

## Lecture 1: Introduction to CFD

CFD: Computational Fluid Dynamics.

Not just limited to Fluid Dynamics, but to any general transport phenomena. (like heat transfer, mass transfer)

### CFD applications:

- Aerospace - Interior: ventilation system, combustion engine  
Exterior: flow over A/C.
- automobile
- biomedical
- chemical - Mixing of chemicals & bubbles at the interface, separation & mixing, injection of streams.
- Electronics - efficient cooling strategies.
- Energy - & dynamic coupling.
- Fluid structure interaction (two way coupling - fluid interacts with structure & structure interacts with fluid)
- Marine (e.g. flow past ships & boats)
- Materials processing (e.g. grain growth; <sup>CFD</sup> need not be deterministic approach; can be stochastic like Monte Carlo as well, but welding, mold filling)  
↳ helps choose the correct filling process to avoid casting defects.  
Find out if the distribution of impurities via their convection across the material is okay for the material itself & viable.
- Microfluidics - microscale fluid flows (micron or sub-micron scales)  
(e.g. mix two streams - good mixing via pulsating flows, droplet dynamics - micro reactor studies, Fluid structure interaction at small scales, flap placement on micro flow - need to optimise the flap movement - flap acting like a mixer & a pump)

- sports: (racing cars, golf balls, running motion etc.)
- Turbomachines: (flow over blade passages)

• Is CFD inevitable?

Numerical vs Analytical vs Experimental.

• Experimental investigation:

- full scale
  - expensive & often impossible
  - measurement errors.

no substitute for this: seeing is believing

- on a scaled model

- simplified
- difficult to extrapolate results
- measurement errors.

(should upscale/downscale in such a manner that the flow physics <sup>doubt</sup> changes)

eg: micro capillary  
scaled up:  
at small scales,  
surface tension  
forces are  
significant

• Theoretical calculation:

- analytical solutions:

- if exists, gives us exact answers
- but exists only for a few cases.

(need to maintain all similarities - kinematic, dynamic etc....)

{ CFD cannot stand on its own without experimental or analytical solutions. B/c we need to benchmark our solution.

• Sometimes complex

- numerical solutions:

- <sup>exists</sup> for almost any problems.

• Continuous nature of the problem is compromised, but at its expense we get answers to complex problems

## Modeling vs Experimentation.

### • Advantages of modeling:

- cheaper
- more complete information (all details of all variables can be obtained)
- can handle any degree of complexity as long as

### • Disadvantages of modeling:

- deals with a mathematical description not a reality

Numerical solution is as good as the input fed to the problem

- Mathematical description can be inadequate

(Governing eq<sup>ns</sup> may / may not capture the correct physics).

- multiple solutions can exist

(Non-linear problems may have multiple different solutions depending on the IC).

In conclusion: no real substitute for experimentation, but experimen-

-tation is limited by many restrictions & cannot handle multiple trials).

Usual plan of action: try analytical solution  $\rightarrow$  do numerical

Simulation  $\rightarrow$  create good experimental design & validate results.

from ~~the~~ numerical simulation.

$\rightarrow$  cross validate the goodness of ~~exp~~ <sup>numerical</sup> ~~solutions~~ <sup>solutions</sup>.

## Lecture 2: Classification of PDEs.

$$\frac{\partial}{\partial t}(\rho\phi) + \nabla \cdot (\rho \vec{v} \phi) = \nabla \cdot (\Gamma \nabla \phi) + S$$

— 2<sup>nd</sup> order PDE — form  $\nabla \cdot (\Gamma \nabla \phi)$ .

Consider a 2<sup>nd</sup> order PDE of the form:

$$A\phi_{xx} + B\phi_{xy} + C\phi_{yy} + \underbrace{D\phi_{xx} + E\phi_y + F\phi + G}_H = 0. \quad \text{--- ①}$$

$$\begin{aligned} \phi_x &\rightarrow \frac{\partial \phi}{\partial x} \\ \phi_y &\rightarrow \frac{\partial \phi}{\partial y} \\ \phi_{xx} &\rightarrow \frac{\partial^2 \phi}{\partial x^2} \\ \phi_{xy} &\rightarrow \frac{\partial^2 \phi}{\partial x \partial y} \\ \phi_{yy} &\rightarrow \frac{\partial^2 \phi}{\partial y^2} \end{aligned}$$

$A, B, C$  need not be const. They can be fns / lower order partial derivatives / independent variables.

$$\phi = \phi(x, y).$$

One classification —

1. Linear:  $A, B, C \rightarrow$  fns of  $x, y$ .  ~~$D, E, F, G$  fns~~ <sup>remaining terms</sup> linear fns of  $\phi, \phi_x, \phi_y$ .
2. ~~Non linear~~ Quasi-linear:  $A, B, C \rightarrow$  fns of  $x, y, \phi, \phi_x, \phi_y$ .

Characteristics of the PDE:

Highest order derivatives in a PDE may be continuous or discontinuous in the domain of consideration. There may be lines <sup>at which</sup> ~~where~~ which these highest order derivatives <sup>could be</sup> ~~are~~ discontinuous. Such lines are called characteristics <sup>lines</sup> of the PDE.

Why they're important? B/C in our numerical method, we have to account for such discontinuities a priori.

Objective: to get characteristics of the PDE

$$\phi_x = \phi_x(x, y) \Rightarrow d\phi_x = \frac{\partial \phi_x}{\partial x} dx + \frac{\partial \phi_x}{\partial y} dy.$$

$$d\phi_x = \phi_{xx} dx + \phi_{xy} dy. \quad \text{--- ②}$$

$$\text{Similarly } \phi_y = \phi_y(x, y) \Rightarrow d\phi_y = \phi_{yx} dx + \phi_{yy} dy.$$

$$\therefore \phi_{xy} = \phi_{yx}, \quad d\phi_y = \phi_{xy} dx + \phi_{yy} dy. \quad \text{--- ③}$$

From ①, ②, ③.

③

$$\begin{bmatrix} A & B & C \\ dx & dy & 0 \\ 0 & dx & dy \end{bmatrix} \begin{bmatrix} \phi_{xx} \\ \phi_{xy} \\ \phi_{yy} \end{bmatrix} = \begin{bmatrix} -H \\ d\phi_x \\ d\phi_y \end{bmatrix}$$

We are interested in the case where the solution of this matrix expression  $\rightarrow \begin{bmatrix} \phi_{xx} \\ \phi_{xy} \\ \phi_{yy} \end{bmatrix}$  doesn't exist.

Analogy with algebraic eq<sup>ns</sup>:

$$\begin{cases} x+y=2 \\ 2x+2y=5 \end{cases} \text{ soln doesn't exist.}$$

$$\begin{bmatrix} 1 & 1 \\ 2 & 2 \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix} = \begin{bmatrix} 2 \\ 5 \end{bmatrix}$$

$$\hookrightarrow \Delta = 0.$$

we want to find locus of pts across which we have discontinuities in  $\phi_{xx}$ ,  $\phi_{yy}$ , &  $\phi_{xy}$ . That is possible when  $\det \begin{pmatrix} A & B & C \\ dx & dy & 0 \\ 0 & dx & dy \end{pmatrix} = 0$

For  $\phi_{xx}$ ,  $\phi_{yy}$ ,  $\phi_{xy}$  to be discontinuous,

$$\Delta = 0.$$

$$\begin{vmatrix} A & B & C \\ dx & dy & 0 \\ 0 & dx & dy \end{vmatrix} = 0 \Rightarrow$$

{Only coeffs of highest order derivatives matters}

$$A(dy)^2 - B(dx dy) + C(dx)^2 = 0$$

$$A\left(\frac{dy}{dx}\right)^2 - B\left(\frac{dy}{dx}\right) + C = 0$$

$$\frac{dy}{dx} = \frac{B \pm \sqrt{B^2 - 4AC}}{2A}.$$

Number of real characteristics existing will depend on whether  $B^2 - 4AC \geq 0$  or  $< 0$ .

If  $B^2 - 4AC = 0 \rightarrow$  only one real characteristic.  
(Parabolic).

$B^2 - 4AC < 0 \rightarrow$  no real characteristics (elliptic PDE)

$B^2 - 4AC > 0 \rightarrow$  2 real characteristics (hyperbolic PDE)

## Lecture 3: Examples of PDEs

Ex 1:  $\nabla^2 \phi = 0$  (Laplace eq<sup>n</sup> - most commonly encountered simple PDE)

$\nabla^2 T = 0$  (T: Temperature, uniform heat conductivity, steady state, no heat source).

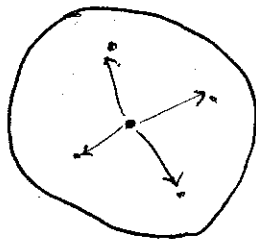
2D example:

$$\frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$$

$A=1, B=0, C=1$

$B^2 - 4AC = -4 \rightarrow$  elliptic equation.

Say, we have a domain with uniform temperature throughout. Put a heat source at a pt  $\rightarrow$  acts as a disturbance that propagates in all directions in the domain at infinite speed. i.e. Thermal disturbance propagates in all directions with infinite speed.



This disturbance tries to nullify the temperature differences at different points in the domain to make the temp distribution homogeneous everywhere.

- So while numerically formulating the problem, a point on the domain under our consideration will be influenced by all the other points in our domain.
- B/C has to be specified at the boundary of the domain. The B/C can be discontinuous. For eg: half the boundary may be in contact with steam & the other half in contact with ice, so discontinuities are possible on the boundary. But since the disturbance travels at infinite speeds (message propagation is fast) means that there will not be discontinuities within the domain.
- This type of problem  $\rightarrow$  BVP.

$$E \propto -2 \quad \frac{\partial \phi}{\partial t} = \alpha \frac{\partial^2 \phi}{\partial x^2}$$

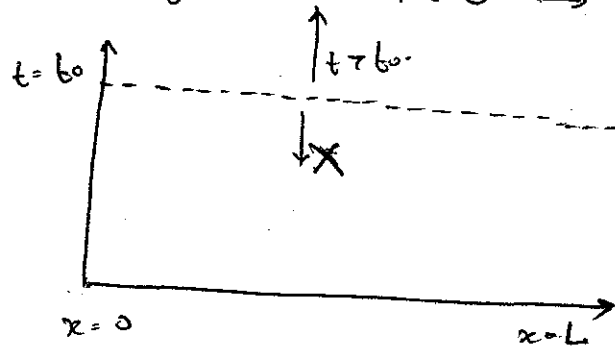
Eg of 1D unsteady heat conduction.

$$A = \alpha.$$

$$B = 0.$$

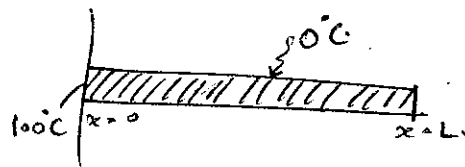
$$C = 0.$$

$$B^2 - 4AC = 0 \rightarrow \text{parabolic} : \text{one characteristic}$$



Say we have a 1D rod at uniform temp. At  $t = t_0$ , create a sudden disturbance in  $T$  at one end of the rod.

Absrupt disturbance at  $t = t_0$  will propagate in a direction forward in time.



The disturbance at  $t = t_0$  will influence what will happen for  $t > t_0$ . It cannot influence back what has already happened sometime back.

~~disturbance~~  $\rightarrow$  time marching problems.

- will have only one type of discontinuity, that discontinuity at reference time at which the disturbance is imposed.

- At time  $t > t_0$ , the abrupt disturbance originated at  $t = t_0$  may have made its presence known throughout some part of the domain. That part is called domain of influence.

- The <sup>total</sup> region in the domain where the presence of disturbance can potentially make its presence known is called domain of disturbance.

$t > t_0 \rightarrow$  domain of influence

~~to all~~  $t \leq t_0 \rightarrow$  domain of disturbance

In elliptic case  $\rightarrow$  entire domain is the domain of influence.

Initial-boundary value problem.

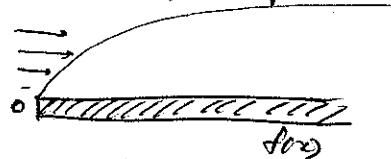
~~Exp~~ since  $t \rightarrow t_0$  disturbance introduced.

As  $t \rightarrow t_0$ , steady state reached  $\Rightarrow \frac{\partial}{\partial t} = 0$  or it becomes elliptic

So it has some elliptic nature build into it.

So more correct way of saying would be that this eq<sup>n</sup> is parabolic in time and elliptic in space.

. Another eg. BL over a flat plate



- Space marching problems - disturbance at  $x=0$ . whatever happens before  $x=0$  is not influenced by the disturbance at  $x=0$ .

It's possible b/c of high Re. High Re  $\Rightarrow$  high inertial forces. Inertial forces are predominantly uni-directional compared to viscous forces which spread out in all directions. High Re  $\Rightarrow$  disturbances predominantly carried uni-directionally.

(not until & unless B/L separation occurs).

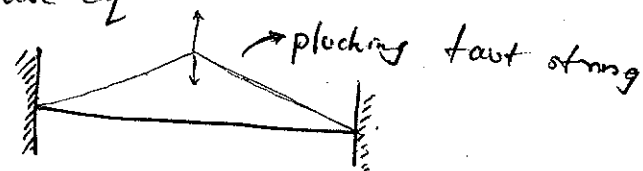
Ex-3:  $\frac{\partial^2 \phi}{\partial t^2} = c^2 \frac{\partial^2 \phi}{\partial x^2}$  wave eq<sup>n</sup>.

Use

$x = X$

$t = Y$

} to formulate in the form  $A\phi_{xx} + B\phi_{xy} + C\phi_{yy} = -H$ .



$A = +c^2$

$B = 0$

$C = -1$

$B^2 - 4AC = 4c^2$

$\frac{dy}{dx} \frac{dY}{dX} = \frac{\pm 2c}{2c^2} = \pm \frac{1}{c}$

$\frac{dt}{dx} = \pm \frac{1}{c}$

$\therefore \frac{dx}{dt} = \pm c$



Integrating,

$$x = \pm ct + C_1 \rightarrow \text{Hyperbolic eqns.}$$

Forget  $C_1$  (It just shifts the sol<sup>n</sup> by a const. amount everywhere).

main characteristics,  $x - ct = \xi$ .

$$x + ct = \eta.$$

Effect is combined spatio-temporal effect.

It's possible to write the entire eq<sup>n</sup> in terms of characteristic variables  $\xi$  and  $\eta$ .

Lecture 4: Examples of partial differential equations (contd).

$$\frac{\partial^2 \phi}{\partial t^2} = c^2 \frac{\partial^2 \phi}{\partial x^2} \quad : \quad \left. \begin{array}{l} \xi = x - ct \\ \eta = x + ct \end{array} \right\} \text{two characteristic variables.}$$

$$\phi(x, t) \rightarrow \phi(\xi, \eta).$$

Sol<sup>n</sup> can be written in terms of characteristic variables

$$\frac{\partial \phi}{\partial t} = \frac{\partial \phi}{\partial \xi} \frac{\partial \xi}{\partial t} + \frac{\partial \phi}{\partial \eta} \frac{\partial \eta}{\partial t}$$

$$\begin{aligned} \frac{\partial^2 \phi}{\partial t^2} &= \frac{\partial}{\partial \xi} \left[ -c \frac{\partial \phi}{\partial \xi} + c \frac{\partial \phi}{\partial \eta} \right] \frac{\partial \xi}{\partial t} + \frac{\partial}{\partial \eta} \left[ -c \frac{\partial \phi}{\partial \xi} + c \frac{\partial \phi}{\partial \eta} \right] \frac{\partial \eta}{\partial t} \\ &= c^2 \frac{\partial^2 \phi}{\partial \xi^2} + c^2 \frac{\partial^2 \phi}{\partial \eta^2} - 2c^2 \frac{\partial^2 \phi}{\partial \xi \partial \eta} \quad \dots (1) \end{aligned}$$

$$\frac{\partial \phi}{\partial x} = \frac{\partial \phi}{\partial \xi} \frac{\partial \xi}{\partial x} + \frac{\partial \phi}{\partial \eta} \frac{\partial \eta}{\partial x}$$

$$\begin{aligned} \frac{\partial^2 \phi}{\partial x^2} &= \frac{\partial}{\partial \xi} \left( \frac{\partial \phi}{\partial \xi} + \frac{\partial \phi}{\partial \eta} \right) \frac{\partial \xi}{\partial x} + \frac{\partial}{\partial \eta} \left( \frac{\partial \phi}{\partial \xi} + \frac{\partial \phi}{\partial \eta} \right) \frac{\partial \eta}{\partial x} \\ &= \frac{\partial^2 \phi}{\partial \xi^2} + \frac{\partial^2 \phi}{\partial \eta^2} + 2 \frac{\partial^2 \phi}{\partial \xi \partial \eta} \quad \dots (2) \end{aligned}$$

Apply (1) and (2) in original wave eq<sup>n</sup>.

$$\text{gives,} \quad 4 \frac{\partial^2 \phi}{\partial \xi \partial \eta} = 0 \quad \text{or} \quad \frac{\partial^2 \phi}{\partial \xi \partial \eta} = 0 \rightarrow \frac{\partial}{\partial \xi} \left( \frac{\partial \phi}{\partial \eta} \right) = 0.$$

$$\Rightarrow \frac{\partial \phi}{\partial n} = f_1(n).$$

Integrating,  $\phi = F(n) + G(\xi).$

Conclusion: general sol<sup>n</sup> can be written in terms of characteristic variables.

IC  $\rightarrow$  At  $t=0$ ,  $\phi = f(x).$

$$t=0, \frac{\partial \phi}{\partial t} = g(x).$$

$$F(x) + G(x) = f(x)$$

$$cF'(x) - cG'(x) = g(x)$$

$$\Rightarrow F(x) - G(x) = \frac{1}{c} \int_0^x g(\tau) d\tau.$$

$$F(x) = \frac{1}{2c} \int_0^x g(\tau) d\tau + C_1$$

$$G(x) = \frac{1}{2} f(x) - \frac{1}{2c} \int_0^x g(\tau) d\tau.$$

$$\phi = F(x) + G(\xi).$$

$$= F(x+ct) + G(x-ct)$$

$$= \frac{1}{2} [f(x+ct) + f(x-ct)] + \frac{1}{2c} \int_{x-ct}^{x+ct} g(\tau) d\tau$$

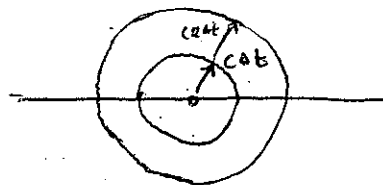
Say disturbance propagates in a fluid medium.

disturbance speed = sonic speed.

$c$  = sonic speed.

$u$  = speed of the source of disturbance.

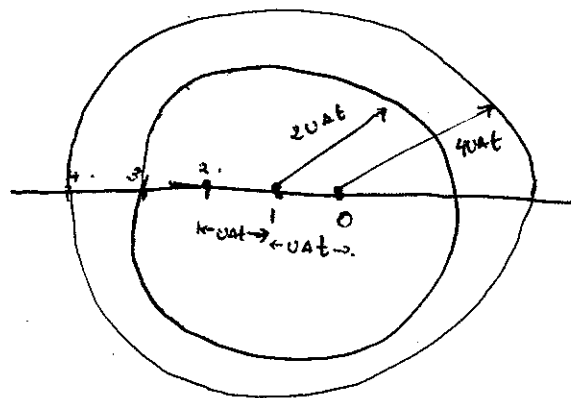
Ex-1  $Ma = 0$  (source of disturbance  $\{Ma = u/c\}$  doesn't move).



$\leftarrow$  Take  $\Delta t$ ,  $2\Delta t$ ,  $3\Delta t$  etc....

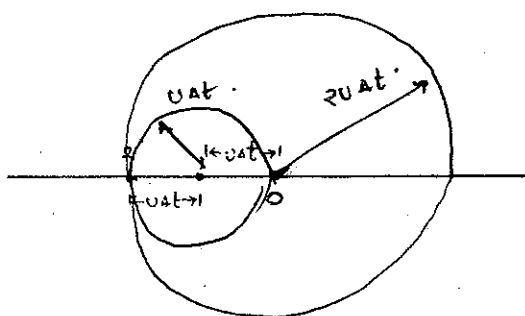
Ex-2

$$Ma = \frac{1}{2} \rightarrow \frac{u}{c} = \frac{1}{2}$$



Ex-3

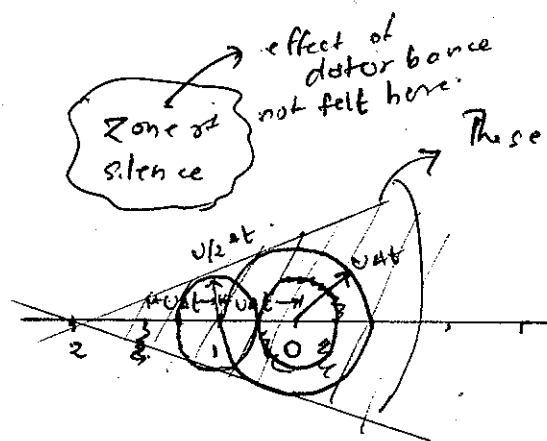
$$Ma = 1 \rightarrow u = c$$



Disturbance wave doesn't propagate more than where the source is located.

Ex-4

$$Ma > 1 \quad Ma = 2 \rightarrow u = 2c$$



These two straight lines are characteristic lines (weak discontinuities)

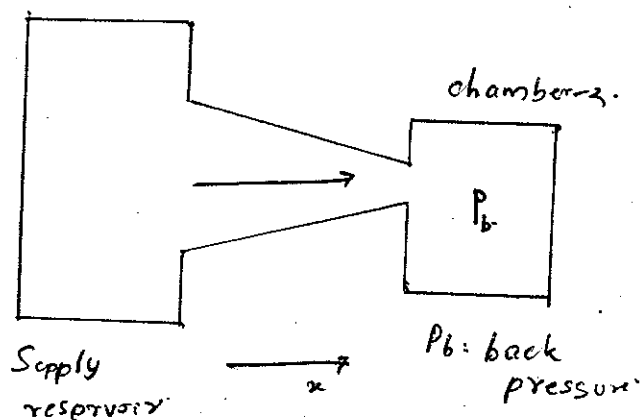
Mach cone - imaginary cone within which the effects of disturbance is felt.

Highly incompressible flows - disturbance can propagate only with finite speed

(unlike in incompressible case where speed is infinite).

As such the effect of disturbance gets accumulated

Say we have a converging nozzle.



To increase flow rates,  $P_b \downarrow$ .

$\Rightarrow n \uparrow$

$P_b$  regulation ~~is~~ essentially creates ~~inflow~~ ~~creating~~ a disturbance in chamber-2 which is propagated upstream & gives message to supply reservoir to respond to that & send more mass flow.

But what actually happens is That the mass flow rate can be increased up to  $Ma=1$  by decreasing  $P_b$ . Beyond that limit it cannot be increased.

Explanation:  
 $\vec{V}_{Df}$  = velocity of disturbance relative to flow  
 $= \vec{V}_b - \vec{V}_f$

$$\vec{V}_f = +c$$

$$\vec{V}_{bf} = -c$$

$$\Rightarrow \vec{V}_b = 0$$

In hyperbolic cases, where source of disturbance moves faster than the disturbance itself, it leads to discontinuities in the flow medium which has to be taken into account while designing the numerical simulation.

# Lecture 5: Nature of the characteristics of partial differential equations

2<sup>nd</sup> order  
General pde form:

$$\sum_i \sum_j A_{ij} \frac{\partial^2 \phi}{\partial x_i \partial x_j} + B = 0. \quad \text{lower order terms}$$

Nature of

Characteristics depends on eigen value of A.

(coeff matrix)

To get eigs, use  $\det |A - \lambda I| = 0 \rightarrow \lambda_i$ .

- If any  $\lambda$  is zero  $\rightarrow$  parabolic
- If none is zero and all  $\lambda$ s are of the same sign  $\rightarrow$  elliptic
- If none is zero and all but one  $\lambda$  is opposite sign  $\rightarrow$  hyperbolic.

Ex  $(1 - M_\infty^2) \frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0 \rightarrow$  eq<sup>n</sup> relating velocity potential for isentropic inviscid compressible flow over air bodies with slender shapes.

$$A_{11} \frac{\partial^2 \phi}{\partial x^2} + A_{22} \frac{\partial^2 \phi}{\partial y^2}$$

$$A_{12} \frac{\partial^2 \phi}{\partial x_1 \partial x_2} + A_{21} \frac{\partial^2 \phi}{\partial x_2 \partial x_1} + A_{22} \frac{\partial^2 \phi}{\partial x_2^2} + B = 0$$

$$x_1 = x$$

$$x_2 = y$$

$$A_{11} = 1 - M_\infty^2$$

$$A_{12} = A_{21} = B = 0$$

$$A_{22} = 1$$

$$\left| \begin{array}{cc} (1 - M_\infty^2) - \lambda & 0 \\ 0 & 1 - \lambda \end{array} \right| = 0$$

$$(1 - \lambda - M_\infty^2)(1 - \lambda) = 0$$

$$\lambda = 1,$$

$$\lambda = 1 - M_\infty^2$$

If  $M_\infty = 1 \rightarrow$  parabolic

$M_\infty < 1 \rightarrow$  elliptic

$M_\infty > 1 \rightarrow$  hyperbolic

So the same stream eq<sup>n</sup> depending on  $M_\infty$  can be parabolic, hyperbolic or elliptic.

- Ex. • Heat transfer  
• Unsteady  
• 1D  
• low  $\frac{k}{\rho c_p} \rightarrow 0$   
•  $U = U_\infty$

$$\frac{\partial}{\partial t} (\rho T) + \nabla (\rho \vec{V} T) = \nabla \left( \frac{k}{c_p} \nabla T \right) + S$$

Using the conditions given,,

$$\nabla (\rho \vec{V} T) = \frac{d}{dx} (\dots)$$

$$\nabla \left( \frac{k}{c_p} \nabla T \right) \rightarrow 0$$

$$\Rightarrow \frac{\partial T}{\partial t} + U_\infty \frac{\partial T}{\partial x} = S$$

$$T = T(x, t)$$

$$dT = \frac{\partial T}{\partial x} dx + \frac{\partial T}{\partial t} dt$$

$$\begin{bmatrix} 1 & U_\infty \\ dt & dx \end{bmatrix} \begin{bmatrix} \frac{\partial T}{\partial t} \\ \frac{\partial T}{\partial x} \end{bmatrix} = \begin{bmatrix} S \\ dT \end{bmatrix}$$

$$| | = 0 \rightarrow dx - U_\infty dt = 0 \rightarrow \frac{dx}{dt} = U_\infty \leftarrow \text{characteristics.}$$

What is the nature of characteristics here? Not parabolic!

