

# **Deep Learning Aided Sensor Fusion for Drift Reduced IMU Orientation Estimation**

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## Abstract

Inertial Measurement Units (IMUs) are widely used in a variety of applications such as Body Sensor Networks (BSNs) for orientation estimation, however the gyroscope suffers from drift due to sensor bias and noise that when integrated accumulate over time. This project investigates a deep learning-based approach which aims to mitigate gyroscopic errors which can be integrated with sensor fusion techniques to achieve more accurate orientation estimates. The proposed deep-learning architecture leverages both neural networks and a temporal history to learn complex and nonlinear error patterns in IMU data, exploring if it outperforms a standard Kalman Filter without learned corrections. The network outputs a correction for the incoming gyroscope sample and ad the measurement noise covariance dependent on the incoming acceleration and magnetometer updates. The data used in training, testing and validating the model come from simulations through MATLAB's Navigation Toolbox and from public datasets such as Berlin Robust Orientation Estimation Assessment Dataset (BROAD).

## Acknowledgements

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I would also like to acknowledge the authors and maintainers of the publicly available datasets of BROAD and RepoIMU. Their efforts in collecting and labelling high quality IMU data and optical motion capture ground truth enabled the training, testing, and validation of the neural network. I am grateful for their contributions and cite them accordingly in this report.

# Contents

<b>Abstract</b>	<b>1</b>
<b>Acknowledgements</b>	<b>1</b>
<b>List of Figures</b>	<b>2</b>
<b>List of Tables</b>	<b>2</b>
<b>1 Introduction</b>	<b>3</b>
1.1 Inertial Measurement Units (IMUs) and their applications . . . . .	3
1.2 Problem Statement: IMU Drift and Its Impact . . . . .	3
1.3 Research Question and Hypotheses . . . . .	3
<b>2 Deep Learning Drift Mitigation: Literature Review</b>	<b>4</b>
<b>3 IMU Basics and Operational Principles</b>	<b>5</b>
3.1 Gyroscope: Operations and Errors . . . . .	5
3.1.1 Angular Rate Integration . . . . .	5
3.1.2 Gyroscope Error Sources . . . . .	6
3.2 Accelerometer: Operations and Errors . . . . .	7
3.2.1 Accelerometer Error Sources . . . . .	7
3.3 Magnetometer: Operations and Errors . . . . .	7
<b>4 Deep Learning Architecture</b>	<b>8</b>
<b>5 Data: Training, Testing, and Validation</b>	<b>9</b>
<b>6 Conclusion and Next Steps</b>	<b>10</b>

## List of Figures

1 Bias Effect on Gyro <b>FIXME: cite</b> . . . . .	6
2 Effect of ARW on Gyro Output <b>FIXME: cite</b> . . . . .	6

## List of Tables

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# 1 Introduction

## 1.1 Inertial Measurement Units (IMUs) and their applications

IMUs are composed of multiple sensors which include a gyroscope, accelerometer, and occasionally a magnetometer. These three sensors measure the angular rate, linear acceleration, and the local magnetic field vector respectively. These measurements can be used to estimate the orientation of an object through integration of the angular rate. This data acquisition is essential for several applications such as BSNs **FIXME: cite**, robotics, and autonomous vehicles where these systems rely on high-rate orientation updates.

IMUs have emerged as a key technology due to the ability to work in a self-contained environment. In environments where external references of orientation are unavailable or unreliable, technologies such as IMUs are attractive. Filters, such as the Kalman filter, allow the use of accelerometers and magnetometers to guide and correct state estimation, but these also suffer from failures like magnetic disturbances and high linear accelerations corrupting a gravitational reference. Due to these limitations, there is significant motivation to explore data-driven methods that compensate for these conditions and errors.

**WORDS:** 170

## 1.2 Problem Statement: IMU Drift and Its Impact

IMUs have shown promise in determining the orientation of an object in motion. However, IMUs suffer from a limitation called drift. IMU drift is characterised by the accumulation of errors through the integration of the angular rate. The sources of these errors include constant bias, scale factor errors and others expanded in section 3.1.2. Errors are not exclusive to the gyroscope but also affect the accelerometer and magnetometer. Drift is also dependent on the type of IMU that is used. Lower cost/grade IMUs suffer from drift at a higher magnitude which results orientation inaccuracies much quicker compared to higher cost/grade IMUs. Errors then lead to inaccuracies in the orientation estimation of an object determined by the IMU.

