## NATIONAL UNIVERSITY OF SINGAPORE

### **EXAMINATION FOR**

(Semester I: 2020/2021)

## EE4302 - ADVANCED CONTROL SYSTEMS

November/December 2020 - Time Allowed: 2 Hours

### **INSTRUCTIONS TO CANDIDATES:**

- 1. Please write your student number only. Do not write your name.
- 2. This question paper contains **FOUR** (4) questions and comprises **Nine** (9) pages.
- 3. Answer **ALL** questions.
- 4. Note that the Questions do not carry equal marks.
- 5. This is an **OPEN BOOK** examination.
- 6. Relevant data are provided at the end of this examination paper.
- 7. Graphics/Programmable calculators are not allowed.

Q1 Consider the system (in the open-loop) given by

$$\begin{aligned}
 \dot{x}_1(t) &= x_2(t) \\
 \dot{x}_2(t) &= -w_0^2 x_1(t) + u(t) \\
 y(t) &= x_1(t)
 \end{aligned}$$

where  $w_0$  is a positive-valued constant. Here y(t) is the measured output of the system to be controlled, and r(t) is a set-point command signal which will be applied to the closed-loop system.

It is desired to use the state-feedback method (with scaling gain)

$$u(t) = -k_1 x_1(t) - k_2 x_2(t) + k_s r(t)$$

to attain to a stable (and sufficiently fast) closed-loop with both closed-loop poles at  $-3w_0$ , and also with 0 dB steady-state gain.

Using Ackermann's formula (the formula may be found in the Data Sheet at the end of this Examination script), calculate the required values of  $k_1$  and  $k_2$  to achieve this. Show clearly all the steps in your calculation.

Next, using the Bass-Gura's formula (the formula may also be found in the Data Sheet at the end of this Examination script), likewise calculate the required values of  $k_1$  and  $k_2$  to achieve this. Again, show clearly all the steps in your calculation.

Finally, for the case with  $w_0 = 1$ , calculate the required value of the scaling gain  $k_s$ to attain the specified 0 dB steady-state gain in the closed-loop. Show clearly all the steps in your calculation.

[22 marks]

Q2 For the same open-loop system given in Question 1, consider instead another approach where an augmented state-variable signal  $x_I(t)$  is generated as

$$\dot{x}_I(t) = y(t) - r(t)$$

where y(t) is the measured output of the system to be controlled, and r(t) is the set-point command signal.

Using all necessary detailed equations, block diagrams, analysis and descriptions, show that this alternate approach can also attain to a stable (and sufficiently fast) closed-loop (say, with all closed-loop poles at  $-1.5w_0$ ), and also with 0 dB steadystate gain. Show clearly all the steps in your analysis and calculations.

[13 marks]

### Q3 Consider the process

$$G_p(s) = \frac{Y(s)}{U(s)} = \frac{1}{s}e^{-s}$$

and the relay with describing function

$$N(a) = \frac{4M}{\pi a^2} (\sqrt{a^2 - h^2} - jh)$$

where the hysteresis h = 0.1 and amplitude M = 1. A limit-cycle is obtained when they are connected in a negative feedback loop as shown in Figure Q3.

a) Give the complex number equation for the limit-cycle.

[5 marks]

b) Obtain 2 simultaneous equations by considering the magnitude and phase of the complex number equation in Part (a). Solve the 2 simultaneous equations graphically by plotting w against a for  $a=0.7,\ 0.8,\ 0.9,\ 1,\ 1.1,$  and estimating the solutions from the intersection of the 2 curves. Note that the magnitude and phase of  $\frac{1}{jw}e^{-jw}$  are  $\left(\frac{1}{w}\right)$  and  $\left(-\frac{\pi}{2}-w\right)$  respectively.

[10 marks]

c) Sketch one cycle of the limit-cycle for y(t). On the same plot, superimpose e(t) and u(t).