Converting to  
2nd order  
form

$$\frac{\partial T}{\partial t} + U_\infty \frac{\partial T}{\partial x} = 0$$

$$\frac{\partial T}{\partial t} + U_\infty \frac{\partial T}{\partial x} = 0$$

$$\frac{\partial^2 T}{\partial t \partial x} + U_\infty \frac{\partial^2 T}{\partial x^2} = 0 \dots \textcircled{1}$$

$$\frac{\partial^2 T}{\partial t^2} + U_\infty \frac{\partial^2 T}{\partial x \partial t} = 0 \dots \textcircled{2}$$

$$\textcircled{1} \times U_\infty - \textcircled{2} \rightarrow$$

$$U_\infty^2 \frac{\partial^2 T}{\partial x^2} - \frac{\partial^2 T}{\partial t^2} = 0$$

$$\frac{\partial^2 T}{\partial t^2} = U_\infty^2 \frac{\partial^2 T}{\partial x^2} \rightarrow \underline{\text{hyperbolic!}}$$

HW:  $\left. \begin{aligned} \frac{\partial u}{\partial x} &= \frac{\partial v}{\partial y} \\ \frac{\partial u}{\partial y} &= -v \end{aligned} \right\} \text{Find the nature of pde:}$

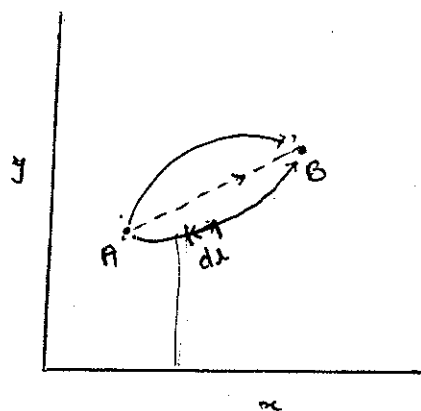
HW: Make a chart for the features of parabolic, hyperbolic, elliptic eq<sup>ns</sup>  
Nature of the characteristics, how many characters, zone of influence,  
Zone of disturbance, speed of propagation of disturbance.

# Lecture 6: Euler-Lagrangian Equation

- Error minimization — key principle with which many numerical methods are founded.
  - B/c a good approximate sol<sup>n</sup> is the one which incurs least error.
  - includes many considerations — one such is variations.

- Calculus of variations in brief:

Say we have two pts. Objectives: find the path with least distance b/w them.



$$dl = \sqrt{dx^2 + dy^2}$$

$$= \sqrt{1 + \left(\frac{dy}{dx}\right)^2} \cdot dx$$

$$y' \equiv \frac{dy}{dx}, \quad y'' \equiv \frac{d^2y}{dx^2}$$

$$L = \int dl = \int \sqrt{1 + y'^2} dx$$

Problem statement becomes:

Find the path AB that minimizes  $L$

$$\Rightarrow \text{minimizes } I = \int \sqrt{1 + y'^2} dx$$

$$I = \int F(x, y, y') dx$$

To minimize  $I$ , take  $\delta I$ .

$\delta I$  — arbitrarily small virtual change in  $I$ .

$\delta I$  will only involve changes in dependent variables, not on independent variable ( $x$  here).

Why? We are trying to find what the  $y$  should

be to minimize  $I$ , so that ~~we~~<sup>ourselves</sup> fall on the desired path i.e. the straight line.

$F(x, y, y')$   
 independent variable:  $x$  — themselves fns of  $x$   
 $y$  and  $y'$  are functions of  $x$   
 $\therefore F$  is a functional

$$\delta I = \int \left[ \frac{\partial F}{\partial y} \delta y + \frac{\partial F}{\partial y'} \delta y' \right] dx$$

$y$  is fixed at  $A$  and  $B$ .  $\therefore \delta y = 0$  at  $A, B$ .

Simplify 2<sup>nd</sup> term ~~by~~<sup>using</sup> integration by parts:

$$\delta I = \int \left[ \frac{\partial F}{\partial y} \right] \delta y + \left[ \frac{\partial F}{\partial y'} \delta y \right]_A^B - \int \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \delta y dx$$

Minimize  $I = \int F(x, y, y') dx$  subject to constraint

$$\therefore 2^{nd} \text{ term } \left[ \frac{\partial F}{\partial y'} \delta y \right]_A^B = 0$$

$y @ A, y @ B$  specified.

$$(\delta y @ A = \delta y @ B = 0).$$

$$\therefore \delta I = \int \left[ \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \right] \delta y dx$$

For min  $I$ ,  $\delta I = 0$  for any arbitrary  $\delta y$ .

This is possible when integrand = 0.

$$\Rightarrow \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) = 0 \leftarrow \text{Euler-Lagrange equation.}$$

for minimization of dist b/w two pts is this example.

$$I = \int \sqrt{1+y'^2} dx \Rightarrow F(x, y, y') = \sqrt{1+y'^2}$$

$$\frac{\partial F}{\partial y} = 0, \quad \frac{\partial F}{\partial y'} = \frac{2y'}{2\sqrt{1+y'^2}} = \frac{y'}{\sqrt{1+y'^2}}$$

$$\therefore \text{E-L eqn: } 0 - \frac{d}{dx} \left( \frac{y'}{\sqrt{1+y'^2}} \right) = 0$$

$$\text{or } \frac{d}{dx} \left( \frac{y'}{\sqrt{1+y'^2}} \right) = 0 \Rightarrow \frac{y'}{\sqrt{1+y'^2}} = \text{const}$$

$\Rightarrow y' = \text{const} = C$   
or  $\frac{dy}{dx} = C \rightarrow \text{path is a straight line.}$



# Lecture 7: Approximate Solutions of Differential Equations

(Prob) Show that an alternative form of the Euler-Lagrange eq<sup>n</sup> is given

by  $\frac{\partial F}{\partial x} - \frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) - \frac{\partial F}{\partial x} = 0$ , where  $F(x, y, y')$ . ①

(Ans)  $dF = \frac{\partial F}{\partial x} dx + \frac{\partial F}{\partial y} dy + \frac{\partial F}{\partial y'} dy'$

$$\Rightarrow \frac{dF}{dx} = \frac{\partial F}{\partial x} + \frac{\partial F}{\partial y} \frac{dy}{dx} + \frac{\partial F}{\partial y'} \frac{dy'}{dx}$$

$$\frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) = y' \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) + \frac{dy'}{dx} \left( \frac{\partial F}{\partial y'} \right)$$

$$\therefore \frac{dy'}{dx} \left( \frac{\partial F}{\partial y'} \right) = \frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) - y' \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right)$$

Sub into main expression:  $\frac{dF}{dx}$

$$\Rightarrow \frac{dF}{dx} = \frac{\partial F}{\partial x} + \frac{\partial F}{\partial y} \frac{dy}{dx} + \frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) - y' \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \dots \textcircled{2}$$

$$= \frac{\partial F}{\partial x} + \frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) + y' \left[ \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \right]$$

$$\Rightarrow \frac{dF}{dx} = \frac{d}{dx} \left( y' \frac{\partial F}{\partial y'} \right) - \frac{\partial F}{\partial x} = 0$$

(since this is same as original E-L eq<sup>n</sup>).

(Prob-2) Hence show that the closed curve that minimizes the perimeter for a given area is a circle.

$$P = \int dl = \int \sqrt{dx^2 + dy^2} = \int \sqrt{1 + y'^2} dx, \quad y' = \frac{dy}{dx}$$

$$A = A^* = \int y dx$$

Obj: minimize  $P$  s.t.  $A$  is a const.

Do this via Lagrange multiplier.

Introduce  $F = P + dA$ , where  $d$  is Lagrange multiplier.

$$I = \int (P + \lambda A) = \int \underbrace{(\sqrt{1+y'^2} + \lambda y)}_{F(x,y,y')} dx$$

Use alternate form of E-L eq<sup>n</sup>.

$$\frac{d}{dx} \left[ F - y' \frac{\partial F}{\partial y'} \right] - \frac{\partial F}{\partial x} = 0$$

$$\Rightarrow F - y' \frac{\partial F}{\partial y'} = \text{const } C$$

$$\sqrt{1+y'^2} + \lambda y - \frac{(y')^2}{\sqrt{1+y'^2}} = C$$

$$\Rightarrow 1 + \cancel{y'^2} + \lambda y - \frac{\cancel{y'^2}}{\sqrt{1+y'^2}} = C \sqrt{1+y'^2}$$

$$1 = (C - \lambda y) \sqrt{1+y'^2}$$

$$\text{Use } y' = \tan \theta = \frac{dy}{dx}$$

$$\cos \theta = C - \lambda y$$

$$\frac{dy}{dx} = \frac{dy/d\theta}{dx/d\theta}$$

$$\frac{d}{d\theta} (\cos \theta) = \frac{d}{d\theta} (C - \lambda y)$$

$$-\sin \theta = -\lambda \frac{dy}{d\theta}$$

$$\text{So, } \frac{dy}{d\theta} = \frac{1}{\lambda} \sin \theta$$

$$\text{So, } \frac{dy}{dx} = \frac{\frac{1}{\lambda} \sin \theta}{\left( \frac{dx}{d\theta} \right)}$$

$$\tan \theta = \frac{\frac{1}{\lambda} \sin \theta}{\left( \frac{dx}{d\theta} \right)}$$

$$\therefore \frac{dx}{d\theta} = \frac{1}{\lambda} \cos \theta$$

$$\therefore x = \frac{\sin \theta}{\lambda} + C_1$$

$$\sin \theta = \lambda(x - C_1)$$

$$\cos \theta = C - \lambda y$$

$$\sin^2 \theta + \cos^2 \theta = 1$$

$$\Rightarrow [\lambda(x - C_1)]^2 + [C - \lambda y]^2 = 1$$

$$\lambda^2 (x - C_1)^2 + \lambda^2 (y - C_2)^2 = 1$$

This is of the form

$$(x - a)^2 + (y - b)^2 = r^2$$

$\Rightarrow$  eq<sup>n</sup> of circle

Thus circle minimizes

perimeter for const area shapes!

Ex: Functionals involving higher order derivatives:

Say,  $I = \int F(x, y, y', y'') dx$

$$\delta I = \int \left[ \frac{\partial F}{\partial y} \delta y + \frac{\partial F}{\partial y'} \delta y' + \frac{\partial F}{\partial y''} \delta y'' \right] dx$$

to denote first order variations (usually omitted)

$$\left[ \frac{\partial F}{\partial y'} \delta y \right]_A^B - \int \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \delta y dx$$

$$- \left[ \frac{d}{dx} \left( \frac{\partial F}{\partial y''} \right) \delta y' \right]_A^B + \int \frac{d^2}{dx^2} \left( \frac{\partial F}{\partial y''} \right) \delta y dx$$

$$\left[ \frac{\partial F}{\partial y''} \delta y' \right]_A^B - \int \frac{d}{dx} \left( \frac{\partial F}{\partial y''} \right) \delta y' dx$$

Boundary terms:

$$\left[ \frac{\partial F}{\partial y'} \delta y \right]_A^B, \left[ \frac{\partial F}{\partial y''} \delta y' \right]_A^B, - \left[ \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) \delta y \right]_A^B$$

Boundary terms = 0 if  $\delta y, \delta y' = 0$ .

So assume  $y, y'$  are specified at A & B:

$$\Rightarrow \delta I = \int \left[ \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) + \frac{d^2}{dx^2} \left( \frac{\partial F}{\partial y''} \right) \right] \delta y dx = 0$$

$$\delta I = \int \left[ \frac{\partial F}{\partial y} + \frac{d^2}{dx^2} \left( \frac{\partial F}{\partial y''} \right) \right] \delta y dx = 0$$

$$\Rightarrow \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) + \frac{d^2}{dx^2} \left( \frac{\partial F}{\partial y''} \right) = 0 \quad \text{to minimize } I.$$

For more higher order eq<sup>n</sup>, we have the form:

$$\left\{ \frac{\partial F}{\partial y} - \frac{d}{dx} \left( \frac{\partial F}{\partial y'} \right) + \frac{d^2}{dx^2} \left( \frac{\partial F}{\partial y''} \right) - \frac{d^3}{dx^3} \left( \frac{\partial F}{\partial y'''} \right) + \frac{d^4}{dx^4} \left( \frac{\partial F}{\partial y^{(4)}} \right) - \dots = 0 \right\}$$

↳ Euler-Poisson form.

## Approximate solutions of differential equations Through variational formulation;

Eg: 1D steady state heat conduction with constant heat source;

$$\frac{\partial}{\partial t}(\rho T) + \nabla(l \vec{V} T) = \nabla\left(\frac{k}{C_p} \nabla T\right) + \frac{S}{C_p}$$

Also assume: const. Thermal properties  $k, C_p$  etc const.

$$\text{Steady state} \Rightarrow \frac{\partial}{\partial t}(\dots) = 0$$

Conduction problem  $\Rightarrow$  no flow velocity involved.

$$\therefore \nabla(l \vec{V} T) = 0$$

$$1D \text{ problem} \Rightarrow \nabla(\dots) = \frac{d}{dx}(\dots)$$

$$\text{Final form: } \frac{d}{dx}\left(k \frac{dT}{dx}\right) + S = 0. \rightarrow \text{"D" form (Differential form)}$$

To make it a variational form, multiply with a  $\rightarrow$  Strong form.  
variational parameter  $v$  & integrate over the domain.

$$\int \left[ \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S \right] v dx = 0$$

$v$  carries the meaning of  $\delta T$ .

Use integration by parts:

$$\left[ v k \frac{dT}{dx} \right]_1^2 - \int k \frac{dv}{dx} \frac{dT}{dx} dx + \int S v dx = 0 \quad \left\{ \begin{array}{l} \text{don't mix the} \\ \text{expression!} \end{array} \right.$$

Boundary conditions possible:

$\rightarrow T$  specified.

$\rightarrow \delta T = 0$  or  $v = 0$ .

$\rightarrow \frac{dT}{dx}$  specified.

} either of the two.

Ex:  $T$  specified at both boundaries.

$$\Rightarrow \boxed{\int k \frac{dT}{dx} \frac{dv}{dx} dx = \int S v dx} \rightarrow \text{Weak form.}$$

"V" form.

## Lecture 8: Variational formulation

In the weak form, it requires continuity only upto the first order derivative, while in the strong form, continuity upto the second order derivative is required. Hence why the name - weak form

generic form of "V" form:

$$a(T, v) = \underbrace{\quad}_{\text{bilinear operator}} \underbrace{\quad}_{\text{linear operator}}$$

why this form?

Make some observations:

obs-1

$$a(\alpha_1 T + \alpha_2 v, \beta_1 T + \beta_2 v) = \int k \frac{d}{dx} (\alpha_1 T + \alpha_2 v) \frac{d}{dx} (\beta_1 T + \beta_2 v) dx$$

$$= \alpha_1 \beta_1 \int k \frac{dT}{dx} \frac{dT}{dx} dx + \alpha_1 \beta_2 \int k \frac{dT}{dx} \frac{dv}{dx} dx + \alpha_2 \beta_1 \int k \frac{dv}{dx} \frac{dT}{dx} dx +$$

$$\alpha_2 \beta_2 \int k \frac{dv}{dx} \frac{dv}{dx} dx$$

$$= \alpha_1 \beta_1 a(T, T) + \alpha_1 \beta_2 a(T, v) + \alpha_2 \beta_1 a(v, T) + \alpha_2 \beta_2 a(v, v).$$

$\Rightarrow a(\cdot, \cdot)$  is bilinear (i.e. linear in each slot)

obs-2

$$L(\alpha T + \beta v) = \int k \frac{d}{dx} (\alpha T + \beta v) \frac{d}{dx} (\alpha T + \beta v) dx$$

$$= \alpha L(T) + \beta L(v), \text{ if such a property is satisfied,}$$

$L$  is a linear operator

obs-3

$$a(T, v) = a(v, T) \Rightarrow a \text{ is symmetric (self adjoint)}$$

obs-4

$$a(v, v) = \int k \left( \frac{dv}{dx} \right)^2 dx \text{ is a true definite operator}$$

$\downarrow$  integral over the domain is true

$a$  is a scalar product on  $V$ .

Say we have a function  $g(\epsilon) = \frac{1}{2} a(T + \epsilon v, T + \epsilon v) - l(T + \epsilon v)$

M) problem = Minimize  $g$  at  $\epsilon = 0 \rightarrow \frac{1}{2} a(T, T) - l(T)$

Assumptions:

$\rightarrow a$  is bilinear &  $l$  is linear

$$g(\epsilon) = \frac{1}{2} a(T, T) + \frac{\epsilon}{2} a(T, v) + \frac{\epsilon}{2} a(v, T) + \frac{\epsilon^2}{2} a(v, v) - l(T) - \epsilon l(v)$$

To minimize 'g',  $g'(\epsilon)|_{\epsilon=0} = 0$

$$g'(\epsilon)|_{\epsilon=0} = \frac{1}{2} a(T, v) + \frac{1}{2} a(v, T) + \cancel{\epsilon a(v, v)} - l(v) = 0$$

$$\frac{1}{2} (a(T, v) + a(v, T)) = l(v)$$

From obs-3,  $\Rightarrow a(T, v) = l(v)$  which is same as 'V' form.

i.e.  $a$  is symmetric  $\Rightarrow a(T, v) = a(v, T)$

$$\Rightarrow a(T, v) = l(v) \rightarrow \text{"V" form}$$

Conclusion: 'M' form  $\Rightarrow$  'V' form, provided  $a$  is bilinear,  $l$  is linear &  $a$  is symmetric.

Question is: starting from 'V' form, is it possible to reach 'M' form?

Consider  $a$  as symmetric.

$$g(\epsilon) = \frac{1}{2} a(T, T) - l(T) + \epsilon [a(T, v) - l(v)] + \frac{\epsilon^2}{2} a(v, v)$$

$$\text{If 'V' form is true, } \epsilon [a(T, v) - l(v)] = 0$$

Then  $g(\epsilon) \geq \frac{1}{2} a(T, T) - l(T)$ , provided  $a(v, v)$  is true.

It is true when  $a$  is true definite.

$$\therefore g(\epsilon) = \frac{1}{2} a(T, T) - l(T) \text{ is the minimum of } g(\epsilon) \text{ at } \epsilon = 0.$$

for V to M, we require additional constraint re.  $a$  is true definite.

$\hookrightarrow$  'M' form

$$\int k \frac{dT}{dx} \frac{dv}{dx} dx = \int S v dx$$

sub  $\delta T$  in place of  $v$ ,

$$\int \underbrace{\left[ \frac{1}{2} k \left( \frac{dT}{dx} \right)^2 \right]}_{\pi} dx - \int S T dx = 0.$$

Essentially, we are minimizing  $\Pi$ .

Q) How to get 'D'-form from 'V' form?

$$\text{'V' form: } \int \underbrace{k \frac{dT}{dx}}_1 \underbrace{\frac{d(\delta T)}{dx}}_2 dx = \int S \delta T dx.$$

Integrate by parts:

$$\left[ k \frac{dT}{dx} \delta T \right]_1^2 - \int \frac{d}{dx} \left( k \frac{dT}{dx} \right) \delta T dx = \int S \delta T dx.$$

$$\int \left[ \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S \right] \delta T dx = 0$$

$$\Rightarrow \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0 \rightarrow \text{'D' form.}$$

So far we took BC as  $T$  is specified. we can do the same with  $\frac{dT}{dx}$  specified instead.

• Boundary conditions in the variational formulation:

$T$  specified: variable for which variation appears in the boundary terms

B/c  $v$  is variation of  $T$ . (primary variable)

$k \frac{dT}{dx}$  specified  $\rightarrow$  coeff. of variation in the boundary term (secondary variable)

Specifying the primary variable at boundary is called as essential BC.

Specifying the secondary variable at boundary is called as natural BC. Natural BC that forms automatically appears in the eqn (i.e. of heat flux).

# Lecture-9: Example of Variational formulation and introduction to Weighted Residual Method.

Eg:  $\frac{d^2}{dx^2} \left[ a(x) \frac{d^2 y}{dx^2} \right] + b(x) = 0.$

Obj + cast in variational formulation:

$$\int \left[ \frac{d^2}{dx^2} \left[ a(x) \frac{d^2 y}{dx^2} \right] + b(x) \right] v dx = 0.$$

Integrate by parts:

$$\int_{x=0}^{x=L} \frac{d^2}{dx^2} \left[ a(x) \frac{d^2 y}{dx^2} \right] v dx + \int_{x=0}^{x=L} b(x) v dx = 0.$$

$$\left. \frac{d}{dx} \left[ a(x) \frac{d^2 y}{dx^2} \right] \right|_{x=0}^{x=L} - \int_{x=0}^{x=L} \frac{dv}{dx} \left[ a(x) \frac{d^2 y}{dx^2} \right] dx + \int_{x=0}^{x=L} \frac{d^2 v}{dx^2} a(x) \frac{d^2 y}{dx^2} dx + \int_{x=0}^{x=L} b(x) v dx = 0.$$

$\downarrow$   
 Primary variable  $\rightarrow y$  (E.B.C)  
 Secondary variable  $\rightarrow \frac{d}{dx} \left[ a(x) \frac{d^2 y}{dx^2} \right]$  (N.B.C)

$\downarrow$   
 Primary variable  $\rightarrow \frac{dv}{dx}$  (E.B.C)  
 Secondary variable  $\rightarrow a(x) \frac{d^2 y}{dx^2}$  (N.B.C)

Boundary terms need not be zero after applying B/Cs.

The terms that we get after BC applications can be dropped with L(v) term.

Specifying B/Cs.

Let  $y=0$  at  $x=0$ .

$$\frac{d}{dx} \left[ a(x) \frac{d^2 y}{dx^2} \right] = c_1 \text{ at } x=L.$$

$$\frac{dy}{dx} = 0 \text{ at } x=0$$

$$a(x) \frac{d^2 y}{dx^2} = c_2 \text{ at } x=L.$$



$$\Rightarrow V_L C_1 - \frac{dv}{dx} \Big|_{x=L} + \int_{x=0}^{x=L} a(x) \frac{d^2 v}{dx^2} \frac{dy}{dx^2} dx + \int_{x=0}^{x=L} b(x) v dx = 0.$$

Write this in the form  $A(y, v) = L(v)$

$$A(y, v) = \int_{x=0}^{x=L} a(x) \frac{dy}{dx^2} \frac{dv}{dx^2} dx$$

$$L(v) = - \int_{x=0}^{x=L} b(x) v dx - V_L C_1 + \frac{dv}{dx} \Big|_{x=L}$$

$A(y, v) = L(v)$  is the required variational formulation.

Approximate solutions of differential equations:

Weighted residual approach:

: gets a clue from the variational form.

$\int(\cdot) v = 0$  form.  $v$  is an arbitrarily small variation.

Say we wish to solve  $\frac{d^2 y}{dx^2} = 0$ .

Call linear operator  $L(y) = 0 \rightarrow L = \frac{d^2}{dx^2}$ .

$v$  till now is an abstract variational parameter. We try to make it non-abstract by looking at the possibilities of functions we can use in place of  $v$ .

In the problem  $\frac{d^2 y}{dx^2} = 0$ , to convert it into an algebraic eq<sup>n</sup> (b/c algebraic are easier to solve), we replace  $y$  with an approximate  $y_{\text{approx}}$  polynomial & solve for it.

But  $\frac{d^2}{dx^2}(y_{\text{approx}}) \neq 0$ , in general.

Then,  $L(y) - L(y_{\text{approx}}) = R$ .  $R \rightarrow \text{residual}$

Our objective is to minimize  $R$  in an integral sense over the domain.

$$\int_{\Omega} R w \, d\Omega = 0$$

Try to minimize the error or the residual in a weighted integral sense.

$\rightarrow y^*$  ( $y_{\text{approx}}$ )

[trial function]

$\rightarrow w$

[weighting function]

Note: These functions need not be as rigorous as the variation formulations. Restrictions like  $a(\cdot, \cdot)$  symmetry & positive definiteness not needed.

### Lecture 10: Weighted Residual Method.

Say, governing differential eq?  $L(y) = 0$ .

Substituting  $y_{\text{approx}}$ ,  $L(y_{\text{approx}}) \neq 0$ .

$$L(y_{\text{approx}}) = R.$$

$$\int_{\Omega} R w \, d\Omega = 0.$$

In 1D problem, say  $y = f(x)$ ,

$$\int_{\Omega} R w \, dx = 0. \quad \text{try to make sure } y_{\text{approx}} \text{ is appropriate to minimise the } R.$$

Trial function  $\rightarrow y_{\text{approx}}$ ,

- polynomial — most convenient form.
- Should satisfy the essential BC (key requirements)
- should be continuous
- derivatives of trial function must be square integrable.  
 $\int \left( \frac{dy_{\text{approx}}}{dx} \right)^2 dx < \infty$  — shows integral is not unbounded.

+  $H^1(\Omega)$ : 1<sup>st</sup> derivative is square integrable.

Requirements for weighting function:  $\rightarrow w$

- should satisfy homogeneous part of the EBC.

{ Say if  $y=5$  is EBC, then  $w=0$  is the homogeneous part }

Why? B/c if  $y$  is specified, then variation is  $y=0$ .  $w$  has similar meaning as that of variation in  $y$ .  $\therefore w=0$  at Boundary

- should be continuous.

Some specific examples:

Prob: 1D, steady state heat transfer with uniform Thermal conductivity  $k$ , Source  $S$ .

Governing DE:  $k \frac{d^2 T}{dx^2} + S = 0$ .

$$\frac{d^2 T}{dx^2} + \left( \frac{S}{k} \right) = 0. \quad \begin{matrix} 100, \text{ say} \\ S \end{matrix}$$

BCs:  $x=0, T=0$   
 $x=10, T=0$ .

$\rightarrow \frac{d^2 y}{dx^2} + 100 = 0$	At $x=0, y=0$ $x=10, y=0$
--	------------------------------

Obj: find approx. sol<sup>n</sup>.

Ex-1 Least squared method

$y \rightarrow y_{\text{approx}}$

$$\frac{d^2 y_{\text{approx}}}{dx^2} + 100 = R$$

Interested in minimizing  $R^2$ .

$\rightarrow \int R^2 dx \rightarrow \text{minimized}$   
 $\equiv \text{sum of square of errors.}$

$y_{\text{approx}} \rightarrow 2^{\text{nd}}$  order polynomial with 1 parameter.

