

Homework 1

Mikko Karkkainen

August 27, 2017

1.1

$$\mu = 2 \cdot 10^{-2} \text{ Nsm}^{-2}$$

$$V = 61.0 \text{ cms}^{-1} = 0.61 \text{ ms}^{-1}$$

$$Y = 2 \text{ mm} = 2 \cdot 10^{-3} \text{ m}$$

For flow between two plates, one moving and one stationary:

$$\tau_{xy} = -\mu \frac{\partial v_x}{\partial y} \quad (1)$$

where τ is the momentum flux in the positive y-direction. For steady state:

$$\begin{aligned} -\mu \frac{\partial v_x}{\partial y} &= -\mu \frac{V}{Y} \\ &= -2 \cdot 10^{-2} \text{ Nsm}^{-2} \cdot \frac{0.61 \text{ ms}^{-1}}{2 \cdot 10^{-3} \text{ m}} \\ &= -6.1 \text{ Nm}^{-2} \end{aligned} \quad (2)$$

The direction of momentum transfer is from the top plate to the bottom plate (-y direction)

1.2

$$v_x = 3y - y^3$$

$$\rho = 10^3 \text{ kgm}^{-3}$$

$$v = 7 \cdot 10^{-7} \text{ m}^2\text{s}^{-1}$$

$$\mu = v\rho = 7 \cdot 10^{-4} \text{ kgm}^{-1}\text{s}^{-1}$$

a)

$$\begin{aligned}
 \frac{\partial v_x}{\partial y} &= 3 - 3y^2 \\
 \left. \frac{\partial v_x}{\partial y} \right|_{x=x_1, y=0} &= 3 \text{ cms}^{-1} \\
 &= 3 \cdot 10^{-2} \text{ ms}^{-1} \\
 \tau_{xy}|_{x=x_1, y=0} &= -\mu \left. \frac{\partial v_x}{\partial y} \right|_{x=x_1, y=0} \\
 &= -7 \cdot 10^{-4} \text{ kgm}^{-1}\text{s}^{-1} \cdot 3 \cdot 10^{-2} \text{ ms}^{-1} \\
 &= -2.1 \cdot 10^{-5} \text{ kgs}^{-2}
 \end{aligned} \tag{3}$$

The shear stress at $x = x_1, y = 0$ is $-2.1 \cdot 10^{-5} \text{ kgs}^{-2}$

b)

$$\begin{aligned}
 \frac{\partial v_x}{\partial y} &= 3 - 3y^2 \text{ cms}^{-1} \\
 \left. \frac{\partial v_x}{\partial y} \right|_{y=0.8mm} &= 1.08 \text{ cms}^{-1} \\
 &= 1.08 \cdot 10^{-2} \text{ ms}^{-1} \\
 \tau_{xy}|_{y=0.8mm} &= -\mu \left. \frac{\partial v_x}{\partial y} \right|_{x=x_1, y=0.8mm} \\
 &= -7 \cdot 10^{-4} \text{ kgm}^{-1}\text{s}^{-1} \cdot 1.08 \cdot 10^{-2} \text{ ms}^{-1} \\
 &= -7.56 \cdot 10^{-6} \text{ kgs}^{-2}
 \end{aligned}$$

The shear stress at $x = x_1, y = 0.8$ is $-7.56 \cdot 10^{-6} \text{ kgs}^{-2}$

c)

Momentum flux in the x-direction:

$$\tau_{yx} = -\mu \frac{\partial v_y}{\partial x} \tag{4}$$

Since the velocity profile has no component in y-direction, momentum flux in x-direction is 0

