

3. The solution of the nodal value is,

U1 = 0.00000E+00 /m U2 = 0.00000E+00 /rad

U3 = 0.00000E+00 /m U4 = -0.13021E-01 /rad

U5 = 0.28646E+00 /m U6 = -0.33854E-01 /rad

The solutions match well with the hand calculation results.

Sample data and output:

*** ECHO OF THE INPUT DATA STARTS ***

Hw-6-1:

```
3 0 0          MODEL, NTYPE, ITEM
0 2           IELEM, NEM
1 0           ICONT, NPRNT
0.0 10 10      DX(I)
0.0 0.0        AX0, AX1
4.0E6 0.0      BX0, BX1
0.0 0.0        CX0, CX1
500.0 0.0 0.0  FX0, FX1, FX2
3             NSPV
1 1 0.0        ISPV(1,1), ISPV(1,2), VSPV(1)
1 2 0.0        ISPV(2,1), ISPV(2,2), VSPV(2)
2 1 0.0        ISPV(3,1), ISPV(3,2), VSPV(3)
0             NSSV
0             NNBC
0             NMPC
```

**** ECHO OF THE INPUT DATA ENDS ****

OUTPUT from program FEM1D by J N REDDY

Hw-6-1:

*** ANALYSIS OF MODEL 3, AND TYPE 0 PROBLEM ***
(see the code below)

MODEL=1,NTYPE=0: A problem described by MODEL EQ. 1
MODEL=1,NTYPE=1: A circular DISK (PLANE STRESS)
MODEL=1,NTYPE>1: A circular DISK (PLANE STRAIN)
MODEL=2,NTYPE=0: A Timoshenko BEAM (RIE) problem
MODEL=2,NTYPE=1: A Timoshenko PLATE (RIE) problem
MODEL=2,NTYPE=2: A Timoshenko BEAM (CIE) problem
MODEL=2,NTYPE>2: A Timoshenko PLATE (CIE) problem
MODEL=3,NTYPE=0: A Euler-Bernoulli BEAM problem
MODEL=3,NTYPE>0: A Euler-Bernoulli Circular plate
MODEL=4,NTYPE=0: A plane TRUSS problem
MODEL=4,NTYPE=1: A Euler-Bernoulli FRAME problem
MODEL=4,NTYPE=2: A Timoshenko (CIE) FRAME problem

Element type (0, Hermite,>0, Lagrange)..= 0

No. of deg. of freedom per node, NDF....= 2

No. of elements in the mesh, NEM.....= 2
No. of total DOF in the model, NEQ.....= 6
Half bandwidth of matrix [GLK], NHBW ...= 4
No. of specified primary DOF, NSPV.....= 3
No. of specified secondary DOF, NSSV.....= 0
No. of specified Newton B. C.: NNBC.....= 0
No. of speci. multi-pt. cond.: NMPC.....= 0

Boundary information on primary variables:

1 1 0.00000E+00
1 2 0.00000E+00
2 1 0.00000E+00

Global coordinates of the nodes, {GLX}:

0.00000E+00 0.10000E+02 0.20000E+02

Coefficients of the differential equation:

AX0 = 0.0000E+00 AX1 = 0.0000E+00
BX0 = 0.4000E+07 BX1 = 0.0000E+00
CX0 = 0.0000E+00 CX1 = 0.0000E+00

SOLUTION (values of PVs) at the NODES:

0.00000E+00 0.00000E+00 0.00000E+00 -0.13021E-01 0.28646E+00
-0.33854E-01

x is the global coord. if ICONT=1 and it is the local coord. if ICONT=0

x Deflect. Rotation B. Moment Shear Force

0.00000E+00 0.00000E+00 0.00000E+00 0.10417E+05 -0.31250E+04
0.12500E+01 -0.17802E-02 0.26449E-02 0.65104E+04 -0.31250E+04
0.25000E+01 -0.61035E-02 0.40690E-02 0.26042E+04 -0.31250E+04
0.37500E+01 -0.11444E-01 0.42725E-02 -0.13021E+04 -0.31250E+04
0.50000E+01 -0.16276E-01 0.32552E-02 -0.52083E+04 -0.31250E+04
0.62500E+01 -0.19073E-01 0.10173E-02 -0.91146E+04 -0.31250E+04
0.75000E+01 -0.18311E-01 -0.24414E-02 -0.13021E+05 -0.31250E+04
0.87500E+01 -0.12461E-01 -0.71208E-02 -0.16927E+05 -0.31250E+04
0.10000E+02 0.00000E+00 -0.13021E-01 -0.20833E+05 -0.31250E+04
0.10000E+02 0.00000E+00 -0.13021E-01 -0.20833E+05 0.25000E+04
0.11250E+02 0.20142E-01 -0.19043E-01 -0.17708E+05 0.25000E+04
0.12500E+02 0.47201E-01 -0.24089E-01 -0.14583E+05 0.25000E+04
0.13750E+02 0.79956E-01 -0.28158E-01 -0.11458E+05 0.25000E+04
0.15000E+02 0.11719E+00 -0.31250E-01 -0.83333E+04 0.25000E+04
0.16250E+02 0.15767E+00 -0.33366E-01 -0.52083E+04 0.25000E+04
0.17500E+02 0.20020E+00 -0.34505E-01 -0.20833E+04 0.25000E+04
0.18750E+02 0.24353E+00 -0.34668E-01 0.10417E+04 0.25000E+04
0.20000E+02 0.28646E+00 -0.33854E-01 0.41667E+04 0.25000E+04

Stop - Program terminated.