$\left\{ \begin{array}{l} \int R dx \text{ --- not correct as } \\ \text{some errors can be +ve \& } \\ \text{-ve \& sum to 0, can be} \\ \text{misleading} \\ \int |R| dx \text{ --- not used b/c of} \\ \text{tedious algebra.} \end{array} \right\}$

General form:  $a x (10-x) = y_{\text{approx}}$

→ polynomial, 2<sup>nd</sup> order

Find out  $a$  s.t.  $\int R^2 dx$  is minimized.

$$y_{\text{approx}} = a x (10-x) \\ = 10 a x - a x^2$$

$$\frac{dy_{\text{approx}}}{dx} = 10a - 2ax$$

$$\frac{d^2 y_{\text{approx}}}{dx^2} = -2a$$

$$R = y - y_{\text{approx}} \\ = -2a + 100$$

$$\frac{\partial}{\partial a} \left( \int R^2 dx \right) = 0$$

$$\Rightarrow \int 2R \frac{\partial R}{\partial a} dx = 0$$

$$\Rightarrow \int R \frac{\partial R}{\partial a} dx = 0$$

So  $\frac{\partial R}{\partial a}$  is the weighting function.

$$\frac{\partial R}{\partial a} = \frac{\partial}{\partial a} (-2a + 100) = -2$$

$$\int_0^{10} (-2a + 100) (-2) dx = 0$$

$$\Rightarrow -2a + 100 = 0$$

$$\text{or } a = 50$$

# Lecture 11: Point Collocation method, Galerkin's method & The 'M' form.

## Ex-2. Point Collocation method:

$$w = \delta(x - x_i) \quad \delta: \text{Dirac-Delta function.}$$

Idea: you try to satisfy the value of the function at chosen points  $x_i$

$$\int_x R w dx = 0.$$

Keep trial function same:  $ax(10-x)$

$$R_{x=x_i} = 0 \quad \text{consider only 1 collocation point, say } x=5.$$

$$R_{x=5} = 0$$

$$-2a + 100 = 0 \Rightarrow a = 50$$

## Ex-3 Galerkin's method:

It considers the weighting function as the trial function.

$$w = x(10-x) \quad \text{Putting } a \text{ doesn't matter as it's just a const & it will go away in the expression } \int R w dx = 0.$$

$$\int R w dx = 0.$$

$$\int_0^{10} (-2a + 100) x(10-x) dx = 0$$

$$\Rightarrow -2a + 100 = 0$$

$$\text{or } a = 50$$

Going through routes of the 'M' form:

$$\int_0^{100} \left( \frac{dy}{dx^2} + 100 \right) v dx = 0$$

$$\left[ v \frac{dy}{dx} \right]_0^{100} - \int_0^{100} \frac{dv}{dx} \frac{dy}{dx} dx + \int_0^{100} 100 v dx = 0$$

$$\text{PV: } y$$

$$\text{SV: } \frac{dy}{dx}$$

$$y=0 \text{ at } x=0$$

$$(\text{given})$$

$$\Rightarrow \int_0^{100} \frac{dv}{dx} \frac{dy}{dx} dx = \int_0^{100} 100 v dx$$

$$a(y, v) = \ell(v)$$

$$a(y, y) > 0. \text{ here } \Rightarrow \text{'M' form exists here!}$$

N form:

$$\begin{aligned} \text{Minimize } \Pi &= \frac{1}{2} a(y, y) - \text{Lap.} \\ &= \frac{1}{2} \int_0^{10} \left( \frac{dy}{dx} \right)^2 dx - \int_0^{10} 100y dx \end{aligned}$$

Use  $y_{\text{approx}} \rightarrow$  in place of  $y$  & minimize  $\Pi$ ,   
 & can have reduced requirement for continuity.

Here we don't require any weighting fn. Only final function is necessary.

This convenience comes at the cost of additional requirement of positive definiteness of  $a(y, y)$ .

$$\frac{\partial \Pi}{\partial a} = 0 \quad \text{Use trial fn } y_{\text{approx}} = ax(10-x)$$

— Rayleigh — Ritz method.

$$\frac{dy_{\text{approx}}}{dx} = 10a - 2ax$$

$$\begin{aligned} \Pi &= \frac{1}{2} \int_0^{10} \left( \frac{dy_{\text{approx}}}{dx} \right)^2 dx - \int_0^{10} 100y_{\text{approx}} dx \\ &= \frac{1}{2} \int_0^{10} (10a - 2ax)^2 dx - \int_0^{10} 100(10a - 2ax) dx \end{aligned}$$

$$\begin{aligned} &= \frac{1}{2} \int_0^{10} (100a^2 - 40a^2x + 4a^2x^2) dx - \int_0^{10} (1000a - 200ax) dx \\ &= \frac{1}{2} \left[ 100a^2x - 20a^2x^2 + \frac{4a^2x^3}{3} \right]_0^{10} - \left[ 1000ax - 100ax^2 \right]_0^{10} \\ &= \frac{1}{2} (1000a^2 - 2000a^2 + \frac{4000a^2}{3}) - (10000a - 10000a) = 0 \end{aligned}$$

$$\frac{\partial \Pi}{\partial a} = 0 \Rightarrow a = 100 \int_0^{10} \dots$$

$$\frac{dy}{dx} = -100x + C_1$$

$$y = -\frac{100x^2}{2} + C_1x + C_2$$

$$C_2 = 0 \rightarrow 0 = -100 \times \frac{1000}{2} + 100C_1 \rightarrow C_1 = 500$$

$$y = 50(10x - x^2) = 50x(10-x)$$

✓ exact!

How to reduce calculations associated with higher order polynomials?

Divide the domain into smaller & smaller subdomains & use lower order polynomials in those subdomains. B/c even though across the entire domain our  $f_n$  may be quite complex, in smaller subdomains, such functions may be simpler. Now the solution will be fitted b/w discrete points rather than over the entire domain  $\rightarrow$  basic idea behind discretization.

Discretization: here we lose the continuous nature of the domain.

- Divide the domain into a number of discrete subdomains.  
(element, control volume, ...)
- Each subdomain is represented by a discrete set of points.  
(grid points, nodes, ...)
- Objective is to convert the governing DE into a system of algebraic equations valid at each of these discrete points.

### Lecture 12: Finite Element Method (FEM) of discretization

Discretization principles:

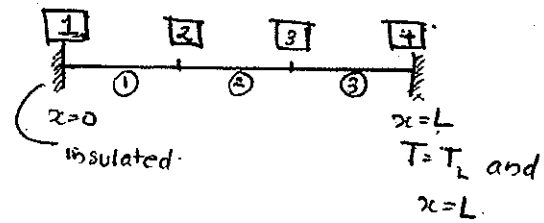
- $\rightarrow$  Divide the domain into a number of discrete subdomains, each subdomain being ~~characterized~~ represented by a number of discrete points.
  - $\rightarrow$  Derive algebraic equations from the governing diff. eq<sup>ns</sup>, valid at these discrete points.
  - $\rightarrow$  Solve the system of algebraic equations to obtain values of the dependent variables at the discrete points.
- 
- $\rightarrow$  Broad steps in overall analysis:
- $\rightarrow$  Pre-processing: set-up geometry, discretized eq<sup>ns</sup>, input data (property data), initial cond, BCs.
  - $\rightarrow$  Solution: algebraic eq<sup>ns</sup>
  - $\rightarrow$  Post-processing: graphical representation of the obtained results.

# Finite Element Method (FEM)

$$F \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0$$

1, 2, 3, 4 → nodes

①, ②, ③ → elements.



Prepare node-element connectivity chart:

Element	node i	node j
1	1 (,)	2 (,)
2	2 (,)	3 (,)
3	3 (,)	4 (,)

Consider any isolated element



Writing an algebraic eq<sup>n</sup> corresponding to the governing diff eq<sup>n</sup>.

$$\int_{x=0}^{x=L} \left[ \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S \right] w dx = 0$$

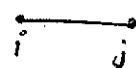
Integrate by parts:

$$w k \frac{dT}{dx} \Big|_0^L - \int_0^L \frac{dw}{dx} k \frac{dT}{dx} dx + \int_0^L S w dx = 0.$$

for the isolated element, this would become.

$$w k \frac{dT}{dx} \Big|_{x_i}^{x_j} - \int_{x_i}^{x_j} k \frac{dw}{dx} \frac{dT}{dx} dx + \int_{x_i}^{x_j} S w dx = 0.$$

If it were not a two-noded element, we'd require a higher order polynomial to approximate  $T$ .



$$T = a_0 + a_1 x$$

(trial function)

$$\text{At } x = x_i, T = T_i$$

$$x = x_j, T = T_j$$

$$T_i = a_0 + a_1 x_i$$

$$T_j = a_0 + a_1 x_j$$

$$a_1 = \frac{T_j - T_i}{x_j - x_i}$$

$$a_0 = T_i - a_1 x_i = T_i - \left( \frac{T_j - T_i}{x_j - x_i} \right) x_i$$

We choose trial function for each element, not for the whole domain.

We finally get piecewise continuous function for the whole domain



$$a_0 = T_i - \left( \frac{T_j - T_i}{x_j - x_i} \right) x_i \quad \left| \quad a_1 = \frac{T_j - T_i}{x_j - x_i} \right.$$

$$= \frac{T_i x_j - T_j x_i}{x_j - x_i}$$

$$T = \left( \frac{T_i x_j - T_j x_i}{x_j - x_i} \right) + \left( \frac{T_j - T_i}{x_j - x_i} \right) x$$

$$= \underbrace{\left( \frac{x_j - x}{x_j - x_i} \right)}_{N_i} T_i + \underbrace{\left( \frac{x - x_i}{x_j - x_i} \right)}_{N_j} T_j$$

$$T = N_i T_i + N_j T_j$$

Interpolation functions / shape functions.

Property of shape functions:

$N_i = 1$  at node  $i$ ,  $= 0$  at node  $j$

$N_j = 0$  at node  $j$ ,  $= 1$  at node  $i$ .

Writing in matrix form

$$T = \underbrace{[N_i \quad N_j]}_{[N]} \underbrace{\begin{bmatrix} T_i \\ T_j \end{bmatrix}}_{[T]}$$

$$w = [N]^T \underbrace{\begin{bmatrix} w_i \\ w_j \end{bmatrix}}_{[w]} \text{ is Galerkin form.}$$

$$\Rightarrow w^T = [w]^T [N]^T$$

$$\therefore w_k \frac{dT}{dx} \Big|_{x_i}^{x_j} - \int_{x_i}^{x_j} k \frac{dw}{dx} \frac{dT}{dx} dx + \int_{x_i}^{x_j} S w dx = 0.$$

$$\rightarrow [w]^T [N]^T q^u \Big|_i^j - [w]^T \int_{x_i}^{x_j} \left[ \frac{dN}{dx} \right]^T k \left[ \frac{dN}{dx} \right] \begin{bmatrix} T_i \\ T_j \end{bmatrix} dx + \int_{x_i}^{x_j} S [N]^T dx = 0.$$

# Lecture 13: Finite Element Method of Discretization (contd.)

Since  $[w]^T$  is arbitrary,

$$\underbrace{\left[ -[N]^T q'' \right]_i}_{\text{Term-1}} - \underbrace{\left[ \int_i^j \left[ \frac{dN}{dx} \right]^T_k \left[ \frac{dN}{dx} \right] [T] \right]}_{\text{Term-2}} + \int_i^j S [N]^T dx = 0$$

$$\text{Term-1} = \left[ - \begin{bmatrix} N_i \\ N_j \end{bmatrix} q'' \right]_i$$

$$= - \begin{bmatrix} 0 \\ q''_i \end{bmatrix} + \begin{bmatrix} q''_i \\ 0 \end{bmatrix} = \begin{bmatrix} +q''_i \\ -q''_j \end{bmatrix}$$

$$\text{Term-2} = \int_i^j \left[ \frac{dN_i}{dx} \quad \frac{dN_j}{dx} \right]^T_k \begin{bmatrix} \frac{dN_i}{dx} & \frac{dN_j}{dx} \end{bmatrix} dx$$

$$\begin{aligned} \frac{dN_i}{dx} &= \frac{-1}{\underbrace{x_j - x_i}_{le}} \\ &= \frac{-1}{le} \end{aligned} \quad \begin{aligned} \frac{dN_j}{dx} &= \frac{1}{\underbrace{x_j - x_i}_{le}} \\ &= \frac{+1}{le} \end{aligned}$$

$$\begin{aligned} \Rightarrow \text{Term-2} &= \int_i^j \frac{k}{le^2} \begin{bmatrix} -1 \\ 1 \end{bmatrix} \begin{bmatrix} -1 & 1 \end{bmatrix} dx \\ &= \int_i^j \frac{k}{le^2} \begin{bmatrix} +1 & -1 \\ -1 & +1 \end{bmatrix} dx \\ &= \frac{k}{le^2} (x_j - x_i) \\ &= \frac{k}{le} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \\ &= \end{aligned}$$

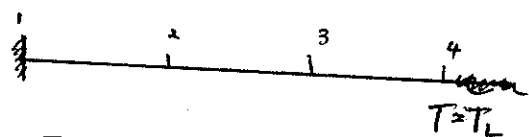
$$\text{Term-3: } \int_i^j S [N]^T dx = S \int_i^j \begin{bmatrix} \frac{x_j - x}{le} \\ \frac{x - x_i}{le} \end{bmatrix} dx = \begin{bmatrix} \frac{S le}{2} \\ \frac{S le}{2} \end{bmatrix}$$

entire effect of the element is manifested by the behavior of nodes.  
Total is shared equally b/w nodes.

Assembling all terms together,

$$(\text{Term-2}) \cdot [T] = (\text{Term-1}) + (\text{Term-3})$$

$$\Rightarrow \frac{k}{2e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix} \begin{bmatrix} T_i \\ T_j \end{bmatrix} = \begin{bmatrix} q_i'' \\ -q_j'' \end{bmatrix} + S_{le} \begin{bmatrix} 1/2 \\ 1/2 \end{bmatrix}$$



This is of the form:  $\rightarrow$  Acts like a force term.

$[K][T] = [F] \rightarrow$  Similar to spring mass system.  
 as if behaves like displacement  
 as if like the stiffness of the system.

for 2 nodes we get 2x2 form. Similarly for 4 nodes we get 4x4 form.

This part of coeff matrix activated for 1st element

1	2	3	4
1	1		
2	-1	1	
3		-1	1
4			-1

This part for the third element

This part activated for 2nd element.

$$\begin{bmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{bmatrix} = \begin{bmatrix} q_1'' \\ -q_2'' + q_3'' \\ -q_3'' + q_4'' \\ -q_4'' \end{bmatrix} + S_{le} \begin{bmatrix} 1/2 \\ 1/2 + 1/2 \\ 1/2 + 1/2 \\ 1/2 \end{bmatrix}$$

Important assumptions: ① Thermal conductivity  $k$  as constant.

② All elements have same length. It need not be the case in reality.

Final form:

$$\begin{bmatrix} 1 & -1 & 0 & 0 \\ -1 & 2 & -1 & 0 \\ 0 & -1 & 2 & -1 \\ 0 & 0 & -1 & 1 \end{bmatrix} \begin{bmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \end{bmatrix} = \begin{bmatrix} q_1'' \\ -q_2'' + q_3'' \\ -q_3'' + q_4'' \\ q_4'' \end{bmatrix} + S_{le} \begin{bmatrix} 1/2 \\ 1 \\ 1 \\ 1/2 \end{bmatrix}$$

Complete the problem by imposing BCs.

At  $x=0$ ,  $q''=0$  (heat flux is zero as insulated)

At  $x=L$ ,  $T_4$  is specified.

Even if  $T_4$  is specified, computer may try to find  $T_4$  from with first BC. In that case, we require matching b/w specified  $T_4$  & computed  $T_4$ .

$$k_{41} T_1 + k_{42} T_2 + k_{43} T_3 + k_{44} T_4 = R.$$

Let  $T_{4, \text{specified}} = T_4^*$

Use small computational trick. Replace  $R$  with  $\frac{L}{k_{44}} T_4^*$  & replace  $k_{44}$  with  $k_{44} + L$ , where  $L$  is a large number.

$$\text{Then } k_{41} T_1 + k_{42} T_2 + k_{43} T_3 + (k_{44} + L) T_4 = L T_4^*.$$

$$T_4 = \frac{-k_{41} T_1}{k_{44} + L} - \frac{k_{42} T_2}{k_{44} + L} - \frac{k_{43} T_3}{k_{44} + L} + \frac{L T_4^*}{k_{44} + L}$$

If  $L$  is very large,

$$\frac{-k_{41}}{k_{44} + L} \rightarrow 0, \quad \frac{-k_{42}}{k_{44} + L} \rightarrow 0, \quad \frac{-k_{43}}{k_{44} + L} \rightarrow 0 \quad \&$$

$$\frac{L}{k_{44} + L} \rightarrow 1.$$

Thus we numerically get,

$$T_4 = T_4^*.$$

# Lecture 14: Finite Difference Method (FDM) of discretization.

(H/W)

(Prob-1) Consider the DE:  $\frac{d^2 u}{dx^2} + u + x = 0$ .

with BC  $u(0) = u(1) = 0$ .

Solve the above eq<sup>n</sup> using (1) Least square

(2) Point collocation

(3) Galerkin

(4) Rayleigh Ritz method.

Choose trial function  $\rightarrow u = a \sin \pi x$

(Prob-2) Consider a heat conduction problem with the following governing DE:

$$\frac{d}{dz} \left( A k \frac{dT}{dz} \right) + Q = 0, \quad A = 10 \text{ m}^2, k = 5 \text{ J/Kms},$$

$$Q = 100 \text{ J/sm}.$$

Domain  $2 \text{ cm} \leq x \leq 8 \text{ cm}$ .

BCs:  $T(x = 2 \text{ cm}) = 0^\circ \text{C}$

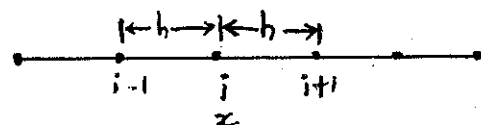
$$T''(x = 8 \text{ cm}) = 15 \text{ J/m}^2 \text{s}.$$

Obtain temperature distribution in the domain using FEM with three linear elements, and compare with the analytical solution.

• If solving a structural mechanics problem, 'M' form is essentially a statement of minimization of potential energy of that system, which governs the stability of the system at equilibrium.

• In FDM, we deal with the 'D'-form directly.

Express derivatives in terms of suitable algebraic differences by using Taylor series expansion.



Consider a 1D domain.

We represent the domain with a collection of discrete grid points.

(Don't have the concept of discrete elements here).

$$f(x+h) = f(x) + h f'(x) + \frac{h^2}{2!} f''(x) + \dots \quad \text{--- (1)}$$

$$f(x-h) = f(x) - h f'(x) + \frac{h^2}{2!} f''(x) - \dots \quad \text{--- (2)}$$

We are interested in an algebraic expression for  $f'(x)$ .  
From (1)

$$\text{Thus, } f'(x) = \frac{f(x+h) - f(x)}{h} + \frac{h}{2!} f''(x) - \dots$$

here continuous derivative is represented as discrete algebraic quantities.

$$\text{Error incurred } - \frac{h}{2!} f''(x) - \dots \rightarrow \text{truncation error (TE)} \\ \sim O(h) \quad \downarrow \text{dictated by the leading order term.}$$

From (2);

$$f'(x) = \frac{f(x) - f(x-h)}{h} + \frac{h}{2!} f''(x) + \dots \rightarrow \text{truncation error (TE)}$$

-(2) + (1) gives

$$f'(x) = \frac{f(x+h) - f(x-h)}{2h} + \frac{h^2}{3!} f'''(x) + \dots \rightarrow \text{TE} \sim O(h^2)$$

$$f'(x) = \frac{f(x+h) - f(x)}{h} + O(h) \rightarrow \text{Forward difference}$$

$$f'(x) = \frac{f(x) - f(x-h)}{h} + O(h) \rightarrow \text{Backward difference}$$

$$f'(x) = \frac{f(x+h) - f(x-h)}{2h} + O(h^2) \rightarrow \text{Central difference}$$

(1) + (2) gives

$$f(x+h) + f(x-h) = 2f(x) + \frac{2 \cdot h^2}{2!} f''(x) + \frac{2 \cdot h^4}{4!} f^{(4)}(x) + \dots$$

$$f^{(4)}(x) = \frac{f(x+h) + f(x-h) - 2f(x)}{h^2} - \frac{h^2}{12} f^{(4)}(x) - \dots \rightarrow \text{Central difference for 2nd order PDE.}$$

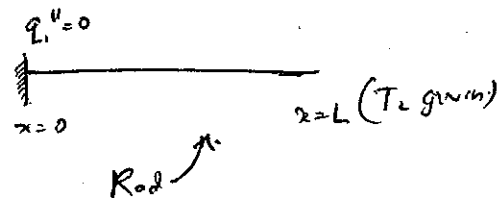
$$f''(x) = \frac{f'(x+h) - f'(x)}{h} \quad \text{Forward-difference}$$

$$= \frac{\frac{f(x+h) - f(x)}{h} - \frac{f(x) - f(x-h)}{h}}{h}$$

$$= \frac{f(x+h) + f(x-h) - 2f(x)}{h^2}$$

Ex Consider 1D, steady-state heat conduction problems.

$$\frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0$$



Assumption:  $k, S$  both constants

$$k \frac{T_{i+1} + T_{i-1} - 2T_i}{h^2} + S = 0$$

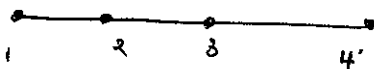
[What determines the whether we should choose large  $h$  or small  $h$ ?

$$T_{i+1} + T_{i-1} - 2T_i + \frac{Sh^2}{k} = 0.$$

↳ algebraic eqn.

It depends upon the temperature gradients in the domain. It may so happen that at some regions, the gradients may be steep. In such regions, we use finer value of  $h$ . At other regions, where the gradient is less, we may opt for a larger  $h$ . It all depends on the physics of the problem & our understanding about it.]

Consider 4 grid pts:



The above eqn is valid for internal grid pts not at the boundary. (B/c we don't have  $T_{i-1}$  @ left boundary &  $T_{i+1}$  at right boundary).

Grid 2  $\rightarrow$

$$T_3 - T_1 - 2T_2 + \frac{Sh^2}{k} = 0$$

Grid 3  $\rightarrow$

$$T_4 - T_2 - 2T_3 + \frac{Sh^2}{k} = 0$$

Grid 4  $\rightarrow$

$$BC: T_4 = T_L \text{ (given)}$$

Grid 1  $\rightarrow$

$$BC: q'' = 0$$

$$\Rightarrow k \frac{dT}{dx} \Big|_1 = 0$$

$$So: \frac{T_2 - T_1}{h} = 0 \text{ FD formula}$$

$$\Rightarrow \cancel{T_2} = \cancel{T_1} \quad T_1 = T_2$$

(Assign the value at the boundary with the interior value)



## Lecture 15: Well Posed Boundary Value Problem.

(21)

• Well posed <sup>BVP</sup> problem requirements:

→ Existence of solution

→ Uniqueness of sol<sup>n</sup>.

→ A small perturbation in BC shouldn't lead to large changes in the solution.

(This is important bc when a perturbation may be unwittingly

introduced through round-off errors; B/c of it, it may

lead to large change in sol<sup>n</sup> → oversensitive BC).

• Possible types of BCs: (2<sup>nd</sup> order problems)

Dirichlet BC:

1. Value of the dependent variable is specified → EBC.

2. Neumann BC:

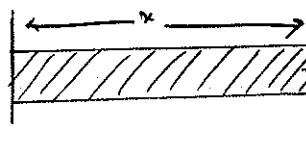
Value of the gradient of the dependent variable is specified

- NBC.

3. Mixed BC:

Value of the dependent variable is expressed as a function of the grad.

Eg: convective heat transfer BC.



$T_0 < T_L$

$$-k \frac{dT}{dx} \bigg|_{x=L} = h(T_L - T_0)$$

This expression can be used even in

unsteady case. (replace  $\frac{dT}{dx}$  with  $\frac{\partial T}{\partial x}$ )

This <sub>BC says</sub> Whatever is the heat flux coming at  $x=L$  via conduction, the same is the heat flux

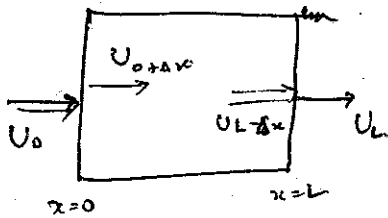
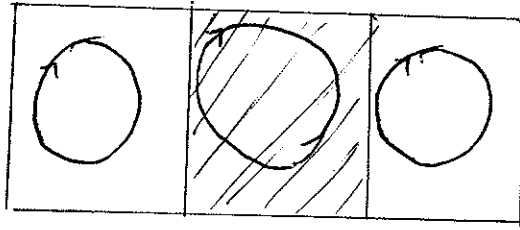
leaving  $x=L$  via convection. That remains true even if unsteady.

Here temperature at  $x=L$  is expressed as a

function of gradient of the temperature  $(-k \frac{dT}{dx})$ .

Hence it's a mixed BC.

#### 4. Periodic BC:



$$U_L = U_{\Delta x}$$

$$U_0 = U_{L-\Delta x}$$

$$\text{periodicity} = L - \Delta x$$

Here again we see boundary <sup>form</sup> being represented in terms of interior terms, not the opposite.

Q) Is any condition specified at the boundary, a boundary condition?

Ans) Consider a simple 1D, steady-state, heat conduction problem,

$$q_0'' = 1 \text{ W/m}^2$$

$$q_L'' = 1 \text{ W/m}^2$$

$$S = 0, k = \text{const.}$$

→ Neumann BC at both the boundaries.

$$\frac{d^2 T}{dx^2} = 0$$

$$\frac{dT}{dx} = C_1$$

$$T = C_1 x + C_2$$

Say,  $k = 1 \text{ W/mK}$ .

$$x=0, -k \frac{dT}{dx} = 1$$

$$\Rightarrow \frac{dT}{dx} = -1 \Rightarrow C_1 = -1$$

Similarly at  $x=L$ ,

$$-k \frac{dT}{dx} = 1 \Rightarrow \frac{dT}{dx} = -1 \Rightarrow C_1 = -1$$

Basically

$$T = -x + C_2$$

Cannot determine  $C_2$ .

→ Violates requirement for uniqueness of sol?

