Project-1 Report

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1 Trajectory Optimization

The objective function J(z) is defined as:

$$J(z) = \sum_{k=0}^{N} \frac{1}{2} (x_k - x_t)^T Q(x_k - x_t) + \sum_{k=0}^{N-1} \frac{1}{2} u_k^T R u_k$$

Where:

$$Q = diag(q_{px}, q_{vx}, q_{py}, q_{vy}, q_{\theta}, q_{\omega}), \quad R = diag(r_{u1}, r_{u2})$$
$$x_k = (p_{xk}, v_{xk}, p_{yk}, v_{yk}, \theta_k, \omega_k), \quad u_k = (u_{1k}, u_{2k})$$

 x_k and u_k are the current states and control inputs, respectively. x_t are target states.

Parameters:

$$q_{px}=100.0, q_{vx}=1.0, q_{py}=100.0, q_{vy}=1.0, q_{\theta}=10.0, q_{\omega}=10.0, r_{u1}=1.0, r_{u2}=1.0,$$

System Dynamics Using the Euler Method

$$\begin{split} \mathbf{P}_{x_{k+1}} &= P_{x_k} + V_{x_k} \, \Delta t \\ V_{x_{k+1}} &= V_{x_k} + \left(-\frac{\left(u_{1_k} + u_{2_k}\right) \sin \theta_k}{m} \right) \Delta t \\ P_{y_{k+1}} &= P_{y_k} + V_{y_k} \, \Delta t \\ V_{y_{k+1}} &= V_{y_k} + \left(\frac{\left(u_{1_k} + u_{2_k}\right) \cos \theta_k - mg}{m} \right) \Delta t \\ \theta_{k+1} &= \theta_k + \omega_k \, \Delta t \\ \omega_{k+1} &= \omega_k + \left(\frac{r(u_{1_k} - u_{2_k})}{I} \right) \Delta t \end{split}$$

Inequality Constraints

$$0 \le u_1 \le 10$$

 $0 \le u_2 \le 10$

Targets

The targets are set at each time step to complete the circular trajectory with in given duration:

$$angular_speed = \frac{2\pi}{DURATION}$$

$$\begin{split} & p_{x,target} = r \cdot \cos(angular_speed \cdot t), \\ & p_{y,target} = r \cdot \sin(angular_speed \cdot t), \\ & v_{x,target} = \frac{dp_{x,target}}{dt}, \\ & v_{y,target} = \frac{dp_{y,target}}{dt}, \\ & a_{x,target} = -r \cdot (angular_speed^2) \cdot \cos(angular_speed \cdot t), \\ & a_{y,target} = -r \cdot (angular_speed^2) \cdot \sin(angular_speed \cdot t), \\ & \theta_{target} = -\arctan 2 \left(a_{x,target}, a_{y,target} + G\right), \\ & \omega_{target} = \frac{d\theta_{target}}{dt}. \end{split}$$

Sequential Quadratic Programming (SQP) Algorithm Implementation

1. Initialization:

- The algorithm starts with initial guesses for the states and controls, and Lagrange multipliers.
- Key parameters such as maximum iterations, tolerance, and line search settings are defined.

2. Quadratic Programming Subproblem:

- At each iteration, the algorithm computes the cost, gradient, Hessian, constraints, and their Jacobians.
- A Quadratic Programming (QP) problem is formulated using these values to determine the search direction (p) and updates for multipliers using qpsolvers(CVXOPT).

3. Line Search:

• A line search with merit function is performed to determine the step size (α) for updating the states and controls.

4. Constraint Violation Handling:

• The algorithm tracks both inequality and equality constraint violations.

• Penalties for violations are incorporated into the merit function to prioritize feasible solutions.

5. Convergence and Iteration:

- Convergence is checked using primal and dual optimality conditions.
- The process repeats until the solution meets the predefined tolerance or the maximum number of iterations is reached.

