 = \frac{1}{12}(50)(152)^3 = 14.633 \times 10^6 \text{ mm}^4$

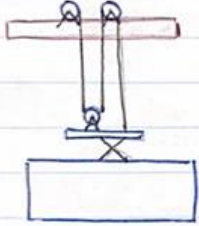
$E_{\text{joist}} = 10 \text{ GPa}$

Taking a conservative approach. Assuming the worst case scenario, the force acts through the centre of beam, I will use the maximum deflection formula for a simply supported beam.



$\delta_{\text{max}} = \frac{PL^3}{48EI}$

I will calculate the stiffness using this formula
I also calculate the δ_{max} separately taking the load into account to assess if the joist is a degree of freedom



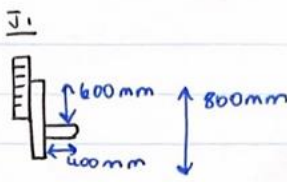
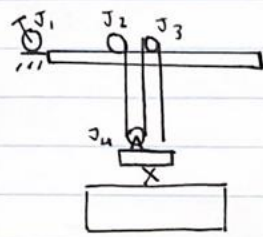
$P = 3m_{\text{HUB+ROPE}} + m_{\text{load spread}} + m_{\text{payload}}$
 $= 3(25 + 18) + (110) + (300) = 539 \text{ Kg}$

$\delta_{\text{max}} = \frac{(539 \times 9.81)(1)^3}{(48 \times 10 \times 10^9)(14.6 \times 10^{-6})}$ $= 7.52 \times 10^{-4} \text{ m}$ $= 0.75 \text{ mm}$	$\therefore k = \frac{48EI}{L^3}$ $= \frac{48(10 \times 10^9)(14.6 \times 10^{-6})}{1^3}$ $= 7008 \text{ kN/m}$
--	---

The joist when compared to the stiffness of the rope (**what**) is greater by a factor of 10 therefore the joist will not be considered as degrees of freedom. The calculated deflection is less by a factor than the minimum length the ropes will extend (4m) solidifying this assumption and decision.

Moments of Inertia

Each pulley will have a moment of inertia. 3 of the 4 pulleys will have the same moment of inertia value however the pulley with crank mechanism will have a different moment of inertia this will be explored later after the crank mechanism has been investigated.


$$J_1 = J_{hub} + J_{wheel} + J_{arm} + J_{handle}$$

J_{arm} , arm will be modelled as a thin rod rotating about its end

$$\therefore J_a = \frac{m l^2}{3} = \frac{(2.79)(0.8^2)}{3} = 0.595 \text{ kgm}^2$$

J_{handle} , the handle will be modelled as a solid cylinder about its central axis & parallel axis theorem

$$\therefore J_h = \frac{m R^2}{2} + m d^2 = \frac{(0.486)(0.01)^2}{2} + 0.486(0.6)^2 = 0.355 \text{ kgm}^2$$
$$J_1 = J_{hub} + J_{wheel} + J_{arm} + J_{handle} = 0.1 + 0.6 + 0.595 + 0.355$$
$$= 1.65 \text{ kgm}^2$$
$$J_2 = J_{hub} + J_{wheel} = 0.1 + 0.6 = 0.7 \text{ kgm}^2$$

Outcome

The investigation has led to the following outcomes and assumptions.

It will be assumed that the load spreader and payload will not deform throughout operation and are connected by a light inextensible rope. Therefore, will not be considered a degree of freedom.

The joist is stiff when compared to the rope by more than a factor of 10 therefore it will not deform and does not need to be considered as a degree of freedom.

The pulleys are assumed to remained in a fixed position and are frictionless throughout operation and the length of the rope in contact with pulleys remain constant.

The crank arm and handle are stiff when compared to rope by more than a factor of 10 therefore will not be considered a degree of freedom and will not deform. The arm is welded the pulley and is considered rigid.

The pulley with the crank mechanism has a moment of inertia different to the other 3 which all share the same value.