4. The solution of the nodal value is,

U1 = 0.00000E+00	U2 = -0.19667E+05	U3 = 0.58667E+05	U4 = -0.46667E+04
U5 = 0.44000E+05	U6 = 0.10333E+05	U7 = 0.00000E+00	U8 = 0.46667E+04
U9 = 0.13333E+05	U10 = -0.60000E+04		

The solutions match well with the hand calculation results.

Sample data and output:

```
*** ECHO OF THE INPUT DATA STARTS ***
Hw-6-1:
3 0 0          MODEL, NTYPE, ITEM
0 4           IELEM, NEM
0 0           ICONT, NPRNT

      5
      1      2      4.0          NNM
      0.0 0.0          AX0, AX1    NOD(1,J),GLX(1)
      1.0 0.0          BX0, BX1
      0.0 0.0          CX0, CX1
      0.0 0.0 0.0      FX0, FX1, FX2
      2      3      4.0          NOD(2,J),GLX(2)
      0.0 0.0          AX0, AX1
      1.0 0.0          BX0, BX1
      0.0 0.0          CX0, CX1
      0.0 0.0 0.0      FX0, FX1, FX2
      3      4      4.0          NOD(3,J),GLX(3)
      0.0 0.0          AX0, AX1
      1.0 0.0          BX0, BX1
      0.0 0.0          CX0, CX1
      1000.0 0.0 0.0    FX0, FX1, FX2
      4      5      4.0          NOD(4,J),GLX(4)
      0.0 0.0          AX0, AX1
      1.0 0.0          BX0, BX1
      0.0 0.0          CX0, CX1
      1000.0 0.0 0.0    FX0, FX1, FX2
2          NSPV
1 1 0.0      ISPV(1,1), ISPV(1,2), VSPV(1)
4 1 0.0      ISPV(2,1), ISPV(2,2), VSPV(2)
2          NSSV
2 1 2500     ISSV(1,1), ISSV(1,2), VSSV(1)
2 2 2500     ISSV(2,1), ISSV(2,2), VSSV(2)
0          NNBC
0          NMPC
**** ECHO OF THE INPUT DATA ENDS ****
```

OUTPUT from program FEM1D by J N REDDY

Hw-6-1:

*** ANALYSIS OF MODEL 3, AND TYPE 0 PROBLEM ***
(see the code below)

MODEL=1,NTYPE=0: A problem described by MODEL EQ. 1
MODEL=1,NTYPE=1: A circular DISK (PLANE STRESS)

MODEL=1,NTYPE>1: A circular DISK (PLANE STRAIN)
MODEL=2,NTYPE=0: A Timoshenko BEAM (RIE) problem
MODEL=2,NTYPE=1: A Timoshenko PLATE (RIE) problem
MODEL=2,NTYPE=2: A Timoshenko BEAM (CIE) problem
MODEL=2,NTYPE>2: A Timoshenko PLATE (CIE) problem
MODEL=3,NTYPE=0: A Euler-Bernoulli BEAM problem
MODEL=3,NTYPE>0: A Euler-Bernoulli Circular plate
MODEL=4,NTYPE=0: A plane TRUSS problem
MODEL=4,NTYPE=1: A Euler-Bernoulli FRAME problem
MODEL=4,NTYPE=2: A Timoshenko (CIE) FRAME problem

Element type (0, Hermite,>0, Lagrange)..= 0
No. of deg. of freedom per node, NDF.....= 2
No. of elements in the mesh, NEM.....= 4
No. of total DOF in the model, NEQ.....= 10
Half bandwidth of matrix [GLK], NHBW ...= 4
No. of specified primary DOF, NSPV.....= 2
No. of specified secondary DOF, NSSV.....= 2
No. of specified Newton B. C.: NNBC.....= 0
No. of speci. multi-pt. cond.: NMPC.....= 0

Boundary information on primary variables:

1 1 0.00000E+00
4 1 0.00000E+00

Boundary information on secondary variables:

2 1 0.25000E+04
2 2 0.25000E+04

Properties of Element = 1

Element length, H = 0.4000E+01
AX0 = 0.0000E+00 AX1 = 0.0000E+00
BX0 = 0.1000E+01 BX1 = 0.0000E+00
CX0 = 0.0000E+00 CX1 = 0.0000E+00
FX0 = 0.0000E+00 FX1 = 0.0000E+00 FX2 = 0.0000E+00

Properties of Element = 2

Element length, H = 0.4000E+01
AX0 = 0.0000E+00 AX1 = 0.0000E+00
BX0 = 0.1000E+01 BX1 = 0.0000E+00
CX0 = 0.0000E+00 CX1 = 0.0000E+00
FX0 = 0.0000E+00 FX1 = 0.0000E+00 FX2 = 0.0000E+00