[15 marks]

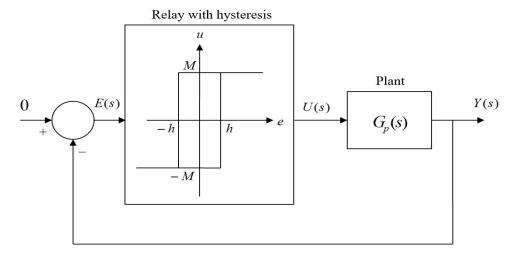


Figure Q3

Q4 Consider the process

$$\dot{x}_1 = x_2^2 - u^2 
\dot{x}_2 = -x_1^2 + 1 
y = x_2$$

where the input u=1 and the states are at the equilibrium point of  $\bar{x}_1=\bar{x}_2=1$ .

a) Find the linearized transfer function about the given equilibrium point.

[20 marks]

b) For u = 1, find all equilibrium points.

[5 marks]

c) Using the linearized transfer function found in Part (a), find y when there is a step change in u from 1 to 1.01 at t=0.

[10 marks]

# END OF QUESTIONS

### DATA SHEET:

1. For the matrices  $\mathbf{A} \in \mathbf{R}^{n \times n}$ ,  $\mathbf{C} \in \mathbf{R}^{1 \times n}$ , and  $\mathbf{L} \in \mathbf{R}^{n}$ , the eigenvalues of the matrix (A - LC) can be arbitrarily assigned by a suitable choice of L as long as

$$O(\mathbf{A}, \mathbf{C}) = \left[ egin{array}{c} \mathbf{C} \\ \mathbf{C}\mathbf{A} \\ dots \\ \mathbf{C}\mathbf{A}^{(n-1)} \end{array} 
ight]$$

is non-singular.

2. For the linear system

$$\dot{\mathbf{x}} = \mathbf{F}\mathbf{x} + \mathbf{G}u$$
$$y = \mathbf{H}\mathbf{x}$$

where  $\mathbf{x} \in \mathbf{R}^n$  and  $u, y \in \mathbf{R}^1$ , the controllability matrix of the system is given by

$$C(\mathbf{F},\mathbf{G}) = \left[ \begin{array}{cccc} \mathbf{G} & \mathbf{F}\mathbf{G} & \dots & \mathbf{F}^{(n-1)}\mathbf{G} \end{array} \right]$$

If the characteristic polynomial of  $\mathbf{F}$  is given by

$$\alpha(s) = s^{n} + a_{1}s^{n-1} + a_{2}s^{n-2} + \dots + a_{n}$$

then the state-feedback  $u = -\mathbf{K}\mathbf{x}$  which yields the closed-loop characteristic polynomial

$$\alpha_c(s) = s^n + \alpha_1 s^{n-1} + \alpha_2 s^{n-2} + \dots + \alpha_n$$

can be calculated using the Bass-Gura's formula

$$\mathbf{K} = \begin{bmatrix} \alpha_1 - a_1 & \alpha_2 - a_2 & \dots & \alpha_n - a_n \end{bmatrix} \{C(\mathbf{F}, \mathbf{G})W\}^{-1}$$

where

$$W = \begin{bmatrix} 1 & a_1 & a_2 & \dots & a_{n-1} \\ 0 & 1 & a_1 & \dots & a_{n-2} \\ \vdots & \vdots & \ddots & \dots & \vdots \\ 0 & 0 & 0 & \dots & 1 \end{bmatrix}$$

Equivalently and alternatively, it can also be calculated using the Ackermann's formula

$$\mathbf{K} = \begin{bmatrix} 0 & 0 & \dots & 0 & 1 \end{bmatrix} \{ C(\mathbf{F}, \mathbf{G}) \}^{-1} \alpha_c(\mathbf{F})$$

where

$$\alpha_c(\mathbf{F}) = \mathbf{F}^n + \alpha_1 \mathbf{F}^{n-1} + \alpha_2 \mathbf{F}^{n-2} + \dots + \alpha_n \mathbf{I}$$

