

# Jittering effects analysis and beam training design for UAV mmWave communications

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# Outline

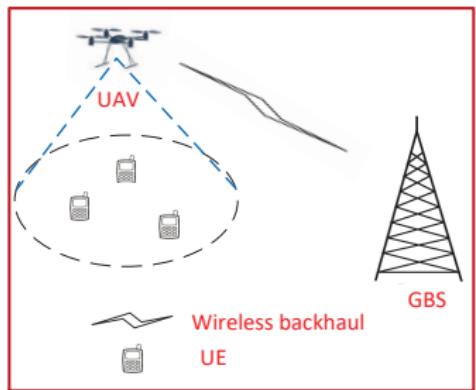
- 1 Background and Motivations
- 2 The Effects of UAV Jitter on mmWave Channel
- 3 Navigation-information-aided Beam Training Design
- 4 Numerical Results
- 5 Conclusion



# UAV communications

## Background

- UAV communications
  - High operational flexibility and controllable mobility
- MmWave communications
  - Directional transmission and high throughput
- UAV + mmWave communications
  - An on-demand solution to high-capacity wireless backhaul in cellular networks



## Motivations

- Jittering effects are a key factor that characterizes UAV communications
  - Detrimental to UAV mmWave backhaul with a directional narrow beam

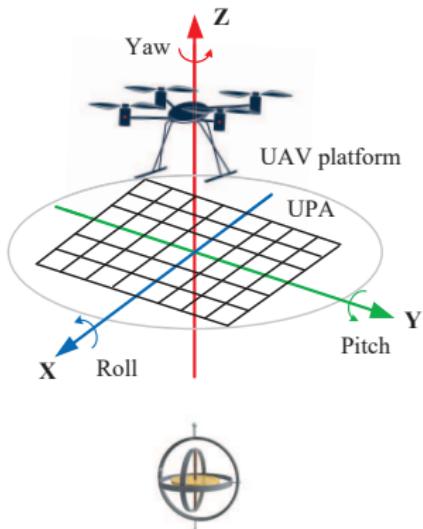


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# What is UAV jitter?



- **UAV jitter:** The unintended high-frequency change of UAV attitude/orientation;
- **Euler angles** describe the orientation of a rigid body with respect to a fixed coordinate system

$$\mathbf{R} = \mathbf{R}_{Yaw}(\alpha)\mathbf{R}_{Pitch}(\beta)\mathbf{R}_{Roll}(\gamma)$$

- **UAV jitter modelled by Euler angles**

$$\alpha = \bar{\alpha} + \Delta\alpha$$

$$\beta = \bar{\beta} + \Delta\beta$$

$$\gamma = \bar{\gamma} + \Delta\gamma$$

where  $\bar{\alpha}, \bar{\beta}, \bar{\gamma}$  are the desired attitude angles, and  $\Delta\alpha, \Delta\beta, \Delta\gamma$  refer to the fluctuations caused by UAV jitter.

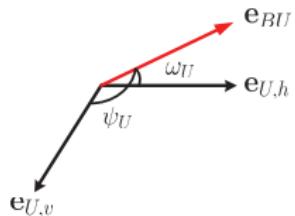
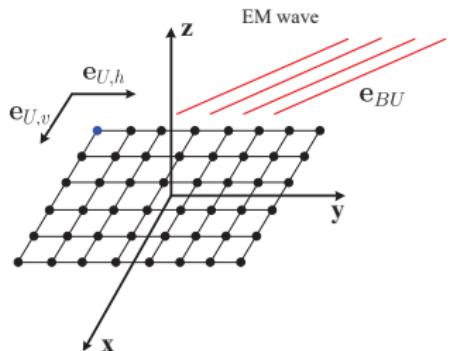


# Narrow-band mmWave Channel Model

## Channel model

$$\mathbf{H} = \sum_{l=0}^{L-1} \beta_l \mathbf{v}(\Psi_{U,l}, \Omega_{U,l}) \mathbf{v}^H(\Psi_{B,l}, \Omega_{B,l})$$

- $\mathbf{v}(\Psi_U, \Omega_U)$  and  $\mathbf{v}(\Psi_B, \Omega_B)$  are array response (steering) vectors at UAV side and BS side.
- LoS component dominates, i.e.,  $|\beta_0| >> |\beta_l|, l \neq 0$
- $\mathbf{H}$  is characterized by  $(\Psi_{U,0}, \Omega_{U,0})$  and  $(\Psi_{B,0}, \Omega_{B,0})$ , which are termed as cosine AoA/AoD or direction cosines.

