

Back to Flying Car Nanodegree

Building an Estimator

审阅
代码审阅
HISTORY

Meets Specifications

Spectacular work! Great writeup and inclusion of figures to demonstrate your quadrotor passing!

You are ready for Term 2!

Writeup

The writeup / README should include a statement and supporting figures / images that explain how each rubric item was addressed, and specifically where in the code each step was handled.

Implement Estimator

The calculated standard deviation should correctly capture ~68% of the sensor measurements. Your writeup should describe the method used for determining the standard deviation given the simulated sensor measurements.

The improved integration scheme should result in an attitude estimator of < 0.1 rad for each of the Euler angles for a duration of at least 3 seconds during the simulation. The integration scheme should use quaternions to improve performance over the current simple integration scheme.

The prediction step should include the state update element (**PredictState()** function), a correct calculation of the Rgb prime matrix, and a proper update of the state covariance. The acceleration should be accounted for as a command in the calculation of gPrime. The covariance update should follow the classic EKF update equation.

The update should properly include the magnetometer data into the state. Note that the solution should make sure to correctly measure the angle error between the current state and the magnetometer value (error should be the short way around, not the long way).

The estimator should correctly incorporate the GPS information to update the current state estimate.

Flight Evaluation

For each step of the project, the final estimator should be able to successfully meet the performance criteria with the controller provided. The estimator's parameters should be properly adjusted to satisfy each of the performance criteria elements.

The controller developed in the previous project should be de-tuned to successfully meet the performance criteria of the final scenario (<1m error for entire box flight).

返回 PATH

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