Plotting in  $T-x$  plane gives all sol<sup>ns</sup> to be parallel straight lines; no 1 sol?

→ not legitimate BCs.

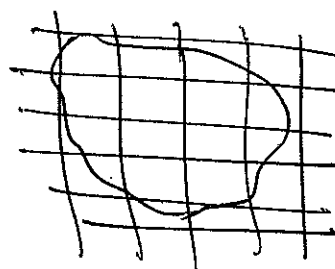
→ not well posed problem

# Lecture 16: Finite Volume Method (FVM) of Discretization.

FDM  $\rightarrow$  simple

Issues: ① Complex geometry.

One has to tediously create ~~a~~ handle the boundary while using cartesian grid.



Taylor series expansion:

$$f(x+h) = f(x) + h f'(x) + \frac{h^2}{2} f''(x) + \dots$$

truncate from here

② But what if  $f''(x)$  is large enough to be non-negligible?  
(B/c  $h$  cannot be tending to zero, it is still finite).

Ex:  $f(x) = e^x \rightarrow f''(x) \approx \text{non-negligible}$

$\Rightarrow$  significant errors while truncating.

$\rightarrow$  limitations of Taylor series based method.

Q) What do we expect from the discretization?

(Ans)  $\rightarrow$  ① Conservativeness  $\rightarrow$  Discretized version of conservation eq<sup>ns</sup> should exhibit that conservative nature.

$\rightarrow$  ② Boundedness

$\rightarrow$  ③ Transportiveness

Finite Difference

• ~~FD~~ discretization may not satisfy conservativeness b/c we haven't explicitly enforced that condition while expanding out the function in Taylor series form and ~~not~~ truncation. Conservativeness is not built while coming up with FDM.

- Boundedness: Say we have a rod:



We are interested in the temperature distribution in the rod.

We expect the values in the rod to lie b/w  $0$  &  $100^\circ\text{C}$ . The discretization should also ensure the physical nature of boundedness in the problem.

This boundedness is also not ensured while coming up with FDM.

- Transportiveness: If there is a predominant directionality of the flow involved in the problem, the transport properties should also have a predominant transport direction based on the flow direction.

(For high Re flow, for eg. enthalpy should predominantly be transported downstream).

---

FEM — relatively more complicated, <sup>when</sup> compared to FDM.

- Strong mathematical basis in error minimization.

- not intuitive physically: all the V-formulation & M-formulation needs to have some physical meaning, which can be difficult to come up with.

For fluid flow, mass flow, conservation is important; while for structural mechanics, minimization of potential energy is important.

- Can handle complex geometries.

## Finite Volume Method (FVM):

- Step-1: Divide the domain into a number of <sup>finite size</sup> sub-domains (control volumes). Each sub-domain is represented by a finite no. of grid pts. (control volumes)
- Step-2: Integrate the governing differential equation over each subdomain.
- Step-3: Consider a profile assumption for the dependent variable for evaluating the above integrals, to express the result in terms of algebraic quantities at the grid points.

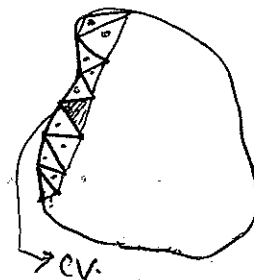
### Lecture 17: Illustrative Examples of Finite Volume Method

Ex Steady state convection diffusion with  $S=0$ .

General transport equation:

$$\frac{\partial (\rho \phi)}{\partial t} + \nabla \cdot (\rho \vec{V} \phi) = \nabla \cdot (\Gamma \nabla \phi) + S$$

$$\int_V \nabla \cdot \underbrace{(\rho \vec{V} \phi - \Gamma \nabla \phi)}_{\vec{J} \cdot \hat{n}} dV = 0$$



$\rho \vec{V} \phi \rightarrow$  advection flux  
 $\Gamma \nabla \phi \rightarrow$  diffusion flux.

Using divergence theorem,

$$\int_{C.S} \vec{J} \cdot \hat{n} dS = 0.$$

Grid pts located at the center of each CV.

Q) Why the name 'finite' volume method?

Ans) While deriving the transport equations, we considered an infinitesimally small control element. Then we were integrating back to apply it to a finite volume. Hence the name finite volume method.

Key step  $\rightarrow$  step-2.

Similarity:

It can be thought of having  $\omega$  in

$$\int_{CV} \nabla \cdot (\rho \nabla \phi - \Gamma \nabla \phi) \omega dV = 1$$

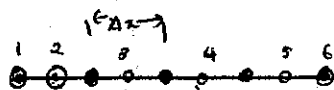
Galerkin method with  $\omega$ .

- Requirement of conservation when applied to each control-volume will satisfy the conservation across the whole domain.
- Unlike FDM, FVM takes into account conservation requirement implicitly.
- Profile assumption method is used just for step-3. Afterwards in post analysis, we no longer require profile assumption. Hence we have more flexibility in choosing profile assumption as compared to FEM.

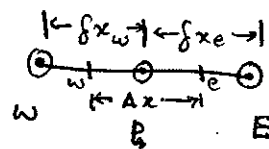
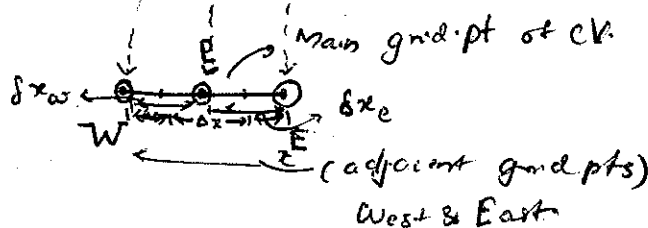
Illustration: 1D steady state heat conduction, with  $S = \text{const.}$

$$\frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0$$

Divide 1-D domain into CVs.



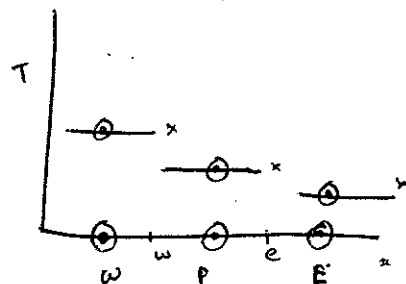
In addition to centroids of CVs, we also consider grid pts at the boundary (just so we can impose BCs at those pts).



$$\int_w^e \frac{d}{dx} \left( k \frac{dT}{dx} \right) dx + \int_w^e S dx = 0$$

$$\left[ k \frac{dT}{dx} \right]_e - \left[ k \frac{dT}{dx} \right]_w + S \Delta x = 0$$

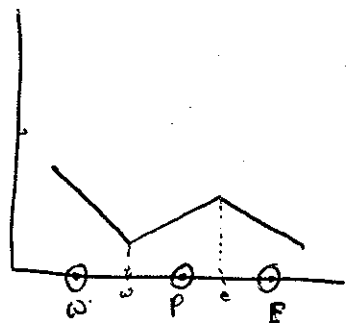
Choosing a profile assumption.



Can consider piecewise continuous function for each CV.

But not here. B/c we need  $\frac{dT}{dx}$  here. Since  $T = \text{const}$  for each CV, this is not a valid profile assumption. (There discontinuity is not a problem).

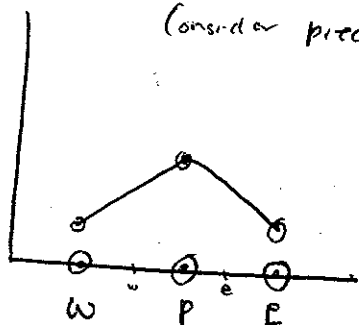
Piecewise linear non-const profile.



Problems with this case:

$k \frac{dT}{dx}$  is not continuous, i.e. physically it means heat flux is not continuous, which is incorrect.  
 $\Rightarrow$  not an acceptable profile.

Consider piecewise linear profile b/w the grid pts.



This will work b/c  $\frac{dT}{dx}$  is not evaluated @ p, but rather at the faces of each control volume w.e.  
 $\Rightarrow$  valid profile assumption

Profile assumption: Piecewise linear & b/w grid pts.

$$\Rightarrow k \left. \frac{dT}{dx} \right|_e - k \left. \frac{dT}{dx} \right|_w + S \Delta x = 0$$

$$\rightarrow k_e \frac{T_E - T_P}{\Delta x_e} - k_w \frac{T_P - T_W}{\Delta x_w} + S \Delta x = 0$$

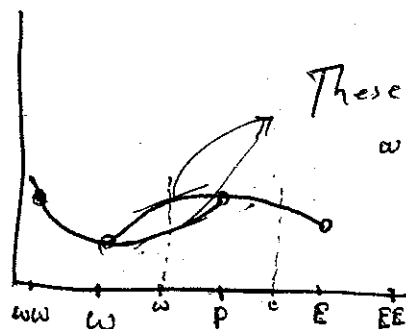
$$\Rightarrow -\left(\frac{k_e}{\Delta x_e} + \frac{k_w}{\Delta x_w}\right) T_P + \frac{k_e}{\Delta x_e} T_E + \frac{k_w}{\Delta x_w} T_W + S \Delta x = 0$$

$$a_P T_P = a_E T_E + a_W T_W + b$$

$$\text{where } a_E = \frac{k_e}{\Delta x_e}, \quad a_W = \frac{k_w}{\Delta x_w}, \quad a_P = a_E + a_W, \quad b = S \Delta x$$

$\frac{k}{\Delta x} \rightarrow$  conductance (physical meaning)

Would it have been better to consider a higher order interpolation function?



These slopes need not necessarily be equal. It happens only when  $w$  lies exactly on b/w  $W$  and  $P$ . From Rolle's Thm., a chord b/w two pts has a slope that will be attained by the curve b/w those two pts at some pt b/w them, given the curve is differentiable at all points. From MVT, this will lie at the midpoint of the curve. ~~It only happens when  $w$  is~~

So higher order interpolate functions need not necessarily give more accurate results in FVM (counter-intuitive).

### Lecture-18: Illustrative Examples of Finite Volume Method (Contd)

1-D steady state heat conduction equation.

$$\frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0$$

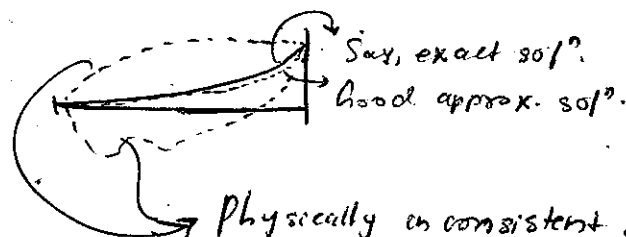
If  $k, S \rightarrow \text{fn}(T) \rightarrow$  non linear eq<sup>n</sup>  $\rightarrow$  solved via iterative process

Requirements of discretization:

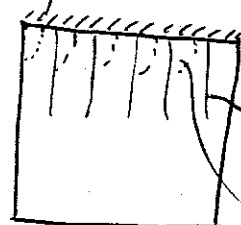
- (1) Physical consistency
- (2) Overall balance.

Eg:

$T=0^\circ\text{C}$   $T=100^\circ\text{C}$  Assume  $S=0$   
 $k = \text{const.}$



insulated



iso Thermal lines in 2d domain

These isotherms are physically consistent (normal to insulated surface)

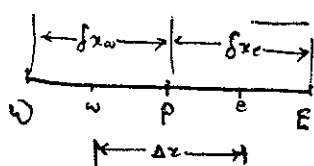
These isotherms are physically inconsistent.

for eg.,  
(Boundedness invalidated)



Variable  $S : \rightarrow$  let  $S(T)$

$E_x = S$  is a linear fn of  $T$ .  
i.e.  $S = a + bT$ .



$$k \frac{dT}{dx} \Big|_e - k \frac{dT}{dx} \Big|_w + \int_w^e (a + bT) dx = 0.$$

→ piecewise const  $T$  profile within each CV.

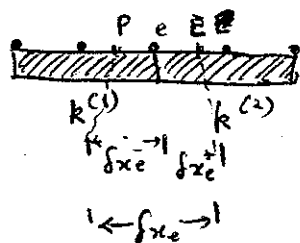
Goodie assumption: piecewise linear  $T$  b/w grid pts.

$$\Rightarrow k_e \frac{(T_E - T_P)}{\Delta x_e} - k_w \frac{(T_P - T_W)}{\Delta x_w} + (a + bT_P) \Delta x = 0.$$

$$a_P T_P = a_E T_E + a_W T_W + b.$$

$$\text{where } a_E = \frac{k_e}{\Delta x_e}, \quad a_W = \frac{k_w}{\Delta x_w}, \quad a_E + a_W - b \Delta x = a_P, \quad b = a \Delta x.$$

Composite material with position dependent  $k$ :



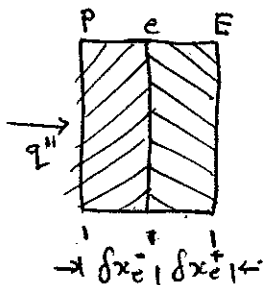
→ 4 CVs

→ Requirement: Thermal conductivity at the interface.

Since interface shared by both materials, an equivalent thermal conductivity needs to be described.

If  $k_P$  &  $k_E$  is known, intuitive:  $k_e = \frac{k_E + k_P}{2}$  → linear interpolation

Physical assessment of this  $k_e$  needs to be performed: → AN formulation.



$$q''_{\text{left}} = \left( \frac{T_E - T_P}{\Delta x_e^-} \right) k_P$$

$$q''_{\text{left}} = q''_{\text{right}}$$

$$= \frac{T_E - T_P}{\left( -\frac{\Delta x_e^-}{k_P} \right)}$$

$$q''_{\text{right}} = \frac{T_E - T_P}{\Delta x_e^+} (-k_P)$$

$$= \frac{T_E - T_P}{\left( -\frac{\Delta x_e^+}{k_P} \right)}$$

$$\frac{T_P - T_E}{\frac{\Delta x_e^-}{k_P}} = \frac{T_E - T_P}{\frac{\Delta x_e^+}{k_E}} = \frac{T_P - T_E}{\frac{\Delta x_e}{k_e}} = \frac{T_P - T_E}{\frac{\Delta x_e^-}{k_P} + \frac{\Delta x_e^+}{k_E}}$$

→ where  $k_e$  is the equivalent thermal conductivity.

→ If  $\int x_e^- = \int x_e^+$ ,

$$\frac{2}{k_e} = \frac{1}{k_p} + \frac{1}{k_E}$$

→ H.M. formulation

$$\text{or } k_e = \frac{2k_p k_E}{k_p + k_E} = \frac{2}{\frac{1}{k_p} + \frac{1}{k_E}}$$

• Limiting cases:

$$k_p \gg k_E$$

$$\text{Then AM} \rightarrow k_e \approx \frac{k_p}{2}$$

$$\text{HM} \rightarrow k_e \approx \frac{k_E}{2}$$

For interfacial conductivity variation, HM formulation is physically much more appealing. why?

Say  $k_E = 0$ . That is equivalent to highly insulated material-E. In that case  $k_e$  should also be 0 at the interface. This is reflected in HM, while not at all in AM formulation.

### Lecture 19: Basic rules of finite Volume Discretization.

4 Basic rules (of 1D steady state diffusion type problem):

(1) Physical consistency of fluxes at control volume faces.

→ Profile should be chosen in such a way that there is no discontinuity of flux at control volume faces.

(2) All coefficients in the discretized equation must be of the same sign.

Eg: say,  $b=0$   
 $10T_p = 15T_E - 5T_w$

Say  $T_E = 10$

$T_w = 100$

$$\therefore T_p = \frac{15 \times 10 - 5 \times 100}{10}$$

$$= -35$$

$T_p$  is not bounded b/w  $T_E$  &  $T_w$ .

⇒ Physically inconsistent.

This inconsistency has originated b/c of the -ve sign in the discretization eqn ( $10T_p = 15T_E - 5T_w$ )

$$\text{If, } 10 T_p = 5 T_E + 5 T_w$$

$$T_p = \frac{5 \times 10 + 5 \times 100}{10}$$

$$= 55$$

$$\text{Hence } T_E \leq T_p \leq T_w$$

Hence consistent.

(Same sign coefficients)

By sign convention, we will consider that sign to be +ve.

(3) If the source term is linearized as  $S = S_c + S_p T_p$

Then  $S_p$  must be -ve  $\rightarrow$  Extension of consideration - #2.

(4) If a linear governing DE is discretized, its discretized version should satisfy the following requirement:

If  $T$  is a sol<sup>n</sup>, then  $T+c$  is also a sol<sup>n</sup>.

We are interested to see

$$a_p (T_p + c) \stackrel{?}{=} a_E (T_E + c) + a_w (T_w + c) \dots \textcircled{1}$$

We already know

$$a_p T_p = a_E T_E + a_w T_w \dots \textcircled{2}$$

$\textcircled{1} - \textcircled{2}$

$$\Rightarrow a_p c = (a_E + a_w) c$$

$$\text{or } a_p = a_E + a_w$$

This linearity is satisfied.

Q) What if the source term is non-linear?

Method to linearize a non-linear source term  $\rightarrow$

Source term linearization

Ex-1 Say,  $S = 3 + 4T$   $S = S_c + S_p T_p$

$$S_c = ? \quad S_p = ?$$

From the form, it may appear that  $S_c = 3$ ,  $S_p = 4$ .

Just now we've seen that  $S_p$  should be taken -ve. So the above form is not valid.

Use iterative process. initially take:

$$S_c = 3 + 4T_p^*, S_p = 0.$$

If the iteration has a tendency to diverge fast, considering appropriate initial value for  $S_p$  may be required.

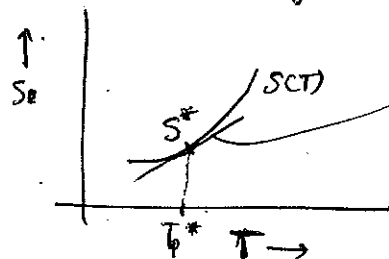
Ex-2  $S = 3 - 4T$

Here  $S_c = 3$ ,  $S_p = -4$  is valid & the best choice.

To slow down the convergence, we may choose  $S_c = 3 + 4T_p^*$   
 $S_p = -8.$

Ex-3  $S = 3 - 4T^3.$

Say we have any arbitrary  $S(T)$ .



Best linear function for this  $\rightarrow$  tangent at  $T_p^*$ .

(why? Same first order derivative as tangent as that of the non-linear curve).

$$\frac{S - S^*}{T - T_p^*} = \left. \frac{dS}{dT} \right|_{T_p^*}$$

$$S - S^* = \left. \frac{dS}{dT} \right|_{T_p^*} (T - T_p^*)$$

$$S^* = 3 - 4T_p^{*3}$$

$$\left. \frac{dS}{dT} \right|_{T_p^*} = -12T_p^{*2}$$

$$\begin{aligned} S - S^* &= S - (3 - 4T_p^{*3}) \\ &= -12T_p^{*2} (T - T_p^*) \end{aligned}$$

$$\rightarrow S = 3 + 8T_p^{*3} - 12T_p^{*2}T$$

$$\therefore S_c = (3 + 8T_p^{*3})$$

$$S_p = (-12T_p^{*2}) \rightarrow S_p \text{ -ve satisfied as } T_p^{*2} \geq 0$$

Ex-4

$$S = 3 + 4T^3$$

$$S^* = 3 + 4T_p^{*3} \left\{ \frac{dS}{dT} \bigg|_{T_p^*} = 12T_p^{*2} (T - T_p^*) \right\}$$

$$S - [3 + 4T_p^{*3}] = 12T_p^{*2} (T - T_p^*)$$

$$S = 3 - 8T_p^{*3} + 12T_p^{*2}T$$

$$\Rightarrow S_c = 3 - 8T_p^{*3}$$

$$S_p = 12T_p^{*2} \rightarrow \text{Here the } S_0 \text{ cannot work!}$$

What to do then?

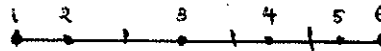
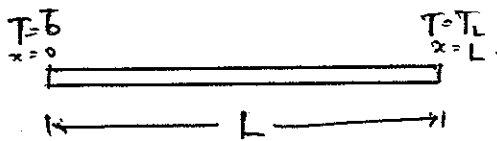
Dump the entirety to  $S_c$ .

$$S_c = 3 + 4T_p^{*3}, \quad S_p = 0$$

This linearization is mathematically correct, but will give physically inconsistent solution

## Lecture 20: Implementation of boundary conditions in FVM.

Ex-1



$$a_p T_p = a_E T_E + a_W T_W + b$$

For grid point-2,

$$a_2 T_2 = a_3 T_3 + a_1 T_1 + b$$

But at point-1,  $T_1 = T_0$  (given)

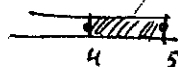
(Use penalty approach in FEM for further analysis)

Ex-2



To implement BC, consider CV at the boundary.

Take  $\frac{1}{2}$  of a CV. (Smaller length can capture sharper gradients accurately)



$$\text{G-de: } \frac{d}{dx} \left( k \frac{dT}{dx} \right) + S = 0$$

Integrating govt over the half CV  $\rightarrow$

$$\int_4^5 \frac{d}{dx} \left( k \frac{dT}{dx} \right) dx + \int_4^5 S dx = 0$$

Total length of  $\frac{1}{2}$  CV =  $\frac{\Delta x}{2}$ .

$$\Rightarrow k \frac{dT}{dx} \Big|_5 - k \frac{dT}{dx} \Big|_4 + S \frac{\Delta x}{2} = 0.$$

$$\downarrow$$
  
 $-q_L''$

Use profile assumption of piecewise linear nonconstant T.

$$\Rightarrow k \frac{T_5 - T_4}{\left(\frac{\Delta x}{2}\right)}$$

$$\Rightarrow -q_L'' - \frac{2k}{\Delta x} (T_5 - T_4) + S \frac{\Delta x}{2} = 0.$$

$$\therefore \frac{2k}{\Delta x} (T_5 - T_4) = -q_L'' + S \frac{\Delta x}{2}$$

$$T_5 = T_4 + \frac{S \Delta x^2}{4k} - \frac{q_L'' \Delta x}{2k}$$

(Expression written as

boundary term as a function of interior terms)

Eq. of the form:

$$a_5 T_5 = a_4 T_4 + b$$

where  $a_5 = 1$

$a_4 = 1$ ,

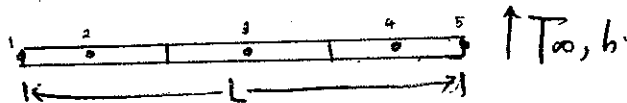
$$b = \left( \frac{S \Delta x^2}{4k} - \frac{q_L'' \Delta x}{2k} \right).$$

FD

$$q_L'' = -k \left( \frac{T_5 - T_4}{\frac{\Delta x}{2}} \right)$$

$$T_5 = T_4 - \frac{q_L'' \Delta x}{2k}$$

Ex-3 (Mixed type BC)



$$-k \frac{dT}{dx} \Big|_{x=L} = h (T_{x=L} - T_\infty)$$

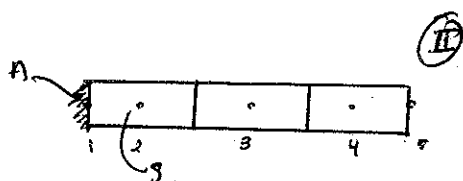
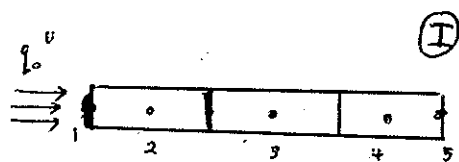
$\downarrow$   
 $q_L''$

$$-q_L'' - \frac{2k}{\Delta x} (T_5 - T_4) + S \frac{\Delta x}{2} = 0$$

$$\downarrow$$

$$h(T_5 - T_\infty) \rightarrow ( ) T_5 = ( ) T_4 + ( )$$

$\rightarrow$  very much analogous to Neumann BC case.



$$S_{extra} = \frac{q_0'' A_{face}}{V_{cv}}$$
 Volumetric heat generation (Heat source term)

$\rightarrow$  total rate of heat transfer.

I  $\rightarrow$  Heat flux entering through left boundary.

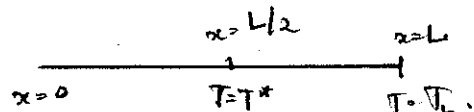
II  $\rightarrow$  Source at 2. Requirement: equivalent heat flux passing through face b/w 2 & 3 same as in I. Boundary 1 insulated.

Case-I is usually treated equivalently as case-II by dumping all the flux through corresponding heat source term ( $S_{extra}$ ). Already present source term unaffected. Why do so?

More rapid convergence to solution.

In case I, heat flux has to penetrate through 1 & then through face 2-3. While in case II, with the introduction of source, flux through 2-3 face is handled in 1 shot.

- Difference observed is marginal in most cases.



Can we specify the B/C at an internal Grid pt & still expect the problem to be well-posed?

$$\frac{d}{dx} \left( k \frac{dT}{dx} \right) = 0 \quad (\text{Take } S=0 \text{ for this eg.})$$

$$k = \text{const.}$$

$$\Rightarrow \frac{dT}{dx} = C_1$$

$$T = C_1 x + C_2$$

$x = L/2, T = T^* \rightarrow$

$x = L, T = T_L \rightarrow$

can obtain both  $C_1$  &  $C_2$ .

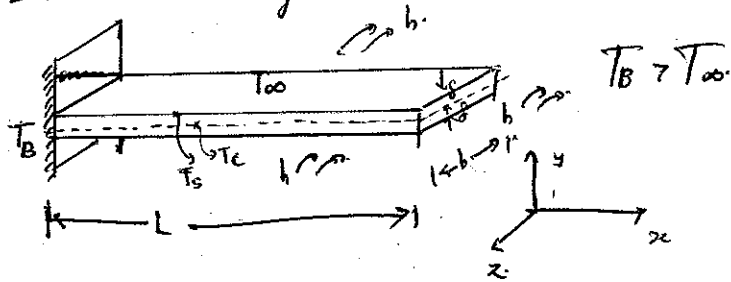
perfectly valid BE. Only thing is that it isn't specified at the physical boundary.

Cannot do the same for IVP. why?

B/C time is a one way coordinate system. B/c whatever is happening now cannot influence what has already happened a while back.

## Lecture 21: Implementation of boundary conditions in FVM (contd)

Ex 1-D steady state heat conduction in a fin.



Thermal resistance in  $x$  is most important. Out of  $y$  &  $z$ , since  $b \gg 2\delta$ , thermal resistance along  $z$  direction is the next significant one.