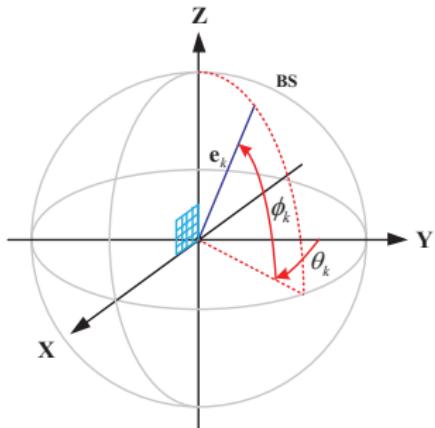


$$\Omega_U = \cos \omega_U = \mathbf{e}_{U,h}^T \mathbf{e}_{BU}$$

$$\Psi_U = \cos \psi_U = \mathbf{e}_{U,v}^T \mathbf{e}_{BU}$$



# A Deep Look Into Angles



$\phi_k$ : Elevation angle

$\theta_k$ : Azimuth angle

$e_k$ : Direction vector  
from user k to BS

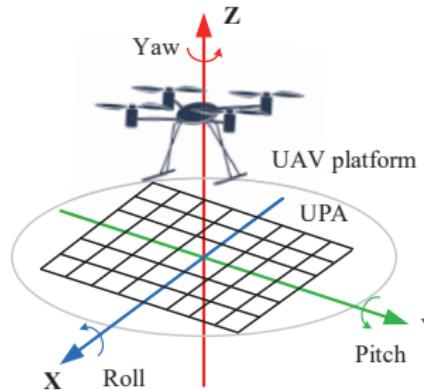
## (1) Azimuth angle & Elevation angle

- Angles in spherical coordinate that are used to identify the **position** of UAV
- The direction vector from UAV to BS is  $e_{BU} = [\cos \phi \cos \theta, \cos \phi \sin \theta, \sin \phi]^T$  where  $\phi$  is elevation angle and  $\theta$  is azimuth angle of the LoS path.
- Can be obtained via GPS and barometer.**

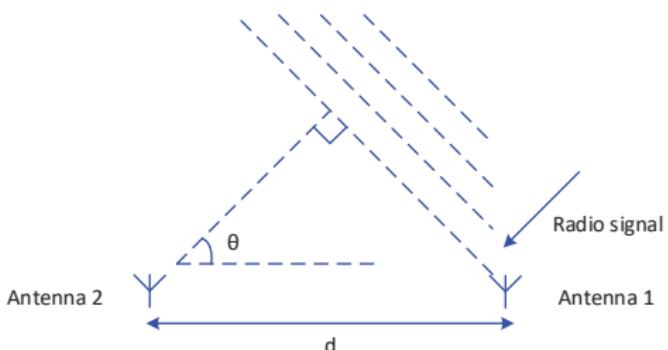


# A Deep Look Into Angles

## (2) Yaw, pitch & roll angles



## (3) Angle of arrival (AoA) & Angle of departure (AoD)



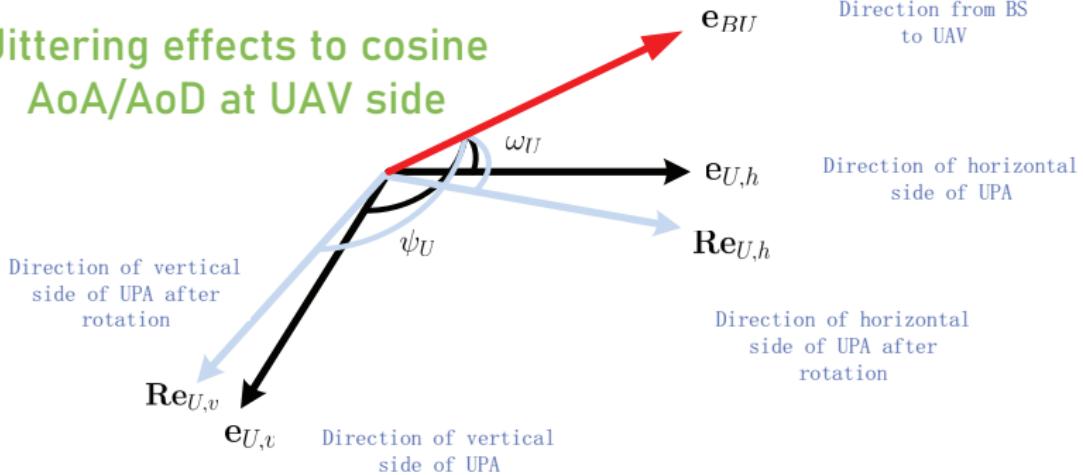
- Can be measured via **Gyroscope**.