3. For the system

$$\begin{array}{rcl}
 \dot{x}_1 & = & x_2 \\
 \dot{x}_2 & = & x_3 \\
 & \vdots \\
 \dot{x}_n & = & -a_n x_1 - a_{n-1} x_2 - \dots - a_1 x_n + b_0 u \\
 y & = & x_1
 \end{array}$$

the characteristic polynomial is

$$\alpha_o(s) = s^n + a_1 s^{n-1} + \dots + a_n$$

and the transfer function is

$$\frac{Y(s)}{U(s)} = \frac{b_0}{s^n + a_1 s^{n-1} + \dots + a_n}$$

4. For the triple

$$A_{m} = \begin{bmatrix} 0 & 1 & 0 \\ a_{1} & a_{2} & a_{3} \\ 1 & 0 & 0 \end{bmatrix}$$

$$b_{m} = \begin{bmatrix} 0 \\ 0 \\ -1 \end{bmatrix}$$

$$c_{m} = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix}$$

the equivalent transfer function is

$$c_m^{\top}[sI - A_m]^{-1}b_m = \frac{-a_3}{s^3 - a_2s^2 - a_1s - a_3}$$

5. In the development of a reduced-order observer, note that a state-variable system with state vector  $\mathbf{x}$  of order n, where the first  $n_1$  state-variables, in a vector  $\mathbf{x}_1$  are essentially measurable, can be written as:

$$\dot{\mathbf{x}}_1 = \mathbf{F}_{11}\mathbf{x}_1 + \mathbf{F}_{12}\mathbf{x}_2 + \mathbf{G}_1 u$$
  
$$\dot{\mathbf{x}}_2 = \mathbf{F}_{21}\mathbf{x}_1 + \mathbf{F}_{22}\mathbf{x}_2 + \mathbf{G}_2 u$$

with the remaining  $n_2$  state-variables, in a vector  $\mathbf{x}_2$  to be estimated (or observed). Here,  $\mathbf{F}_{11}$ ,  $\mathbf{F}_{12}$ ,  $\mathbf{F}_{21}$  and  $\mathbf{F}_{22}$  are all known system matrices (obtained by calibration data, for example), and the measurement is typically given by

$$\mathbf{y}_m = \mathbf{H}_1 \mathbf{x}_1$$

where  $\mathbf{H}_1$  is also a known  $(n_1 \times n_1)$  system matrix. Under these circumstances, a suitable form for the estimator for  $\mathbf{x}_2(t)$  is

$$\begin{aligned} \hat{\mathbf{x}}_2 &= & \mathbf{L}\mathbf{y}_m + \mathbf{z} \\ \dot{\mathbf{z}} &= & \bar{\mathbf{F}}\mathbf{z} + \bar{\mathbf{G}}\mathbf{y}_m + \bar{\mathbf{H}}u \end{aligned}$$

where a suitable set of matrices defining the reduced-order observer are given by:

$$\begin{split} \bar{\mathbf{F}} &= \mathbf{F}_{22} - \mathbf{L} \mathbf{H}_1 \mathbf{F}_{12} \\ \bar{\mathbf{G}} &= \left\{ \mathbf{F}_{21} - \mathbf{L} \mathbf{H}_1 \mathbf{F}_{11} + \bar{\mathbf{F}} \mathbf{L} \mathbf{H}_1 \right\} \mathbf{H}_1^{-1} \\ \bar{\mathbf{H}} &= \mathbf{G}_2 - \mathbf{L} \mathbf{H}_1 \mathbf{G}_1 \end{split}$$

