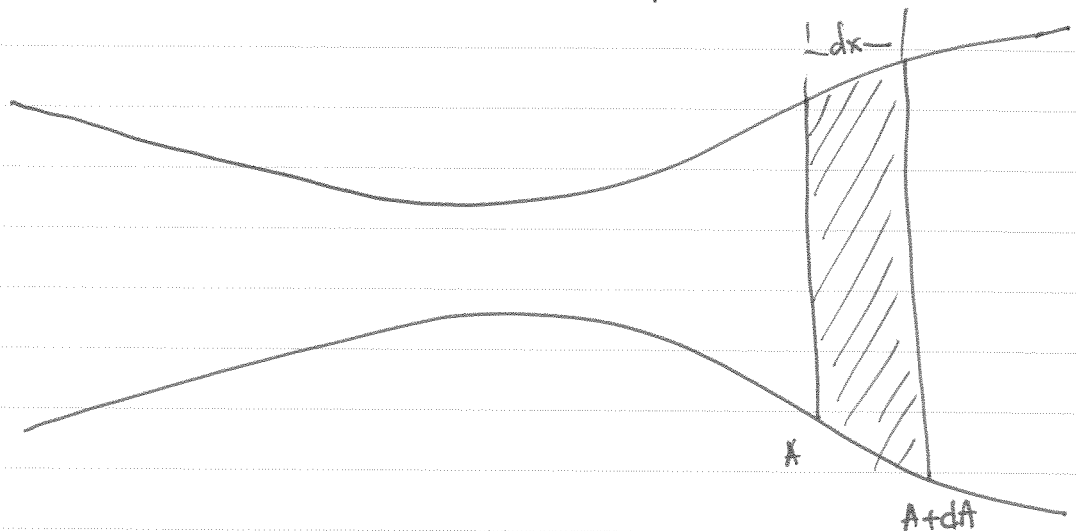


Quasi-1D Euler Equations

Derivation of continuity equation,

$$\frac{d}{dt} \int_{cv} \rho dV + \int_{cs} \rho \vec{V} \cdot \vec{n} dA = 0.$$

The control volume for a quasi-1D flow,



$$\frac{\partial}{\partial t} \int_{cv} \rho dV = \frac{\partial}{\partial t} \left(\int_{cv} \rho A dx \right)$$

where $A = A(x)$

$$\begin{aligned} \int_{cs} \rho (\vec{V} \cdot \vec{n}) dA &= \rho(-u)A + (\rho + d\rho)(u + du)(A + dA) \\ &= -\rho u A + (\rho + d\rho)(u A + u dA + du A + d u dA) \\ &= -\rho u A + \rho u A + \rho u dA + \rho du A \\ &\quad + d\rho u A + d\rho u dA + A d\rho du \end{aligned}$$

In the limit of $dx \rightarrow 0$, products of $d\rho$, du and dA will go to zero faster.

$$\int_{cs} \rho (\vec{V} \cdot \vec{n}) dA = \int \rho u dA + \rho A du + u A d\rho = d(\rho u A)$$

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Therefore if the equations are written for a single discrete control volume instead of the entire nozzle, then

$$\frac{\partial}{\partial t}(\rho A dx) + d(\rho A u) = 0$$

Assume that $A = A(x)$ only and not $A = A(x, t)$, then,

$$\frac{\partial \rho}{\partial t} (A dx) + d(\rho A u) = 0$$

$$\frac{d\rho}{dt} + \frac{1}{A} \frac{d}{dx}(\rho A u) = 0$$

- Derivation of momentum equation,

$$\frac{\partial}{\partial t} \int_{cv} \rho u dV + \int_{cs} \rho u (\vec{V} \cdot \vec{n}) dA = - \int_{cs} p dA$$

$$\begin{aligned} \rightarrow \frac{\partial}{\partial t} \int_{cv} \rho u dV &= \frac{\partial}{\partial t} \int \rho u A(x) dx \\ &= \left[\frac{\partial}{\partial t} (\rho u) \right] A dx \end{aligned}$$

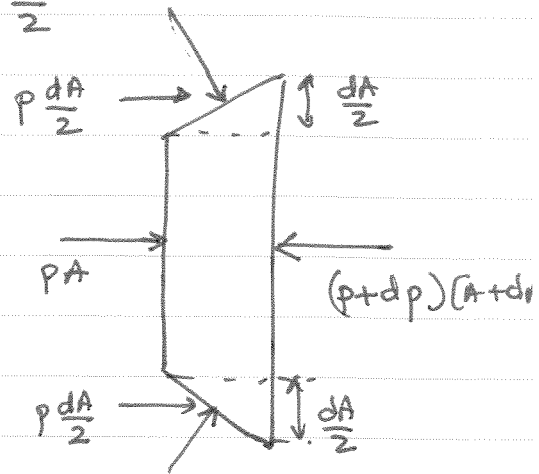
$$\begin{aligned} \rightarrow \int_{cs} \rho u \vec{V} \cdot \vec{n} dA &= \rho u (-u) A + (\rho + d\rho)(u + du)^2 (A + dA) \\ &\approx d(\rho u^2 A) \end{aligned}$$

$$\rightarrow - \int p dA = pA - (p+dp)(A+dA) + 2p \frac{dA}{2}$$

$$= pA - pA - pdA - Adp - dA dp + 2p \frac{dA}{2}$$

$$= -pdA - Adp + pdA$$

$$= -Adp$$



Collecting all terms, we have for a single discrete volume

$$\left[\frac{\partial}{\partial t} (\rho u) \right] A dx + d(\rho u^2 A) = -Adp = -d(pA) + p dA$$

$$\frac{\partial}{\partial t} (\rho u) + \frac{1}{A} \frac{d}{dx} (\rho u^2 A + pA) = \frac{p}{A} \frac{dA}{dx}$$

\rightarrow Derivation of energy equation,

Similarly from abax,

$$\frac{\partial e}{\partial t} + \frac{1}{A} \frac{d}{dx} [(e+p)uA] = 0$$

Then the complete differential form can be written as,

$$\frac{\partial W}{\partial t} + \frac{1}{A} \frac{\partial}{\partial x}(FA) = Q.$$

$$\text{where } W = \begin{bmatrix} p \\ pu \\ e \end{bmatrix}, \quad F = \begin{bmatrix} pu \\ pu^2 + p \\ (e+p)u \end{bmatrix}, \quad Q = \begin{bmatrix} 0 \\ \frac{p}{A} \frac{dA}{dx} \\ 0 \end{bmatrix}$$

$$p = (r-1) \rho \left(\frac{e}{\rho} - \frac{u^2}{2} \right) \quad \text{and} \quad A = A(x)$$

where A is the cross-sectional area as a function of x .

In integral form, we can write the equation as,

$$\frac{\partial W}{\partial t} + \frac{1}{V} \int_{cs} F dA = \frac{1}{V} \int_{cv} Q dV$$

Boundary Conditions for Euler and Navier-Stokes Equations

- To solve for values along inlet and outlet boundaries as well as the farfield, the method of Characteristics can be employed.
- It can be employed for solving hyperbolic sets of equations in two coordinates systems but not three.