- Can be obtained through estimating **phase differences** of the elements of an **antenna array**.



# A Deep Look Into Angles

## Jittering effects to cosine AoA/AoD at UAV side



$$\Omega_U = \cos \omega_U = \mathbf{e}_{U,h}^T \mathbf{e}_{BU}$$

$$\Psi_U = \cos \psi_U = \mathbf{e}_{U,v}^T \mathbf{e}_{BU}$$

Rotation: R

$$\Omega_U = \cos \omega_U = \mathbf{e}_{U,h}^T \mathbf{R}^T \mathbf{e}_{BU}$$

$$\Psi_U = \cos \psi_U = \mathbf{e}_{U,v}^T \mathbf{R}^T \mathbf{e}_{BU}$$



# Jittering Effects on Cosine AoA/AoD

Two-dimensional AoA/AoD	UAV position related angles (i.e., $\phi, \theta$ )	UAV attitude related angles (i.e., $\alpha, \beta, \gamma$ )
$(\Psi_B, \Omega_B)$	Dependent	Independent
$(\Psi_U, \Omega_U)$	Dependent	Dependent

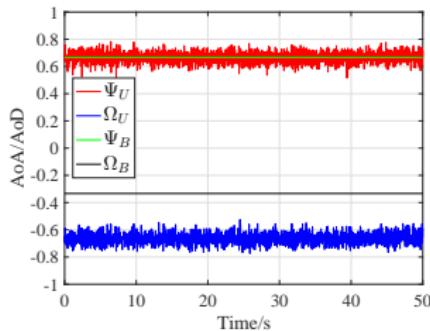
**Table 1:** Dependency of the two-dimensional AoA/AoD (BS side and UAV side) on UAV position and UAV attitude

- UAV Jittering Effects on cosine AoA/AoD at UAV side cannot be ignored
- UAV Jittering Effects on cosine AoA/AoD at BS side are negligible

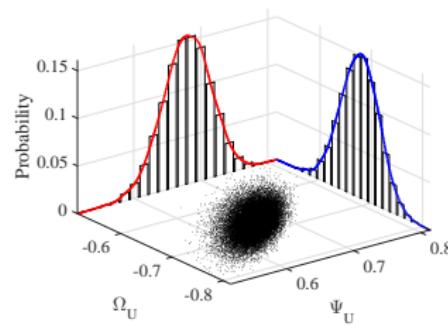
# Jittering Effects on Cosine AoA/AoD - Numerical Examples



Modelling of UAV jitter:  $\sigma_\alpha = \sigma_\beta = \sigma_\gamma = 0.05$



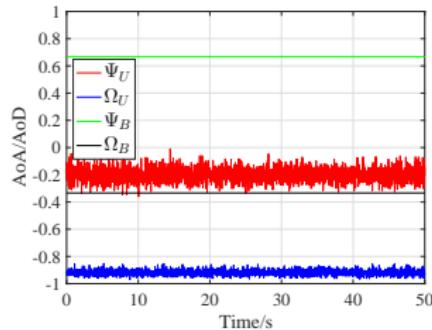
(a) Variation of  $\Psi_U$ ,  $\Omega_U$ ,  $\Psi_B$  and  $\Omega_B$  over time (Scenario 1)



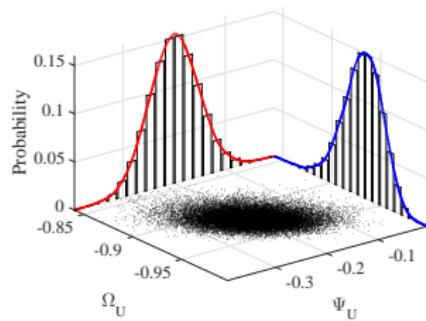
(b) Empirical marginal probability density function of  $\Psi_U$  and  $\Omega_U$  (Scenario 1)