Kianifar et al. explored using IMUs for automated orientation estimation in a clinical setting. They found that for rotation angles parallel to gravity, drift due to gyroscope bias cannot be compensated by the accelerometer. **FIXME: cite**. Even with multiple sensors, it is still challenging to find an accurate orientation estimation. Thus it is important to try and address the orientation problem by addressing gyroscopic drift.

Therefore, this project aims to address gyroscopic drift by using deep-learning methods to learn complex and non-linear nature of biases and errors.

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## 1.3 Research Question and Hypotheses

The question this project aims to answer is: **Can Deep Learning be used to learn the drift pattern to get drift free orientation estimation?** While this is the overarching question, it can be broken down to the following:

- How effectively can deep learning learn the drift experienced?
- Can this model be generalised or will the model only apply to a single dataset?
- What are the advantages or disadvantages of using Deep Learning compared to traditional approaches?

The hypothesis that I state is that by incorporating a deep-learning model, we will be able to see a percentage decrease in the orientation error over a period of time.

## **2 Deep Learning Drift Mitigation: Literature Review**

### 3 IMU Basics and Operational Principles

As mentioned before, IMUs consists of tri-axial gyroscope, accelerometer, and magnetometer. They measure angular rate, specific force, and the local magnetic field vector. The sensors are mounted so that it measures their components in the three orthogonal axes  $x^b, y^b, z^b$ .

#### 3.1 Gyroscope: Operations and Errors

The gyroscope is the main component that is used to determine the orientation of the object. It measures the angular rate in its orthogonal axes  $\omega^x, \omega^y, \omega^z$  which is used to determined the object's orientation at a discrete time. The orientation is determined through the integration of angular rate. There are multiple different ways to represent the orientation of the object, these being, Euler angles, rotation matrices, and quaternions. Euler angles give rise to the singularity problem due to the loss of one degree-of-freedom whenever two axes of the rotations are parallel [FIXME: cite](#). Rotation matrices are also used, however they are less concise as they are represented as 3x3 matrix, where 6 of these elements are redundant[FIXME: cite](#). Therefore in this chapter, the quaternion representation is selected due to being more computationally efficient than the others [FIXME: cite prof paper](#).

##### 3.1.1 Angular Rate Integration

Starting in continuous-time kinematics, we can define a quaternion that is represented by the angular rate, where  $\omega$  is the real angular rate,  $\omega_q(t)$  is the pure quaternion.

$$\omega_q(t) = [0, \omega_x(t), \omega_y(t), \omega_z(t)] \quad (1)$$

The orientation evolves according to[FIXME: cite](#)

$$\dot{q}(t) = \frac{1}{2} q(t) \otimes \omega_q(t) \quad (2)$$

where  $\otimes$  denotes quaternion multiplication. The solution over  $[t_0, t]$  can be written using the quaternion exponential, this assumes that the  $\omega_q$  is constant over  $\tau$ .

$$q(t) = q(t_0) \otimes \exp\left(\frac{1}{2} \int_{t_0}^t \omega_q(\tau) d\tau\right) \quad (3)$$

These equations show that the orientation update is determined by integrating the angular rate over time and mapping the resulting rotation into a quaternion via the exponential.

Moving to discrete-time kinematics, we can restate of defintions in terms  $\Delta t$ . Here the orientation evolves according to

$$\hat{q}_k = \hat{q}_{k-1} \otimes \Delta q_k \quad (4)$$

where  $\Delta q_k$  is

$$\Delta q_k = \exp\left(\frac{1}{2} \omega_{q,k} \Delta t\right) \quad (5)$$

This was all done by using an ideal  $\omega_k$ , if were to model the angular rate from a gyroscope as [FIXME: cite](#)

$$\omega_{m,k} = \omega_k + b_k + n_k \quad (6)$$

Where  $\omega_{m,k}$  is the measured angular rate,  $\omega_k$  is the true angular rate,  $b_k$  is the bias term, and  $n_k$  is the noise term. The real quaternion orientation update is as follows.

$$\hat{q}_k = \hat{q}_{k-1} \otimes \exp\left(\frac{1}{2} (\omega_k + b_k + n_k)_q \Delta t\right) \quad (7)$$

This shows that the inclusion of the bias and noise accumulates over samples as we move forward to the next k,  $q_{k-1}$  will incorporate the previous error terms. This results in an accumulated orientation error.