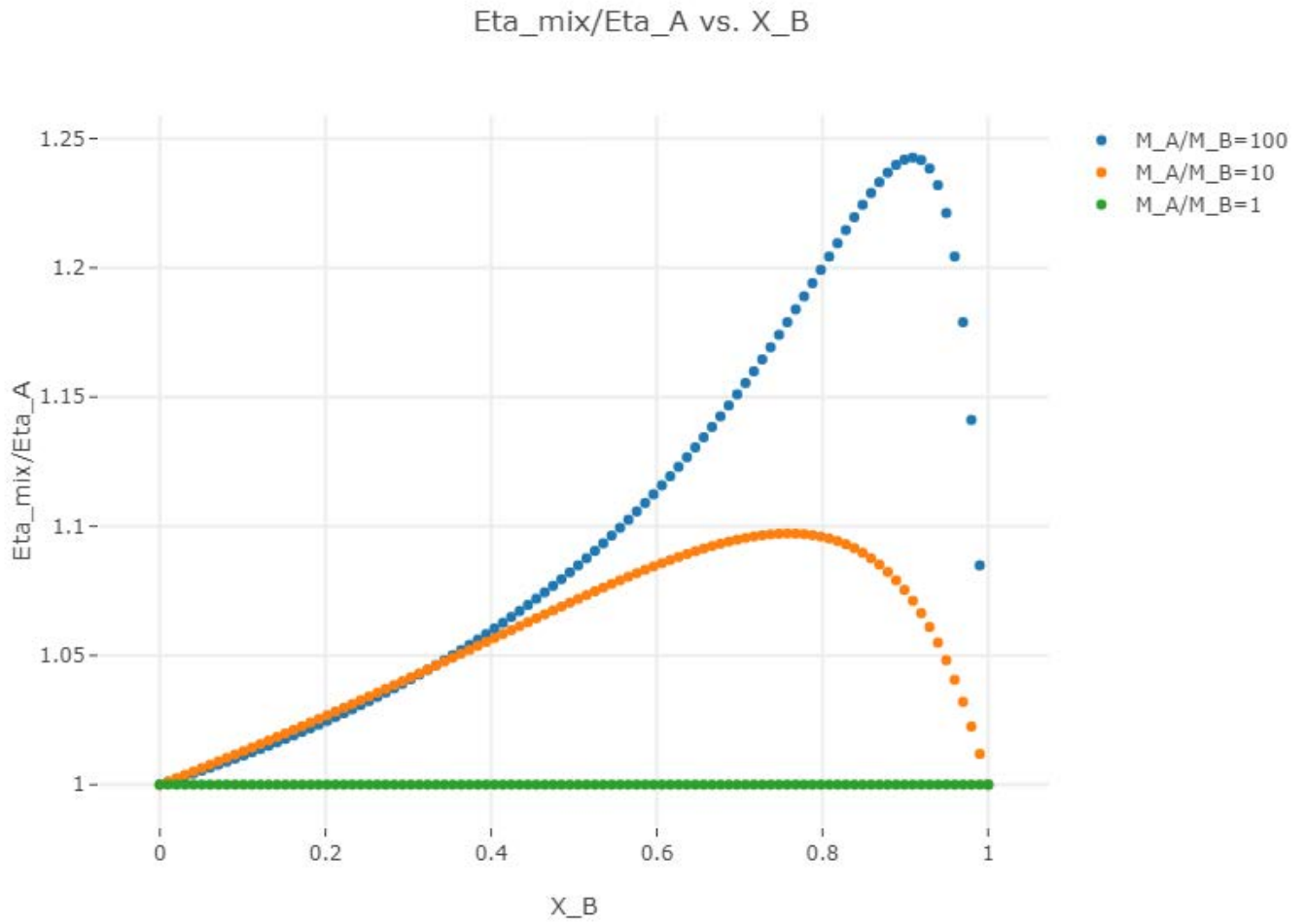
1.4

$$\eta_{mix} = \sum_{i=1}^n \frac{x_i \eta_i}{\sum_{j=1}^n x_j \phi_{ij}} \tag{5}$$

$$\phi_{ij} = \frac{1}{\sqrt{8}} \left[1 + \frac{M_i}{M_j} \right]^{-1/2} \left[1 + \left[\frac{\eta_i}{\eta_j} \right]^{1/2} \left[\frac{M_j}{M_i} \right]^{1/4} \right]^2 \tag{6}$$

For a binary gas with $\eta_A = \eta_B$:

$$\frac{\eta_{mix}}{\eta_A} = \frac{(1 - x_B)}{1 - x_B + x_B \left(\frac{1}{\sqrt{8}} \left[1 + \frac{M_A}{M_B} \right]^{-1/2} \left[1 + \left[\frac{M_B}{M_A} \right]^{1/4} \right]^2 \right)} + \frac{x_B}{x_B + (1 - x_B) \left(\frac{1}{\sqrt{8}} \left[1 + \frac{M_B}{M_A} \right]^{-1/2} \left[1 + \left[\frac{M_A}{M_B} \right]^{1/4} \right]^2 \right)} \quad (7)$$



1.8

$$\begin{aligned}T &= 2273 \text{ K} \\T_M &= 1575 \text{ K} \\M &= 52 \text{ gmol}^{-1} = 5.2 \cdot 10^{-2} \text{ kgmol}^{-1} \\\rho &= 7100 \text{ kgm}^{-3} \\\delta &= 0.272 \text{ nm}\end{aligned}$$

$$\begin{aligned}\frac{\epsilon}{K_B} &= 5.20 T_M \text{ K} \\&= 5.20 \cdot 1575 \text{ K} \\&= 8.19 \cdot 10^3 \text{ K}\end{aligned} \tag{8}$$

$$\begin{aligned}T^* &= \frac{K_B T}{\epsilon} \\&= \frac{2273 \text{ K}}{8.19 \cdot 10^3 \text{ K}} \\&= 0.2775\end{aligned} \tag{9}$$

From Figure 1.10 in the book:

$$\eta^*(V^*)^2 = 2.05$$

$$\begin{aligned} V^* &= \frac{1}{n\delta^3} \\ &= \frac{1}{\left[\frac{6.023 \cdot 10^{23} \text{ atoms}}{5.2 \cdot 10^{-2} \text{ kg}} \right] \left[\frac{7100 \text{ kg}}{\text{m}^3} \right] \left(0.272 \cdot 10^{-9} \text{ m} \right)^3} \\ &= 0.604 \end{aligned} \tag{10}$$

$$\eta^* = \frac{2.05}{0.604^2} = 5.62$$

$$\begin{aligned} \eta &= \frac{\eta^*(MRT)^{1/2}}{\delta^2 N_A} \\ &= \frac{5.62 \cdot \left(5.2 \cdot 10^{-2} \text{ kgmol}^{-1} \cdot 8.3144 \text{ Jmol}^{-1}\text{K}^{-1} \cdot 2273 \text{ K} \right)^{1/2}}{\left(0.272 \cdot 10^{-9} \text{ m} \right)^2 \cdot 6.023 \cdot 10^{23} \text{ mol}^{-1}} \\ &= 3.954 \cdot 10^{-3} \text{ kgm}^{-1}\text{s}^{-1} \end{aligned} \tag{11}$$

The viscosity for Cr at 2273K is approximately $3.954 \cdot 10^{-3} \text{ kgm}^{-1}\text{s}^{-1}$