**Figure 1:** UAV is hovering at the position  $\mathbf{p}_U = [-100, 100, 50]^T$  (Cartesian coordinates) with its desired flight attitude being  $\bar{\alpha} = \bar{\beta} = \bar{\gamma} = 0$

# Jittering Effects on Cosine AoA/AoD - Numerical Examples



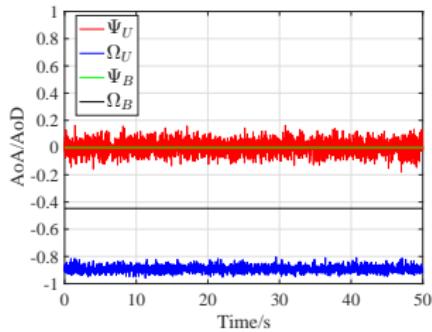
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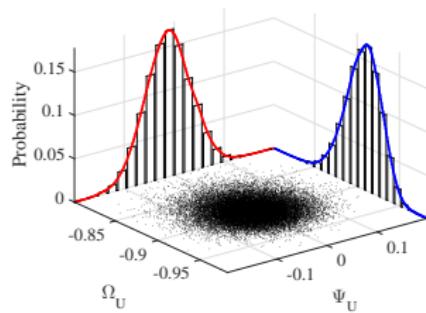
(b) Empirical marginal probability density function of  $\Psi_U$  and  $\Omega_U$  (Scenario 1)

**Figure 2:** UAV is hovering at the position  $\mathbf{p}_U = [-100, 100, 50]^T$  (Cartesian coordinates) with its desired flight attitude being  $\bar{\alpha} = 1$ , and  $\bar{\beta} = \bar{\gamma} = 0$

# Jittering Effects on Cosine AoA/AoD - Numerical Examples



(a) Variation of  $\Psi_U$ ,  $\Omega_U$ ,  $\Psi_B$  and  $\Omega_B$  over time (Scenario 1)

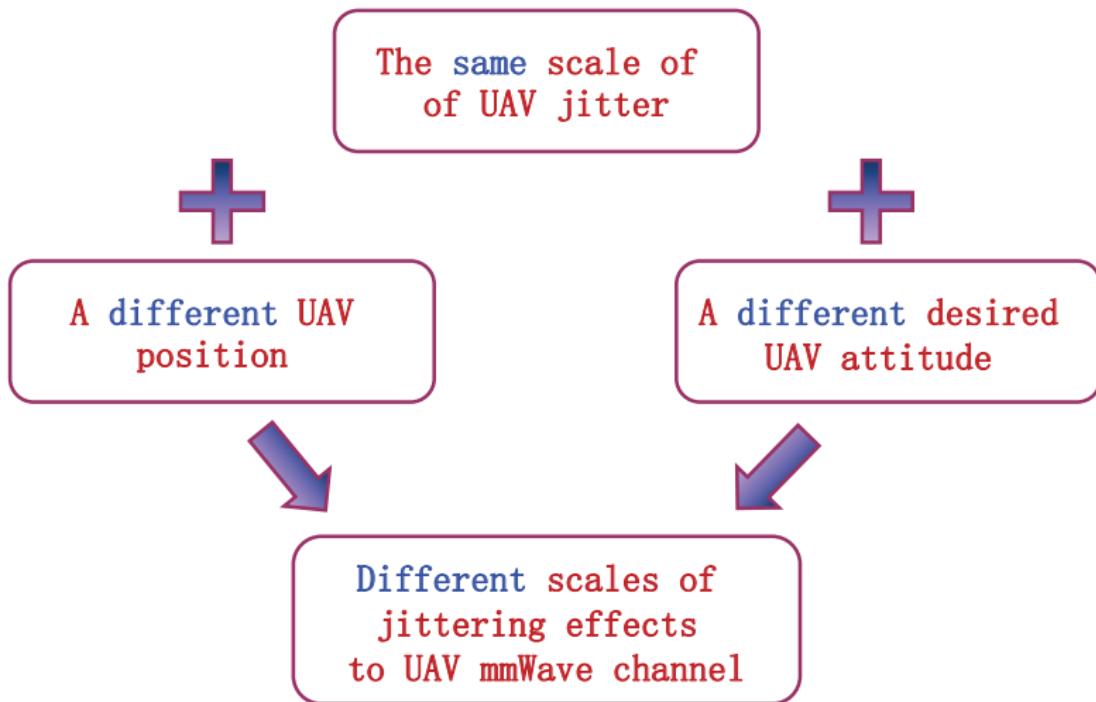


(b) Empirical marginal probability density function of  $\Psi_U$  and  $\Omega_U$  (Scenario 1)