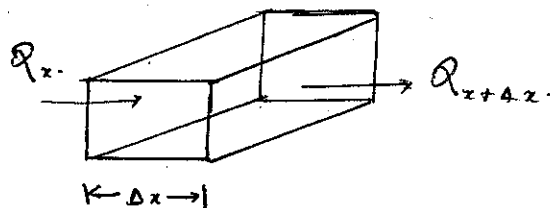
$\frac{\delta z}{kA_z} \cdot \frac{b}{kA_x}$  | Either the consideration  $b \gg 2\delta$ , or  $2\delta \gg b$  makes this problem 2D.

$$k \frac{T_c - T_b}{\delta} \sim h(T_s - T_\infty)$$

$$\frac{T_c - T_s}{T_s - T_\infty} \sim \frac{h\delta}{k} \quad \rightarrow B_i \rightarrow \text{assume this to be small.}$$

IF  $\frac{T_c - T_s}{T_s - T_\infty} \ll 1 \Rightarrow T_c \sim T_s \Rightarrow$  no necessity to analyse the temp difference b/w centreline & outside surface  
 $\Rightarrow$  no necessity to analyse heat transfer in  $y$  direction  
 $\Rightarrow$  1-D problem.

Take a section out of the fin:



$$Q_x - A Q_{conv} = Q_{x+\Delta x}$$

$$Q_x = Q_{x+\Delta x} \quad Q_x = q_x'' A$$

$$Q_{x+\Delta x} = q_{x+\Delta x}'' A$$

$$Q_x - Q_{x+\Delta x} = (q_x'' - q_{x+\Delta x}'') A$$

$$q_{x+\Delta x}'' = q_x'' + \Delta x \frac{dq_x''}{dx} + \dots$$

$$\Rightarrow Q_x - Q_{x+\Delta x} = - \left[ \frac{dq_x''}{dx} \Delta x + \dots \right] A.$$



$$Top = 2hb \Delta x (T - T_{\infty})$$

$$Front = 2h(2\delta) \Delta x (T - T_{\infty})$$

$$A Q_{conv} = Ph(T - T_{\infty}) \Delta x$$

$P = \text{perimeter} = 2(b + 2\delta)$

$$Q'_x - Q'_{x+\Delta x} = A Q_{conv}$$

$$- \frac{dq''_x}{dx} \Delta x + \dots = \frac{Ph(T - T_{\infty}) \Delta x}{A}$$

Take limit as  $\Delta x \rightarrow 0$ .

$$\Rightarrow - \frac{d q''_x}{dx} A = Ph(T - T_{\infty})$$

$$\text{and } q''_x = -kA \frac{dT}{dx}$$

$$+ \frac{d}{dx} (kA \frac{dT}{dx}) = Ph(T - T_{\infty})$$

$$\frac{d}{dx} (kA \frac{dT}{dx}) - Ph(T - T_{\infty}) = 0$$

↳ Governing DE.

Discretize using FVM:

The above eq<sup>n</sup> is of the form

$$\frac{d}{dx} (k \frac{dT}{dx}) + S = 0$$

$$\text{with } S = - \frac{Ph}{A} (T - T_{\infty})$$

↑  
implicitly linear source terms.

$$S = S_c + S_p T$$

$$S_c = \frac{Ph T_{\infty}}{A}, \quad S_p = - \frac{Ph}{A}$$

(-ve).

⇒ It's a well posed problem

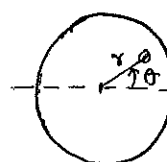
$$(4/w) \quad (1) \text{ At } x=0, T = T_B$$

$$(2) \text{ At } x=L, \frac{dT}{dx} = 0$$

Non-dim extn equations:  $\bar{x} = \frac{x}{L}$

Find Temp distribution with FDM, FEM, FVM.

Ex 1D steady state heat conduction in cylindrical coordinates.

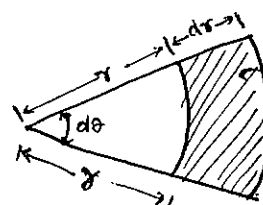


Consider  $L$  large  $\Rightarrow$  no variation along  $z$ .

Axially symmetric  $\Rightarrow$  no variation along  $\theta$ .

Governing DE:

$$\frac{1}{r} \frac{d}{dr} (rk \frac{dT}{dr}) + S = 0$$



$$dA = r d\theta dr$$

$$dV = r d\theta dr dz$$

$$dV = (r dr) (d\theta) (dz)$$

form integral eq<sup>n</sup> using this in  $\int$

$$\int \frac{1}{r} \frac{d}{dr} (rk \frac{dT}{dr}) r dr + \int S r dr = 0$$

↳ Multiplying with  $r dr$

remove singularity induced by  $\frac{1}{r}$ .

↳ length of the CV.

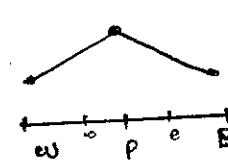
$$\left[ rk \frac{dT}{dr} \right]_{r+\Delta r} - \left[ rk \frac{dT}{dr} \right]_r + \int_r^{r+\Delta r} (S_c + S_p T) r dr$$

$\omega = 0$

(Assume)  $S = S_c + S_p T$

Constant temp over CV.

Make piecewise linear profile ~~ass~~  
assumptions b/w the grid pts.



$$\gamma_E k_E \frac{T_E - T_P}{\delta x_E} - \gamma_W k_W \frac{T_P - T_W}{\delta x_W} + (S_c + S_P T_P) \frac{\gamma_E^2 - \gamma_W^2}{2} = 0$$

Organise the eq<sup>n</sup> in the form:

$$a_P T_P = a_E T_E + a_W T_W + b$$

where  $a_P = \frac{\gamma_E k_E}{\delta x_E} + \frac{\gamma_W k_W}{\delta x_W}$

$$a_E = \frac{\gamma_E k_E}{\delta x_E} \quad a_W = \frac{\gamma_W k_W}{\delta x_W}$$

$$a_P = a_E + a_W - S_P \left( \frac{\gamma_E^2 - \gamma_W^2}{2} \right)$$

$$b = S_c \left( \frac{\gamma_E^2 - \gamma_W^2}{2} \right)$$

- If it were spherical coord system,  
eq<sup>n</sup> would have been

$$\frac{1}{r^2} \frac{d}{dr} \left( r^2 \frac{dT}{dr} \right) + S = 0$$

and while forming the integral eq<sup>n</sup>,  
we'd have to multiply with  $r^2 dr$   
instead of  $dx$  ~~and~~ before integrating  
to remove singularities induced by  $\frac{1}{r^2}$   
term.

## Lecture 22: 1D - Unsteady state Diffusion Problem.

FVM for:

1D unsteady state diffusion problem:  
(diffusion; so no fluid flow terms)

$$\frac{\partial}{\partial t} (\rho C_p T) + \nabla \cdot (\rho C_p T \vec{V}) = \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) + S$$

unsteady term      conduction term

$$\Rightarrow \frac{\partial}{\partial t} (\rho C_p T) = \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) + S$$

Our obj: how to take care of the <sup>bew</sup> term  
 $\frac{\partial}{\partial t} (\rho C_p T)$  ?

Integrate eq<sup>n</sup> over the domain.

Here domain consists of both  $t$  &  $x$ .

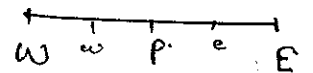
So our elemental domain consists of both  
 $dt$  &  $dx$ .

{Take  $S=0$ }

$$\int_W^E \int_t^{t+\Delta t} \frac{\partial}{\partial t} (\rho C_p T) dt dx = \int_t^{t+\Delta t} \int_W^E \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) dx dt$$

Treat both terms

separately

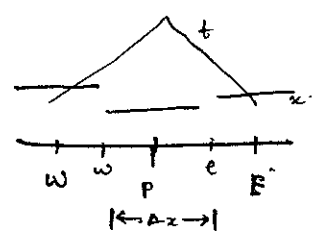


$$\int_W^E \int_t^{t+\Delta t} \frac{\partial}{\partial t} (\rho C_p T) dt dx = \int_W^E (\rho C_p T)^{t+\Delta t} - (\rho C_p T)^t dx$$

Make profile assumption for  $T$  variation  
in space:

Simplest assumption: piecewise constant  
profile

Why? Here no need to take piecewise  
linear forms about grid pts b/c there are  
no derivative terms involved.



Assume  $\rho C_p$  a constant.

$$\therefore \int_v^e [\rho C_p T]^{t+\Delta t} - [\rho C_p T]^t dx$$

$$= \rho C_p \Delta x (T_P^{t+\Delta t} - T_P^t)$$

Term = 2:

$$\int_t^{t+\Delta t} \int_w^e \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) dx dt$$

$$= \left( \left[ k \frac{dT}{dx} \right]^e - \left[ k \frac{dT}{dx} \right]^w \right) dt$$

Profile assumption for  $t$ :  
piecewise linear  $t$ .

$$\Rightarrow \int_t^{t+\Delta t} \left\{ k_e \left( \frac{T_E - T_P}{\Delta x_e} \right) - k_w \left( \frac{T_P - T_w}{\Delta x_w} \right) \right\} dt$$

$$\int_t^{t+\Delta t} T dt = ??$$

Make profile assumption:

$$= \left[ (1-f) T_P^t + f T^{t+\Delta t} \right] \Delta t$$

$$\Rightarrow \left[ \frac{k_e}{\Delta x_e} \left\{ (1-f) T_E^t + f T_E^{t+\Delta t} \right\} - \frac{k_e}{\Delta x_e} \left\{ (1-f) T_P^t + f T_P^{t+\Delta t} \right\} - \frac{k_w}{\Delta x_w} \left\{ (1-f) T_w^t + f T_P^{t+\Delta t} \right\} \right] \Delta t$$

$$+ \frac{k_w}{\Delta x_e} \left\{ (1-f) T_w^t + f T_w^{t+\Delta t} \right\} \Delta t$$

For notational convenience, write

$$T^t = T^0$$

$$T^{t+\Delta t} = T^1 = T$$

Term 1 = Term 2.

$$\Rightarrow \rho C_p T_P = a_E T_E + a_w T_w + \underbrace{a_P^0 T_P^0}_{\text{new terms appears!}} + b$$

where,

$$a_E = \frac{k_e f}{\Delta x_e}, \quad a_w = \frac{k_w f}{\Delta x_e}$$

$$a_P^0 = \frac{k_e (1-f)}{f} (a_E + a_w) + \rho C_p \frac{\Delta x}{\Delta t}$$

$$b = \frac{k_e}{\Delta x_e} (1-f) T_E^0 + \frac{k_w}{\Delta x_w} (1-f) T_w^0$$

$$a_P = \rho C_p \frac{\Delta x}{\Delta t} + \frac{k_e f}{\Delta x_e} + \frac{k_w f}{\Delta x_w}$$

$$= a_E + a_w + \rho C_p \frac{\Delta x}{\Delta t}$$

Earlier there were two neighbors E and W (spatial neighbors)

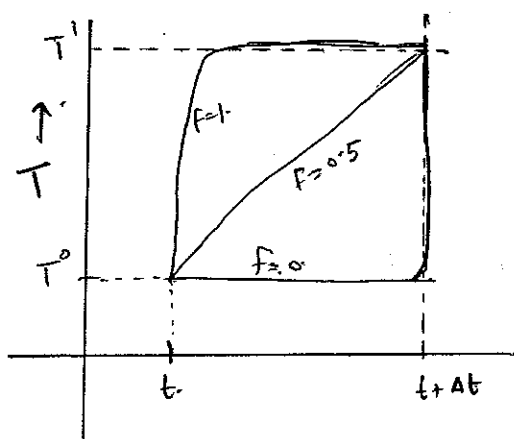
Now we have a temporal neighbor

$$a_P^0$$

We only have one neighbor for time  $t$  but whatever has happened at time  $t + \Delta t$  is influenced by the events at time  $t$  and not by events at time  $t + 2\Delta t$ .

That's why 2 space neighbors & 1 <sup>time</sup> neighbor

## Choices of $f$



- Remember  
 $0 \leq f \leq 1$

$$T^k = T^0$$

$$T^{k+\Delta t} = T^1 = T$$

$f=0 \rightarrow$  fully explicit scheme

$f=1 \rightarrow$  implicit scheme

$f=1/2 \rightarrow$  Crank-Nicholson scheme.

$f=0:$

$$q_E = 0, \quad q_W = 0, \quad q_P = \rho C_P \frac{\Delta x}{\Delta t},$$

$$q_P^0 = \rho C_P \frac{\Delta x}{\Delta t} - \frac{k_E}{\Delta x_E} - \frac{k_W}{\Delta x_W}$$

$$b = \frac{k_E}{\Delta x_E} T_E^0 + \frac{k_W}{\Delta x_W} T_W^0$$

$$q_P T_P = q_P^0 T_P^0 + b$$

$$T_P = \frac{q_P^0 T_P^0 + b}{q_P}$$

$T_P = f(T_E^0, T_P^0, T_W^0)$  expressed in an explicit form.

$f=1:$

$$q_E = \frac{k_E}{\Delta x_E}, \quad q_W = \frac{k_W}{\Delta x_W}, \quad q_P = q_E + q_W + \rho C_P \frac{\Delta x}{\Delta t}$$

$$q_P^0 = \rho C_P \frac{\Delta x}{\Delta t}, \quad b = 0.$$

Lecture 23: 1-D unsteady state diffusion  
Problem (contd.)

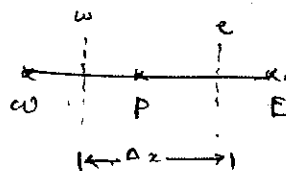
We should have  $q_P^0 \geq 0$ .

$f=0:$

$$q_P^0 = \rho C_P \frac{\Delta x}{\Delta t} - \frac{k_E}{\Delta x_E} - \frac{k_W}{\Delta x_W}$$

Let  $k_E = k_W = k$  (for algebraic simplicity)

$$\Delta x_E = \Delta x_W = \Delta x = \Delta x.$$



$$\Delta x_W = \Delta x_E = \Delta x$$

$$q_P^0 = \rho C_P \frac{\Delta x}{\Delta t} - \frac{2k}{\Delta x}$$

Conditions for  $q_P^0 \geq 0$ :

$$\rightarrow \rho C_P \frac{\Delta x}{\Delta t} \geq \frac{2k}{\Delta x}$$

$$\Rightarrow \frac{\alpha \Delta t}{\Delta x^2} \leq \frac{1}{2} \quad \alpha = \text{Thermal diffusivity}$$

$$\frac{\alpha t_c}{L^2} = F_o \rightarrow \text{Fourier number}$$

characteristic time  
characteristic length

Stability criterion for the explicit scheme.

Round off errors can propagate & amplify with calculations. If such a thing happens & it is inherent to the scheme itself, then such a scheme is an unstable scheme.

Key requirement of stable scheme:

→ physically consistent: coeffs are of same sign. If temp is increased at a pt, then that change will cause the temperature at neighbouring pts to increase as well & not decrease.

B/c of the condition  $\frac{\alpha \Delta t}{\Delta x^2} \leq \frac{1}{2}$ , change in grid spacing also affects the <sup>size of</sup> time step to be chosen.

$\frac{\alpha t_c}{L^2} = \frac{b_1}{t_c}$   $t_c \rightarrow \Delta t$   
 $t_c \rightarrow \frac{\Delta x^2}{\alpha}$  → characteristic time over which a thermal disturbance propagates by thermal diffusion in a medium.

→ conditionally stable schemes: the scheme works as long as  $\Delta t, \Delta x$  constraint is followed.

$f=1 \rightarrow$

all coeffs same sign

⇒ unconditionally stable

$f=0.5 \rightarrow$

$$\rho C_p \frac{\Delta x}{\Delta t} - \frac{2k}{\Delta x} \cdot \frac{1}{2} \geq 0$$

$$\Rightarrow \frac{\alpha \Delta t}{\Delta x^2} \leq 1.$$

## Lecture 24: Consequences of Discretization of Unsteady State Problems.

Consequences of time-discretization is

Finite Difference:

Errors associated with any Taylor series based discretizations:

→ Consistency: - characteristic of a numerical scheme.

Consistent if in the limit of grid size & time step size  $\rightarrow 0$ , the algebraic eqns imitate the same behavior as that of its parent diff. eqn.

This happens when error is nullified at such refined scales. ⇒ nullification of truncation error as grid size & time step <sup>size</sup> tends to zero in the limit.

⇒ discretized eqn tends to behave same as g.d.e.

→ Stability: Just like consistency talks about truncation errors, stability talks of round off errors.  $\Delta$  errors in a numerical scheme is similar to physical perturbation. How strongly these perturbations propagate/amplify in the presence of numerical calculation determines stability.

Stable  $\Rightarrow$  no amplification of numerical perturbations due to propagation of round-off errors.

→ Convergence: In the limit of grid size and time step size tends to 0, numerical soln  $\rightarrow$  exact solution.

Lax equivalence theorem: for linear problems, consistency + stability  $\Rightarrow$  convergence.

Consistency + stability need not ensure convergence for a non-linear problems. Such problems can have multiple solutions. To test for convergence in non-linear problems, the following is done. At a finite grid size & time step size evaluate the soln. Then take finer grid & smaller time step & evaluate soln. As finer & finer grid & time step is used, if the soln is found to be grid-independent & time-step independent, the non-linear problem has convergence.

Finite difference schemes on 1-D unsteady state diffusion problems:

$$\text{G.d.e: } \rho C_p \frac{\partial T}{\partial t} = k \frac{\partial^2 T}{\partial x^2}$$

Assume constant properties.

$$\frac{\partial T}{\partial t} = \alpha \frac{\partial^2 T}{\partial x^2} \quad (\alpha = \frac{k}{\rho C_p})$$

Use FTCS.

FT  $\rightarrow$  Forward Time

CS  $\rightarrow$  Central Space.

Corresponding Taylor series:

$$\begin{array}{ccc} x-\Delta x & x & x+\Delta x \\ | & | & | \\ i-1 & i & i+1 \\ (w) & (P) & (E) \end{array}$$

Superscript:

$n \rightarrow$  current time ( $t$ )

$n-1 \rightarrow$  previous time ( $t-\Delta t$ )

$n+1 \rightarrow$  next time ( $t+\Delta t$ )

$$T_i^{n+1} = T_i^n + \left. \frac{\partial T}{\partial t} \right|_i \Delta t + \left. \frac{\partial^2 T}{\partial t^2} \right|_i \frac{\Delta t^2}{2} + O(\Delta t^3)$$

$$\left. \frac{\partial T}{\partial t} \right|_i = \frac{T_i^{n+1} - T_i^n}{\Delta t} = \left. \frac{\partial^2 T}{\partial t^2} \right|_i \frac{\Delta t}{2} + O(\Delta t^2)$$

Truncated representation of forward time derivative:

$$\left. \frac{\partial T}{\partial t} \right|_i = \frac{T_i^{n+1} - T_i^n}{\Delta t}$$

$$T_{i+1}^n = T_i^n + \left. \frac{\partial T}{\partial x} \right|_i \Delta x + \left. \frac{\partial^2 T}{\partial x^2} \right|_i \frac{\Delta x^2}{2} + O(\Delta x^3)$$

$$\left. \frac{\partial^2 T}{\partial x^2} \right|_i = \frac{\Delta x^2}{6} + \left. \frac{\partial^4 T}{\partial x^4} \right|_i \frac{\Delta x^4}{24} + O(\Delta x^5)$$

$$T_{i+1}^{(1)} - T_i^{(1)} = T_i^{(1)} - \frac{\partial T}{\partial x} \left[ \Delta x + \frac{\partial^2 T}{\partial x^2} \frac{\Delta x^2}{2} \right] - \frac{\partial^3 T}{\partial x^3} \frac{\Delta x^3}{6} + \frac{\partial^4 T}{\partial x^4} \frac{\Delta x^4}{24} + O(\Delta x^5)$$

$$T_{i+1}^{(1)} + T_{i-1}^{(1)} - 2T_i^{(1)} = 2T_i^{(1)} + \frac{\partial^2 T}{\partial x^2} \Delta x^2 +$$

$$\frac{\partial^4 T}{\partial x^4} \frac{\Delta x^4}{12} + O(\Delta x^6)$$

$$\frac{T_{i+1}^{(1)} + T_{i-1}^{(1)} - 2T_i^{(1)}}{\Delta x^2} = -\frac{\partial^2 T}{\partial x^2} \frac{\Delta x^2}{12} + O(\Delta x^4)$$

Sub into g.d.e.

$$\frac{T_i^{(n+1)} - T_i^{(n)}}{\Delta t} = \left[ \frac{\partial^2 T}{\partial t^2} \frac{\Delta t}{2} \right] + O(\Delta t^2)$$

$$= \frac{T_{i+1}^{(n)} + T_{i-1}^{(n)} - 2T_i^{(n)}}{\Delta x^2} \left[ \frac{\partial^4 T}{\partial x^4} \frac{\Delta x^2}{12} \right] + O(\Delta x^4)$$

$$g.d.e \rightarrow \frac{\partial^2}{\partial t^2} \Rightarrow \frac{\partial^3 T}{\partial t \partial x^2} = \alpha \frac{\partial^4 T}{\partial x^4}$$

$$g.d.e \rightarrow \frac{\partial}{\partial t} \Rightarrow \frac{\partial^2 T}{\partial t^2} = \alpha \frac{\partial^3 T}{\partial t \partial x^2}$$

$$\frac{\partial^2 T}{\partial t^2} = \alpha \frac{\partial^3 T}{\partial t \partial x^2} = \alpha^2 \frac{\partial^4 T}{\partial x^4}$$

Term ① - Term ②.

$$\Rightarrow \alpha \frac{\partial^4 T}{\partial x^4} \left[ \frac{\alpha \Delta t}{2} - \frac{\Delta x^2}{12} \right]$$

$$\hookrightarrow O(\Delta t), O(\Delta x^2)$$

First order in time,

Second order in space.

$$\alpha \frac{\partial^4 T}{\partial x^4} \left[ \frac{\alpha \Delta t}{2} - \frac{\Delta x^2}{12} \right] = 0.$$

$$\text{when } \frac{\alpha \Delta t}{2} - \frac{\Delta x^2}{12} = 0$$

$$\Rightarrow \Delta t = \frac{\alpha \Delta x^2}{6} = \frac{1}{6}.$$

Then  $O(\Delta t^2), O(\Delta x^4)$  are errors in time & space respectively.

Is it consistent? Yes! as  $\Delta x \rightarrow 0, \Delta t \rightarrow 0$

error  $\rightarrow 0$ .

### Lecture 25: FTCS scheme.

$$FTCS: \frac{\partial T}{\partial t} = \alpha \frac{\partial^2 T}{\partial x^2}$$

$$\frac{T_i^{(n+1)} - T_i^{(n)}}{\Delta t} = \alpha \frac{T_{i+1}^{(n)} + T_{i-1}^{(n)} - 2T_i^{(n)}}{\Delta x^2}$$

Checking for stability: (whether numerical perturbations amplify while propagating or not).

$$T_i^{(n+1)} - T_i^{(n)} = \alpha [T_{i+1}^{(n)} + T_{i-1}^{(n)} - 2T_i^{(n)}]$$

$$\Rightarrow T_i^{(n+1)} = (1-2\alpha) T_i^{(n)} + \alpha T_{i+1}^{(n)} + \alpha T_{i-1}^{(n)} \quad \text{--- ①}$$

→ analogy with FV discretization.

→ Explicit scheme.  $T_i^{(n+1)}$  described in terms of previous time step.

Say, we get an approximate soln. That soln must also satisfy above eq?

i.e.

$$T_i^{(n+1)} = (1-2\alpha) T_i^{(n)} + \alpha T_{i+1}^{(n)} + \alpha T_{i-1}^{(n)} \quad \text{--- ②}$$

$$\textcircled{1} - \textcircled{2} \Rightarrow$$

$$(T_i - T_i^*)^{n+1} = (1-2r)(T_i - T_i^*)^n +$$

$$r(T_{i+1} - T_{i+1}^*)^n + r(T_{i-1} - T_{i-1}^*)^n$$

$$\text{Say } e = T - T^*$$

Then

↳ (error)

$$E_i^{n+1} = (1-2r)E_i^n + rE_{i+1}^n + rE_{i-1}^n$$

errors satisfy the same eq<sup>n</sup> as discretization variables

$$E = E(x, t)$$

Write this in terms of a Fourier series.

$$E(x, t) = \sum A \sum_j e^{ijkx} \quad i = \sqrt{-1}$$

What is the particular form of  $\sum$ ?

A convenient form is  $e^{at}$ , where  $a = \ln(j)$

Why  $e^{at}$ ? exponential fn helps evaluate the growth/decay behavior easily.

$$\text{If } e^{at+4t} < e^{at} \rightarrow \text{decay,}$$

$$e^{at+4t} > e^{at} \rightarrow \text{growth.}$$

Assess whether there is exponential growth/decay.

work space, there is a periodicity.

Over length of  $\frac{1}{\lambda}$  CV, repeatability observed.

Corresponding  $\omega$  are number (k) (number of waves over a time period), corresponding  $\omega$  are length = cell length.

$$e^{at} e^{ijkx} \quad j = \sqrt{-1}$$

$$e^{a(t+4t)} \cdot e^{ijkx}$$

$$= (1-2r) e^{at} e^{ijkx} + r e^{at} e^{jk(x+4x)} + r e^{at} e^{jk(x-4x)}$$

$$A = \text{Amplification factor} = \frac{e^{a(t+4t)}}{e^{at}}$$

$$A = (1-2r) + r(e^{jk4x} + e^{jk(-4x)})$$

Check for stability.  $|A| < 1$

If regardless of  $r$ ,  $|A| < 1$ , then

unconditionally stable.

If regardless of  $r$ ,  $|A| > 1$ , then unconditionally unstable.