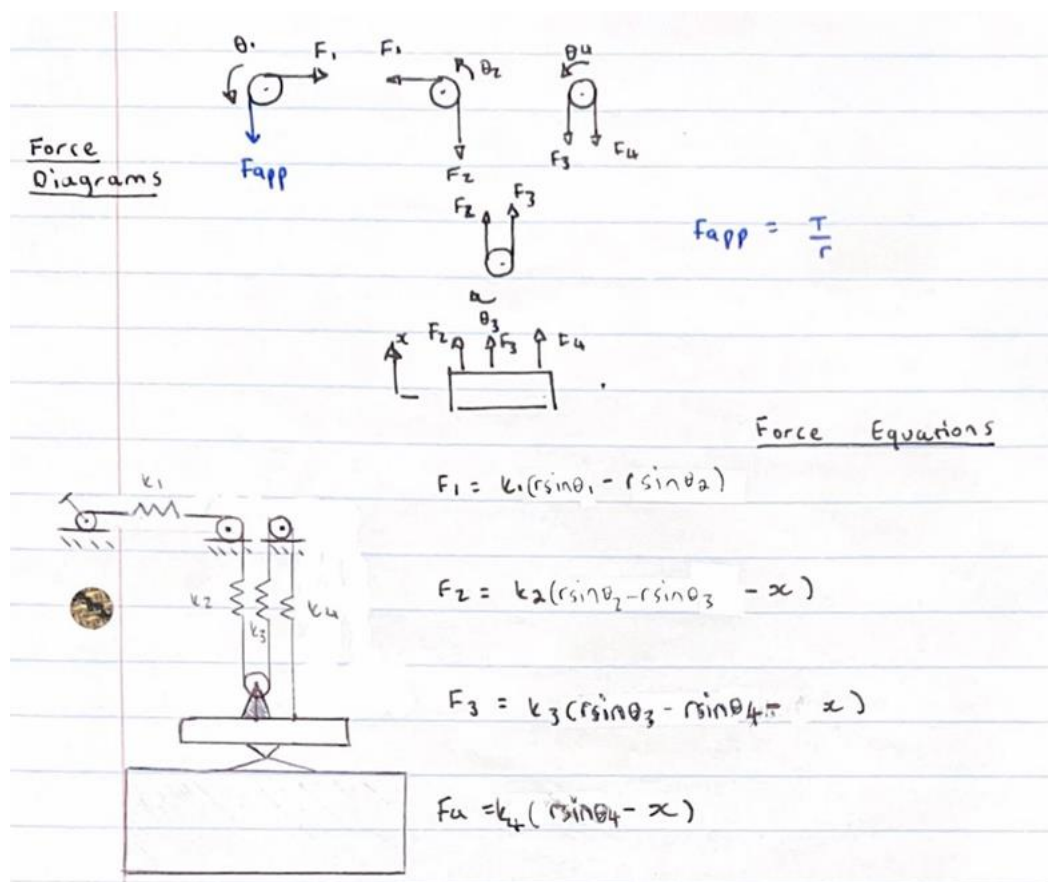
The rope can safely carry the load and the mass of the rope can be neglected as it less than the payload mass by a factor of 10.

There will be 3 K values in the model.

Therefore, the chosen degrees of freedom are 5. One DOF for each pulley and a degree of freedom for the payload and load spreader together.

Equations of Motion

A Newtonian approach was used.



x

$$\sum F_x = m_{beam} \ddot{x} = F_4 + F_3 + F_2$$

$$= k_4(r \sin \theta_4 - x) + k_3(r \sin \theta_3 - r \sin \theta_4 - x) + k_2(r \sin \theta_2 - r \sin \theta_3 - x)$$

θ_4

$$\sum M = -F_4 r + F_3 r = -k_4(r \sin \theta_4 - x) r + k_3(r \sin \theta_2 - r \sin \theta_3 - x) r$$

$$J \ddot{\theta}_4 = -k_4(r \sin \theta_4 - x) r + k_3(r \sin \theta_3 - r \sin \theta_4 - x) r$$

θ_3

$$\sum M = -F_3 r + F_2 r = -k_3(r \sin \theta_3 - r \sin \theta_4 - x) r + k_2(r \sin \theta_2 - r \sin \theta_3 - x) r$$

$$J \ddot{\theta}_3 = k_3(-r \sin \theta_3 + r \sin \theta_4 + x) r + k_2(r \sin \theta_2 - r \sin \theta_3 - x) r$$

