

① - ③

xx prerequisite
[moment of inertia (I) - (G)
STRENGTH of MATERIALS]

BENDING & TORSION (STRESS, DEFORMATION)

1. BENDING

Consider a beam in fig. below.

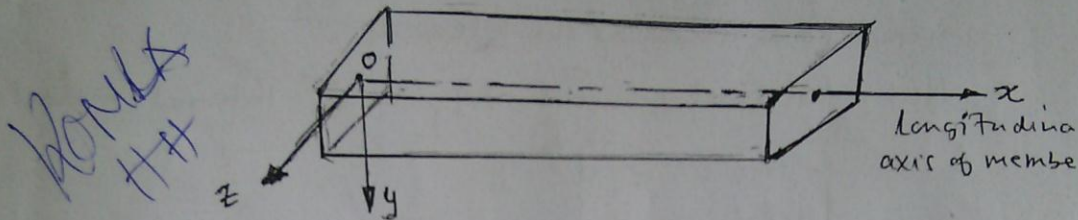
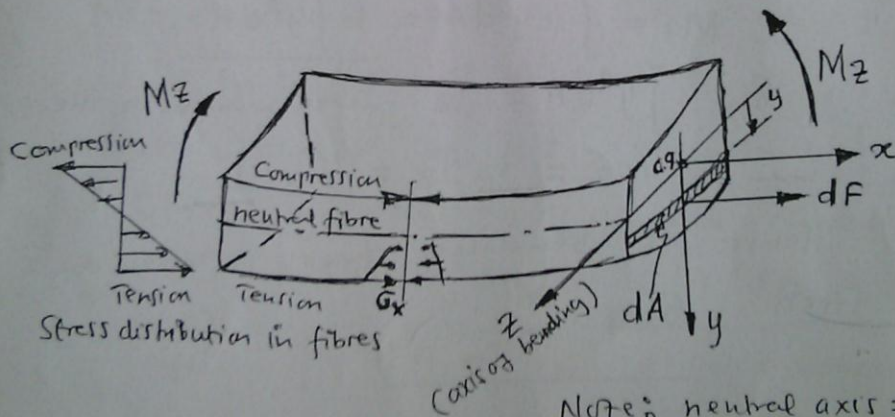


Fig: Beam (Coordinate system)

Now let the above beam be subjected to a bending moment M_z (ive) about z-axis. (fig. below)



Note: neutral axis = C.G. axis in straight beams.

Fig: Bent beam

Therefore the stress is not uniform over the fibres of the cross-section. (fig. above). That is the stress varies from tension to compression.

Assuming a linear variation, let the stress at the fibre a distance 'y' from the axis of bending (z-axis) be ' σ '

$$\therefore \sigma = ky \quad \text{or} \quad k = \frac{\sigma}{y}, \quad k = \text{constant}$$

A differential force at the fibre is $dF = \sigma dA$ where dA = area of the element.

Now External moment $M_z = \sum$ internal moments

$$\therefore M_z = \sum dF \cdot y$$

Note: dF , y are both +ve and -ve below and above the neutral fibre.

$\therefore dF \cdot y$ all +ve

$$\therefore M_z = \int dF \cdot y; \quad dF = \sigma dA = ky dA$$

$$\therefore M_z = \int ky^2 dA = \underline{k \int y^2 dA}$$

But $\int y^2 dA = I_z$ area moment of inertia.

$$\therefore M_z = k \cdot I_z = \frac{\sigma_x}{y} I_z$$

Hence the bending stress is given by.

$$\text{Stress} \quad \sigma_x = \frac{M_z \cdot y}{I_z} \quad (1)$$

Note: σ is max. at outer fibres (-ve or +ve) and $\sigma = 0$ at neutral fibre i.e. $y = 0$.

TORSION

Consider a solid circular bar in torsion (fig. below).

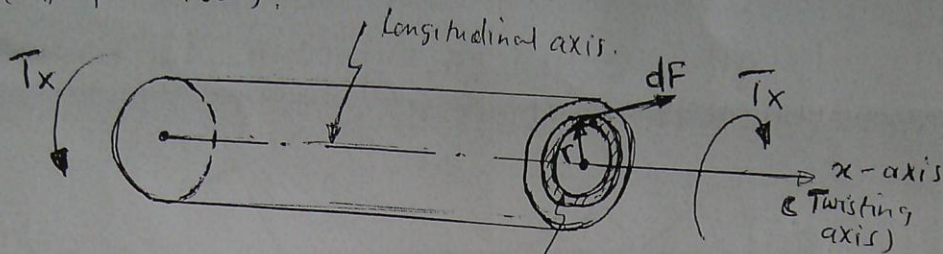


Fig: bar in torsion

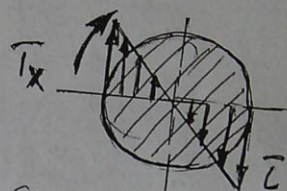


Fig: Stress distribution

Now External torque $T_x = \sum$ internal torque.