If for some values of  $r$ ,  $|A| < 1$ , then

conditionally stable.

$$A = (1-2r) + r(e^{j\theta} + e^{-j\theta})$$

$$= (1-2r) + r(\cos\theta + j\sin\theta + \cos\theta - j\sin\theta)$$

$$= (1-2r) + 2r \cos\theta$$

$$= 1 - 2r(1 - \cos\theta)$$

$$= 1 - 4r \sin^2\left(\frac{\theta}{2}\right)$$

For stability.  $|A| \leq 1$ .

$$\Rightarrow -1 \leq A \leq 1$$

$$\left[ -1 \leq 1 - 4r \sin^2\left(\frac{\theta}{2}\right) \leq 1 \right]$$

$$\frac{-5/2}{25} \rightarrow 4r \sin^2\left(\frac{\theta}{2}\right) \leq 2$$

$$r \leq \frac{1}{2 \sin^2(\theta/2)}$$

Conservative upper limit  $\Rightarrow r \leq \frac{1}{2}$



Right hand limit

$$1 - 4r \sin^2\left(\frac{\theta}{2}\right) \leq +1$$

$$\Rightarrow 4r \sin^2\left(\frac{\theta}{2}\right) \geq 0 \rightarrow \text{always true!}$$

$$r = \frac{\alpha \Delta t}{\Delta x^2} > F_0 \leq \frac{1}{2}$$

Lecture 26: CTCS scheme (Leap Frog Scheme) and Dufort-Frankel Scheme

Ex CTCS Scheme (Leap frog scheme).

$$\frac{\partial T}{\partial t} = \alpha \frac{\partial^2 T}{\partial x^2}$$

$$\frac{T_i^{n+1} - T_i^{n-1}}{2\Delta t} = \alpha \frac{T_{i+1}^n + T_{i-1}^n - 2T_i^n}{\Delta x^2}$$

$$T_i^{n+1} - T_i^{n-1} = \frac{2\alpha}{\Delta x^2} [T_{i+1}^n + T_{i-1}^n - 2T_i^n] \rightarrow \text{①}$$

Approx soln:

$$T_i^{n+1} - T_i^{n-1} = 2r [T_{i+1}^n + T_{i-1}^n - 2T_i^n] \rightarrow \text{②}$$

① - ②  $\rightarrow$

$$E_i^{n+1} - E_i^{n-1} = 2r [E_{i+1}^n + E_{i-1}^n - 2E_i^n]$$

Sub the form  $e^{at} e^{jkx}$

$$e^{a(t+\Delta t)} e^{jkx} - e^{a(t-\Delta t)} e^{jkx}$$

$$= 2r [e^{at} e^{jk(x+\Delta x)} + e^{at} e^{jk(x-\Delta x)} - 2e^{at} e^{jkx}]$$

Divide both sides by  $e^{at} e^{jkx}$

$$e^{a\Delta t} - e^{-a\Delta t} = 2r [e^{jk\Delta x} + e^{-jk\Delta x} - 2]$$

$$A - \frac{1}{A} = 2r [e^{j\theta} + e^{-j\theta} - 2]$$

$$A - \frac{1}{A} = 2r [2\cos\theta - 2]$$

$$A - \frac{1}{A} = 4r (\cos\theta - 1) = -8r \sin^2(\theta/2)$$

$$A^2 + 8r \sin^2(\theta/2) A - 1 = 0$$

Quadratic eqn in A.

Product of the roots has a magnitude of 1. But

$$A = \frac{-8r \sin^2(\theta/2) \pm \sqrt{64r^2 \sin^4(\theta/2) + 4}}{2}$$

$$= -4r \sin^2(\theta/2) \pm \sqrt{16r^2 \sin^4(\theta/2) + 1}$$

Greater magnitude of A  $\rightarrow$

$$|-4r \sin^2(\theta/2) - \sqrt{16r^2 \sin^4(\theta/2) + 1}| \rightarrow \geq 1$$

$\Rightarrow$  unconditionally unstable!

But the need for higher accuracy

(CT  $\sim O(\epsilon^2)$ , FT  $\sim O(\epsilon)$ ) leads to

us using CTCS over FTCS. But CTCS is unconditionally unstable while FTCS is <sup>conditionally</sup> stable.

Modification of this scheme will provide accuracy as well as stability. Such a modification  $\rightarrow$  Dufort-Frankel Scheme.

$$\frac{\partial T}{\partial t} \propto \frac{\partial^2 T}{\partial x^2}$$

$$\frac{T_i^{n+1} - T_i^{n-1}}{2\Delta t} = \alpha \left[ \frac{T_{i+1}^n + T_{i-1}^n - 2T_i^n}{\Delta x^2} \right]$$

Make adhoc change:

$$T_i^n = \frac{T_i^{n+1} + T_i^{n-1}}{2}$$

(Temperature at a grid pt at an instant = average temperature at that grid pt at previous time and the next time).

This adhoc change: CTCS  $\rightarrow$  Dufort-Frankel Scheme.

$$T_i^{n+1} - T_i^{n-1} = 2\alpha \left[ \frac{T_{i+1}^n + T_{i-1}^n - T_i^{n+1} - T_i^{n-1}}{2} \right]$$

$$E_i^{n+1} - E_i^{n-1} = 2\alpha \left[ \frac{E_{i+1}^n + E_{i-1}^n - E_i^{n+1} - E_i^{n-1}}{2} \right]$$

$$e^{a(t+\Delta t)} e^{jkx} - e^{a(t-\Delta t)} e^{jkx}$$

$$= 2\alpha \left[ e^{at} e^{jk(1+\Delta x)} + e^{at} e^{jk(x-\Delta x)} - e^{a(t-\Delta t)} e^{jkx} - e^{a(t+\Delta t)} e^{jkx} \right]$$

Divide by  $e^{at} e^{jkx}$ ,

$$A - A^{-1} = 2\alpha \left[ e^{jk\Delta x} + e^{-jk\Delta x} - A - A^{-1} \right]$$

$$A = \frac{1}{A}$$

$$A(1+2\alpha) - \frac{1}{A}(1-2\alpha) = 2\alpha(2\cos\theta)$$

$$(1+2\alpha)A^2 - 4\alpha\cos\theta A - (1-2\alpha) = 0$$

$$A = \frac{4\alpha\cos\theta \pm \sqrt{16\alpha^2\cos^2\theta + 4(1-4\alpha^2)}}{2(1+2\alpha)}$$

$$= \frac{2\alpha\cos\theta \pm \sqrt{4\alpha^2\cos^2\theta + 1-4\alpha^2}}{1+2\alpha}$$

$$= \frac{2\alpha\cos\theta \pm \sqrt{1-4\alpha^2\sin^2\theta}}{1+2\alpha}$$

Case 1:  $2\alpha < 1$

Then  $4\alpha^2 < 1$

$$\therefore 4\alpha^2\sin^2\theta < 1$$

$$A = \frac{2\alpha\cos\theta \pm \sqrt{1-4\alpha^2\sin^2\theta}}{1+2\alpha}$$

Take (+) for more conservative estimate,

$$A = \frac{(2\alpha\cos\theta + \sqrt{1-4\alpha^2\sin^2\theta})}{1+2\alpha}$$

$$\leq \frac{1+2\alpha\cos\theta}{1+2\alpha} \leq 1$$

Case 2:  $2r > 1$

$$A = \frac{2r \cos \theta \pm j \sqrt{4r^2 \sin^2 \theta - 1}}{1 + 2r}$$

$$|A| = \frac{4r^2 \cos^2 \theta + 4r^2 \sin^2 \theta - 1}{(1 + 2r)^2}$$

$$= \frac{4r^2 - 1}{(1 + 2r)^2} = \frac{(2r-1)(2r+1)}{(1+2r)(1+2r)} < 1$$

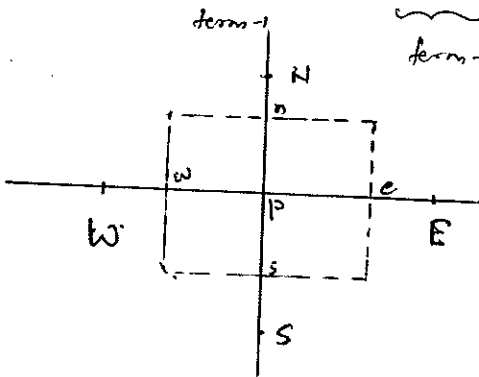
This scheme  $\rightarrow$  conditionally stable, but not consistent (H/W: prove)

W.I.C. truncation error as  $\Delta t \rightarrow 0$  &  $\Delta x \rightarrow 0$  doesn't nullify itself.

Lecture 27: FV Discretization of 2D unsteady state Diffusion Type Problems:

Eg: heat conduction:

$$\rho C_p \frac{\partial T}{\partial t} = \underbrace{\frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right)}_{\text{Term-2}} + \underbrace{\frac{\partial}{\partial y} \left( k \frac{\partial T}{\partial y} \right)}_{\text{Term-3}} + \underbrace{S}_{\text{Term-4}}$$



$$\text{Term-1} \rightarrow \iiint_{s,w,t} \rho C_p \frac{\partial T}{\partial t} dt dx dy$$

$\rho C_p (T_p - T_p^0) \Delta x \Delta y$   $\rightarrow$  can use piecewise constant temperature profile (no derivative terms involved). Then  $T = T_p$ .

Term-2:

$$\iiint_{t,s,w} \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) dx dy dt$$

$\int \frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) dx$  Take piecewise-linear profile.

$$= k_e \frac{T_E - T_p}{\Delta x_e} - k_w \frac{T_p - T_w}{\Delta x_w}$$

Final integration form:

= Consider a fully implicit scheme.

$$\Rightarrow \left( k_e \frac{T_E - T_p}{\Delta x_e} - k_w \frac{T_p - T_w}{\Delta x_w} \right) \Delta t \Delta y$$

Term 3:

$$\iiint_{t,w,s} \frac{\partial}{\partial y} \left( k \frac{\partial T}{\partial y} \right) dy dx dt$$

$$\rightarrow \left( k_n \frac{T_N - T_p}{\Delta y_n} - k_s \frac{T_p - T_s}{\Delta y_s} \right) \Delta t \Delta x$$

Term 4:

$$\iiint_{t,s,w} (S_c + S_p T_p) dx dy dt$$

$$= (S_c + S_p T_p) \Delta x \Delta y \Delta t$$

Assemble terms:

Divide all terms by  $\Delta t$ :

Then we get eqn of the form:

$$a_p T_p = a_E T_E + a_w T_w + a_s T_s + a_N T_N + a_p^0 T_p^0 + b$$

$$a_E = \frac{k_e}{\Delta x_e} \Delta y, \quad a_w = \frac{k_w}{\Delta x_w} \Delta y,$$

$$a_s = \frac{k_s}{\Delta y_s} \Delta x, \quad a_N = \frac{k_n}{\Delta y_n} \Delta x, \quad a_p^0 = \frac{\rho C_p \Delta x \Delta y}{\Delta t}$$

$$a_p = a_E + a_W + a_S + a_N + \rho C_p \frac{\Delta x \Delta y}{\Delta t}$$

$$b = S_c \Delta x \Delta y - S_p \Delta x \Delta y$$

of the form:

$$a_p T_b = \sum a_{nb} T_{nb} + b \quad (\text{nb} \rightarrow \text{neighbour})$$

The coeffs  $a_E, a_W, a_S, a_N$  etc physically represent thermal conductance. why?

Take  $a_E = \frac{k_c A y}{\Delta x}$ . Assuming unit length normal to the x-y plane,

$(\Delta y \times 1) = \text{area of face}$ . So

$a_E$  is of the form  $a_E = \frac{k_c A}{L}$ , which

is the formula for conductance,  $\frac{L}{k_c A} \rightarrow$  resistance.

By setting  $\Delta t$  to very large number, this unsteady problem can be converted into a steady state problem as terms

containing  $(\Delta t)^{-1}$  becomes negligibly small.

(U/W): repeat same exercise for a fully explicit scheme.

• Solving system of algebraic eq<sup>ns</sup>:-

$$\textcircled{1} \quad x + y = 2$$

$$2x + 3y = 5$$

$$E_1 \times 2 - E_2$$

$$-y = -1$$

$$\Rightarrow y = 1$$

$$E_1: x = 2 - y = 2 - 1 = 1$$

using method of elimination

$$(x, y) = (1, 1)$$

$$\textcircled{2} \quad x + y = 2$$

$$2x + 2y = 4$$

line arly dependent eq<sup>ns</sup>.

No. of independent eq<sup>s</sup>.

< no. of unknowns.

~~④~~

$$\textcircled{3} \quad x + y = 2$$

$$y = 2 - x$$

$$\begin{bmatrix} x \\ y \end{bmatrix} = \begin{bmatrix} 2 \\ 2-x \end{bmatrix} = \begin{bmatrix} 1 \\ -1 \end{bmatrix} x + \begin{bmatrix} 2 \\ 2 \end{bmatrix}$$

$$\textcircled{3} \quad x + y = 2$$

$$2x + 2y = 5$$

$$E_1 \times 2 \rightarrow 2x + 2y = 4$$

$$\text{From } E_2 \rightarrow 2x + 2y = 5$$

$$\text{Equate them both} \Rightarrow 4 = 5 \quad \times$$

$\rightarrow$  no sol<sup>n</sup>.

(parallel straight lines — no intersection pts).

$$\textcircled{4} \quad x + y = 0$$

$$2x + 3y = 0$$

} If RHS = 0  $\rightarrow$  homogeneous system of eq<sup>ns</sup>.

$$x = 0, y = 0 \rightarrow$$

trivial sol<sup>ns</sup>.

no non-trivial sol<sup>ns</sup>.

$$\textcircled{5} \quad x + y = 0$$

$$2x + 2y = 0$$

trivial sol<sup>n</sup>  $x = 0, y = 0$

$\infty$  non-trivial sol<sup>ns</sup>.

$$y = -x$$

what if number of eqns are large?

Use matrix forms.

$$\textcircled{1} \rightarrow \underbrace{\begin{bmatrix} 1 & 1 \\ 2 & 3 \end{bmatrix}}_{\substack{\text{Coeff} \\ \text{matrix}}} \begin{bmatrix} x \\ y \end{bmatrix} = \begin{bmatrix} 2 \\ 5 \end{bmatrix}$$

$$\left[ \begin{array}{cc|c} 1 & 1 & 2 \\ 2 & 3 & 5 \end{array} \right] \rightarrow \text{Augmented matrix.}$$

Lecture 28: Solutions to linear algebraic equations (contd).

Identify rank of coeff matrix & augmented matrix. Why rank? B/c it gives an idea about the linear dependence/independence of equations in the system.

Rank of a matrix is  $r$ , if

- (i) It has atleast one non-zero minor of order  $r$ .
- (ii) all minors of order  $> r$  vanishes ( $=0$ ).

• For eg: 1,

Take  $R_c = 2 \rightarrow$  coeff matrix

$R_A = 2 \rightarrow$  aug. matrix.

Observation:  $R_c = R_A = 2 = \frac{n}{n}$  (no. of eqns = no. of unknowns).

• For eg: 2,

$$\left[ \begin{array}{cc|c} 1 & 1 & 2 \\ 2 & 2 & 4 \end{array} \right]$$

$$R_c = 1$$

$$R_A = 1$$

Observation:  $R_A = R_c = 1 < n$

• For eg: 3,

$$\left[ \begin{array}{cc|c} 1 & 1 & 2 \\ 2 & 2 & 5 \end{array} \right]$$

$$R_c = 1$$

$$R_A = 2$$

Obs:  $R_c \neq R_A$

For a system of non-homogeneous eqns:

$\rightarrow R_c = R_A = n \Rightarrow$  unique sol<sup>n</sup>.

$\rightarrow R_c = R_A < n \Rightarrow$  infinitely large no. of solutions.

$\rightarrow R_c \neq R_A \Rightarrow$  no solution.

For homogeneous eqns, since RHS is all 0, there

is no need for augmented matrix.

For system of homogeneous eqns:

$\Delta \neq 0 \rightarrow$  only trivial sol<sup>n</sup>

$\Delta = 0 \rightarrow$  infinitely large no. of solutions

It's important to see the nature of sol<sup>n</sup> for system of algebraic eqns. Why? For eg, for well-posed problems, uniqueness is an important criterion. But if the system of non-homogeneous eqns has infinitely many solutions, the problem itself is ill defined & needs modifications.

Solution techniques for systems of linear algebraic equations:

- Elimination
- Iteration
- Gradient search method.

Elimination method:

$$\begin{aligned} \bullet \quad x_1 + x_2 &= 7. \quad (E_1) \\ 2x_1 + 3x_2 &= 5. \quad (E_2) \\ 2E_1 - E_2 &\Rightarrow \text{Effort is to eliminate } x_1 \end{aligned}$$

$$\begin{aligned} \bullet \quad x_1 + x_2 + x_3 &= 3 \quad (E_1) \\ 2x_1 + 2x_2 + 3x_3 &= 7 \quad (E_2) \\ 3x_1 + x_2 + 2x_3 &= 6. \quad (E_3) \end{aligned}$$

$$\begin{aligned} 2 \times E_1 - E_2 &\Rightarrow x_3 = 1. \\ E_3 - 3E_1 &\Rightarrow -2x_2 - 2x_3 = -3 \\ -2x_2 - 2(1) &= -3. \\ x_2 &= \frac{1}{2} \end{aligned}$$

$$\left[ \begin{array}{ccc|c} 1 & 1 & 1 & 3 \\ 2 & 2 & 3 & 7 \\ 3 & 1 & 2 & 6 \end{array} \right] \rightarrow \left[ \begin{array}{ccc|c} 1 & 1 & 1 & 3 \\ 0 & 0 & 1 & 1 \\ 0 & 0 & 0 & 0 \end{array} \right]$$

Reorder:

$$\begin{aligned} x_1 + x_2 + x_3 &= 3. \\ 3x_1 + x_2 + 2x_3 &= 6. \\ 2x_1 + 2x_2 + 3x_3 &= 7 \end{aligned}$$

$$\left[ \begin{array}{ccc|c} 1 & 1 & 1 & 3 \\ 0 & -2 & -1 & -3 \\ 0 & 0 & 1 & 1 \end{array} \right] \rightarrow \text{Converted into upper triangular form.}$$

Now,  $x_3 = 1$

$$x_2 = \frac{3 - x_3}{2} = 1$$

$$\Rightarrow x_1 = 3 - x_2 - x_3 = 1$$

We can break this method into two parts:

Part-I: forward elimination.

$\Rightarrow$  Convert  $C$  to upper  $\Delta$  form.

part-II: Backwards substitution.

$$\begin{aligned} \bullet \quad x_1 + x_2 + x_3 &= 3. \\ 2x_1 + x_2 + 3x_3 &= 6. \\ 3x_1 + 4x_2 + 2x_3 &= 9. \end{aligned}$$

$$\left[ \begin{array}{ccc|c} 1 & 1 & 1 & 3 \\ 2 & 1 & 3 & 6 \\ 3 & 4 & 2 & 9 \end{array} \right] \quad \begin{aligned} E_2 - 2E_1 &\rightarrow -x_2 + x_3 = 0 \\ E_3 - 3E_1 &\rightarrow x_2 - x_3 = 0 \end{aligned}$$

linear dependency involved!

$$(E_3 - 5E_1 - E_2).$$

Lecture 29: Elimination Methods.

→

$$E_1: 10x_1 + x_2 + x_3 = 12. \quad \text{--- } (E_1)$$

$$x_1 + 10x_2 + x_3 = 12. \quad \text{--- } (E_2)$$

$$x_1 + x_2 + 10x_3 = 12. \quad \text{--- } (E_3)$$

Step-1: Row 1 as the pivot row

$$(E_2) \rightarrow (E_2) - \frac{1}{10} \times (E_1)$$

also

$$\begin{bmatrix} 10 & 1 & 1 & 12 \\ 1 & 10 & 1 & 12 \\ 1 & 1 & 10 & 12 \end{bmatrix}$$

$$(E_3) \rightarrow (E_3) - \frac{1}{10} (E_1)$$

$$\rightarrow \begin{bmatrix} 10 & 1 & 1 & 12 \\ 0 & 9.9 & 0.9 & 10.8 \\ 0 & 0.9 & 9.9 & 10.8 \end{bmatrix}$$

Step-2:

$$(E_3) \rightarrow (E_3) - \frac{0.9}{9.9} (E_2)$$

$$\rightarrow \begin{bmatrix} 10 & 1 & 1 & 12 \\ 0 & 9.9 & 0.9 & 10.8 \\ 0 & 0 & 9.818 & 9.818 \end{bmatrix}$$

2 steps required for 3 equations.

So in general, for  $n$  equations, it would require  $n-1$  steps.

→ Upto this forward elimination.

Next: Backward substitution:

$$\text{From } (E_3), \quad x_3 = \frac{9.818}{9.818} = 1.$$

$$\text{From } (E_2), \quad 9.9x_2 = 10.8 -$$

& finally get  $x_1$  from  $(E_1)$

## Generalization of Gaussian Elimination: (36)

$$a_{11}x_1 + a_{12}x_2 + a_{13}x_3 + \dots + a_{1n}x_n = b_1 \quad \text{--- } E_1$$

$$a_{21}x_1 + a_{22}x_2 + \dots + a_{2n}x_n = b_2 \quad \text{--- } E_2$$

⋮

$$a_{n1}x_1 + a_{n2}x_2 + \dots + a_{nn}x_n = b_n \quad \text{--- } E_n$$

Row-1 pivot

$$E_2 \rightarrow E_2 - \frac{a_{21}}{a_{11}} E_1$$

$$E_3 \rightarrow E_3 - \frac{a_{31}}{a_{11}} E_1$$

⋮

$$E_i \rightarrow E_i - \frac{a_{i1}}{a_{11}} E_1$$

for step-1:

$$a_{ij} \rightarrow a_{ij} - \frac{a_{i1}}{a_{11}} \times a_{1j}$$

for step-2:

$$a_{ij} \rightarrow a_{ij} - \frac{a_{i2}}{a_{22}} \times a_{2j}$$

Generalize for step no.  $k$ :  $(k \text{ for pivot})$

$$a_{ij} \rightarrow a_{ij} - \frac{a_{ik}}{a_{kk}} \times a_{kj}$$

# Lecture 30: Gaussian Elimination and LU Decomposition. methods.

Formalize forward elimination:

```
for k=1 to n-1
  for j=k+1 to n+1 augmented matrix considered
    for i=k+1 to n
```

$a_{ik}$  is already ready if 3rd loop comes in 2nd loop's position. So

```
for k=1 to n-1
  for i=k+1 to n
    R =  $a_{ik}/a_{kk}$ ;
    for j=k+1 to n+1
       $a_{ij} = a_{ij} - R * a_{kj}$ ;
    end
  end
end
```

Formalize backward substitution:

$$x_n = \frac{b_n}{a_{nn}} \rightarrow a_{nn} x_n = b_n$$

$$a_{n-1,n-1} x_{n-1} + a_{n-1,n} x_n = b_{n-1}$$

$$x_{n-1} = \frac{b_{n-1} - a_{n-1,n} x_n}{a_{n-1,n-1}}$$

$$a_{n-2,n-2} x_{n-2} + a_{n-2,n-1} x_{n-1} + a_{n-2,n} x_n = b_{n-2}$$

$$x_{n-2} = \frac{1}{a_{n-2,n-2}} \left[ b_{n-2} - a_{n-2,n-1} x_{n-1} - a_{n-2,n} x_n \right]$$

$$x_{n-i} = \frac{1}{a_{n-i,n-i}} \left[ b_{n-i} - \sum_{j=i+1}^n a_{n-i,j} x_j \right]$$

Formal algo for backward substitution:

```
22  $x_n = \frac{b_n}{a_{nn}}$ ;
for i=1 to n-1
  sum = 0
  for j=i+1 to n
    sum = sum +  $a_{n-i,j} x_j$ 
  end
   $x_{n-i} = (b_{n-i} - \text{sum}) / a_{n-i,n-i}$ ;
end
```

Assessment of number of computations:-

Forward elimination  $\rightarrow O(n^3) \rightarrow (n \text{ for each loop})$

Backward substitution  $\rightarrow O(n^2)$

Rate determining step: forward elimination.

$\therefore$  Computational complexity of the algo  $\rightarrow O(n^3)$ .

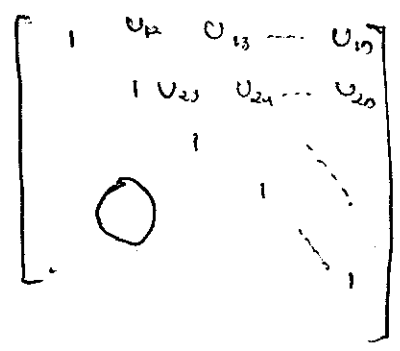
LU decomposition evolved  $\phi$  reduce the complexity.  
It is  $O(n^3)$ .

L-U decomposition technique:

Factorize  $A = L U$ .  $\rightarrow$  Upper triangular matrix  
 $\hookrightarrow$  Lower triangular matrix.

$$\begin{bmatrix} a_{11} & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{22} & a_{23} & \dots & a_{2n} \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} \end{bmatrix} = \begin{bmatrix} l_{11} & 0 & 0 & \dots & 0 \\ l_{21} & l_{22} & 0 & \dots & 0 \\ l_{31} & l_{32} & l_{33} & \dots & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ l_{n1} & l_{n2} & l_{n3} & \dots & l_{nn} \end{bmatrix} \times$$





Crawt's method:

$$\begin{aligned} a_{11} &= l_{11} & l_{11} &= a_{11} \\ a_{21} &= l_{21} & l_{21} &= a_{21} \\ a_{31} &= l_{31} & l_{31} &= a_{31} \\ &\vdots & & \\ a_{n1} &= l_{n1} & l_{n1} &= a_{n1} \end{aligned}$$

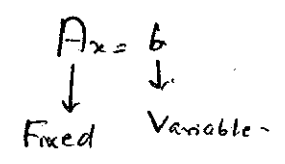
$$\begin{aligned} AX &= b \\ L[U]X &= b \\ L Z &= b \end{aligned} \quad \left| \quad \begin{aligned} Z &= L^{-1}b \rightarrow O(n^2) \quad (\text{Forward elim.}) \\ UX &= Z \\ X &= U^{-1}Z \rightarrow O(n^2) \quad (\text{Back sub.}) \end{aligned}$$

L is triangular matrix.

Even though complexity may seem  $O(n^2)$ , calculations required to factorize is has complexity  $O(n^3)$ . Thus it is no better than Gaussian elimination.

Then why use L-U decomposition method at all?

Lecture 31: Illustrative example of elimination method.



→ Special case: When A is symmetric and +ve definite.  $\Rightarrow U = L^T$  (Cholesky's L-U factorization).  
Here number of calculations become half (But doesn't become  $O(n^3) \rightarrow O(n^2)$ ; remains  $O(n^3)$  itself)  
Symmetric:  $A = A^T$   
+ve definite:  $V^T A V = I$

Ex.  $\epsilon x_1 + x_2 = 1$  (E1)  
 $x_1 + x_2 = 2$  (E2)  $\epsilon \rightarrow$  a small no.

