

Feedback Linearization for a Nonlinear Quarter-Car Model

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Final Project

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Overview

1. Model
2. Controller Design
3. Simulation

Model

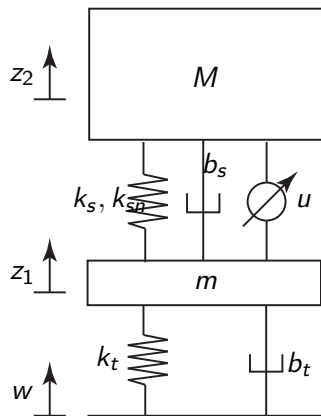


Figure: Quarter-Car Model

$$m\ddot{z}_1 = F_s + F_d - F_t - F_b - u \quad (1)$$

$$M\ddot{z}_2 = -F_s - F_d + u \quad (2)$$

The forces produced by the springs and dampers are the following:

$$F_s = k_s(z_2 - z_1) + k_{sn}(z_2 - z_1)^3 \quad (3)$$

$$F_d = b_s(\dot{z}_2 - \dot{z}_1) \quad (4)$$

$$F_t = k_t(z_1 - w) \quad (5)$$

$$F_b = b_t(\dot{z}_1 - \dot{w}) \quad (6)$$

State Space

Define state variables

$$x_1 = z_2, \quad x_2 = \dot{z}_2, \quad x_3 = z_1, \quad x_4 = \dot{z}_1$$

$$\dot{x}_1 = x_2 \quad (7) \quad \text{where}$$

$$\dot{x}_2 = \frac{1}{M}(-F_s - F_d + u) \quad (8) \quad F_s = k_s(x_1 - x_3) + k_{sn}(x_1 - x_3)^3 \quad (12)$$

$$\dot{x}_3 = x_4 \quad (9) \quad F_d = b_s(x_2 - x_4) \quad (13)$$

$$\dot{x}_4 = \frac{1}{m}(F_s + F_d - F_t - F_b - u) \quad (10) \quad F_t = k_t(x_3 - w) \quad (14)$$

$$y = x_1 \quad (11) \quad F_b = b_t(x_4 - \dot{w}) \quad (15)$$

System

The system is represented in the following form

$$\begin{aligned}\dot{x} &= f(x) + g(x)u \\ y &= h(x)\end{aligned}\tag{16}$$

where $x = [x_1 \ x_2 \ x_3 \ x_4]$ is the state vector, u is the control input, and $y = x_1$ is the output. $f(x)$, $g(x)$, and $h(x)$ are given as follows

$$f(x) = \begin{bmatrix} x_2 \\ \frac{1}{M}(-F_s - F_d) \\ x_4 \\ \frac{1}{m}(F_s + F_d - F_t - F_b) \end{bmatrix}; \quad g(x) = \begin{bmatrix} 0 \\ \frac{1}{M} \\ 0 \\ \frac{-1}{m} \end{bmatrix}; \quad h(x) = [x_1]\tag{17}$$

Lie Derivatives and Coordinate Change

$$y = x_1 = y_1 \quad (18)$$

$$\dot{y} = \dot{y}_1 = \dot{x}_1 = x_2 = y_2 \quad (19)$$

$$\ddot{y} = \dot{y}_2 = \dot{x}_2 = \frac{1}{M}(-F_s - F_d) + \frac{1}{M}u \quad (20)$$

$$L_f h = \frac{\partial h}{\partial x} \times f(x) = x_2 \quad (21)$$

$$L_g h = \frac{\partial h}{\partial x} \times g(x) = 0 \quad (22)$$

$$L_f^2 h = \frac{1}{M}(-F_s - F_d) \quad (23)$$

$$L_g L_f h = \frac{1}{M} \quad (24)$$

the system relative degree is 2. Also, $n - r = 4 - 2 = 2$

Coordinate Change and Zero Dynamics

$$\begin{bmatrix} y_1 \\ y_2 \\ z_1 \\ z_2 \end{bmatrix} = \begin{bmatrix} h(x) \\ L_f(x) \\ \phi_1(x) \\ \phi_2(x) \end{bmatrix} = \begin{bmatrix} x_1 \\ x_2 \\ -x_3 + w \\ Mx_2 + mx_4 - mw \end{bmatrix} \quad (25)$$