**Figure 3:** UAV is hovering at the position  $\mathbf{p}_U = [0, 100, 50]^T$  (Cartesian coordinates) with its desired flight attitude being  $\bar{\alpha} = \bar{\beta} = \bar{\gamma} = 0$



# Jittering Effects on Cosine AoA/AoD





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# UAV Beam Training Under Jittering Effects

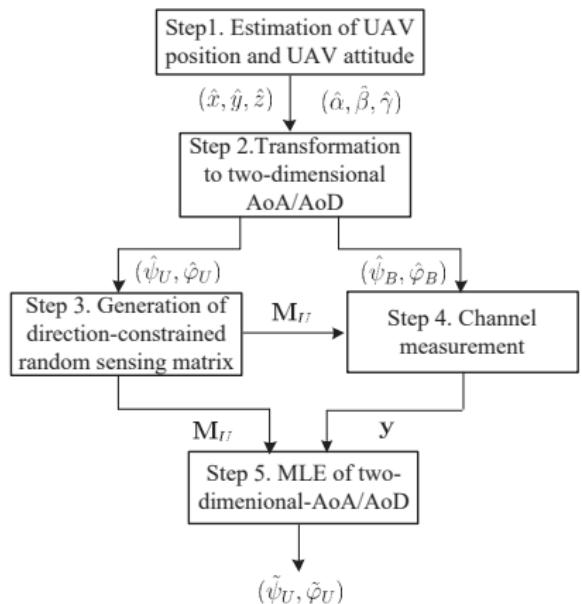
The Objective of Beam Training: Estimate AoA/AoD of the strongest path in mmWave channel to support the subsequent beam alignment operation.

- Ideally, with the relative position (**Azimuth angle & Elevation angle**) and attitude (**Euler angles**) of UAV, AoA/AoD can be accurately obtained.
- However, UAV platform faces the following challenges
  - Gyroscope and accelerometer are very **sensitive** to the jitter/vibration induced by the engine and wind gust;
  - The **estimation error** increases with the degree of UAV jitter.



# UAV Beam Training Under Jittering Effects

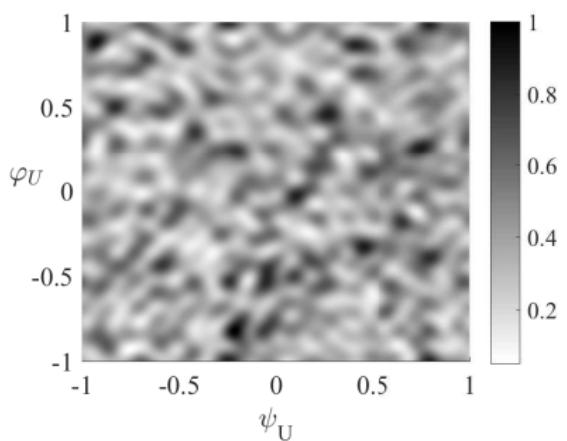
Utilizing navigation information to facilitate compressed sensing (CS) based UAV beam training.



- Obtain a rough estimate of cosine AoA/AoD from navigation information according to the relationship between angles.
- Narrow down the search range of CS-based beam training



# How to Design Sensing Matrix



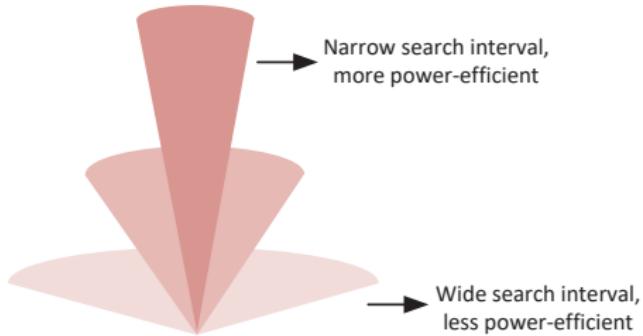
**Figure 4:** Fully random case with the sensing range being  $\Psi_U, \Omega_U \in (-1, 1)$

- Sensing matrix needs to satisfy restricted isometry property (RIP)
- Randomly generated (under **constant modulus constraint**) sensing matrix satisfies RIP with high probability
- However, fully random sensing matrix is **semi-omnidirectional** and **power inefficient** and will result in heavier training overload.