θ_2

$$\sum M = -F_2 r + F_1 r = -k_2(r \sin \theta_2 - r \sin \theta_3 - x) r + k_1(r \sin \theta_1 - r \sin \theta_2) r$$

$$J \ddot{\theta}_2 = -k_2(r \sin \theta_2 - r \sin \theta_3 - x) r + k_1(r \sin \theta_1 - r \sin \theta_2) r$$

θ_1

$$\sum M = -F_1 r$$

$$J \ddot{\theta}_1 = -k_1(r \sin \theta_1 - r \sin \theta_2) r$$

EOM

$$x \quad m\ddot{x} - k_4 r \theta_4 + k_4 x - k_3 r \theta_3 + k_3 r \theta_4 + k_3 x + k_2 x + k_2 r \theta_3 - k_2 r \theta_2 = 0$$

$$m\ddot{x} - k_4 r \theta_4 + k_3 r \theta_4 - k_3 r \theta_3 + k_2 r \theta_3 + k_4 x + k_2 x - k_2 r \theta_2 = 0$$

$$\theta_4 \quad J\ddot{\theta}_4 + k_4 r \theta_4 r - k_4 x r - k_3 r \theta_3 r + k_3 r \theta_4 r + k_3 x r = 0$$

$$I\ddot{\theta}_4 + k_4 r^2 \theta_4 - k_4 x r + k_3 x r - k_3 r^2 \theta_3 + k_3 r^2 \theta_4 = 0$$

$$\theta_3 \quad J\ddot{\theta}_3 - k_3 r^2 \theta_3 + k_3 r^2 \theta_4 + k_3 x r + k_2 r^2 \theta_2 - k_2 r^2 \theta_3 - k_2 x r = 0$$

$$J\ddot{\theta}_3 + k_3 r^2 \theta_3 + k_2 r^2 \theta_3 - k_3 r^2 \theta_4 - k_2 r^2 \theta_2 - k_3 x r + k_2 x r = 0$$

$$\theta_2 \quad J\ddot{\theta}_2 + k_2 r^2 \theta_2 - k_2 r^2 \theta_3 - k_2 x r - k_1 r^2 \theta_1 + k_1 r^2 \theta_2 = 0$$

$$J\ddot{\theta}_2 + k_2 r^2 \theta_2 + k_1 r^2 \theta_2 - k_2 r^2 \theta_3 - k_2 x r - k_1 r^2 \theta_1 = 0$$

$$\theta_1 \quad J\ddot{\theta}_1 + k_1 r^2 \theta_1 - k_1 r^2 \theta_2 = 0$$

Matrix Form

$$\begin{bmatrix} J_0 & 0 & 0 & 0 & 0 \\ 0 & J_0 & 0 & 0 & 0 \\ 0 & 0 & J_0 & 0 & 0 \\ 0 & 0 & 0 & J_0 & 0 \\ 0 & 0 & 0 & 0 & m \end{bmatrix} \begin{bmatrix} \ddot{\theta}_1 \\ \ddot{\theta}_2 \\ \ddot{\theta}_3 \\ \ddot{\theta}_4 \\ \ddot{x} \end{bmatrix} + \begin{bmatrix} k_1 r^2 & -k_1 r^2 & 0 & 0 & 0 \\ -k_1 r^2 & r^2(k_1 + k_2) & -k_2 r^2 & 0 & -k_2 r \\ 0 & -k_2 r^2 & r^2(k_3 + k_2) & -k_3 r^2 & r(k_2 - k_3) \\ 0 & 0 & -k_3 r^2 & r^2(k_3 + k_4) & r(k_3 - k_4) \\ 0 & -k_2 r & r(k_2 - k_3) & r(k_3 - k_4) & k_4 + k_3 + k_2 \end{bmatrix} \begin{bmatrix} \theta_1 \\ \theta_2 \\ \theta_3 \\ \theta_4 \\ x \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

$$= \begin{bmatrix} 1 & 1 & 1 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Modal Analysis