ϕ_1 and ϕ_2 are selected such that $L_g\phi = 0$, and T is a diffeomorphism
Check for zero dynamics

$$\dot{z} = \begin{bmatrix} \dot{\phi}_1 \\ \dot{\phi}_2 \end{bmatrix} = \begin{bmatrix} -\frac{z_2 - My_2}{m} \\ k_t z_1 - b_t \frac{z_2 - My_2}{m} \end{bmatrix} \quad (26)$$

$$\dot{z} = f_z(z, 0, 0) = \begin{bmatrix} -\frac{1}{m} z_2 \\ k_t z_1 - \frac{b_t}{m} z_2 \end{bmatrix} \quad (27)$$

is Horowitz or can be proven a.s using a Lyapunov function

Controller and Closed Loop Dynamics

$$u = \frac{1}{L_g L_f h(x)} [v - L_f^2 h(x)] = M[v + k_s(x_1 - x_3) + k_{sn}(x_1 - x_3)^3 + b_s(x_2 - x_4)] \quad (28)$$

close loop dynamics:

$$A = \begin{bmatrix} 0 & 1 \\ -\gamma_1 & -\gamma_2 \end{bmatrix}$$

Choose $-\gamma_1$ and $-\gamma_2$ such that $Real(\lambda_1, \lambda_2) < 0$ for the controller to be stable

Controller

y is required to track a smooth signal y_d

$$u = \frac{1}{L_g L_f h(x)} [v - L_f^2 h(x)] = M[v + k_s(x_1 - x_3) + k_{sn}(x_1 - x_3)^3 + b_s(x_2 - x_4)] \quad (29)$$

where

$$\begin{aligned} v &= \ddot{y}_d - K_1 e - K_2 \dot{e} \\ e &= y - y_d \end{aligned} \quad (30)$$

This will lead to

$$\begin{aligned} \ddot{y} &= \ddot{y}_d - K_1 e - K_2 \dot{e} \\ \ddot{e} + K_2 \dot{e} + K_1 e &= 0 \end{aligned} \quad (31)$$

when $K_1 > 0$ and $K_2 > 0$, the error will decay to zero. This means that the output will follow the desired trajectory.

Parameters

Parameters	Values
M	600 kg
m	100 kg
k_s	15000 N/m
b_s	1500 Ns/m
k_t	200000 N/m
b_t	2000 Ns/m
k_{sn}	1000 N/m

Table: System parameters

$$w = \begin{cases} 1 - \cos(8\pi t) & ; 1 \leq t \leq 1.5 \\ 0 & ; \text{otherwise} \end{cases}$$

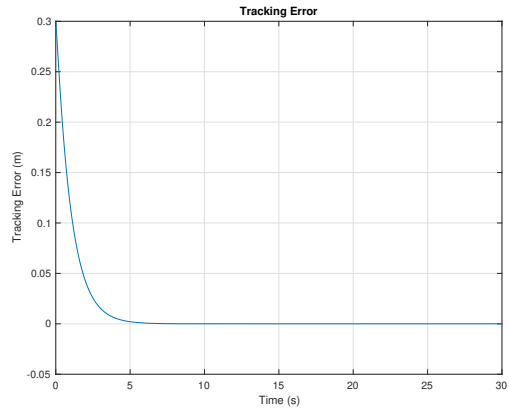
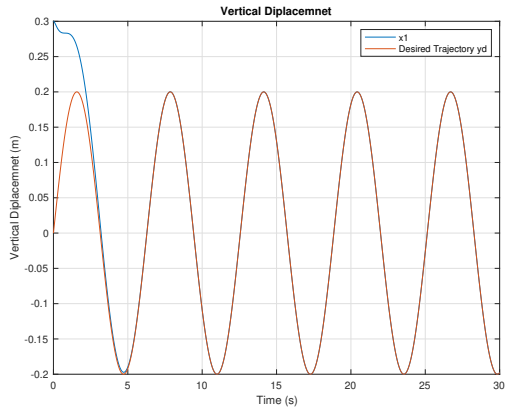
$$y_d = 0.2 \sin(t)$$

$$x = [0.3 \quad 0 \quad 0 \quad 0]^T$$

$$K_1 = 10$$

$$K_2 = 11$$

Results



Results Cont.