Results

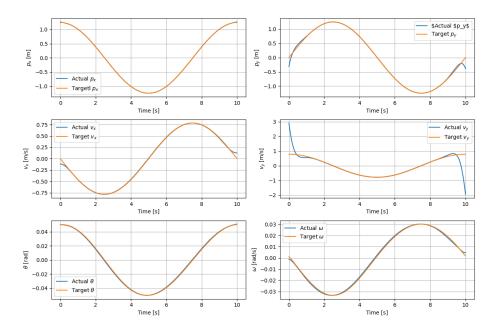


Figure 1: States as function of time

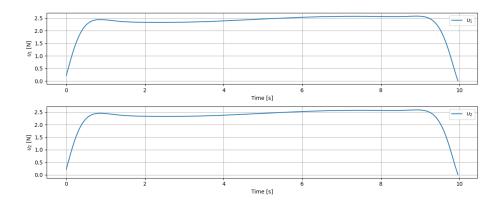


Figure 2: Controls as function of time

2 Model Predictive Control

The objective function J(z) is defined as:

$$J(z) = \sum_{k=0}^{N} \frac{1}{2} (x_k - x_t)^T Q(x_k - x_t) + \sum_{k=0}^{N-1} \frac{1}{2} u_k^T R u_k$$

Where:

$$Q = diag(q_{px}, q_{vx}, q_{py}, q_{vy}, q_{\theta}, q_{\omega}), \quad R = diag(r_{u1}, r_{u2})$$
$$x_k = (p_{xk}, v_{xk}, p_{yk}, v_{yk}, \theta_k, \omega_k), \quad u_k = (u_{1k}, u_{2k})$$

 x_k and u_k are the current states and control inputs, respectively. x_t are target states

Parameters:

$$q_{px} = 300.0, q_{vx} = 1.0, q_{py} = 150.0, q_{vy} = 1.0, q_{\theta} = 10.0, q_{\omega} = 10.0, r_{u1} = 1.0, r_{u2} = 1.0, q_{\theta} = 10.0, q_{\psi} = 10$$

The system Dynamics, Inequality constraints, and targets are same as Trajectory Optimization

MPC Controller Design

- 1. Initial Check for Hover Condition:
 - If the current time step $(t_{current})$ exceeds the total horizon (N), the controller outputs a hover thrust(M * G /2) command to maintain stability.
- 2. Define Prediction Horizon:

• Set the prediction horizon (N_p) based on the current time step $(t_{current})$ and the remaining horizon $(N - t_{current})$.

3. Initialize Variables:

- Initialize the decision variable (z_0) containing states and control inputs for the entire prediction horizon.
- The initial guess for states uses the current state $(x_{current})$, while control inputs are initialized to hover thrust.

4. Set Final State Target:

- If the prediction horizon extends within the total time horizon, set the final state target (x_{N_n}) using predefined trajectory targets $(p_x, v_x, p_y, v_y, \theta, \omega)$.
- Otherwise, use the last predicted state as the final state.

5. Solve the MPC Optimization Problem:

• Call the SQP solver(with CVXOPT), which solves the constrained optimization problem to minimize the cost function while satisfying system dynamics and constraints.

6. Extract First Control Input:

• From the solution , extract the first control input (u_0) for immediate application.

Control Law

The optimal control law is derived iteratively using the Sequential Quadratic Programming (SQP) algorithm. At each step:

- 1. A Quadratic Programming (QP) subproblem is solved to determine the best control inputs and states for the prediction horizon.
- 2. Constraints (e.g., system dynamics, control limits) are enforced during the optimization process.
- 3. The first control input from the optimized sequence is extracted and applied to the system.