Properties of Element = 3

Element length, H = 0.4000E+01
AX0 = 0.0000E+00 AX1 = 0.0000E+00
BX0 = 0.1000E+01 BX1 = 0.0000E+00
CX0 = 0.0000E+00 CX1 = 0.0000E+00
FX0 = 0.1000E+04 FX1 = 0.0000E+00 FX2 = 0.0000E+00

Properties of Element = 4

Element length, H = 0.4000E+01
AX0 = 0.0000E+00 AX1 = 0.0000E+00
BX0 = 0.1000E+01 BX1 = 0.0000E+00
CX0 = 0.0000E+00 CX1 = 0.0000E+00
FX0 = 0.1000E+04 FX1 = 0.0000E+00 FX2 = 0.0000E+00

SOLUTION (values of PVs) at the NODES:

0.00000E+00 -0.19667E+05 0.58667E+05 -0.46667E+04 0.44000E+05
0.10333E+05 0.00000E+00 0.46667E+04 0.13333E+05 -0.60000E+04

x is the global coord. if ICONT=1 and it is the local coord. if ICONT=0
x Deflect. Rotation B. Moment Shear Force

0.00000E+00	0.00000E+00	-0.19667E+05	0.56155E-05	0.18750E+04
0.50000E+00	0.97943E+04	-0.19432E+05	0.93750E+03	0.18750E+04
0.10000E+01	0.19354E+05	-0.18729E+05	0.18750E+04	0.18750E+04
0.15000E+01	0.28445E+05	-0.17557E+05	0.28125E+04	0.18750E+04
0.20000E+01	0.36833E+05	-0.15917E+05	0.37500E+04	0.18750E+04
0.25000E+01	0.44284E+05	-0.13807E+05	0.46875E+04	0.18750E+04
0.30000E+01	0.50562E+05	-0.11229E+05	0.56250E+04	0.18750E+04
0.35000E+01	0.55435E+05	-0.81823E+04	0.65625E+04	0.18750E+04
0.40000E+01	0.58667E+05	-0.46667E+04	0.75000E+04	0.18750E+04
0.00000E+00	0.58667E+05	-0.46667E+04	0.50000E+04	-0.62500E+03
0.50000E+00	0.60388E+05	-0.22448E+04	0.46875E+04	-0.62500E+03
0.10000E+01	0.60937E+05	0.20833E+02	0.43750E+04	-0.62500E+03
0.15000E+01	0.60393E+05	0.21302E+04	0.40625E+04	-0.62500E+03
0.20000E+01	0.58833E+05	0.40833E+04	0.37500E+04	-0.62500E+03
0.25000E+01	0.56336E+05	0.58802E+04	0.34375E+04	-0.62500E+03
0.30000E+01	0.52979E+05	0.75208E+04	0.31250E+04	-0.62500E+03
0.35000E+01	0.48841E+05	0.90052E+04	0.28125E+04	-0.62500E+03
0.40000E+01	0.44000E+05	0.10333E+05	0.25000E+04	-0.62500E+03
0.00000E+00	0.44000E+05	0.10333E+05	0.38333E+04	-0.26250E+04
0.50000E+00	0.38409E+05	0.11922E+05	0.25208E+04	-0.26250E+04
0.10000E+01	0.32187E+05	0.12854E+05	0.12083E+04	-0.26250E+04
0.15000E+01	0.25664E+05	0.13130E+05	-0.10417E+03	-0.26250E+04
0.20000E+01	0.19167E+05	0.12750E+05	-0.14167E+04	-0.26250E+04
0.25000E+01	0.13023E+05	0.11714E+05	-0.27292E+04	-0.26250E+04
0.30000E+01	0.75625E+04	0.10021E+05	-0.40417E+04	-0.26250E+04
0.35000E+01	0.31120E+04	0.76719E+04	-0.53542E+04	-0.26250E+04
0.40000E+01	0.00000E+00	0.46667E+04	-0.66667E+04	-0.26250E+04
0.00000E+00	0.00000E+00	0.46667E+04	-0.66667E+04	0.20000E+04
0.50000E+00	-0.15417E+04	0.15833E+04	-0.56667E+04	0.20000E+04
0.10000E+01	-0.16667E+04	-0.10000E+04	-0.46667E+04	0.20000E+04
0.15000E+01	-0.62500E+03	-0.30833E+04	-0.36667E+04	0.20000E+04
0.20000E+01	0.13333E+04	-0.46667E+04	-0.26667E+04	0.20000E+04
0.25000E+01	0.39583E+04	-0.57500E+04	-0.16667E+04	0.20000E+04
0.30000E+01	0.70000E+04	-0.63333E+04	-0.66667E+03	0.20000E+04
0.35000E+01	0.10208E+05	-0.64167E+04	0.33333E+03	0.20000E+04
0.40000E+01	0.13333E+05	-0.60000E+04	0.13333E+04	0.20000E+04

Stop - Program terminated.