### 6. Prototype Response Tables

```
Pole Locations for \omega_0 = 1 \ rad/s^a
         k
ITAE
         1 s + 1
         2 \quad s + 0.7071 \pm 0.7071 j^b
        3 (s + 0.7081)(s + 0.5210 \pm 1.068j)
        4 (s + 0.4240 \pm 1.2630j)(s + 0.6260 \pm 0.4141j)
        5 (s + 0.8955)(s + 0.3764 \pm 1.2920j)(s + 0.5758 \pm 0.5339j)
Bessel 1 s+1
        2 \quad s + 0.8660 \pm 0.5000 j^b
         3 (s + 0.9420)(s + 0.7455 \pm 0.7112j)
        4 (s + 0.6573 \pm 0.8302j)(s + 0.9047 \pm 0.2711j)
           (s + 0.9264)(s + 0.5906 \pm 0.9072j)(s + 0.8516 \pm 0.4427j)
```

<sup>&</sup>lt;sup>a</sup> Pole locations for other values of  $\omega_0$  can be obtained by substituting  $s/\omega_0$  for s.

<sup>&</sup>lt;sup>b</sup> The factors (s+a+bj)(s+a-bj) are written as  $(s+a\pm bj)$  to conserve space.

# Laplace Transform Table

Laplace Transform,	Time Function,
F(s)	f(t)
1	$\delta(t)$ (unit impulse)
$\frac{1}{s}$	u(t) (unit step)
$\frac{1}{s^2}$	t
$\frac{\frac{1}{s^2}}{\frac{1}{s^n}}$	$\frac{t^{n-1}}{(n-1)!}$ (n = positive integer)
$\frac{1}{s+a}$	$e^{-at}$
$\frac{\frac{a}{s(s+a)}}{\frac{a}{s(s+a)}}$	$1 - e^{-at}$
$\frac{1}{(s+a)^2}$ $a^2$	$te^{-at}$
$\frac{a^2}{s(s+a)^2}$	$1 - e^{-at}(1 + at)$
$\frac{1}{(s+a)^n}$	$\frac{1}{(n-1)!}t^{n-1}e^{-at} \ (n = \text{positive integer})$
$\frac{1}{(s+a)(s+b)}$	$\frac{e^{-at} - e^{-bt}}{b - a}$
$\frac{1}{s(s+a)(s+b)}$	$\frac{e^{-at}-e^{-bt}}{b-a}$ $\frac{1}{ab}\left[1+\frac{1}{a-b}(be^{-at}-ae^{-bt})\right]$
$\frac{\omega}{s^2 + \omega^2}$	$\sin \omega t$
$\frac{\frac{\omega}{s^2 + \omega^2}}{\frac{\omega^2}{s(s^2 + \omega^2)}}$ $\frac{\frac{s}{s^2 + \omega^2}}{\frac{\omega}{(s+a)^2 + \omega^2}}$	$1-\cos\omega t$
$\frac{s}{s^2+\omega^2}$	$\cos \omega t$
$\frac{\omega}{(s+a)^2+\omega^2}$	$e^{-at}\sin\omega t$
$\frac{s+a}{s+a}$ $\frac{s+a}{(s+a)^2+\omega^2}$ $\omega_n^2$	$e^{-at}\cos\omega t$
$\frac{\omega_n^2}{s^2 + 2\zeta \omega_n s + \omega_n^2}$	$\frac{\frac{\omega_n}{\sqrt{1-\zeta^2}}e^{-\zeta\omega_n t}\sin\omega_n\sqrt{1-\zeta^2}t}{-\frac{1}{\sqrt{1-\zeta^2}}e^{-\zeta\omega_n t}\sin(\omega_n\sqrt{1-\zeta^2}t-\phi)}$
$\frac{s}{s^2 + 2\zeta\omega_n s + \omega_n^2}$	$-\frac{1}{\sqrt{1-\zeta^2}}e^{-\zeta\omega_n t}\sin(\omega_n\sqrt{1-\zeta^2}t-\phi)$
	$\phi = \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta}$
$\frac{\omega_n^2}{s(s^2+2\zeta\omega_n s+\omega_n^2)}$	$1 - \frac{1}{\sqrt{1-\zeta^2}}e^{-\zeta\omega_n t}\sin(\omega_n\sqrt{1-\zeta^2}t + \phi)$
	$\phi = \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta}$

## END OF PAPER