Level 0

Laden

Modal Frequencies (Hz):

['0.0000', '8.6868', '22.5269', '36.7840', '99.3305']

Mode Shapes (normalized eigenvectors):

[['-0.4301' '-0.4176' '-0.2606' '0.1036' '0.4100'
 '-0.4301' '-0.4065' '-0.2140' '0.0542' '-1.0147'
 '-0.2867' '-0.3164' '0.7220' '-0.8500' '0.0506'
 '-0.1434' '-0.1728' '0.8368' '0.8233' '-0.0025'
 '-0.0347' '0.0351' '0.0014' '-0.0001' '0.0003']]

Unladen

Modal Frequencies (Hz):

['0.0000', '13.3843', '22.5611', '36.7844', '99.3364']

Mode Shapes (normalized eigenvectors):

[['0.5380' '-0.2592' '-0.2657' '-0.1038' '-0.4099'
 '0.5380' '-0.2429' '-0.2180' '-0.0543' '1.0147'
 '0.3587' '-0.2496' '0.7160' '0.8500' '-0.0506'
 '0.1793' '-0.1562' '0.8332' '-0.8232' '0.0025'
 '0.0434' '0.0846' '0.0070' '0.0005' '-0.0012']]

Level 1

Laden

Modal Frequencies (Hz):

['0.0000', '10.1298', '26.2088', '42.8860', '99.9964']

Mode Shapes (normalized eigenvectors):

[['-0.4301' '-0.4193' '-0.2652' '0.1080' '0.4041'
 '-0.4301' '-0.4042' '-0.2010' '0.0380' '-1.0190'
 '-0.2867' '-0.3144' '0.7259' '-0.8459' '0.0709'
 '-0.1434' '-0.1717' '0.8333' '0.8271' '-0.0049'
 '-0.0347' '0.0351' '0.0014' '-0.0001' '0.0004']]

Unladen

Modal Frequencies (Hz):

['0.0000', '15.6095', '26.2446', '42.8862', '100.0074']

Mode Shapes (normalized eigenvectors):

[['0.5380' '-0.2621' '-0.2702' '-0.1081' '-0.4040'
 '0.5380' '-0.2396' '-0.2046' '-0.0381' '1.0190'
 '0.3587' '-0.2459' '0.7203' '0.8459' '-0.0709'
 '0.1793' '-0.1537' '0.8299' '-0.8271' '0.0049'
 '0.0434' '0.0846' '0.0067' '0.0004' '-0.0017']]

Level 2

Laden

Modal Frequencies (Hz):

['0.0000', '12.6602', '32.5844', '53.5515', '101.5237']

Mode Shapes (normalized eigenvectors):

[['-0.4301' '-0.4230' '-0.2751' '0.1184' '0.3905'
 '-0.4301' '-0.3991' '-0.1722' '-0.0012' '-1.0270'
 '-0.2867' '-0.3102' '0.7341' '-0.8350' '0.1185'
 '-0.1434' '-0.1693' '0.8252' '0.8356' '-0.0135'
 '-0.0347' '0.0351' '0.0012' '0.0000' '0.0007']]

Unladen

Modal Frequencies (Hz):

['0.0000', '19.5120', '32.6184', '53.5515', '101.5504']

Mode Shapes (normalized eigenvectors):

[['0.5380' '-0.2684' '-0.2798' '-0.1184' '-0.3902'
 '0.5380' '-0.2324' '-0.1750' '0.0012' '1.0269'
 '0.3587' '-0.2378' '0.7295' '0.8350' '-0.1184'
 '0.1793' '-0.1484' '0.8222' '-0.8356' '0.0135'
 '0.0434' '0.0847' '0.0058' '-0.0000' '-0.0026']]

Level 3

Laden

Modal Frequencies (Hz):

['0.0000', '19.1207', '48.1024', '79.9978', '108.6443']

Mode Shapes (normalized eigenvectors):

[['-0.4301' '-0.4365' '-0.3104' '0.1697' '0.3246']
['-0.4301' '-0.3803' '-0.0575' '-0.2128' '-1.0249']
['-0.2867' '-0.2945' '0.7592' '-0.7520' '0.3431']
['-0.1434' '-0.1604' '0.7885' '0.8659' '-0.1043']
['-0.0347' '0.0351' '0.0004' '0.0005' '0.0014']]