### 3.1.2 Gyroscope Error Sources

#### Constant Bias

The bias of a gyroscope is the average output from the gyroscope when it is not undergoing any rotation **FIXME: cite**. This is measured in  $^{\circ}/h$  and can be estimated by taking an average of the output. As this bias is constant, the drift it causes grows linearly with time. However, as discussed further in this section other error sources can make this difficult to determine. **FIXME: Figure** shows how constant bias changes the output.

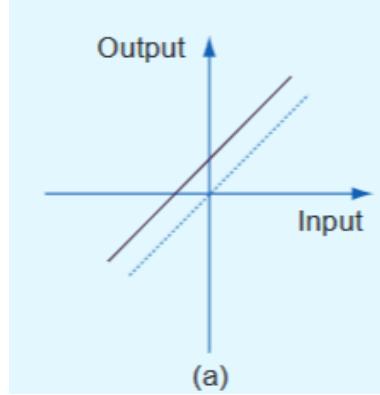


Figure 1: Bias Effect on Gyro **FIXME: cite**

#### White Noise / Angle Random Walk (ARW)

The gyroscope is also affected by some white noise **FIXME: cite**. This white noise sequence is zero-mean uncorrelated random variables between samples and across axes. When the gyroscope signal is integrated to obtain an angle, this white noise produces an ARW. The units of ARW is denoted by  $^{\circ}/\sqrt{h}$ , this shows that the deviation of angle error grows proportionally to  $\sqrt{t}$ . As ARW is due to random variables, it is classified as the  $1\sigma$  of the orientation error. Analog devices shows this in the **FIXME: figure**, where the ARW is  $0.17^{\circ}/\sqrt{h}$ .

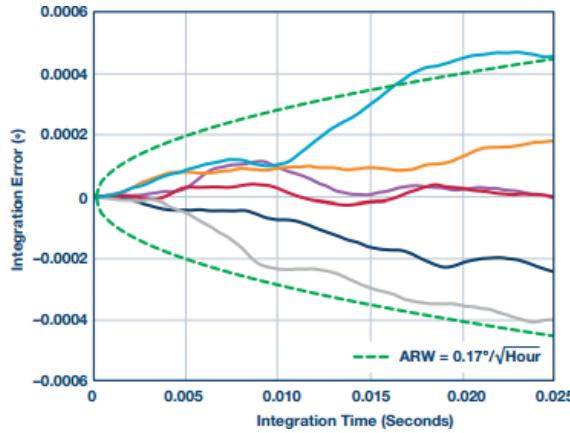


Figure 2: Effect of ARW on Gyro Output **FIXME: cite**

#### Flicker Noise / Bias Stability

The gyroscope suffers from flicker noise in electronics **FIXME: cite**. Flicker noise is  $1/f$ , due to this the effects are observed at lower frequencies while, higher frequencies the noise is dominated by white noise.

## Temperature

Temperature fluctuations due to changes in the environment and sensor heating induce a movement in the bias, this relationship is also highly non-linear **FIXME: cite**. Therefore, it can be very difficult to model and subsequently subtract from the gyroscope measurements compared to a constant bias.

## 3.2 Accelerometer: Operations and Errors

The accelerometer is another component used in determining an orientation estimate of an object. It measures the specific force along its orthogonal axes  $a^x, a^y, a^z$ . These measurements can be rotated in the navigation frame, gravity vector is subtracted, and through integration of the accelerometer, a velocity and subsequently the position can be derived for an object. However, the focus of this project is to achieve drift-free orientation, and hence this will not be covered.

The main purpose of the accelerometer is to offer a gravity reference to the system such that, an accelerometer is capable of providing drift-free inclination estimates **FIXME: cite**. However, the accelerometer, as stated above, does not only measure a gravity reference but also any linear and centripetal accelerations that the object also experiences. Under conditions of high accelerations, the reference is unreliable and is only reliable for static or slowly moving objects **FIXME: cite**. Additionally, the accelerometer cannot detect rotations about the gravity vector, so it cannot provide a yaw/heading of the object. However, like the gyroscope, it also suffers from a variety of errors which causes drift in the output.

### 3.2.1 Accelerometer Error Sources

## 3.3 Magnetometer: Operations and Errors

## 4 Deep Learning Architecture

## **5 Data: Training, Testing, and Validation**

## **6 Conclusion and Next Steps**