# How to Design Sensing Matrix

How to design the (random) CS sensing matrix within an arbitrary sensing range?



- Design of direction-constrained CS sensing matrix is challenging due to
  - Constant modulus constraint of analog array antenna
  - Restrained sensing range



# Direction-Constrained Sensing Matrix

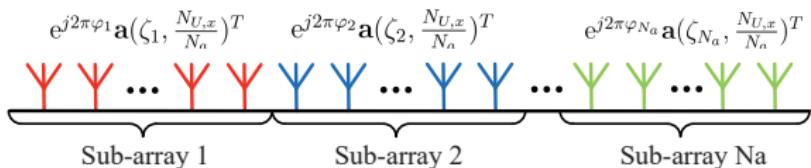


Figure 5: Visualization of sub-array partition along one dimension

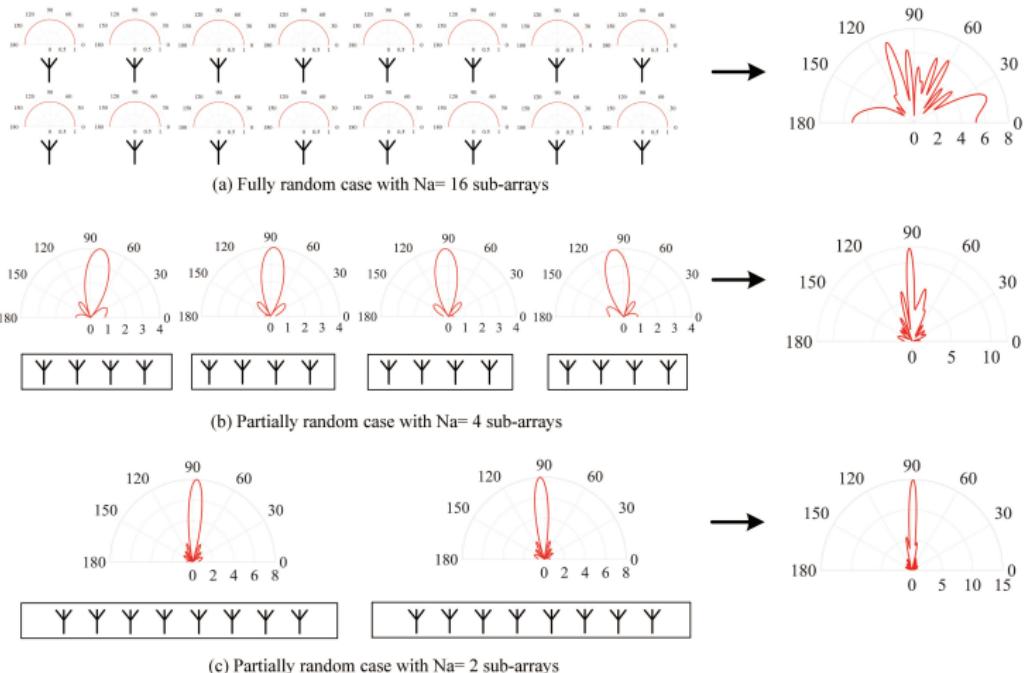
## Sub-array based method:

Radiation range of the sub-array:  $\left(-\frac{N_a}{N_{U,x}} + \zeta_{n_a}, \frac{N_a}{N_{U,x}} + \zeta_{n_a}\right)$

- **Step1.** Restrict center angle  $\zeta_{n_a}$  and sub-array size  $N_{U,x}$  to restrain radiation range;
- **Step2.** Randomize  $\varphi_{n_a}$  and partially randomize center angle  $\zeta_{n_a}$  to satisfy RIP.