Solve by Gaussian elimination

Forward elimination:

$$\begin{aligned} (E_2) - (E_1) \times \frac{1}{\epsilon} &\Rightarrow (1 - \frac{1}{\epsilon})x_2 = (2 - \frac{1}{\epsilon}) \\ x_2 &= \frac{2 - \frac{1}{\epsilon}}{1 - \frac{1}{\epsilon}} \approx 1 \\ x_1 &= \frac{1 - x_2}{\epsilon} = 0. \end{aligned} \quad \} \times$$

These sol<sup>n</sup> doesn't satisfy the equations, They aren't the correct solutions.

Test whether reordering eq<sup>n</sup> help or not:

Reorder the eq<sup>n</sup>:

$$\begin{aligned} x_1 + x_2 &= 2 \quad (E_1) \\ \epsilon x_1 + x_2 &= 1 \quad (E_2) \end{aligned}$$

Forward elimination:

$$(E_2) - \epsilon(E_1) \Rightarrow (1 - \epsilon)x_2 = 1 - 2\epsilon$$

Backward substitution:

$$x_2 = \frac{1 - 2\epsilon}{1 - \epsilon} \approx 1 \Rightarrow x_1 = 2 - x_2 = 1 \quad \checkmark$$

Reordering works!

Origin of the problem: Pivot/coefficient/diagonal entry being small

Reordering the equation makes diagonal entries not small.

What is the issue with diagonals being small?

- during forward elimination, division by diagonal element is required. Division by small number makes the resulting numbers very large. This blowing off can outweigh any other <sup>coeff-</sup> numbers in the equation making them insignificant & making it difficult to extract out the difference b/w coeff numbers  $\rightarrow$  leading to errors.

Reorder equations to reassign pivot rows  $\rightarrow$  pivotization.

- Gaussian elimination need not work well in cases where there are issues in diagonal dominance.

## Lecture 32: Tri-diagonal Matrix Algorithm (TDMA)

- has complexity:  $O(N)$
- also known as Thomas algorithm.

$$\begin{array}{ccc} i-1 & i & i+1 \\ \hline w & p & e \end{array}$$

$$a_p \phi_p = a_e \phi_e + a_w \phi_w + b$$
$$= \sum a_{nb} \phi_{nb} + b.$$

$$a_i \phi_i = b_i \phi_{i+1} + c_i \phi_{i-1} + d_i \quad \text{--- (1)}$$

$$\begin{array}{ccccccc} 1 & 2 & 3 & 4 & \dots & N \end{array}$$

$$\begin{bmatrix} \times & \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \times & \times \\ \times & \times & \times & \times & \times & \times \end{bmatrix} \begin{bmatrix} \phi_1 \\ \phi_2 \\ \phi_3 \\ \vdots \\ \phi_N \end{bmatrix} = \begin{bmatrix} w \\ \times \\ \times \\ \times \\ \times \\ \times \end{bmatrix}$$

Tri-diagonal matrix.

Instead of using 2 indices for storage of an element, use 1 index for storing entries related to three diagonals together.  $\rightarrow$  2D array storage system to linear storage system.

MEAS 7, 12

$$a_1 \phi_1 = b_1 \phi_2 + d_1 \quad (c_1 = 0)$$

$$\phi_1 = \frac{b_1 \phi_2 + d_1}{a_1} = f_1(\phi_2)$$

$$a_2 \phi_2 = b_2 \phi_3 + c_2 \phi_1 + d_2$$

$$\phi_2 = f_2(\phi_3)$$

$$\phi_3 = f_3(\phi_4) \quad \text{const } (c_3 = 0 \text{ no } \phi_4, \text{ it's just a const})$$

In general,

$$\phi_i = P_i \phi_{i+1} + Q_i \rightarrow \text{in the form of a linear function.}$$

Immediate step before:

$$\phi_{i-1} = P_{i-1} \phi_i + Q_{i-1} \quad (*)$$

Sub this recurrence formula in  $(*)$ :

$$a_i \phi_i = b_i \phi_{i+1} + c_i [P_{i-1} \phi_i + Q_{i-1}] + d_i$$

$$\Rightarrow (a_i - c_i P_{i-1}) \phi_i = b_i \phi_{i+1} + d_i + c_i Q_{i-1}$$

$$\phi_i = \frac{b_i}{a_i - c_i P_{i-1}} \phi_{i+1} + \frac{d_i}{a_i - c_i P_{i-1}} + \frac{c_i}{a_i - c_i P_{i-1}} Q_{i-1}$$

—  $(**)$

Compare  $(*)$  and  $(**)$  :-

$$\Rightarrow P_i = \frac{b_i}{a_i - c_i P_{i-1}}$$

$$Q_i = \frac{d_i + c_i Q_{i-1}}{a_i - c_i P_{i-1}}$$

where  $c_1 = 0$  ( $\because$  There is not  $\phi_0$ ) &  
 $b_N = 0$  ( $\because$  There is no  $\phi_{N+1}$ )

$$P_i = \frac{b_i}{a_i}, \quad Q_i = \frac{d_i}{a_i}$$

$$\phi_n = Q_n \quad (\because \phi_n = P_n \phi_{n+1} + Q_n)$$

$\therefore$  There is no  $\phi_{n+1}$ ,  
 $\phi_n = Q_n$ .

Above  $\uparrow$  stuff  $\rightarrow$  like  
 elimination.

$$\phi_i = P_i Q_{i+1} + Q_i \rightarrow \text{use to backward sub.}$$

Summary of TBMA:

Input  $a_i, b_i, c_i, d_i$

$$P_i = \frac{b_i}{a_i}, \quad Q_i = \frac{d_i}{a_i}$$

for  $i = 2, N$   
 forward elimination:  $P_i = \frac{b_i}{a_i - c_i P_{i-1}}; \quad Q_i = \frac{d_i + c_i Q_{i-1}}{a_i - c_i P_{i-1}}$   
 end

$$Q_N = \phi_N = Q_N$$

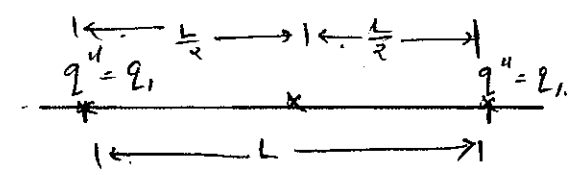
for  $i = N-1, i, i-1$   
 backward substitution:  $\phi_i = P_i \phi_{i+1} + Q_i$   
 end

$O(N)$  complexity!

Will TBMA always work?

Take an example:-

$g = 0$



Find steady state T distribution.

Steps:

→ FV discretization / FD discretization.

→ sol<sup>n</sup> of eq<sup>n</sup> by TDMA

Governing equation:

$$\frac{d}{dx} \left( k \frac{dT}{dx} \right) = 0.$$

If  $k$  const  $\rightarrow \frac{d^2 T}{dx^2} = 0.$

Applied to grid-point (2)  
Fb discretization:

$$\frac{T_3 + T_1 - 2T_2}{(L/2)^2} = 0$$

$$\Rightarrow T_2 = \frac{T_1 + T_3}{2} \rightarrow 2T_2 = T_1 + T_3.$$

For b/c at ①,

$$q'' = -k \frac{dT}{dx} \Big|_1 = k \left( \frac{T_1 - T_2}{(L/2)} \right)$$

$$T_1 = T_2 + \frac{q'' L}{2k}$$

$$\Rightarrow T_1 = T_2 + \frac{1}{2} a$$

BC at ③,

$$T_3 = T_2 - \frac{1}{2} a \text{ in a similar way.}$$

Physically obvious  $q''$  heat is transferred from higher to lower temperature.  
 $T_3 < T_2$

$$\text{and } a_1 T_1 = b_1 T_2 + d_1$$

$$a_1 = 1, b_1 = 1, d_1 = \alpha.$$

$$a_2 T_2 = b_2 T_3 + c_2 T_1 + d_2.$$

$$a_2 = 2, b_2 = 1, c_2 = 1, d_2 = 0.$$

$$a_3 T_3 = \frac{1}{2} c_3 T_1 + d_3.$$

$$a_3 = 1, c_3 = 1, d_3 = -\alpha.$$

$$P_1 = \frac{b_1}{a_1} = 1, Q_1 = \frac{d_1}{a_1} = \alpha.$$

$$P_2 = \frac{b_2}{a_2 - c_2 P_1}$$

$$= \frac{1}{2 - 1 \times 1} = \frac{1}{1}$$

$$Q_2 = \frac{d_2 + c_2 Q_1}{a_2 - c_2 P_1}$$

$$= \frac{0 + 1 \times \alpha}{1}$$

$$= \alpha$$

$$P_3 = \frac{b_3}{a_3 - c_3 P_2}$$

$$= \frac{0}{1 - 1 \times 1} = \frac{0}{0} \rightarrow \text{indeterminate.}$$

~~na~~

TDMA breaks down in

this case.

Why? Coeff matrix is singular (det=0).

$$\text{In } Ax = b$$

$$x = A^{-1} b,$$

inverse of A  $A^{-1}$

$$= \frac{\text{adj}(A)}{\det(A)}.$$

If  $\det(A) = 0 \rightarrow A^{-1}$  has a problem.

ill posed BVP.

Why? 2 flux conditions don't give additional information regarding the system.

~~It is~~ If ill posedness & physical unreality of the problem isn't detected, the mathematics of the problem will naturally reveal it.

## Lecture 33: Elimination Methods: Error Analysis

Prob Given for L-U factorization by Crout's method, following steps are to be executed.

$$l_{ii} = a_{ii} \text{ for } i \in [1, n]$$

$$l_{ij} = a_{ij} - \sum_{k=1}^{j-1} l_{ik} u_{kj} \text{ where}$$

$$u_{ij} = \frac{a_{ij}}{l_{ii}} \text{ for } j \in [2, n]$$

$$u_{ij} = \frac{1}{l_{ii}} \left[ a_{ij} - \sum_{k=1}^{i-1} l_{ik} u_{kj} \right] \text{ for } i, j \in [i+1, n]$$

Estimate the operational count if  $A_{N \times N}$ .

Numerical error  $\rightarrow$  combined effect of errors intrinsic to method and errors due to the machine on which the algo is performed.

— Errors intrinsic to elimination methods:—

{Terminology}

\* norm of a vector:

Say, we have a vector  $x$ ,

$$x = \{x_1, x_2, \dots, x_n\}$$

element of  $x \rightarrow x_i$

$$\|x\|_p = \left[ \sum_i |x_i|^p \right]^{1/p}$$

$$\underline{\text{Ex}} \quad x = \{1, -2, 3, -4\}$$

$$\|x\|_1 = |1| + |-2| + |3| + |-4| = 10$$

$$\|x\|_2 = (1^2 + 2^2 + 3^2 + 4^2)^{1/2} = \sqrt{30}$$

length of a vector.

$$\|x\|_\infty = \max |x_i| = 4$$

Norm of a matrix:

$\|A\|$ ?

Introduce a vector  $z$  and  $\|Az\|$ .

$$\|A\| \rightarrow \frac{\|Az\|}{\|z\|} \text{ s.t. } \|z\|=1$$

Prob:  $A = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix}$ . Find  $\|A\|_2$ .

$$Ax = Bb; \quad x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}, \text{ s.t. } \|x\|_2 = 1$$

$$Ax = \begin{bmatrix} 1 & 1 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} = \begin{bmatrix} x_1 + x_2 \\ x_1 \end{bmatrix}$$

$$\|Ax\|_2 = \sqrt{(x_1 + x_2)^2 + x_1^2}$$

subject to the constraint:

$$\|x\|_2 = 1 \Rightarrow \sqrt{x_1^2 + x_2^2} = 1 \quad \text{--- (1)}$$

$$\|Ax\|_2 = \sqrt{x_1^2 + x_2^2 + 2x_1x_2}$$

Using (1),  $\|Ax\|_2$  can be written in terms of either  $x_1$  or  $x_2 \rightarrow$  unambiguously.

B/c of that we say,  $\|A\|_2 = \max_{\|x\|_2=1} \|Ax\|_2$

Effectively find out

$$\max (x_1^2 + x_2^2 + 2x_1x_2), \text{ given}$$

$$x_2 = \sqrt{1 - x_1^2}$$

$$y = 2x_1^2 + 1 - x_1^2 + 2x_1\sqrt{1-x_1^2}$$

$$y = x_1^2 + 1 + 2x_1\sqrt{1-x_1^2}$$

$$\text{for max } y, \frac{dy}{dx} = 0$$

$$\rightarrow 2x_1 + 2\left[\sqrt{1-x_1^2} + \frac{x_1(-2x_1)}{2\sqrt{1-x_1^2}}\right] = 0$$

$$\Rightarrow x_1\sqrt{1-x_1^2} + \left[1-x_1^2 - x_1^2\right] = 0$$

$$\Rightarrow x_1\sqrt{1-x_1^2} = 2x_1^2 - 1$$

$$x_1^2(1-x_1^2) = 4x_1^4 - 4x_1^2 + 1$$

$$\Rightarrow 5x_1^4 - 5x_1^2 + 1 = 0$$

$$x_1^2 = \frac{5 \pm \sqrt{5}}{2 \times 5} = \frac{1}{2} \pm \frac{1}{2\sqrt{5}}$$

$$x_1^2 = \frac{1}{2} \pm \frac{\sqrt{5}}{10}$$

Find condition for maxima, out of the two possible roots.

That gives  $x_1^2 \rightarrow x_1 \rightarrow x_2 \rightarrow$   
elements of  $Ax$ .

## Lecture 34: Elimination method: Error Analysis (Contd).

$$y = 2x_1^2 + 1 - x_1^2 + 2x_1\sqrt{1-x_1^2}$$

$$y = x_1^2 + 1 + 2x_1\sqrt{1-x_1^2}$$

$$\text{For max } y, \frac{dy}{dx} = 0$$

$$\Rightarrow 2x_1 + 2\sqrt{1-x_1^2} + 2x_1 \times \frac{1(-2x_1)}{2\sqrt{1-x_1^2}} = 0$$

$$\Rightarrow x_1\sqrt{1-x_1^2} + 1-x_1^2 - x_1^2 = 0$$

$$\Rightarrow x_1\sqrt{1-x_1^2} = 2x_1^2 - 1$$

$$x_1^2(1-x_1^2) = 4x_1^4$$

Norms with special meaning for matrices:  
1-norm &  $\infty$ -norm.

1-norm  $\|A\|_1 \rightarrow$  column sum norm.

$$\max_j \sum_i |A_{ij}|$$

$\infty$ -norm  $\|A\|_\infty \rightarrow$  maximum row sum norm.

$$\max_i \sum_j |A_{ij}|$$

$$\begin{bmatrix} 1 & 4 & 6 \\ -5 & -2 & 1 \\ -8 & -1 & 3 \end{bmatrix} \quad \|A\|_1 \rightarrow \max(14, 7, 10) = 14$$

$$\|A\|_\infty \rightarrow \max(11, 9, 12) = 12$$

A

Some important properties of matrix norms:

$$1. \|kA\| = |k| \|A\|$$

$$2. \|A+B\| \leq \|A\| + \|B\|$$

$$3. \|AB\| \leq \|A\| \|B\| \rightarrow \max_{\|x\|=1} \|ABx\| = \max_{\|x\|=1} \|A(Bx)\|$$

$$= \max_{\|x\|=1} \|A\| \|Bx\| = \max_{\|x\|=1} \|A\| \max_{\|y\|=1} \|Bx\|$$

$$= \max_{\|x\|=1} \|A\| \max_{\|y\|=1} \|B\| = \|A\| \|B\|$$

~~Let~~  $\|AB\| = \max \frac{\|Ay\|}{\|y\|} \max \frac{\|Bx\|}{\|x\|}$   
 $\leq \|A\| \cdot \|B\|$

Error analysis of elimination methods:

Consider  $Ax=b$

Let  $x_{approx}$  be the approximate numerical sol?

$$\underbrace{A(x - x_{approx})}_{e(\text{error})} = \underbrace{b - Ax_{approx}}_{r(\text{residue})}$$

In error analysis, we look for an upper bound of error without actually knowing the exact solution. It involves estimation, not exact quantification.

Say we have  $e_1 \sim 10^{-10}$  and  $e_2 \sim 10^{-5}$ . In which case error is more/less? We cannot really tell as it depends on the  $x$  itself.

What we require is relative error, not absolute error.

Relative error indicator:  $\frac{\|e\|}{\|x\|}$

$$\|Ax\| \leq \|A\| \cdot \|x\|$$

$$\|b\| \leq \|A\| \cdot \|x\|$$

$$\|x\| \geq \frac{\|b\|}{\|A\|}$$

$$x = A^{-1}b$$

$$\|x\| = \|A^{-1}b\| \leq \|A^{-1}\| \cdot \|b\|$$

~~Let~~ If you have  $Ae=r$ ,

$$\|e\| \geq \frac{\|r\|}{\|A\|} \quad \text{and}$$

$$\|e\| \leq \|A^{-1}\| \cdot \|r\|$$

Bounds:

$$\frac{\|e\|_{min}}{\|x\|_{max}} \leq \frac{\|e\|}{\|x\|} \leq \frac{\|e\|_{max}}{\|x\|_{min}}$$

$$\frac{\|r\|}{\|A\| \cdot \|A^{-1}\| \cdot \|b\|} \leq \frac{\|e\|}{\|x\|} \leq \frac{\|A^{-1}\| \cdot \|r\| \cdot \|A\|}{\|b\|}$$

Focus on upper bound (for conservative approach):

$$\frac{\|e\|}{\|x\|} \leq \boxed{\|A^{-1}\| \|A\|} \cdot \frac{\|r\|}{\|b\|}$$

Conclusion: Even with small residual, the relative error may be large if  $\|A^{-1}\| \cdot \|A\|$  is large. Therefore the largeness of  $\|A^{-1}\| \|A\|$  determines the condition for accuracy of the system of equations one is solving.

Condition number:  $C(A) = \|A\| \cdot \|A^{-1}\|$

Large  $C(A) \Rightarrow$  even a small  $\frac{\|r\|}{\|b\|}$  can lead to large  $\frac{\|e\|}{\|x\|}$ .

$$\|AA^{-1}\| \leq \|A\| \cdot \|A^{-1}\|$$

$$I \leq \|A\| \cdot \|A^{-1}\|$$

$$\Rightarrow C(A) \geq 1$$

Closer to 1, is better.

$C(A) \rightarrow$  very critical parameter for estimating error.

## Lecture 35: Iteration Methods.

Example problem:

Given,  $A = \begin{bmatrix} 2 & 1 \\ 2 & 1.01 \end{bmatrix}$ .

Find  $CCA$ .

$$CCA = \|A\| \cdot \|A^{-1}\|$$

$$A^{-1} = \begin{bmatrix} 1.01 & -1 \\ -2 & 2 \end{bmatrix} \cdot \frac{1}{(2 \cdot 0.02 - 2)}$$

$$= \frac{1}{0.02} \begin{bmatrix} 1.01 & -1 \\ -2 & 2 \end{bmatrix} = \begin{bmatrix} 50.05 & -50 \\ -100 & 100 \end{bmatrix}$$

$$\|A\|_{\infty} = 3.01$$

$$\|A^{-1}\|_{\infty} = 200$$

$$CCA = \|A\|_{\infty} \|A^{-1}\|_{\infty}$$

$$= 602 \rightarrow \text{quite large}$$

$\Rightarrow$  ill conditioned system.

Source of this largeness: large smallness of the determinant.

Concl: Inference: Smaller the determinant, greater the chances of ~~of~~ larger condition number & ill-condition of the system.

## Iteration methods:

Basic philosophy: start with an initial guess for solution & iterate on it till you get a final solution that converges.

Say, we have equations:

$$5x_1 + x_2 = 6 \quad \text{--- (E}_1\text{)}$$

$$x_1 + 5x_2 = 6 \quad \text{--- (E}_2\text{)}$$

$$x_1 = \frac{6 - x_2}{5} \quad x_2 = \frac{6 - x_1}{5}$$

Make an initial guess:

$$x_1^{(0)} = 0, \quad x_2^{(0)} = 0.$$

Now try to update on this initial guess:

Make an iterative formula out of the given system:

$$x_1^{(k+1)} = \frac{6 - x_2^{(k)}}{5}, \quad x_2^{(k+1)} = \frac{6 - x_1^{(k)}}{5}$$

$\Rightarrow$  If we write iteration formula in this manner, it is called Jacobi's method.

Jacobi's iteration scheme.

$$S-1 \quad x_1^{(1)} = \frac{6 - 0}{5} = \frac{6}{5} \quad x_2^{(1)} = \frac{6 - 0}{5} = \frac{6}{5}$$

$$S-2 \quad x_1^{(2)} = \frac{6 - x_2^{(1)}}{5} = \frac{6 - (6/5)}{5} = \frac{24}{25}$$

$$x_2^{(2)} = \frac{6 - x_1^{(1)}}{5} = \frac{6 - (6/5)}{5} = \frac{24}{25}$$

S-3

S-4

S-5

$\rightarrow x_1 \rightarrow 1, \quad x_2 \rightarrow 1$  convergence.



Convergence  $\Rightarrow$  result b/w the current & previous steps doesn't differ substantially.

(within some tolerance)

To update iterations with a faster rate,

use

$$x_1^{(k+1)} = \frac{6 - x_2^k}{5}$$

$$x_2^{(k+1)} = \frac{6 - x_1^{(k+1)}}{5}$$

(instead of  $x_1^k$ , we use more updated version).

If you do that, then this becomes

Gauss-Seidel method.

$$x_1^{(1)} = \frac{6 - x_1^{(0)}}{5} = 6/5$$

$$x_2^{(1)} = \frac{6 - x_1^{(1)}}{5} = \frac{6 - (6/5)}{5} = \frac{24}{25}$$

$$x_1^{(2)} = \frac{6 - x_1^{(1)}}{5} = \frac{6 - 24/25}{5} = \frac{126}{125}$$

$$x_2^{(2)} = \frac{6 - x_1^{(2)}}{5} = \frac{6 - 126/125}{5} = \frac{624}{625}$$

Here within few steps, ~~the~~ sol<sup>n</sup> is converged very fast.

• Where is the guarantee that the scheme will converge or not?

## Lecture 36: Generalized Analysis of Iterative Method.

Generalized analysis of the iterative

(41)

methods:

$$a_{11}x_1 + a_{12}x_2 + a_{13}x_3 + \dots + a_{1n}x_n = b_1$$

$$a_{21}x_1 + a_{22}x_2 + a_{23}x_3 + \dots + a_{2n}x_n = b_2$$

$$\vdots$$

$$a_{n1}x_1 + a_{n2}x_2 + a_{n3}x_3 + \dots + a_{nn}x_n = b_n$$

$$[A][x] = [b]$$

$$x_1^{(k+1)} = \frac{b_1 - (a_{12}x_2^{(k)} + a_{13}x_3^{(k)} + \dots + a_{1n}x_n^{(k)})}{a_{11}}$$

Jacobi's method:

$$x_1^{(k+1)} = \frac{b_1 - (a_{12}x_1^{(k)} + a_{13}x_3^{(k)} + \dots + a_{1n}x_n^{(k)})}{a_{11}}$$

Gauss-Seidel:

$$x_2^{(k+1)} = \frac{b_2 - (a_{21}x_1^{(k+1)} + a_{23}x_3^{(k)} + \dots + a_{2n}x_n^{(k)})}{a_{22}}$$

$$[A][x] = [b]$$



$$[L] + [D] + [U]$$

lower diagonal  
Alor.

upper  
Alor.

Don't confuse with LU decomposition

$$L = \begin{bmatrix} a_{21} & & & 0 \\ a_{31} & a_{32} & & \\ a_{41} & a_{42} & a_{43} & \\ & a_{n1} & a_{n2} & \dots & a_{nn-1} \end{bmatrix}$$

$$D = \begin{bmatrix} a_{11} & & & 0 \\ & a_{22} & & \\ & & a_{33} & \\ & & & \ddots & \\ & & & & a_{nn} \end{bmatrix}$$

$$U = \begin{bmatrix} & a_{12} & a_{13} & \dots & a_{1n} \\ & & a_{23} & a_{24} & \dots & a_{2n} \\ & & & a_{34} & a_{35} & \dots & a_{3n} \\ & & & & \ddots & & \\ & & & & & & a_{nn} \end{bmatrix}$$

$$\text{Jacobi: } x_i^{(k+1)} = \frac{b_i - (a_{i1}x_1^{(k)} + a_{i2}x_2^{(k)} + \dots + a_{in}x_n^{(k)})}{a_{ii}}$$

Gauss-Seidel:

$$x_i^{(k+1)} = \frac{b_i - (a_{i1}x_1^{(k+1)} + a_{i2}x_2^{(k)} + \dots + a_{in}x_n^{(k)})}{a_{ii}}$$

In place of  $Ax=b$ ,

$$(L+D+U)x=b.$$

$$\text{Jacobi: } Dx^{(k+1)} + (L+U)x^{(k)} = b.$$

$$x^{(k+1)} = -D^{-1}(L+U)x^{(k)} + D^{-1}b.$$

$$x^{(k+1)} = Mx^{(k)} + C,$$

$$\text{where } M = -D^{-1}(L+U)$$

$$C = D^{-1}b$$

Gauss-Seidel:

$$Dx^{(k+1)} + Lx^{(k+1)} + Ux^{(k)} = b.$$

$$(L+D)x^{(k+1)} = -Ux^{(k)} + b$$

$$x^{(k+1)} = -(L+D)^{-1}Ux^{(k)} + (L+D)^{-1}b.$$

$$x^{(k+1)} = Mx^{(k)} + C$$

$$M = -(L+D)^{-1}U$$

$$C = (L+D)^{-1}b$$

$$x^{(k+1)} = Mx^{(k)} + C.$$

Say,  $x^*$  is actual sol<sup>n</sup>

$$x^* = Mx^* + C.$$

$$x^{(k+1)} - x^* = M(x^{(k)} - x^*)$$

error in the  $(k+1)^{\text{th}}$  step.

$$e^{(k+1)} = Me^{(k)}$$

$$e^{(1)} = Me^{(0)}$$

$$e^{(2)} = Me^{(1)}$$

$$= M^2 e^{(0)}$$

$$\vdots$$

$$e^{(k)} = M^k e^{(0)}$$

$$\frac{\|e^{(k)}\|}{\|e^{(0)}\|} < 1 \rightarrow \text{requirement for convergence.}$$

$$\Rightarrow \|M^k\| < 1$$

Not easy for raw computation of  $M^k$ .

Use eigenvalues & eigenvectors of  $M$  for ease of representation.