Unladen

Modal Frequencies (Hz):

['0.0000', '29.4649', '48.1090', '80.0120', '108.7669']

Mode Shapes (normalized eigenvectors):

[['0.5380' '-0.2929' '-0.3126' '-0.1683' '-0.3236']
['0.5380' '-0.2034' '-0.0578' '0.2112' '1.0246']
['0.3587' '-0.2054' '0.7578' '0.7528' '-0.3415']
['0.1793' '-0.1275' '0.7873' '-0.8658' '0.1034']
['0.0434' '0.0847' '0.0021' '-0.0022' '-0.0054']]

Discussion

The modal frequencies and mode shapes explain the behavior of the lifting operation at different levels and under different loading conditions.

Level 3

The modal frequencies at the top level are similar between the laden and unladen systems suggesting that the presence or absence of a payload has minimal effect on behavior of the system at this level.

The mode shapes at the top level also show similarities between the laden and unladen systems. The primary mode shapes involve large vertical displacements (X1 and X2) and little lateral displacements (X3 and X4), with minimal involvement of the payload (X5).

Level 2

The modal frequencies show little variation between the laden and unladen systems. This indicates that the dynamic behavior remains consistent regardless of the presence of a payload.

Similar behavior to the top level, the mode shapes show trends between the laden and unladen systems, with vertical displacements (X1 and X2) and small lateral displacements (X3 and X4).

Level 1

The modal frequencies at the 1st level show minimal differences between the laden and unladen systems, suggesting consistent dynamic behavior.

The mode shapes at the 2nd level also show similarities between the laden and unladen systems, with significant vertical displacements and minor lateral displacements.

Level 0

The modal frequencies at the bottom level show marginal changes between the laden and unladen systems.

The mode shapes at the bottom level also demonstrate similarities between the laden and unladen systems, with dominant vertical displacements and minor lateral displacements.

Results Discussion

Mode shapes remain relatively consistent across different levels and loading conditions, suggesting that the dynamic behavior of the lifting operation is primarily influenced by structural and geometric characteristics rather than the presence of a payload.

Part 2

Numerical Approximation

Dropped Payload Investigation

Discussion of model variants

Unsuitable scenarios

To conduct a MDOF analysis a linear system is assumed, meaning components in the system affect each other proportionally. However, this is not the case but modelling systems in this way is possible if on a low scale and the characteristics of the system match a linear system.

When a system's characteristics are far from linear or affect each other in a nonlinear fashion the assumption can no longer stand as the results of the model will be very inaccurate. An alternative approach to this system is to conduct a finite element analysis.

Another important consideration is the topic of deflections. In our model the deflection of the joist could be ignored because it was far less than the length of the rope therefore will not influence the results by a large degree however if it was similar this deflection cannot be ignored and will result in the system not following a linear pattern.

Rope

When the platform is at the lowest level deflections are most severe this is the most relevant case. If a stiffer rope results in a better insight into the deflection between each level, this will make the system and model more precise. A stiffer rope also improves the safety of the system ensuring the rope does not deflect laterally.

A stiffer rope will transmit higher loads to the pulley where they are attached. This means stronger pulleys will need to be sourced and could make for a more complex system. The maximum deflection is decreased but now more rope length is required which can increase the cost of the system.

Bicycle Wheels

If bicycle wheels are incorporated into the system the system will need to be modified due to different mass, stiffness, and geometric properties.

Bicycles have a lower mass and are narrower than car wheels, they have spokes and have less volume. The empty spaces vs the thick rims results in the bicycle having a smaller moment of inertia, negatively affecting the performance of the pulley system.

The bike wheels also differ in stiffness and damping properties and are designed to operate at lower forces and torques. The system will have to be modelled differently to account for this.

Bicycles also have a bigger radius on average than car wheels, this means a different length rope is required to reach the same levels. Also, bicycles are lighter and therefore they will need to be mounted in a different way to ensure they do not shift during operation.