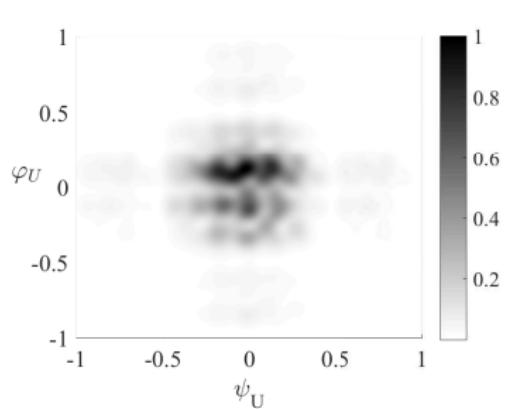
# Direction-Constrained Sensing Matrix



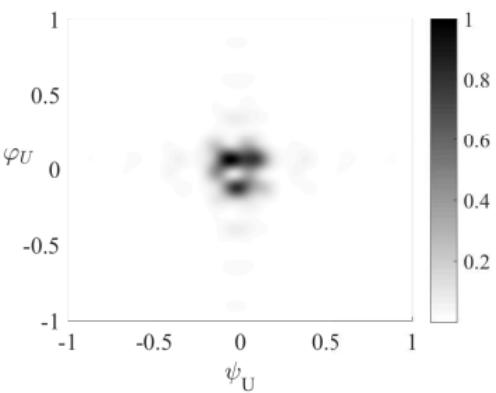
**Figure 6:** Visualization of sub-array based design of random sensing vector (the scale of angle is  $\cos^{-1} \Psi_U$ )



# Direction-Constrained Sensing Matrix



(a)  $N_a = 4$  sub-arrays, and the sensing range is  $\Psi_U, \Omega_U \in (-0.4, 0.4)$



(b)  $N_a = 2$  sub-arrays, and the sensing range is  $\Psi_U, \Omega_U \in (-0.225, 0.225)$

Figure 7: Beam space of partially random sensing matrices under constant modulus constraint

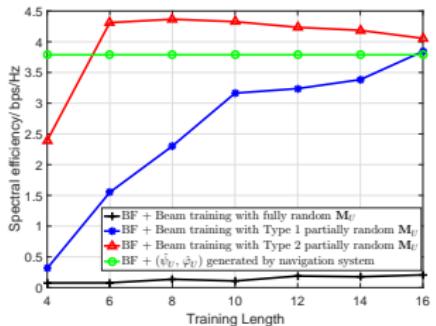
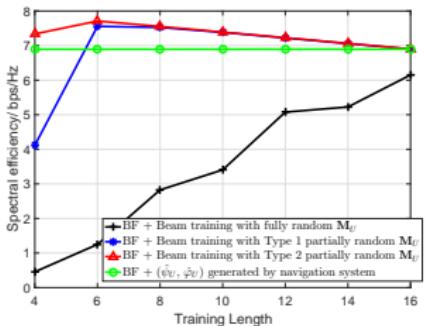
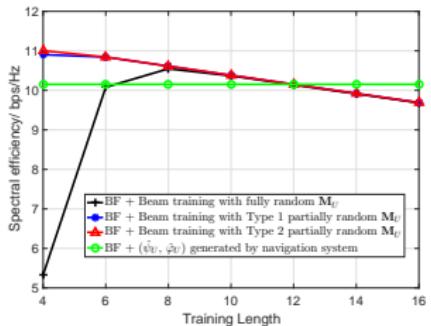
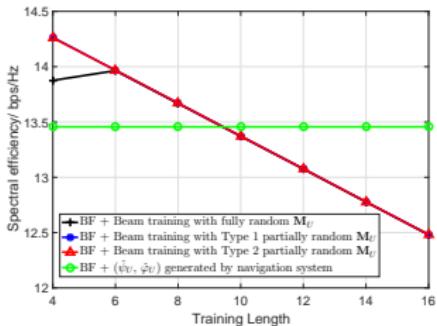


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# Spectral Efficiency Comparison

(a) When transmit power is  $-10\text{dBm}$ (b) When transmit power is  $0\text{dBm}$ (c) When transmit power is  $10\text{dBm}$ (d) When transmit power is  $20\text{dBm}$



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# Conclusion

- We analytically build the connection between UAV jitter and its effects on mmWave channel;
- We propose a navigation-information-aided beam training for UAV mmWave communications;
- UAV beam training scheme assisted by navigation information can achieve better accuracy with **reduced training length** in AoA/AoD estimation.

# Q&A