Let  $\lambda_i$  &  $v_i$  be correspondingly the eigenvalues & eigenvectors of  $M$ .

$$e^0 = a_1 v_1 + a_2 v_2 + a_3 v_3 + \dots + a_n v_n.$$

(eigenvalues are arranged in a way that

$$|\lambda_1| > |\lambda_2| > |\lambda_3| \dots > |\lambda_n|)$$

$$Me^0 = a_1 Mv_1 + a_2 Mv_2 + a_3 Mv_3 + \dots + a_n Mv_n$$

$$Mv_1 = \lambda_1 v_1; Mv_2 = \lambda_2 v_2; \dots; Mv_n = \lambda_n v_n.$$

$$M^0 e^0 = a_1 d_1 M V_1 + a_2 d_2 M V_2 + \dots + a_n d_n M V_n.$$

$$= a_1 d_1^2 V_1 + a_2 d_2^2 V_2 + \dots + a_n d_n^2 V_n$$

$$M^k e^0 = a_1 d_1^k V_1 + a_2 d_2^k V_2 + \dots + a_n d_n^k V_n.$$

Compare leading order term of  $M^k e^0$  with that of  $e^0$ .

$$\Rightarrow |\lambda_i|^k < 1 \rightarrow \frac{\|e^k\|}{\|e^0\|}$$

$$= \frac{\|a_1 d_1^k V_1 + \text{LOT}\|}{\|a_1 d_1 V_1 + \text{LOT}\|}$$

$$\sim \frac{\|a_1 d_1^k V_1\|}{\|a_1 d_1 V_1\|}$$

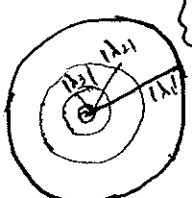
$$\sim |\lambda_1|^k$$

$|d_{\max}| \rightarrow$  spectral radius of convergence.

In certain physical problems, eigenvalues represents the natural frequencies of the system. Something to do with frequency  $\rightarrow$  "spectral" radius

For convergence

$$\left\{ \begin{array}{l} \text{Given } |\lambda_1| > |\lambda_2| > |\lambda_3| \dots \\ |\lambda_1| = |d_{\max}| \end{array} \right\}$$



$\frac{1}{2}$  radius of the 'biggest circle' should be less than 1.

Sufficient condition for convergence:

(If that condition is satisfied, you'll definitely satisfy convergence. But you may also have convergence without satisfying that condition).

Rate of convergence:

Requirement: no. of iterations to converge.

Say we require 'm' decimal accuracy.

$$\text{Then, } \frac{\|e^k\|}{\|e^0\|} < 10^{-m}$$

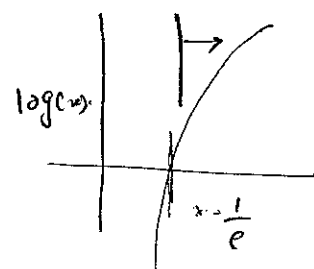
$$e^k < 10^{-m}$$

$$k \log_{10} \rho < -m$$

$$m < k \log_{10} \left(\frac{1}{\rho}\right)$$

$$k > \frac{m}{\log_{10} \left(\frac{1}{\rho}\right)} \rightarrow R = \text{rate of convergence.}$$

### Lecture 37: Further discussion on Iterative Methods.



Smaller spectral radius  $\Rightarrow$  better rate of convergence.

$$M V = \lambda V$$

$$\|M V\| \leq \|M\| \cdot \|V\|$$

$$\downarrow$$

$$|\lambda| \|V\| \leq \|M\| \cdot \|V\|$$

$$\Rightarrow |\lambda| \leq \|M\| \Rightarrow \rho \leq \|M\|$$

Spectral radius upper bound.

Which norm to choose?

$$\text{An estimate of } \rho \rightarrow \max [\|M\|_1, \|M\|_\infty]$$

• Sufficient condition for convergence:

$$\max(\|M\|_1, \|M\|_\infty) \leq 1$$

Jacobi's method:

$$M = -D^{-1}(L+U)$$

$$\begin{bmatrix} a_{11} & & & \\ & a_{22} & & \\ & & \ddots & \\ & & & a_{nn} \end{bmatrix} \rightarrow \begin{bmatrix} a_{21} & a_{22} & & \\ a_{31} & a_{32} & & \\ a_{41} & a_{42} & \ddots & \\ a_{n1} & a_{n2} & \ddots & a_{nn} \end{bmatrix}$$

$D$                        $L$

$$\begin{bmatrix} a_{12} & a_{13} & \dots & a_{1n} \\ & a_{23} & a_{24} & \dots & a_{2n} \\ & & a_{34} & \dots & a_{3n} \\ & & & \ddots & \\ & & & & a_{n-1,n} \end{bmatrix}$$

$U$

$$\|M\|_R = \|M\|_\infty \rightarrow \sum_{i \neq j} \frac{|a_{ij}|}{|a_{ii}|} \rightarrow$$

Should be  $\leq 1$  to satisfy sufficient condition for convergence.

$$\downarrow$$

$$\sum \frac{|a_{ij}|}{|a_{ii}|} \leq 1$$

for 1D system  $\rightarrow$  TDN

2D  $\rightarrow$  Pentadiagonal systems

3D  $\rightarrow$  7-diag systems

But matrix is generally sparse

(sparse matrices).

Scarborough Criteria for sufficient condition for convergence in Gauss-Seidel method.

$$\sum \frac{|a_{nb}|}{|a_p|} \leq 1 \text{ for all eqns}$$

$$< 1 \text{ for at least one eqn.}$$

Lecture 38: Illustrative Examples of Iterative methods

Ex 1-D steady state heat conduction in a rod with uniform  $k$ ,  $S=0$

$$\begin{array}{c} \xrightarrow{\quad} \\ \text{1} \quad \quad \quad \text{2} \quad \quad \quad \text{3} \end{array} \quad q'' \text{ (given)}$$

gde:  $\frac{d^2 T}{dx^2} = 0$

$$\text{CD: } \frac{T_2 + T_1 - 2T_2}{\Delta x^2} = 0$$

$$\Rightarrow T_2 = \frac{T_1 + T_3}{2}$$

For gp1,  $q'' = -k \frac{(T_2 - T_1)}{\Delta x}$

$$T_1 = T_2 + \left( \frac{q'' \Delta x}{k} \right)^c$$

For gp3,  $q'' = -k \frac{(T_3 - T_2)}{\Delta x}$

$$T_3 = T_2 + \left( \frac{q'' \Delta x}{k} \right)^c$$

$$Eq 1 \rightarrow T_1 = T_2 + T_3 \rightarrow \sum \frac{|a_{nb}|}{|a_p|} = 1$$

$$Eq 2 \rightarrow 2T_2 = T_1 + T_3 \rightarrow \sum \frac{|a_{nb}|}{|a_p|} = \frac{1}{2} < 1$$

$$Eq 3 \rightarrow T_3 = T_2 - T_1 \rightarrow \sum \frac{|a_{nb}|}{|a_p|} = \frac{1}{1} = 1$$

Ill-posed problem: we need at least Temp specified at a boundary.

In all cases  $\sum \frac{|a_{nb}|}{|a_p|} \leq 1$ . In none of

The cases is it  $< 1$ . Can't satisfy

Scarborough's criteria.

So specify at gp1  $T_{gun} = T^*$  instead of  $2''$

Then for gp2,

$$\begin{aligned} 2T_2 &= T_1 + T_3 \\ \Rightarrow 2T_2 &= T_1^* + T_3 \end{aligned} \quad \left\{ \begin{array}{l} Eq 2 \text{ no more valid} \\ \text{as } 2'' \text{ at gp1} \\ \text{not known} \end{array} \right.$$

$\therefore T_1^*$  is already known mathematically it is no longer a neighbour to  $T_2$ . only  $T_3$  is the neighbour.

$\therefore$  for modified eqn 2,

$$\sum \frac{|a_{nb}|}{|a_p|} = \frac{1}{2} \neq 1 < 1$$

With redefinition of the problem,

Scarborough criterion is satisfied.

Ex Consider the system,

$$2x_1 + 3x_2 + 10x_3 = 10$$

$$5x_1 - 2x_2 + 2x_3 = 5$$

$$x_1 + 10x_2 + 5x_3 = 6$$

Q: Is it possible to follow an iterative method (say Jacobi iteration) for the above system with guaranteed convergence? If yes, what is the estimated no. of iterations to achieve 4 decimal accuracy?

$$\sum_{j \neq i} \frac{|a_{ij}|}{|a_{ii}|} \leq 1 \rightarrow \text{representative of the diagonal dominance.}$$

Diagonal terms are dominating over the sum of the off diagonal terms.

$$R_1 \rightarrow \frac{|3| + |10|}{|2|} \neq 1 \text{ doesn't satisfy sufficient condition.}$$

How to somehow satisfy the suff cond?

Find an eqn from the set of eqns where the first term's coeff is largest. Here eq-2 has met that criteria. So swap eq-1 with eq-2 [eq-1\* = eq-2].

$$\text{Now } R_1^* \rightarrow \frac{|-2| + |2|}{|5|} = \frac{4}{5} \leq 1$$

sufficient condition is satisfied!

11<sup>try</sup> make eq-3  $\rightarrow$  eq-2\* & eq-1  $\rightarrow$  eq-3\*

Thus reordering equations can help satisfy the suff. cond.

$$\left. \begin{array}{l} \text{eq-1}^* \quad \frac{|2| + |2|}{|5|} = \frac{4}{5} \\ \text{eq-2}^* \quad \frac{|5| + |1|}{|10|} = \frac{6}{10} \\ \text{eq-3}^* \quad \frac{|2| + |3|}{|10|} = \frac{5}{10} \end{array} \right\} \begin{array}{l} \text{Row sum norm} = 0.8 \\ \text{as} \\ \text{max} = 0.8 \end{array}$$

$$\rho = \max(\|M\|_R, \|M\|_C)$$

Column sum norm  $\|M\|_C$

$$\left. \begin{array}{l} \begin{array}{ccc} 5 & -2 & 2 \\ 1 & 10 & 5 \\ 2 & 8 & 10 \end{array} \\ \begin{array}{l} \text{col-1} = 3/5 \\ \text{col-2} = 5/10 \\ \text{col-3} = 7/10 \end{array} \end{array} \right\} \begin{array}{l} \text{max} = 0.7 \\ \|M\|_C = 0.7 \end{array}$$

$$\rho = \max(0.8, 0.7) = 0.8$$

$$\therefore k = \log_{10} \left( \frac{1}{\rho} \right) = \log_{10}(1.25)$$

Ex. For a linear system of size  $N$ , the

$i^{\text{th}}$  eigenvalue of the Jacobi iteration matrix  $[M = -D^{-1}(L+U)]$  is given by

$$\lambda_i = \cos \left( \frac{i\pi}{N+1} \right) \text{ where } i = 1, 2, \dots, N.$$

It is also known that for sufficiently large values of  $N$ ,  $\cos \frac{\pi}{N+1} \approx \frac{1 - \pi^2}{2(N+1)^2} \approx \exp \left( \frac{-\pi^2}{2N^2} \right)$ .

If the size of the coefficient matrix changes from size  $10^3 \times 10^3$  to  $10^4 \times 10^4$ , to what

proportion would the total no. of iterations expected to achieve a desired level of accuracy will decrease?

for  $(\lambda_{\max})$ , where  $i=1$ ,

$$\cos \left( \frac{\pi}{N+1} \right) \approx \exp \left( \frac{-\pi^2}{2N^2} \right)$$

$$k \rightarrow \frac{M}{R}$$

$$\frac{k_2}{k_1} \approx \frac{R_1}{R_2} \quad \left( \text{keeping same level of accuracy, but diff coeff matrix} \right)$$

$$\rightarrow \frac{k_2}{k_1} = \frac{-\log_{10}(|\lambda_1|_{\max})}{-\log_{10}(|\lambda_1|_{\max})} \rightarrow N=10^3$$

$$\approx 10^{-2}$$

$$k_2 = 10^{-2} k_1$$

### Lecture 39: Gradient Search Based Methods

f: function

Grad  $f \rightarrow$  represents maximum rate of change of  $f$ .

Say,  $f = \frac{1}{2} x^T A x$  Obj: Solve  $Ax = b$ .

$$-b^T x + c,$$

$c$  arbitrary const.

Two restrictions:

- $A$  symmetric
- $A$  pos definite.

Next obj, find cond. for min  $f = ?$

$$\min f \Rightarrow \nabla f = 0.$$

$f =$

$$\frac{1}{2} [x_1 \ x_2 \ \dots \ x_n] \begin{bmatrix} a_{11} & a_{12} & a_{13} & \dots & a_{1n} \\ a_{21} & a_{22} & a_{23} & \dots & a_{2n} \\ \vdots & \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & a_{n3} & \dots & a_{nn} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_n \end{bmatrix}$$

$$- [b_1 \ b_2 \ b_3 \ \dots \ b_n] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_n \end{bmatrix} + C$$

$$\nabla f = \begin{bmatrix} \frac{\partial f}{\partial x_1} \\ \frac{\partial f}{\partial x_2} \\ \vdots \\ \frac{\partial f}{\partial x_n} \end{bmatrix}$$

$$f = \frac{1}{2} [x_1 \ x_2 \ \dots \ x_n] \begin{bmatrix} a_{11}x_1 + a_{12}x_2 + \dots + a_{1n}x_n \\ a_{21}x_1 + a_{22}x_2 + \dots + a_{2n}x_n \\ \vdots \\ a_{n1}x_1 + a_{n2}x_2 + \dots + a_{nn}x_n \end{bmatrix}$$

$$- [b_1 \ b_2 \ \dots \ b_n] \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + C$$

$$f = \frac{1}{2} x_1 (a_{11}x_1 + \dots + a_{1n}x_n) + \frac{1}{2} x_2 (a_{21}x_1 + \dots + a_{2n}x_n) + \dots + \frac{1}{2} x_n (a_{n1}x_1 + \dots + a_{nn}x_n)$$

$$- (b_1x_1 + b_2x_2 + \dots + b_nx_n) + C$$

$$\frac{\partial f}{\partial x_1} = \frac{1}{2} (a_{11}x_1 + \dots + a_{1n}x_n) + \frac{1}{2} x_1 (a_{11}) + \frac{1}{2} a_{21}x_2 + \frac{1}{2} a_{31}x_3 + \dots + \frac{1}{2} a_{n1}x_n - b_1$$

$$= \frac{1}{2} (a_{11}x_1 + \dots + a_{1n}x_n) + \frac{1}{2} (a_{11}x_1 + \dots + a_{n1}x_n) - b_1$$

$$\therefore A \text{ is symmetric, } a_{ij} = a_{ji}$$

$$\Rightarrow \frac{\partial f}{\partial x} = (a_{11}x_1 + a_{12}x_2 + \dots + a_{1n}x_n) - b_1$$

$$= \begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} - b_1$$

$$\frac{\partial f}{\partial x} = \begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} - b_1$$

$$\Rightarrow \nabla f = \begin{bmatrix} \frac{\partial f}{\partial x_1} \\ \frac{\partial f}{\partial x_2} \\ \vdots \\ \frac{\partial f}{\partial x_n} \end{bmatrix}$$

$$= \begin{bmatrix} a_{11} & a_{12} & \dots & a_{1n} \\ a_{21} & a_{22} & \dots & a_{2n} \\ \vdots & \vdots & \ddots & \vdots \\ a_{n1} & a_{n2} & \dots & a_{nn} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} - \begin{bmatrix} b_1 \\ b_2 \\ \vdots \\ b_n \end{bmatrix}$$

$$= Ax - b$$

$$\nabla f = 0$$

$$\Rightarrow Ax - b = 0$$

$$\Rightarrow Ax = b$$

Getting a sol<sup>n</sup>  $Ax = b$  is as good as  
extremizing  $f = \frac{1}{2} x^T Ax - b^T x + c$ .

we're actually doing minimization of  $f$  &  
not maximization

How to show it? why minimization?

$$f(x) = \frac{1}{2} x^T Ax - b^T x + c$$

Let  $x = x + e$

if  $f(x+e) < f(x)$   $\forall e$ , then  $f(x)$  is  
minimum

$$f(x+e) = \frac{1}{2} (x+e)^T A (x+e) - b^T (x+e) + c$$

$$= \frac{1}{2} (x^T Ax + e^T Ax + x^T Ae + e^T Ae) - b^T x - b^T e + c$$

$$= \left[ \frac{1}{2} (x^T A x) - b^T x + c \right] + e^T A x$$

$$\begin{aligned} \bullet x^T A e &= (x^T A e)^T \\ &\text{(transpose of scalar is scalar itself)} \\ &= (A e)^T (x^T)^T \\ &= e^T A^T x \\ &= e^T A x \quad (A \text{ symmetric; } A^T = A) \end{aligned}$$

$$\begin{aligned} \bullet b^T e &= (b^T e)^T \quad (\text{again, scalar}) \\ &= e^T (b^T)^T \\ &= e^T b \end{aligned}$$

$$\begin{aligned} \bullet f(x+e) &= f(x) + e^T A x - e^T b + \frac{1}{2} e^T A e \\ &= f(x) + e^T (A x - b) + \frac{1}{2} e^T A e \end{aligned}$$

$$\text{B/c } A x = b, \quad e^T (A x - b) = 0.$$

$$\begin{aligned} &= f(x) + e^T (A x - b) + \frac{1}{2} e^T A e \\ &\geq 0 \text{ for arbitrary } e \\ &(\because A \text{ is positive definite}) \end{aligned}$$

$$\Rightarrow f(x+e) \geq f(x).$$

$\Rightarrow f(x)$  is a minimum.

Many gradient search methods!

1. Steepest descent method.

Solve  
Ex  $A x = b$ , where  $A = \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix}$ ,  $b = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$

$$\text{Ans } f = \frac{1}{2} x^T A x - b^T x + c$$

$$f = \frac{1}{2} \begin{bmatrix} x_1 & x_2 \end{bmatrix} \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} - \begin{bmatrix} 1 & 1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + c$$

$$\begin{aligned} &= \frac{1}{2} \begin{bmatrix} x_1 & x_2 \end{bmatrix} \begin{bmatrix} x_1 \\ 2x_2 \end{bmatrix} - [x_1 + x_2] + c \\ &= \frac{1}{2} x_1^2 + x_2^2 - x_1 - x_2 + c \end{aligned}$$

What does  $f=0$  represent?

$$\begin{aligned} &\frac{1}{2} (x_1^2 - 2x_1 + \frac{1}{2}) + (x_2^2 - x_2 + \frac{1}{2}) \\ &\quad + c = 0 \\ &\quad \quad \quad -\frac{3}{4} \end{aligned}$$

$$= \frac{(x_1 - 1)^2}{2} + \frac{(x_2 - \frac{1}{2})^2}{1} = \frac{3}{4} - c$$

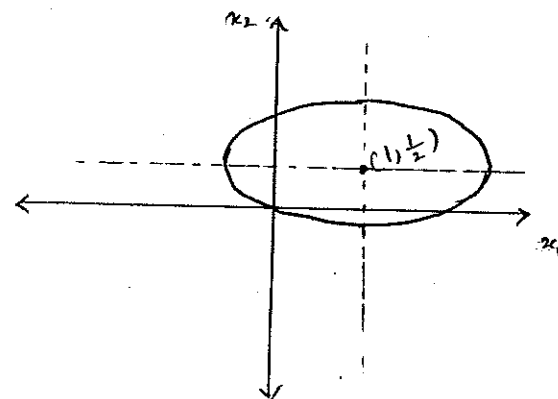
If  $f$  satisfies  $(0,0)$ ,

$$\begin{aligned} \text{Then } \frac{1}{2} + \frac{1}{4} &= \frac{3}{4} - c \\ \Rightarrow c &= 0 \end{aligned}$$

$\therefore$  choice of  $c$  is arbitrary, we can choose our curve to pass through a pt s.t.  $c=0$ .

$$\frac{(x_1 - 1)^2}{2} + \frac{(x_2 - \frac{1}{2})^2}{1} = \frac{3}{4}$$

$$\frac{(x_1 - 1)^2}{(\sqrt{3/2})^2} + \frac{(x_2 - \frac{1}{2})^2}{(\sqrt{3/2})^2} = 1.$$





Say we start with  $(x_1, x_2) = (0, 10)$  & try to reach the actual solution. We move  $\alpha$  along the gradient direction.

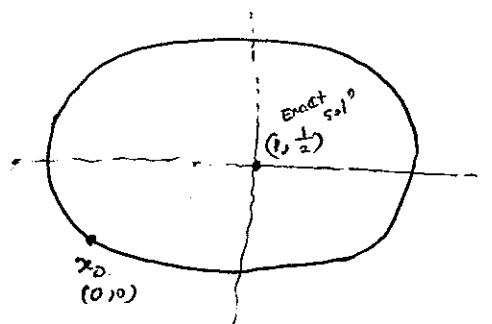
$$\nabla f = Ax - b \\ = -r_1$$

After moving some distance. We again stop & move in the gradient direction. The objective then becomes <sup>to find</sup> how far much to travel in each segment.

### Lecture 40: Steepest descent method

~~f(x)~~ Steepest descent method:

- move along direction of maximum rate of change.



$$A = \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \quad b = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$$

$$\nabla f = Ax - b \\ = -r_1$$

$$x^1 = x^0 + \alpha_0 r_0$$

$$r_0 = \nabla f|_{x^0}$$

$\alpha_0 \rightarrow$  how much to move in that direction.

$r_0 \rightarrow$  tells direction in which you're moving

$$f(x^1) = \frac{1}{2} (x^0 + \alpha_0 r_0)^T A (x^0 + \alpha_0 r_0)$$

$$- b^T (x^0 + \alpha_0 r_0) + c$$

For  $f$  to be min,  $\frac{\partial f}{\partial \alpha_0} = 0$ .

$$\frac{1}{2} r_0^T A (x^0 + \alpha_0 r_0) + \frac{1}{2} (x^0 + \alpha_0 r_0)^T A r_0$$

$$- b^T r_0 = 0$$

$$\alpha_0 r_0^T A r_0 + \frac{1}{2} r_0^T A x^0 + \frac{1}{2} x^0^T A r_0 - b^T r_0 = 0$$

$$x^0^T A r_0 = (x^0^T A r_0)^T$$

$$= (A r_0)^T (x^0)^T$$

$$= r_0^T A^T x^0$$

$$= r_0^T A x^0 \quad (A^T = A)$$

$$b^T r_0 = (b^T r_0)^T = r_0^T b$$

$$\alpha_0 r_0^T A r_0 + r_0^T (A x^0 - b) = 0$$

$$\Rightarrow \alpha_0 = \frac{r_0^T r_0}{r_0^T A r_0} \quad \text{--- ①}$$

$$x^1 = x^0 + \alpha_0 r_0$$

$$\alpha_1 = \frac{r_1^T r_1}{r_1^T A r_1}$$

$$x^2 = x^1 + \alpha_1 r_1$$

Relation b/w directions of  $r_0$  &  $r_1$ :

$$r_0^T r_1 = r_0^T [b - A x_1]$$

$$= r_0^T [b - A (x^0 + \alpha_0 r_0)]$$

$$= r_0^T [(b - A x^0) - \alpha_0 A r_0]$$

$$= r_0^T r_0 - \alpha_0 r_0^T A r_0$$

$$= 0 \quad \rightarrow \text{Using ①}$$

$\Rightarrow r_0$  and  $r_1$  are orthogonal to each other.

$\Rightarrow$  we'll be moving mutually perpendicular directions till we reach the solution.

$$r_0 = b - Ax^0 = \begin{bmatrix} 1 \\ 1 \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \begin{bmatrix} 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ 1 \end{bmatrix}$$

$$r_0^T r_0 = \begin{bmatrix} 1 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \end{bmatrix} = 2$$

$$r_0^T A r_0 = \begin{bmatrix} 1 & 1 \end{bmatrix} \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \begin{bmatrix} 1 \\ 1 \end{bmatrix} = 3$$

$$\alpha_0 = \frac{r_0^T r_0}{r_0^T A r_0} = \frac{2}{3}$$

$$x^1 = x^0 + \alpha_0 r_0$$

$$= \begin{bmatrix} 0 \\ 0 \end{bmatrix} + \frac{2}{3} \begin{bmatrix} 1 \\ 1 \end{bmatrix} = \begin{bmatrix} 2/3 \\ 2/3 \end{bmatrix}$$

$$r_1 = b - Ax^1 = \begin{bmatrix} 1 \\ 1 \end{bmatrix} - \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \begin{bmatrix} 2/3 \\ 2/3 \end{bmatrix}$$

↓

$$\alpha_1 \rightarrow \alpha_2 \rightarrow \dots \text{ until } r_k = 0$$

Substantial no. of steps required for the naive method.

Improvement → Conjugate Gradient method.

### Lecture 41: Conjugate Gradient Method

$$P_0 = r_0$$

$$P_1 = k(x_0 - x^1)$$

Initial direction same as steepest descent method

Suppose we want to reach  $x_c$  from  $x^1$  is 1 step. Then we'll have to move according to.

$$AP_1 = k(Ax_c - Ax^1)$$

$\underbrace{Ax_c}_{b}$  is correct sol<sup>n</sup>.

$$= k(r_1)$$

$$r_0^T r_1 = 0$$

Multiply LHS & RHS by  $r_0^T$

$$\rightarrow r_0^T A P_1 = k r_0^T r_1 = 0$$

$$\Rightarrow P_0^T A P_1 = 0$$

$P_0$  is 'A orthogonal' to  $P_1$ .

Form,

$$P_1 = r_1 - \alpha_2 P_0$$

↳ make new direction from old directions (Gram-Schmidt Conjecture).

$$P_0^T A (r_1 - \alpha_2 P_0) = 0$$

$$\Rightarrow \alpha_2 = \frac{P_0^T A r_1}{P_0^T A P_0}$$

We get direction  $P_1$  from this.

Next to find: how much to go?

$$x^2 = x^1 + \alpha_1 P_1$$

In steepest descent, we used  $x^2 = x^1 + \alpha_1 r_1$

(we moved along  $r_1$ . Now we move along different direction  $P_1$  with the hope of reaching the target in 1 shot).

$\alpha_1$  should be such that  $f$  should be a min.

$$\Rightarrow f(x^2) = \frac{1}{2}(x^1 + \alpha_1 P_1)^T A (x^1 + \alpha_1 P_1) - b^T (x^1 + \alpha_1 P_1) + c$$

For  $f$  to be minimum

$$\frac{df}{d\alpha_1} = 0 \rightarrow$$