$$\therefore T_x = \int r dF, \quad dF = \tau \cdot dA$$

dA = elemental area of ring at a radius r

$$\therefore T_x = \int r \tau dA$$

But $\tau = kr$ or $k = \frac{\tau}{r} = \text{constant from}$

Stress distribution (assuming linear)

$$\therefore T_x = \int kr^2 dA = k \int r^2 dA$$

But $\int r^2 dA = J_x$ i.e. polar moment of inertia.

$$\therefore T_x = k \cdot J_x = \frac{\tau}{r} \cdot J_x$$

\therefore Torsional shear stress τ_{xy} is given

by
$$\tau_{xy} = T_x \cdot r / J_x \quad (1)$$

(2)

Strain

Consider again a bent beam (fig. below).
Let ρ - radius of curvature for the deflected beam at a distance x .

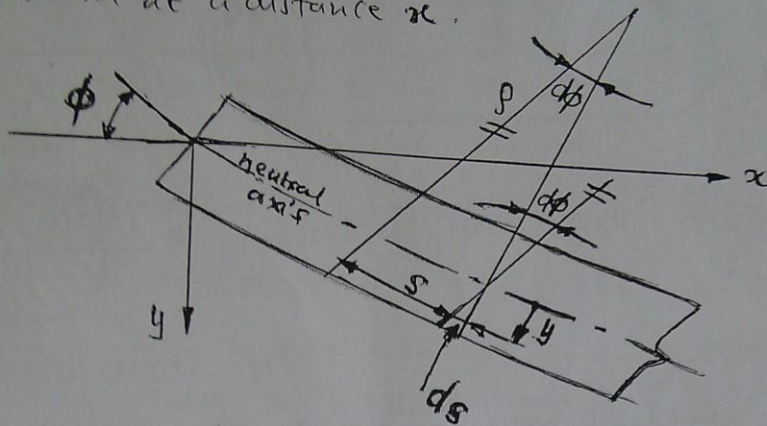


fig: beam curvature

Then the fibre at a distance ' y ' from the neutral axis a length portion ' s ' will stretch by an amount ' ds ' if deflected through a small angle ' $d\phi$ ' (fig. above),

geometry: $\frac{ds}{y} = \frac{s}{\rho}$

Strain: $\epsilon_x = \frac{ds}{s} = \frac{y}{\rho}$ and $\sigma_x = E \cdot \epsilon_x$

$\therefore \sigma_x = \frac{E \cdot y}{\rho}$ (2)

Eqs (1) and (2) therefore give

$$\boxed{\frac{\sigma_x}{y} = \frac{M_z}{I_z} = \frac{E}{\rho}} \quad (3) \quad \text{The bending equation.}$$

(3)

$$\text{or } \frac{\tau}{r} = \frac{T}{J}$$

Note: τ is maximum at outer fibres
 $\tau = 0$ at $r = 0$ (axis of twisting).

Deflection

Consider again a bar in torsion. (Circular solid)

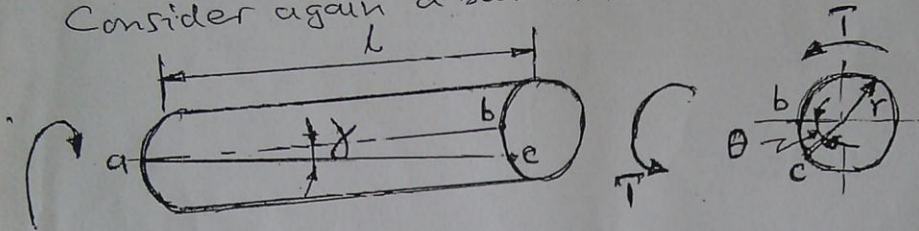


fig: bar in torsion

Let the length of the bar be l

∴ fibre 'ab' on the surface will deflect through angle ' γ ' to 'ac' if ' T ' is applied.

∴ $bc = \gamma l$, and $\gamma = \frac{\tau}{G}$ shear strain

The deflection of the bar at one end with the other end ' θ ' will be given by

$bc = r\theta$, $r = \text{radius of the bar. (fig. above)}$

$$\therefore \gamma l = r\theta \quad \text{or} \quad \frac{\tau}{G} = \frac{r\theta}{l}$$

$$\text{Hence } \frac{G\theta}{l} = \frac{\tau}{r} \quad (2)$$

Combining equations (1) and (2) we obtain

$$\boxed{\frac{\tau}{r} = \frac{T}{J} = \frac{G\theta}{l}} \quad (3)$$

The torsional equation.

Jose Welding
Dchona Rück

STRUCTURES TERMINOLOGY

TRUSSES

1. TIE — Tension member
2. STRUT — Compression member