Results

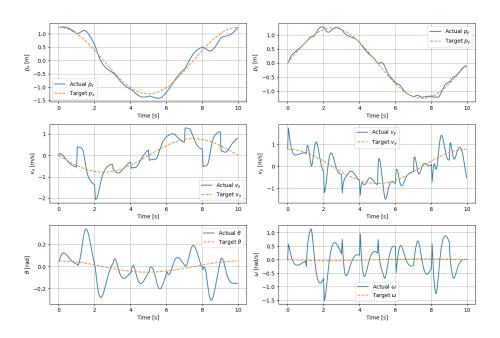


Figure 3: States with disturbance as function of time

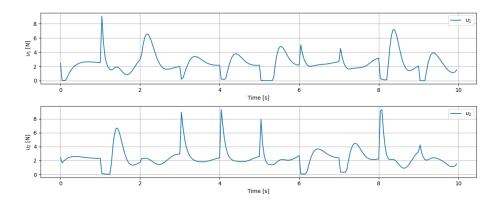


Figure 4: Controls with disturbance as function of time

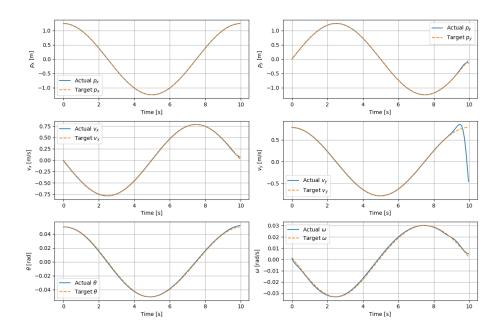


Figure 5: States without disturbance as function of time

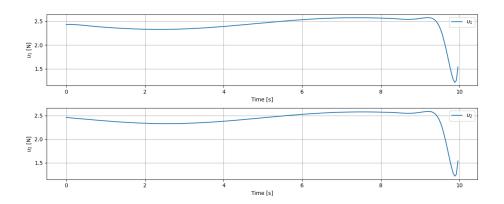


Figure 6: Controls without disturbance as function of time

3 Bonus

State constraint added for the quadrotor to maintain positive altitude.

$$-p_y \le 0$$

Results

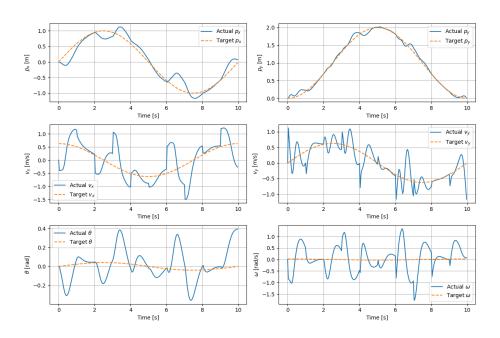


Figure 7: States with disturbance as function of time

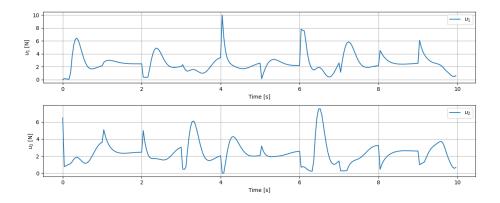


Figure 8: Controls with disturbance as function of time

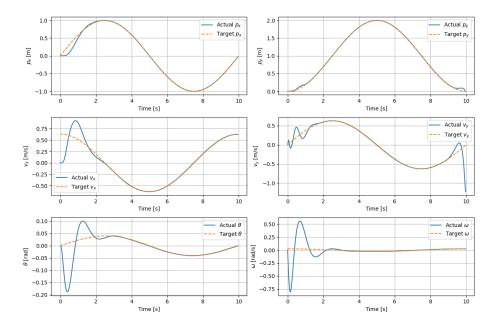


Figure 9: States without disturbance as function of time

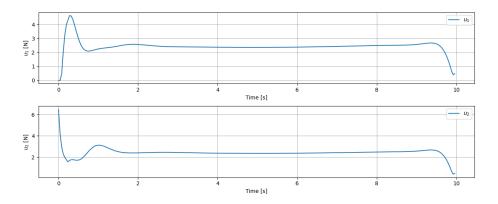


Figure 10: Controls without disturbance as function of time

Note: All animations are available in the runnable Jupyter Notebook or as videos included in the submitted zip folder.